

# Isogeometric collocation for implicit dynamics of three-dimensional beams undergoing finite motions

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## Abstract

We propose a novel approach to the implicit dynamics of shear-deformable geometrically exact beams, based on the isogeometric collocation method combined with the Newmark time integration scheme extended to the rotation group  $SO(3)$ . The proposed formulation is fully consistent with the underlying geometric structure of the configuration manifold. The method is highly efficient, stable, and does not suffer from any singularity problem due to the (material) incremental rotation vector employed to describe the evolution of finite rotations. Consistent linearization of the governing equations, variables initialization and update procedures are the most critical issues which are discussed in detail in the paper. Numerical applications involving very large motions and different boundary conditions demonstrate the capabilities of the method and reveal the critical role that the high-order approximation in space may have in improving the accuracy of the solution.

*Keywords:* Isogeometric collocation, Implicit dynamics, Geometrically nonlinear Timoshenko beams, Finite rotations, Newmark method

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## 1. Introduction

For many complex engineering problems nonlinear beam models, able to accurately reproduce large three-dimensional motions, are often the preferred choice due the low computational cost compared to higher-dimensional models. In pursuing increased efficiency,

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5 robustness and geometric capabilities with respect to existing methods, in this work for the  
6 first time the isogeometric collocation (IGA-C) method is used to solve the implicit dynamic  
7 problem of geometrically exact beams employing the Newmark time integration scheme ex-  
8 tended to the rotation group  $SO(3)$ . The IGA-C method was proposed in [1, 2] with the aim  
9 of exploiting the higher smoothness of NURBS basis functions used in isogeometric analy-  
10 sis (IGA) [3, 4] and the low computational cost of collocation. In IGA higher-order basis  
11 functions with tailorable smoothness, used both for the geometry representation and the  
12 space discretization of the differential problem, have proven to achieve increased accuracy  
13 and robustness on a per degree-of-freedom basis compared with standard Finite Element  
14 Analysis (FEA) [5–8]. Moreover, mesh generation and refinement are significantly simplified  
15 and, once the initial mesh is generated, refinements do not affect the geometric approxi-  
16 mation. High-order basis functions require a larger number of quadrature points causing  
17 a fast growth of the computational cost. Countermeasures against this limitation are still  
18 being investigated, although significant progress has been made in [9–13]. IGA-C represents  
19 an extreme remedy since the need for numerical quadrature is completely removed due to  
20 the discretization of the strong form of the governing equations. IGA-C requires only one  
21 evaluation point per degree of freedom, regardless of the approximation degree, resulting in  
22 a much faster method compared to standard Galerkin-based IGA [14].

23 IGA-C proved excellent performances in a wide range of applications, such as linear prob-  
24 lems [1, 2, 14], phase-field modeling [15], contact problems [16, 17], hyperelasticity [17]. New  
25 relations between Galerkin and collocation methods were found in [18]. Timoshenko beam  
26 formulations were successfully proposed in [19–23]. Bernoulli-Euler beams and Kirchhoff  
27 plates were addressed in [24], and Reissner-Mindlin plate and shell problems in [25] and [26],  
28 respectively. Kirchhoff-Love plate and shell problems were studied in [27]. In [28–30] IGA-C  
29 was extended to geometrically nonlinear three-dimensional shear-deformable beams. Non-  
30 linear planar Kirchhoff rods were formulated in [31]. In linear dynamics, an explicit IGA-C  
31 formulation was introduced in [2] and more recently an explicit higher-order space- and time-  
32 accurate method for elastodynamics was proposed in [32]. IGA-C methods for the nonlinear  
33 dynamics of geometrically exact beams, to our best knowledge, have been investigated so far  
34 only in [33] through an implicit quaternion-based formulation and in [34] through an explicit

35 formulation based on the spatial incremental rotation vector.

36 In the present paper we propose an implicit scheme based on a full material description  
37 of the rotational unknowns. In a three-dimensional Timoshenko beam model, the (finite)  
38 rotation of each beam cross section is described by a time-dependent orthogonal operator  
39 belonging to the non-commutative Lie group  $SO(3)$ . On  $SO(3)$  rotation updates, namely  
40 transformations of  $SO(3)$  onto itself, are consistently performed by the composition of an  
41 incremental rotation with the current rotation. Such an operation, called translation, is  
42 non-additive and non-commutative. The latter attribute leads to a substantial difference  
43 between right translation (spatial description of the motion) and left translation (material  
44 description of the motion) on  $SO(3)$  [35–37]. Apart from the implicitness of the method,  
45 the fundamental difference with respect to our previous formulations [28, 30, 34] lies in that  
46 incremental rotations are now considered in the material setting, therefore rotation updates  
47 are performed via left translations. This change, necessary for an optimal combination  
48 with the  $SO(3)$ -extended version of the classical Newmark scheme originally proposed in  
49 [36], requires a new derivation of the linearized equations and new formulas for the update  
50 procedure. These two central issues are discussed in detail in the present paper.

51 In the standard FEA framework, after the seminal papers by Simo & Vu-Quoc [36] and  
52 Cardona & Geradin [38], several time integration schemes and nonlinear beam formulations  
53 have been proposed and discussed over the years [33, 37, 39–53]. Our choice of combining  
54 the  $SO(3)$  Newmark scheme with a kinematic model based on the material incremental  
55 rotation vector is made to endow the formulation with the following attributes: (i) high  
56 stability; (ii) full geometric consistency, namely the main operations of linearization, variables  
57 initialization and kinematic update are made consistently with the geometric structure of  
58 the configuration manifold; (iii) high efficiency, since rotational unknowns are represented  
59 by a minimum number of variables due to the use of the rotation-vector parameterization.  
60 Moreover, the time stepping algorithm is used in a full material framework. This is an  
61 improvement with respect to [36], where the primary rotational unknowns are expressed in  
62 the spatial form, while the time stepping algorithm is set in the material form. Thus, we  
63 avoid the repetitive and time-consuming pull-back and push-forward operations between the  
64 material and spatial manifolds; (iv) absence of singularities, since incremental rotations are

65 always small.

66 The outline of the paper is as follows: in Section 2 we briefly review the three-dimensional  
 67 shear-deformable beam kinematics with a focus on the construction of the material tangent  
 68 space. In Section 3 the material strong form of the governing equations is presented in terms  
 69 of kinematic variables only. Strain measures and the constitutive law are also briefly recalled.  
 70 In Section 4 the main features of the Newmark time stepping scheme are discussed. Section 5,  
 71 complemented with Appendix A, addresses the discretization of the governing equations and  
 72 their consistent linearization. In Section 6 we describe the solution procedure focusing on the  
 73 variables initialization and update formulas. In Section 7 we apply the proposed formulation  
 74 to solve problems involving very large displacements and rotations with different boundary  
 75 conditions. Finally, in Section 8, we draw the main conclusions of our work.

## 76 **2. The configuration manifold and its (material) tangent space**

77 In this section we briefly recall the geometric structure underlying the kinematics of the  
 78 Timoshenko beam model.

79 The motion of any material particle  $\mathbf{p} \in \mathcal{B}$  of a shear-deformable beam is expressed as  
 80  $\boldsymbol{\varphi}(t, \mathbf{p}) = \mathbf{c}(t, \mathbf{q}) + \mathbf{R}(t, \mathbf{q})(\mathbf{p} - \mathbf{q})$  where  $t$  is the time,  $\mathbf{q}$  is the material position of the  
 81 centroid of the beam cross section containing point  $\mathbf{p}$ .  $\mathcal{S} \subset \mathcal{B}$  is the centroid line, namely a  
 82 one-dimensional space containing the centroids of all cross sections of the beam. On  $\mathcal{S}$  we  
 83 define a coordinate system  $s : \mathcal{S} \rightarrow [0, L] \subset \mathbb{R}$ , where  $L$  is the length of the beam centroid  
 84 line in the initial configuration. The configuration manifold is the set

$$\mathcal{C} = \{(\mathbf{c}, \mathbf{R}) \mid \mathbf{c} : \mathbf{T} \times \mathcal{S} \rightarrow \mathbb{R}^3, \mathbf{R} : \mathbf{T} \times \mathcal{S} \rightarrow \text{SO}(3)\}, \quad (1)$$

85 where  $\mathbf{c}(t, \mathbf{q})$  is the spatial location of the center of mass of the beam cross section at time  
 86  $t$  and point  $\mathbf{q} \in \mathcal{S}$ , and  $\mathbf{R}(t, \mathbf{q})$  is the rigid rotation of the same cross section at the same  
 87 time.  $\mathbf{T} = [0, T] \subset \mathbb{R}$  denotes the time domain.

88 As opposed to our previous formulations [28, 30, 34], where the spatial formulation was  
 89 used, here we need to introduce the material form of the tangent space to the configuration  
 90 manifold at point  $(\mathbf{c}, \mathbf{R}) \in \mathcal{C}$ . The tangent space is denoted by  $T_{(\mathbf{c}, \mathbf{R})}\mathcal{C} = T_{\mathbf{c}}\mathbb{R}^3 \times T_{\mathbf{R}}\text{SO}(3)$ ,  
 91 where  $T_{\mathbf{c}}\mathbb{R}^3$ , the tangent space to  $\mathbb{R}^3$  at  $\mathbf{c}$ , is simply  $\mathbb{R}^3$ , namely the set of vectors  $\boldsymbol{\eta}$

92 applied in  $\mathbf{c}$ ; whereas the (material) tangent space to  $\text{SO}(3)$  at  $\mathbf{R}$  is given by  $T_{\mathbf{R}}\text{SO}(3) =$   
 93  $\left\{ \mathbf{R}\tilde{\boldsymbol{\Theta}} \mid \tilde{\boldsymbol{\Theta}} \in \text{so}(3), \mathbf{R} \in \text{SO}(3) \right\}$ . The construction of the material tangent space is made  
 94 similarly to the spatial case [28] with the difference that the incremented rotation  $\mathbf{R}_\varepsilon$  is  
 95 obtained through left translation instead of right translation [35–37]. Namely, the (material)  
 96 tangent space at  $(\mathbf{c}, \mathbf{R})$  is obtained by  $(d\boldsymbol{\gamma}/d\varepsilon)_{\varepsilon=0}$  where  $\varepsilon \mapsto \boldsymbol{\gamma}(\varepsilon) = (\mathbf{c}_\varepsilon, \mathbf{R}_\varepsilon)$ , with  $\varepsilon \in \mathbb{R}$ ,  
 97 is a curve on  $\mathcal{C}$  defined by  $\mathbf{c}_\varepsilon = \mathbf{c} + \varepsilon\boldsymbol{\eta}$  (standard translation on  $\mathbb{R}^3$ ) and  $\mathbf{R}_\varepsilon = \mathbf{R}\exp(\varepsilon\tilde{\boldsymbol{\Theta}})$   
 98 (left translation on  $\text{SO}(3)$ ), such that  $\boldsymbol{\gamma}(0) = (\mathbf{c}, \mathbf{R})$ . From the kinematic point of view,  
 99  $\boldsymbol{\eta} \in \mathbb{R}^3$  represents an incremental displacement superimposed to the current configuration  
 100 of the centroid line  $\mathbf{c}$ ; whereas  $\tilde{\boldsymbol{\Theta}}$ , such that  $\mathbf{R}\tilde{\boldsymbol{\Theta}} \in T_{\mathbf{R}}\text{SO}(3)$ , represents an incremental  
 101 rotation superimposed to the rotation  $\mathbf{R}$ . Note that in the construction of the curve  $\boldsymbol{\gamma}$   
 102 we used the exponential map  $\exp : \text{so}(3) \rightarrow \text{SO}(3)$  which maps the line  $\varepsilon\tilde{\boldsymbol{\Theta}}$  at  $\text{so}(3)$  onto  
 103 the one parameter subgroup  $\exp(\varepsilon\tilde{\boldsymbol{\Theta}}) \in \text{SO}(3)$  [35, p. 160]. This means that  $\exp(\varepsilon\tilde{\boldsymbol{\Theta}})$  is  
 104 a rotation occurring around the fixed direction  $\boldsymbol{\Theta} = \text{axial}(\tilde{\boldsymbol{\Theta}})^1$ . A fundamental aspect the  
 105 present formulation relies on is that for  $\text{SO}(3)$  the exponential map is expressed by an exact  
 106 (Rodrigues) formula [54–57].

107 For a more detailed discussion on the construction of tangent spaces to  $\text{SO}(3)$  reference  
 108 is made to [36–38, 42, 44].

### 109 3. Balance equations in local form

110 We start this section by recalling the strong form of the balance equations using the  
 111 material description. Since in this paper we develop a primal formulation, the equations are  
 112 expressed in terms of kinematic quantities only. For this reason, we shortly review also the  
 113 strain measures and the constitutive law.

The strong form of the balance equations [58] can be written in the material form as

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<sup>1</sup>With the symbol  $\sim$  we denote elements of  $\text{so}(3)$ , that is the set of  $3 \times 3$  skew-symmetric matrices that, in this context, represent infinitesimal incremental rotations. Furthermore, for any skew-symmetric matrix  $\tilde{\mathbf{a}} \in \text{so}(3)$ ,  $\mathbf{a} = \text{axial}(\tilde{\mathbf{a}})$  indicates the axial vector of  $\tilde{\mathbf{a}}$  such that  $\tilde{\mathbf{a}}\mathbf{h} = \mathbf{a} \times \mathbf{h}$ , for any  $\mathbf{h} \in \mathbb{R}^3$ .

follows

$$\mu \mathbf{R}^\top \mathbf{a} = \widetilde{\mathbf{K}} \mathbb{C}_N \boldsymbol{\Gamma}_N + \mathbb{C}_N \boldsymbol{\Gamma}_{N,s} + \mathbf{R}^\top \bar{\mathbf{n}}, \quad (2)$$

$$\mathbf{J} \mathbf{A} + \widetilde{\mathbf{W}} \mathbf{J} \mathbf{W} = \widetilde{\mathbf{K}} \mathbb{C}_M \mathbf{K}_M + \mathbb{C}_M \mathbf{K}_{M,s} + \mathbf{R}^\top \mathbf{c}_{,s} \times \mathbb{C}_N \boldsymbol{\Gamma}_N + \mathbf{R}^\top \bar{\mathbf{m}}, \quad (3)$$

114 valid for any  $s \in (0, L)$  and  $t \in (0, T]$ . In the above equations  $\mu$  is the mass per unit length  
 115 of the beam;  $\mathbf{a}$  is the spatial acceleration vector of the cross section centroid;  $\widetilde{\mathbf{K}} = \mathbf{R}^\top \mathbf{R}_{,s}$   
 116 is the beam curvature (skew-symmetric tensor) in the material form;  $\boldsymbol{\Gamma}_N$  and  $\mathbf{K}_M$  are the  
 117 strain measure vectors in the material form (better detailed later);  $\bar{\mathbf{n}}$  and  $\bar{\mathbf{m}}$  are the dis-  
 118 tributed external forces and moments per unit length in spatial form;  $\mathbf{J}$  is the material  
 119 (time-independent) inertia tensor;  $\widetilde{\mathbf{W}} = \mathbf{R}^\top \dot{\mathbf{R}}$  is the material skew-symmetric angular veloc-  
 120 ity tensor and  $\mathbf{W} = \text{axial}(\widetilde{\mathbf{W}})$  its axial vector;  $\mathbf{A} = \dot{\mathbf{W}}$  is the material angular acceleration  
 121 vector;  $\mathbb{C}_N = \text{diag}(GA_1, EA, GA_3)$  and  $\mathbb{C}_M = \text{diag}(EJ_1, GJ, EJ_3)$ , where  $GA_1$  and  $GA_3$  are  
 122 the shear stiffnesses along the cross section principal axes,  $EA$  is the axial stiffness,  $GJ$  is  
 123 the torsional stiffness, and  $EJ_1$  and  $EJ_3$  are the principal bending stiffnesses. Partial deriva-  
 124 tives with respect to the coordinate  $s : \mathcal{S} \rightarrow [0, L] \subset \mathbb{R}$  are indicated with  $(\cdot)_{,s}$ , whereas  $(\dot{\cdot})$   
 125 indicates the derivative with respect to time.

Boundary and initial conditions are given as follows

$$\boldsymbol{\eta} = \bar{\boldsymbol{\eta}}_c \quad \text{or} \quad \mathbf{N} = \mathbf{R}^\top \bar{\mathbf{n}}_c \quad \text{with} \quad s = \{0, L\}, t \in [0, T], \quad (4)$$

$$\boldsymbol{\Theta} = \bar{\boldsymbol{\Theta}}_c \quad \text{or} \quad \mathbf{M} = \mathbf{R}^\top \bar{\mathbf{m}}_c \quad \text{with} \quad s = \{0, L\}, t \in [0, T], \quad (5)$$

$$\mathbf{v} = \mathbf{v}_0 \quad \text{with} \quad s \in (0, L) \quad \text{and} \quad t = 0, \quad (6)$$

$$\mathbf{W} = \mathbf{W}_0 \quad \text{with} \quad s \in (0, L) \quad \text{and} \quad t = 0, \quad (7)$$

126 where  $\mathbf{N}$  and  $\mathbf{M}$  are the material internal forces and moments, respectively;  $\bar{\mathbf{n}}_c$  and  $\bar{\mathbf{m}}_c$   
 127 are the external concentrated forces and moments applied to any of the beam ends in the  
 128 current configuration;  $\mathbf{v}$  is the spatial velocity vector of the cross section centroid;  $\bar{\boldsymbol{\eta}}_c$  and  
 129  $\bar{\boldsymbol{\Theta}}_c$  are the prescribed displacement (spatial) and rotation (material) vectors at any of the  
 130 beam ends.

131 We recall that the deformation measures in the material form are given by [58–60]

$$\boldsymbol{\Gamma}_N = \boldsymbol{\Gamma} - \boldsymbol{\Gamma}_0 = \mathbf{R}^\top \mathbf{c}_{,s} - \mathbf{R}_0^\top \mathbf{c}_{0,s} \quad \text{and} \quad \mathbf{K}_M = \text{axial}(\widetilde{\mathbf{K}} - \widetilde{\mathbf{K}}_0) = \mathbf{K} - \mathbf{K}_0, \quad (8)$$

132 where  $\mathbf{\Gamma} = \mathbf{R}^\top \mathbf{c}_{,s} - [0\ 1\ 0]^\top$  and  $\mathbf{\Gamma}_0 = \mathbf{R}_0^\top \mathbf{c}_{0,s} - [0\ 1\ 0]^\top$ .  $\mathbf{\Gamma}_N$  describes the axial and shear  
133 strains, whereas  $\mathbf{K}_M$  describes the bending and torsional strains;  $\widetilde{\mathbf{K}}_0 = \mathbf{R}_0^\top \mathbf{R}_{0,s}$  is the beam  
134 initial curvature (skew-symmetric tensor) in the material form;  $\mathbf{c}_0$  represents the centroid line  
135 in the initial configuration;  $\mathbf{R}_0 \in \text{SO}(3)$  is the rotation operator that expresses the rotation  
136 of the beam cross section in the initial configuration [59, 60].

137 Under the assumption of a Saint Venant-Kirchhoff constitutive model, the material in-  
138 ternal forces and moments are linearly related to the material strain measures as follows  
139 [38, 44, 61]

$$\mathbf{N} = \mathbb{C}_N \mathbf{\Gamma}_N \quad \text{and} \quad \mathbf{M} = \mathbb{C}_M \mathbf{K}_M. \quad (9)$$

#### 140 4. Implicit time stepping algorithm

141 Before addressing the core of the formulation, in this section we review the Newmark time  
142 integration scheme extended to the rotation group  $\text{SO}(3)$  [36] and focus on some preparatory  
143 (but fundamental for the computational formulation) geometric aspects related to the use  
144 of different tangent spaces to  $\text{SO}(3)$ .

##### 145 4.1. The Newmark scheme

The material form of the Newmark algorithm for  $\text{SO}(3)$  [36] is given as follows

$$\mathbf{R}^{n+1} = \mathbf{R}^n \exp(\widetilde{\mathbf{\Theta}}^n), \quad (10)$$

$$\mathbf{\Theta}^n = h \mathbf{W}^n + h^2 \left[ \left( \frac{1}{2} - \beta \right) \mathbf{A}^n + \beta \mathbf{A}^{n+1} \right], \quad (11)$$

$$\mathbf{W}^{n+1} = \mathbf{W}^n + h \left[ (1 - \gamma) \mathbf{A}^n + \gamma \mathbf{A}^{n+1} \right]. \quad (12)$$

The superscript  $n = 0, 1, \dots$  is used to denote any temporal discrete and approximate quan-  
tity at time  $t^n = nh$ , where  $h$  is the time step size.  $\beta \in [0, \frac{1}{2}]$  and  $\gamma \in [0, 1]$  are the standard  
Newmark parameters. It is convenient for the developments in the next sections to express  
angular acceleration and velocity at time  $t^{n+1}$  in terms of quantities at  $t^n$ . By exploiting  
Eqs. (11) and (12) we have

$$\mathbf{A}^{n+1} = \frac{1}{\beta h^2} \mathbf{\Theta}^n - \mathbf{A}_*^n, \quad (13)$$

$$\mathbf{W}^{n+1} = \frac{\gamma}{\beta h} \mathbf{\Theta}^n + \mathbf{W}_*^n, \quad (14)$$

where we have set

$$\mathbf{A}_*^n = \frac{1}{h\beta} \mathbf{W}^n + \left(\frac{1}{2\beta} - 1\right) \mathbf{A}^n, \quad (15)$$

$$\mathbf{W}_*^n = \left(1 - \frac{\gamma}{\beta}\right) \mathbf{W}^n + \left(1 - \frac{\gamma}{2\beta}\right) h \mathbf{A}^n. \quad (16)$$

The algorithm used to integrate the motion of the beam centroid line is the standard Newmark for nonlinear dynamics, which, for the sake of completeness, is reported in the following

$$\mathbf{c}^{n+1} = \mathbf{c}^n + \boldsymbol{\eta}^n, \quad (17)$$

$$\boldsymbol{\eta}^n = h \mathbf{v}^n + h^2 \left[ \left(\frac{1}{2} - \beta\right) \mathbf{a}^n + \beta \mathbf{a}^{n+1} \right], \quad (18)$$

$$\mathbf{v}^{n+1} = \mathbf{v}^n + h \left[ (1 - \gamma) \mathbf{v}^n + \gamma \mathbf{a}^{n+1} \right]. \quad (19)$$

As done for the rotational quantities, using Eqs. (18) and (19), angular acceleration and velocity at time  $t^{n+1}$  can be expressed as follows

$$\mathbf{a}^{n+1} = \frac{1}{\beta h^2} \boldsymbol{\eta}^n - \mathbf{a}_*^n, \quad (20)$$

$$\mathbf{v}^{n+1} = \frac{\gamma}{\beta h} \boldsymbol{\eta}^n + \mathbf{v}_*^n, \quad (21)$$

where we have set

$$\mathbf{a}_*^n = \frac{1}{h\beta} \mathbf{v}^n + \left(\frac{1}{2\beta} - 1\right) \mathbf{a}^n, \quad (22)$$

$$\mathbf{v}_*^n = \left(1 - \frac{\gamma}{\beta}\right) \mathbf{v}^n + \left(1 - \frac{\gamma}{2\beta}\right) h \mathbf{a}^n. \quad (23)$$

#### 146 4.2. Relation between different tangent spaces to SO(3)

147 The numerical formulation we employ crucially relies on the correct identification of the  
 148 tangent space the incremental rotations belong to [36, 37]. Roughly speaking, the tangent  
 149 space is a geometric structure which permits to approximate locally the nonlinear manifold  
 150 (SO(3) in our case) with a vector space ( $\mathbb{R}^3$  in our case) where standard (additive and  
 151 commutative) operations can be performed. The local nature of the tangent space represents  
 152 the main complexity behind the numerical schemes involving finite rotations.



Starting from Eq. (10), an incremented rotation from  $\mathbf{R}^{n+1}$  can be obtained in two different ways, in both cases by using left (material) translation, as follows

$$\mathbf{R}_\varepsilon^{n+1} = \mathbf{R}^{n+1} \exp(\varepsilon \delta \tilde{\boldsymbol{\Theta}}^{n+1}), \quad (24)$$

$$\mathbf{R}_\varepsilon^{n+1} = \mathbf{R}^n \exp(\tilde{\boldsymbol{\Theta}}^n + \varepsilon \delta \tilde{\boldsymbol{\Theta}}^n). \quad (25)$$

153 Note that  $\delta \tilde{\boldsymbol{\Theta}}^{n+1}$  and  $\delta \tilde{\boldsymbol{\Theta}}^n$  are both incremental rotations but they refer to two different  
 154 tangent spaces, namely  $\delta \tilde{\boldsymbol{\Theta}}^{n+1} \in T_{\mathbf{R}^{n+1}} \text{SO}(3)$ , whereas  $\delta \tilde{\boldsymbol{\Theta}}^n \in T_{\mathbf{R}^n} \text{SO}(3)$ . By exploiting  
 155 Eq. (10) and the orthogonality property of the rotation operators, the two above equations  
 156 lead to

$$\exp(\tilde{\boldsymbol{\Theta}}^n) \exp(\varepsilon \delta \tilde{\boldsymbol{\Theta}}^{n+1}) = \exp(\tilde{\boldsymbol{\Theta}}^n + \varepsilon \delta \tilde{\boldsymbol{\Theta}}^n). \quad (26)$$

157 As demonstrated in [38], the above equation provides the relation between the incremental  
 158 rotations belonging to two different tangent spaces. Namely, there exists a linear invertible  
 159 mapping  $\mathbf{T}(\boldsymbol{\Theta}^n) : T_{\mathbf{R}^n} \text{SO}(3) \rightarrow T_{\mathbf{R}^{n+1}} \text{SO}(3)$  such that<sup>2</sup>

$$\delta \boldsymbol{\Theta}^{n+1} = \mathbf{T}(\boldsymbol{\Theta}^n) \delta \boldsymbol{\Theta}^n \quad \text{and} \quad \delta \boldsymbol{\Theta}^n = \mathbf{T}^{-1}(\boldsymbol{\Theta}^n) \delta \boldsymbol{\Theta}^{n+1}. \quad (27)$$

160 Given the fundamental role of  $\mathbf{T}^{-1}(\boldsymbol{\Theta}^n)$ , as it will appear clear in the next section, we  
 161 report its explicit expression leaving the details of the derivation in [38, 62]

$$\mathbf{T}^{-1}(\boldsymbol{\Theta}^n) = \mathbf{e}^n \otimes \mathbf{e}^n + \frac{\|\boldsymbol{\Theta}^n\|/2}{\tan(\|\boldsymbol{\Theta}^n\|/2)} (\mathbf{id} - \mathbf{e}^n \otimes \mathbf{e}^n) + \frac{1}{2} \tilde{\boldsymbol{\Theta}}^n, \quad (28)$$

162 where  $\mathbf{e}^n = \boldsymbol{\Theta}^n / \|\boldsymbol{\Theta}^n\|$ .

163 As regards the translational motion of the cross section centroid, the mapping  $\mathbf{T}$  turns  
 164 out to be the identity map, namely  $\delta \boldsymbol{\eta}^{n+1}$  and  $\delta \boldsymbol{\eta}^n$  coincide.

## 165 5. Discretization of the governing equations and consistent linearization

166 In this section we first introduce the time discretized version of the governing equations.  
 167 Then we proceed with the consistent linearization of the equations and finally we introduce  
 168 the spatial discretization. Importantly, we remark that, unlike in [36] where the problem

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<sup>2</sup>An abuse of notation is made here since the operator  $\mathbf{T}(\boldsymbol{\Theta}^n)$  actually maps axial vectors and not the associated skew-symmetric matrices.

169 is solved in the spatial setting and the time integration is performed in the material frame,  
 170 here we propose a procedure which is fully carried out in the material setting, resulting in  
 171 an increased efficiency since the numerous pull-backs and push-forwards between the two  
 172 settings are avoided.

### 173 5.1. Time-discretized governing equations

The balance equations (2) and (3) together with the Neumann boundary conditions Eqs. (4) and (5) must be satisfied at each time instant. We write the equations at  $t^{n+1}$  as follows

$$-\mu \mathbf{R}^{\top n+1} \mathbf{a}^{n+1} + \widetilde{\mathbf{K}}^{n+1} \mathbb{C}_N \mathbf{\Gamma}_N^{n+1} + \mathbb{C}_N \mathbf{\Gamma}_{N,s}^{n+1} + \mathbf{R}^{\top n+1} \bar{\mathbf{n}}^{n+1} = \mathbf{0}, \quad (29)$$

$$\begin{aligned} & -(\mathbf{J} \mathbf{A}^{n+1} + \widetilde{\mathbf{W}}^{n+1} \mathbf{J} \mathbf{W}^{n+1}) + \widetilde{\mathbf{K}}^{n+1} \mathbb{C}_M \mathbf{K}_M^{n+1} + \\ & \mathbb{C}_M \mathbf{K}_{M,s}^{n+1} + \mathbf{R}^{\top n+1} \mathbf{c}_{,s}^{n+1} \times \mathbb{C}_N \mathbf{\Gamma}_N^{n+1} + \mathbf{R}^{\top n+1} \bar{\mathbf{m}}^{n+1} = \mathbf{0}, \end{aligned} \quad (30)$$

$$\mathbb{C}_N \mathbf{\Gamma}_N^{n+1} - \mathbf{R}^{\top n+1} \bar{\mathbf{n}}_c^{n+1} = \mathbf{0}, \quad (31)$$

$$\mathbb{C}_M \mathbf{K}_M^{n+1} - \mathbf{R}^{\top n+1} \bar{\mathbf{m}}_c^{n+1} = \mathbf{0}. \quad (32)$$

### 174 5.2. Linearization of the time-discretized governing equations

175 Linearizations are performed making use of the directional derivatives reported in Ap-  
 176 pendix A. With the symbol  $(\hat{\cdot})$  we denote any quantity evaluated at the configuration  $t^{n+1}$   
 177 around which the linearization takes place. We start with the linearization of the transla-  
 178 tional equations and then proceed with the rotational equations. For the sake of clarity, we  
 179 proceed systematically term by term.

#### 180 5.2.1. Translational equations

The linearization of the inertia term of Eq. (29) leads to

$$\begin{aligned} L[\mu \mathbf{R}^{\top n+1} \mathbf{a}^{n+1}] &= \mu \hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{a}}^{n+1} + \mu \frac{d}{d\varepsilon} (\mathbf{R}_\varepsilon^{\top n+1} \mathbf{a}_\varepsilon^{n+1})_{\varepsilon=0} = \\ & \mu \hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{a}}^{n+1} + \mu \left( \widetilde{\hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{a}}^{n+1}} \right) \delta \boldsymbol{\Theta}^{n+1} + \frac{\mu}{h^2 \beta} \hat{\mathbf{R}}^{\top n+1} \delta \boldsymbol{\eta}^{n+1}. \end{aligned} \quad (33)$$

181 Note that to obtain the above result we used the directional derivative of the acceleration  
 182 given as follows

$$\frac{d}{d\varepsilon} (\mathbf{a}_\varepsilon^{n+1})_{\varepsilon=0} = \frac{\delta \boldsymbol{\eta}^{n+1}}{\beta h^2}, \quad (34)$$

183 where use of Eq. (20) has been made.

Linearization of the second term of Eq. (29) is made as follows

$$\begin{aligned}
L[\widetilde{\mathbf{K}}^{n+1} \mathbb{C}_N \mathbf{\Gamma}_N^{n+1}] &= \widetilde{\mathbf{K}}^{n+1} \mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1} + \frac{d}{d\varepsilon} \left( \widetilde{\mathbf{K}}_\varepsilon^{n+1} \mathbb{C}_N \mathbf{\Gamma}_{N\varepsilon}^{n+1} \right)_{\varepsilon=0} = \\
&\widetilde{\mathbf{K}}^{n+1} \mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1} + \left[ \widetilde{\mathbf{K}}^{n+1} \mathbb{C}_N \left( \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right) - \left( \mathbb{C}_N \widetilde{\hat{\mathbf{\Gamma}}_N^{n+1}} \right) \widetilde{\mathbf{K}}^{n+1} \right] \delta \boldsymbol{\Theta}^{n+1} - \\
&\left[ \mathbb{C}_N \widetilde{\hat{\mathbf{\Gamma}}_N^{n+1}} \right] \delta \boldsymbol{\Theta}_{,s}^{n+1} + \left[ \widetilde{\mathbf{K}}^{n+1} \mathbb{C}_N \hat{\mathbf{R}}^{\text{T}n+1} \right] \delta \boldsymbol{\eta}_{,s}^{n+1}, \tag{35}
\end{aligned}$$

184 where we have used Eqs. (A.7) and (A.13).

Linearization of the third term of Eq. (29) is made as follows

$$\begin{aligned}
L[\mathbb{C}_N \mathbf{\Gamma}_{N,s}^{n+1}] &= \mathbb{C}_N \hat{\mathbf{\Gamma}}_{N,s}^{n+1} + \frac{d}{d\varepsilon} \left( \mathbb{C}_N \mathbf{\Gamma}_{N\varepsilon,s}^{n+1} \right)_{\varepsilon=0} = \\
\mathbb{C}_N \hat{\mathbf{\Gamma}}_{N,s}^{n+1} + \mathbb{C}_N &\left[ \left( \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \hat{\mathbf{c}}_{,ss}^{n+1}} \right) - \left( \widetilde{\mathbf{K}} \left( \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right) \right) \right] \delta \boldsymbol{\Theta}^{n+1} + \mathbb{C}_N \left[ \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right] \delta \boldsymbol{\Theta}_{,s}^{n+1} - \\
&\mathbb{C}_N \left[ \widetilde{\mathbf{K}}^{n+1} \hat{\mathbf{R}}^{\text{T}n+1} \right] \delta \boldsymbol{\eta}_{,s}^{n+1} + \mathbb{C}_N \left[ \hat{\mathbf{R}}^{\text{T}n+1} \right] \delta \boldsymbol{\eta}_{,ss}^{n+1}, \tag{36}
\end{aligned}$$

185 where we have used Eq. (A.14).

Linearization of the fourth term of Eq. (29) is made as follows

$$L[\mathbf{R}_\varepsilon^{\text{T}n+1} \bar{\mathbf{n}}^{n+1}] = \hat{\mathbf{R}}^{\text{T}n+1} \bar{\mathbf{n}}^{n+1} + \left[ \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \bar{\mathbf{n}}^{n+1}} \right] \delta \boldsymbol{\Theta}^{n+1}, \tag{37}$$

186 where Eq. (A.4) has been used.

Linearization of the first term of the boundary condition (31) is made as follows

$$L[\mathbb{C}_N \mathbf{\Gamma}_N^{n+1}] = \mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1} + \frac{d}{d\varepsilon} \left( \mathbb{C}_N \hat{\mathbf{\Gamma}}_{N\varepsilon}^{n+1} \right)_{\varepsilon=0} = \tag{38}$$

$$\mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1} + \mathbb{C}_N \left( \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right) \delta \boldsymbol{\Theta}^{n+1} + \mathbb{C}_N \hat{\mathbf{R}}^{\text{T}n+1} \delta \boldsymbol{\eta}_{,s}^{n+1}, \tag{39}$$

187 where Eq. (A.13) has been used.

Linearization of the last term of boundary condition (31) is made as follows

$$L[\mathbf{R}^{\text{T}n+1} \bar{\mathbf{n}}_c^{n+1}] = \hat{\mathbf{R}}^{\text{T}n+1} \bar{\mathbf{n}}_c^{n+1} + \frac{d}{d\varepsilon} \left( \mathbf{R}_\varepsilon^{\text{T}n+1} \bar{\mathbf{n}}_c^{n+1} \right)_{\varepsilon=0} = \tag{40}$$

$$\hat{\mathbf{R}}^{\text{T}n+1} \bar{\mathbf{n}}_c^{n+1} + \left( \widetilde{\hat{\mathbf{R}}^{\text{T}n+1} \bar{\mathbf{n}}_c^{n+1}} \right) \delta \boldsymbol{\Theta}^{n+1}, \tag{41}$$

188 where Eq. (A.4) has been used.

Linearization of the inertia term of Eq. (30) is given as follows

$$L[\mathbf{J}\mathbf{A}^{n+1} + \widetilde{\mathbf{W}}^{n+1} \mathbf{J}\mathbf{W}^{n+1}] = \mathbf{J}\hat{\mathbf{A}}^{n+1} + \widehat{\mathbf{W}}^{n+1} \mathbf{J}\hat{\mathbf{W}}^{n+1} + \frac{d}{d\varepsilon} \left( \mathbf{J}\mathbf{A}_\varepsilon^{n+1} + \widetilde{\mathbf{W}}_\varepsilon^{n+1} \mathbf{J}\mathbf{W}_\varepsilon^{n+1} \right)_{\varepsilon=0} = \mathbf{J}\hat{\mathbf{A}}^{n+1} + \widehat{\mathbf{W}}^{n+1} \mathbf{J}\hat{\mathbf{W}}^{n+1} + \left[ \frac{1}{\beta h^2} \mathbf{J} - \frac{\gamma}{\beta h} \left( \widetilde{\mathbf{J}\hat{\mathbf{W}}^{n+1}} - \widehat{\mathbf{W}}^{n+1} \mathbf{J} \right) \right] \mathbf{T}^{-1}(\hat{\boldsymbol{\Theta}}^n) \delta \boldsymbol{\Theta}^{n+1}. \quad (42)$$

Note that to obtain the above results we used the directional derivative of the angular velocity and accelerations. We first note that

$$\mathbf{A}_\varepsilon^{n+1} = \frac{\boldsymbol{\Theta}^n + \varepsilon \delta \boldsymbol{\Theta}^n}{\beta h^2} - \mathbf{A}_*^n, \quad (43)$$

$$\mathbf{W}_\varepsilon^{n+1} = \frac{\gamma (\boldsymbol{\Theta}^n + \varepsilon \delta \boldsymbol{\Theta}^n)}{\beta h} + \mathbf{W}_*^n, \quad (44)$$

from which it follows that

$$L[\mathbf{W}_\varepsilon^{n+1}] = \hat{\mathbf{W}}^{n+1} + \frac{\gamma}{\beta h} \delta \boldsymbol{\Theta}^n = \hat{\mathbf{W}}^{n+1} + \frac{\gamma}{\beta h} \mathbf{T}^{-1}(\hat{\boldsymbol{\Theta}}^n) \delta \boldsymbol{\Theta}^{n+1}, \quad (45)$$

$$L[\mathbf{A}_\varepsilon^{n+1}] = \hat{\mathbf{A}}^{n+1} + \frac{1}{\beta h^2} \delta \boldsymbol{\Theta}^n = \hat{\mathbf{A}}^{n+1} + \frac{1}{\beta h^2} \mathbf{T}^{-1}(\hat{\boldsymbol{\Theta}}^n) \delta \boldsymbol{\Theta}^{n+1}, \quad (46)$$

190 where we have used Eq. (27).

Linearization of the second term of Eq. (30) is made as follows

$$L[\widetilde{\mathbf{K}}^{n+1} \mathbb{C}_M \mathbf{K}_M^{n+1}] = \widehat{\mathbf{K}}^{n+1} \mathbb{C}_M \hat{\mathbf{K}}_M^{n+1} + \frac{d}{d\varepsilon} \left( \widetilde{\mathbf{K}}_\varepsilon^{n+1} \mathbb{C}_M \mathbf{K}_{M\varepsilon}^{n+1} \right)_{\varepsilon=0} = \widehat{\mathbf{K}}^{n+1} \mathbb{C}_M \hat{\mathbf{K}}_M^{n+1} + \left[ \widehat{\mathbf{K}}^{n+1} \mathbb{C}_M \widehat{\mathbf{K}}^{n+1} - \left( \widetilde{\mathbb{C}_M \hat{\mathbf{K}}_M} \right) \widehat{\mathbf{K}} \right] \delta \boldsymbol{\Theta}^{n+1} + \left[ \widehat{\mathbf{K}}^{n+1} \mathbb{C}_M - \left( \widetilde{\mathbb{C}_M \hat{\mathbf{K}}_M} \right) \right] \delta \boldsymbol{\Theta}_{,s}^{n+1}, \quad (47)$$

191 where we used Eqs. (A.7) and (A.8).

Linearization of the third term of Eq. (3) is given by

$$L[\mathbb{C}_M \mathbf{K}_{M,s}^{n+1}] = \mathbb{C}_M \hat{\mathbf{K}}_{M,s}^{n+1} + \frac{d}{d\varepsilon} \left( \mathbb{C}_M \mathbf{K}_{M\varepsilon,s}^{n+1} \right)_{\varepsilon=0} = \mathbb{C}_M \hat{\mathbf{K}}_{M,s}^{n+1} + \mathbb{C}_M \widehat{\mathbf{K}}_{,s}^{n+1} \delta \boldsymbol{\Theta}^{n+1} + \mathbb{C}_M \widehat{\mathbf{K}}^{n+1} \delta \boldsymbol{\Theta}_{,s}^{n+1} + \mathbb{C}_M \delta \boldsymbol{\Theta}_{,ss}^{n+1}, \quad (48)$$

192 where we used Eq. (A.12).

Linearization of the fourth term of Eq. (30) is made as follows

$$\begin{aligned}
L[\mathbf{R}^{\top n+1} \mathbf{c}_{,s}^{n+1} \times \mathbb{C}_N \mathbf{\Gamma}_N^{n+1}] &= \hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{c}}_{,s}^{n+1} \times \mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1} + \frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon}^{\top n+1} \mathbf{c}_{\varepsilon,s}^{n+1} \times \mathbb{C}_N \mathbf{\Gamma}_{N\varepsilon}^{n+1})_{\varepsilon=0} = \\
&\hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{c}}_{,s}^{n+1} \times \mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1} + \left[ \left( \widetilde{\hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right) \mathbb{C}_N - \left( \widetilde{\mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1}} \right) \right] \left( \widetilde{\hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right) \delta \boldsymbol{\Theta}^{n+1} + \\
&\left[ \left( \widetilde{\hat{\mathbf{R}}^{\top n+1} \hat{\mathbf{c}}_{,s}^{n+1}} \right) \mathbb{C}_N - \left( \widetilde{\mathbb{C}_N \hat{\mathbf{\Gamma}}_N^{n+1}} \right) \right] \hat{\mathbf{R}}^{\top n+1} \delta \boldsymbol{\eta}_{,s}^{n+1}, \tag{49}
\end{aligned}$$

193 where use has been made of Eqs. (A.4), (A.1) and (A.13).

Linearization of the last term of Eq. (30) is made as follows

$$\begin{aligned}
L[\mathbf{R}^{\top n+1} \bar{\mathbf{m}}^{n+1}] &= \hat{\mathbf{R}}^{\top n+1} \bar{\mathbf{m}}^{n+1} + \frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon}^{\top n+1} \bar{\mathbf{m}}^{n+1})_{\varepsilon=0} = \\
&\hat{\mathbf{R}}^{\top n+1} \bar{\mathbf{m}}^{n+1} + \left( \widetilde{\hat{\mathbf{R}}^{\top n+1} \bar{\mathbf{m}}^{n+1}} \right) \delta \boldsymbol{\Theta}^{n+1}, \tag{50}
\end{aligned}$$

194 where use has been made of Eq. (A.4).

Linearization of the first term of boundary condition (32) is made as follows

$$L[\mathbb{C}_M \mathbf{K}_M^{n+1}] = \mathbb{C}_M \hat{\mathbf{K}}_M^{n+1} + \frac{d}{d\varepsilon} (\mathbb{C}_M \mathbf{K}_{M\varepsilon}^{n+1})_{\varepsilon=0} = \tag{51}$$

$$\mathbb{C}_M \hat{\mathbf{K}}_M^{n+1} + \mathbb{C}_M \widetilde{\hat{\mathbf{K}}_M^{n+1}} \delta \boldsymbol{\Theta}^{n+1} + \mathbb{C}_M \delta \boldsymbol{\Theta}_{,s}^{n+1}, \tag{52}$$

195 where we used Eq. (A.8).

Linearization of the last term of boundary condition (32) is made as follows

$$L[\mathbf{R}^{\top n+1} \bar{\mathbf{m}}_c^{n+1}] = \hat{\mathbf{R}}^{\top n+1} \bar{\mathbf{m}}_c^{n+1} + \frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon}^{\top n+1} \bar{\mathbf{m}}_c^{n+1})_{\varepsilon=0} = \tag{53}$$

$$\hat{\mathbf{R}}^{\top n+1} \bar{\mathbf{m}}_c^{n+1} + \left( \widetilde{\hat{\mathbf{R}}^{\top n+1} \bar{\mathbf{m}}_c^{n+1}} \right) \delta \boldsymbol{\Theta}^{n+1}, \tag{54}$$

196 where we used Eq. (A.4).

### 197 5.3. Space discretization of the linearized equations

The linearized governing equations discussed above are written at time  $t^{n+1}$  but in space are still valid for any point  $\mathbf{q} \in \mathcal{S}$ . The linearized problem is turned into an algebraic system of equations first by discretizing the primary variables  $\delta \boldsymbol{\Theta}^{n+1}$  and  $\delta \boldsymbol{\eta}^{n+1}$  and then

by collocating the time and space discretized equations in a set of points named collocation points. Space discretization is made using NURBS basis functions as follows

$$\mathbf{c}^{n+1}(u) = \sum_{j=0}^n R_{j,p}(u) \check{\mathbf{p}}_j^{n+1} \quad \text{with } u \in \mathcal{I}_u, \quad (55)$$

$$\delta \boldsymbol{\Theta}^{n+1}(u) = \sum_{j=0}^n R_{j,p}(u) \delta \check{\boldsymbol{\Theta}}_j^{n+1} \quad \text{with } u \in \mathcal{I}_u, \quad (56)$$

$$\delta \boldsymbol{\eta}^{n+1}(u) = \sum_{j=0}^n R_{j,p}(u) \delta \check{\boldsymbol{\eta}}_j^{n+1} \quad \text{with } u \in \mathcal{I}_u, \quad (57)$$

198 where  $\mathcal{I}_u = [0, 1]$  is the normalized one-dimensional domain of the  $j$ th NURBS basis function  
 199  $R_{j,p}$ .  $\check{\mathbf{p}}_j^{n+1}$  is the  $j$ th control point defining the centroid line;  $\delta \check{\boldsymbol{\Theta}}_j^{n+1}$  and  $\delta \check{\boldsymbol{\eta}}_j^{n+1}$  are the  $j$ th  
 200 incremental control variables of the kinematic fields, which for  $j = 0, \dots, n$ , form the set of  
 201  $2 \times 3 \times (n + 1)$  unknowns of the algebraic system.

202 Recent studies proposed alternative choices for collocation points that can achieve im-  
 203 proved convergence rates [18, 63–65]; however, in the present study we collocate at the  
 204 standard Greville points [1].

## 205 6. Step by step description of the time integration scheme

206 At the time instant  $t^{n+1}$  the solution of the linearized problem must be searched for it-  
 207 eratively until convergence is achieved within a classical Newton-Raphson scheme. In this  
 208 section we present the two fundamental steps that need to be accomplished: the initializa-  
 209 tion of the variables and the update procedure. Both operations must be performed in a  
 210 geometrically consistent way.

211 Note that in the following we use the subscript  $i$  to denote the  $i$ th collocation point (of  
 212 parametric coordinate  $u_i^c$ ), whereas superscripts  $n, k$  are used to denote the time step and  
 213 the iteration counter, respectively.

### 214 6.1. Initialization of variables at $t^{n+1}$

215 A correct initialization of the system matrix, namely the initialization of all  $(\hat{\cdot})$  quantities  
 216 appearing in the linearized equations, has a crucial role in the reliability of the method.  
 217 Assume that configuration  $(\mathbf{c}_i^n, \mathbf{R}_i^n)$ , velocities  $\mathbf{W}_i^n, \mathbf{v}_i^n$ , accelerations  $\mathbf{A}_i^n, \mathbf{a}_i^n$ , and the strain

218 measures  $\mathbf{\Gamma}_{Ni}^n, \mathbf{K}_{Mi}^n$  (and their derivatives) are known at  $t^n$  in each collocation point  $u_i^c$  with  
 219  $i = 0, \dots, n$ .

First  $\mathbf{\Theta}_i^n$  and  $\boldsymbol{\eta}_i^n$  are initialized by using the predictors in Eqs. (11) and (18) as follows

$$\mathbf{\Theta}_i^{n,0} = h\mathbf{W}_i^n + h^2 \left( \frac{1}{2} - \beta \right) \mathbf{A}_i^n, \quad (58)$$

$$\boldsymbol{\eta}_i^{n,0} = h\mathbf{v}_i^n + h^2 \left( \frac{1}{2} - \beta \right) \mathbf{a}_i^n, \quad (59)$$

220 With Eqs. (58) and (59) at hand, all kinematic quantities can be initialized at time  $t^{n+1}$   
 221 as follows:

- The configuration is initialized as follows

$$\check{\mathbf{p}}_j^{n+1,0} = \check{\mathbf{p}}_j^n + \check{\boldsymbol{\eta}}_j^{n,0}, \quad (60)$$

$$\mathbf{R}_i^{n+1,0} = \mathbf{R}_i^n \exp(\tilde{\mathbf{\Theta}}_i^{n,0}), \quad (61)$$

222 where incremental displacements  $\check{\boldsymbol{\eta}}_j^{n,0}$  are obtained from the corresponding collocated  
 223 quantities  $\boldsymbol{\eta}_i^{n,0}$  given in Eq. (59).

- Accelerations and velocities are initialized by making use of Eqs. (13), (14), (20) and (21) as follows

$$\mathbf{A}_i^{n+1,0} = \frac{1}{\beta h^2} \mathbf{\Theta}_i^{n,0} - \mathbf{A}_{*i}^n, \quad (62)$$

$$\mathbf{W}_i^{n+1,0} = \frac{\gamma}{\beta h} \mathbf{\Theta}_i^{n,0} + \mathbf{W}_{*i}^n, \quad (63)$$

$$\mathbf{a}_i^{n+1,0} = \frac{1}{\beta h^2} \boldsymbol{\eta}_i^{n,0} - \mathbf{a}_{*i}^n, \quad (64)$$

$$\mathbf{v}_i^{n+1,0} = \frac{\gamma}{\beta h} \boldsymbol{\eta}_i^{n,0} + \mathbf{v}_{*i}^n. \quad (65)$$

- Curvature tensors. By recalling the definition of the material curvature tensor, which in the time discretized form reads  $\widetilde{\mathbf{K}}^{n+1,0} = \mathbf{R}^{\Gamma n+1,0} \mathbf{R}_{,s}^{n+1,0}$ , and knowing that  $\mathbf{R}^{n+1,0} = \mathbf{R}^n \exp(\tilde{\mathbf{\Theta}}^{n,0})$ , we obtain at the  $i$ th collocation point

$$\widetilde{\mathbf{K}}_i^{n+1,0} = \exp(-\tilde{\mathbf{\Theta}}_i^{n,0}) \widetilde{\mathbf{K}}_i^n \exp(\tilde{\mathbf{\Theta}}_i^{n,0}) + d \exp_{\tilde{\mathbf{\Theta}}_i^{n,0}}(\tilde{\mathbf{\Theta}}_{i,s}^{n,0}). \quad (66)$$

224 In the above formula we made use of the rule  $(\exp(\tilde{\mathbf{\Theta}}(s)),_s = \exp(\tilde{\mathbf{\Theta}})(d \exp_{\tilde{\mathbf{\Theta}}}) \tilde{\mathbf{\Theta}}_{,s}$ ,  
 225 where the operator  $d \exp_{\tilde{\mathbf{\Theta}}}$  is expressed by a series involving nested Lie bracket (see  
 226 [28, Appendix B] and [66, p. 15] for further details).

In a similar way the initialization formula for the derivative of the material curvature tensor is obtained as follows

$$\begin{aligned}
\widetilde{\mathbf{K}}_{i,s}^{n+1,0} &= \exp(-\widetilde{\boldsymbol{\Theta}}_i^{n,0}) d \exp_{\widetilde{\boldsymbol{\Theta}}_i^{n,0}}(-\widetilde{\boldsymbol{\Theta}}_{i,s}^{n,0}) \widetilde{\mathbf{K}}_i^n \exp(\widetilde{\boldsymbol{\Theta}}_i^{n,0}) + \\
&\quad \exp(-\widetilde{\boldsymbol{\Theta}}_i^{n,0}) \widetilde{\mathbf{K}}_{i,s}^n \exp(\widetilde{\boldsymbol{\Theta}}_i^{n,0}) + \\
&\quad \exp(-\widetilde{\boldsymbol{\Theta}}_i^{n,0}) \widetilde{\mathbf{K}}_i^n \exp(\widetilde{\boldsymbol{\Theta}}_i^{n,0}) d \exp_{\widetilde{\boldsymbol{\Theta}}_i^{n,0}}(\widetilde{\boldsymbol{\Theta}}_{i,s}^{n,0}) + \\
&\quad \left( d \exp_{\widetilde{\boldsymbol{\Theta}}_i^{n,0}}(\widetilde{\boldsymbol{\Theta}}_{i,s}^{n,0}) \right)_{,s} .
\end{aligned} \tag{67}$$

227 The series expressing the operator  $d \exp_{\widetilde{\boldsymbol{\Theta}}}$  permits also to compute the second derivative  
228 of the exponential map  $(d \exp_{\widetilde{\boldsymbol{\Theta}}} \widetilde{\boldsymbol{\Theta}}_{,s})_{,s}$  appearing in Eq. (67) keeping terms up to the  
229 desired order. The derivatives of the incremental rotation appearing in the above  
230 formulas are computed by using the discretization  $\boldsymbol{\Theta}^{n,0} = \sum_{j=0}^n R_{j,p}(u) \check{\boldsymbol{\Theta}}_j^{n,0}$ , where the  
231 incremental rotations  $\check{\boldsymbol{\Theta}}_j^{n,0}$  are obtained from the corresponding collocated quantities  
232  $\boldsymbol{\Theta}_i^{n,0}$  given in Eq. (58)<sup>3</sup>.

- Strain measures defined in Eq. (8) and their derivatives are initialized using the above quantities as follows

$$\boldsymbol{\Gamma}_{Ni}^{n+1,0} = \boldsymbol{\Gamma}_i^{n+1,0} - \boldsymbol{\Gamma}_{0i} \tag{68}$$

$$\boldsymbol{\Gamma}_{Ni,s}^{n+1,0} = \boldsymbol{\Gamma}_{i,s}^{n+1,0} - \boldsymbol{\Gamma}_{0i,s} \tag{69}$$

$$\widetilde{\mathbf{K}}_{Mi}^{n+1,0} = \widetilde{\mathbf{K}}_i^{n+1,0} - \widetilde{\mathbf{K}}_{0i} \tag{70}$$

$$\widetilde{\mathbf{K}}_{Mi,s}^{n+1,0} = \widetilde{\mathbf{K}}_{i,s}^{n+1,0} - \widetilde{\mathbf{K}}_{0i,s} \tag{71}$$

233 where  $\boldsymbol{\Gamma}_i^{n+1,0} = \mathbf{R}_i^{\top n+1,0} \mathbf{c}_{i,s}^{n+1,0} - [0 \ 1 \ 0]^{\top}$  and  $\boldsymbol{\Gamma}_{i,s}^{n+1,0} = -\widetilde{\mathbf{K}}_i^{n+1,0} \mathbf{R}_i^{\top n+1,0} \mathbf{c}_{i,s}^{n+1,0} + \mathbf{R}_i^{\top n+1,0} \mathbf{c}_{i,ss}^{n+1,0}$ .

234 Derivatives of the centroid position are calculated by using the initialized control points  
235 given in Eq. (59).

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<sup>3</sup>First and second-order derivatives with respect to the physical coordinate  $s \in [0, L]$  need to be calculated taking into account the change of parameterization required since NURBS basis functions are defined on the normalized domain  $\mathcal{I}_u = [0, 1]$ . Namely, for any quantity  $\mathbf{g} : \mathcal{S} \rightarrow \mathbb{R}^3$ , we have that  $\mathbf{g}_{,s} = \mathbf{g}_{,u} / j_0$  and  $\mathbf{g}_{,ss} = \mathbf{g}_{,uu} / j_0^2 - \mathbf{g}_{,u} (\mathbf{c}_{0,u} \cdot \mathbf{c}_{0,uu}) / j_0^4$ , where  $j_0 = \|\mathbf{c}_{0,u}\|$  is the jacobian and  $\cdot$  indicates the scalar product.



236 *6.2. Update procedure*

237 Once the initialization procedure described in the above section has been accomplished,  
 238 the iteration procedure starts. Assume that at the  $k$ th iteration (with  $k = 0, 1, 2 \dots$ ) the  
 239 configuration  $(\mathbf{c}_i^{n+1,k}, \mathbf{R}_i^{\top n+1,k})$  and all other relevant kinematic variables are known. The  
 240 solution of the system matrix provides control incremental rotation and displacement vectors  
 241  $\delta\check{\Theta}_j^{n+1,k}, \delta\check{\eta}_j^{n+1,k}$ . With a procedure similar to the one used in [28], but with the fundamental  
 242 difference that here we employ the material incremental rotation vector, the geometrically  
 243 consistent update procedure is performed as follows.

244 Control points are updated by exploiting the standard translation in  $\mathbb{R}^3$

$$\check{\mathbf{p}}_j^{n+1,k+1} = \check{\mathbf{p}}_j^{n+1,k} + \delta\check{\eta}_j^{n+1,k}, \quad (72)$$

245 from which we update the configuration of the centroid line

$$\mathbf{c}_i^{n+1,k+1} = \sum_{j=0}^n R_{j,p}(u_i^c) \check{\mathbf{p}}_j^{n+1,k+1}. \quad (73)$$

246 For the rotation variables, we compute the incremental material rotation vector

$$\delta\Theta_i^{n+1,k} = \sum_{j=0}^n R_{j,p}(u_i^c) \delta\check{\Theta}_j^{n+1,k}, \quad (74)$$

247 then the rotation operator is consistently updated as follows

$$\mathbf{R}_i^{n+1,k+1} = \mathbf{R}_i^{n+1,k} \exp(\delta\tilde{\Theta}_i^{n+1,k}). \quad (75)$$

Update of the curvature tensor and its derivative is made through formulas similar to those used for their initialization (see Eqs. (66) and (67)). Namely we have

$$\tilde{\mathbf{K}}_i^{n+1,k+1} = \exp(-\delta\tilde{\Theta}_i^{n+1,k}) \tilde{\mathbf{K}}_i^{n+1,k} \exp(\delta\tilde{\Theta}_i^{n+1,k}) + d \exp_{\tilde{\Theta}_i^{n+1,k}}(\delta\tilde{\Theta}_{i,s}^{n+1,k}), \quad (76)$$

and

$$\begin{aligned} \tilde{\mathbf{K}}_{i,s}^{n+1,k+1} &= \exp(-\delta\tilde{\Theta}_i^{n+1,k}) d \exp_{-\delta\tilde{\Theta}_i^{n+1,k}}(-\delta\tilde{\Theta}_{i,s}^{n+1,k}) \tilde{\mathbf{K}}_i^{n+1,k} \exp(\delta\tilde{\Theta}_i^{n+1,k}) + \\ &\quad \exp(-\delta\tilde{\Theta}_i^{n+1,k}) \tilde{\mathbf{K}}_{i,s}^{n+1,k} \exp(\delta\tilde{\Theta}_i^{n+1,k}) + \\ &\quad \exp(-\delta\tilde{\Theta}_i^{n+1,k}) \tilde{\mathbf{K}}_i^{n+1,k} \exp(\delta\tilde{\Theta}_i^{n+1,k}) d \exp_{\delta\tilde{\Theta}_i^{n+1,k}}(\delta\tilde{\Theta}_{i,s}^{n+1,k}) + \\ &\quad \left( d \exp_{\delta\tilde{\Theta}_i^{n+1,k}}(\delta\tilde{\Theta}_{i,s}^{n+1,k}) \right)_{,s}. \end{aligned} \quad (77)$$

248 The above updating formulas are obtained by following a very similar procedure used for  
 249 Eqs. (66) and (67) with the difference that in this case  $\widetilde{\mathbf{K}}^{n+1,k+1} = \mathbf{R}^{\top n+1,k+1} \mathbf{R}_s^{n+1,k+1}$  where  
 250  $\mathbf{R}^{n+1,k+1} = \mathbf{R}^{n+1,k} \exp(\delta \widetilde{\boldsymbol{\Theta}}^{n+1,k})$ .

251 The updates of the strain measure vectors  $\mathbf{K}_{Mi}^{n+1,k+1}$  and  $\boldsymbol{\Gamma}_{Ni}^{n+1,k+1}$  and their derivatives  
 252 are made straightforwardly by using the updated kinematic variables presented above.

Finally we need to update accelerations and velocities. Making use of Eqs. (13) and (14) expressed once in terms of  $\boldsymbol{\Theta}^{n,k}$  and once in terms of  $\boldsymbol{\Theta}^{n,k+1}$ , see [36] for the details, we obtain the following update formulas

$$\mathbf{A}_i^{n+1,k+1} = \mathbf{A}_i^{n+1,k} + \frac{1}{\beta h^2} \left( \boldsymbol{\Theta}_i^{n,k+1} - \boldsymbol{\Theta}_i^{n,k} \right), \quad (78)$$

$$\mathbf{W}_i^{n+1,k+1} = \mathbf{W}_i^{n+1,k} + \frac{\gamma}{\beta h} \left( \boldsymbol{\Theta}_i^{n,k+1} - \boldsymbol{\Theta}_i^{n,k} \right), \quad (79)$$

where  $\boldsymbol{\Theta}_i^{n,k}$  is known from the previous iteration, whereas we still have to compute  $\boldsymbol{\Theta}_i^{n,k+1}$ . To this end we recall that  $\boldsymbol{\Theta}_i^{n,k+1}$  is such that

$$\mathbf{R}_i^{n+1,k+1} = \mathbf{R}_i^n \exp(\widetilde{\boldsymbol{\Theta}}^{n,k+1}), \quad (80)$$

from which we extract the incremental rotation vector by making use of the inverse of the exponential operator [36, 37, 67] as follows

$$\widetilde{\boldsymbol{\Theta}}_i^{n,k+1} = \exp^{-1} \left( \mathbf{R}_i^{\top n} \mathbf{R}_i^{n+1,k+1} \right), \quad (81)$$

253 where  $\mathbf{R}_i^{\top n+1,k}$  is given by  $\mathbf{R}_i^{n+1,k} \exp(\delta \widetilde{\boldsymbol{\Theta}}^{n+1,k})$ .

In a similar way, but without the complexity related to SO(3), update formulas for linear acceleration and velocity are given by

$$\mathbf{a}_i^{n+1,k+1} = \mathbf{a}_i^{n+1,k} + \frac{1}{\beta h^2} \delta \boldsymbol{\eta}_i^{n+1,k}, \quad (82)$$

$$\mathbf{v}_i^{n+1,k+1} = \mathbf{v}_i^{n+1,k} + \frac{\gamma}{\beta h} \delta \boldsymbol{\eta}_i^{n+1,k}. \quad (83)$$

254 Once all kinematic variables are consistently updated  $k+1 \rightarrow k$  and a new system  
 255 matrix and residual vector can be defined. The algorithm proceeds until the  $L_2$  norm of the  
 256 incremental vector  $[\delta \check{\boldsymbol{\Theta}}_j^{n+1,k}, \delta \check{\boldsymbol{\eta}}_j^{n+1,k}]^{\top}$  is reduced below a given tolerance; after that  $n+1 \rightarrow n$   
 257 and a new time step starts with the initialization procedure discussed at the beginning of  
 258 this section.

## 259 7. Numerical results and discussion

260 In this section we present the results of three numerical examples selected to test the  
261 capabilities of the proposed formulation in different conditions, including very fast dynamics  
262 with high-frequency vibrations, very large two- and three-dimensional deformations as well  
263 as different boundary conditions. In all cases we use  $\beta = 0.25$  and  $\gamma = 0.5$  to ensure  
264 second-order time-accuracy.

### 265 7.1. Cantilever beam

266 We begin with the case of the cantilever beam that we analyzed in [34] with an IGA-C ex-  
267 plicit formulation. The test, originally proposed in [68], consists of a beam of length 1 m and  
268 with a square cross section with side 0.01 m. The Young's modulus is  $E = 210 \times 10^9$  N/m<sup>2</sup>,  
269 the Poisson's ratio is  $\nu = 0.2$  and the material density is  $\rho = 7800$  kg/m<sup>3</sup>. With respect  
270 to a Cartesian reference system  $(x_1, x_2, x_3)$ , initially the beam axis is placed along  $x_2$  and  
271 the deformation occurs in the  $(x_2, x_3)$  plane due to a constant concentrated transversal tip  
272 force  $\bar{n}_{c_3}$ . In Figure 1 the time histories of the beam tip displacements are shown. Two load  
273 intensities:  $\bar{n}_{c_3} = -10$  N and  $\bar{n}_{c_3} = -100$  N, the same as in [68] and [34], respectively, are  
274 considered. The loads are applied with constant intensity for a duration of 0.5s through a  
275 step-function without any ramp. For both cases  $p = 4$  and  $n = 20$ . An excellent agreement  
276 is found with [68] (for the small amplitude vibrations case) and with [34] for both small  
277 and large amplitude vibrations cases. The present implicit formulation appears particularly  
278 efficient since it is able to reproduce very fast nonlinear dynamics with impulsive loads (no  
279 load ramp functions are applied to any of the two load intensities) with a time step 500 times  
280 larger than the explicit formulation. Note that four iterations per time step are required in  
281 the Newton-Raphson algorithm with a tolerance on the  $L_2$  norm of the incremental vector  
282 of  $10^{-10}$ .

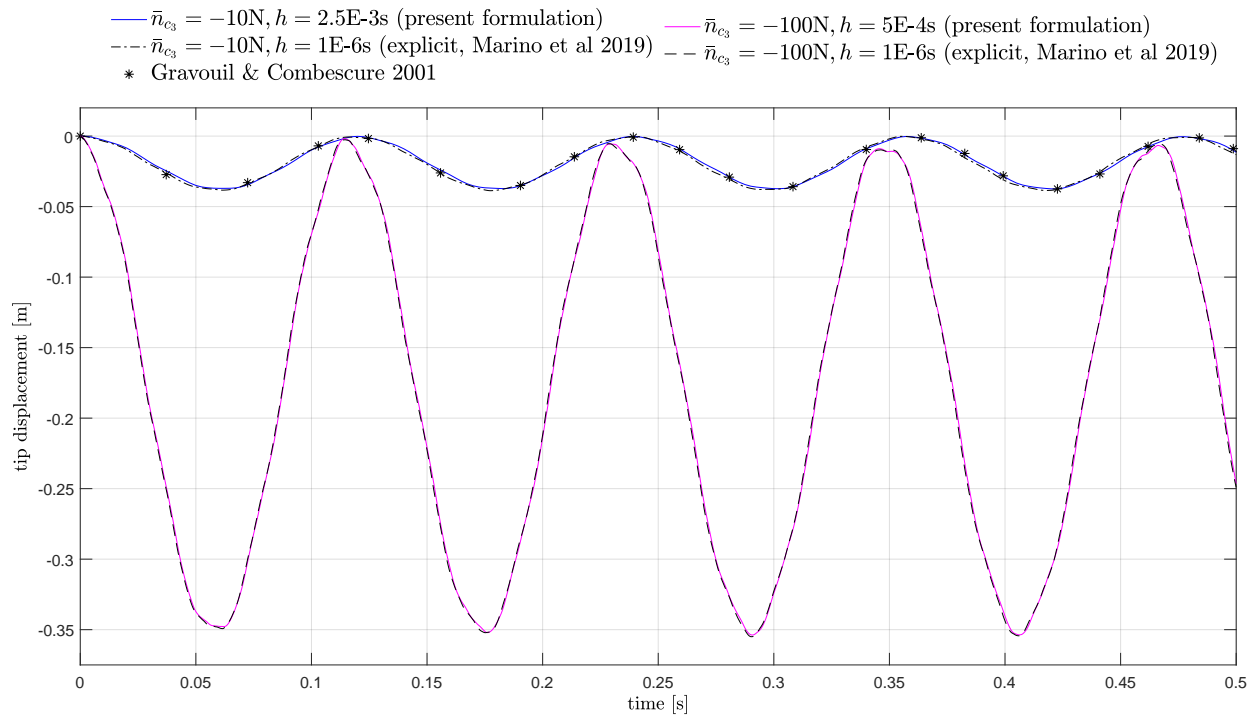


Figure 1: Tip displacement of a cantilever beam subjected to a tip transversal load  $F_3$  with two different intensities. In both cases  $p = 4$  and  $n = 20$ . Comparisons are made with results obtained in Marino et al 2019 [34] and Gravouil & Combescure 2001 [68].

283 *7.2. Swinging flexible pendulum*

284 Unlike in the previous numerical application, where a very stiff beam is considered, in this  
285 second example we study a highly flexible beam moving like a pendulum. The performance  
286 of the present formulation is analyzed through a direct comparison with the results obtained  
287 in [33, 46] and with our results obtained in [34] through an explicit formulation. The test  
288 consists of an initially straight beam of length 1 m with a circular cross section of diameter  
289 0.01 m. The Young's modulus is  $E = 5 \times 10^6 \text{ N/m}^2$ , the Poisson's ratio is  $\nu = 0.5$  and  
290 the material density is  $\rho = 1100 \text{ kg/m}^3$ . With respect to a Cartesian reference system  
291  $(x_1, x_2, x_3)$ , the beam, initially placed along  $x_2$ , is hinged at the end located at  $(0, 0, 0)$  and  
292 is free at the other end. The motion occurs in the  $(x_2, x_3)$  plane under the effect of the  
293 gravity only acting along the  $x_3$  direction. The distributed external force per unit length is  
294  $\bar{\mathbf{n}} = [0, 0, -0.8475]^\top \text{ N/m}$ .

295 Figure 2 shows eleven snapshots taken from time 0 to 1 s with increments of 0.1 s. Results  
296 associated with different combinations of basis function degrees, number of collocation points  
297 and time step sizes are shown. Furthermore, considering the solution with  $p = 6$ ,  $n =$   
298  $30$ ,  $h = 5 \times 10^{-3} \text{ s}$  (see black line in the figure) as the most accurate one among the six  
299 analyses performed, some additional observations can be made. Up to approximately 0.5 s,  
300 the differences between all cases are almost indistinguishable. After that time, the results  
301 with  $p = 4$  exhibit some loss of accuracy, while the results with  $p = 6$  are always very  
302 accurate, also when using a coarser mesh (red line) or when doubling the time step size  
303 (green line), indicating that the error due to the space discretization dominates and can be  
304 easily (and efficiently) reduced by order elevation. In the IGA-C context, where efficiency  
305 is one of the major goals, such an attribute is highly desirable since order elevation is made  
306 almost at no additional computational cost.

307 The time history of the tip displacement is shown in Figure 3, where an excellent agree-  
308 ment with the results obtained in [33, 34, 46] is found.

309 *7.3. Three-dimensional flying beam*

310 The third numerical example has been chosen to assess the capabilities of the present  
311 formulation when very large and complex three-dimensional deformations occur. The test,

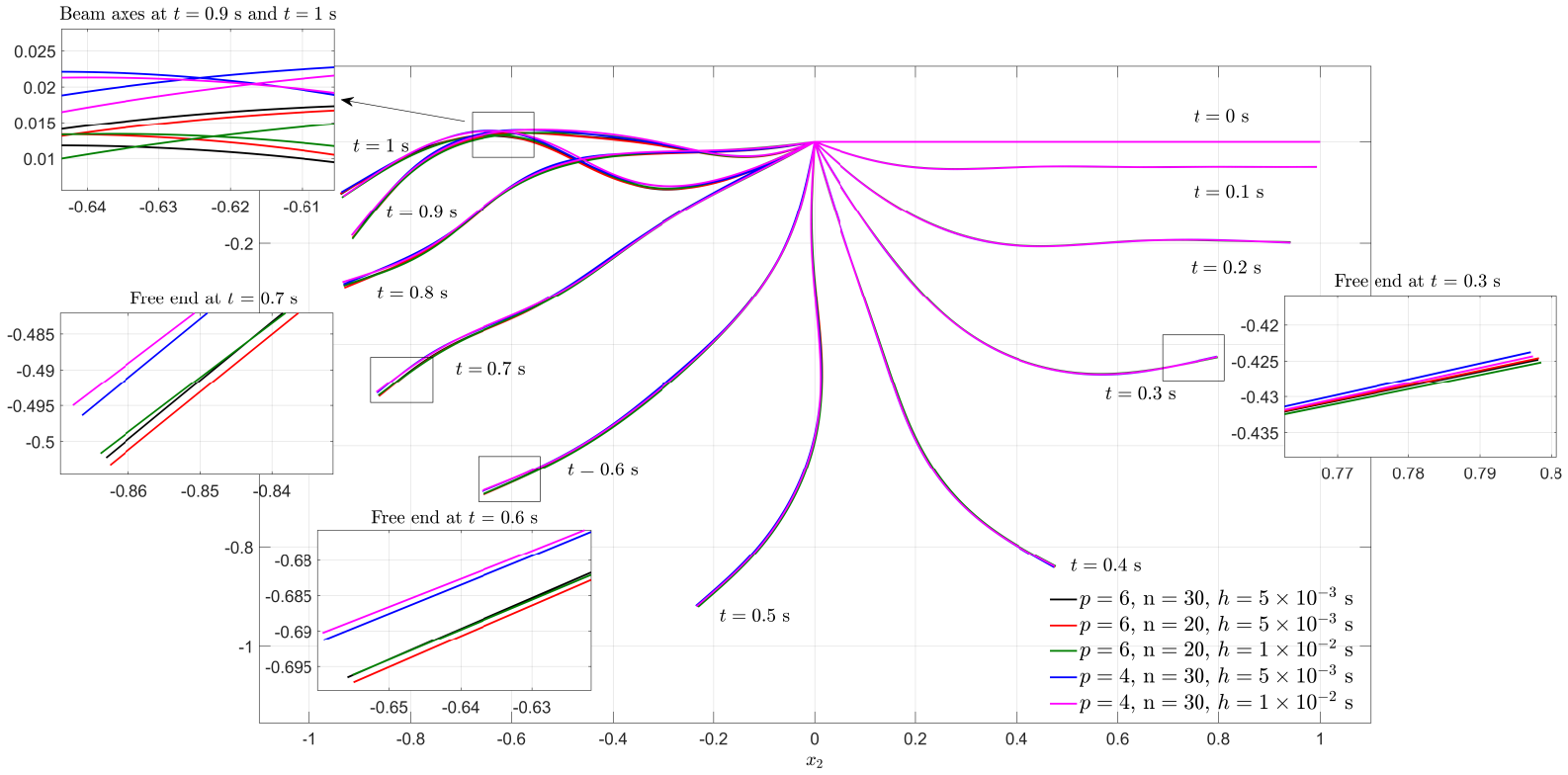


Figure 2: Snapshots of a swinging flexible pendulum from time 0 to 1 s with increments of 0.1 s for different basis function degrees, number of collocation points and time step spans.

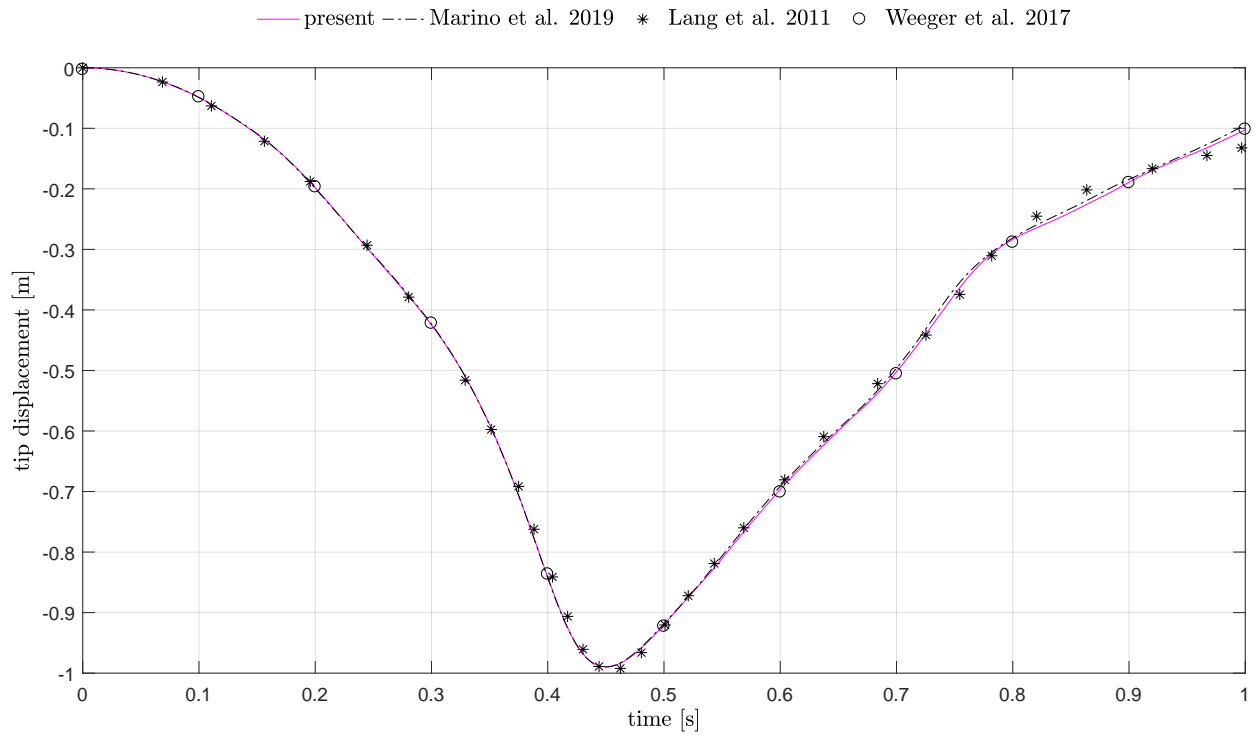
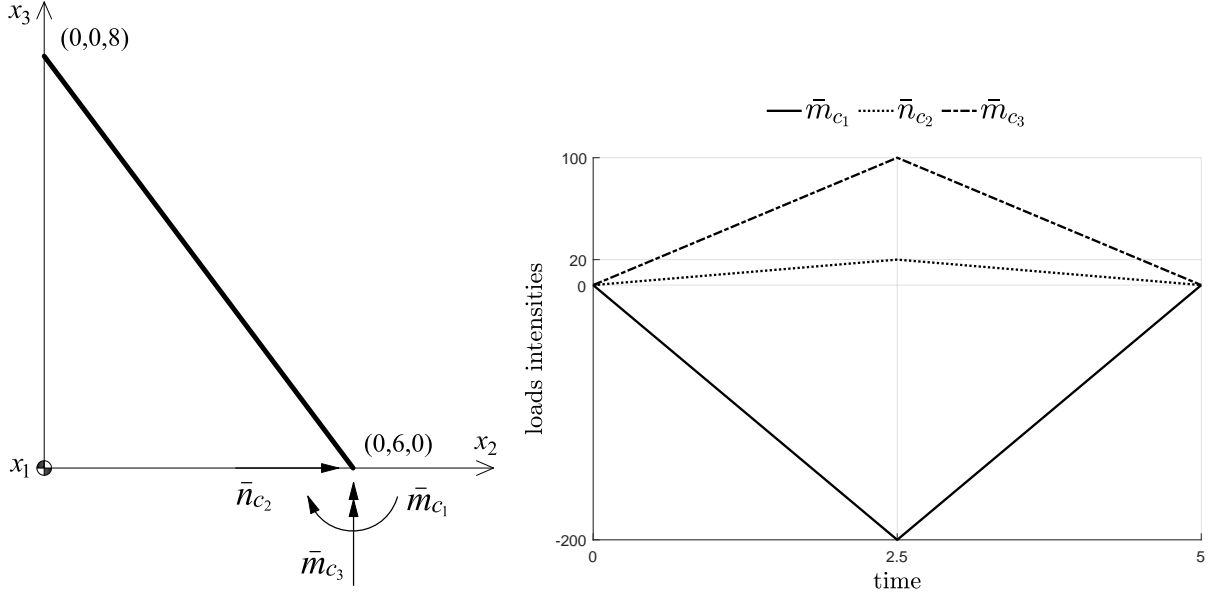


Figure 3: Vertical tip displacement of a swinging flexible pendulum. Comparisons are made with results obtained in Marino et al 2019 [34], Lang et al 2011 [46] and Weeger et al 2017 [33]. Results of the present formulation are obtained with  $p = 6$ ,  $n = 30$ ,  $h = 5 \times 10^{-3}$  s.



(a) Flying flexible beam subjected to tip force and moments. (b) Loads time histories for the flying flexible beam.

Figure 4: Flying flexible beam: initial configuration and loads.

312 proposed originally by Simo & Vu-Quoc in [36] and later studied also in [48, 49, 53, 69, 70],  
 313 consists of an initially straight free beam with length  $L = 10$  placed in the plane  $(x_2, x_3)$  (see  
 314 Figure 4(a)) subjected at the lower end to three different time-varying concentrated loads  
 315 applied simultaneously (see Figure 4(b)). Under these loads the beam undergoes a forward  
 316 translation due to  $\bar{n}_{c_2}$ , a forward tumbling due to  $\bar{m}_{c_1}$  and an out-of-plane deformation due  
 317 to  $\bar{m}_{c_3}$ .

318 First we test the high order accuracy of the formulation. The convergence curves of the  
 319  $L_2$  norm of the error evaluated at  $t = 2$  s vs. the number of collocation points are shown  
 320 in Figure 5. The error is calculated as  $err_{L_2} = \|\mathbf{u}^h - \mathbf{u}^r\|_{L_2} / \|\mathbf{u}^r\|_{L_2}$ , where  $\mathbf{u}^h$  and  $\mathbf{u}^r$   
 321 are the approximate and reference displacements, respectively. The reference solution  $\mathbf{u}^r$  is  
 322 obtained with  $p = 8$ ,  $n = 200$  (approximately 2.3 on the abscissa of Figure 5) and  $h = 0.1$ .  
 323 Very good convergence rates are observed up to  $p = 5$ . They are  $p$  for even degrees and  
 324  $p - 1$  for odd degrees, which is the typical behavior in isogeometric collocation using Greville  
 325 points [1, 2, 30]. For higher degrees, especially for  $p = 8$ , as the number of collocation  
 326 points increases, the temporal error becomes dominant and slightly affects the quality of the  
 327 convergence rate.



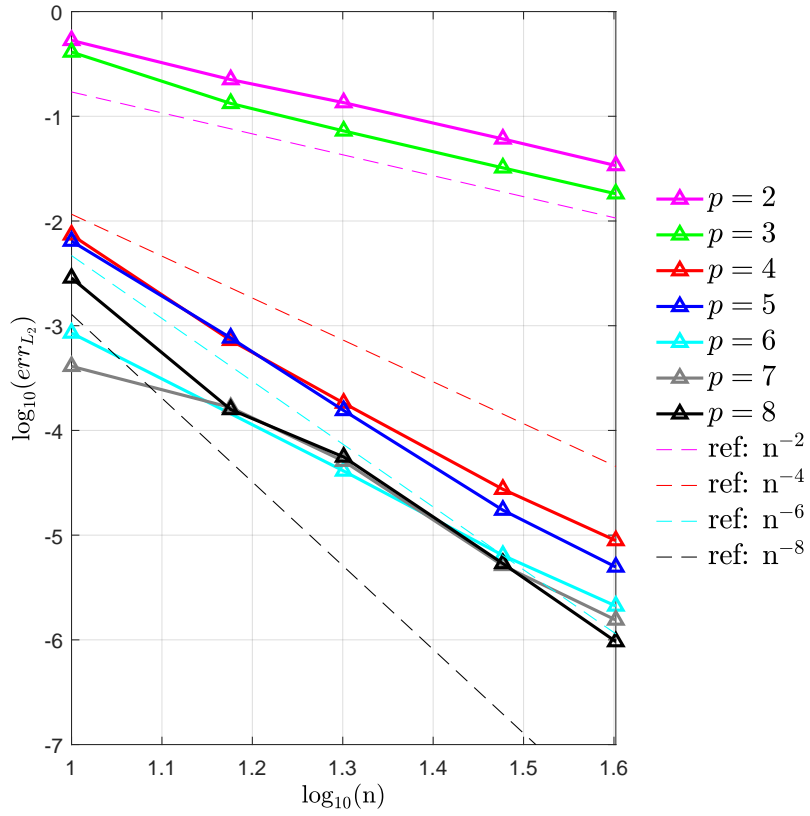


Figure 5:  $L_2$  norm of the error evaluated at  $t = 2$  s vs. the number of collocation points for the free flying beam for NURBS basis functions of degrees  $p = 2, \dots, 8$ . Dashed lines indicate reference orders of convergence. Reference solution computed with  $p = 8$ ,  $n = 200$ ,  $h = 0.1$ .

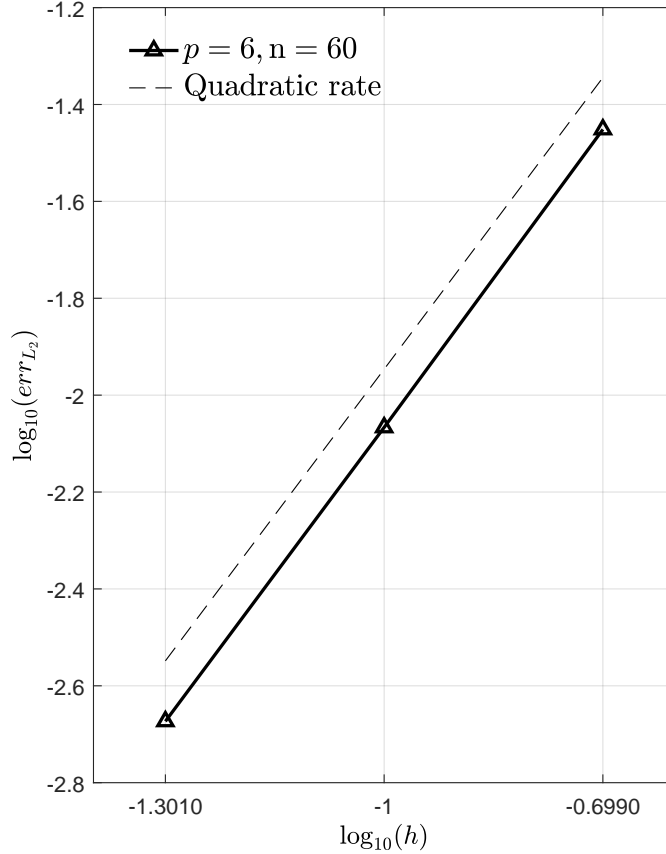
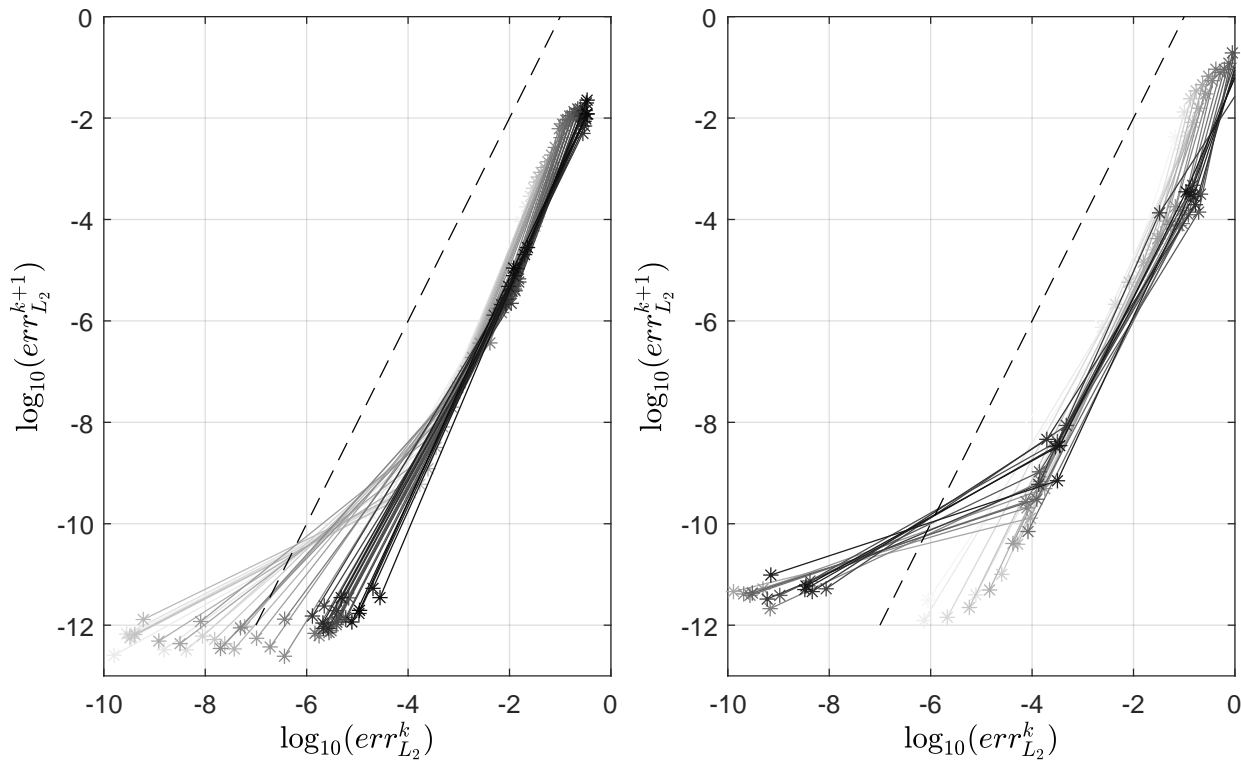


Figure 6:  $L_2$  norm of the error vs. time step sizes of 0.05, 0.1, 0.2 for the free flying beam. The error is evaluated by comparing the beam configuration at  $t = 5$  with a reference solution obtained with  $h = 0.01$ . In all cases  $p = 6$  and  $n = 60$ .

328 With our choice of  $\beta$  and  $\gamma$ , the standard Newmark time integration scheme is second-  
 329 order accurate in time [36]. To verify that this attribute is preserved in the present IGA-C  
 330 formulation on  $\text{SO}(3)$ , we show in Figure 6 the error in  $L_2$  norm associated with time step  
 331 sizes of 0.05, 0.1, and 0.2. The error is evaluated by comparing the beam configuration at  
 332  $t = 5$  with a reference solution obtained with  $h = 0.01$ . In all cases  $p = 6$  and  $n = 60$ . A  
 333 perfectly quadratic rate is observed.

334 The correctness of the linearization procedure and the construction of the tangent matrix  
 335 is confirmed by the convergence curves of the  $L_2$  norm of the incremental vector during the  
 336 iterations of the Newton-Raphson algorithm shown in Figure 7. Two cases are considered:

337 one with with  $h = 0.1$  (left panel) and one with  $h = 0.2$  (right panel). Both simulations are  
 338 performed with  $p = 6$ ,  $n = 60$  and are 5 time units long. The color of the curves shades from  
 339 light gray (initial time steps) to darker gray (last time steps). Second order convergence rate  
 340 is observed for all time instants and for both time step sizes. With  $h = 0.1$  the tolerance  
 341 of  $10^{-10}$  is reached always in four iterations, whereas with  $h = 0.2$  for some time steps five  
 342 iterations are needed to achieve the same tolerance. When at the  $(k + 1)$ th iteration the  
 343 norm of the incremental vector is already pretty small (around  $10^{-9}$ ), at the next iteration  
 344 a lower bound around  $10^{-12}$  is observed. This bound is due to multiple reasons, such as the  
 345 truncation error of the series used to express the derivatives of the exponential map involved  
 346 in the beam curvature and its derivative (see Section 6.1) and the effect of the condition  
 number on the machine precision.



(a) Time step size  $h = 0.1$  (50 curves).

(b) Time step size  $h = 0.2$  (25 curves).

Figure 7: Convergence curves of the  $L_2$  norm of the incremental vector during the Newton-Raphson iterations. The color of the curves shades from light gray (initial time steps) to darker gray (last time steps). Dashed line indicates the quadratic reference rate. Both simulations are obtained with  $p = 6$ ,  $n = 60$ . Total simulation time equals 5.

348 Figures 8 and 9 show some snapshots of the beam centroid line projected on the  $(x_2, x_3)$   
 349 and  $(x_1, x_3)$  planes, respectively; a three-dimensional view is shown in Figure 10. In all the  
 350 figures, results associated with five combinations of  $p$  and  $h$  are shown. Note that in some  
 351 cases snapshots at slightly different time instants with respect to [36] have been selected  
 352 since we considered, in addition to  $h = 0.1$ , the case with  $h = 0.2$ . When the largest time  
 353 step is considered ( $h = 0.2$ ), as expected, the temporal error dominates the spatial one.  
 354 Reducing the time step size to 0.1, the accuracy increases significantly. The beneficial effect  
 355 of  $p = 6$  over  $p = 4$  for both time steps can be qualitatively appreciated at a magnified scale  
 (see Figure 9).

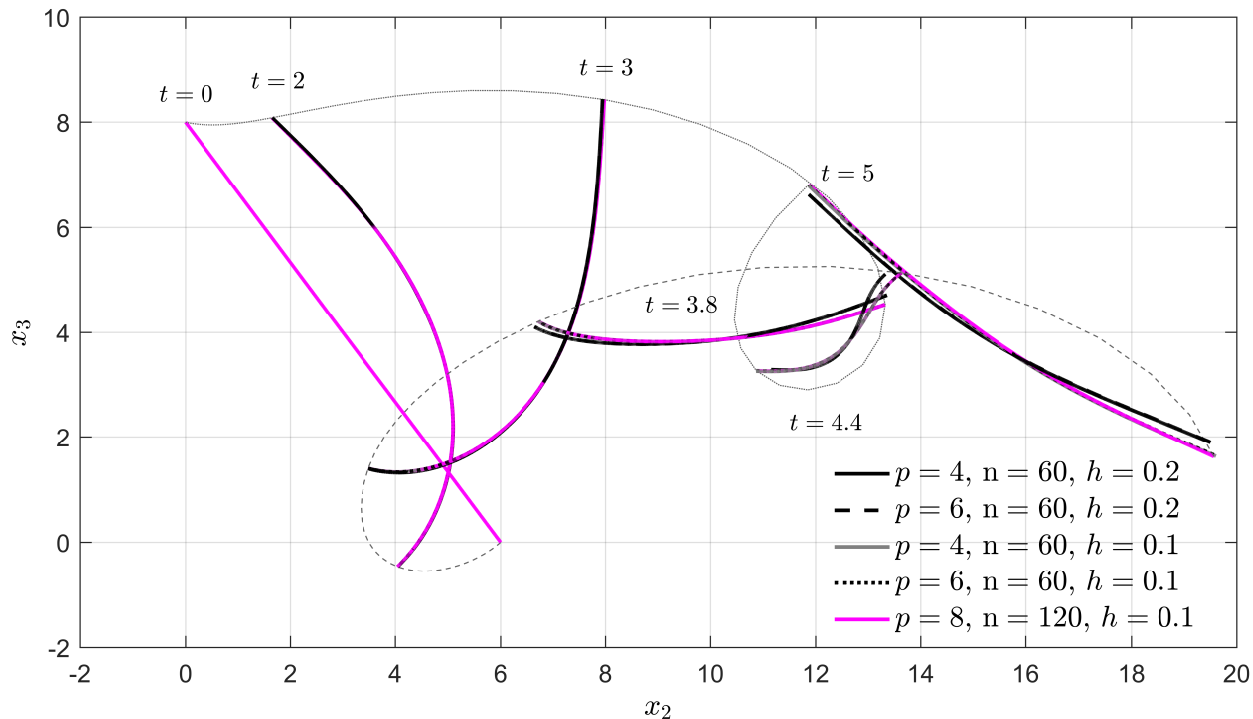


Figure 8: Snapshots of the free flying beam in the early tumbling stage projected on the  $(x_2, x_3)$  plane for different combinations of polynomial degrees, number of collocation points, and time step sizes. Gray dashed and dotted lines indicate the trajectories of the beam end points.

356

357 Figure 11 shows the material stress resultants  $\mathbf{N}$  and  $\mathbf{M}$  (see Eq. (9)) at  $t = 2.5$  for  $p = 6$ ,  
 358  $n = 60$ ,  $h = 0.1$ . No oscillatory behavior is observed and almost identical results have been  
 359 obtained with more refined meshes as well as with smaller time step sizes (these results are  
 360 not reported here as they are almost indistinguishable in the figure). The good convergence

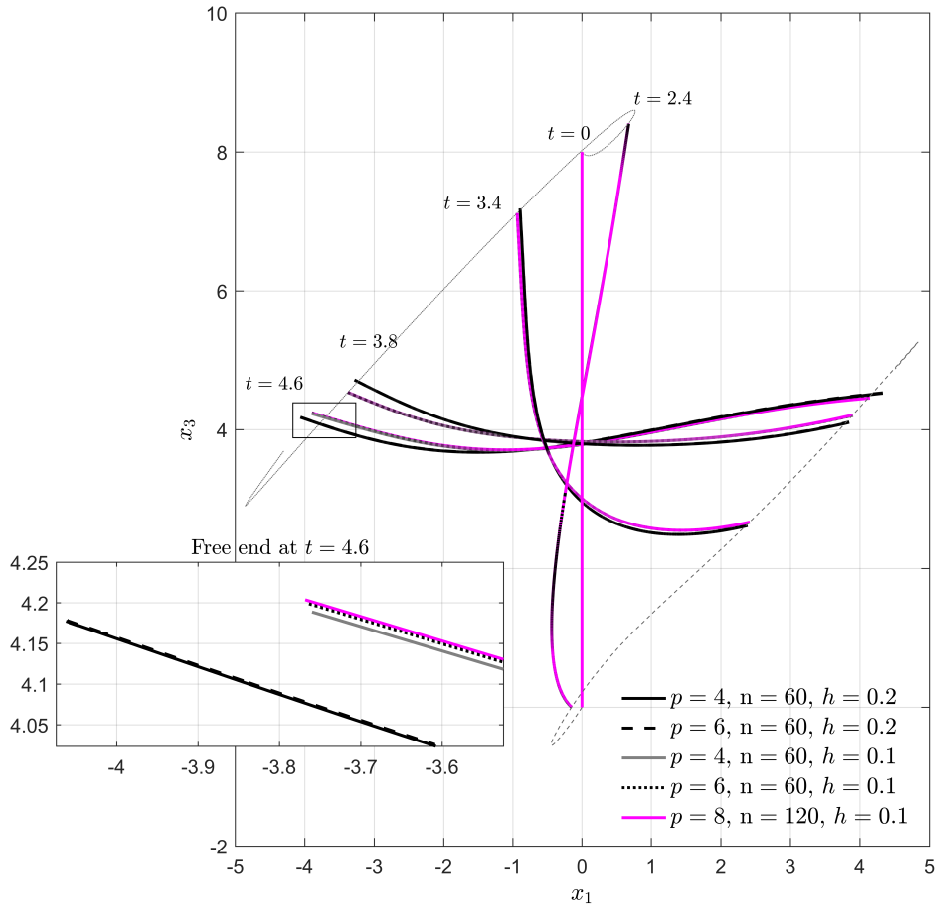


Figure 9: Snapshots of the free flying beam in the early tumbling stage projected on the  $(x_1, x_3)$  plane for different combinations of polynomial degrees, number of collocation points, and time step sizes. Gray dashed lines indicate the trajectories of the beam end points.

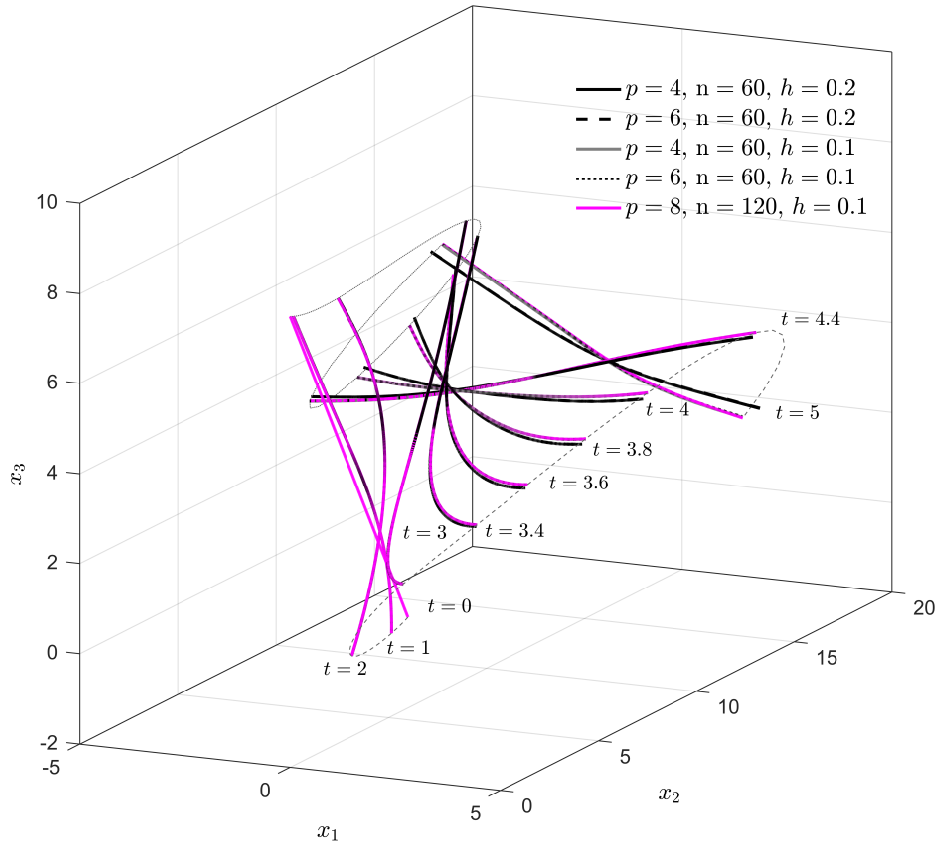


Figure 10: Snapshots of the free flying beam in the early tumbling stage in a three-dimensional view for different combinations of polynomial degrees, number of collocation points, and time step sizes. Gray dashed and dotted lines indicate the trajectories of the beam end points.

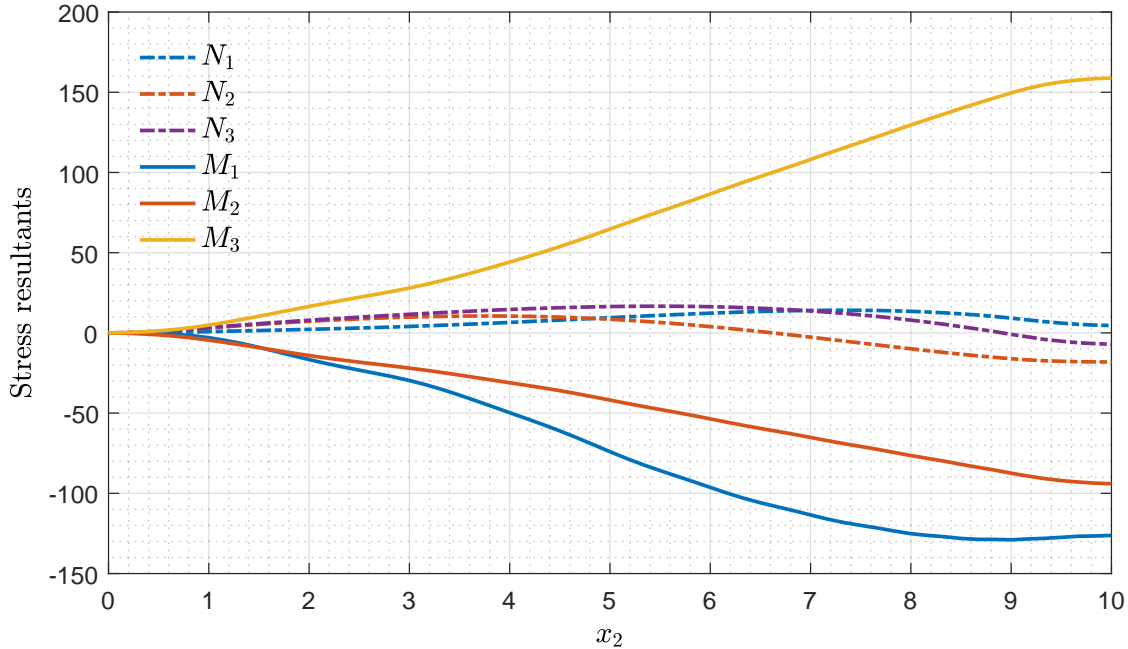


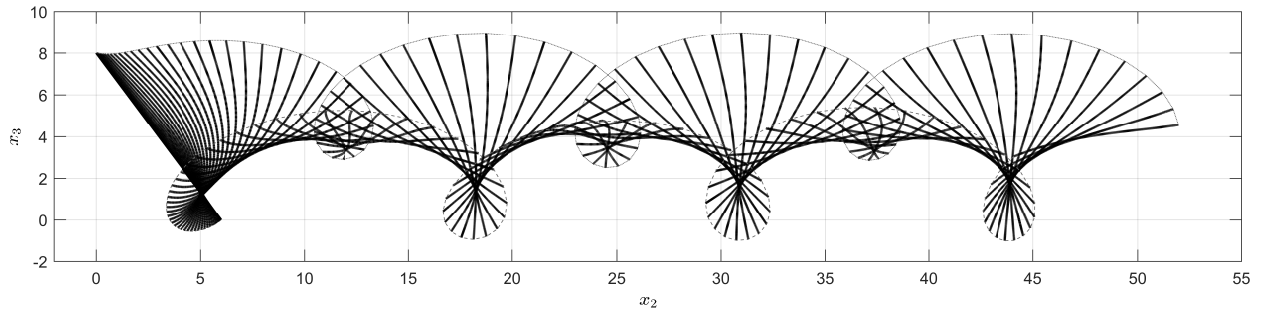
Figure 11: Free flying beam: stress resultants ( $\mathbf{N}$ ,  $\mathbf{M}$ , see Eq. (9)) at  $t = 2.5$  (maximum load intensities) over the beam length in the material setting.

361 rates observed in Figure 5 along with the smooth behavior of the stress resultants indicate  
 362 that the results are not affected by locking effects.

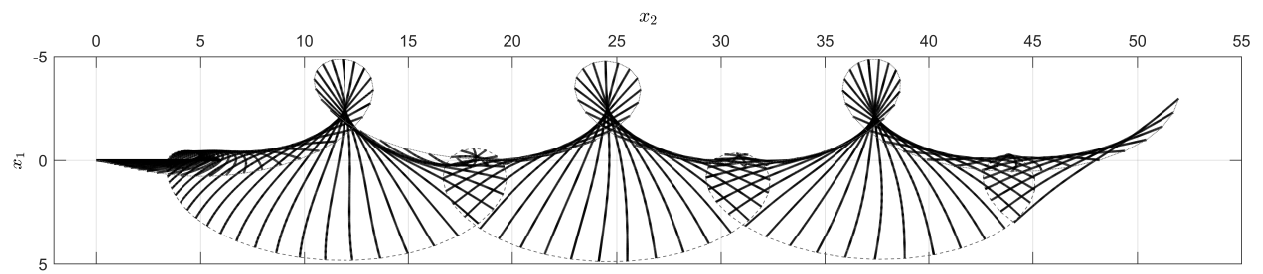
363 Note that at  $x_2 = 10$  and at time  $t = 2.5$  the stress resultants shown in Figure 11, once  
 364 rotated to the spatial setting through  $\bar{\mathbf{n}} = \mathbf{R}\mathbf{N}$  and  $\bar{\mathbf{m}} = \mathbf{R}\mathbf{M}$ , coincide with the assigned  
 365 loads given in Figure 4(b).

366 The performance of the formulation is also assessed for long simulations. Different views  
 367 of a series of configurations taken with time increments of 0.1 up to a final time of 11.5 are  
 368 shown in Figure 12.

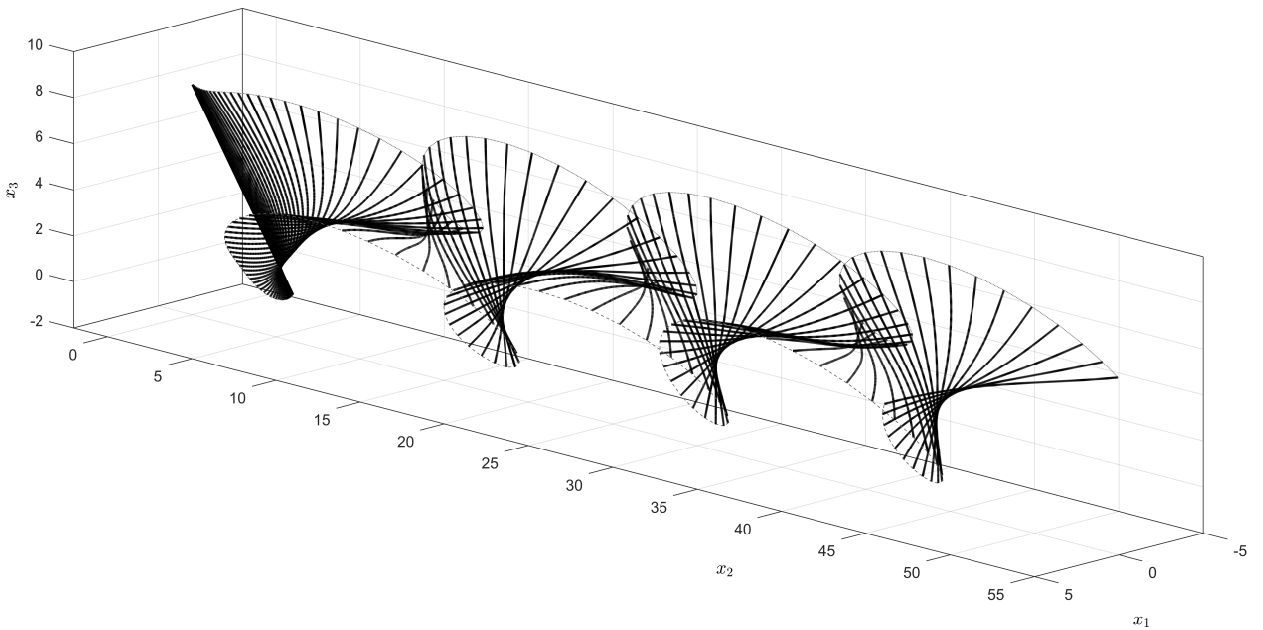
369 Finally, we remark that the investigation of conserving properties is out of the scope  
 370 of the present paper. However, since the Newmark scheme does not conserve energy nor  
 371 momentum, loss of accuracy may occur for long-term simulations. The study of energy  
 372 preserving schemes for nonlinear beams, see for example [43, 53, 71–75], is definitely an  
 373 important direction for the future developments of this work.



(a) Projections on the  $(x_2, x_3)$  plane.



(b) Projections on the  $(x_1, x_2)$  plane.



(c) Three-dimensional view with observer at  $(135^\circ, 15)$  (azimuth and vertical elevation).

Figure 12: Free flying beam: snapshots from time 0 to 11.5 with increments of 0.1.



## 374 8. Conclusions

375 With this paper we extended the field of applicability of the isogeometric collocation  
376 method to the dynamics of geometrically exact shear-deformable beams using a  $SO(3)$ -  
377 consistent version of the implicit Newmark scheme. The central issues of consistent lin-  
378 earization of the governing equations, variables initialization and update procedures are  
379 discussed in detail. In addition to very high stability, the proposed formulation ensures full  
380 consistency with the underlying geometric structure of the configuration manifold, is highly  
381 efficient due to the use of the rotation-vector parameterization, avoids the repetitive use of  
382 pull-backs and push-forwards since it is entirely formulated in the material setting, and is  
383 singularity free due to the use of the incremental rotation instead of the total rotation vector.  
384 We applied the proposed formulation to problems involving very large rotations and differ-  
385 ent boundary conditions. Correctness of the linearization and update procedures is proved  
386 by the quadratic convergence rate of the Newton-Raphson algorithm obtained in cases in-  
387 volving complex and large rotations even with a large time step size. In all cases a very  
388 good agreement with literature results is obtained. We observed that the method is stable  
389 and accurate also in cases where impulsive motions occur with loads applied without any  
390 ramp function. In addition, order elevation improves the overall accuracy significantly. In a  
391 context where efficiency is one of the major goals, this is a remarkable feature considering  
392 that order elevation is made at almost no additional computational cost. It is well known  
393 that the Newmark scheme does not conserve neither energy nor momentum and this can be  
394 a problem for long-term simulations. Future works should be oriented towards energy and  
395 momentum preserving methods with isogeometric collocation.

## 396 Acknowledgements

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399 niques in Solid Mechanics”.

400 **Appendix A. Directional derivatives**

401 In this Appendix we provide the complete set of directional derivatives needed for the lin-  
 402 earization of the governing equations discussed in Section 5. Directional derivatives are  
 403 carried out using the incremental rotation vector  $\delta\boldsymbol{\Theta}$  in the material form. Thus, this Ap-  
 404 pendix complements the formulas provided in Appendix A of [28], where the spatial form of  
 405 the incremental rotation vector was used.

*Directional derivative of  $\mathbf{c}_{,s}$ .*

$$\frac{d}{d\varepsilon} (\mathbf{c}_{\varepsilon,s})_{\varepsilon=0} = \frac{d}{d\varepsilon} [(\mathbf{c} + \varepsilon\delta\boldsymbol{\eta})_{,s}]_{\varepsilon=0} = \delta\boldsymbol{\eta}_{,s} . \quad (\text{A.1})$$

*Directional derivative of  $\mathbf{c}_{,ss}$ .*

$$\frac{d}{d\varepsilon} (\mathbf{c}_{\varepsilon,ss})_{\varepsilon=0} = \delta\boldsymbol{\eta}_{,ss} . \quad (\text{A.2})$$

*Directional derivative of  $\mathbf{R}$ .*

$$\frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon})_{\varepsilon=0} = \frac{d}{d\varepsilon} \left( \mathbf{R} \exp(\varepsilon\delta\tilde{\boldsymbol{\Theta}}) \right)_{\varepsilon=0} = \mathbf{R}\delta\tilde{\boldsymbol{\Theta}} . \quad (\text{A.3})$$

*Directional derivative of  $\mathbf{R}^{\top}$ .*

$$\frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon}^{\top})_{\varepsilon=0} = \frac{d}{d\varepsilon} \left( \exp(-\varepsilon\delta\tilde{\boldsymbol{\Theta}})\mathbf{R}^{\top} \right)_{\varepsilon=0} = -\delta\tilde{\boldsymbol{\Theta}}\mathbf{R}^{\top} . \quad (\text{A.4})$$

*Directional derivative of  $\mathbf{R}_{,s}$ .*

$$\frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon,s})_{\varepsilon=0} = \mathbf{R}\tilde{\mathbf{K}}\delta\tilde{\boldsymbol{\Theta}} + \mathbf{R}\delta\tilde{\boldsymbol{\Theta}}_{,s} . \quad (\text{A.5})$$

*Directional derivative of  $\mathbf{R}_{,s}^{\top}$ .*

$$\frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon,s}^{\top})_{\varepsilon=0} = \delta\tilde{\boldsymbol{\Theta}}\tilde{\mathbf{K}}\mathbf{R}^{\top} - \delta\tilde{\boldsymbol{\Theta}}_{,s}\mathbf{R}^{\top} . \quad (\text{A.6})$$

406 *Directional derivative of  $\tilde{\mathbf{K}}$ .* Making use of (A.4) and (A.5), we obtain

$$\frac{d}{d\varepsilon} \left( \tilde{\mathbf{K}}_{\varepsilon} \right)_{\varepsilon=0} = \frac{d}{d\varepsilon} \left( \mathbf{R}_{\varepsilon}^{\top}\mathbf{R}_{\varepsilon,s} \right)_{\varepsilon=0} = [\tilde{\mathbf{K}}, \delta\tilde{\boldsymbol{\Theta}}] + \delta\tilde{\boldsymbol{\Theta}}_{,s} , \quad (\text{A.7})$$

407 where the Lie bracket has been used  $[\tilde{\mathbf{K}}, \delta\tilde{\boldsymbol{\Theta}}] = \tilde{\mathbf{K}}\delta\tilde{\boldsymbol{\Theta}} - \delta\tilde{\boldsymbol{\Theta}}\tilde{\mathbf{K}}$ .

408 The corresponding axial vector is found by exploiting the Jacobi identity  $(\tilde{\mathbf{K}}\delta\tilde{\boldsymbol{\Theta}} - \delta\tilde{\boldsymbol{\Theta}}\tilde{\mathbf{K}})\mathbf{h} =$   
 409  $\tilde{\mathbf{K}}\delta\boldsymbol{\Theta} \times \mathbf{h}$  for any  $\mathbf{h} \in \mathbb{R}^3$ , which leads to

$$\frac{d}{d\varepsilon} (\mathbf{K}_{\varepsilon})_{\varepsilon=0} = \tilde{\mathbf{K}}\delta\boldsymbol{\Theta} + \delta\boldsymbol{\Theta}_{,s} . \quad (\text{A.8})$$

Directional derivative of  $\mathbf{R}_{,ss}$ .

$$\frac{d}{d\varepsilon} (\mathbf{R}_{\varepsilon,ss})_{\varepsilon=0} = \mathbf{R} \left( \widetilde{\mathbf{K}}^2 + \widetilde{\mathbf{K}}_{,s} \right) \delta \widetilde{\boldsymbol{\Theta}} + 2\mathbf{R}\widetilde{\mathbf{K}}\widetilde{\boldsymbol{\Theta}}_{,s} + \mathbf{R}\delta \widetilde{\boldsymbol{\Theta}}_{,ss} . \quad (\text{A.9})$$

Directional derivative of  $\widetilde{\mathbf{K}}_{,s}$ . We start with the derivative of  $\widetilde{\mathbf{K}}_{,s}$  and recall that

$$\widetilde{\mathbf{K}}_{\varepsilon,s} = (\mathbf{R}_{\varepsilon}^{\top} \mathbf{R}_{\varepsilon,s})_{,s} = \mathbf{R}_{\varepsilon,s}^{\top} \mathbf{R}_{\varepsilon,s} + \mathbf{R}_{\varepsilon}^{\top} \mathbf{R}_{\varepsilon,ss} , \quad (\text{A.10})$$

from which, by making use of equations (A.6), (A.5), (A.4), and (A.9), and after some manipulations, we obtain

$$\frac{d}{d\varepsilon} \left( \widetilde{\mathbf{K}}_{\varepsilon,s} \right)_{\varepsilon=0} = \left[ \widetilde{\mathbf{K}}, \delta \widetilde{\boldsymbol{\Theta}}_{,s} \right] + \left[ \widetilde{\mathbf{K}}_{,s}, \delta \widetilde{\boldsymbol{\Theta}} \right] + \delta \widetilde{\boldsymbol{\Theta}}_{,ss} , \quad (\text{A.11})$$

410 where again the Lie bracket have been used.

411 Again, by exploiting the Jacobi identity, the corresponding axial vector is obtained as  
412 follows

$$\frac{d}{d\varepsilon} (\mathbf{K}_{\varepsilon,s})_{\varepsilon=0} = \widetilde{\mathbf{K}} \delta \boldsymbol{\Theta}_{,s} + \widetilde{\mathbf{K}}_{,s} \delta \boldsymbol{\Theta} + \delta \boldsymbol{\Theta}_{,ss} . \quad (\text{A.12})$$

413 Directional derivative of  $\boldsymbol{\Gamma}_N$ . Making use of (A.4) and (A.1), it follows that

$$\frac{d}{d\varepsilon} (\boldsymbol{\Gamma}_{N\varepsilon})_{\varepsilon=0} = \frac{d}{d\varepsilon} (\boldsymbol{\Gamma}_{\varepsilon} - \boldsymbol{\Gamma}_0)_{\varepsilon=0} = \mathbf{R}^{\top} \boldsymbol{\eta}_{,s} + (\widetilde{\mathbf{R}^{\top} \mathbf{c}_{,s}}) \delta \boldsymbol{\Theta} . \quad (\text{A.13})$$

Directional derivative of  $\boldsymbol{\Gamma}_{N,s}$ .

$$\frac{d}{d\varepsilon} (\boldsymbol{\Gamma}_{N\varepsilon,s})_{\varepsilon=0} = \left( -\delta \widetilde{\boldsymbol{\Theta}}_{,s} + \delta \widetilde{\boldsymbol{\Theta}} \widetilde{\mathbf{K}} \right) \mathbf{R}^{\top} \mathbf{c}_{,s} - \widetilde{\mathbf{K}} \mathbf{R}^{\top} \boldsymbol{\eta}_{,s} - \delta \widetilde{\boldsymbol{\Theta}} \mathbf{R}^{\top} \mathbf{c}_{,ss} - \mathbf{R}^{\top} \boldsymbol{\eta}_{,ss} . \quad (\text{A.14})$$

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