

# 1 A new cold cooling system using krypton for the future upgrade of 2 the LHC after the Long Shutdown 4 (LS4)

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9

10 ABSTRACT

11 Future silicon detectors for High Energy Physics Experiments will require operation at lower  
12 temperatures to cope with radiation damage of the sensors and consequent increase of the dark  
13 current, beyond the limit of the current CO<sub>2</sub> evaporative cooling system. This, together with many  
14 other requirements such as mass minimization and high radiation hardness, pushes the need of a new  
15 advanced cooling technology. The new coolant shall be able to approach ultra-low temperatures below  
16 -60 °C, withstand high radiation levels while having cooling lines diameters comparable to the currently  
17 achieved with the CO<sub>2</sub> technology. Among different natural working fluids, krypton appears as a  
18 promising coolant for the thermal management of future detectors in high-irradiated environments.  
19 The thermodynamic properties of krypton do not allow the use of a pumped loop cycle but rather  
20 impose a need of a novel cooling technology. A new ejector-supported krypton cycle is presented,  
21 highlighting the cycle dynamics involved due to the different temperature levels normally encountered  
22 during the detector lifetime.

23 Keywords: Krypton, CO<sub>2</sub>, Detector, Supercritical, Ejector, Cooling

## 24 1. Introduction

25 Silicon detector trackers are used in High-Energy Physics (HEP) experiments to track the path and  
26 momentum of particles created by the collisions inside the beam pipe. Those sensors are uniformly  
27 distributed along the detector volume. Construction material used around the detector must be  
28 tolerant of ionizing radiation and of low mass density. The wanted signals from charged particles  
29 crossing those sensors are very short pulses of current. The dark current of the sensors has to be kept  
30 low by low operating temperatures and the heat generated by the dark current in the sensor and the  
31 heat of surrounding electronics needs to be removed by the cooling system to maintain the sensors  
32 thermally stable after radiation damage, preventing the phenomena of thermal runaway [1]. Different  
33 generation of detectors have required over the years a continuous upgrade of the cooling system  
34 according to the thermal requirements. In the 1990s, single phase cooling with water under sub-  
35 atmospheric pressures was used but drawbacks such as water freezing point and low efficiency in

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36 terms of heat removal capability of the fluid made two-phase cooling the preferred choice [2].  
37 Alternatively, mixture of water and glycol was also considered.

38 One advantage of evaporative cooling is the isothermal extraction of the heat (the two-phase fluid  
39 temperature does not change other than due to pressure drops along the cooling pipe). In the ATLAS  
40 ID an evaporative cooling system based on perfluorocarbons  $C_3F_8$  [3] is used, which provides low  
41 evaporation temperature with a low pressure fluid, while presenting a high radiation hardness. Low-  
42 operating pressure allows for thinner tube wall, reducing the mass but as drawback the tube size is  
43 increased to limit the pressure drops and related temperature drops which are amplified by the low  
44 reduced pressure. The system was firstly based on a compression driven cycle before being changed  
45 to a gravity-driven thermosyphon cycle [4]. The latter has the advantage that it avoids the challenges  
46 deriving from employment of oil-free compression, which normally requires much more maintenance  
47 than oil-lubricated machines. In contrast, single-phase cooling using  $C_6F_{14}$  was used for the CMS  
48 detector tracker via a pumped loop cycle since it was considered more robust than relying on a large  
49 two-phase system for which there existed little experience at the time. Despite the high-radiation  
50 hardness of  $C_3F_8$ ,  $CO_2$  has proven to have excellent thermal performance in small-diameter-tube  
51 evaporators which is extremely welcome for low-mass detector design [5]. A mechanically pumped  
52 loop concept with a two-phase accumulator was developed at NIKHEF [6] for the Alpha Magnetic  
53 Spectrometer (AMS) for the International Space Station (ISS) and used in the Vertex Locator (Velo) of  
54 the LHCb detector [7]. The suitability of  $CO_2$  to withstand high radiation with excellent thermophysical  
55 properties, as well as ensuring a remote control of the detector operational temperature, have led to  
56 the adoption of  $CO_2$  evaporative cooling systems in other detectors at CERN, CMS Pixel detector [8]  
57 and the ATLAS IBL tracking layer [9]. However, the freezing point of  $CO_2$  and the subcooling required  
58 at the pump inlet limit the use of the 2PACL for the ultra-low temperature range [10].

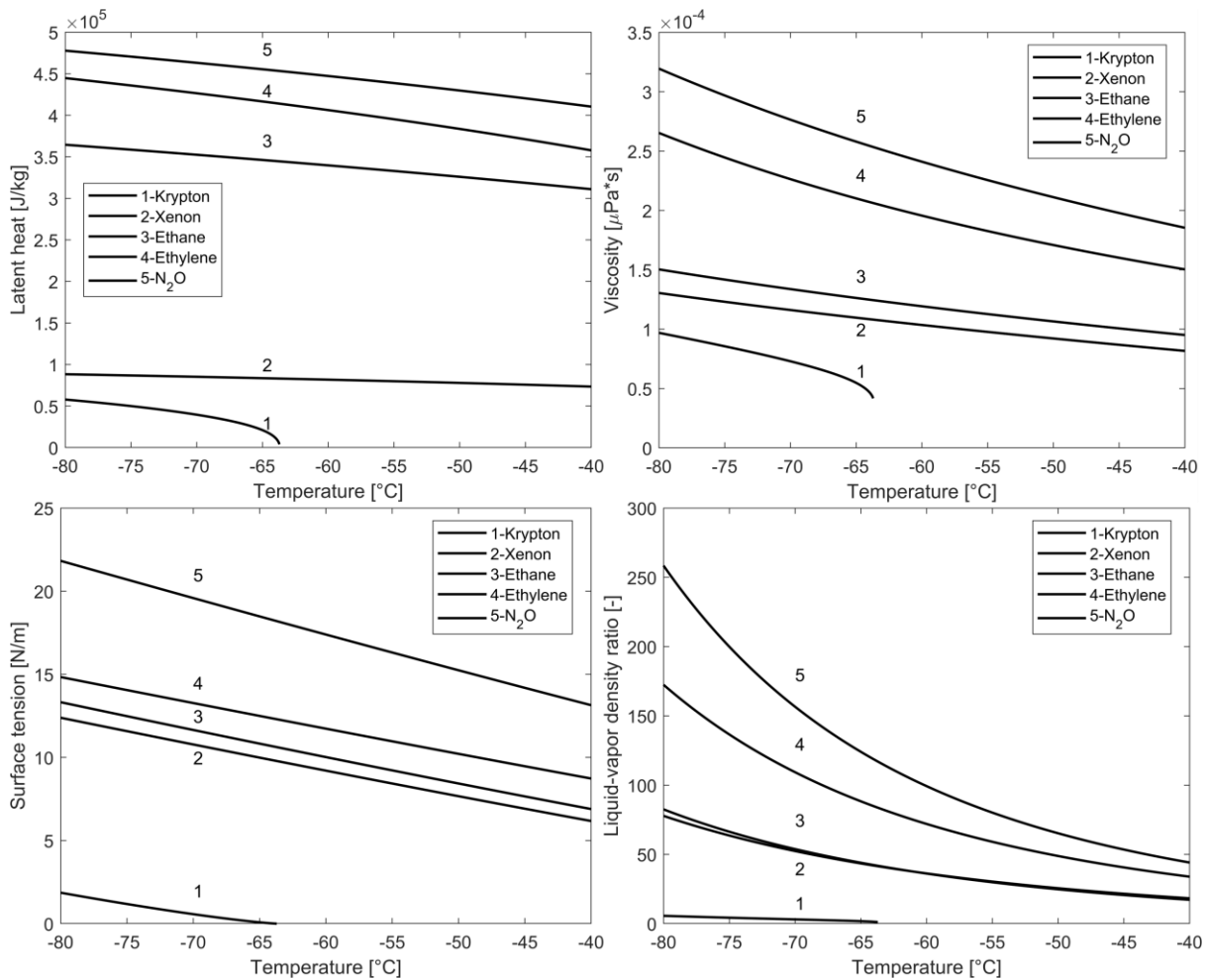
59 Since  $CO_2$  cannot be used at these very low evaporation temperatures, a new environmental and  
60 performant coolant must be identified [11,12]. In this article different natural working fluids are  
61 compared, considering relevant thermophysical properties and their thermal performance. Among  
62 different candidates, krypton is a promising candidate and its thermophysical properties have been  
63 studied to design and propose a suitable cooling technology. The new cooling concept developed  
64 allows to provide cooling either in supercritical cold conditions or under flow boiling at the detector.  
65 Even though the control strategy presented is designed primarily for detector cooling, the working  
66 principle can be extended to any other cooling system operating in the supercritical area which  
67 requires a slow and controlled cooling in case the object to be cooled is sensitive to fast temperature  
68 gradients that could potentially harm its integrity.

## 69 **2. Challenges for cooling of silicon detectors in High Energy Physics**

70 Detector cooling presents peculiarities that go beyond any conventional refrigeration system in terms  
71 of available space, reliability, maintenance, amount of heat dissipated and maximum  $\Delta T$  allowed  
72 between inlet-outlet. The hardware for cooling inside the detector must be minimized through  
73 efficient system-level solution. In this respect, the use of a working fluid enabling the integration of  
74 small cooling channels while concurrently preserving high thermal performance throughout the pipe,  
75 is of primary importance. The trackers require long cooling lines to reach all the heat sources.  
76 Refrigerants should have low global warming potential and be radiation hard. It is known that  
77 traditional fluids containing hydrogen in their molecular composition could decompose due to the  
78 phenomena of molecular chain breaking [13]. A range of working fluids, comprising nitrous oxide,  
79 ethane and ethylene, and the noble gases krypton and xenon, has been compared for the ultra-low

80 temperature range. Potential interesting mixtures as CO<sub>2</sub>+N<sub>2</sub>O have not been considered since their  
81 thermal performance will not differ substantially from that of pure fluids.

82 Nitrous oxide has similar properties to CO<sub>2</sub> but a lower freezing point, making it suitable to be used as  
83 a freezing depressant [14]. However, the operating pressure remains extremely low leading to poor  
84 thermal performance due to the high dependency of temperature on the vapor pressure at low  
85 reduced pressure, which in turns would negatively affect the temperature of the sensors according to  
86 the heat path of the fluid. Furthermore, N<sub>2</sub>O is an oxidizing agent which can decompose explosively  
87 under specific conditions and therefore its use in pure form is not recommended [15]. A mixture of  
88 CO<sub>2</sub> and N<sub>2</sub>O can alleviate this issue but on the other hand the minimum possible operating  
89 temperature constrained by the freezing point rises as a function of the CO<sub>2</sub> concentration in the  
90 mixture. The hydrocarbons ethane and ethylene present interesting thermal properties but they are  
91 flammable, limiting their use due to safety concerns. In addition, if a further extension of the operating  
92 envelope towards colder temperatures is required, these coolants have in the normal boiling point  
93 (NBP) their limitation (-88.58°C for ethane and -103.8°C for ethylene). Operating below the NBP would  
94 require working under vacuum conditions and increases the risk of air infiltration into the system  
95 causing performance degradation. In case of an air infiltration into the system, the presence of non-  
96 condensable gases (i.e. nitrogen) can partially clog capillary tubes, creating intermittent unbalance of  
97 the mass flow rates along the cycle and increasing the power consumption of the system due to sudden  
98 reduction of the cooling capacity. If this happens, the system needs to be stopped and the entire  
99 refrigerant charge recovered before proceeding with the new filling. This points out the importance of  
100 the NBP as limiting factor in the fluid selection. The noble gases xenon and especially krypton have low  
101 NBP and they are the most stable elements in nature, which is certainly an attractive quality [16]. It is  
102 worth to mention that CO<sub>2</sub> is a fluid with unique physical properties: the other fluids previously  
103 mentioned have a freezing point at very low pressures, significantly below the atmospheric pressure.  
104 Nonetheless, when considering thermal performance, it is crucial to focus on the NBP of these fluids.  
105 The thermal performance comparison of different fluids will be based on relevant thermophysical  
106 properties such as latent heat of vaporization, viscosity, surface tension and liquid-vapor density ratio  
107 (Figure 1).



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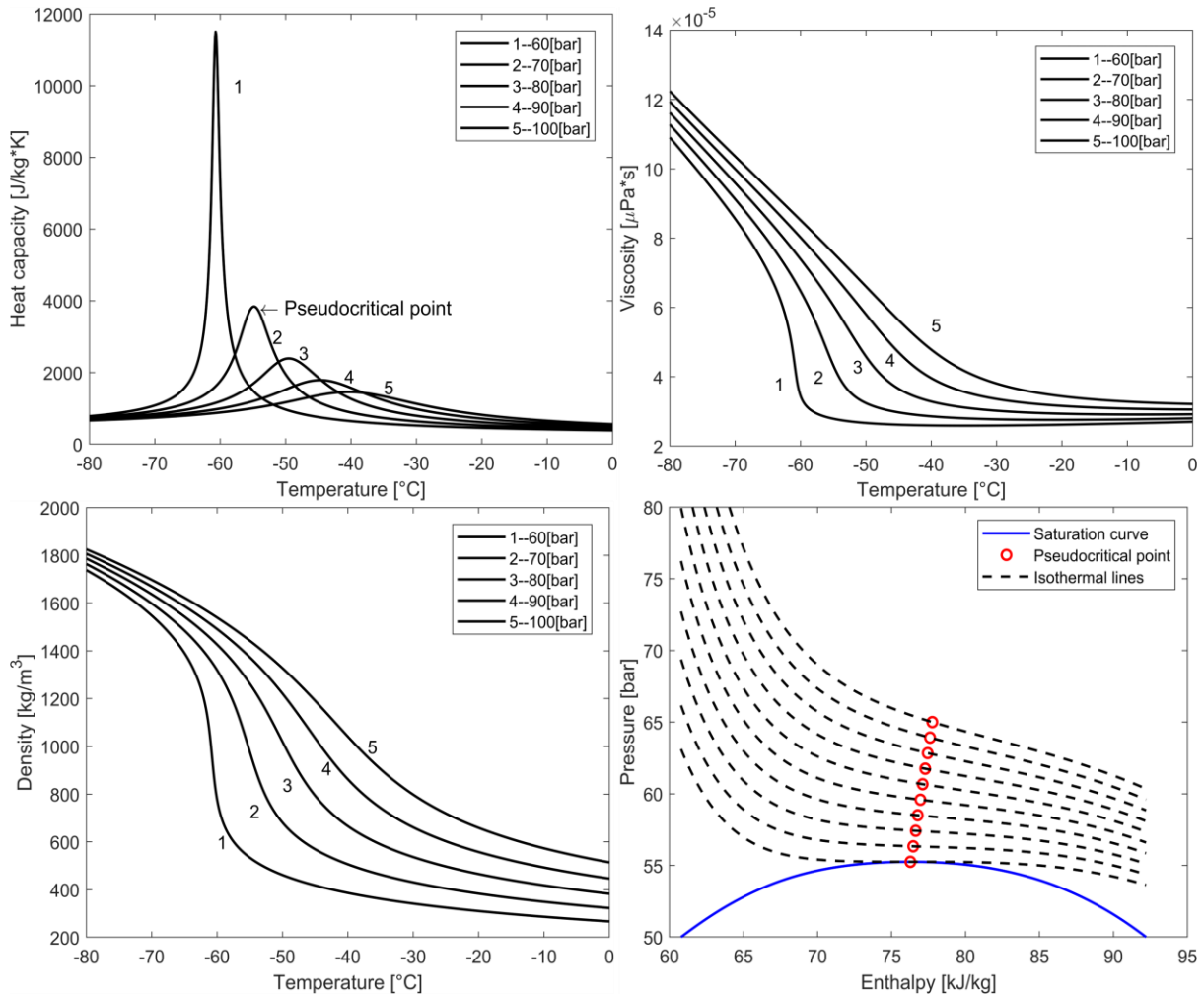
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**Figure 1: Relevant thermodynamic properties of the fluids investigated.**

110 Latent heat represents the total amount of heat that is absorbed by a fluid before completely turning  
 111 into vapor. A larger latent heat is normally preferred since the flow can be reduced and consequently  
 112 the associated pressure gradients are smaller. The dynamic viscosity is a measure of the resistance of  
 113 a fluid to flow, while surface tension indicates the forces holding the molecules together that must be  
 114 overcome to initiate the boiling process. The ratio of liquid to vapor density is more complex: in two-  
 115 phase flow a low-density ratio promotes a more homogenous flow, impacting flow pattern and heat  
 116 transfer. More importantly, a high-density ratio causes larger pressure drops between inlet and outlet,  
 117 causing an uneven temperature distribution. Therefore, having a fluid with high vapor density results  
 118 in lower temperature drops for the same pressure drops.

119 In the range between -60 to -80 °C [12], it is important to distinguish between the two-phase and  
 120 supercritical state of krypton which has a critical temperature around -64 °C. Above this temperature,  
 121 sharp changes in fluid density as well as in the heat capacity occur. Two-phase cooling at high pressures  
 122 presents extremely high heat transfer coefficients and small volume thanks to gas compression whilst  
 123 supercritical fluids near the critical point present smaller values of heat transfer [17] but low pressure  
 124 drops thanks to low viscosity levels. However, the advantages of two-phase cooling are strongly  
 125 dependent on operating pressure and molecular weight of the fluid. Supercritical fluids are especially  
 126 interesting because of the high heat capacity and low resistance to flow in proximity of the  
 127 pseudocritical points. A pseudocritical point is the thermodynamic condition given by a temperature  
 128 and pressure above the critical values at which a maximum of specific heat capacity is registered. Peaks  
 129 in specific heat capacity and thermal conductivity quickly decrease as we move away from the critical

130 point. The temperature distribution of a fluid can be understood through its specific heat capacity.  
 131 When the specific heat capacity is larger, the behavior of a supercritical fluid resembles that of a two-  
 132 phase fluid thanks to a nearly constant temperature profile. For cooling of detector trackers where  
 133 heat transfer performances are crucial for the thermal management of the sensors, working in  
 134 proximity of those pseudocritical points seems beneficial.



135  
 136 **Figure 2: Thermodynamic properties of supercritical krypton at different pressures and temperatures.**

137 This mono-phase region is not only interesting in mini-channel applications but also for micro-  
 138 channels. In two-phase systems using micro-channels, fluid resistance given by micro-orifices are  
 139 required at the channel inlet to promote an even flow distribution and suppress flow instabilities. This  
 140 challenge is of a great importance to avoid risk of dry-out considering that micro-channels are much  
 141 more susceptible to flow maldistribution than mini-channels. Two-phase flow in micro-channels  
 142 imperatively requires collecting the flow exiting each channel in the outlet manifold:  
 143 the hindrance posed by the two-phase state of the flow for further flow distribution in the subsequent  
 144 channel becomes feasible by employing a supercritical fluid. In supercritical state, all the flow can pass  
 145 by multiple chips in series rather than being distributed through multiple channels in parallel [18] as  
 146 would occur in a two-phase distribution system. The thermal budget in terms of  $\Delta T$ - $\Delta P$  would pose the  
 147 number of how many chips can be efficiently cooled by one single stream. Although supercritical fluids  
 148 could offer a remarkable alternative to two-phase cooling while preserving the advantages of single-  
 149 phase fluids, their characterization is still far from the well-known two-phase area. For this reason, the  
 150 natural working fluids studied here are all compared under boiling conditions.

### 151 3. Fluid study comparison

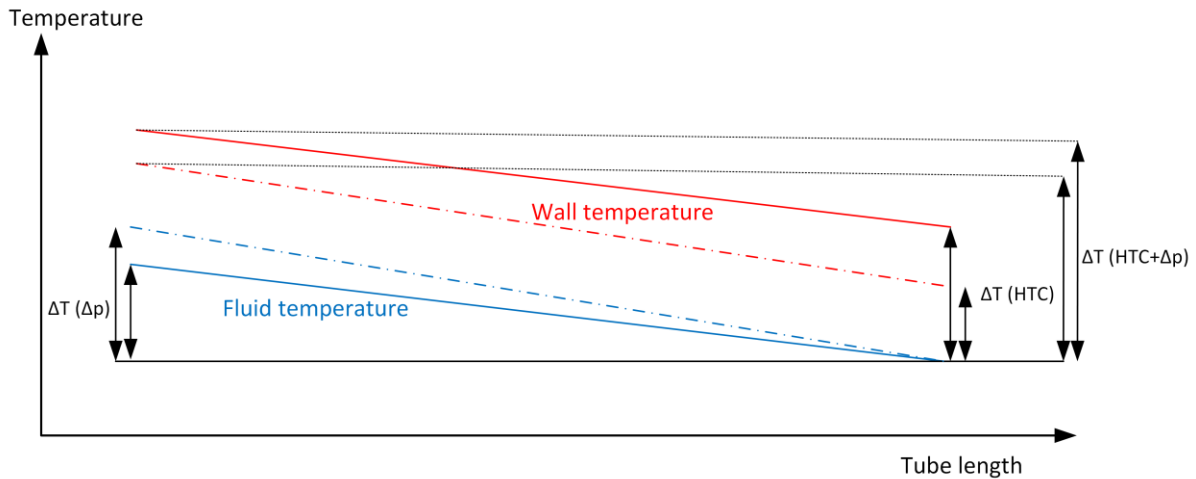
152 Although the optimization of the cooling system comprises the fluid but also the support of the cooling  
153 structure in which the fluid is embedded, a single fluid-based approach has been used here [13]. The  
154 optimal physical performance of the detector requires the mass minimization of the full structure  
155 around it. Mass minimization normally refers to a parameter called radiation length [19] which is a  
156 scale of length for the degradation of particle trajectories due to scattering and radiation. In principle  
157 this means that a smaller cooling tube does not necessarily lead to the best scenario if the missing  
158 volume is replaced by a material with a shorter radiation length. The simplest way to identify the most  
159 promising coolant in HEP is to use a non-conventional definition of the thermal performance of a fluid:  
160 the volumetric heat transfer coefficient as defined in Eq. (1).

$$161 \quad VHTC = \frac{Q}{Volume * (\Delta T(\Delta p) + \Delta T(HTC))} \quad (1)$$

162 Unlike the conventional definition of the heat transfer coefficient that measures locally the thermal  
163 performance of the fluid, the volumetric heat transfer coefficient is a scan of the overall performance  
164 of the fluid for a given operating condition and tube geometry. It combines important requirements  
165 for the detector that need to be fulfilled to achieve an efficient cooling:

- 166 • low temperature losses on the fluid side, as a function of the fluid pressure drops and therefore  
167 described via the term  $\Delta T(\Delta p)$ .
- 168 • Minimization of the Thermal Figure of Merit (TFM), a parameter normally used within the  
169 detector community to quantify the efficiency of the conductive (through tube wall and  
170 materials surrounding the cooling pipe) and convective (fluid dependent) heat transfer  
171 processes. High heat transfer coefficients make the applied cooling method more efficient,  
172 contributing to the minimization of convective term described in the VHTC as  $\Delta T(HTC)$ . The  
173 conductive term is a function of the thermal impedance of the passive resistance of the  
174 support structure.

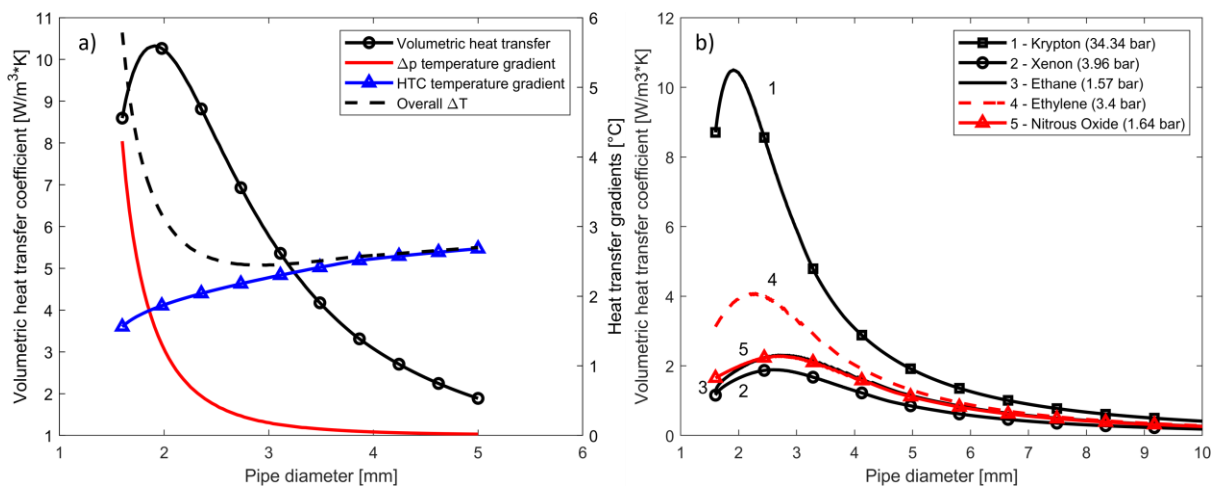
175 The warmest point during flow boiling is normally located at the entrance of the pipe where the fluid  
176 temperature and pressure is highest. The tube diameter has an impact on longitudinal gradients in  
177 terms of pressure and temperature, which in turns affect the heat transfer coefficient and the  
178 temperature gradient along the heat path sensor-cooling tube (transversal gradient).



179

180 **Figure 3: Typical temperature distribution along a two-phase cooling tube where temperature gradients with**  
 181 **respect to pressure drops and local heat transfer coefficients are illustrated (dotted lines illustrate the case**  
 182 **with reduced diameter).**

183 The volumetric heat transfer has been calculated using MATLAB [20], where correlations of heat  
 184 transfer (Kandlikar [21] and pressure drop (Friedel [22]) have been implemented. The outlet pressure  
 185 and quality are fixed, therefore requiring an iterative solution to find the mass flow rate entering the  
 186 detector under saturated conditions. Figure 4 gives a graphical representation of the volumetric heat  
 187 transfer coefficient considering a typical detector cooling pipe: the optimum tube diameter can be  
 188 calculated as a trade-off between an increase of the flow speed and larger pressure gradients.

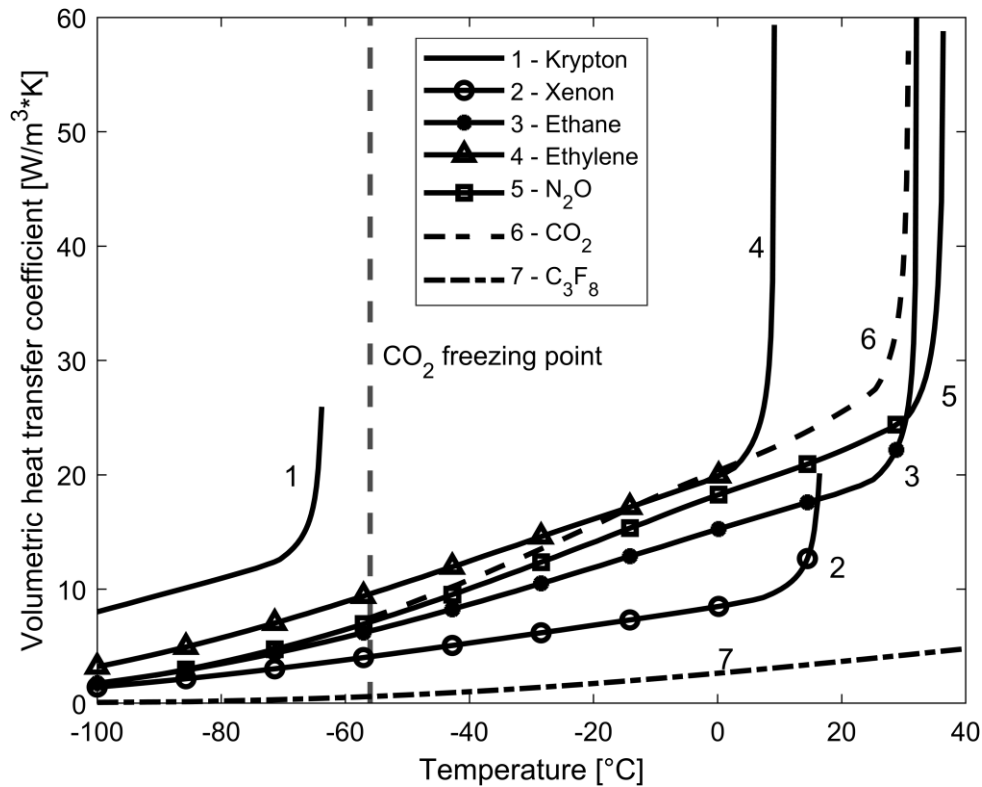


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190 **Figure 4: Cooling tube performance optimization for krypton (a) & comparison of the VHTC of the different**  
 191 **fluids investigated (b) considering standard detector geometry (length = 2[m], Q = 200[W], outlet vapor**  
 192 **quality = 35%, T = -80 °C)).**

193 In the same manner, all the other fluids have been compared. The noble gas krypton outperforms the  
 194 other candidates showing a peak in the volumetric heat transfer around a diameter of 2 mm, in the  
 195 same order of what is currently used in the CO<sub>2</sub> cooling system [23]. The main reason is the operating  
 196 pressure: at higher pressures the temperature drops are less sensitive to pressure drops, the vapor  
 197 phase remains more compressed allowing a volume reduction of the pipe. Other potential candidates,  
 198 especially N<sub>2</sub>O and Xenon, are low-pressure fluids and thus extremely sensitive to pressure drops. Their  
 199 optimal diameters are larger, and they have a lower heat transfer maximum. Figure 5 shows the  
 200 estimated volumetric heat transfer coefficients for a wide range of temperatures, including the warmer

201 range where CO<sub>2</sub> is identified as the most performant and neutral (not flammable and toxic) coolant.  
 202 The perfluorocarbon C<sub>3</sub>F<sub>8</sub> has much worse thermal performance compared to all the others, while  
 203 being a banned fluid nowadays. It is worth to notice that the maximum performance is registered in  
 204 proximity of the critical point where the specific heat capacity is maximum, despite the low latent heat  
 205 which tends to zero at the critical point. For the ultra-low temperature range krypton stands out as the  
 206 best candidate for thermal management of future detectors.

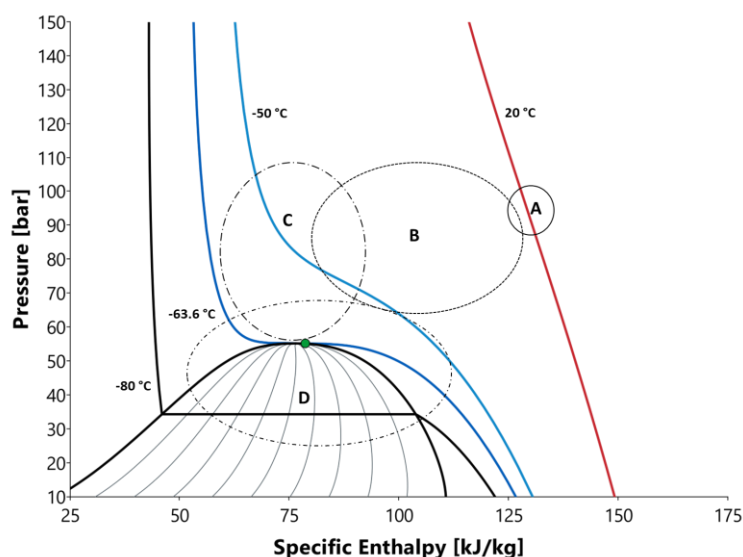


207  
 208 **Figure 5: Comparison VHTC for different fluids over a wide range of temperatures.**

209 **4. Challenges with krypton cooling**

210 As demonstrated above, the noble gas krypton outperforms all the other candidates in the  
 211 temperature range of interest. Regardless of the associated cooling system, silicon detector trackers  
 212 require during their lifetime to be kept at different temperature levels. For instance, during the  
 213 commissioning phase, cooling around ambient conditions is needed. Therefore, the cooling system  
 214 must be able to cool down the trackers under all intermediate temperature levels in a stable manner,  
 215 while guaranteeing to remove the heat dissipated from the sensors that can vary from full load to no  
 216 load according to the type of detector. Those constraints, together with the thermophysical properties  
 217 of krypton, pose challenges never experienced before both with the old cooling system using C<sub>3</sub>F<sub>8</sub> [3]  
 218 and the current 2PACL [24] using CO<sub>2</sub>. Indeed, those refrigerants at ambient conditions are in liquid  
 219 phase, or vapor or two-phase state according to the charge of fluid in the system. Figure 6 represents  
 220 the pressure-enthalpy diagram of krypton: at ambient conditions it is in gas/supercritical state  
 221 according to the the amount of refrigerant charge in the system, where there is no distinction between  
 222 phases (liquid-vapor).





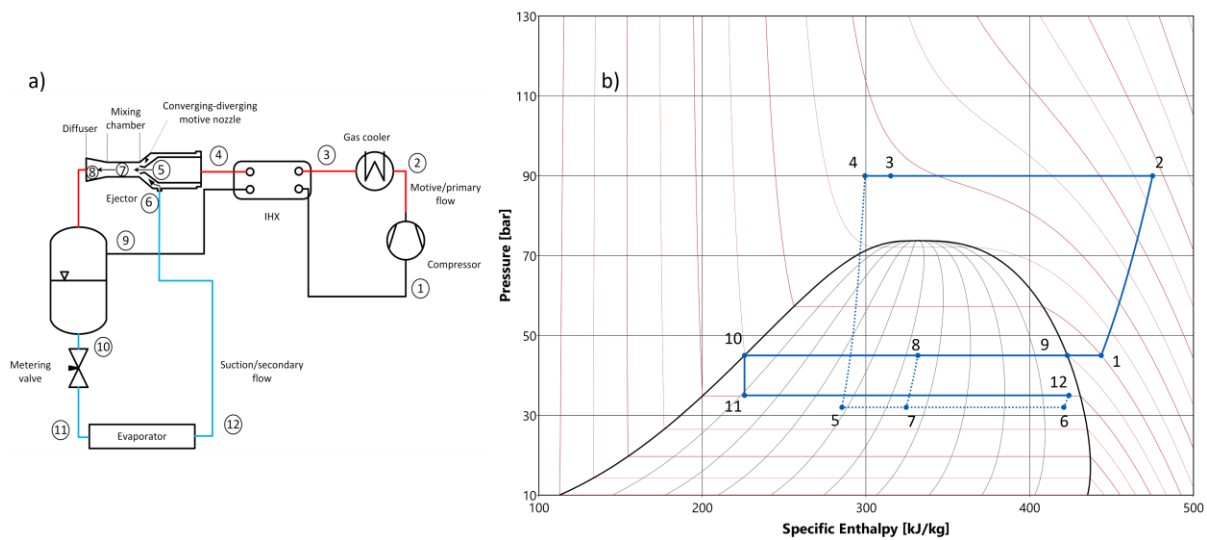
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224 **Figure 6: Pressure-enthalpy diagram of krypton, highlighting the most important isothermal lines and the**  
 225 **different working regimes encountered during the detector lifetime. The zones A-D are explained in the text.**

226 At room temperature (zone A), the fluid behaves as gas and it does not allow the use of a pumped loop  
 227 cycle as currently in use for the CO<sub>2</sub> cooling. The use of a vapor compression system poses many  
 228 challenges: firstly, the system must rely on oil-free machines such as turbocompressors, since it is very  
 229 hard to avoid oil contamination in the refrigerant going through the detectors. If this happens, under  
 230 strong irradiation the oil droplets could polymerize potentially clogging the cooling lines or produce  
 231 corrosive compounds. Secondly, designing a turbomachine for a wide range of operating conditions is  
 232 challenging, mainly due to density changes while moving from the warm to the cold state. At last,  
 233 during the startup, the activation of the compressor may cause a thermal shock in the detector. Indeed,  
 234 in any refrigeration system the compressor startup causes a rapid decrease of the suction pressure  
 235 which is associated with a temperature drop. In a detector application this may lead to undesirable  
 236 fluctuations of the fluid temperature entering the detector. As common practice in particle detectors,  
 237 a gradual and controlled cooldown of about 1 K/min is desirable [13]. Figure 6 highlights all the  
 238 transient scenarios encountered by the detector: start-up (A), supercritical cooldown (B), supercritical  
 239 operation (C), transcritical operation (D) including the transition mode between supercritical to two-  
 240 phase mode. Different working envelopes involve different control strategies (always prioritizing the  
 241 detectors), due to the significant distinction between the supercritical and two-phase states. In the  
 242 supercritical zone, the fluid behaves as a single-phase fluid, whereas in the two-phase state, there is a  
 243 coexistence of liquid and vapor phase. As a result, the regulation of pressure is carried out in  
 244 completely different manners. The amount of refrigerant stored in the cycle determines the achievable  
 245 pressure levels. It is common practice in any vapor compression system to have a separator to help  
 246 deliver only vapor to the compressor suction port while liquid to the evaporator section. In the two-  
 247 phase state, the liquid separator functions as a buffer tank, and it is used to manage refrigerant charge  
 248 fluctuations due to variation in system pressures, unsteady operation caused by sudden changes in the  
 249 evaporator load (i.e. detector heat load) and variability of external conditions. Conversely, in the  
 250 supercritical state, adjusting the pressure to the desired level can only be achieved through the  
 251 injection or removal of refrigerant charge from the cycle. In addition, in supercritical state pressure  
 252 and temperature are independent of each other and therefore pressure control does not necessarily  
 253 impose control of temperature.

254 **5. Ejector as flow regulator through the detector**

255 The evaporative cooling system using C<sub>3</sub>F<sub>8</sub> had a similar functionality to that of a standard vapor  
 256 compression cycle. Because of the already explained requirements and preferred working area  
 257 involving low-vapor quality regimes, heaters were placed inside the detector to allow for warm return  
 258 lines, thus eliminating the need of thermal insulation as well as the risk of compressor damage in  
 259 presence of liquid droplets. From an energy point of view, electrical heating is an inefficient way that  
 260 can be overcome by means of an ejector. In a traditional vapor-compression cycle isenthalpic  
 261 expansion is an irreversible process that constantly generates entropy while expanding. For some fluids  
 262 exergetic losses are quite remarkable and they limit the coefficient of performance of the system, as  
 263 occurring in CO<sub>2</sub> systems during transcritical operation [25,26]. A simple and cost-efficient way of using  
 264 the potential energy of the expanding fluid is an ejector [27], being a robust component that involves  
 265 no moving parts (Figure 7).



266  
 267 **Figure 7: System layout of transcritical ejector CO<sub>2</sub> system (a) & representation in the p-h diagram (b).**

268 The fluid exiting the high-pressure gas cooler section (point 4) is normally called primary or motive  
 269 flow, while the entrained flow from the evaporator outlet (point 12) is called secondary or suction flow.  
 270 The primary flow is expanded through a nozzle, accelerating up to sonic conditions (Mach n° = 1) and  
 271 further accelerates to supersonic conditions in the nozzle diverging section (point 5). The increase of  
 272 kinetic energy corresponds to a pressure decrease which by means of a local depressurization zone  
 273 (point 5) drives the flow from the secondary inlet (point 12) into the suction chamber (point 6). The  
 274 two streams are then mixed in the mixing chamber (point 7) where they exchange mass, momentum  
 275 and heat. The flow is later decelerating in the diffuser where there is an increase of static pressure,  
 276 which corresponds to the diverging area located at the ejector outlet. Normally two parameters are  
 277 computed to describe the ejector performance: mass entrainment ratio and pressure lift (Eq. (2)-(3),  
 278 respectively where the subscripts indicate the thermodynamic state referred to Figure 7).

279

280 
$$\theta_m = \frac{\dot{m}_{12}}{\dot{m}_4} \quad (2)$$

281

282 
$$P_{lift} = p_8 - p_7 \quad (3)$$

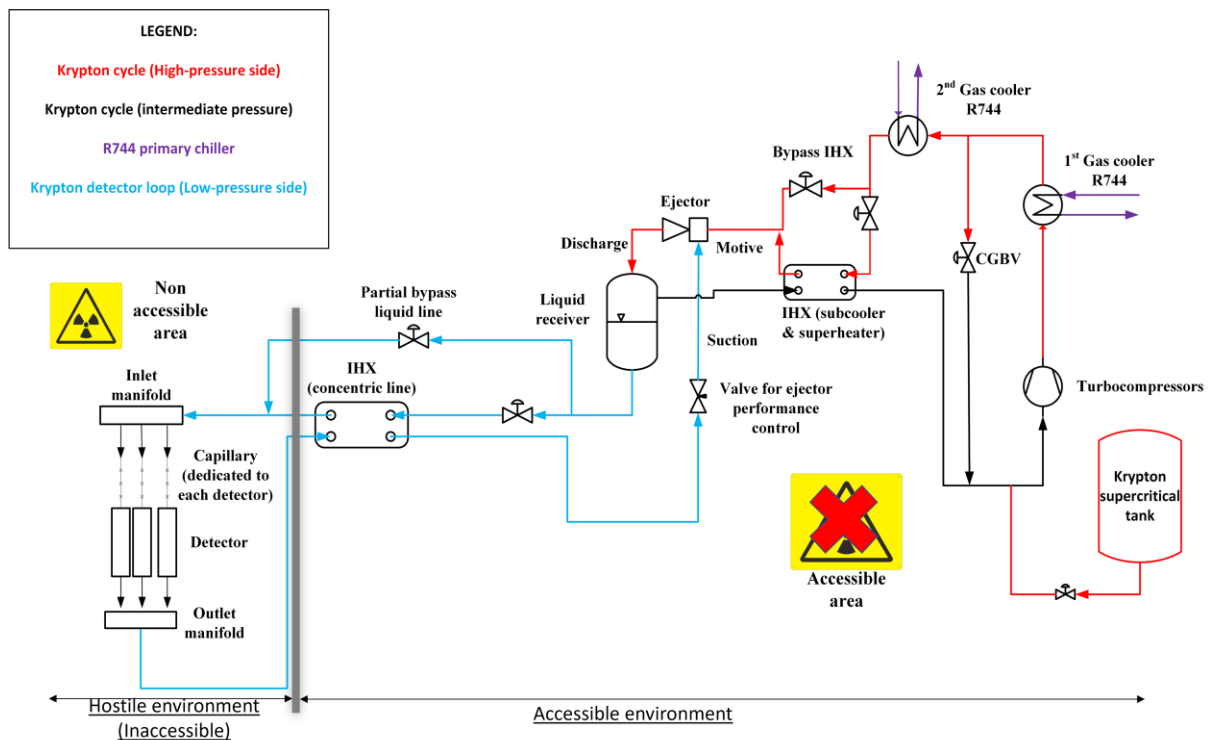
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284 Those two parameters must be considered simultaneously because they measure two separate effects  
285 of the ejector. A given amount of kinetic energy in the motive flow can either be used to pre-compress  
286 a large amount of secondary flow across a small pressure difference or vice versa. A trade-off exists  
287 among those two quantities. However, attention should be given to the controllability of such device:  
288 if the ejector is static, i.e., with constant-geometry motive nozzle, the high-pressure side cannot be  
289 actively controlled, compromising the amount of the secondary flow entrained in part load. As  
290 consequence, the ejector will perform in a suboptimal way.

291 In a properly designed ejector, the convergent-divergent motive nozzle produces subsonic flow in the  
292 convergent section, reaching locally sonic condition at the throat (minimum cross-sectional area) and  
293 further accelerates to supersonic condition in the divergent area. The velocity will increase after the  
294 narrowest point from sonic to supersonic (Mach  $n^\circ > 1$ ) as the flow expands in the diverging section.  
295 The isentropic expansion of the primary flow to supersonic Mach number causes the static pressure  
296 and temperature to decrease from the throat to the pre-mixing chamber, hence the amount of  
297 expansion work defines the exit pressure and temperature. With a fixed geometry ejector it is not  
298 possible to adjust the mass flow rate while maintaining constant motive conditions in terms of  
299 temperature and pressure. When the motive flow reaches critical conditions, the secondary flow  
300 cannot be adjusted by manipulating the downstream pressures (discharge or suction pressures). The  
301 potential to entrain the secondary stream is strongly influenced by the high-pressure control which  
302 could result in no suction flow delivered during part-load operation in case of a static ejector. The  
303 secondary flow, which in this application refers to the krypton flow through the detector, owing to the  
304 entrainment effect can be adjusted in a controlled-geometry ejector using a needle that moves  
305 towards and away from the nozzle throat, regulating the flow by restricting or increasing the flow area  
306 [28,29].

## 307 **6. Cold krypton cycle**

308 The cycle presented here is an ejector-supported krypton cycle where the heat is rejected to a CO<sub>2</sub>  
309 system with the feature of controlling the evaporating level such to avoid excessive high temperature  
310 differences between the two fluids. It consists of a cascade refrigeration system where the high-  
311 temperature circuit is a primary transcritical CO<sub>2</sub> cooling unit, while the low-temperature circuit is the  
312 krypton unit connected to the detectors. It can maintain the refrigerant entering the detector area  
313 either in supercritical cold conditions or in subcritical conditions in subcooled state to ensure that in  
314 this latter case boiling starts at the entrance of the evaporator. Important features of the cycle are  
315 firstly the possibility to extend the temperature range towards colder temperature without a full  
316 upgrade of the cycle, and secondly the design of the cycle is based on the well-known two-phase area  
317 where expertise in the community exist, while the supercritical area has been less explored so far [12].  
318 The simplified piping and instrumentation diagram (P&ID) of the cycle is illustrated below in Figure 8.  
319 The transition between different operating modes (supercritical – transcritical) is achieved by  
320 activating different components of the system in response to changes in the temperature and pressure  
321 level.



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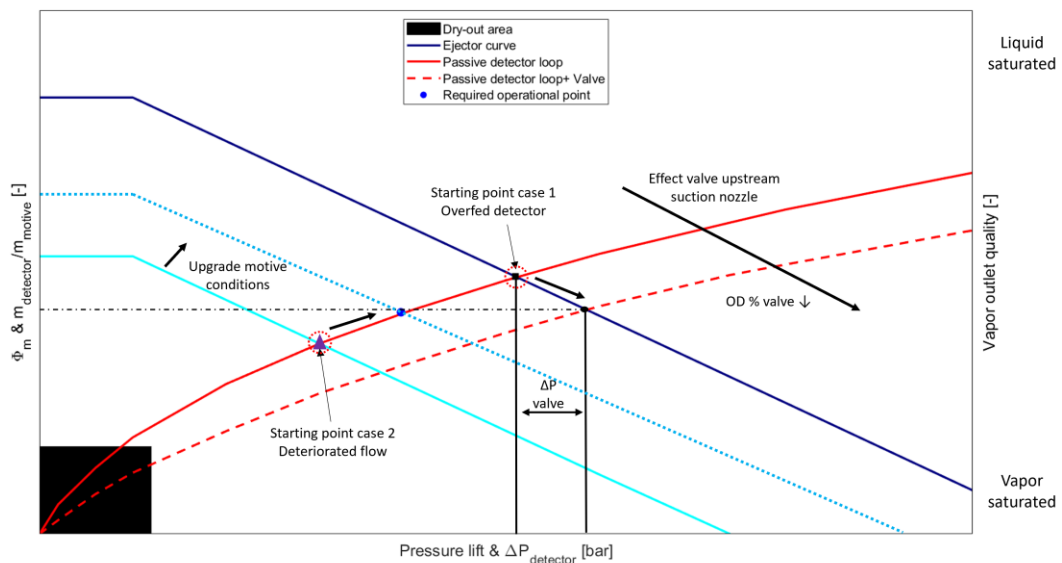
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**Figure 8: Simplified piping and instrumentation diagram (P&ID) of the krypton cooling unit.**

324 The proposed ejector-supported cycle involves three different pressure levels: low (detector),  
 325 intermediate and high-pressure level. The system comprises a compression stage, a gas cooler section  
 326 with a bypass, an internal heat exchanger, a controllable-ejector geometry and a loop used to  
 327 distribute the coolant to the detectors. The bypass (CGBV) downstream the first gas cooler serves as a  
 328 flow regulator in opposition of a variable speed drive compressor, and it is chosen and used here to  
 329 enhance the reliability and stability. A high-pressure tank is also used to condition the system prior to  
 330 startup and to sustain supercritical operation during the gradual cooldown, by further injecting krypton  
 331 into the system. The ejector works as a high-pressure control device while it enables recirculation of  
 332 the cold fluid coming from the detector outlet by using the expansion work available, in a very similar  
 333 manner to a traditional ejector vapor compression cycle [27]. The low-pressure side is very similar to  
 334 what is in use nowadays with CO<sub>2</sub> within the 2PACL system: the long distance is covered by a tube-in-  
 335 tube arrangement to shield the liquid from ambient heating, as well as due to space constraints. This  
 336 counter-flow heat exchanger is needed for conditioning the evaporator inlet flow to a low vapor quality  
 337 during flow boiling operation. Expansion devices such as capillaries are installed at the detector inlet  
 338 to ensure expansion before starting boiling whilst promoting a homogenous distribution of the flow  
 339 through the multiple parallel channels, considering that the heat loads in the individual branches can  
 340 be different, and vary over time. Passive expansion devices must be used to the inaccessibility for  
 341 maintenance in an irradiated environment. Because reliability is one of the main concerns, an  
 342 expansion device is also installed immediately upstream the ejector suction nozzle: changing the  
 343 opening of the valve allows to reduce the flow invoked by the ejector, substantially wasting part of the  
 344 expansion work available in favor of one extra degree of freedom to control flow through the detector.

345 It should be noted that the ejector can potentially lift a large or small amount of flow according to the  
 346 jump in pressure of the low-pressure fluid. In any conventional refrigeration system, trade-off among  
 347 those quantities is controlled via a metering valve (Figure 7, point 10-11). In a detector application,  
 348 passive expansion devices do not allow a regulation of the flow area but rather the pressure drop will  
 349 increase quadratically as the flow increases. This emphasizes the need for a controllable ejector

350 geometry as described in the Figure 9 below, which is developed considering a fixed outlet detector  
351 temperature (-70 °C). The graphical representation of the ejector and detector performance curves  
352 serves only as illustrative example, since the mass flow for a given pressure drop across the detector  
353 can be computed only for a specific geometry of the passive loop. The blue solid curve represents a  
354 typical ejector characteristic curve which remains unchanged under constant boundary conditions at  
355 the motive and suction nozzle. The entrainment ratio remains initially constant before dropping: this  
356 phenomenon is related to double choked conditions in both motive and suction nozzles. The flow  
357 results to be choked when the fluid velocity reaches the speed of sound (sonic conditions) at the throat  
358 location and a further increase of the pressure difference does not lead to any increase of the mass  
359 flow rate. However, this curve will mainly vary according to changes in the motive rather than due to  
360 the suction conditions. The red curve represents the passive loop behavior as a function of the flow  
361 crossing the detector, normalized with respect to the motive flow to scale up the curve and have a fair  
362 comparison. For a specific operating temperature in the detector, one working point (flow) allows to  
363 remove the heat dissipated while guaranteeing an exhaust two-phase flow from the detector with a  
364 vapor content around 35% (indicated by the blue circle). Higher flows than the desired one would  
365 produce an overflow with a drastic increase of the liquid content at the detector outlet while a reduced  
366 flow leads to unstable scenario and possible dry-out (black filled area). The intersection of the detector  
367 and ejector curves represents a stable working point of the cycle, dictated by the ejector performance  
368 in terms of entrainment ratio. When this occurs, the ejector can accurately control the detector outlet  
369 pressure and therefore temperature, being in two-phase state. However, two possible scenarios can  
370 occur during operation: the flow through the detector for any reason (i.e. load change) may need to  
371 be reduced (case 1) or to be increased (case 2). In both cases the ejector is the device used for  
372 conditioning the detector. In the first case, turning down the metering valve installed upstream of the  
373 suction nozzle (Figure 8) increases the pressure lift seen by the ejector inducing a reduction of the flow  
374 entrained. In Figure 9 the new intersection point is given by the same ejector curve and new detector  
375 curve which accounts for the extra  $\Delta P$  introduced by the valve (red dotted line). An excessive closing  
376 of the valve leads to a fast increase of the vapor content due to the reduction of the flow invoked by  
377 the ejector under the same heating power. In the second case the flow needs to be increased and  
378 therefore starting point 2 can be seen as a deteriorated flow scenario. The entrainment potential of  
379 the ejector is enhanced by upgrading the ejector curve mainly varying the motive conditions, especially  
380 in terms of pressure. The performance map is adapted to load change conditions by varying the motive  
381 nozzle entering conditions. This in principle translates to a floating control of the discharge pressure  
382 while simultaneously controlling the receiver pressure via the CGBV to the desired setpoint.



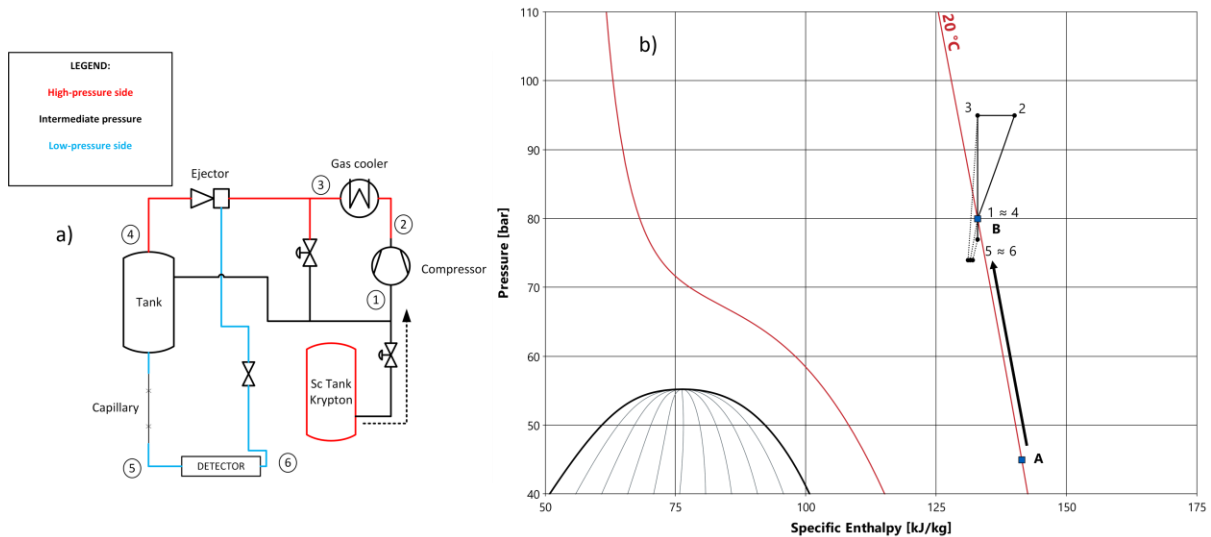
383

384 **Figure 9: Ejector characteristic curve (light blue & blue) & normalized detector curve (red) respect to the motive**  
 385 **flow for a fixed outlet temperature in the detector (-70 °C), considering the detector powered (450 W). The**  
 386 **blue circle represents the desired operational point (flow) in the detector while the black square and purple**  
 387 **triangle illustrate two different initial flow conditions through the detector which require an adjustment of**  
 388 **the valve upstream of the suction nozzle or modulation of the ejector motive conditions (case 1 and 2,**  
 389 **respectively).**

390 **7. Supercritical operation**

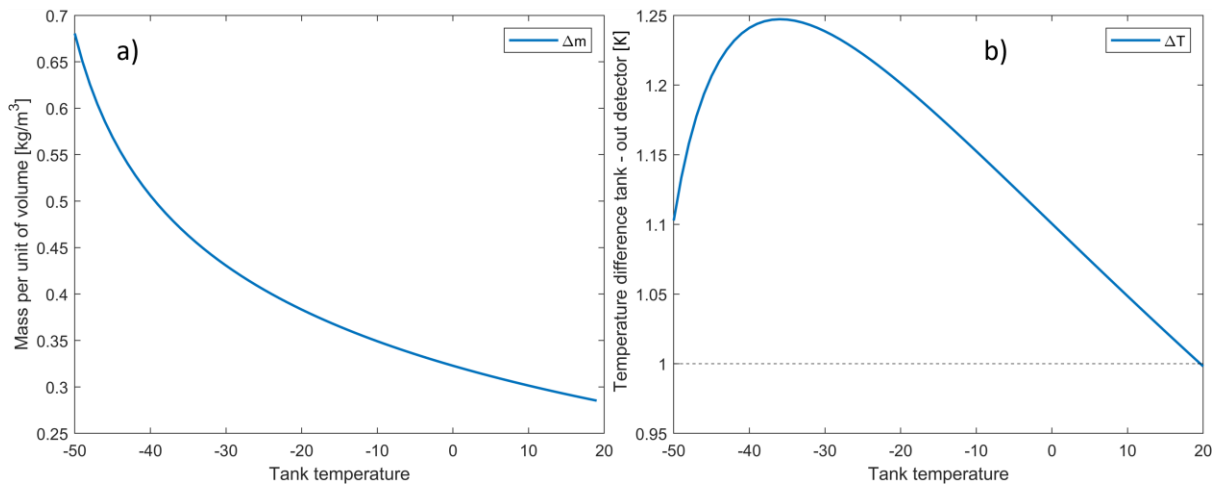
391 The cycle presented above undergoes different scenarios during the detector's lifetime. Those working  
 392 regimes cover both supercritical and transcritical operation, due to the combination of krypton  
 393 properties and gradual cooldown of the detectors. The cycle start-up represents the first challenge  
 394 (Figure 10): detectors are light-weight components with a non-uniform and limited heat capacity over  
 395 all the structure. Therefore, to address a gradual cooldown, it is mandatory to control the fluid  
 396 temperature entering the detectors. The first step is to fill the stagnant loop with an appropriate  
 397 krypton mass inventory to reach the desired starting pressure (from point A to B). An improper charge  
 398 can negatively affect the operation. It can lead to a mismatch in the initial pressure of the system  
 399 compared to the established pressure startup procedure. If the pressure is excessively high additional  
 400 mass is needed, while lower pressures pose a greater risk of fast cooling. The magnitude of the specific  
 401 heat capacity which is described by the distance between the isothermal lines, under lower pressures,  
 402 causes a larger temperature gradient per unit of mass. When the compressor is turned on the krypton  
 403 fluid within the high-pressure leg becomes more dense displacing mass from the intermediate  
 404 pressure side. A high volume ratio dampens this effect, though not entirely. In fact, the increase of the  
 405 high pressure as a function of time is the main factor to consider. The rate at which the pressure  
 406 increases can lead to possible thermal shock scenarios: the compressor discharge temperature would  
 407 increase and the inherent delay of the high-temperature system (CO<sub>2</sub>) in responding to an abrupt load  
 408 change in the gas cooler may not satisfy process constraints, especially concerning the precise  
 409 temperature control on the sensor. Any increase of the cycle pressure if not followed by a proper  
 410 temperature control leads to density change, which in turns corresponds to a mass displacement and  
 411 variation of the intermediate-low pressure levels. This action activates a sequence of transients that  
 412 may bring the cycle out of the wanted operational envelope. Motivated by this, a dedicated control  
 413 logic must be implemented. Heat rejection and high-pressure regulation are coupled to ensure a

414 pressurization of the detector with as little flow as possible throughout all supercritical cooldown  
 415 scenarios. Any injection or withdrawal of mass to/from the system can cause instabilities in terms of  
 416 temperature fluctuations and longer time to achieve steady-state conditions. In this sense, volume  
 417 ratios between the different sections of the cycle are important parameters to be predicted in order  
 418 to provide the best control logic for the inventory control management. The latter refers to the  
 419 common name used to describe the charge control of a CO<sub>2</sub> supercritical Brayton cycle [30], which is  
 420 extended and used here. As a representative example, a simplified representation of the cycle on the  
 421 p-h diagram is illustrated in Figure 10 where the non-active components have been excluded for sake  
 422 of clarity.



423  
 424 **Figure 10: Simplified architecture of the krypton cycle during startup (a) and associated representation in the**  
 425 **pressure-enthalpy diagram (b).**

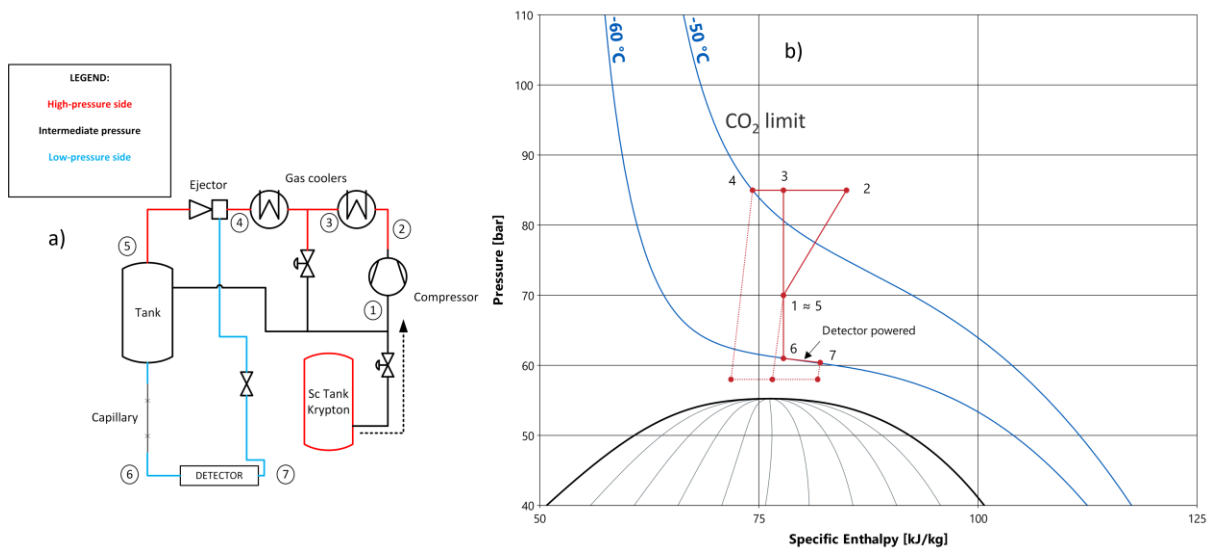
426 After stabilizing the cycle, the supercritical cooldown begins. The high-pressure side is cooled, bringing  
 427 the cycle towards the colder area. The colder the temperature, the denser the fluid becomes. In the  
 428 cycle only one pressure level can be actively controlled without any external loop (i.e. charging tank).  
 429 Although one pressure level is controlled, the remaining two are affected by the remaining mass  
 430 distributed as a function of their volume, pressure and temperature (density). Therefore, to sustain  
 431 the cycle pressure, to avoid a fall into the two-phase area with consequent thermal shock, the high-  
 432 pressure tank can inject mass into the system. Figure 11 (a) illustrates the extra charge required to  
 433 sustain a pressure of 70 bar considering a volume of 1 liter. It is assumed that the pressure lift provided  
 434 by the ejector is 1.5 bar between the tank and the detectors, which allows to maintain a small  
 435 temperature gradient during the whole transition (see Figure 11 (b)). Therefore, maintaining a constant  
 436  $\Delta p$  along the loop (tank-suction nozzle ejector) protects the detector from fast overcooling once the  
 437 tank pressure and temperature are well controlled. It is expected that the flow would increase, for the  
 438 same pressure drop, while moving towards the colder area. This can be deduced by the supercritical  
 439 fluid which behaves like a gas at warmer temperatures.



440

441 **Figure 11: Charge upgrade during gradual cooldown of a system volume of 1 liter to maintain 70 bar pressure**  
 442 **(a). Temperature difference between the tank (70 bar) and outlet of the detector while maintaining a fixed**  
 443  **$\Delta p$  of 1.5 bar (b).**

444 During the duration of the supercritical cooldown with no power to be dissipated, only the first stage  
 445 of gas cooling is active. It needs to reject the heat generated by the compressor only. Once the detector  
 446 is powered, the second gas cooler is activated to reach thermal stability of the cycle.



447

448 **Figure 12: Simplified architecture of the krypton cycle during supercritical operation (a) and associated**  
 449 **representation in the pressure-enthalpy diagram (b).**

450

451 The thermodynamic limitation of the cycle is given by the triple point of CO<sub>2</sub>: colder temperatures than  
 452 -50/-55°C are not achievable due to the already low evaporating temperature of the high-temperature  
 453 circuit. Furthermore, when the detector is powered (5-6) different legs of the cycle are characterized  
 454 by different density zones. In the high-pressure side the fluid is denser, and it stores more mass  
 455 compared to the preceding scenario where the detector was unpowered. The extra mass is displaced  
 456 from the intermediate pressure side with a consequent depressurization of the intermediate-low  
 457 pressure level, according to the mass movement towards the high-pressure side. Volume ratios  
 458 between different sections of the cycle are extremely important to anticipate any action to be  
 459 performed, either charging or discharging, they also have a drastic impact on the cycle dynamic. It is

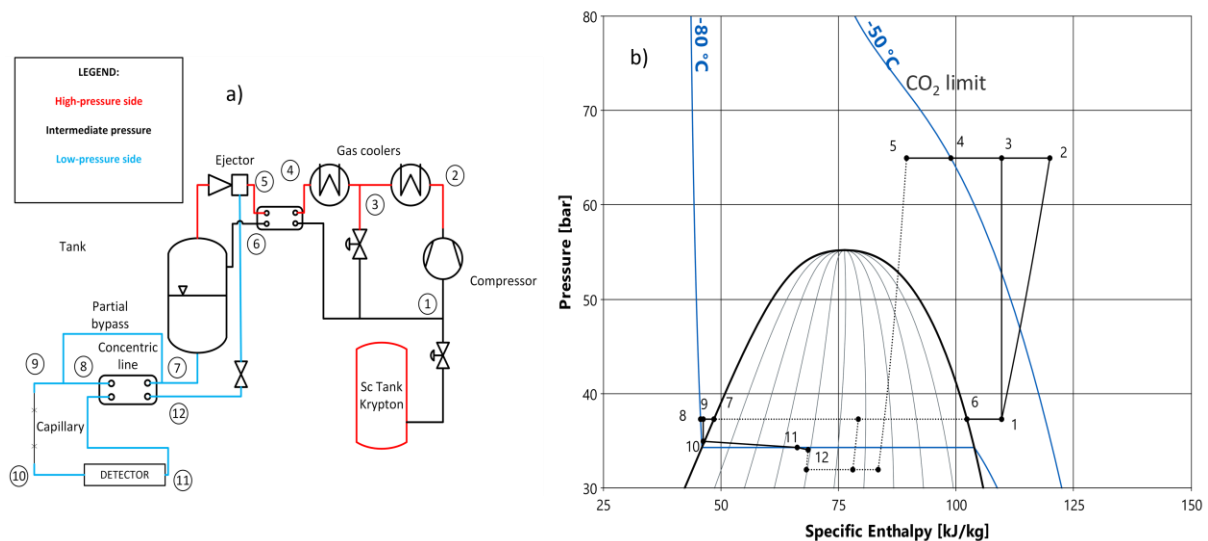


460 unpractical to continuously perform an injection or withdrawal of mass considering the extreme  
 461 variability of the detector load profile. A larger volume on the intermediate-low pressure level can  
 462 alleviate those oscillations but still a slight change of the operational point in the tank could occur. If  
 463 this happens and assuming that the mass stored is insufficient to maintain the tank pressure at the  
 464 desired setpoint, the valve upstream of the suction nozzle can be potentially used to reduce the  
 465 entrained flow by the ejector maintaining almost unchanged the detector setpoint.

466 Detector cooling in the supercritical state resembles a gas heating process: pressure drops along the  
 467 detector are welcome to promote an isothermal process while a larger flow potentially protects the  
 468 detector from warming up. Indeed, if the flow through the detector decreases, the heat picked up by  
 469 the krypton flow would correspond to a larger enthalpy change making the outlet the warmest spot,  
 470 differently than in an evaporative process.

471 **8. Transcritical cycle**

472 If the detector needs colder temperatures, an evaporative cooling method is possible. The cycle must  
 473 migrate from the supercritical to the transcritical area, gradually reaching new steady-state conditions.  
 474 To perform this transition, charge shall be removed from the cycle causing a pressure drop in the  
 475 system. When the supercritical tanks turn into a phase separator, the concept of the cycle aligns with  
 476 that of a traditional ejector vapor compression system. Now the internal heat exchanger and the  
 477 concentric lines are used to deliver superheated vapor (state point 6 -1) and subcooled liquid at the  
 478 capillary inlets (state point 7-8), respectively. A partial bypass of the liquid is required before expanding  
 479 through the capillary due to the compressibility of liquid krypton at high-pressures (state point 9). The  
 480 CGBV (state point 3 – 1) still works as a capacity regulator, in the same manner as a variable speed  
 481 drive compressor would. A parallel modulation of the latter together with the controllable ejector  
 482 geometry ensures a precise control of the detector outlet temperature, as well as of the receiver  
 483 pressure.



484  
 485 **Figure 13: Simplified architecture of the krypton cycle during transcritical operation (a) and associated**  
 486 **representation in the pressure-enthalpy diagram (b).**

## 487 9. Technical challenges for the krypton demonstrator

488 The new cooling concept developed needs to be experimentally tested. In preparation for the  
489 experimental campaign, thermal design and consequent dynamic modelling of the full cycle is required  
490 and extremely helpful to understand the complex dynamics, as well as for improving the robustness  
491 of the necessary control logic to handle the different transient scenarios. The krypton prototype is  
492 currently under construction in Varmeteknisk laboratory at NTNU (Trondheim). However, many  
493 challenges arise from using the rare gas krypton that can be summarized as follows:

- 494 • Noble gases such as neon, krypton, xenon are extremely expensive compared to conventional  
495 fluids, for instance CO<sub>2</sub>. They are harvested exclusively from air as byproduct in large air  
496 separation units via cryogenic distillation of air. Their main use covers window insulation and  
497 lighting (krypton), semiconductors (neon) and as detector material in investigation of dark  
498 matter (xenon) whilst their use as refrigerants has not been explored so far.
- 499 • The cost of krypton is less than Xenon due to higher abundance in the air ( $\approx$  factor 10) but they  
500 both suffer of high-price variations according to the market dynamic and continuous  
501 technological improvements (i.e. LEDs applications do not longer required krypton). The  
502 estimated price in [31] could not anticipate the recent world events and the market is not  
503 transparent. Two major rare gas suppliers were Russia and Ukraine but after Russia's invasion  
504 the supply market was strongly affected, remaining so until alternative producers emerge.  
505 Although Russia is still producing those gases, international sanctions will isolate them from  
506 the global market. Potential alternatives are China and US, while Germany remains the only  
507 reliable supplier in Europe. To understand the magnitude of change for krypton, its price  
508 quadrupled in the first months of 2022 in Japan [32].
- 509 • The cost of krypton and operation in the supercritical state require reduction of the system  
510 volume to limit the refrigerant charge needed to operate at high pressures.
- 511 • As krypton has never been used before in a vapor compression system, specific Krypton-based  
512 components are not available in the market. Therefore, high-pressure CO<sub>2</sub> rated components  
513 such as compressor and gas coolers will be used in the prototype. This constraint affects the  
514 ideal system design which would require the supercritical tank to be the largest volume in the  
515 system. By doing so, rapid changes of temperature and pressure can be alleviated by damping  
516 the gradients through a larger volume, considering that the tank represents the "entering  
517 conditions" to the detector. Even by using a very small CO<sub>2</sub> compressor for extremely low  
518 capacity, the compressor volume could store the largest amount of mass of the system,  
519 strongly impacting the behavior of the cycle. This is normally the opposite in a two-phase  
520 system, where the compressor is operated in the low-density vapor region and large part of  
521 the mass is stored in the buffer tank and condenser (due to the liquid phase).

## 522 10. Theoretical assessment of typical operational point of the krypton cycle

523 As illustrative examples, numerical simulations of the supercritical and transcritical cycle have been  
524 performed to analyze the thermal behavior of the cycle under different operating conditions. The cycle  
525 layout was implemented within the simulation environment Dymola using the TIL-Suite and TILMedia-  
526 Suite, which are a commercial Modelica model library for thermal components-systems ([33,34]) and  
527 a software package for determination of the thermophysical properties [35], respectively. The  
528 assumptions used for the modelling are summarized in Table 1. Considering the multi cooling branch  
529 layout around the detectors, three cooling branches were selected and used in the examples. Details

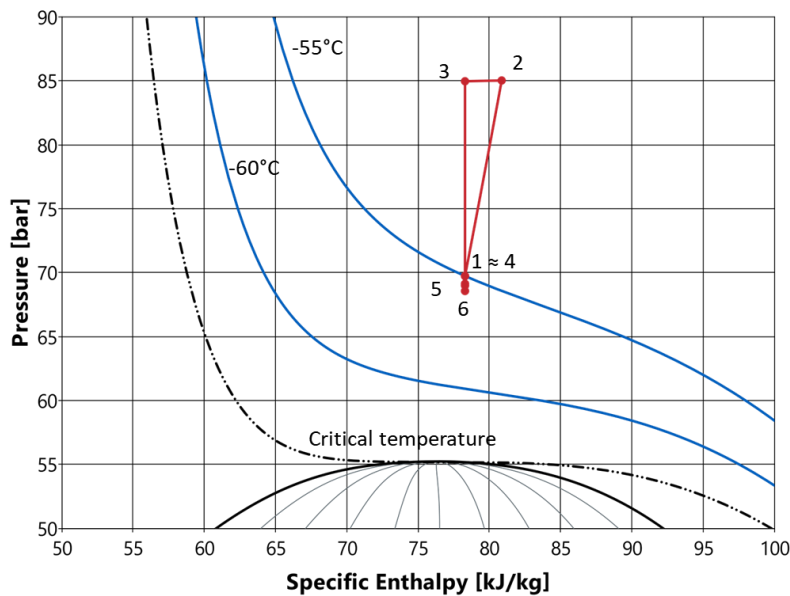
530 of heat transfer and pressure drop correlations, as well as geometrical characterization of the  
 531 components are not reported because it is out of the scope of this work.

532 Table 1: Assumptions used in thermal modelling of the cycle.

Component	Assumption
Compressor	Fixed isentropic and volumetric efficiency (60%)
Ejector	Constant efficiency (25%)
Detector	Load varied in the range 0-150 W
Detector cooling pipe (standard as in use with CO <sub>2</sub> )	Length = 1 [m], inner diameter 2 [mm]

533

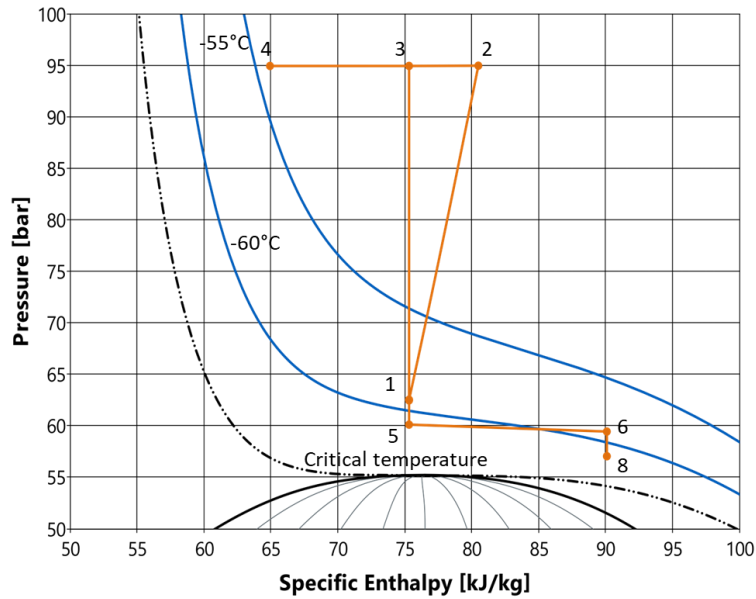
534 The first two simulations investigate supercritical operation. The system was initially simulated without  
 535 heat dissipation from the detector, resembling one of the points encountered during the supercritical  
 536 cooldown. As described in Figure 10, only the first gas cooler is active to dissipate the thermal load  
 537 introduced by the compression. The ejector works as a flow circulator through the detector while  
 538 guaranteeing a minimal change in temperature and pressure along the loop. Figure 14 shows the  
 539 thermodynamic points of the cycle in the pressure-enthalpy diagram.



540

541 **Figure 14: Steady-state result during supercritical operation with detector unpowered. Schematic of the cycle**  
 542 **in the p-h diagram (points corresponds to Figure 10).**

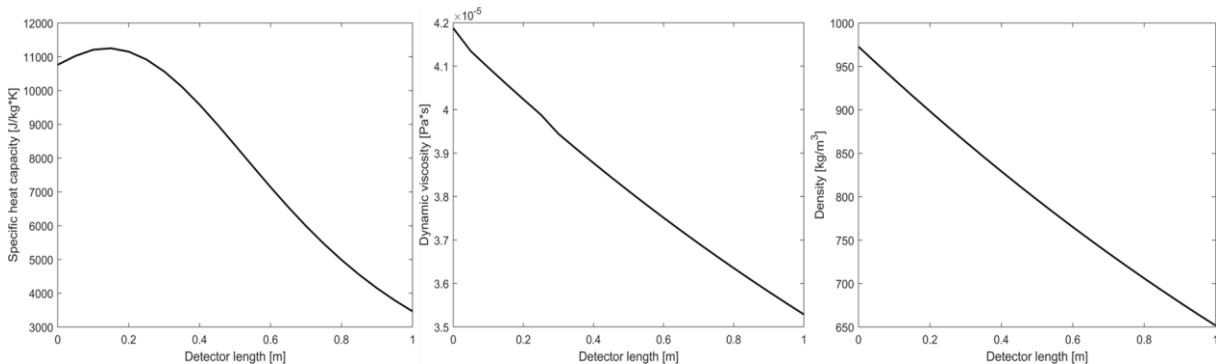
543 Secondly, when the detector is powered the second gas cooler is switched on to dissipate the thermal  
 544 power absorbed (Figure 15). The operational point in the detector has been selected considering one  
 545 of the pseudocritical points illustrated in Figure 2, which can be interpreted as performance map to  
 546 achieve the best thermal performance offered by the fluid in supercritical state. As described earlier,  
 547 gas heating in the supercritical state (points 5-6) may cause a temperature increase between the inlet  
 548 and the outlet, according to the pressure gradient along the detector which helps to follow an almost  
 549 isothermal process.



550

551 **Figure 15: Steady-state result during supercritical operation with detector powered. Schematic of the cycle in**  
 552 **the p-h diagram (points referred to Figure 12) considering a setpoint around -60°C. Additional pressure drops**  
 553 **defined by the points (6-8) correspond to the return line (tube-in-tube) which is bypassed on the inlet side,**  
 554 **due to the single phase state of the fluid.**

555 In the case simulated, the entrainment potential of the ejector is increased by further raising the high  
 556 pressure to cope with the power absorbed by the detector. An overflow through the detector would  
 557 lead to colder fluid temperatures, as well as having an impact on the mass distribution within the  
 558 system. The detector setpoint was chosen to ensure cooling while exploiting the region with high  
 559 values of specific heat capacity (Figure 16). However, a degradation of the specific heat capacity is  
 560 observed by moving further from the inlet. The reason of such degradation is explained at the end of  
 561 this chapter. Low viscosities in the supercritical state are observed: in the region of interest (above the  
 562 critical point) the fluid presents viscosity levels like the gas phase while having high densities typical of  
 563 the liquid phase.

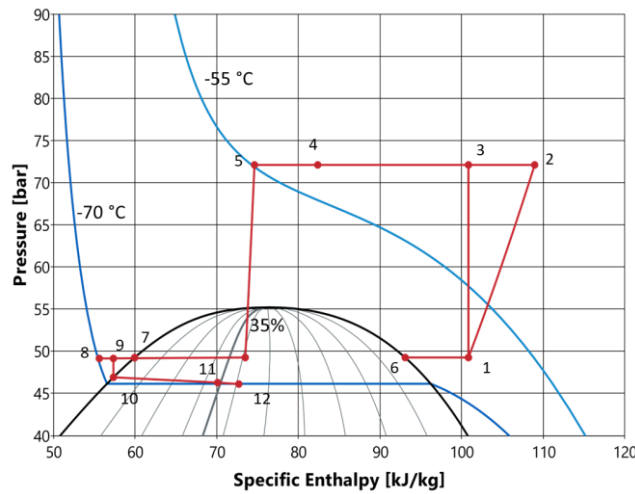


564

565 **Figure 16: Variation of the specific heat capacity, viscosity and density along the detector during gas heating.**

566 At last, the krypton cycle was simulated for the coldest working conditions in the detector (-70°C). This  
 567 scenario refers to the transcritical operation (Figure 17). The main difference compared to a  
 568 conventional ejector cycle is the great amount of liquid in the exhaust two-phase flow at the detector  
 569 outlet. Thermodynamic limitations are given by the CO<sub>2</sub> triple point which limits the lowest heat  
 570 rejection temperature achievable in the second gas cooler. The ejector motive conditions are, for the

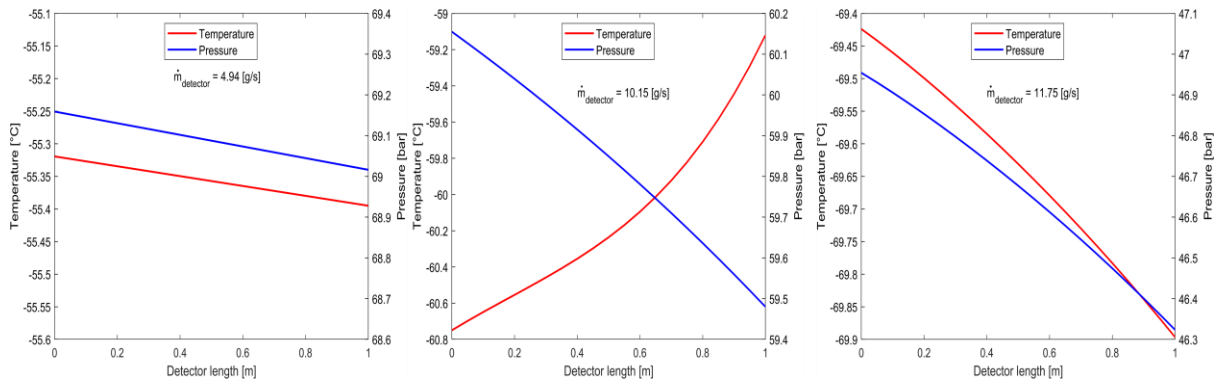
571 reason above, limited to “warmer” temperatures but with the flexibility of adjusting the pressure by  
572 moving the needle installed in front of the motive nozzle throat in and out.



573

574 **Figure 17: Steady-state result during transcritical operation with detector powered with setpoint set to -70°C**  
575 **(outlet vapor quality  $\approx$  35% as required by design). Schematic of the cycle in the p-h diagram (points**  
576 **corresponding to Figure 10).**

577 The difference between supercritical to transcritical operation can be seen in the pressure-  
578 temperature distribution along the detector, as shown in Figure 18. During gradual cooldown of the  
579 detectors (left side), the flow entrained by the ejector is relatively small as cooling is not yet required.  
580 Without power absorbed, the temperature changes by approximately 0.1 K. During gas heating  
581 conditions (supercritical state), the pressure decreases along the pipe length while the temperature  
582 increases by approximately 1.6 K. The detector cooling channel may potentially be designed to  
583 introduce larger pressure drops to follow up the isothermal line throughout the gas heating process.  
584 Typically, a temperature difference of 5 K is allowed between the inlet and the outlet. Considering the  
585 flow boiling conditions occurring at colder temperatures, the performance in supercritical and  
586 transcritical states can be traded off against each other. The pressure does not decrease in a linear  
587 manner as typical for the single-phase state: in supercritical state above the critical point sharp changes  
588 of density and viscosity occur. Overall (in-out) it resembles a two-phase flow behavior near the critical  
589 point. It is worth noticing that in the right plot (flow boiling) the pressure profile lies between a linear  
590 and a parabolic curve. The two-phase correlation used [36] in the evaporator was characterized by a  
591 relatively low (below 2) two-phase multiplier. The Friedel correlation is based on the separated flow  
592 approach wherein the estimation of the two-phase pressure drop involves treating the entire flow as  
593 single-phase liquid while accounting for the larger pressure drops given by the vapor phase through  
594 the two-phase multiplier. High reduced pressure and low vapor quality as required by detector cooling  
595 are the main causes of such behavior. Closer to the critical point vapor and liquid density tend to be  
596 the same: the slip ratio and vapor velocity decreases, resembling a homogeneous vapor-droplet flow.  
597 More importantly, in a more homogenous flow the dependency of pressure drops on the vapor content  
598 is weaker. This improves the equalization of flows in different cooling branches with different heat  
599 loads at high reduced pressures.



600

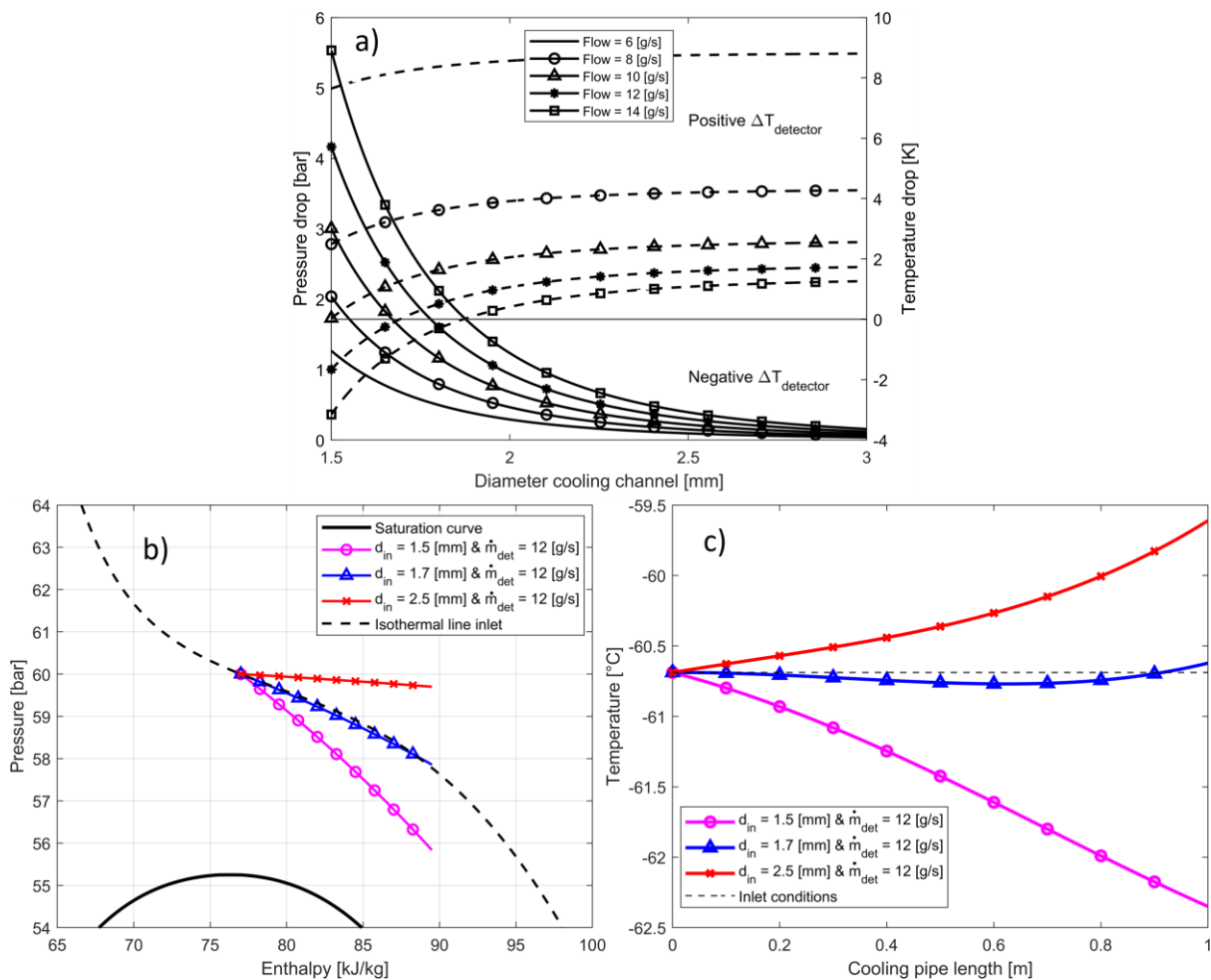
601 **Figure 18: Temperature-pressure distribution in the detector for three different cases: supercritical operation**  
 602 **with detector unpowered and powered (left and middle), flow boiling operation during transcritical**  
 603 **operation (right side).**

604 In section 3, a fluid case study was carried out to identify the optimal tube diameter. The comparison  
 605 was made with respect to the two-phase area: as one specific tube diameter does not lead to the best  
 606 thermal performance over a wide range of reduced pressure, neither does it in supercritical state. If  
 607 then, the detectors need temperatures in the range of -60 down to -80°C, three possible scenarios  
 608 exist. The cooling channel inside the detectors can be optimized based exclusively on the supercritical  
 609 state, on the two-phase area or considering a trade-off between the two cases. Pressure drops which  
 610 are normally unwanted during vaporization of the fluid promote a more homogenous fluid  
 611 temperature profile along the detector during gas heating near the critical point. Nevertheless, for a  
 612 given geometry, the rise of the fluid temperature can be reduced by overflowing the detector,  
 613 corresponding to a smaller enthalpy change. This in turn has an effect at system level: overflowing the  
 614 detector means an increase of the total pressure lift provided by the ejector due to the fixed fluid  
 615 resistance introduced by the capillary upstream of the detector. As a side-effect, the whole cycle moves  
 616 up in pressure requiring extra krypton mass for the pressurization. The desired operational strategy,  
 617 which is also affected by temperature requirements from the detectors, is therefore object of further  
 618 study.

619 An illustrative example of a fluid case study in supercritical state focusing exclusively on pressure drops  
 620 is presented below in Figure 19. The input conditions (p-T) and the heat load (150 W, equally  
 621 distributed<sup>2</sup>) were fixed, while the channel diameter and the flow rate were varied. The flow rate can  
 622 be adjusted by the ejector up to a certain extent, dependent on its design. The range of channel size  
 623 interesting for detector cooling is around or below 2 mm, which was previously considered as optimal  
 624 for flow boiling operation. In that area a colder outlet temperature than the inlet. Although the  
 625 estimation of the pressure drops is based on a single-phase correlation not developed for supercritical  
 626 fluids, the optimization concept does not change. An increase or decrease of the mass flow rate  
 627 influences the outlet fluid enthalpy. A smaller pipe diameter, for a given mass flow rate, produces larger  
 628 frictional losses along the cooling pipe. The combination of these effects can be seen in the  
 629 optimization study in Figure 19. A more intuitive representation can be found in Figure 19 (b-c), where  
 630 pressure-temperature profiles are plotted in the pressure-enthalpy diagram. The fluid temperature can  
 631 increase, decrease or have a sinusoidal profile. A similarity can be considered between gas heating and  
 632 the vaporization process: excessive low flow rates have a potentially harmful effect on the integrity of

<sup>2</sup> It is common to consider for sake of simplicity constant heat flux conditions along the detector stave. The heat load dissipated is a function of the sensor's temperature, which in turn is dependent on the temperature gradient (in first approximation only dependent on the fluid heat transfer coefficient if thermal resistivity of the support structure is considered temperature-independent). However, a short gap between the different sensors glued to the detector stave is typically present, resulting in alternating powered and unpowered sections.

633 the sensors in both cases. Indeed, during flow boiling low flow rates can cause dry-out with a reduction  
 634 of the heat removal capability, mainly due to the low thermal conductivity of the vapor phase. During  
 635 gas heating, regardless of the local fluid heat transfer coefficient, the non-uniformity of the fluid  
 636 temperature is in principle reflected in the temperature gradient along the thermal path between the  
 637 sensor and the cooling pipe. Furthermore, larger enthalpy changes suggest moving away from the  
 638 region near the pseudocritical points with an expected degradation of the heat transfer performance  
 639 (drop of the specific heat capacity). Figure 2 showed that the peaks in the specific heat capacity quickly  
 640 decreased with increasing temperature. Introducing pressure losses does not only help to trigger a  
 641 more uniform flow temperature distribution along the detector but also to remain close to the critical  
 642 point. The area characterized by those peaks is also very narrow and temperature dependent. This  
 643 suggests to design for a limited enthalpy gain of the fluid, while achieving a nearly zero or negative  
 644 temperature gradient between the inlet and the outlet.



645  
 646 **Figure 19: Fluid-study optimization (a) of the cooling channel during supercritical operation (solid line =  $\Delta P$ ,**  
 647 **dotted line =  $\Delta T$ ,  $d_{\text{in}}$  = inner diameter). Entering conditions are fixed ( $p = 60$  [bar],  $h = 77$  [kJ/kg]). The Swamee-**  
 648 **Jaime correlation was used for the pressure drop calculation [37]. Pressure profile along the detector for a**  
 649 **given flow of 12 [g/s] while changing the channel diameter (b). Temperature distribution (c) along the pipe for**  
 650 **the cases plotted in (b).**

## 651 11. Conclusion

652 Based on the HL-LHC plan, a future upgrade of the detector is planned to take place in 2034. To address  
 653 the challenges arising from a highly irradiated environment, a cooling fluid allowing for colder

654 operation than with CO<sub>2</sub> is required. A fluid-based comparison showed that the noble gas krypton is a  
655 promising candidate for the thermal management of the detectors. The gradual cooldown occurs in  
656 supercritical phase thus requiring a new cooling technology with a dedicated control logic. A new  
657 ejector-supported cycle is presented with a short description of the transient scenarios occurring  
658 during the detector lifetime. Some design guidelines have been drawn based on analysis of the fluid  
659 behavior in the supercritical and two-phase state, emphasizing the need of a distinct or combined  
660 optimization of the cooling channel inside the detector. The new krypton cycle is a candidate for the  
661 Vertex Locator (Velo) of the LHCb detector and for the NA62 experiment, which are experiments at  
662 CERN of limited cooling capacity (in the order of few kW).

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**Nomenclature**

***Abbreviations and variable names***

d	Diameter (mm)
$\Delta p$	Pressure drop (bar)
$\Delta T$	Temperature drop (K)
OD	Opening degree (-)
IHX	Internal heat exchanger
$\dot{m}$	Mass flow rate (kg/s)
P	Pressure (bar)
Q	Heat load (W)
VHTC	Volumetric heat transfer coefficient (W/m <sup>3</sup> K)
NBP	Normal Boiling Point (°C)

***Greek symbols***

$\theta_m$	Mass entrainment ratio (-)
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***Subscripts***

det	Detector
in	Inner

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