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1 **Coexistence of natural and forced vortex dislocations in step cylinder flow**

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6 The wake behind a step cylinder (consisting of a small diameter cylinder ( $d$ ) and a large  
 7 diameter cylinder ( $D$ )) with diameter ratio  $D/d = 2$  at Reynolds number  $Re_D=200$  (the  
 8 mode  $A^*$  regime) is simulated by direct numerical simulations. New detailed information  
 9 of the interaction between natural vortex dislocations and forced vortex dislocations is de-  
 10 scribed. In the large cylinder wake, the forced and natural vortex dislocations coexist. The  
 11 regular formation of forced vortex dislocation is found to be delayed under the effect of  
 12 natural vortex dislocations. The occurrence of natural vortex dislocations is suppressed in  
 13 the large cylinder wake close to the small cylinder. Moreover, the effect of vortex disloca-  
 14 tions on structural loads are described. The results in this paper provide a more thorough  
 15 understanding of the formation and interaction between the natural and forced vortex dis-  
 16 locations.

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17 **I. INTRODUCTION**

18 As a fundamental transition feature in cylindrical structural wakes, vortex dislocations are usu-  
19 ally referred to as the flow region where the neighboring spanwise vortices move out of phase due  
20 to different shedding frequencies<sup>1</sup>. There are mainly two types of vortex dislocations: Natural  
21 vortex dislocation (NVD) and forced vortex dislocation (FVD).

22 Natural vortex dislocations form naturally when bluff-body wakes exhibit three-dimensional  
23 transitions. For example, the wake behind a circular cylinder becomes three-dimensional when  
24 the Reynolds number ( $Re_D = UD/\nu$  where  $U$  represents the uniform inflow velocity,  $D$  is the  
25 cylinder diameter, and  $\nu$  is the kinematic viscosity of the fluid) exceeds around 190, due to the  
26 mode A instability, originating from the elliptic instability<sup>2</sup>. A stable state of mode A occurs  
27 first at  $190 < Re < 193^{2,3}$  and is characterized by two counter-rotating streamwise vortices forming  
28 a mode A vortex loop with a spanwise wavelength around  $4D$ . At  $193 < Re < 230$ , this flow  
29 becomes unstable and is denoted mode A\*. Here an intermittent large-scale vortex dislocation  
30 (also called the spot-like vortex dislocation<sup>4</sup>) occurs randomly in the wake both in time and over  
31 the spanwise position<sup>2,5-7</sup>. As  $Re$  further increases to above 230, the wake flow transforms to  
32 mode B<sup>2,8</sup> and further on towards turbulence<sup>9,10</sup> where the natural vortex dislocation still exists  
33 but with a decreased probability and duration of occurrence<sup>10</sup>.

34 Forced vortex dislocations occurs in the wake behind a nonuniform geometry, such as step  
35 cylinders<sup>12,13</sup>, ring-attached cylinders<sup>4</sup>, and cylinders with end effects<sup>1,14,15</sup>. In most of these  
36 cases, several spanwise vortices occur in the wake due to the spanwise disturbance induced by  
37 the geometrical non-uniformity. Due to the different resulting shedding frequencies, the phase  
38 difference between two neighboring vortex cells will continuously accumulate, yielding the forced  
39 vortex dislocations. This process occurs regularly at a beat frequency ( $f_{beat} = f_1 - f_2$ ) between the  
40 frequencies of the two neighboring vortices (where  $f_1$  and  $f_2$  represent the vortex with the higher  
41 and lower shedding frequency, respectively). Detailed investigations<sup>5,7,16,17</sup> of the near wake show  
42 that both natural and forced vortex dislocation induce a decrease in the vortex shedding frequency,  
43 an increase in the base pressure, and an increase in the fluctuation amplitude of spanwise flow in  
44 the surrounding flow region.

45 The step cylinder sketched in figure 1(a) is convenient for investigating vortex dislocations  
46 due to its simple geometry and multiple spanwise vortices in the wake. For a step cylinder with  
47  $D/d > 1.55$  at  $63 < Re_D < 3900$ , three dominating spanwise vortices have been observed in the wake

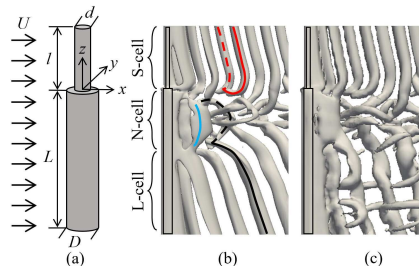
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FIG. 1. (a) A sketch of the step cylinder geometry. The diameters of the small and large cylinders are  $d$  and  $D$ , respectively.  $l$  is the length of the small cylinder, and  $L$  is the length of the large cylinder. The origin locates at the center of the interface between the small and large cylinders. The uniform incoming flow  $U$  is in the positive  $x$ -direction. The three directions are named streamwise ( $x$ -direction), crossflow ( $y$ -direction), and spanwise ( $z$ -direction). (b) The isosurfaces of  $\lambda_2 = -0.05^{18}$  shows the instantaneous wake behind a step cylinder with  $D/d = 2$  at  $Re_D = 150$  from Tian *et al.*<sup>13</sup>, taken at the moment when vortex dislocations occur. The SS-half loop and NL-loop structures are denoted by the red and black curves, where the solid and dashed curves represent the vortex shed from the  $-Y$  and  $+Y$  sides of the step cylinder, respectively. (c) The isosurfaces of  $\lambda_2 = -0.05$  shows the instantaneous wake behind a step cylinder with  $D/d = 2$  at  $Re_D = 200$  from the present study.

48 in previous studies<sup>12,13,19–21</sup>: (i) The S-cell vortex behind the small cylinder with the highest shed-  
 49 ding frequency  $f_S$ , (ii) the L-cell vortex sheds from the large cylinder with the lower shedding  
 50 frequency  $f_L$ , and (iii) the N-cell vortex located between the S- and L-cell vortices with the lowest  
 51 shedding frequency  $f_N$ . The forced vortex dislocation (FVD) occurs periodically between neigh-  
 52 boring vortex cells. When FVD occurs between the S- and N-cell vortices, the connection between  
 53 the corresponding S- and N-cell vortices is broken; two S-cell vortices with opposite rotating di-  
 54 rections connect (i.e., the S-S half loop vortex structure forms<sup>19,20</sup>, as denoted by the red curve in  
 55 figure 1(b)). When FVD starts to occur between the N- and L-cell vortices, one L-cell vortex (e.g.,  
 56 the one denoted by the solid black curve in figure 1(b)) dislocates with its corresponding N-cell  
 57 vortex (as denoted by the blue solid curve in figure 1(b)) and connects to the N-cell vortex on the  
 58 other side of the step cylinder (as denoted by the black dashed curve in figure 1(b)), forming the  
 59 NL-loop structure. These NL vortex loops will be focused on in the present paper as they indi-  
 60 cating the formation of vortex dislocations between the N- and L-cell vortices. As FVD occurs, a

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61 series of NL-loop structures, NN-loop structure, and LL-half loop structure forms. More detailed  
 62 information is referred to Tian *et al.*<sup>13,21</sup>. From an engineering application point of view, it was  
 63 found in previous studies<sup>5,14,17</sup> that as the spanwise coherent vortex structures is destroyed by the  
 64 formation of vortex dislocations, the magnitude of the structural loads (drag and lift) is decreased.

65 Although previous studies have widely investigated the flow characteristics of the natural vortex  
 66 dislocation and the forced vortex dislocation separately, a detailed investigation of the interaction  
 67 between the natural and forced vortex dislocation is still absent. The primary aim of the present  
 68 paper is to investigate the flow interactions when the forced and natural vortex dislocations coexist  
 69 and how such interactions affect the structural loads. To achieve this, the flow past a step cylinder  
 70 with  $D/d=2$  at  $Re_D=200$  (in the mode A\* regime) is studied using a well-validated Direct Nu-  
 71 merical Simulation (DNS) code MGLET<sup>22</sup> to directly solve the three-dimensional Navier-Stokes  
 72 equations.

## 73 II. GOVERNING EQUATIONS, FLOW CONFIGURATION AND COMPUTATIONAL 74 ASPECTS

75 The incoming flow  $U$  is uniform in the positive  $x$ -direction. The diameter ratio of the step  
 76 cylinder is given as 2.0. The Reynolds number is  $Re_D=200$ , based on the diameter of the large  
 77 cylinder. The origin of the coordinate system is at the step as shown in figures 1. The incom-  
 78 pressible flow is governed by the continuity equation and the time-dependent three-dimensional  
 79 incompressible Navier-Stokes equation:

$$80 \quad \nabla \cdot \mathbf{u} = 0, \quad (1)$$

$$81 \quad \frac{\partial \mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla) \mathbf{u} = \nu \nabla^2 \mathbf{u} - \frac{1}{\rho} \nabla p, \quad (2)$$

82 where  $\mathbf{u}$  is the velocity vector, while  $\rho$ ,  $p$ , and  $t$  denote the constant density, pressure, and time,  
 83 respectively.

84 A finite-volume numerical code MGLET<sup>22</sup> is used to conduct Direct numerical simulations  
 85 (DNS). This code has been thoroughly validated in previous works for various applications, for  
 86 example, the flow around step cylinders<sup>13,23</sup>, the flow past two tandem cylinders<sup>24,25</sup>, and the  
 87 oscillatory flow through a hexagonal sphere pack<sup>26</sup>. A staggered numerical grid is applied, where  
 88 the pressure is located in the middle of the grid cell, and the velocities are evaluated in the middle  
 89 of the grid face. The step cylinder geometry is handled by an immersed boundary method<sup>27,28</sup>.

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91 A third-order Runge-Kutta scheme<sup>29</sup> is applied for the time integration. A constant time step  $\Delta t$   
 92 is used to ensure a CFL (Courant-Friedrichs-Lewy) number smaller than 0.5. Stone's implicit  
 93 procedure (SIP)<sup>30</sup> is applied to solve the elliptic pressure correction equation.

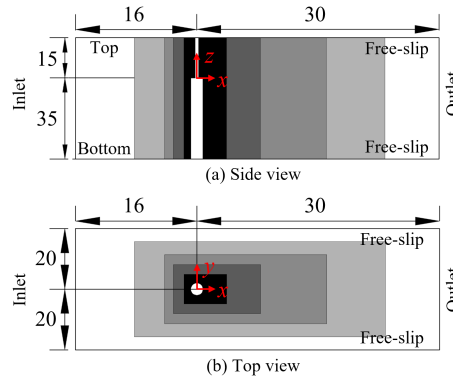


FIG. 2. Computational domain and coordinate system: (a) Side view, (b) Top view. The three directions are named streamwise ( $x$ -direction), crossflow ( $y$ -direction) and spanwise ( $z$ -direction). The grid refinement regions are schematically illustrated and marked with darker shades of grey for finer regions. The length unit is the cylinder diameter  $D$ .

94 Figure 2 shows the computational domain and the coordinate system. The inlet and outlet are  
 95 placed  $16D$  and  $30D$  away from the cylinder axis, respectively. The distance between the top and  
 96 bottom is  $50D$ , where the length of the small and large cylinders is  $15D$  and  $35D$ , respectively.  
 97 The computational domain applied here is comparable to those used in the previous studies<sup>20</sup>  
 98 for modeling the step cylinder wake at  $Re_D=300$ . At the inlet, a constant velocity profile ( $u=U$ ,  
 99  $v=w=0$ ) is applied. At the outlet, a Neumann condition ( $\partial u/\partial x = \partial v/\partial x = \partial w/\partial x = 0$ ) is applied.  
 100 A free-slip boundary condition is applied for the other four sides of the computational domain  
 101 (for the two vertical sides  $v = 0$ ,  $\partial u/\partial y = \partial w/\partial y = 0$ ; for the two horizontal sides:  $w = 0$ ,  
 102  $\partial u/\partial z = \partial v/\partial z = 0$ ). Neumann conditions are applied for the pressure, except at the outlet where  
 103 the pressure is set equal to zero. A no-slip and impermeability condition ( $u = v = w = 0$ ) is  
 104 applied on the surface of the step cylinder through an immersed boundary method<sup>27</sup>. Around  
 105 the cylinder, a local grid refinement is achieved by embedding zonal grids<sup>22</sup>. Figure 2 shows a  
 106 schematic illustration of the grid design, where the darker shades represent the finer grid regions.  
 107 The grid is equally sized in the  $x$ -,  $y$ -, and  $z$ -directions within each grid region; the ratio of

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108 the grid size between the neighboring regions is equal to two. Two cases with the finest grid  
 109 resolution  $0.0125D$  (G0125) and  $0.01D$  (G01) have been simulated to conduct a grid convergence  
 110 study. Table I shows the Strouhal number of the three dominating vortex cells ( $St_S = f_S D/U$ ,  
 111  $St_N = f_N D/U$ , and  $St_L = f_L D/U$ ) behind the step cylinder obtained by a fast Fourier transform  
 112 (FFT) of the time series of the streamwise velocity  $u$  along a vertical sampling line positioned  
 113 at  $(x/D, y/D)=(1.6, 0.4)$ . The differences in  $St$  of the same vortex cell between these two cases  
 114 is small. Figure 3 shows the time-averaged streamwise velocity distribution along the line AB  
 115 (as indicated in inset (a2)), illustrating the flow variation above the step surface just in front  
 116 of the small cylinder. The zoom-in view in the inset figure 3(a) shows that the results from the  
 117 two cases almost coincide. All flow characteristics and the underpinning mechanisms discussed  
 118 in the present paper are valid for grid resolutions  $0.0125D$  and  $0.01D$ . Moreover, based on the  
 119 same numerical code MGLET and the same grid resolution  $0.01D$ , Jiang et al.<sup>31</sup> obtained novel  
 120 results in the simulations of flow past a curved cylinder at a higher Reynolds number  $Re = 600$ .  
 121 Therefore, we are convinced that the grid resolution  $0.01D$  is sufficiently fine for the present study.  
 122 The results presented in the following are from the case with the grid resolution  $0.01D$ .

TABLE I. Strouhal numbers of the three dominating vortex cells (S-cell,  $St_S = f_S D/U$ ; N-cell,  $St_N = f_N D/U$ ; L-cell,  $St_L = f_L D/U$ ).

Vortex cell	G0125	G01
$St_S$	0.331	0.332
$St_N$	0.164	0.163
$St_L$	0.185	0.184

### 123 III. NATURAL AND FORCED VORTEX DISLOCATIONS

124 Figure 1(c) shows an overview of the vortex structures in the wake behind the step cylinder with  
 125  $D/d=2$  at  $Re_D=200$  by plotting isosurfaces of  $\lambda_2=-0.05$ <sup>18</sup>. The three spanwise vortex cells  
 126 observed in the previous studies<sup>19,20,23</sup> (as shown in figure 1(b))  
 127 are also observed in the present study (as shown in figure 1(c)). The corresponding frequency  
 128 components  $St_S = f_S D/U$ ,  $St_N = f_N D/U$ , and  $St_L = f_L D/U$  for the S-, N-, and L-cell vortices

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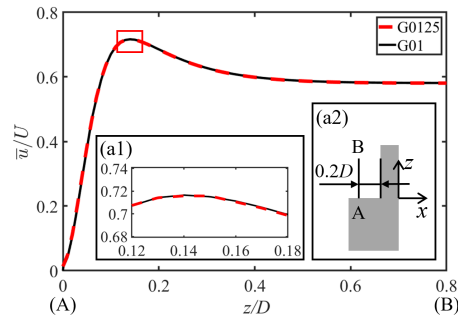


FIG. 3. Distribution of mean streamwise velocity  $\bar{u}/U$  along a vertical sampling line AB in the  $xz$ -plane at  $y/D=0$ . Insets: (a1) a zoomed-in view of the upper part of the curves (red rectangle); (a2) a sketch of the line AB positioned at  $x/D=-0.45$  with a length of  $0.8D$ .

129 are presented in the crossflow velocity spectrum shown in figure 4. Figures 4(a, b) show that the  
 130 spectrum peak in the N-cell region ( $-4 < z/D < 0$ ) and the S-cell region ( $z/D > 0$ ) is more slim  
 131 and dominating than in the L-cell region ( $z/D < -4$ ), due to the regular shedding in the S- and  
 132 N-cell region and the irregular shedding in the L-cell region. The shedding of the S- and N-cell  
 133 vortices and the interactions between them are similar to that discussed in the step cylinder case  
 134 at  $Re_D=150$ <sup>13,21</sup> and  $Re_D=300$ <sup>20</sup>. The irregular shedding of the L-cell vortex is caused by the  
 135 coexistence of the forced vortex dislocation (FVD) and natural vortex dislocation (NVD) in the  
 136 large cylinder wake; the formation of the natural vortex dislocation in the L-cell region changes  
 137 the shedding frequency in the surrounding region and time interval. Our observations show that,  
 138 in contrast to the regular formation of FVD between the N- and L-cell vortices in the  $Re_D = 150$   
 139 and  $Re_D = 300$  cases, the coexistence of FVD and NVD in the present  $Re_D = 200$  case is found to  
 140 be able to delay (but not accelerate) the regular formation of FVD under certain circumstances; a  
 141 detailed discussion will be given below.

142 Figure 5 shows snapshots of instantaneous vortex structures by plotting isosurfaces of  $\lambda_2 =$   
 143  $-0.05$ . In figures 5(a, b, c), the solid and dashed red curves represent the vortex on the -Y and  
 144 +Y sides of the step cylinder, respectively. Due to the different dominating shedding frequencies  
 145 of the N- and L-cell vortices, the forced vortex dislocation occurs periodically between the N-  
 146 and L-cell vortices at  $z/D \approx -4$  around every 48 time units ( $D/U$ ) from  $tU/D=611.1$  (figure 5(a))  
 147 to  $tU/D=708.1$  (figure 5(c)). The L-cell vortex (the red dashed line at  $-10 < z/D < -4$ ) dislocates  
 148 with its counter N-cell vortex (the red dashed line at  $-4 < z/D < 0$ ) on the same cylinder side and



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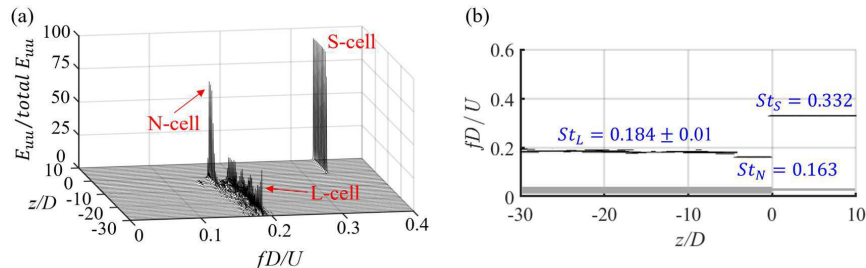


FIG. 4. (a) Streamwise velocity spectra is obtained from a discrete Fourier transform (DFT) of time series of the streamwise velocity  $u$  along a vertical sampling line positioned at  $(x/D, y/D)=(1.6, 0.4)$  over 1500 time units ( $D/U$ ) showing the three dominating vortex cells (the S-, N-, and L-cell vortices). (b) Projection of the 3D plot in (a) into the horizontal plane. Only the points with  $E_{uu}/(total E_{uu}) > 7\%$  are shown, indicating the extension and frequency ( $St_S$ ,  $St_N$ , and  $St_L$  for the S-, N-, and L-cell vortices, respectively).

149 connects to the N-cell (the solid red line at  $-4 < z/D < 0$ ) on the other side cylinder side, forming the  
 150 NL-loop structure. The time interval between these FVDs fits well with the period ( $T_{VD} = 1/f_{beat}$ )  
 151 corresponding to the beat frequency<sup>1</sup> ( $f_{beat} = f_L - f_N$ ). Besides the forced vortex dislocation, the  
 152 one-side and two-side natural vortex dislocation (identified in Tian *et al.*<sup>32</sup>) is also captured here.  
 153 In figures 5(a, c, d), the two vortices marked by the solid green lines simultaneously dislocate  
 154 with the vortex located in-between and marked by the green dashed line, i.e., the two-side NVD.  
 155 In figures 5(b, c, d), the vortices marked by the solid and dashed blue lines dislocate with each  
 156 other, i.e., the one-side NVD. Although the formation and position of these two types of NVD  
 157 are irregular, the regularity of the formation position and period of the FVD from  $tU/D=611.1$  to  
 158  $659.6$  shown in figures 5(a, b) is not affected. During this time interval, no NVD occurs in the  
 159 region  $z/D > -7$ , i.e., the region close to the formation position of FVD at  $z/D=-4$  (i.e., the NL-  
 160 boundary). However, this is not always the case. When NVD do occur in the region  $z/D > -7$ ,  
 161 the formation of the subsequent FVD will be delayed. An example is shown in figure 6: The time  
 162 interval between the FVDs shown in figures 6(a) and (b) is  $80D/U$ , which is much larger than that  
 163 between the FVDs shown in figures 5(b, c, d). From  $tU/D=756.6$  (figure 6(a)) to  $tU/D=834.1$   
 164 (figure 6(b)), a series of NVDs occur at  $z/D > -7$ ; several of them are presented in figures 6(c,  
 165 d, e). The underpinning mechanism is discussed in the forthcoming. It is worth emphasizing that

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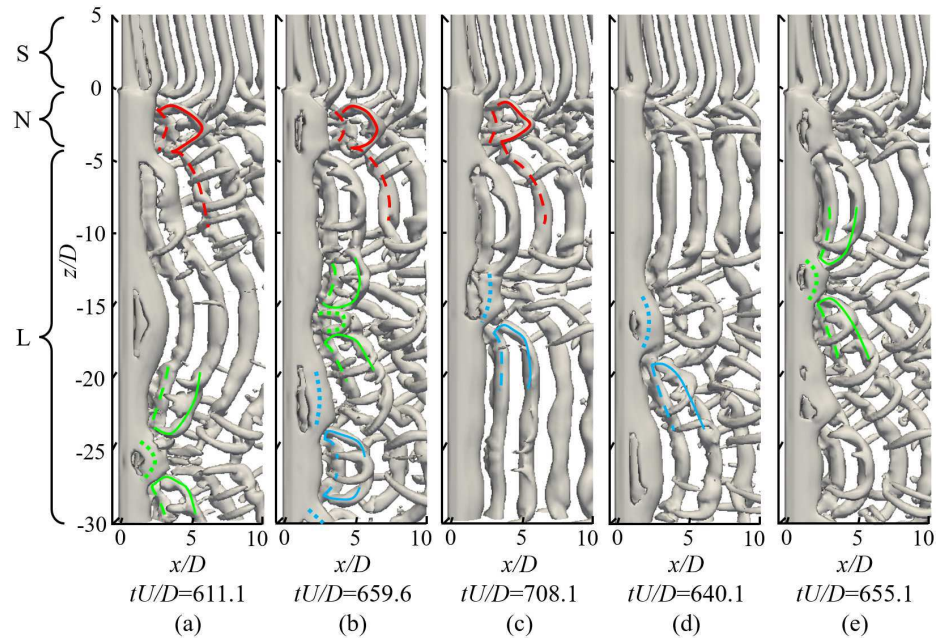


FIG. 5. Isosurfaces of  $\lambda_2=-0.05^{18}$  illustrating the vortex structures in the wake: (a)  $tU/D=611.1$ ; (b)  $tU/D=659.6$ ; (c)  $tU/D=708.1$ ; (d)  $tU/D=640.1$ ; (e)  $tU/D=655.1$ . The flow region covered by the three vortex cells (the S-, N-, and L-cell) is shown. The NL-loop vortex structure forms when the forced vortex dislocation occurs between the N- and L-cell vortices; is denoted by the red curve. The blue and green curves indicate the one-side and two-side natural vortex dislocation<sup>32</sup>, respectively.

166 the identification of the formation of FVD and NVD can be achieved by using the number of the  
 167 neighboring vortices and the formation of the streamwise vortices. The identification will not be  
 168 affected by the selection of the  $\lambda_2$  value. The corresponding detailed information can be found in  
 169 Tian *et al.*<sup>13,21,32</sup>. Furthermore, the hand sketches of the vortex structure topology agree well with  
 170 the vortex rotation axis line obtained by Liutex method<sup>33</sup>. An example is shown in appendix A.

171 The time history of the shedding frequencies for the regular forced vortex dislocations shown  
 172 in figures 5(a, b) and the delayed forced vortex dislocations shown in figures 6(a, b) are presented  
 173 in the left (regular FVD) and right (delayed FVD) column of figure 7. The shedding frequency  
 174 is estimated from the cross-flow velocity component ( $v/U$ ). Figures 7(a, b) are obtained at the  
 175 sampling point  $(x/D, y/D, z/D)=(0.6, 0, -2)$  which is located in the middle of the N-cell region

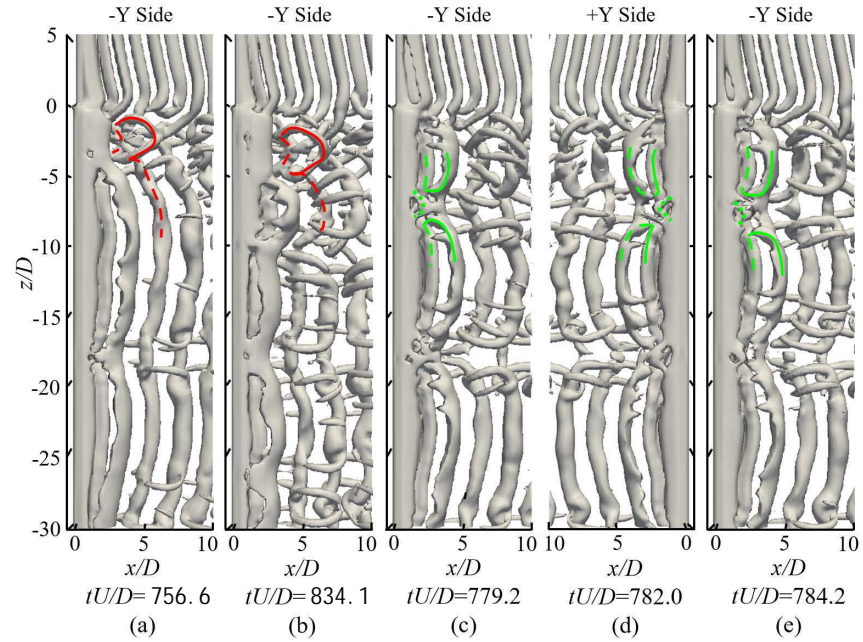
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FIG. 6. Isosurfaces of  $\lambda_2 = -0.05^{18}$  illustrating the vortex structures in the wake : (a)  $tU/D = 756.6$ ; (b)  $tU/D = 834.1$ ; (c)  $tU/D = 779.2$ ; (d)  $tU/D = 782.0$ ; (e)  $tU/D = 784.2$ . The same types of curves applied in figure 5 are also used here to denote the forced and natural vortex dislocations.

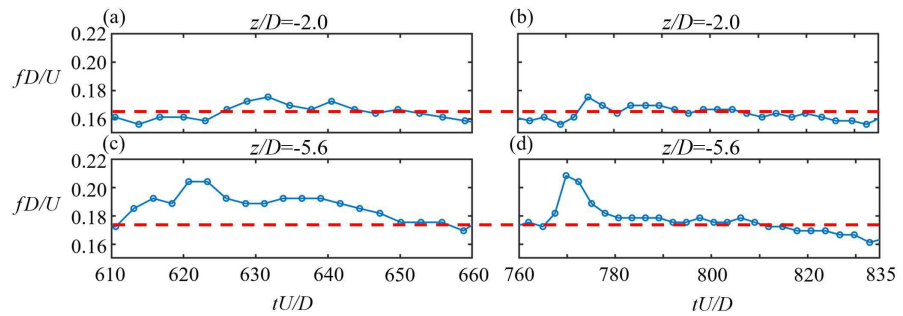
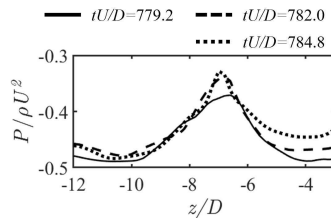


FIG. 7. Time history of the vortex shedding frequency  $fD/U$ : (a, b) at  $z/D = -2.0$  and (c, d) at  $z/D = -5.6$ ; where (a, c) are obtained during the regular forced vortex dislocations shown in figures 5(a, b) and (b, d) are obtained during the delayed forced vortex dislocations shown in figures 6(a, b). The red dashed line is plotted to ease the comparison of the time history between the left and right subplots.

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176 as shown in figure 5(a), representing the shedding frequency of the N-cell vortex  $f_N$ . Figures 7(c),  
 177 d) are obtained at the sampling point  $(x/D, y/D, z/D)=(0.6, 0, -5.6)$  which is located in the L-cell  
 178 region close to the NL-cell boundary and the natural vortex dislocation position as shown in fig-  
 179 ures 6(b, c, d). This frequency-capture method was also applied in Behara and Mittal<sup>14</sup> and Tian  
 180 et al.<sup>32</sup>. The time history of the frequency in figures 7(a) and (b) behave similarly, indicating that  
 181 the shedding frequency in the N-cell region is rarely affected by the natural vortex dislocation.  
 182 However, the time history of the shedding frequency shown in figure 7(d) is overall lower than  
 183 that shown in figure 7(c), implying that the shedding frequency of the L-cell vortex at  $z/D = -5.6$   
 184 clearly decreases during the natural vortex dislocation period. It is known that, in the wake of  
 185 cylindrical structures, the vortex dislocation (both FVD and NVD) between neighboring spanwise  
 186 vortices is caused by the accumulation of the phase difference between these vortices<sup>2,13,32</sup>. In the  
 187 present case, due to the shedding frequency for the N-cell vortex being lower than for the L-cell  
 188 vortex, the accumulated phase difference between the N- and L-cell vortices will cause FVD to  
 189 form between them. When  $f_N$  and  $f_L$  are constant, the period of FVD is constant and equal to  
 190  $T_{VD} = 1/(f_L - f_N)$ . However, as  $f_L$  decreases in the region close to the NL-boundary (e.g.,  $f_L$   
 191 shown in figure 7(d)) due to the formation of natural vortex dislocations (e.g., the NVDs shown  
 192 in figure 6(c, d, e)), the corresponding  $T_{VD}$  increases (see the extended time interval between the  
 193 delayed FVDs shown in figures 6(a, b)). Only the formation of NVD at  $z/D > -7$  can affect the



194 FIG. 8. The spanwise distribution of the instantaneous pressure when NVD occurs at  $z/D \approx -7$  in figures  
 195 6(c, d, e). The solid and dotted curves are along the sampling line at  $(x/D, y/D)=(0.6, -0.2)$ ; the dashed  
 196 curve is at  $(x/D, y/D)=(0.6, 0.2)$ .

196 formation of FVD. This is because the suction pressure decreases locally when NVD occurs. Fig-  
 197 ure 8 shows the spanwise distribution of the pressure at  $(x/D = 0.6, y/D = 0.2)$  when the natural  
 198 vortex dislocation occurs in figures 6(c, d, e). It appears that the suction pressure decreases to the  
 199 local extreme at  $z/D = -7$  where two NVDs occur in figures 6(c, d, e); the corresponding affected

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200 region spans 2–3 diameters ( $D$ ) around the dislocation position. As pointed out in Williamson<sup>1</sup>,  
 201 a decrease in the suction (i.e. an increase in the base pressure) can enlarge the vortex formation  
 202 region locally and further decrease the vortex shedding frequency; the formation of NVD at a  
 203 given location can decrease the vortex shedding frequency in a surrounding region covering 4–6  
 204 diameters in the spanwise direction. When NVD occurs at  $z/D < -7$ , it has limited effect on the  
 205 N- and L-cell vortices around  $z/D = -4$  where FVD occurs, i.e., the regular formation of FVD  
 206 will not be affected.

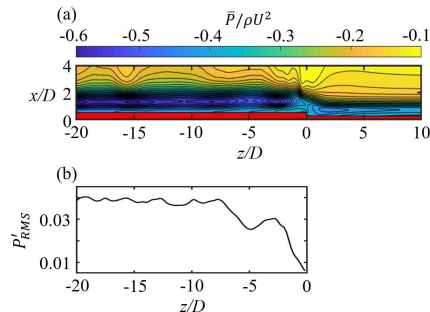


FIG. 9. (a) Time-averaged pressure contour in the  $xz$ -plane at  $y/D=0$ . (b) Spanwise distribution of the fluctuated base pressure along the large cylinder at  $(x/D, y/D)=(0.6, 0)$ .

207 As shown in appendix B, long-time observations based on our numerical simulations reveal that  
 208 NVD can only delay, not accelerate, the formation of FVD in the present case. This is because no  
 209 natural vortex dislocation forms in the N-cell region ( $-4 < z/D < 0$ ), i.e., the shedding frequency of  
 210 the N-cell vortex is not decreased by the formation of the NVD. When only  $f_L$  can be decreased due  
 211 to the NVD, the vortex dislocation period ( $T_{VD}$ ) can only be increased, not decreased, based on the  
 212 equation  $T_{VD} = 1/(f_L - f_N)$ . The absence of NVD in the N-cell region could be due to the effect of  
 213 the weak-suction region behind the small cylinder. Figure 9(a) shows the time-averaged pressure  
 214 contours in the  $xz$ -plane at  $y/D=0$ , indicating that far from the step, the weak-suction region (the  
 215 yellow region) is located closer to the small cylinder than the large cylinder due to the different  
 216 diameters. Close to the step ( $z/D=0$ ), the wakes are mixed behind the small and large cylinders.  
 217 Consequently, the suction pressure in the vicinity of the step decreases on a region behind the  
 218 large cylinder  $-4 < z/D < 0$  (see figure 9(a)) under the effect of the weak-suction region behind the  
 219 small cylinder, which also suppresses the flow instability in this region: Figure 9(b) shows the

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220 distribution of the root-mean-square of the fluctuating base pressure ( $p'_{RMS} = \sqrt{\frac{1}{N} \sum_{i=1}^N (p_i - \bar{p})^2}$   
 221 where  $N$  is the number of values in the sample) along a line behind the large cylinder at  $(x/D,$   
 222  $y/D)=(0.6, 0)$ , indicating that the value of  $p'_{RMS}$  is clearly smaller in the region  $-4 < z/D < 0$  than in  
 223 the rest of the region along the large cylinder. As the base pressure instability is suppressed, the  
 224 vortex shedding frequency in the N-cell region becomes more regular than that in the L-cell region  
 225 (as shown in figure 4), further suppressing the formation of the natural vortex dislocation in the  
 226 N-cell region  $-4 < z/D < 0$ .

227 In general, the period of the forced vortex dislocations between the slower shedding N-cell  
 228 vortex and faster shedding L-cell vortex behind a step cylinder is determined by the difference  
 229 between their shedding frequencies, i.e.,  $T_{VD} = 1/(f_L - f_N)$ . The formation of the forced vor-  
 230 tex dislocation is observed to be delayed when NVD occurs at  $z/D > -7$  (close to the NL-cell  
 231 boundary) due to the corresponding decrease of  $f_L$ . Since the pressure instabilities in the N-cell  
 232 region are suppressed by the weak suction pressure behind the small cylinder, NVD can not oc-  
 233 cur in the N-cell region, and thus NVD can only delay, not accelerate, the formation of FVD in  
 234 the present case. This delay effect has also been observed in flow past a circular cylinder with  
 235 a downstream sphere (figure 17(b) in Zhao<sup>34</sup>), where the downstream sphere plays a similar role  
 236 as the small cylinder does in the present case, leading to that NVD can only occur in the region  
 237 where the faster-shedding vortex is located. It should also be noted that although only the de-  
 238 lay effect of NVD on FVD has been observed, we speculate that an acceleration effect of NVD  
 239 (i.e., a decreases of  $T_{VD}$ ) could exist in other cases where NVD occurs in the region where the  
 240 slower-shedding vortex is located.

#### 241 IV. STRUCTURAL LOAD

242 Since all interactions between the natural and forced vortex dislocations are located on the large  
 243 cylinder side of the step cylinder, the present section focuses on how these vortex dislocations  
 244 affect the structural load of the large cylinder.

245 Previous studies<sup>5,14,17</sup> have observed that the magnitude of the drag and lift coefficients de-  
 246 crease when vortex dislocations occur in cylinder wakes. Behara and Mittal<sup>5</sup> found that both the  
 247 mean drag and the amplitude of the lift reach a minimum when the activity of vortex dislocations  
 248 are at the highest. Figures 10(a, b) show the time history of the drag coefficient  $C_D = \frac{2F_D}{\rho U^2 DL}$  (where  
 249  $F_D$  represents the total drag force on the large cylinder) and the total lift coefficient  $C_L = \frac{2F_L}{\rho U^2 DL}$

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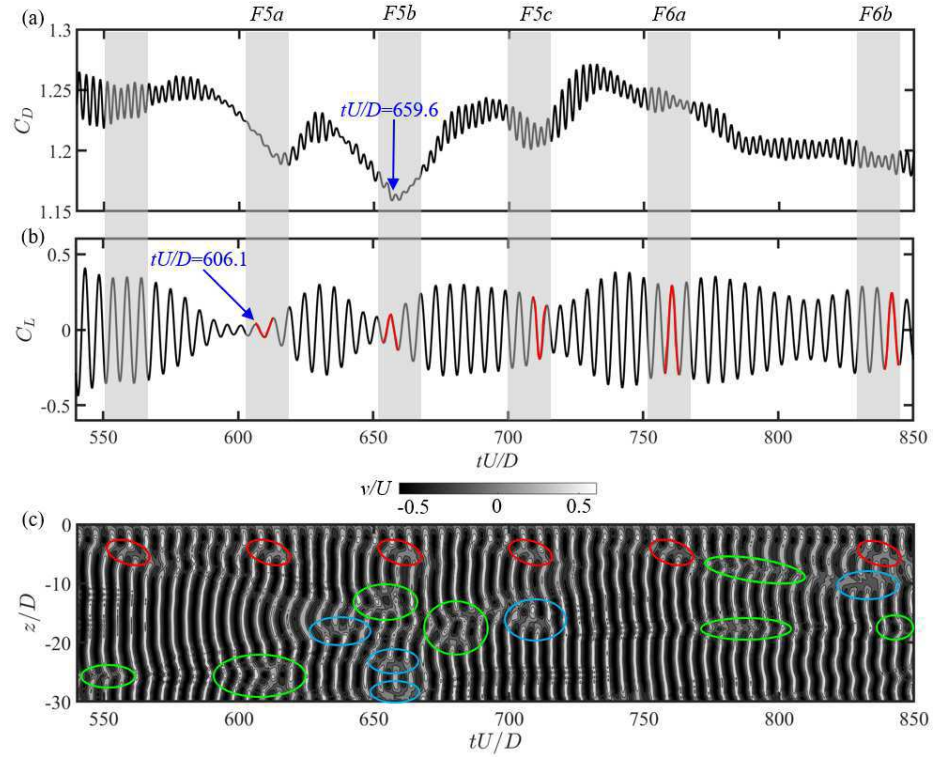


FIG. 10. (a, b) Time history of the drag and lift coefficient on the large cylinder. The time when vortex dislocations occur is highlighted by the transparent rectangles; the one marked by the label  $F5a$  corresponds to the dislocation instant shown in figure 5(a). The meaning of the other labels is analogous. (c) Crossflow velocity component  $v$  as a function of the non-dimensional time, along spanwise sampling line at  $(x/D, y/D)=(0.6, 0)$ . The red, blue and green circles indicate the occurrence of the forced vortex dislocation, the two-side natural vortex dislocation, and the one-side natural vortex dislocation, respectively.

250 (where  $F_L$  represents the local lift force on the large cylinder). The time instant when vortex dis-  
 251 locations occur is highlighted by the shaded gray rectangles; the one marked by the label  $F5a$  is  
 252 where figure 5(a) is located. Figure 10(c) shows contours of the crossflow velocity  $v$ , where the  
 253 occurrence of forced vortex dislocations, two-side natural vortex dislocations, and one-side natu-  
 254 ral vortex dislocations are indicated by red, blue, and green circles, respectively. The number of  
 255 spanwise positions where vortex dislocations (natural and forced) occur is shown in table II. Here

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TABLE II. Number of spanwise positions where vortex dislocations occur

Time	F5a	F5b	F5c	F6a	F6b
Number	3	5	3	1	4

257 one formation of a two-side natural vortex dislocation is treated as two one-side natural vortex  
 258 dislocations forming simultaneously at two spanwise positions<sup>32</sup>. For example, the two-side nat-  
 259 ural vortex dislocation sketched by the green curves in figure 5(e) can be treated as two one-side  
 260 natural vortex dislocation occurring at  $z/D \approx -15$  and  $z/D \approx -12$ . For the other two types of vortex  
 261 dislocations, it is a one-to-one relation between the number of vortex dislocations and the number  
 262 of dislocation positions. Figure 10(a) shows that the amplitude of  $C_D$  and  $C_L$  decreases as vortex  
 263 dislocations occur. The amplitude of  $C_D$  reaches a minimum at  $tU/D = 659.6$  (within F5b) when  
 264 vortex dislocations occur at the largest number of spanwise positions (i.e., when the vortex dislo-  
 265 cation activity becomes the highest) compared to the other instants within  $550 < tU/D < 850$ , as  
 266 indicated in table II and figure 10(c). This is consistent with previous observations<sup>5</sup>. However, the

273 to the coherence of the spanwise vortex<sup>35</sup>, which is affected by the formation of vortex disloca-  
 274 tions but is not proportional to the activity of vortex dislocations. To quantify the coherence of  
 275 the spanwise vortex in the near wake, the spanwise correlation of the crossflow velocity along a  
 276 spanwise sampling line just behind the large cylinder at  $(x/D, y/D) = (0.6, 0)$  is conducted. The  
 277 spanwise correlation coefficient of crossflow velocity  $v$  between two spanwise positions  $z_1$  and  $z_2$   
 278 is calculated using one vortex shedding period of data by:

$$C_{v12}(z_2) = \frac{\overline{(v(z_1) - \overline{v(z_1)})(v(z_2) - \overline{v(z_2)})}}{\sqrt{\overline{(v(z_1) - \overline{v(z_1)})^2}} \sqrt{\overline{(v(z_2) - \overline{v(z_2)})^2}}}, \quad (3)$$

280 where the bar denotes time-averaged taken. The spanwise position where the lift reaches a max-  
 281 imum during the selected period is set up as the reference position ( $z_1$ ) when  $C_{v12}$  is calculated



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282 within a vortex shedding period. For example, the spanwise distribution of  $C_L$  at  $tU/D=606.1$  (as

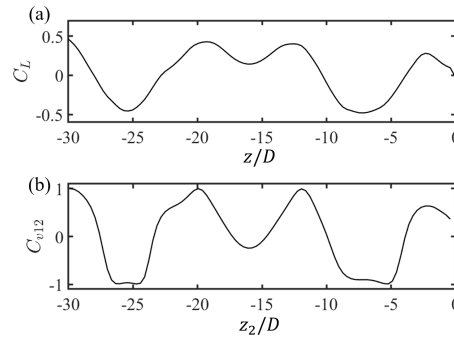


FIG. 11. (a) Spanwise distribution of  $C_L$  along the large cylinder at  $tU/D=601$  as marked in figure 10(b). (b) Distribution of the correlation coefficient  $C_{v12}$  calculated during one vortex shedding period just after  $tU/D=601$  with the reference position  $z_1=-30$ .

283

284

285 marked in figure 10(b)) is shown in figure 11(a), showing a maximum at  $z/D=-30$ . To study why  
 286 the amplitude of  $C_L$  reaches a minimum at  $tU/D=606.1$  in figure 10(b), the correlation coefficient  
 287  $C_{v12}$  is calculated over the corresponding vortex shedding period (colored in red) with the refer-  
 288 ence position  $z_1=-30$ . The result is shown in figure 11(b). When the coefficient is equal to 1, the  
 289 vortex structure at the sampling position is completely in-phase with the vortex at the reference  
 290 position  $z_1=-30$ . This means that the vortex at these two positions shed from the same side of the  
 291 cylinder simultaneously, contributing to the lift force together. When the coefficient is equal to -1,  
 292 the vortex at the sampling position is out-of-phase with the vortex at  $z/D=-30$ , indicating that the  
 293 these two vortices (shed from the different sides of the cylinder) cancel each other out and do not  
 294 contribute to the lift force. Therefore, the coherence of the vortex structure can be evaluated by  
 295 averaging the correlation coefficient  $C_{v12}$  in the spanwise direction. Table III shows  $\overline{C_{v12}}$  for five  
 296 vortex shedding periods within the vortex dislocation duration F5a, F5b, F5c, F6a, and F6b (the  
 297 corresponding time history of  $C_L$  is colored in red in figure 10(b)). First, it is clear that  $\overline{C_{v12}}$  and  
 298 the corresponding amplitude of  $C_L$  shown in figure 10(b) behave qualitatively equal. Secondly, a  
 299 comparison between table II and table III implies that there is no clear relation between the activity  
 300 of vortex dislocations and the vortex coherence.

301 Overall, the formation of vortex dislocations cause the amplitude of both the drag  $C_D$  and lift  
 302  $C_L$  coefficient to become smaller, compared to those when no vortex dislocation occurs. The

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303 amplitude decline of  $C_D$  and  $C_L$  is dominated by the activity of vortex dislocation and the vortex  
 304 coherence in the near wake, respectively. The vortex coherence is affected by but not proportional  
 305 to the activity of vortex dislocation.

TABLE III. Mean correlation coefficient.

Time	F5a	F5b	F5c	F6a	F6b
$\overline{C_{v12}}$	0.06	0.27	0.39	0.52	0.56

## 306 V. CONCLUSION

307 Our present results show good agreement with previous studies, including the three main span-  
 308 wise vortices<sup>19-21,23</sup>, the vortex dislocation mechanism<sup>1,13,17</sup>, the vortex dislocation effects in the  
 309 surrounding flow region<sup>7</sup>, and the different topologies of natural vortex dislocations<sup>32</sup>. More im-  
 310 portantly, the direct numerical simulations of flow around a single step cylinder with  $D/d = 2$  in  
 311 the mode A\* regime ( $Re_D = 200$ ) provide new detailed information about the interaction between  
 312 the forced vortex dislocation process and the natural vortex dislocation process. The new findings  
 313 are as follows:

- 314 - Since the pressure instabilities in the N-cell region are suppressed by the weak suction pres-  
 315 sure behind the small cylinder, natural vortex dislocations can not occur behind the large  
 316 cylinder close to the step (i.e. the N-cell region  $-4 < z/D < 0$ ). Thus, natural vortex dislo-  
 317 cations can only delay, not accelerate, the formation of forced vortex dislocations between  
 318 the N- and L-cell vortices in the present case by locally decreasing the corresponding shed-  
 319 ding frequency of the L-cell vortex.
- 320 - The amplitude of both drag ( $C_D$ ) and lift ( $C_L$ ) coefficients decrease when vortex dislocations  
 321 occur. The activity of vortex dislocations and the vortex coherence in the near wake domi-  
 322 nate the amplitude decline of  $C_D$  and  $C_L$ , respectively. The vortex coherence is affected by,  
 323 but are not proportional to, the activity of vortex dislocation.

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326 computing) under projects nn9352K and nn9191k.

327 **CONFLICT OF INTEREST**

328 The author report no conflict of interest.

329 **DATA AVAILABILITY**

330 The data that support the findings of this study are available from the corresponding author  
331 upon reasonable request.

332 **Appendix A: Topology of vortex**

333 Figure 12(a) shows the isosurface of  $\lambda_2=-0.05$  at  $tU/D=659.6$ , which is directly picked up from  
334 figure 5(b). The vortex dislocations are illustrated by the same types of curves as used in figure  
335 5. At the same instant, figure 12(b) shows transparent isosurfaces of  $\lambda_2=-0.05$  with vortex rotation  
336 axis lines in red calculated based on Liutex method<sup>33</sup>. The zoom-in views of the black rectangles  
337 (where the vortex dislocations occur) in figure 12(b) are shown in figures 12(c, d, e). It is clear  
338 that the topology of vortex structures indicated by the vortex rotation axis lines in figures 12(c-e)  
339 agrees well with that outlined by the hand-sketched curves in figure 12(a).

340 **Appendix B: Crossflow velocity contour**

341 Figure 13 shows the contour of the crossflow velocity  $v$  from  $tU/D=900$  to  $tU/D=1500$ , where  
342 the occurrence of forced vortex dislocations, two-side natural vortex dislocations, and one-side  
343 natural vortex dislocations are marked by the red, blue, and green circles, respectively. The solid  
344 red line marks the spanwise position  $z/D=-7$ . It appears that the formation of forced vortex dislo-  
345 cations (FVD) can only be delayed when natural vortex dislocations occur in the region above the  
346 solid red line (i.e.,  $z/D > -7$ ).

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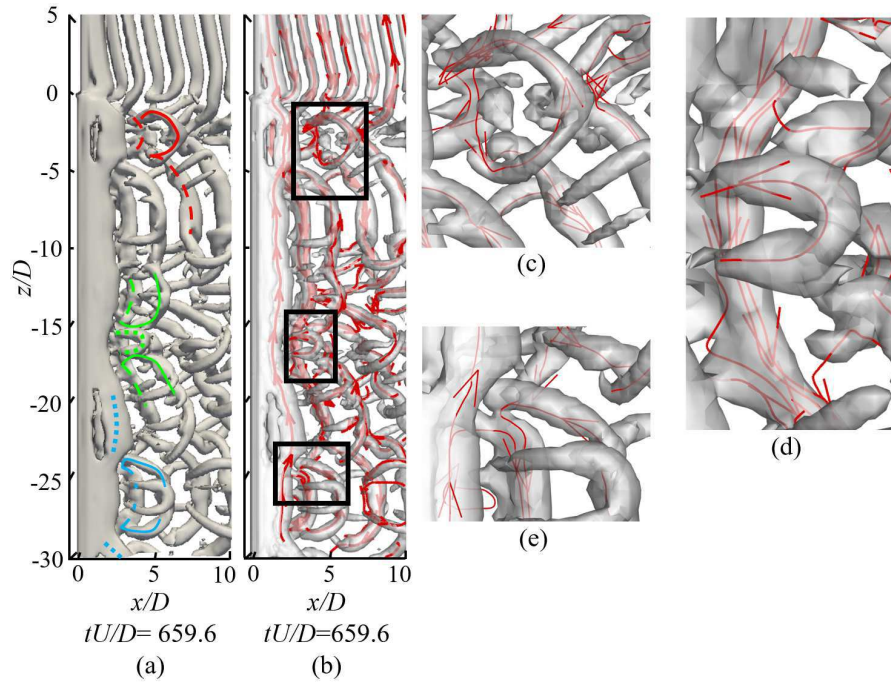


FIG. 12. (a) Isosurfaces of  $\lambda_2 = -0.05$  illustrating the vortex structure in the wake at  $tU/D = 659.6$ . Directly pick up from figure 5(b). The vortex dislocations are illustrated by the same types of curves as used in figure 5. (b) Translucent isosurfaces of  $\lambda_2 = -0.05$  at  $tU/D = 659.6$  with vortex rotation axis lines in red calculated based on Liutex method<sup>33</sup>. (c,d,e) The zoom-in view of the black rectangles in (b).

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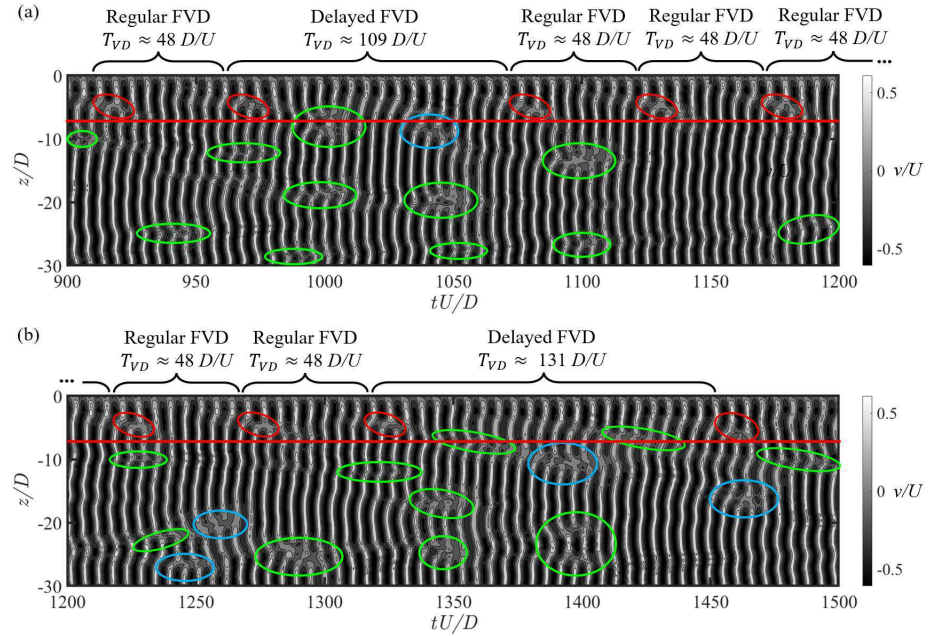


FIG. 13. (a) Crossflow velocity component  $v$  as a function of the non-dimensional time, along spanwise sampling line at  $(x/D, y/D)=(0.6, 0)$  from  $tU/D=900$  to  $tU/D=1200$ . (b) Same as (a) but from  $tU/D=1200$  to  $tU/D=1500$ . The red, blue and green circles indicate the occurrence of the forced vortex dislocation, the two-side natural vortex dislocation, and the one-side natural vortex dislocation, respectively. The spanwise position  $z/D = -7$  is marked by the solid red line.

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