

A virtual calibration strategy and its validation for large-scale models of multi-sheet self-piercing rivet connections

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Abstract

This paper presents a strategy for the virtual calibration of a large-scale model representing a self-piercing rivet (SPR) connection. The connection is formed between a stack of three AA6016-T4 aluminium sheets and one SPR. The calibration process involves material characterisation, a detailed riveting process simulation, virtual joint unit tests and the final large-scale model calibration. The virtual tests were simulated by detailed solid-element FE models of the joint-unit. These detailed models were validated using experimental tests, namely peeling, single-lap joint and cross tests. The virtual parameter calibration was compared to the experimental calibration and finally applied to component test simulations. The paper contains both experiments and numerical models to characterise the mechanical behaviour of the SPR connection under large deformation and failure.

Keywords: Self-piercing rivet, virtual testing, multi-sheet, multi-scale, aluminium.

1. Introduction

Car body structures are made from a variety of components and materials, where each part is designed and placed to fulfil its purpose. The final geometry of the components and the choice of materials are the result of an iterative design process. Modern design philosophies advocate the integrated use of dissimilar materials including steel and aluminium sheets, polymers and foams, as well as metal extrusions and castings [1, 2]. This multi-material design strategy holds particular significance for

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28 lightweighting in cars, and especially for electric vehicles. Different material qualities
29 and part geometries usually require a tailored joining technique. Each car body struc-
30 ture has a large number of different combinations of material qualities and material
31 thicknesses, each demanding its own suitable fastening techniques. A well-established
32 fastening technique for the joining of aluminium sheets is self-piercing riveting (SPR)
33 [3].

34 SPR connections are a popular choice as they do not require pilot drilling of the
35 sheets and can therefore be placed automatically in assembly line production [4]. A
36 semi-tubular rivet is pushed into a material stack; as the rivet deforms, it locks the
37 sheets in place, creating a spot-like connection. Multiple SPR connections are often
38 placed as a seam on component edges, also in combination with structural adhesive
39 [5]. Most SPR connections are made between two mating sheets. However, connec-
40 tions can also be made up of multiple sheets, leading to more complex design possib-
41 ilities and enabling further lightweighting potential using multi-material systems.

42 During the design process of SPR connections, manufacturability as well as correct
43 mechanical performance must be ensured. Strength, failure and fatigue behaviour
44 are particularly important [6]. The connections greatly contribute to the structural
45 stiffness of the car, and moreover have a significant influence on the deformation of
46 components under events like crash and impact. In order to meet worldwide crash
47 regulations, the car body must behave in a predictable and safe manner. The design
48 of full-car body structures and their components employs powerful numerical tools
49 like the finite element method (FEM).

50 With respect to the SPR connection design, FEM can be utilised at different scales.
51 Process simulations facilitate the creation of SPR geometries, enabling the assessment
52 of joint quality through measures such as interlock or by computation of the riveting
53 force. Process simulations are also the basis for detailed solid element models. Re-
54 cent studies on SPR process modelling have been presented by Fang et al. [7], Kappe
55 et al. [8] and Zhao et al. [9]. To study the mechanical behaviour of SPR during
56 severe loading, detailed simulations with solid elements are used, as presented by

57 Hirsch et al. [10], Hönsch et al. [11] and Karim et al. [12]. To reduce the computa-
58 tional effort, surrogate models are used in shell-element-based car crash analyses as
59 shown recently by Duan et al. [13], Leconte et al. [14] and Wang et al. [15]. Under
60 defined conditions and with sufficient knowledge of the material's constitutive beha-
61 viour, mechanical SPR characterisation tests can be modelled [16]. As the mechan-
62 ical characterisation of SPR connections is time and material-consuming, these virtual
63 tests offer great potential. Virtual testing allows for the optimisation of tool geometry
64 and material pairing, easing the testing of multiple joint configurations in the early
65 development phase and aiding in the screening of possible design choices [17].

66 Surrogate or connector models applied in large-scale car crash simulations often
67 require various experiments for calibration [14]. These calibration experiments can
68 to some extent be replaced by detailed simulations of the SPR connection. Detailed
69 simulations of the joint are further referred to as mesoscopic simulations/models,
70 where 3D solid elements are used to discretise the rivet and sheet material. Connector
71 models used in full-scale car crash simulations are further referred to as macroscopic
72 models. Macroscopic models use shell elements to discretise the sheet material, where
73 the element size is equal to or larger than the size of the SPR. Therefore, the detailed
74 geometry of the SPR cannot be modelled using said large shell elements. Instead, the
75 connector model is used to reproduce the force-displacement behaviour of the joint.

76 Limited work has been published on the large-scale modelling of three-sheet SPR
77 connections. Process simulations followed by virtual calibration tests of large-scale
78 shell connector models are rare. Since three-sheet connector models require more
79 tests than two-sheet connections, a virtual calibration procedure is attractive. There-
80 fore, this work presents a novel approach for a virtual calibration strategy and its
81 experimental validation for an SPR connection between three aluminium sheets. The
82 individual steps throughout the calibration procedure are not novel in themselves and
83 can be replicated. The work involves calibrating models based on material and joint
84 tests, with each stage of the process being experimentally validated, including com-
85 ponent tests.

86 2. Calibration framework

87 The virtual calibration strategy proposed in this work had four main steps and
88 is depicted in Fig. 1. The first step involved the material characterisation and con-
89 stitutive modelling for both mesoscopic and macroscopic simulations. In addition,
90 cross section cuts of the SPR connection were made in order to evaluate the accuracy
91 of the following riveting process step.

92 The process simulation allowed for a fast creation of the connector geometry using
93 axisymmetric 2D models, which included the known rivet and die geometries as well
94 as the constitutive material models. This process simulation produced the SPR con-
95 nection geometry including the resulting plastic strain and work hardening history. By
96 adjusting process parameters such as friction coefficients and riveting speed, a close
97 fit to the physical section cut could be achieved. The friction coefficient between the
98 top-middle-sheet, middle-bottom-sheet, bottom-sheet-die and rivet-top-sheet pairings
99 was tuned to optimise the fit between the physical section cut and the deformed mesh
100 contour. The LS-OPT software was used to tune the process parameters to achieve
101 minimum error between experimental and simulated rivet quality measures. The
102 rivet quality measures were: sheet interlock, rivet spread and compression, as further
103 described in Section 2.1.

104 A solid element model of the SPR connection was generated from the resulting
105 geometry and incorporated into mesoscopic cross test models. Then, these meso-
106 scopic cross test models were validated by peeling, single lap joint (SLJ) and free
107 middle sheet cross tests. The responses from the mesoscopic cross tests are referred
108 to as virtual cross test results.

109 A parallel large-scale model calibration was done using the experimental cross
110 test to compare against the virtually calibrated models. The parameter set for the
111 experimentally calibrated model was taken from André et al. [18].

112 The macroscopic models including the calibrated connector models were valid-
113 ated by peeling, SLJ and free middle sheet cross tests. The selection of these tests
114 was motivated by their non-proportional loading mode, which poses a greater chal-

115 lenge to the model compared to the cross tests, which give an almost proportional
 116 loading. In addition, these tests do not enforce failure of a specific sheet since the
 117 middle sheet is unclamped. Finally, the interaction effects of multiple three-sheet SPR
 118 connections on a large-scale level were investigated with a component test. The mac-
 119 roscopic connector models were thereby challenged and the differences between the
 120 experimentally and the virtually calibrated models on the global component beha-
 viour were evaluated.

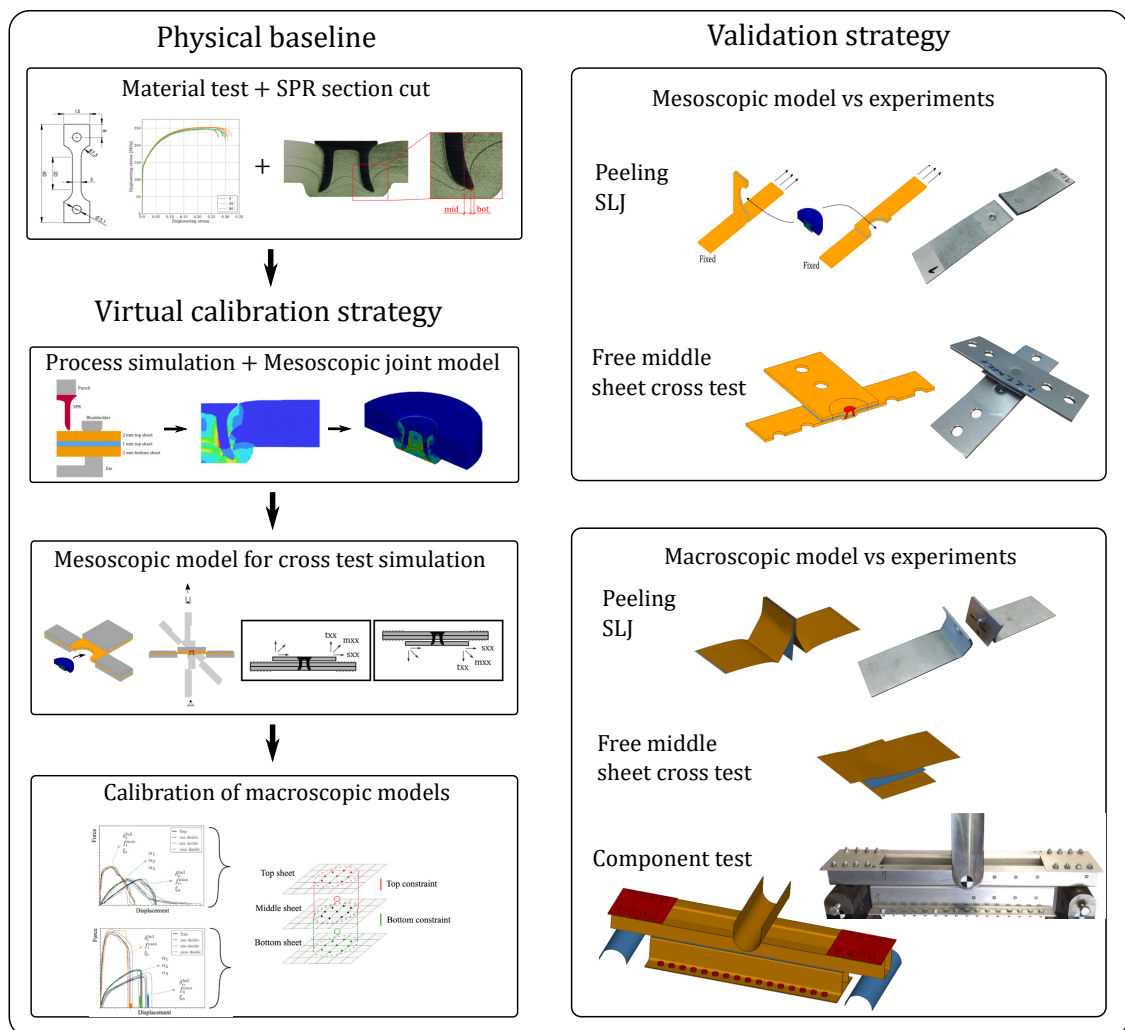


Fig. 1. Calibration and validation strategy / overview

121

122 2.1. Strategy overview

123 The presented calibration strategy is applied to a SPR connection made from three
 124 AA6016 aluminium sheets in T4 temper condition and a rivet from type Böllhoff
 125 RIVSET K 5.3×7. The aluminium sheets were stacked with a 2 mm thick sheet on

126 the top and bottom and a 1mm thick sheet in between, resulting in a total stack thick-
 127 ness of 5 mm. A section cut of the resulting joint is shown in Fig. 2. The rivet had
 128 a shank diameter of 5.3 mm, the head diameter was 7.75 mm and the overall rivet
 129 length was 7 mm, as shown in Fig. 2. The riveting die was shaped with a flat bot-
 130 tom and a diameter of 11 mm. This joint SPR configuration was studied by André
 131 et al. [18] where the effect of different sheet stack-ups on the joint behaviour was
 132 investigated.

133 For two-sheet SPR connections, usually the interlock between rivet and bottom
 134 sheet is measured, which is a practical quality criterion [2]. The bottom interlock
 135 is defined as the radial distance between rivet tip and bottom sheet tip which folds
 136 inwards around the rivet tip. A high interlock measure is usually a good indicator
 137 for a strong mechanical connection between the bottom sheet and the rivet. For the
 138 three-sheet connection, an additional interlock for the middle sheet was defined, as
 139 displayed in Fig. 2. The middle interlock was defined as the radial distance between
 140 bottom sheet tip and inner edge of the middle sheet. Rivet spread and compression
 141 are the differences between deformed and undeformed rivet geometry measured at
 142 the rivet tip. A bottom-sheet interlock of 0.22 mm (“bot” Fig. 2) and a middle-sheet
 143 interlock of 0.40 mm (“mid” Fig. 2) were measured. In the section cut, the rivet got
 144 compressed by 0.80 mm (“comp” Fig. 2) and the tip of the legs were spread out by
 0.74 mm (“spread” Fig. 2).

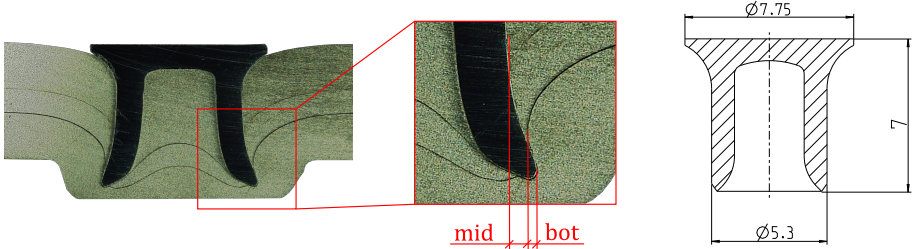


Fig. 2. SPR section cut with interlock, compression and spread measures.

145

146 *2.2. Sheet material characterisation*

147 The mechanical properties of the rolled AA6016-T4 sheets were experimentally
 148 characterised. Tensile specimens with the dimensions given in Fig. 3a were extracted

149 from the sheets in 0° , 45° and 90° with respect to the rolling direction. Four specimens
150 were extracted for each sheet thickness in each direction, which resulted in a total of
151 24 tensile specimens. Dimensional accuracy was ensured by cutting the specimens
152 using wire erosion in a water bath, making sure the aluminium was not affected by
153 heat. The tensile tests were carried out at room temperature, 0.02 mm/s and recor-
154 ded at 1 frame/s with a Basler acA4112 camera and an approximate resolution of 80
155 px/mm. The specimens were loaded with an Instron 5982 universal testing machine
156 and a 100 kN load cell. The specimens were painted with a black-and-white speckle
157 pattern and the surface deformations were computed using digital image correlation
158 (DIC) with the software eCorr [19]. A digital extensometer with a gauge length of
159 18.38 mm was used to obtain the engineering strains. The determined stress-strain
160 curves are displayed in Figs. 3b and 3c for the two thicknesses, respectively. No signi-
161 ficant difference in yield stress, work hardening and failure was observed for all the
162 directions. The higher strength of the 1 mm sheet could be an effect of the thinner
163 rolling but was not further evaluated.

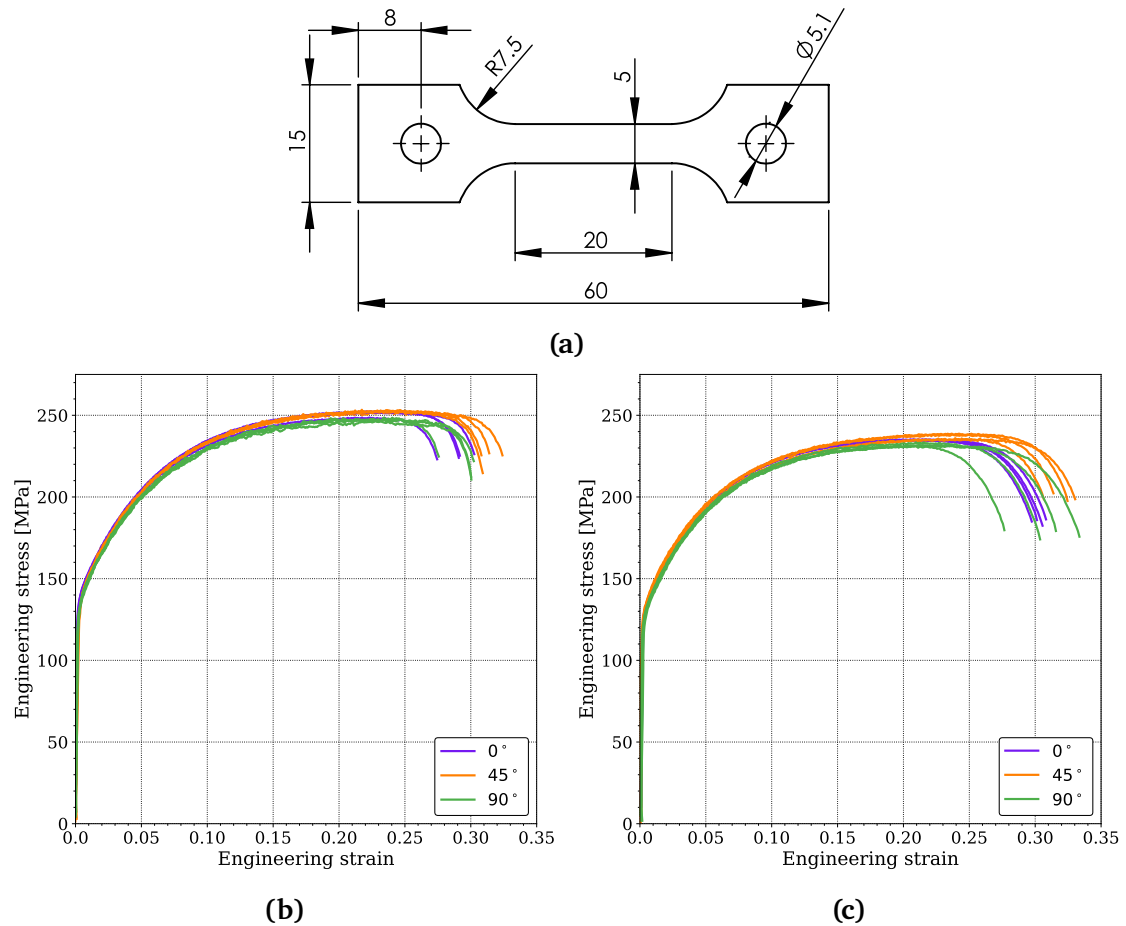


Fig. 3. Tensile test: (a) specimen dimensions in mm, (b) 1 mm sheet and (c) 2 mm sheet.

164 **2.3. Rivet material characterisation**

165 The rivet material was characterised by axial compression of the rivet body between
 166 two flat surfaces without lubrication. Material characterisation by crushing of the rivet
 167 body, previously presented by Khezri et al. [20], Porcaro et al. [21], Baha II [22] and
 168 Hönsch et al. [23], was simplified in this work by testing the original rivet without ad-
 169 ditional machining. The deformation sequence is displayed in Fig. 4a and a bulging
 170 of the rivet's midsection could be observed. DIC was used to track the punch dis-
 171 placement applying the eCorr [19] point tracking algorithm. Three repetitions were
 172 carried out under quasi-static conditions and the force-displacement curves can be
 173 seen in Fig. 4b. The results were repeatable and the response curves were used for
 174 later material model calibration through inverse engineering.

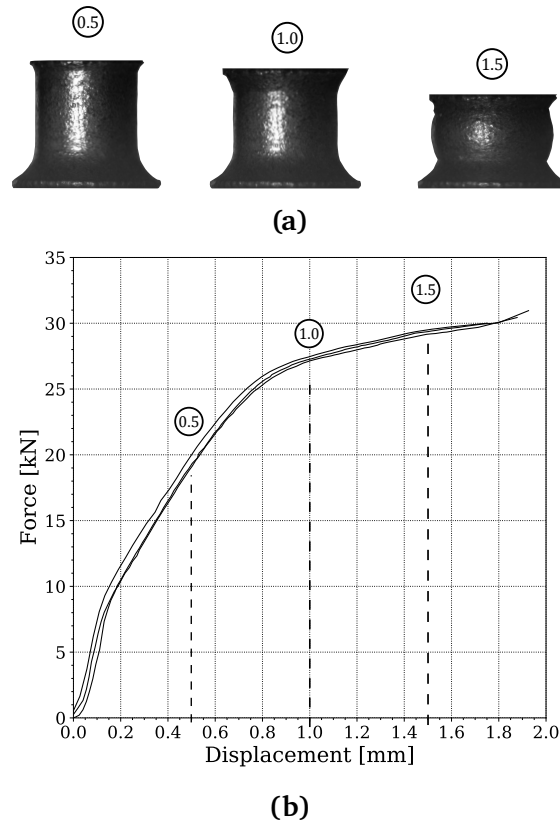


Fig. 4. Rivet compression test: (a) deformation at 0.5, 1.0 and 1.5 mm compression, (b) force-displacement response.

175 *2.4. SPR joint characterisation*

176 The SPR connection investigated in this work was experimentally tested in André
 177 et al. [18], where the joint unit was characterised by cross tests under six loading
 178 modes. Either the top sheet was loaded and the middle and bottom sheets were
 179 clamped or the bottom sheet was loaded and the top and middle sheets were clamped.
 180 By rotating the clamps in the test rig, different loading modes were achieved, see
 181 Fig. 5a. Either the top or bottom sheet was loaded under pure tension, pure shear
 182 and mixed mode. The joint loading modes are depicted in Figs. 5c and 5d and are
 183 referred to as txx, sxx and mxx for top sheet tensile, shear and mixed mode loading.
 184 Bottom layer tension, shear and mixed mode loading are abbreviated with xxt, xxs
 185 and respectively xxm.

186 Each loading configuration was tested with five repetitions resulting in total 30
 187 tests. Fig. 6 shows the resulting force-displacement curves. Two distinct failure modes
 188 were observed. Rivet-pull-through was defined as the top sheet extracting the rivet

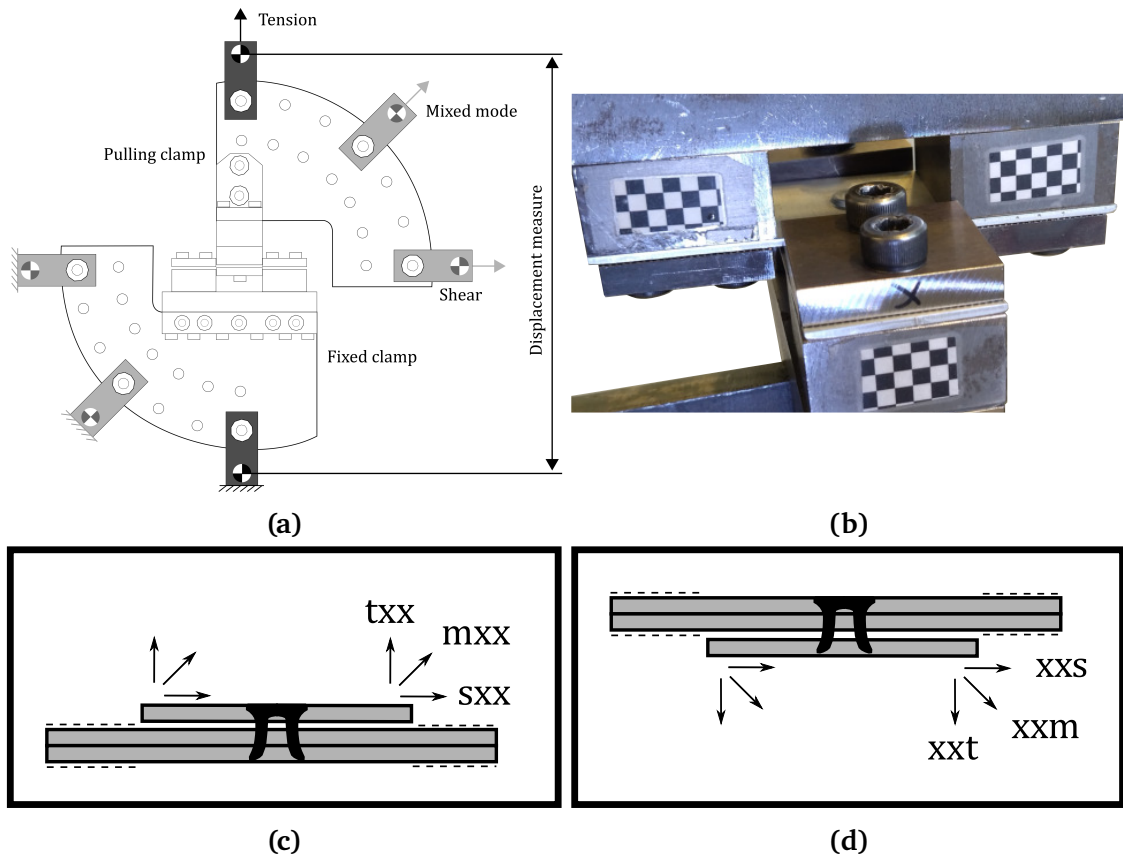


Fig. 5. Cross test: (a) schematic test rig, (b) clamping, (c) top sheet loading and (d) bottom sheet loading.

189 body, leading to the full separation of all sheets. Rivet-pull-out was defined as the
 190 case where the bottom sheet detached from the rivet while the top and middle sheets
 191 were still connected. The assignment of the failure modes to the cross test loadings
 192 is displayed in Fig. 7.

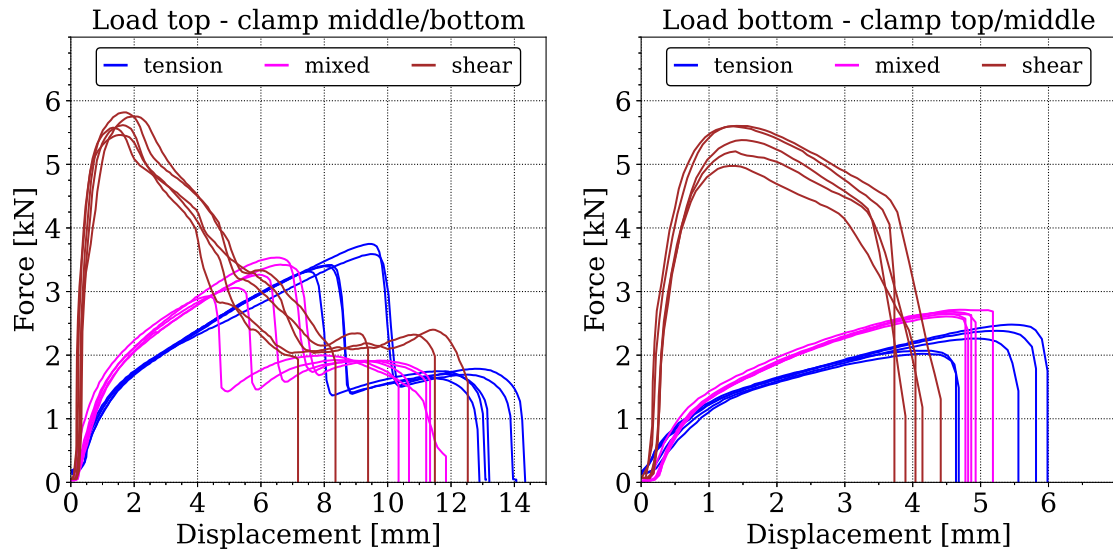


Fig. 6. Experimental cross test results.

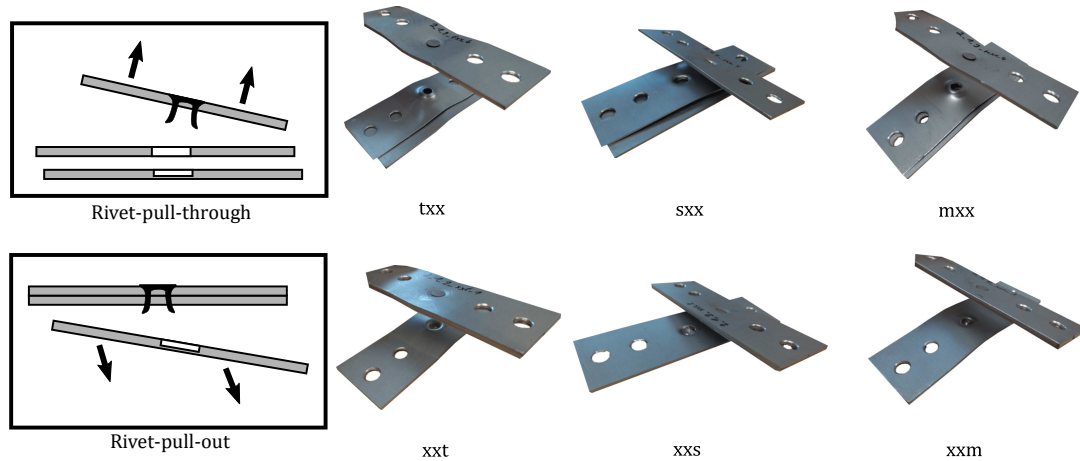


Fig. 7. Assigned cross test failure modes

193 3. Mesoscopic modelling

194 3.1. Constitutive modelling for process and mesoscopic modelling

195 For both the aluminium sheet and the rivet an isotropic plasticity material model
 196 was applied. The *MAT_PIECEWISE_LINEAR_PLASTICITY keyword in LS-DYNA [24]
 197 was used which offers the von Mises yield criterion with isotropic hardening. The
 198 hardening for the aluminium sheets was modelled by the Voce hardening rule ac-
 199 cording to Eq. (A.3). A combination of power-law and Voce hardening was used to
 200 model the work-hardening behaviour of the rivet material as described by Eq. (A.5).
 201 The governing equations for the constitutive modelling are supplied in Appendix A.

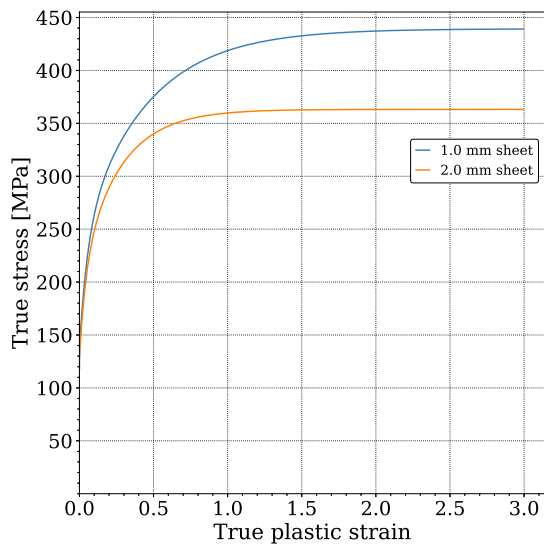
202 The flow curves generated from the hardening rules Eq. (A.3) and Eq. (A.5) were

203 supplied in tabulated form and the evolution of true stress over true plastic strain
 204 for the sheet and the rivet material are shown in Fig. 8a and Fig. 9a, respectively.
 205 Damage in the aluminium and the rivet material was not modelled as no fracture
 206 was observed in the cross tests. Thinning and splitting of the aluminium sheet in the
 207 process simulation was modelled by adaptive remeshing.

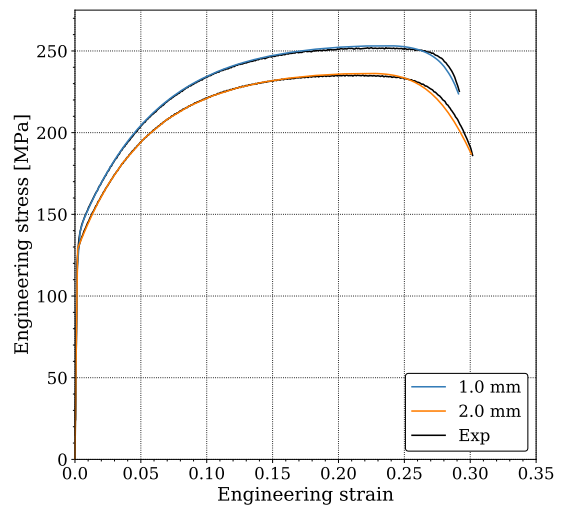
208 The response curves from the inverse solid element models of the tensile test and
 209 the rivet compression test are shown in Fig. 8b and Fig. 9c. The simulations of the
 210 sheet tensile tests showed good agreement with the experiments and the material
 211 model was able to capture the necking behaviour. The hardening model for the rivet
 212 material allowed for the capture of the initial stiffness of the rivet compression tests
 213 with slight deviation for larger punch displacements. As none of these extensive de-
 214 formations occur in the process simulation and the mesoscopic models, the curve fit
 215 as seen in Fig. 9c is sufficient. The hardening parameters for the rivet material are
 216 given in Table 1. The compression test simulation was therefore stopped at 1.8 mm.
 217 A friction coefficient of 0.15 was chosen between the rivet and the punches giving
 218 a similar deformation mode with bulging of the rivets mid section replicated by the
 219 simulation, depicted in Fig. 9b. It should be noted that the friction coefficient might
 220 change the stiffness and deformed shape of the rivet under compression.

Table 1. Material parameters for rivet

E	ν	σ_0	Q	H	c
(GPa)	(-)	(MPa)	(MPa)	(MPa)	(-)
210.0	0.30	1302.2	443.9	314.3	82.1

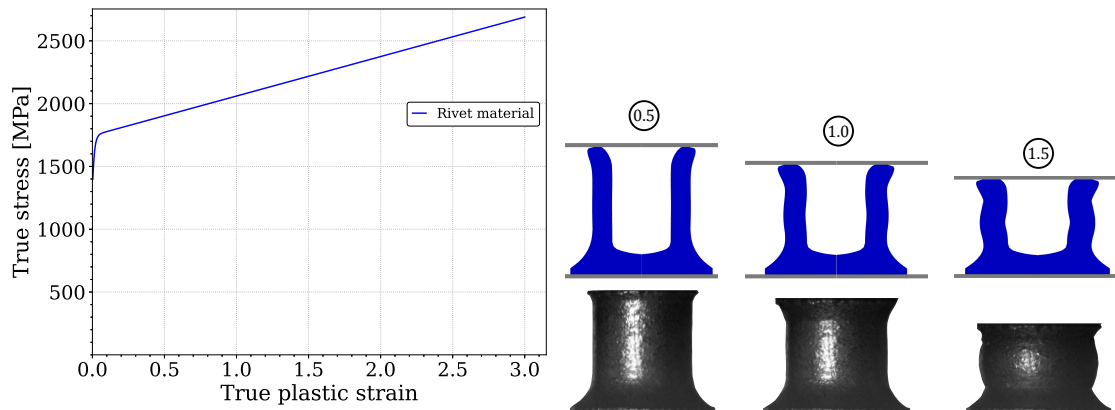


(a)



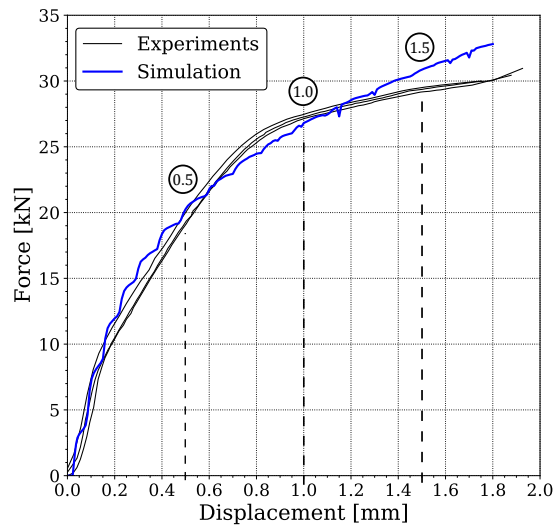
(b)

Fig. 8. Aluminium material calibration: (a) extrapolated flowcurves, (b) inverse tensile test modelling results.



(a)

(b)



(c)

Fig. 9. Rivet material calibration: (a) extrapolated flowcurve, (b) deformation sequence simulation vs experiment and (c) inverse rivet compression model response.

221 *3.2. Riveting process simulation and virtual cross testing*

222 The SPR riveting process was modelled in the explicit LS-DYNA solver version 11
223 [24] using a 2D axisymmetric model. Many studies have been published on the suitability
224 of 2D models for the riveting process. A reduction of numerical costs can thereby
225 be reduced as shown by Porcaro et al. [21] for the modelling of two-sheet aluminium
226 SPR connections. Bouchard et al. [25] and Karathanasopoulos et al. [26] presented
227 the successful application of 2D axisymmetric models for hybrid aluminium and
228 steel SPR joints. The process modelling of three-sheet aluminium SPR connections
229 was shown by Mori et al. [4]. According to Moraes et al. [27], the simulated rivet
230 interlock as a key feature of the SPR joint is greatly influenced by the coefficients of
231 friction between the sheets and the die. Multiple simulations with adjustment of the
232 friction parameters were therefore necessary to find the modelled joint interlock that
233 matched the physical cross section. Multiple options for modelling material fracture
234 are available including simple minimum thickness criteria or complex constitutive
235 models. However, according to Huang et al. [28], an inappropriate failure criterion
236 can lead to unrealistic volume loss or incorrect crack growth, which can cause a deviation
237 between the experimental and simulated joint cross-section and rivet force.

238 The riveting process begins with the sheets being clamped between die and blank-
239 holder Fig. 11a. The punch pushes the rivet into the sheets where the rivet legs penetrate
240 and bend outwards to plastically deform, locking the sheets together. When
241 the rivet head is approximately flush with the top sheet surface, the punch and blank-
242 holder retract.

243 The riveting process model consists of deformable parts, namely the rivet body and
244 the three sheets. Punch, blankholder and die were modelled as rigid bodies. All parts
245 were discretised using four-node volume weighted axisymmetric solid elements. As
246 the sheet material experiences severe plastic deformation, an adaptive remeshing algorithm
247 was applied to overcome excessive element distortion. The rivet and sheet
248 materials were modelled using the isotropic plasticity model described in Section 3.1.
249 In these simulations, the effects of strain-rate and temperature due to adiabatic con-

250 ditions were not accounted for. Modelling of material damage was not accounted for
251 as element erosion could lead to problems in the contact formulation and eventually
252 to an aborted simulation. Material failure due to sheet piercing was modelled using
253 the part adaptive failure keyword from LS-DYNA. This feature allows for the splitting
254 of a sheet into two parts based on a critical thickness, thus creating two new surfaces
255 without deleting elements [21]. The remeshing steps were sufficient to account for
256 the thinning and splitting of the sheet material. A 2D penalty based Coulomb friction
257 formulation was used between the parts. Because the punch displacement, the punch
258 velocity and the friction coefficients have significant influence on the final joint geo-
259 metry, several iterations with adjustment of parameters was necessary to achieve the
260 optimal geometry.

261 The resulting mesh contour is displayed and compared to the physical section
262 cut in Fig. 10. This shape was achieved using a friction coefficient of 0.31 between
263 the sheets, friction coefficient of 0.36 between sheets and die and almost no friction
264 between the rivet and the sheets with a coefficient of 0.08. The friction parameters
265 were found to give a good fit between physical cut and mesh geometry measured by
266 the variables in Table 2. As seen in the table, a higher rivet compression and spread
267 than in the physical cut was achieved. While the bottom interlock was underestimated,
268 the middle interlock showed a higher value than in the physical cut. These
269 values provided the best results in the cross-test simulations, even though they show
a relatively large error in Table 2.

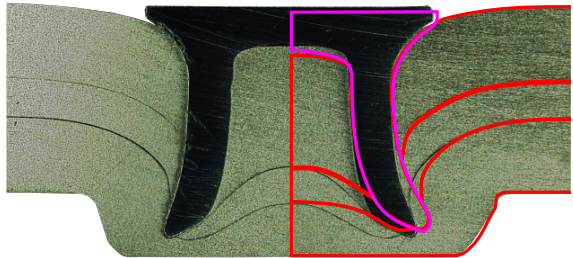


Fig. 10. SPR section: physical cut vs mesh contour.

270
271 After the 2D process simulation, the deformed mesh geometry of rivet and sheets
272 was extracted while storing the equivalent plastic strain and work hardening for each

Table 2. Riveting process quality measures in mm.

	Rivet compression	Rivet spread	Bottom interlock	Middle interlock
Physical cut	0.80	0.74	0.22	0.40
Mesh contour	1.19	1.03	0.17	0.54
Error	49 %	39 %	23 %	35 %

273 element, Fig. 11b. A 3D solid mesh was generated by rotation of the shell elements
274 in 20 angular increments Fig. 11c, Fig. 11d. The resulting model unit was inserted in
275 a cross test specimen mesh, Fig. 11e, further referred to as a mesoscopic model.

276 Two different mesh configurations were used where either the top-loaded sheet
277 or the bottom-loaded sheet were placed transversely to the two clamped sheets. The
278 sheet meshes were sandwiched between rigid bodies replicating the cross test setup
279 shown in Fig. 11f. The tests were modelled using a symmetry plane to reduce the
280 computational costs. The mixed mode and shear cases incorporated a half-sized clamp
281 for clearance of the moving sheet. Each clamp was made from two solid parts which
282 were held together with beam elements, modelling the initial clamp tightening and
283 allowing for slight opening of the clamp Fig. 11g. The clamping load was applied with
284 an initial axial force in the clamping beam elements. The pre-load in the beams was 8
285 kN which was assumed from the hand tightening of the real M8 bolts. Different axial
286 pre-loading values showed no significant differences in the clamping behaviour. The
287 contact between clamps and sheets was realised by a penalty-based Coulomb friction
288 model with a friction coefficient of 0.25 allowing for minimal sliding during the test.

289 The resulting force-displacement curves from the mesoscopic cross test simula-
290 tions are given in Fig. 12 and are compared to the experiments. The models were
291 able to capture the overall curve shape for each loading mode with a good agreement
292 of both peak force and displacement to failure. However, the models of the shear
293 loadings overestimated the peak force, especially in the bottom layer loading xxs.
294 The modelling of sheet material damage and element erosion around the rivet could
295 have decreased the strength but was not evaluated further. All simulations predicted a
296 stiffer joint response than experiments, especially in the bottom layer loading modes.

297 However, all models reproduced the failure modes as observed in the experiments,
 298 which can also be seen by the matching drop in force in the txx and mxx tests where
 299 middle and bottom layer separated around the rivet. The discrepancies observed in
 300 the mesoscopic simulations of the cross-tests are believed to be associated with the
 301 deformed geometry of the rivet and surrounding sheets obtained through the process
 simulations.

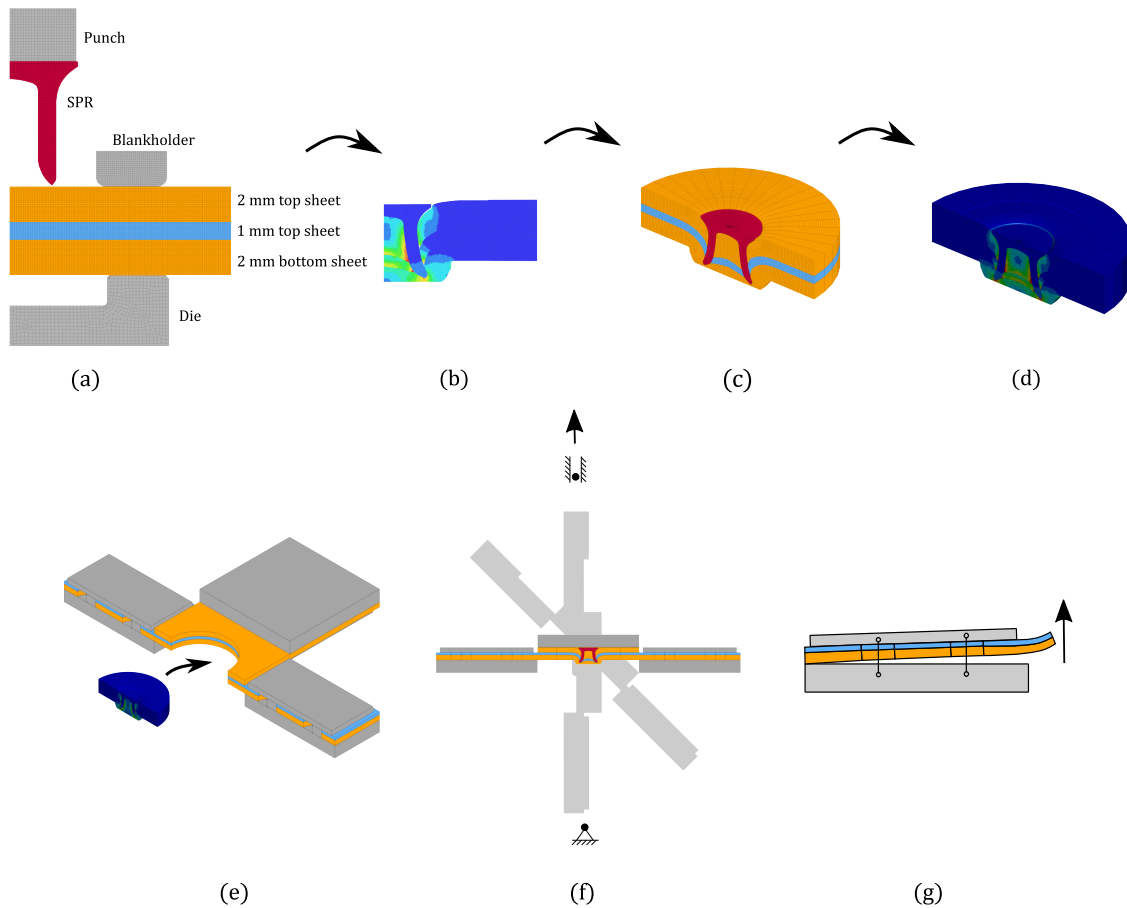


Fig. 11. Mesoscopic model generation: (a) 2D riveting process, (b) resulting 2D mesh, (c) solid element mesh, (d) plastic strain mapping, (e) cross test model, (f) hinged cross test rig model and (g) clamping setup.

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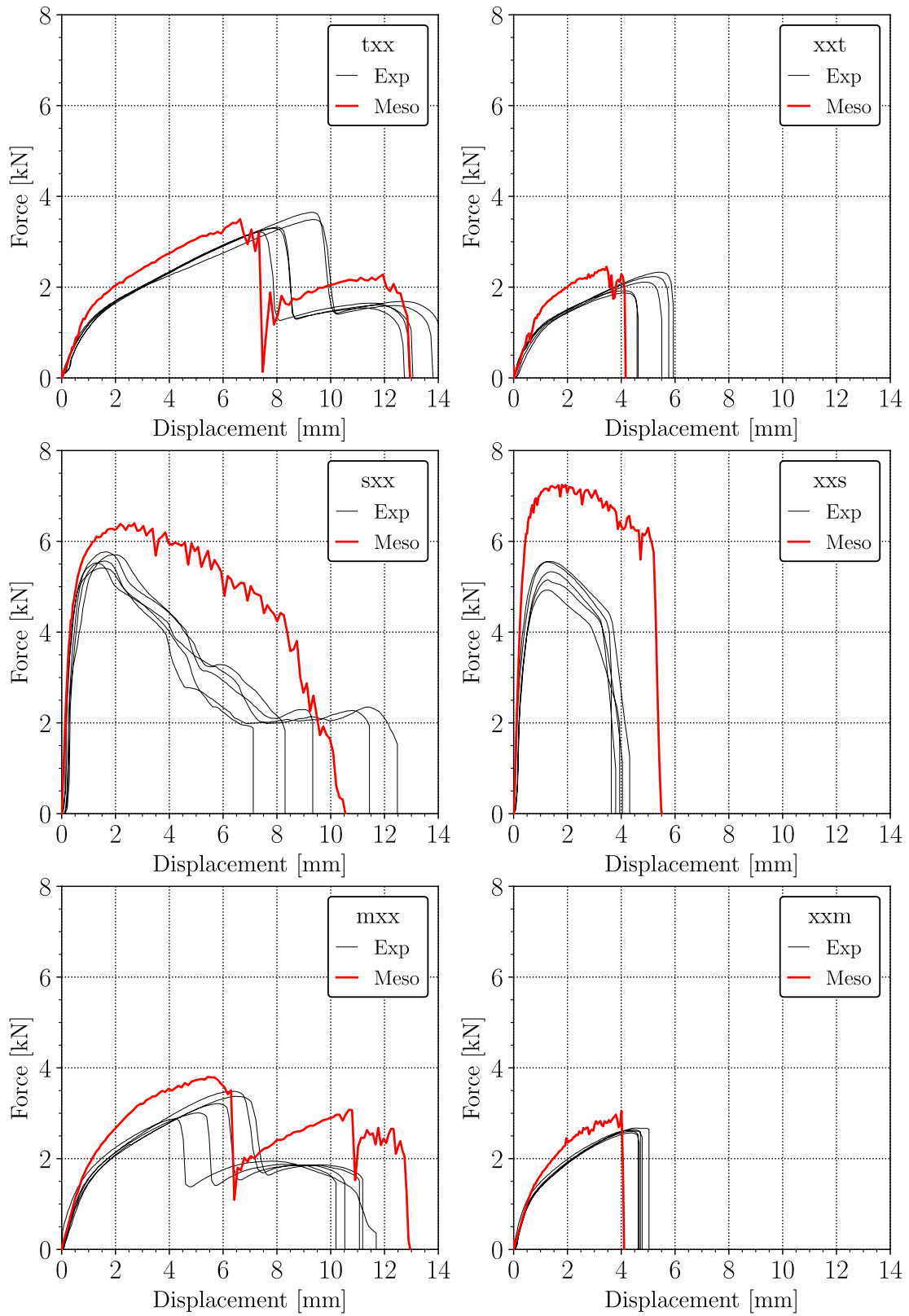


Fig. 12. Force-displacement curves of the mesoscopic cross tests vs the experiments.

303 4. Macroscopic modelling

304 4.1. Constitutive modelling for shell elements

305 The shell-element discretised aluminium sheets in the macroscopic simulations
306 were modelled with an isotropic plasticity model. The Hershey-Hosford yield criterion
307 [29] and Voce-hardening were adopted. The model is based on the work by Costas et
308 al. [30] and is available as *MAT_258 keyword in the explicit LS-DYNA solver [24].
309 The work hardening parameters were taken from [18] and the obtained simulation
310 response is displayed in Fig. 8b. The Voce-hardening parameters were originally ob-
311 tained from the tension tests in 0° direction and are shown in Table 3. That results in
two individual material cards for the 1 mm and the 2 mm sheet material.

Table 3. Material parameters for AA6016 in T4 temper

Sheet thickness (mm)	E (GPa)	ν (-)	σ_0 (MPa)	θ_{R1} (MPa)	Q_{R1} (MPa)	θ_{R2} (MPa)	Q_{R2} (MPa)	θ_{R3} (MPa)	Q_{R3} (MPa)
1.0	70.0	0.33	115.3	26919.2	22.7	1831.9	102.2	451.9	199.3
2.0	70.0	0.33	128.7	1362.4	59.4	604.4	157.6	403.0	17.6

312

313 4.2. Macroscopic connector modelling

314 Modelling of the SPR in large-scale macroscopic models was realised by a con-
315 nector model. In this work, the connector was modelled by the constraint formula-
316 tion proposed by Hanssen et al. [31], which is available as *CONSTRAINED_SPR2
317 keyword in the explicit LS-DYNA solver [24]. While the connector model was origin-
318 ally designed for connections between two sheets, André et al. [18] presented the
319 application of two sandwiched connectors between three sheets for the modelling of
320 three-sheet connections. The stacking of two connector models is sketched in Fig. 13.

321

322 One constraint was placed between each pair of shell element meshes and, de-
323 pending on the relative nodal displacement of the sheets, the resulting forces and
324 moments acting on the sheets were computed. The constraint model accounts for
325 damage and scales down the forces while the state of maximum accumulated dam-
326 age is stored. Each constraint model is governed by nine parameters which are usually

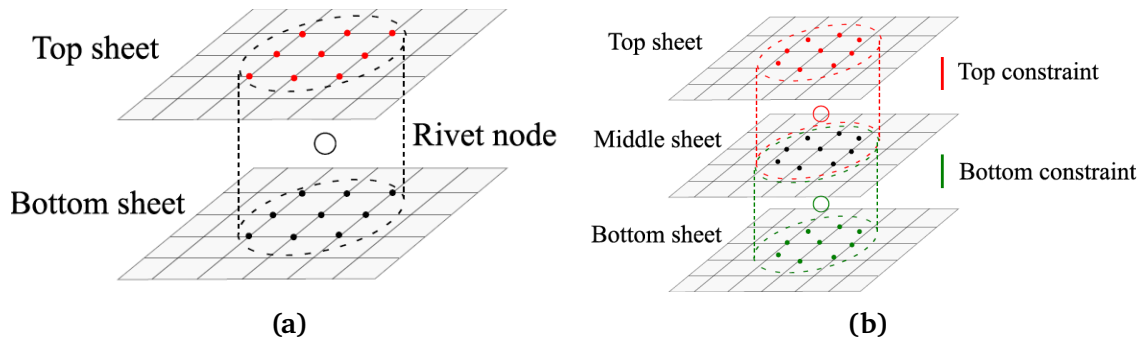


Fig. 13. Constraint modelling technique: (a) general two-sheet and (b) three-sheet connection.

327 fitted by inverse engineering. Governing equations for the constraint model are given
 328 in Appendix B. For in-depth discussion of the constraint model, the work by Hanssen
 329 et al. [31] is referred to.

330 The modelling strategy for three-sheet connections applies two stacked connector
 331 models which rely on the same constraint model formulation but are independent in
 332 the sense that they use two different sets of parameters. The two connector models
 333 are neither connected with each other nor can they communicate. Fig. 14 shows the
 334 proposed calibration process of the 18 parameters for the three-sheet connection. This
 335 method allows for a calibration using a reasonable amount of experiments, which are
 336 the six cross tests. The top sheet loading tests were used for calibration of the top
 337 connector model and the bottom sheet loading tests for calibration of the bottom
 338 connector, respectively.

339 4.3. Connector model calibration: experimental and virtual

340 With the aim of comparing the virtual versus the experimental calibration strategy,
 341 both connector models were calibrated by inverse modelling of either the physical
 342 cross tests or the virtual tests. The response curves from the experimentally calib-
 343 rated macroscopic models are shown in Fig. 15. The cross test model response was
 344 in good agreement with the experiments. The parameters under pure shear loading
 345 were calibrated so that a conservative failure behaviour at lower shear displacement
 346 was achieved. Also failure under top sheet tension and mixed mode loading were
 347 calibrated conservatively to ensure failure at the first force drop. This ensures full

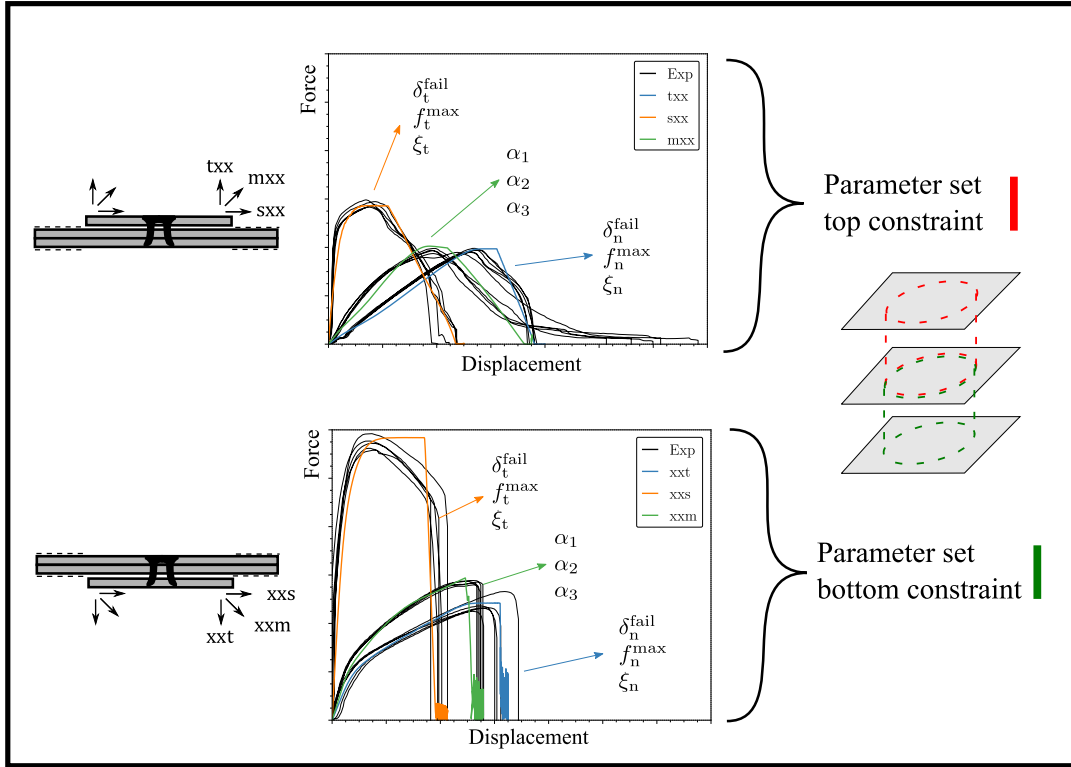


Fig. 14. Calibration process for each single constraint

348 separation of the top sheet from the clamped middle and bottom sheets.

349 Fig. 15 also shows the response from the cross test models with parameters found
 350 by inverse modelling of the virtual mesoscopic cross tests. The macroscopic simula-
 351 tions were in good agreement with the underlying mesoscopic model. Again, a more
 352 conservative parameter set was chosen to accomplish separation of the top sheet un-
 353 der top sheet tension and mixed mode loading. The response from the mixed mode
 354 and free middle sheet test showed deviation in force level and displacement to failure.
 355 The initial stiffness under shear loading could not be fully reached applying the vir-
 356 tually calibrated parameter set. Different rivet diameters were used for the virtually
 357 calibrated models to adjust the initial stiffness by increasing the diameter of influence
 358 and incorporating more nodes. The parameter sets are given in Table 4.

Table 4. Constraint model parameters. Experimental calibration obtained from [18].

Calibration	Constraint	δ_n^{fail} (mm)	δ_t^{fail} (mm)	f_n^{max} (kN)	f_t^{max} (kN)	ξ_n (-)	ξ_t (-)	α_1 (-)	α_2 (-)	α_3 (-)	Diameter (mm)
Experimental	Upper	3.149	3.566	3.392	5.801	0.749	0.705	0.804	0.750	1.489	10.0
	Lower	2.252	3.757	2.261	5.798	0.696	0.749	0.795	0.899	1.402	10.0
Virtual	Upper	2.956	10.356	3.392	6.253	0.749	0.350	1.171	0.386	0.437	15.0
	Lower	2.252	5.369	2.361	7.824	0.696	0.751	0.450	1.154	1.189	20.0

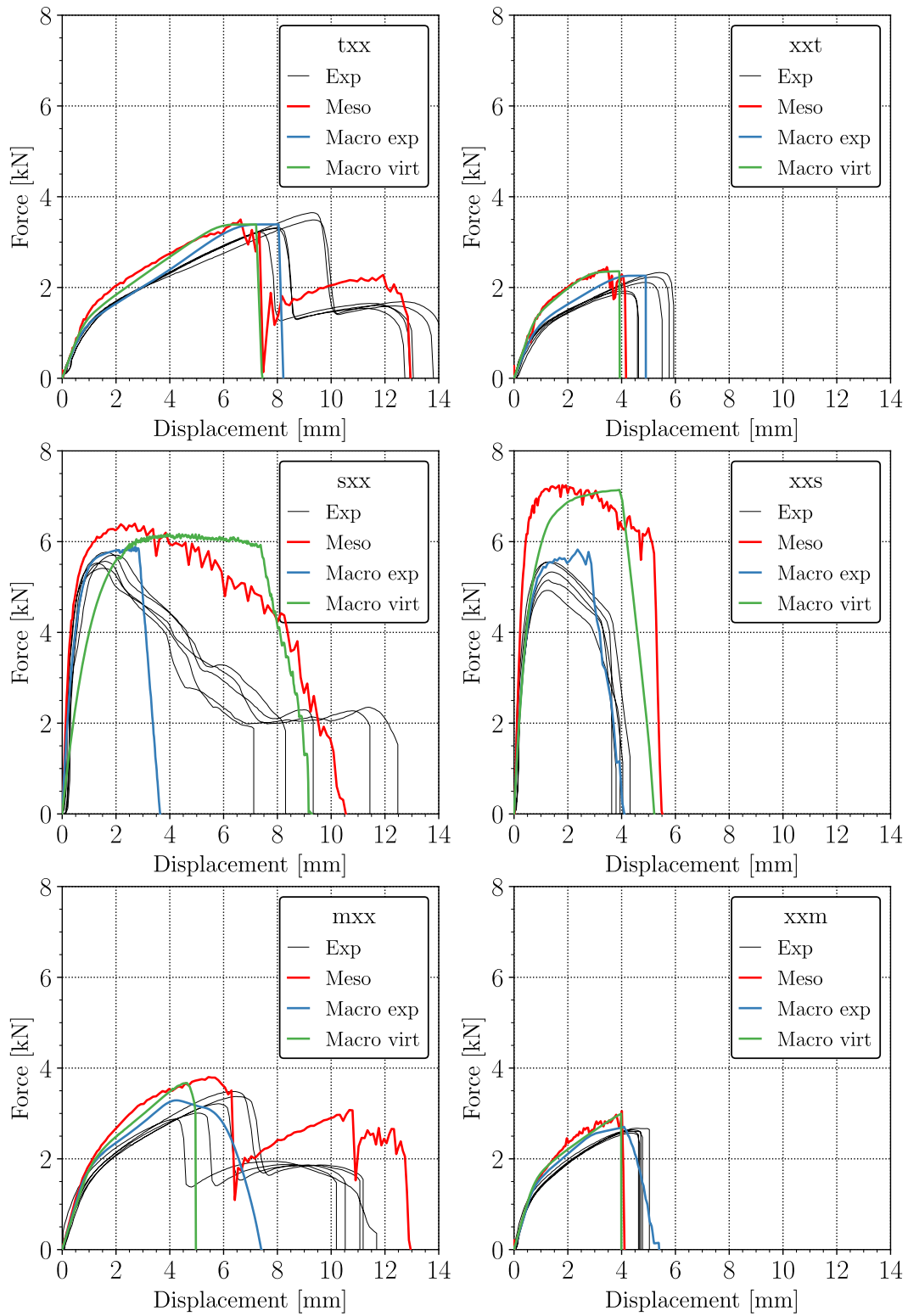


Fig. 15. Results from virtual cross tests, experimentally and virtually calibrated macroscopic models vs the experiments.

359 **5. Strategy validation: Meso- and macroscopic models vs experiments**

360 *5.1. Benchmark tests and simulation*

361 For the validation of the meso- and macroscopic models, three benchmark tests
362 where conducted. The tests consisted of a cross test where the middle sheet was
363 unclamped, a single-lap-joint (SLJ) test and a peeling test.

364 The free middle sheet cross test had the same dimensions as the regular cross tests
365 but the middle layer was not clamped and was therefore a 40×40 mm square. The
366 results from the free middle sheet test (labelled mox) are displayed in Fig. 16. The
367 mesoscopic model and the experimentally calibrated macroscopic model showed a
368 good representation of the experimental behaviour. The virtually calibrated macro-
369 scopic model captured the response from the mesoscopic model but overestimated
peak force and failure displacement slightly.

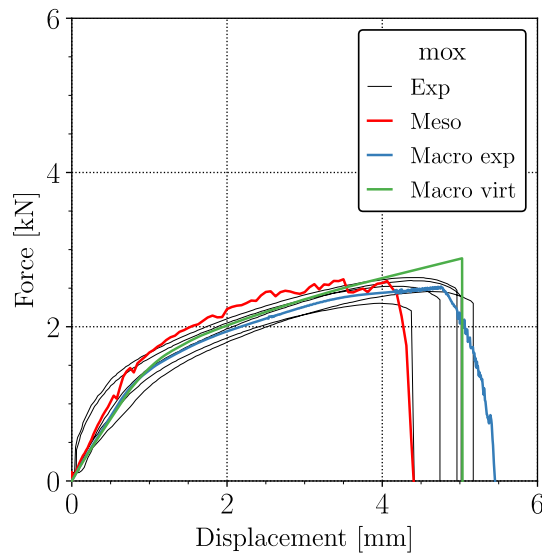


Fig. 16. Free middle sheet test: experiments “Exp”, mesoscopic model “Meso”, macroscopic model experimental calibration “Macro exp”, macroscopic model virtual calibration “Macro virt”.

370

371 The dimensions of the peeling and SLJ specimens are displayed in Fig. 17a and
372 Fig. 18a with the clamping area marked in grey. In the cross test with unclamped
373 middle-sheet, the top-sheet was loaded in mixed mode. Peeling and SLJ tests were
374 done by clamping the bottom-sheet and moving the top-sheet, constraining both
375 clamps in a straight line.

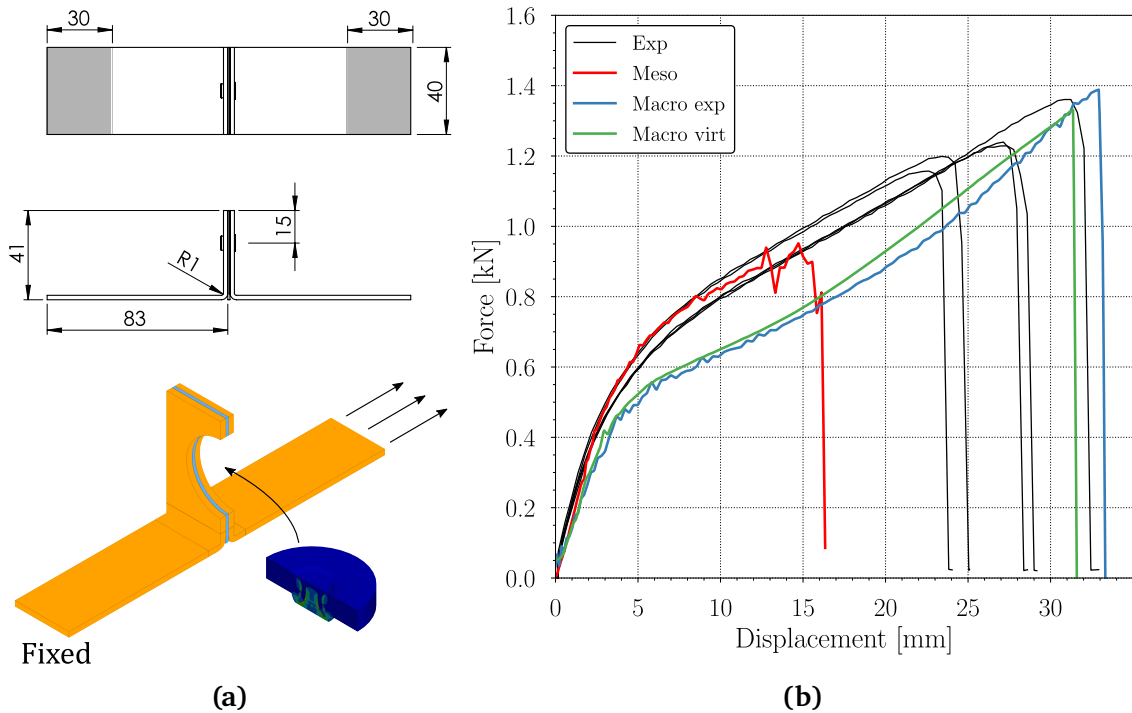


Fig. 17. Peeling tests: (a) specimen dimensions in mm, (b) force-displacement curves. Experiments “Exp”, mesoscopic model “Meso”, macroscopic model experimental calibration “Macro exp”, macroscopic model virtual calibration “Macro virt”.

376 The force-displacement curves from the peeling and SLJ test and models are dis-
 377 played in Fig. 17b Fig. 18b. Slight scatter in the tests with a peak force ranging from
 378 1.18 to 1.4 kN was observed. Displacement at failure was ranging from 24 to 32 mm.
 379 This could be related to minimal differences in the specimen manufacturing and rivet
 380 placement. The peeling specimens failed with the rivet-pull-out mode.

381 The resulting curves from the SLJ tests show a peak force ranging from 4.4 kN to
 382 5.0 kN with a displacement to failure of 2.4 to 3 mm. Also the SLJ tests failed with
 383 the rivet-pull-out mode.

384 The mesoscopic model of the peeling test showed good agreement of the initial
 385 behaviour but fails at a lower peak force of approximately 0.9 kN and displacement
 386 of 15 mm. The deviation could be explained by the smaller bottom interlock in the
 387 mesoscopic model compared to the bottom. Additionally, the peel load introduced
 388 local bending and joint rotation, which lowered the displacement at failure compared
 389 to the mixed and tension loading modes. While the mesoscopic model of the SLJ
 390 tests showed a good representation of the peak force, the displacement to failure was

391 overestimated by 0.5 mm.

392 The results from both the experimentally and the virtually calibrated macroscopic
393 models were relatively similar. The macroscopic model of the peeling tests showed
394 a lower force level in the initial loading but captured the peak force and failure dis-
395 placement well. The SLJ model showed lower stiffness, matches the peak force but
396 overestimates the failure displacement by approximately 1.7 mm.

397 Despite the deviation between the virtual cross test models and the cross test ex-
398 periments, the virtually and the experimentally calibrated macroscopic models per-
399 formed equally well in the peeling and SLJ simulations. Considering that both calib-
400 rations of the macroscopic model led to the same overall response, the general mis-
401 match between the simulations and the experiments likely stems from the modelling
402 of the SLJ experiments with shell elements. In these experiments, initially loaded in
403 pure shear, the sheets may undergo rotation and slight bending which are difficult to
capture with a shell element approach.

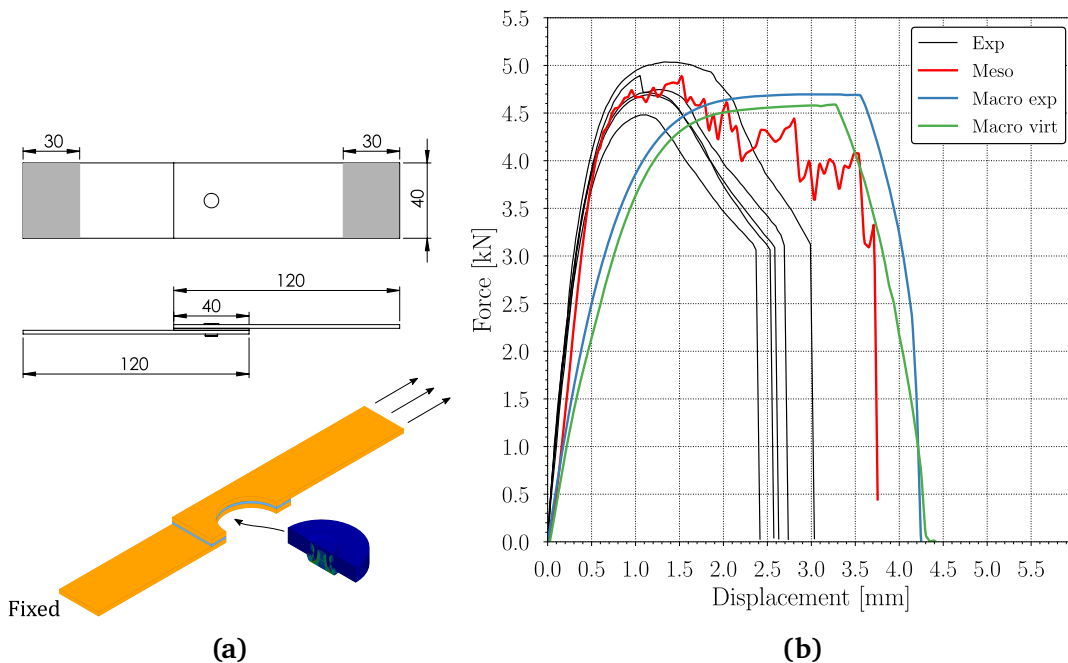


Fig. 18. SLJ tests: (a) specimen dimensions in mm, (b) force-displacement curves. Experiments “Exp”, mesoscopic model “Meso”, macroscopic model experimental calibration “Macro exp”, macroscopic model virtual calibration “Macro virt”.

404

405 *5.2. Component test and simulation*

406 This section presents a test setup for three-sheet SPR connections with a compon-
407 ent based on a design proposed by Reil et al. [32]. The component was loaded under
408 quasi-static conditions in a three-point bending setup, allowing for a subsequent fail-
409 ure of the SPR connections. Two 2 mm thick hat-profile sections were sandwiched to-
410 gether with a 1 mm flat sheet section in between and were joined with a row of eight
411 SPR along the centerline. The stack of hat profiles and middle plate was identical
412 to the investigated SPR connection in this work. The top hat-profile featured two
413 1 mm covering plates at both ends, whereas the bottom profile was covered along
414 the entire length with a 1 mm covering plate as seen in Fig. 19. The covering plates
415 were fastened with a row of M8 bolts which were used to prevent distortion of the
416 cross-section of the profile during the test.

417 The dimensions of the component specimen are shown in Fig. 19a. The three-
418 point bending test rig consisted of two lower posts and a punch with a 15 mm offset
419 from the centerline. The offset ensured the asymmetric opening of the assembly and
420 sequential rivet failure. The posts and the punch had a cylindrical shape with a radius
421 of 25 mm.

422 The macroscopic model of the component test can be seen in Fig. 19c. The sheet
423 metal parts were discretised with 2×2 mm under-integrated shell elements with
424 five integration points through the thickness. The Belytschko-Tsay element formula-
425 tion was applied together with a stiffness based hourglass control. The post and the
426 punch were modelled using a rigid body formulation. A surface-to-surface contact
427 formulation with a general friction coefficient of 0.15 was used between all contact
428 pairs. The closing plates were tied to the hat profiles as no separation of the parts
429 was observed in the tests. In addition, the M8 bolt heads were modelled using solid
430 elements tied to the sheets. The eight connector pairings (upper and lower) were
431 placed along the centerline. The work-hardening from the forming of the profiles was
432 accounted for by including the resulting plastic strains. Mass scaling was applied to
433 achieve a time-step of 3.3×10^{-4} ms with a total simulation time of 2000 ms. A 12%

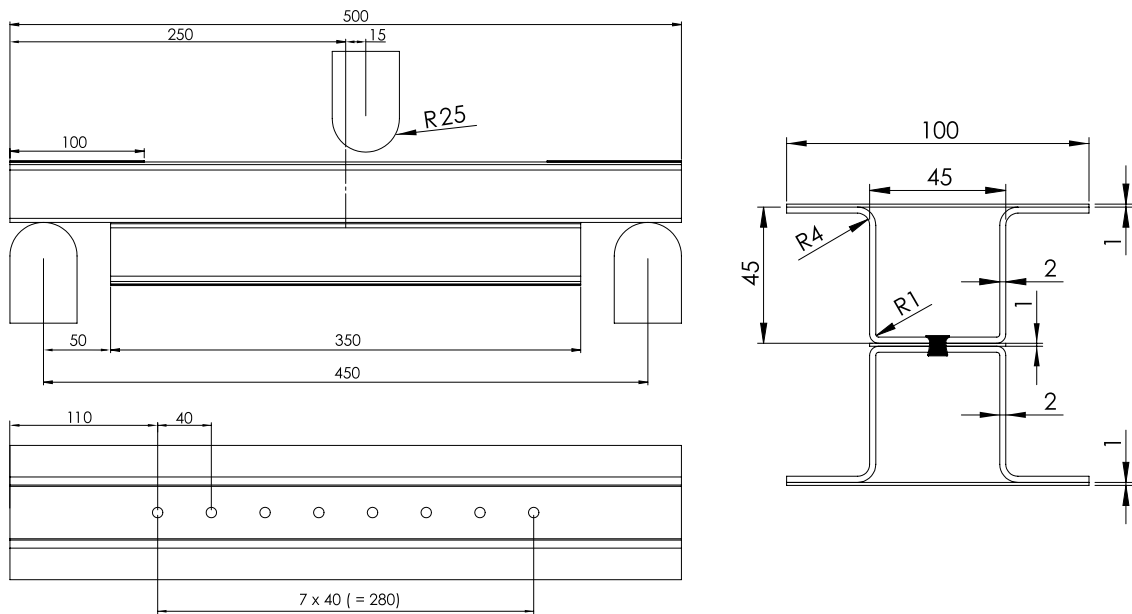
434 mass increase of the deformable components was therefore considered reasonable.

435 The force-displacement curves from the experiments and the macroscopic sim-
436 ulations are given in Fig. 20a. The test was repeated four times and all specimens
437 showed a similar gradual increase of force to approximately 8.7 kN and the same
438 overall deformation mode. Sudden failure of the first rivet at the right outermost pos-
439 ition dropped the force to approximately 6.5 kN. The force then increased slightly for
440 all tests and reached a plateau which was followed by the rapid failure of the second
441 rivet. Another force plateau was followed by the third rivet failure at approximately
442 4.8 kN, followed by a decrease to approximately 3.9 kN. The test was stopped at 65
443 mm with no failure of the remaining 5 rivets. All three rivets failed in rivet-pull-out
444 mode, see Fig. 7. That means that the bottom 2 mm sheet separated from the stack
445 while the 1 mm middle sheet stayed connected to the 2 mm top sheet via the rivet, as
446 seen in Fig. 20d. Figure 20b shows the top hat profile buckling in the punch contact
447 area but no signs of material fracture or cracks were observed.

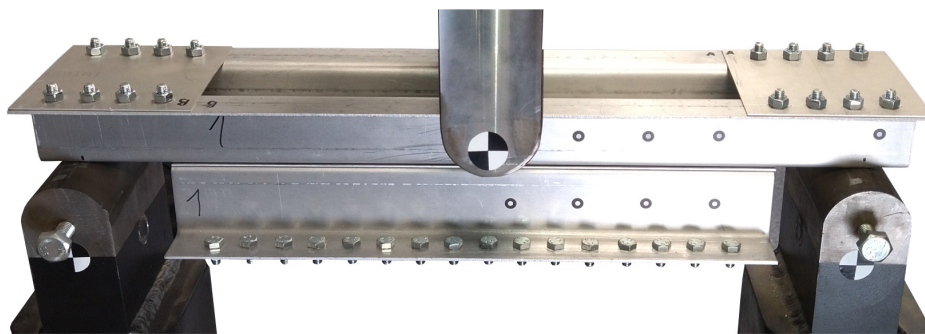
448 The macroscopic models both with the experimentally and the virtually calibrated
449 constraint models showed a good representation of the experimental behaviour. The
450 initial stiffness was well captured by both models. The experimentally-calibrated mac-
451 roscopic model is within the dispersion of the experimental curves. The virtually-
452 calibrated macroscopic model gave an error of around 5 % for the peak force, and
453 around 15 % for the displacements at failure. Both models showed the sequen-
454 tial failure of three bottom connectors, with the matching rivet-pull-out mode, as
455 seen in Fig. 20e. Failure of the first rivet using the experimentally calibrated model
456 was captured at the matching punch displacement but the second and third rivets
457 failed slightly earlier than the rivets in the experiment. The virtually calibrated model
458 showed earlier failure of the first rivet and failure of the second rivet close after. The
459 third rivet failed also earlier accordingly. Buckling of the top hat profile was in good
460 agreement with the experiments as seen in Fig. 20c.

461 The experimentally calibrated SPR models were able to give a satisfactory repres-
462 entation of the component behaviour. Also the virtually calibrated connector models

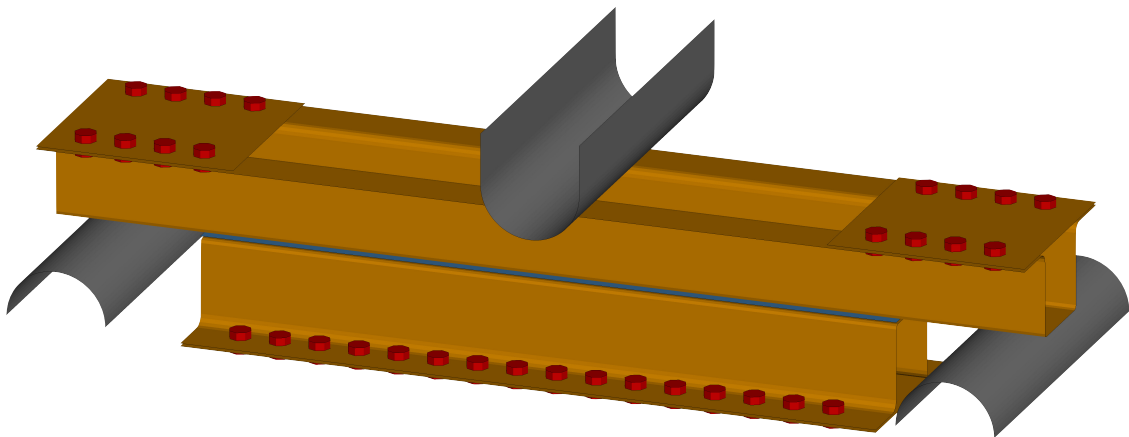
463 gave a satisfactory response of the component model, promoting the use of virtual
464 cross tests for calibration. Despite discrepancies in the cross test modelling where the
465 virtual cross test showed higher strength especially under shear loading, the virtually
466 fitted macroscopic model performed well. It should be noted that this form of com-
467 ponent test enabled a tension-dominated loading of the rivets and therefore does not
468 challenge the macroscopic models in the shear regime.



(a)

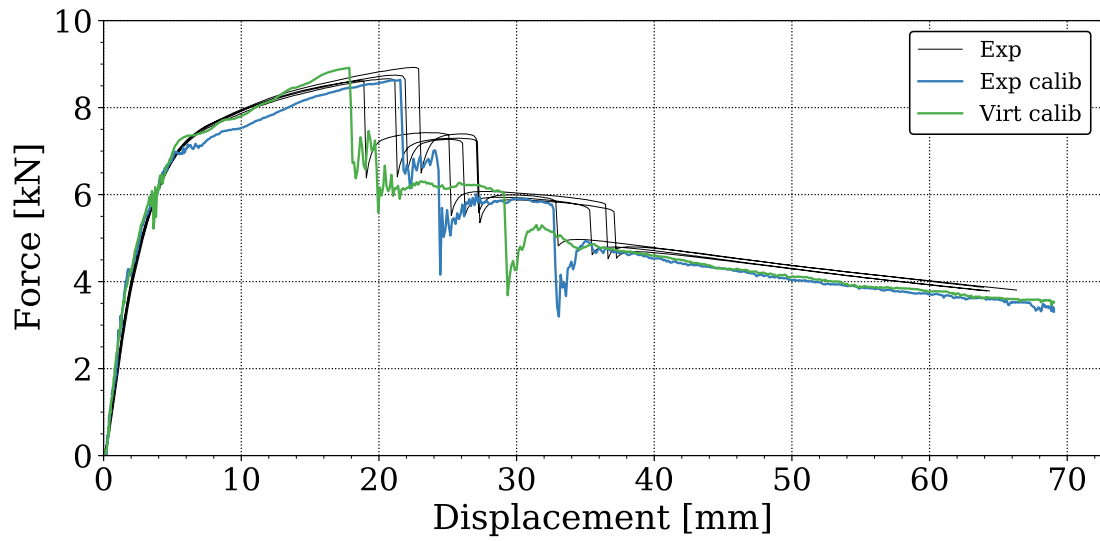


(b)

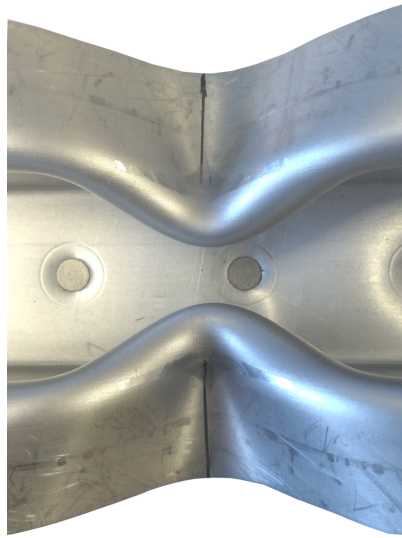


(c)

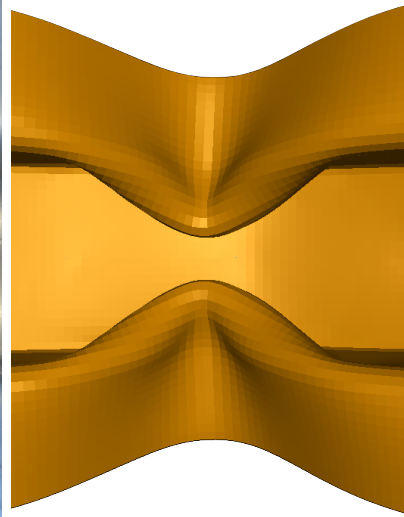
Fig. 19. Component test setup: (a) specimen dimensions in mm, (b) real test specimen and (c) macroscopic component test model.



(a)



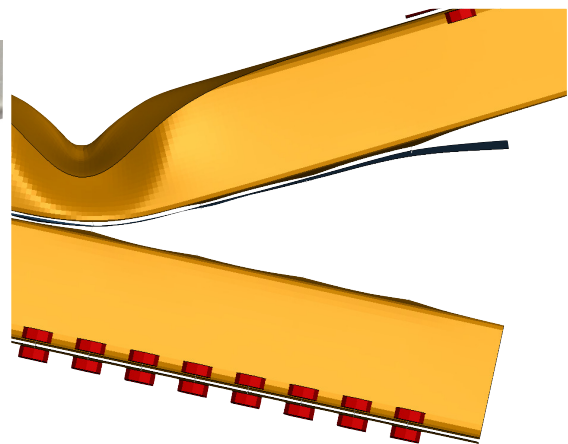
(b)



(c)



(d)



(e)

Fig. 20. Component test results: (a) force-displacement curve from experiments and simulations, (b, c) buckling below punch, (d, e) rivet failure. Experiments “Exp”, macroscopic model experimental calibration “Exp calib”, macroscopic model virtual calibration “Virt calib”.

469 6. Conclusions

470 This work investigated a virtual calibration strategy for a three-sheet SPR connec-
471 tion model with aluminium sheets. In addition, validation methods at the joint unit
472 level as well as the component level were presented. Different scales, from the riv-
473 eting process and single joint unit to a component with multiple connections, were
474 addressed while using both detailed mesoscopic and large-scale macroscopic mod-
475 elling techniques. The performance of a virtually calibrated connector model was
476 assessed. The following main conclusions are drawn:

- 477 • A state-of-the-art riveting process simulation was able to predict the geometry
478 of a three-sheet SPR joint with the process parameters adjusted to approximate
479 the physical geometry. Force measurements from the SPR machine should be
480 used in addition for adjusting the process parameters.
- 481 • Cross tests were successfully replicated with detailed mesoscopic models, al-
482 though deviations likely arose from discrepancies in the riveting process simu-
483 lation. The mismatch could be due to the limitations of the model, as neither
484 strain rate sensitivity nor temperature were accounted for.
- 485 • Despite the deviations seen in the virtual experiments, the macroscopic models
486 calibrated with them nonetheless provided reasonable predictions of the exper-
487 imental tests.
- 488 • The virtually calibrated connector models showed reliable behaviour when ap-
489 plied to the component test simulation, supporting the virtual calibration strategy
490 presented. Slightly earlier connector failure was observed in the virtually calib-
491 rated model, which is explained by a lower failure displacement visible in the
492 tensile and mixed mode cross tests.
- 493 • The virtual calibration process is based on a chain of simulations, where errors
494 in the models can add up and lead to uncertainty in the final large-scale model.
495 However, inaccuracies in the model may be offset by the decreased need for
496 extensive physical testing.

497 7. Acknowledgements

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499 Advanced Structural Analysis, SFI-CASA (Project No. 237885), funded by the Re-
500 search Council of Norway and NTNU.

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591 **Appendix A. Material modelling**

592 An isotropic plasticity model was used for modelling of the sheet and the rivet
593 material. The yield function can be written as

$$f = \sigma_{\text{eq}} - (\sigma_0 + R) \leq 0, \quad (\text{A.1})$$

594 where σ_{eq} is the equivalent stress, σ_0 the initial yield stress and R is the isotropic
595 hardening variable. For the constitutive modelling in the mesoscopic models, the
596 von Mises equivalent stress was used, described as:

$$\sigma_{\text{eq,vM}} = \left[\frac{1}{2} (|\sigma_1 - \sigma_2|^2 + |\sigma_2 - \sigma_3|^2 + |\sigma_3 - \sigma_1|^2) \right]^{\frac{1}{2}}, \quad (\text{A.2})$$

597 where the principal stresses are denoted by σ_1 , σ_2 and σ_3 . For the macroscopic mod-
598 els, the Hershey-Hosford [29] equivalent stress formulation was applied, which is a
599 generalised form of the von Mises equivalent stress equation, denoted as:

$$\sigma_{\text{eq,H}} = \left[\frac{1}{2} (|\sigma_1 - \sigma_2|^m + |\sigma_2 - \sigma_3|^m + |\sigma_3 - \sigma_1|^m) \right]^{\frac{1}{m}}, \quad (\text{A.3})$$

600 where $m = 8$ is a material dependent exponent. For both the mesoscopic and the
601 macroscopic models, the isotropic hardening of the aluminium was realised by a three-
602 term Voce hardening rule, described as:

$$R = \sum_{i=1}^N Q_{\text{R}i} \left(1 - \exp \left(-\frac{\theta_{\text{R}i}}{Q_{\text{R}i}} p \right) \right), \quad (\text{A.4})$$

603 where N is the number of terms used. $Q_{\text{R}i}$ are the final stress values where hardening
604 saturates and $\theta_{\text{R}i}$ are the initial hardening moduli. The term is indicated with i and
605 the equivalent plastic strain is denoted p . The material parameters for the aluminium
606 sheets were taken from [18] et al. The isotropic hardening of the rivet material was
607 realised by a combination of power-law and Voce hardening, resulting in a stronger
608 work-hardening at initial yielding and a linear work-hardening modulus for higher

609 strains, described by:

$$R = (Q + Hp) \cdot (1 - \exp(-cp)), \quad (\text{A.5})$$

610 where Q changes the onset of linear hardening, H is the linear work hardening mod-
611 ulus, c is the initial strain hardening gradient and p denotes the equivalent plastic
612 strain. The Hershey-Hosford yield criterion was not available for the mesoscopic mod-
613 elling (*MAT_PIECEWISE_LINEAR_PLASTICITY in LS-Dyna) and the von Mises yield
614 criterion had to be used instead.

615 **Appendix B. Constraint model**

616 The constraint model by Hansen et al. [31] defines a master and a slave sheet,
617 represented by master and slave node regions. The spatial relative deformation is de-
618 composed into a normal and tangential part, where the normal direction is orthogonal
619 to the sheets' mid-surface and the tangential direction is in-plane of the sheets.

620 The relative nodal displacements between the sheets, also called stretches, are
621 denoted δ_n (normal) and δ_t (tangential) stretch, respectively. The model calculates a
622 normal f_n and a tangential component f_t from these stretch variables. The governing
623 equations for calculation of the forces components are given in Table B.5. The model
624 calculates a damage variable named maximum effective displacement η_{\max} . The dam-
625 age evolves during joint deformation and depends on the loading mode. The damage
626 variable indicates maximum joint opening and drives joint deletion. Nine parameters
627 are needed for model calibration which are found by inverse engineering of experi-
628 mental tests in different loading directions. The moment acting on the master and
629 slave sheets is based on the force components and the sheet thicknesses. The mod-
630 els allow for free rotation around the fastener axis, since connections like SPR show
631 neglectable twisting resistance.

Table B.5. Constraint model, governing equations [31, 33].

<p>Total stretch δ defined as the vector between slave end and original location on the deformed slave sheet. Normal and tangential stretch:</p>	<p>Loading direction:</p>
$\delta = \delta_n + \delta_t, \quad \delta_n = \delta \cdot \hat{n}_m ,$ $\delta_t = \delta \cdot \hat{n}_t , \quad \hat{n}_t = \hat{n}_0 \times \hat{n}_m$	$\theta = \arctan\left(\frac{\delta_t}{\delta_n}\right)$
<p>Forces are calculated directly from mathematical expressions:</p>	<p>Damage variables:</p>
$f_n = \frac{f_n^{\max} \delta_n}{\eta_{\max} \delta_n^{\text{fail}}} \hat{f}_n, \quad f_t = \frac{f_t^{\max} \delta_t}{\eta_{\max} \delta_t^{\text{fail}}} \hat{f}_t,$	$\eta = \left[\xi + \frac{1-\xi}{\alpha} \sqrt{\left(\frac{\delta_n}{\delta_n^{\text{fail}}}\right)^2 + \left(\frac{\delta_t}{\delta_t^{\text{fail}}}\right)^2} \right]$
<p>where</p>	
$\hat{f}_n = \begin{cases} 1 - \left(\frac{\xi_n - \eta_{\max}}{\xi_n}\right)^8, & \eta_{\max} \leq \xi_n \\ 1 - \frac{\eta_{\max} - \xi_n}{1 - \xi_n}, & \eta_{\max} > \xi_n \end{cases}$	$\xi = 1 - \frac{27}{4} \left(\frac{2\theta}{\pi}\right)^2 + \frac{27}{4} \left(\frac{2\theta}{\pi}\right)^3$
$\hat{f}_t = \begin{cases} 1 - \left(\frac{\xi_t - \eta_{\max}}{\xi_t}\right)^8, & \eta_{\max} \leq \xi_t \\ 1 - \frac{\eta_{\max} - \xi_t}{1 - \xi_t}, & \eta_{\max} > \xi_t \end{cases}$	$\alpha = \begin{cases} \frac{\xi_t - \eta_{\max}}{\eta_t} \alpha_1 + \frac{\eta_{\max}}{\eta_t} \alpha_2, & \eta_{\max} < \xi_t \\ \frac{1 - \eta_{\max}}{1 - \xi_t} \alpha_2 + \frac{\eta_{\max} - \xi_t}{1 - \xi_t} \alpha_3, & \eta_{\max} \geq \xi_t \end{cases}$
<p>Exponent of 8 suggested by Hanssen et al. [31] but can be changed in the LS-DYNA solvers.</p>	