

34 INTRODUCTION

35 Increasing exploratory and transportation activities in the Arctic and sub-Arctic regions have improved
36 activity of vessels operating along the Northern Sea Route. These vessels are required to progress
37 through level ice and this has consequently increased the demand of ice-capable ship designs. Also,
38 ensuring adequate safety in the Arctic waters requires a thorough understanding of ice load and how ice
39 deforms and fails. This knowledge is crucial to come up with efficient designs that will enable these
40 vessels to operate competently in the harsh Arctic environment. As the ship's bow comes in contact with
41 ice, the ice is crushed locally until the contact area is wide enough to induce bending failure. The material
42 properties affecting this failure behavior are site-specific and depend on many factors including
43 temperature, grain structure, grain size, salinity, brine volume, loading rate and mechanism of loading
44 (Timco and O'Brien 1994). These factors are best determined by in-situ tests even though the tests are
45 quite difficult to conduct. The severe conditions of the Arctic and sub-Arctic regions demand innovative
46 solutions as conducting such in-situ tests can become dangerous and time-consuming considering the low
47 temperatures, prevailing darkness and harsh weather. Consequently, it would be advantageous to
48 develop accurate numerical techniques that can not only predict the correct stresses and strains at failure
49 but also simulate fracture patterns and fracture locations that correspond well with those reported in the
50 literature.

51 The Finite Element Method (FEM) has been the preferred approach to model the complex failure
52 behaviour and progressive deformation of ice. Varsta (1983) was the first to present a non-linear bending
53 failure model of an ice wedge in FEM using a simplified Tsai-Wu criterion. Since, this model was not
54 thoroughly compared with experiments its accuracy could not be determined correctly. The Tsai-Wu
55 failure surface has also been used much later by Liu et al. (2011) to implement a pressure-dependent
56 material model for ship-iceberg collisions. The first significant contribution in the field of ice-structure
57 interaction came from Derradji-Aouat (2003), who developed a multi-surface failure model of saline ice
58 to capture the complex failure mechanisms associated with ship-ice interaction. This failure model, which
59 takes into account the impact of temperature, strain rate and loading direction on the material properties
60 of ice, has been implemented into an explicit finite element program by Wang and Derradji-Aouat
61 (2009). The major challenge in simulating such interactions has been to model the continuous failure
62 process of ice. Kolari et al. (2009) presented an anisotropic continuum damage mechanics model to
63 predict this continuous failure process by using a model update technique that updated and refined the
64 finite element mesh to capture the progressive fracture more accurately.

65 Although FEM is used for failure analysis of ice, it is not able to predict the continuous failure
66 mechanism satisfactorily as this introduces numerical difficulties caused by severe element distortion and
67 negative volume problems. The problem can be resolved by introducing an erosion criterion, where the
68 distorted elements are removed from the system based on a predefined failure criteria such as pressure,
69 stress, strain, damage and temperature (Swaddiwudhipong et al. 2010). Erosion is essentially a numerical
70 technique used in conjunction with FEM to predict fracture pattern, and the erosion parameters are
71 generally derived empirically or using an ad-hoc approach. Thus, the focus has been to opt for more
72 robust numerical techniques and has naturally paved the way for more innovative solutions. Gürtner
73 (2009) presented the cohesive element approach used with FEM to predict the dynamic fracture of ice.
74 This model required the knowledge of energy release rates of different fracture modes, which were
75 derived heuristically so as to attain good correspondence with experimental results. SPH is another
76 method that can successfully avoid these pitfalls found in FEM. The SPH particles inherently follow
77 deformation and offer a natural transition from a continuum to a fragmented state (Stansby and Ma
78 2016). Anghileri et al. (2005) compared the applicability of three different numerical models of FEM,
79 Arbitrary Lagrangian Eulerian (ALE) and SPH to simulate hailstone impact on aircrafts. It was found
80 that the FEM model could realistically simulate the phenomenon only when the mesh distortion was not
81 high, but the SPH method was extremely useful and showed a good correlation between numerical and
82 experimental investigations. Carney et al. (2006) also used SPH to present a phenomenological failure
83 model for high strain rate behaviour of ice. This model took into account different yield strengths in
84 tension and compression, and strain rate sensitivity of ice. Although the results showed good
85 correspondence with experiments from Kim and Kedward (2000) and Kim et al. (2003), the model
86 required an equation of state and certain parameters that had to be derived empirically to comply with
87 experimental solutions. Using Carney's material model, Keegan et al. (2013) simulated hailstone impact
88 on the leading edge of a turbine blade with the SPH method. This work not only identified an accurate
89 material model able to capture the phenomena but also a numerical tool capable of realistically
90 simulating such an event. The SPH method is a viable alternative to traditional grid-based methods for
91 simulating high-velocity impact problems, where distortions are quite high and using standard Lagrangian
92 grids result in erroneous solutions.

93 However, for low-velocity impact simulations the capability of SPH is not well documented. Some
94 relevant literature is available for deformation and fracture behaviour in rocks. Ice, like rock, shows
95 similar failure characteristics especially under compression. Both the materials exhibit brittle behaviour

96 under high strain rates and ductile behaviour under low strain rates (Renshaw and Schulson 2001).
97 Cleary and Das (2008) initially looked into the potential of SPH to model fracture of rocks and
98 demonstrated its accuracy and applicability by comparing elastic and elastoplastic deformation cases with
99 FEM simulations. Therein, it was concluded that brittle fracture of rocks under compression during low
100 impacts could be modelled with good accuracy using SPH. Ma et al. (2011) further investigated the
101 compressive fracture and crack propagation of heterogeneous rocks under uniaxial and biaxial
102 compression. The material heterogeneity of rock was successfully modelled using the Weibull
103 distribution law (Deb and Pramanik 2013). Under quasi-static conditions SPH has been successfully used
104 to model, rocks. For other rock-like materials, Tang et al. (2013) identified the factors influencing their
105 quasi-static analysis and therein concluded that with a suitable choice of the particle approximation of
106 momentum equation and smoothing length, the SPH method could favourably simulate progressive
107 failure of similar solids under compressive loads.

108 Das (2017) used the SPH method for simulating bending failure of ice. A parametric study was
109 conducted to determine the correct formulation, particle density, smoothing length and scaling factor (to
110 initial smoothing length) required for accurately simulating failure of ice using the SPH method. The
111 article validated the SPH method for simulating linear elastic bending and was subsequently extended to
112 model four-point bending failure of ice. However, the associated computational cost increased
113 considerably. Based on the limitations within which the work was carried out, and the results, it was
114 highlighted that reducing the computing time, without significantly affecting the accuracy was a salient
115 criterion for SPH to be considered as an efficient tool for predicting failure. One way to improve the
116 computational efficiency of SPH was to use a suitable alternative to the conventionally used Gaussian
117 kernel function which is computationally expensive because of its infinite range. Ito (2008) used a kernel
118 function with only even-order terms to simulate hypervelocity axisymmetric impact problems more
119 efficiently and accurately. Parallel computing algorithms have also proved to be computationally very
120 efficient for large-scale SPH simulations (Nishiura et al. 2015). Furthermore, hybrid SPH method of
121 coupling Lagrangian Finite Volume Method (FVM) and SPH was also proposed by Barcarolo (2013) to
122 improve the convergence order, accuracy and efficiency of SPH. Similar to the coupling between FVM
123 and SPH, the hybrid method of SPH-FEM has also been proven to improve computational efficiency
124 and accuracy while mitigating the core disadvantages of FEM related to failure simulation. Most of the
125 work that was carried out using coupled SPH and FEM has been limited to fluid-structure interactions,
126 high-velocity impacts and metal-cutting problems. Such coupled approaches in the field of ice failure are

127 not very well documented. Only recently, Kim (2014) briefly mentioned the coupled SPH-FEM
128 approach in the context of pressure area distribution during accidental collision of ice mass and steel
129 structures. The different SPH-FEM coupling methods (Johnson 1994), domain decomposition
130 (Lapoujade et al. 2014), time scaling (Villumsen and Fauerholdt 2008) and mass scaling (Olovsson and
131 Simonsson 2006) techniques are investigated in this paper to present their computational advantages and
132 capability to simulate bending failure accurately.

133 **METHODOLOGY**

134 **SMOOTHED PARTICLE HYDRODYNAMICS**

135 Smoothed Particle Hydrodynamics (SPH) is a mesh-free Lagrangian method that was first developed by
136 Gingold and Monaghan (1977) to study astrophysical phenomena. Since then the SPH method has
137 evolved and diversified from the initial idea to reach the desired degree of maturity required to simulate
138 complex engineering problems (Barreiro et al. 2016). Traditionally SPH has been applied for
139 simulations related to computational fluid dynamics (Colagrossi and Landrini 2003), ocean and coastal
140 engineering (Wei and Dalrymple 2016). In the field of solid mechanics Allahdadi et al. (1993) were the
141 first to introduce a three dimensional SPH code predicting the dynamic material response under shock.
142 The initial algorithm of SPH for strength of materials has been refined over the years to improve stability
143 as well as convergence and resulted in more widespread application of the method. In the SPH method
144 the problem domain is divided into a set of uniformly or randomly distributed discrete elements referred
145 to as particles. These particles not only act as interpolation points but also possess material properties
146 and interact with the external and internal forces. In the following section, the basic SPH formulations
147 are discussed briefly. For a more comprehensive description of the method the reader is referred to Liu
148 and Liu (2003).

149 Integral representation of a function

150 The integral representation of a function $f(x)$ is written as:

$$f(x) = \int_{\Omega} f(x') \delta(x - x') dx' \quad (1)$$

151 where the problem domain is given by Ω and $\delta(x - x')$ is the Dirac delta function defined as:

$$\delta(x - x') = \begin{cases} 1 & x = x' \\ 0 & x \neq x' \end{cases} \quad (2)$$

152 Kernel approximation of a function

153 For the kernel estimate, the Dirac delta function is replaced by a smoothing function $W(x - x', h)$.

$$\langle f(x) \rangle = \int_{\Omega} f(x') W(x-x', h) dx' \quad (3)$$

154 The smoothing function $W(x-x', h)$ has a compact support domain proportional to the smoothing length
 155 (h). The domain extends up to κh , where κ is a constant such that $W(x-x', h) = 0$ when $|x-x'| > \kappa h$ i.e. the
 156 smoothing function is equal to zero when the particle lies outside the domain. This is also known as the
 157 compact condition (Liu and Liu 2010). Consequently, the integration over the problem domain Ω can be
 158 reduced to κh , the support domain of the kernel function as shown in equation (4):

$$\langle f(x) \rangle = \int_0^{\kappa h} f(x') W(x-x', h) dx' \quad (4)$$

159 Smoothing function

160 A general expression for the smoothing function is written as:

$$W(x, h) = \frac{1}{h(x)^a} \theta(x) \quad (5)$$

161 where h is the smoothing length, a depends on the number of space dimensions and x is the location of
 162 the particle (Hallquist 2006).

163 Gingold and Monaghan (1977) in their first paper on SPH used a Gaussian kernel, which is given by:

$$\theta(d) = C \cdot \begin{cases} e^{-d^2} & 0 \leq d < 3 \\ 0 & d > 3 \end{cases} \quad (6)$$

164 Here, C is the normalisation constant such that $C = \left[\frac{1}{\sqrt{\pi}}, \frac{1}{\pi}, \frac{1}{\pi^{3/2}} \right]$ in $[1D, 2D, 3D]$ to satisfy the
 165 normalisation condition (Liu and Liu 2010); $d = r_{ij}/h$, r_{ij} is the distance between the particles i and j , h is
 166 the smoothing length. However using such a kernel is computationally less efficient considering its
 167 infinite range. More traditionally, for SPH simulations a cubic spline having a narrower support, as
 168 shown in Figure 1, is often used. Such a function is defined by:

$$\theta(d) = C \cdot \begin{cases} 1 - \frac{3}{2}d^2 + \frac{3}{4}d^3 & 0 \leq d \leq 1 \\ \frac{1}{4}(2-d)^3 & 1 \leq d \leq 2 \\ 0 & d > 3 \end{cases} \quad (7)$$

169 For the cubic spline kernel $C = \left[\frac{2}{3}, \frac{10}{7\pi}, \frac{1}{\pi} \right]$ in $[1D, 2D, 3D]$

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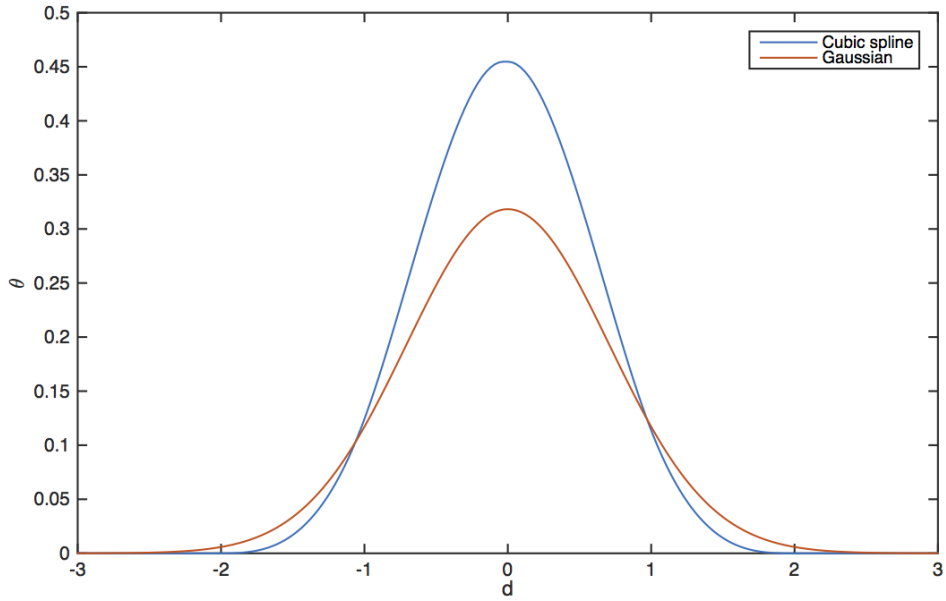


Figure 1: 2D form of cubic spline and Gaussian kernel

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175 Particle approximation

176 The particle approximation of any field variable at a particular SPH node is carried out by substituting
 177 the integral in the kernel approximation by a summation symbol which covers all the particles in the
 178 support domain of the SPH node weighted by the kernel function. Figure 2 shows the neighbouring
 179 particles in the support domain of particle i . The corresponding particle approximation of a field
 180 function $f(x)$ at particle i is written as:

181

$$\langle f(x_i) \rangle = \sum_{j=1}^n w_j f(x_j) \cdot W_{ij} \quad (8)$$

182 where $w_j = m_j / \rho_j$ and $W_{ij} = W(x_i - x_j, h)$

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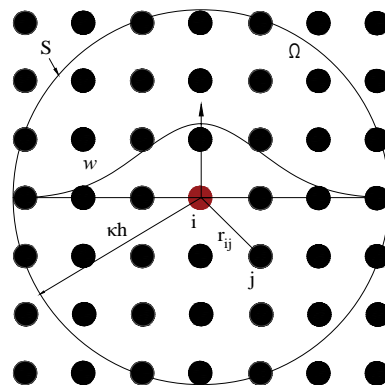
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Figure 2: Particle approximation of particle i with support domain κh

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The particle approximation of the spatial derivative of a function can be similarly expressed as

$$\langle \nabla \cdot f(x_i) \rangle = \sum_{j=1}^n w_j f(x_j) \cdot \nabla_i W(x_i - x_j, h) \quad (9)$$

192

Equation (9) shows that the SPH approximation of the gradient of a field function requires the absolute

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values of the function itself weighted by the gradient of the kernel function, which is shown in (10).

$$\nabla_i W(x_i - x_j, h) = \nabla_i W_{ij} = \frac{\partial W_{ij}}{\partial r_{ij}} \left[\frac{x_i - x_j}{r_{ij}} \right] = A_{ij} \quad (10)$$

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where r_{ij} is the distance between particles i and j

195

Governing equations

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The balance laws of continuum mechanics, in the Lagrangian frame of reference, are given in equations

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[(11)-(13)]. In the following equations d/dt indicates the total time derivative, α and β are the

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coordinate directions and components with repeated indices are summed over.

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i. The continuity equation:

$$\frac{d\rho}{dt} = -\rho \frac{\partial v^\beta}{\partial x^\beta} \quad (11)$$

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ii. The momentum equation:

$$\frac{dv^\alpha}{dt} = \frac{1}{\rho} \frac{\partial \sigma^{\alpha\beta}}{\partial x^\beta} \quad (12)$$

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iii. The energy equation

$$\frac{de}{dt} = -\frac{P}{\rho} \frac{\partial v^\alpha}{\partial x^\beta} \quad (13)$$

203

Here ρ is the density, v^α is the velocity in α direction, $\sigma^{\alpha\beta}$ is the stress tensor, e represents the energy and

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P is the pressure.

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206

Particle approximation of the governing equations

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i. The continuity equation

208 For particle approximation of the continuity equation, the approximation is applied to the divergence
 209 component of equation (11). The particle approximation of the continuity equation is written as:

$$\frac{d\rho_i}{dt} = -\rho_i \sum_{j=1}^n \frac{m_j}{\rho_j} (v_j^\beta - v_i^\beta) A_{ij} \quad (14)$$

210 It is also possible to derive various forms of particle approximation for the same conservation equation
 211 by considering different identities. An alternative expression for the continuity equation is given by:

$$\frac{d\rho_i}{dt} = -\sum_{j=1}^n m_j (v_j^\beta - v_i^\beta) A_{ij} \quad (15)$$

212 ii. The momentum equation

213 The momentum equation in (12) can be re-written as:

$$\frac{dv^\alpha}{dt} = \frac{\partial}{\partial x^\beta} \left(\frac{\sigma^{\alpha\beta}}{\rho} \right) + \left(\frac{\sigma^{\alpha\beta}}{\rho^2} \right) \frac{\partial \rho}{\partial x^\beta} \quad (16)$$

214 The particle approximation of this equation can be written as:

$$\begin{aligned} \frac{dv_i^\alpha}{dt} &= \sum_{j=1}^n \frac{m_j}{\rho_j} \left(\frac{\sigma_j^{\alpha\beta}}{\rho_j} \right) A_{ij} + \frac{\sigma_i^{\alpha\beta}}{\rho_i^2} \left(\sum_{j=1}^n \frac{m_j}{\rho_j} \rho_j A_{ij} \right) \\ \frac{dv_i^\alpha}{dt} &= \sum_{j=1}^n m_j \left(\frac{\sigma_j^{\alpha\beta}}{\rho_j^2} A_{ij} + \frac{\sigma_i^{\alpha\beta}}{\rho_i^2} A_{ij} \right) \end{aligned} \quad (17)$$

215 Since the gradient of a cubic spline is antisymmetric i.e. $A_{ij} = -A_{ji}$ we can write:

$$\frac{dv_i^\alpha}{dt} = \sum_{j=1}^n m_j \left(\frac{\sigma_j^{\alpha\beta}}{\rho_j^2} A_{ij} - \frac{\sigma_i^{\alpha\beta}}{\rho_i^2} A_{ji} \right) \quad (18)$$

216 iii. The Energy equation:

217 The energy equation can be re-written in the form of the continuity equation:

$$\frac{de}{dt} = -\frac{P}{\rho} \frac{\partial v^\alpha}{\partial x^\beta} = \frac{P}{\rho^2} \left(-\rho \frac{\partial v^\alpha}{\partial x^\beta} \right) = \frac{P}{\rho^2} \frac{d\rho}{dt} \quad (19)$$

218 Subsequently, the particle approximation of this form of the energy equation can be written using
 219 equation (15):

$$\frac{de_i}{dt} = -\frac{P_i}{\rho_i^2} \sum_{j=1}^n m_j (v_j^\beta - v_i^\beta) A_{ij} \quad (20)$$

220 In SPH, the particle approximation is carried out at each time step and depends on the current local
 221 distribution of the particles. However, it is necessary for computational reasons to keep the mass and
 222 number of particles inside the influence domain unchanged. Thus, if M denotes the total mass of the
 223 spherical influence domain, then to keep this total mass unchanged dM/dt must be zero and the

224 conservation of mass in (11) will require that the smoothing length is calculated as a function of the
225 divergence of velocity as shown in (21). This equation suggests that the number of neighbours inside the
226 influence domain will increase during expansion and decrease during compression. Moreover, the
227 number of particles inside the support domain has a direct consequence on the simulation result. If the
228 number is too small, the simulation is unable to simulate the interaction between the particles realistically
229 or if it is too large, the local properties of the particles are smoothed out thereby affecting accuracy.

$$\frac{dh}{dt} = \frac{1}{3} h \cdot \text{div}(v) \quad (21)$$

230 In spite of the advantages that the SPH method offers, it suffers from certain disadvantages like lack of
231 consistency, tensile instability and reduced accuracy at the boundary. These inherent drawbacks can be
232 resolved by a combination of corrective measures best described in Swegle et al. (1995), Liu (2003) and
233 Lacombe (2001). Apart from these drawbacks, the SPH is also less efficient computationally compared to
234 other traditional numerical methods. However, since it is based on the Lagrangian approach it can be
235 linked to a standard Lagrangian model like the FEM. Such linked (or coupled) models provide the
236 capability to carry out simulations involving both fracture and structural response at a reduced
237 computational cost (Johnson 1994).

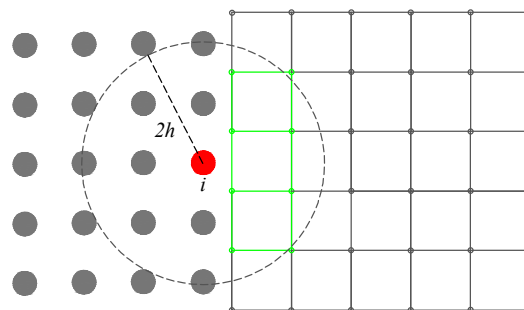
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239 SPH-FEM COUPLING TECHNIQUES

240 The FEM has been extensively used in the field of continuum mechanics for modelling continuous
241 matter (such as solids, fluids etc.). However, one of the most challenging aspects of FEM is its inefficient
242 handling of problems involving large deformation, fracture and fragmentation (Chen et al. 2013). If solid
243 elements are used to predict progressive fracture or deformation, excessive mesh distortion results in
244 zero or negative volume problems, which introduces numerical difficulties like premature termination.
245 To resolve this, the mesh-free approach of SPH can be used. This method successfully mitigates the
246 aforementioned disadvantages of FEM and is well suited for problems involving fracture or
247 fragmentation. However, one major drawback of SPH is the increase in computational resource
248 requirement. To optimize the computational cost, both SPH and FEM can be coupled. The SPH
249 method can be embedded within an existing FEM code by considering SPH particles as FEM nodes. In
250 SPH-FEM coupling, SPH particles are used in regions where fracture or deformation is likely to be
251 predominant, and FEM for discretizing the rest of the model. Such coupling not only helps in reducing
252 the computation cost, by decreasing the number of active SPH particles, but also helps in accurately

253 simulating fracture without the numerical instability found in FEM. Furthermore, the accuracy of the
254 analysis is also improved if the problem domain comprises solid elements in addition to SPH particles
255 (Swaddiwudhipong et al. 2010).

256 An overview of the code structure for SPH and Lagrangian FEM provided in Johnson (1994) showed
257 that the major differences between the two methods involve only the calculation of strains, strain rates
258 and forces. Another important consideration is that the same material model can be used for both SPH
259 particles and finite elements. Thus, it was possible to combine both the methods with appropriate
260 conditions enforced at the SPH-FEM interface. At this SPH-FEM interface, the particles are constrained
261 and moved with the elements. The interface also ensured continuous bonding between the two methods.
262 In Figure 3 the particle "i" includes both finite elements (marked in green) and SPH particles in its
263 influence domain. Certain considerations are required in the way computations are carried out for
264 particles at such interfaces. While determining the strain and strain rate at each SPH particle at the
265 interface, only the SPH nodes within the support domain are considered, whereas while calculating the
266 internal forces, contributions from both finite elements and particles inside the influence domain are
267 required to be included Johnson (1994). This coupling can be done using a tied contact algorithm or
268 using an adaptive coupling technique, both of which are discussed briefly in the following section.



269 Figure 3: Coupling of SPH and FEM at the interface

270 Tied contact

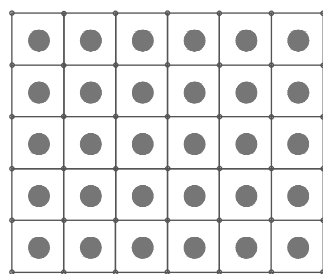
271 A tied contact algorithm can be used to connect the SPH nodes with solid elements. Contact detection
272 algorithm requires a set of nodes called the slave nodes, and a set of surface patches called the master
273 surfaces to be defined (Attaway et al. 1994). During modelling, a node to surface contact was used with
274 the SPH particles as the slave nodes, and FEM elements at the interface as the master surface. During
275 initialization, all the slave nodes within the allowable standard projection from the master surface, was
276 tied and moved to the master surface to ensure zero gaps between FEM and SPH particles. Furthermore,

277 the displacement of the slave nodes was monitored throughout the calculation to check for penetration.
 278 If the slave nodes penetrated the master surface, a contact constraint was applied to push back the slave
 279 nodes on to the master surface. Instead of a tied contact, a tied offset contact could also be used. For
 280 such contact algorithms, the slave nodes were not directly linked to the master surface. Rather, an offset
 281 vector was defined at a specific distance from the master segment, and the slave (SPH) nodes were
 282 connected to the end of this offset vector.

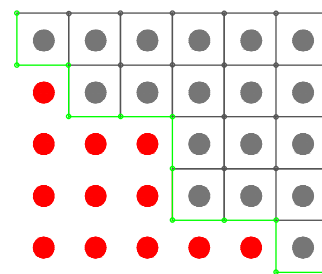
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 284 Adaptive coupling method

285 In FEM simulations fracture is often associated with numerical erosion of the failed element. This
 286 erosion function is not a material property, but a numerical technique and is a very useful way to simulate
 287 fracture. When an element is eroded its mass is either reduced to zero, which violates conservation of
 288 mass, or is redistributed to the corner nodes. Nevertheless, both the compressive strength and the
 289 internal energy of the eroded element are discarded (Kala et al., 2016). One way to solve this erosion
 290 problem is to use adaptive coupling where the SPH elements replace the solid parts once they fail. The
 291 activated SPH particle inherits all the mass, kinematic variables and constitutive properties of the failed
 292 element.

293 In adaptive coupling, the SPH particles are included in the model from the beginning of a simulation.
 294 However, they remain inactive until failure occurs so that the computational requirement is not
 295 increased. Once failure occurs, the failed element is replaced by 1, 2 or 3 SPH particles. The inactive
 296 SPH particles, at the start of a simulation, are marked in grey in Figure 4. Here, only one SPH particle is
 297 placed at the centre of gravity of every element for representation. At the onset of failure, the failed
 298 element is replaced by one SPH particle, marked in red in Figure 4(b). A contact interface, shown in
 299 green, gets activated for the interaction between the undamaged FEM model and the activated SPH
 300 particles (Kala and Hušek 2016).



(a) Finite element mesh with SPH particles in the background



(b) SPH-FEM contact algorithm for adaptive coupling

301

Figure 4

Although the SPH-FEM coupling is not entirely a new concept, the relevant literature on the efficiency and applicability of such methods to simulate quasi-static phenomena is very limited. The outcome from the numerical experiments are compared with the results of a series of four-point bending tests conducted by Kujala et al. (1990) at the Gulf of Finland near Porkkala pilot station and serves as the basis for comparing the efficiency and applicability of the numerical techniques presented in this article. The experiment had a loading rig that was set up to ensure that bending took place upwards to bring about tensile failure, which is the preferred mode of failure for ice going vessels. The loading rig had two moving supports, spaced 1m apart, and was used to bend the beam against two fixed supports, spaced 4m apart. A hydraulic cylinder, regulated by an electronically controlled pressure valve, produced the bending force, which was measured by a force gauge attached to the hydraulic cylinder and linear potentiometers measured deflection at each support and in the middle of the beam. The beams were sawn from sea ice cover by a chainsaw to get straight edges. Furthermore, before conducting the tests, surface temperature and dimensions of the beam were measured, and the loading rig was adjusted such that the fixed top support of the loading platform and the upper surface of ice beams were at the same level. The final deflection was calculated by subtracting the average displacement at the end supports from the net deflection, measured at the middle of the beam and was considered necessary as the top supports protruded slightly into the ice beam. The flexural strength values were about 24% higher compared to results from (Enkvist 1972) and 30% higher values compared to Maattanen (1975). Also, the elastic modulus determined from the experiments was found to be 5% to 20% greater than in cantilever tests. In this article, the results from four-point bending simulations are compared with experimental results from Kujala et al. (1990).

NUMERICAL SIMULATION

The numerical experiments are compared with the bending tests of natural sea ice conducted by Kujala et al. (1990). Table 1 lists the properties of ice measured from the experiments.

Table 1: Material properties of Baltic Sea ice

Elastic Modulus (E) (GPa)	Shear modulus (G) (GPa)	Bulk modulus (K) (GPa)	Density (ρ) (Kg/m ³)	Poisson's ratio (ν)
4.154	1.562	4.073	917.12	0.33

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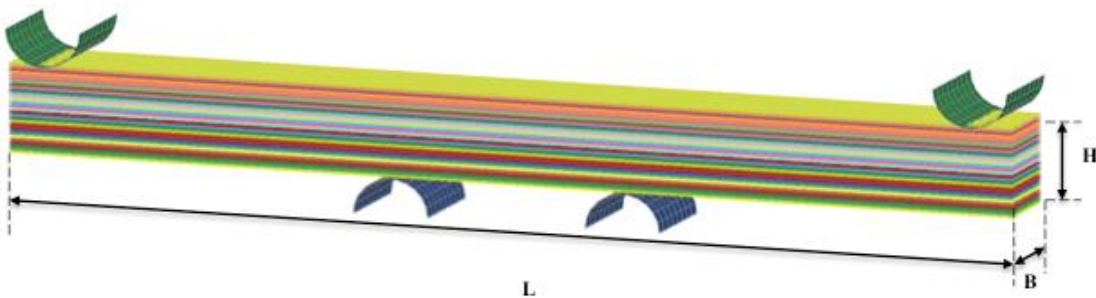
329 Kujala et al. (1990) also presented a sensitivity study to determine how the Young's Modulus and the
330 flexural strength varied with temperature through the thickness of the ice beam. He assumed a linear
331 temperature gradient and constant salinity for the sensitivity study. Table 2 shows the calculated top (E)
332 and bottom elastic moduli (E') and the bending strength at the top surface (S). Here, F = load at failure, L
333 = distance between top supports and δ is the deflection.

334 Table 2: Calculated elastic moduli at top and bottom surface as well as bending strength

E (GPa)	α	E' (GPa)	S (MPa)
8.60	0.17	1.462	0.81

335

336 The SPH model of an ice beam in four-point bending had three parts viz. the sea ice beam, upper and
337 lower supports. The model is shown in Figure 5 where L , B and H denote the length, width and height
338 of the beam respectively. The measurements were directly taken from the in-situ beam bending
339 experiments. The SPH model had 702240 SPH nodes, and the supports were modelled as a rigid
340 material with 352 shell elements having an edge length of 0.05m. A penalty contact was required to be
341 defined to achieve contact between the rigid supports and the SPH ice beam (Hallquist 2006). The upper
342 support was fixed and the lower support was allowed to move in the Z -direction. The different colours in
343 Figure 5 indicate the different layers of the beam with varying material properties. A constant
344 displacement velocity (δ) was applied to the lower supports in the positive z -direction and this velocity
345 was calculated from full-scale test data.



346

347 Figure 5: SPH model of an ice beam

348 MATERIAL MODEL

349 The experimental results showed only brittle elastic failure of the ice beams and consequently an elastic
350 failure model was selected. This model required the values of seven material parameters viz. density (ρ),
351 Shear Modulus (G), yield strength (σ), plastic hardening modulus (E_p), Bulk Modulus (K), pressure cut-

352 off (p) and failure strain (ε). Failure was specified either by the pressure cut-off (p) or failure strain (ε). At
 353 the onset of failure, the deviatoric stress components were reduced to zero, and the material could only
 354 carry compressive loads (Hallquist 2006).

355 The Von Mises yield criterion for the material is given by

$$\phi = J_2 - \frac{\sigma_{yield}^2}{3} \quad (22)$$

356 where J_2 is the second stress invariant and can be defined in terms of deviatoric stress components as

$$J_2 = \frac{1}{2} s_{ij} s_{ij} \quad (23)$$

357 The yield stress (σ_{yield}) is a function of effective plastic strain (ε_{eff}^p) and the plastic hardening modulus (E_p)

$$\sigma_{yield} = \sigma_0 + E_p \varepsilon_{eff}^p \quad (24)$$

358 The plastic hardening modulus (E_p) can be defined in terms of the tangent modulus (E_t) as

$$E_p = \frac{EE_t}{E - E_t} \quad (25)$$

359 The effective plastic strain is defined as:

$$\varepsilon_{eff}^p = \int_0^t d\varepsilon_{eff}^p \quad (26)$$

360 The pressure is given by equation (27), where K is the Bulk Modulus and V is the volume

$$p^{n+1} = K \left(\frac{1}{V^{n+1}} - 1 \right); \quad (27)$$

361 The failure is assumed to occur if

$$p^{n+1} < p_c \text{ OR } \varepsilon_{eff}^p > \varepsilon_f \quad (28)$$

362 where p_c and ε_f are user defined parameters.

363 In the experiment, the effective elastic modulus (E_{eff}) and bending strength were calculated from the
 364 deflection. The flexural strength and elastic modulus depends on the temperature that varied through the
 365 thickness of the beam. A linear temperature gradient and constant salinity was assumed Kujala et al.,
 366 (1990) to present the variation of Young's modulus and flexural strength from the top to the bottom of
 367 the beam. The plastic hardening modulus in Table 3 was assumed to be slightly lower than the elastic
 368 modulus to differentiate between elastic and plastic behaviour. The yield strength was equal to the failure
 369 stress at the top of the beam as shown in Table 3. Since it was preferred to initiate failure based on tensile

370 pressure, the failure strain (ϵ) was assumed to have a large value to not let the beam fail due to strain
 371 criteria. The pressure cut-off value was taken from Soa (2011).

372 Table 3: Input values for the material model

ρ (kg/m ³)	G (GPa)		σ_c (MPa)	E_p (GPa)		K (GPa)		p_c (MPa)
	Top	Bottom		Top	Bottom	Top	Bottom	
917.12	3.233	0.550	0.81	8.428	1.433	8.431	1.433	-0.2645

373

374 RESULTS AND DISCUSSION

375 The results from a parametric study of an SPH bar identified the correct formulation, number of
 376 particles, smoothing length and scaling factor (to the initial smoothing length) required to simulate tensile
 377 and compressive behaviour accurately using the SPH method (Das, 2017). Since beam bending includes
 378 both tensile and compressive stresses on either side of the neutral axis, this study was naturally extended
 379 to simulate four-point bending failure of an ice beam using SPH. In this study, the material properties of
 380 ice presented in Table 3 were used to simulate a FEM model of the same beam in four-point bending.
 381 Although the FEM is one of the most extensively used numerical tools in the field of solid mechanics, it
 382 suffers from a few limitations while simulating fracture or fragmentation. One such challenge is element
 383 distortion and a way to resolve this is through numerical erosion of solid elements. However, the
 384 capability of the finite element model to simulate the physical phenomena accurately depends on the
 385 correlation between the discretization length and the solid elements. Thus, such deletion of elements (or
 386 drop in mass) is a direct violation of the conservation of mass and is not preferred (Das et al., 2014).
 387 Besides, in an SPH analysis, no particles are removed from the problem domain at failure, and the mass
 388 remains constant throughout the simulation (as only the inter-particle connections fail and the particles
 389 are not removed from the problem domain). This is considered to be a notable advantage of SPH over
 390 FEM, especially in simulations where fracture processes play a dominant role. Furthermore, the
 391 comparisons of the FEM results with SPH also showed that the breaking location was similar in both the
 392 models. However, the fracture pattern from the SPH simulation appear more realistic because SPH
 393 offers a natural transition from a continuum state to a fragmented state at the point of failure (Stansby and
 394 Ma 2016). Once failure occurred, the deviatoric stresses in the SPH particles were reduced to zero and
 395 the particles only carried stresses in compression, displaying fluid-like behaviour. Such a solid-to-fluid

396 transition of the SPH particles was regarded as failure. In contrast, for FEM, such a transition state could
397 only be simulated by physically removing the elements from the model (erosion).

398 Table 4: Comparison of SPH and FEM results

METHOD	Failure load (kN)	Failure time (sec)	Deflection (mm)	CPU time (hrs)
Experiment ¹	5.69	0.34	1.07	–
SPH	5.98	0.34	1.16	27.10
FEM	5.56	0.34	1.16	2.500

399 ¹ (Kujala et al. 1990)

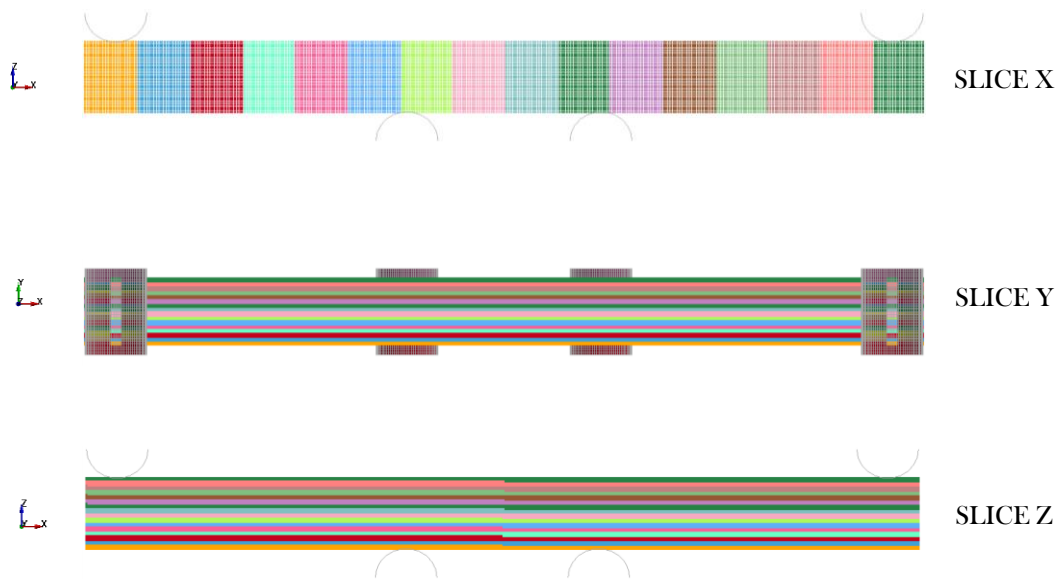
400 While these advantages of particle methods over traditional FEM for predicting fracture is quite clear,
401 one cannot overlook the computational demand of SPH considering that the FEM model was
402 significantly faster than SPH (Table 4). Therefore, this study looked into the different methods that could
403 be applied to reduce the CPU costs associated with an SPH simulation. The applicability of domain
404 decomposition, time scaling, mass scaling and SPH-FEM coupling techniques to improve the
405 computational efficiency were investigated. The results were compared with the experiments to
406 determine the error and the improvement in computational resource requirement. A satisfactory way to
407 present the accuracy of the investigated results were taken from Villumsen et al. (2008), where the
408 percentage differences in failure and displacement outputs were compared with experimental data.
409 However, it was not possible to formulate a generic answer for the acceptable percentage difference. It
410 was important to keep this as low as possible, but taking into consideration, the extreme conditions
411 prevalent in the Arctic, the measuring uncertainty of the mechanical properties of ice, and the accuracy of
412 the experimental setup, an error percentage of less than ten was considered to be satisfactory for the
413 present study.

414

415 DOMAIN DECOMPOSITION

416 The size of numerical models has been continually increasing in recent times, and it is now quite
417 common to handle models with several millions of elements/particles (computational nodes).
418 Consequently, there has been a growing demand for parallel and scalable computations through Shared
419 Memory Parallel (SMP) or Massively Parallel Processing (MPP). The SMP provides a parallel
420 architecture where a model is solved using multiple processors on the same machine. Naturally, the
421 processors share the hardware resources and operating system, and communicate through shared
422 memory. However, the SMP shows limited scalability for more than eight processors. For larger
423 problems, a more favourable option is the MPP that is a synchronized handling of a task by several
424 processors having their own set of hardware resources. The MPP capability of LS-DYNA uses a data-

425 partitioning algorithm that divides a given model into subdomains, each of which is assigned to a
 426 particular processor. Every processor independently advances the solution for its subdomain until the
 427 end of the time step after which the processors communicate through a Message Passing Interface (MPI).
 428 For FEM, this communication only happens while updating the nodal displacements and forces.
 429 Contrary to this, for SPH, the processors are additionally required to communicate with each other.
 430 Firstly, during particle sorting, which determines the neighbour list in the sphere of influence of each
 431 SPH particle. Secondly, during calculating strain rates. And thirdly while computing internal forces on
 432 each SPH particle and its neighbours. Furthermore, in Lagrangian FEM, only the nodes that are located
 433 at the boundary between decomposition subdomains govern the communication between processors.
 434 However, for SPH, any particle having a neighbour in another subdomain will affect the communication
 435 requirements. This means that the processors are required to communicate for every particle that is
 436 located at a distance less than twice the smoothing length. Consequently, the communication
 437 requirements for SPH are not only more frequent, but also the volume of data transferred between
 438 processors is higher. Thus, the data partitioning algorithms of MPP can significantly affect the efficiency
 439 of a given problem Lapoujade et al. (2014). In the present study, three different domain decomposition
 440 methods were investigated by dividing the model along one of the three coordinate axes, viz. SLICE X,
 441 SLICE Y and SLICE Z (Figure 6), to determine the most efficient partitioning algorithm for the problem
 442 at hand.



443 Figure 6: Different domain decomposition techniques
 444 The load deformation response and fracture location for all the three decomposition techniques

445 correlated quite well. However, the computational efficiency for the three decomposition schemes
 446 differed. Table 5 summarizes the results from the different decomposition schemes. SLICE X was
 447 computationally most efficient and was 1.18 times faster than SLICE Y and 1.13 times faster than SLICE
 448 Z. SLICE X has been used for all the numerical experiments presented hereafter in this article.

449 Table 5: Comparison of results for different decomposition schemes

Decomposition Scheme	Deviations in output at failure (%)		CPU time (hrs)
	Failure load	Displacement	
SLICE X	5.10	8.41	27.10
SLICE Y	5.10	8.41	31.88
SLICE Z	5.10	8.41	30.45

450
 451 It is not surprising that the runtimes differ when the way a model is decomposed is changed. Each
 452 processor, apart from the communication between them, also has to perform specific operations. If more
 453 particles or computationally more expensive elements (due to material non-linearities or element
 454 formulations) end up in one processor, it has to work harder. At the same time the remaining processors
 455 have to wait till the end of each time cycle. There are a lot of different factors to consider when dividing
 456 the problem into domains and it is challenging to determine, before the simulation is run, what the
 457 optimal decomposition would look like since the deformation will affect the computational cost. Thus,
 458 although the MPP takes advantage of better scaling and parallel capabilities through domain
 459 decomposition and MPI, it is important to ensure that each processor has a good load balance between
 460 them such that they are assigned at least several thousands of particles. Otherwise, the benefits of using
 461 the MPP version will be quickly overshadowed by the communication requirements.

462

463 TIME SCALING

464 Time scaling is a technique that is used to reduce the simulation time by applying the given load more
 465 quickly than in a quasi-static experiment. The study is applicable as long as the inertial effects are low and
 466 the material behaviour is only slightly influenced by the corresponding strain rates. Two analyses, with
 467 different time scaling factors of 2.5 and 5.0, were carried out to study its effect on the failure output.

468 For simulating the bending experiments, velocity boundary conditions were applied to the lower
 469 supports. The velocities of the lower supports for each test were calculated from full-scale test data. For
 470 time scaling, this applied velocity was artificially increased by factors of 2.5 and 5.0. The velocity profile
 471 was defined by a half sine function curve that was gradually ramped up to the final scaled velocity. The
 472 minimum ramp up time was set around 1.5-2.0 times the average period of the system to avoid inertial

473 effects that could change the response of the system (Hallquist 2006). The study was first conducted for a
474 linear elastic case for different time scaling factors of 2.5 and 5.0. The Normalised Root Mean Square
475 Error (NRMSE) was determined between the simulated forces and theory. For time scaling factors of 2.5
476 and 5.0, the NRMSE between the computed forces and analytical solution was 3.23% and 6.17%
477 respectively for linear elastic bending. The error percentage being within reasonable limits (below 10%
478 error is considered to be acceptable here), the study was subsequently applied to simulate four-point
479 bending failure of ice.

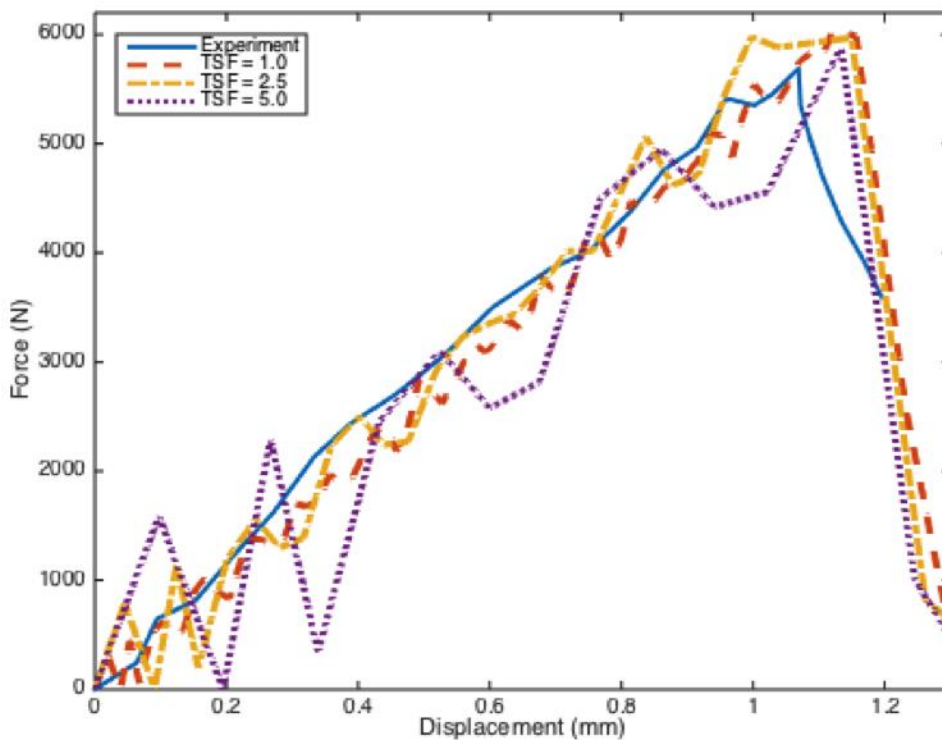
480 As shown in Table 6, using the time scaling technique significantly reduced the computation time. For
481 the analysis with a time scaling factor (TSF) of 2.5, the deviation in the load and displacement at the onset
482 of failure, compared to experimental results, were 4.39 and 7.48% respectively. At the same time the
483 CPU cost was reduced by 59%. The computation time was further reduced when the time scaling factor
484 (TSF) was increased to 5.0, and the corresponding difference in force and displacement outputs at failure
485 with experiments were 2.64% and 5.61% respectively. However, inertia effects slowly came into play as
486 evident from the abrupt spikes in the load-deformation curve for $TSF = 5.0$ (Figure 7). The abrupt
487 changes or fluctuations in the reaction forces (Figure 7) were caused because of the increasing loading
488 velocity, which caused higher reaction forces and possible penetrations at the contact interface between
489 the indenter (lower support) and the SPH ice beam, due to faster movement of the lower
490 supports/indenters. In this study, a penalty-based contact was defined between the supports and the
491 beam. For such contact definition, if a penetration was found (in this case the lower support penetrated
492 into the ice beam because of the higher velocity), a force directly proportional to the penetration depth
493 was applied to resist and eventually eliminate the penetration. Such elimination often caused oscillations
494 in contact responses, and is primarily responsible for the fluctuations in the load-deformation curve. A
495 damping coefficient can reduce such oscillations from penalty contacts. However, contact damping was
496 not used in the present study to ensure that the contact definition for all the numerical experiments
497 presented here was consistent. It can also be seen in Table 6 that the deviations in the output at failure
498 decreased, when compared to experiments, as the time scaling factor was increased. That said, with the
499 increase in time scaling, inertial effects caused higher reaction forces and stresses (pressure) within the
500 beam. Since a pressure-based failure criterion was used, the time-scaled models ($TSF = 2.5$ and $TSF =$
501 5.0) failed earlier and at a lower load and displacement levels than the SPH model with no time scaling
502 ($TSF = 1$). Thus, as the SPH model with $TSF = 1$ gave higher failure results (about 5%) compared to
503 experiments, the failure outputs for the time-scaled models appear to be more accurate and closer to the

504 experimental investigations in Table 6. A better basis of comparison for the time scaling study would
 505 have been the reference model of SPH-40 with no time scaling (TSF = 1), which would have showed that
 506 the accuracy decreased with the increase in TSF. However, this was not done to be consistent with the
 507 other results presented in the manuscript.

508 Table 6: Comparison of results for different time scaling factors

Time scaling factor (TSF)	Deviations in output at failure (%)		CPU time (hrs)
	Failure load	Displacement	
TSF = 1.0	5.10	8.41	27.10
TSF = 2.5	4.39	7.48	11.24
TSF = 5.0	2.64	5.61	5.080

509



510

511 Figure 7: Force versus deflection plots for different values of process time scaling

512 Considering the computational demand of an SPH simulation and the improvement offered by
 513 increasing the loading velocity, invoking time scaling was an efficient way to determine the results within a
 514 reasonable timeframe. However, in a quasi-static experiment the load is applied very slowly such that the
 515 structure deforms at a very low strain rate, and consequently the inertial forces can be ignored.
 516 Therefore, if time scaling was used for a quasi-static case, it was important to check that the inertial effects
 517 were minimal. To ensure this, the total kinetic energy had to be small compared to the total internal
 518 energy of the system, during the simulation (for the models presented here, the total kinetic energy was
 519 about 1% of the total internal energy, at the onset of failure). Also, the force-displacement response had

520 to be independent of the applied velocity. Based on the results from the numerical experiments, a time-
521 scaling factor of 2.5 was recommended to adequately simulate the four-point bending failure of an ice
522 beam without any significant instability and inertial effects.

523

524 MASS SCALING

525 Mass scaling refers to a technique of reducing the computational cost in an explicit analysis by adding a
526 non-physical mass to the structure that increases the critical time step. In standard FEM, mass scaling is
527 often applied to reduce the computational cost for explicit quasi-static simulations Prior (1994). In such
528 an analysis, the time step is calculated internally not only to maintain numerical stability but also to
529 ensure that the stress waves cannot propagate more than the minimum element dimension in one time
530 step. This is referred to as the Courant condition and is expressed in (29)

$$\delta t \leq C_{CFL} \left[\frac{h}{c} \right] \quad (29)$$

531 The time step (δt) is a function of the particle spacing (h), sound speed (c), and a stability constant C_{CFL}
532 (Hallquist 2006). The sound speed (c) depends on the element type and in its simplest form can be
533 written as $c = (E/\rho)^{0.5}$, where E is the Young's Modulus and ρ is the material density. The internally
534 calculated time step for SPH method is slightly modified and takes into account the particle velocity (v),
535 as reflected in (30).

$$\delta t = C_{CFL} \text{Min} \left\{ \frac{h(i)}{c(i) + v(i)} \right\} \quad (30)$$

536 Amongst all the factors controlling the internally calculated time step in an SPH simulation, there are
537 three parameters that can be changed to improve the time step. Firstly, it follows directly from (30) that
538 increasing the mesh size or the element length (h) will increase the time step. This should affect the
539 accuracy, and a convergence study must be conducted prior to determine the optimal mesh size for a
540 given problem. Secondly, Young's Modulus is changed thus directly altering the stiffness of the model
541 and thereby affecting the accuracy. Thirdly, the density of certain critical elements, responsible for
542 bringing down the time step in a simulation, is increased. This is also referred to as mass scaling where a
543 minimum time step size is specified. If an element or particle is responsible for bringing down the time
544 step below the specified minimum value, its density is increased to meet the minimum time step
545 requirements. In this study mass scaling was applied with the expectation to increase the critical time step
546 and consequently reduce the run time

547 An accurate SPH solution requires uniform distribution of particles Lacombe (2001). Naturally, for such a
548 uniform discretization there is no specific particle controlling the time step. Villumsen et al. (2008)
549 indirectly invoked mass scaling of SPH particles in a metal-cutting simulation by directly increasing the
550 density of particles to examine the effect on the final result. In this study, mass scaling was similarly
551 introduced by scaling the density of each particle by ratios of 1.05 and 1.10. Similar to the previous
552 numerical experiment on time scaling, mass scaling was first applied to an elastic four-point bending case.
553 The NRMSE from the analytical solution was 6.06% and 12.71% for mass scaling factors of 1.05 and
554 1.10 respectively. At the same time, the computation time was reduced by 8.34% and 15.24%
555 respectively. The simulations were subsequently extended to the four-point bending failure analysis. The
556 deviations of the results from experiments are presented in Table 7. It can be seen that for a mass scaling
557 factor of 1.05 the force and displacement outputs at failure deviated by 4.22% and 1.87% respectively,
558 and calculation time was reduced by 9%. From the results, it is evident that further increasing the mass
559 scaling factor would reduce both the CPU requirements and the accuracy. Considering a below 10%
560 deviation in output as satisfactory, a mass scaling factor of 1.05 can be adequately used to simulate the
561 overall behavior of the numerical model within a reasonable timeframe. In a quasi-static analysis, mass
562 scaling has been similarly used to increase the critical time step and consequently reduce the run time
563 (Olovsson and Simonsson 2006) Still, the kinetic energy of the system was required to be monitored
564 regularly to ensure that it remained quite small compared to the internal energy, and that inertial effects
565 did not come into play.

566 Table 7: Comparison of percentage difference in results for different values of mass scaling

Mass Scaling Factor (MSF)	Deviations in output at failure (%)		CPU time (hrs)
	Failure load	Displacement	
MSF = 1.00	5.10	8.41	27.10
MSF = 1.05	4.22	1.87	24.84
MSF = 1.10	9.84	2.80	22.97

567

568 SPH-FEM COUPLING

569 The computational time in an SPH simulation primarily depends on the number of discrete particles
570 present in the model. Thus, a viable alternative to reduce the simulation time was to reduce the number
571 of active SPH particles in a simulation. This was achieved by coupling the Lagrangian methods of SPH
572 and FEM. SPH-FEM coupling could be done either by using the adaptive transformation of failed solid
573 elements into SPH particles or by using an offset tied contact to connect the SPH and FEM parts. In
574 adaptive transformation, the new particles assumed the properties of the original finite elements and

575 similar to using a tied contact improved the computational times by reducing the number of active SPH
576 particles in the model.

577 However, it was important to determine the most appropriate SPH and FEM models to couple together
578 and the discretisation error had to be calculated a priori, both for the FEM and SPH parts. The
579 appropriate models were selected from the results of a convergence study conducted for both SPH and
580 FEM beams under four-point bending. The convergence study for SPH was already presented in Das
581 (2017). Therein it was concluded that the height of the SPH beam if discretized using 40 particles (SPH-
582 40) and consequently assigning this inter-particle distance (to ensure uniform particle distribution) for the
583 entire model, the Normalized Root Mean Square Error (NRMSE) was found to be less than 4%. A
584 similar convergence study for FEM indicated that the NRMSE for FEM-20 (the height of the beam was
585 discretized using 20 elements) was around 2.4%, and further reducing the element size to FEM-40 only
586 reduced the error to 2.2%. However, this small improvement in accuracy of FEM-40, compared to FEM-
587 20, came at a significant increase in computational requirements (Table 8). Thus, considering both the
588 CPU resource requirements and accuracy, FEM-20 and SPH-40 were considered to be the most
589 appropriate models to couple together.

590 Table 8: Comparison of NRMSE and CPU costs for different FEM models

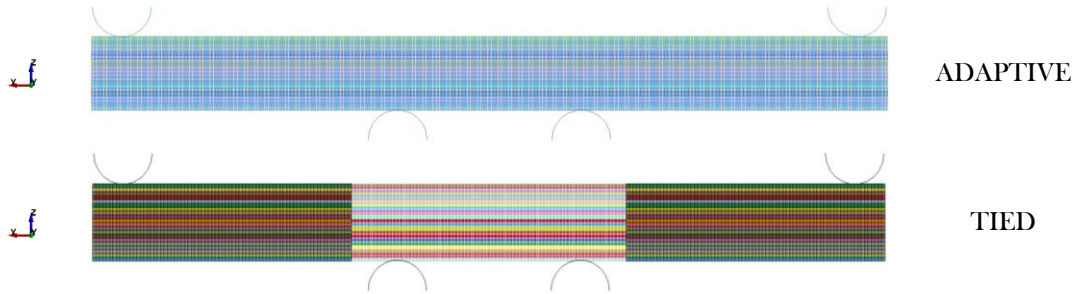
METHOD	Output at t = 0.4 s		NRMSE (%)	CPU time (hrs)
	Load (KN)	Displacement (mm)		
FEM-5	5.25	1.14	16.6	0.04
FEM-10	6.68	1.33	3.40	0.10
FEM-20	6.81	1.34	2.40	1.75
FEM-40	6.90	1.35	2.20	14.0

591 FEM-X indicates that the height of the beam is discretised using "X" number of elements. For instance, FEM-20 denotes that the
592 height of the beam is discretized using 20 elements
593

594 The coupled SPH-FEM methods were first validated for linear elastic four-point bending, by comparing
595 the contact force histories and deflection results with analytical solution and then extended to simulate
596 four-point bending failure of an ice beam. Two different coupled models are discussed in this article and
597 shown in Figure 8. The adaptive model had 11600 elements modelled using eight-noded solid elements
598 and 742400 SPH nodes that are shown in blue colour. Each FEM element was replaced by two SPH
599 particles at the onset of failure, to keep the inter-particle distance in the adaptive model consistent with
600 the pure SPH model. For the tied model, SPH was used for the middle part of the beam where fracture
601 was likely to occur and FEM for the rest of the beam. An offset contact was used to tie the FEM and
602 SPH parts. Such a contact allowed the SPH particles to be tied to the end point of a vector, offset at a
603 specified distance from the master segment of FEM.

604 In Figure 8, both the adaptive model and tied SPH-FEM models are shown. The SPH-FEM models
 605 were first validated for a linear elastic bending case that ensured the applicability of the coupled models
 606 to simulate bending. Thereafter, the SPH-FEM models were applied to simulate the flexural behaviour
 607 of ice in four-point bending.

608



609

Figure 8: SPH-FEM coupling models

610

Validation of SPH-FEM coupling techniques to simulate linear elastic bending

611

The linear elastic bending results from the coupled SPH-FEM models were compared with FEM and a
 612 pure SPH simulation of the same beam. For FEM analysis, the beam was modeled using the default
 613 eight-noded solid element and the outcome from FEM was first matched with the analytical solution of
 614 beam deflection under linear elastic bending. NRMSE further quantified the accuracy between the
 615 numerical results and FEM. Although this normalised error percentage was least for the SPH method,
 616 the adaptive model was the most efficient computationally (Table 9). Also, the NRMSE for the adaptive
 617 model was only 0.47% more than the SPH method. This decrease was quite acceptable, especially
 618 considering the significant improvement in efficiency.

619

Table 9: Comparison of linear elastic bending results between SPH and SPH-FEM models

METHOD	Output at t=0.4 s		NRMSE (%)	CPU time (hrs)
	Load (KN)	Displacement (mm)		
SPH-40	7.03	1.369	2.41	22.08
TIED	6.38	1.376	4.76	13.90
ADAPTIVE	6.79	1.354	2.88	2.170

620

621

Furthermore, the comparison of deflection (Figure 9) and load versus deformation plots (Figure 10)
 622 between the coupled models, SPH and FEM, show good correlation between the different approaches
 623 thereby contributing to the validation of the coupled models to simulate elastic bending behavior of ice.

624



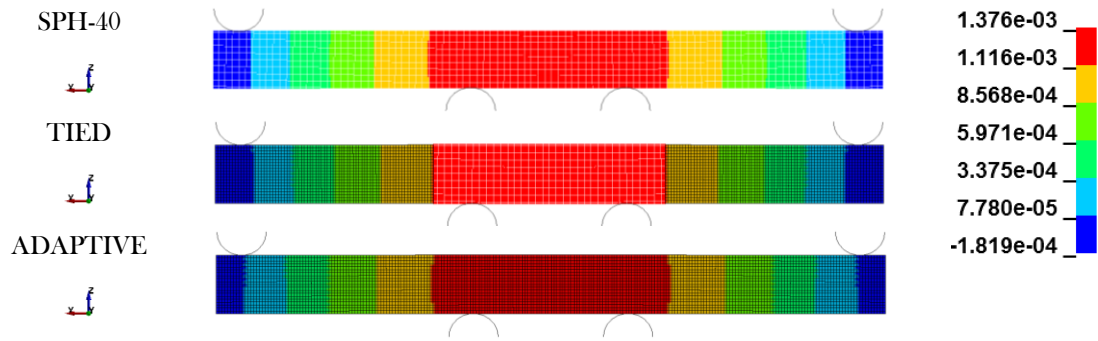


Figure 9: Comparison of deflection at $t = 0.40$ s

625
626

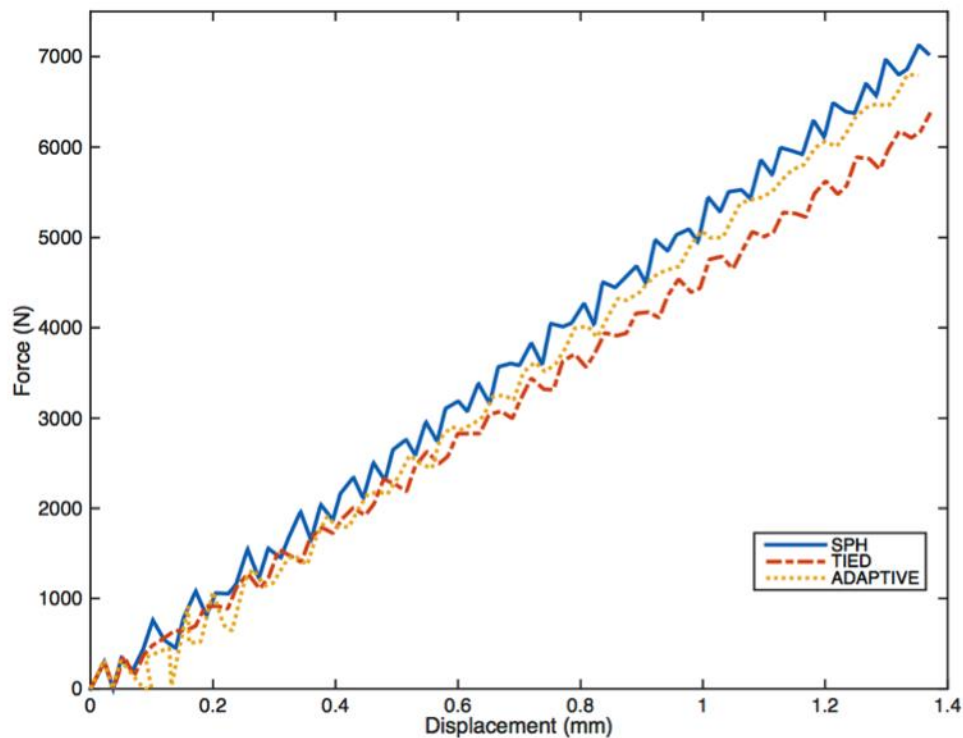


Figure 10: Comparison of force-displacement plots between SPH and SPH-FEM models

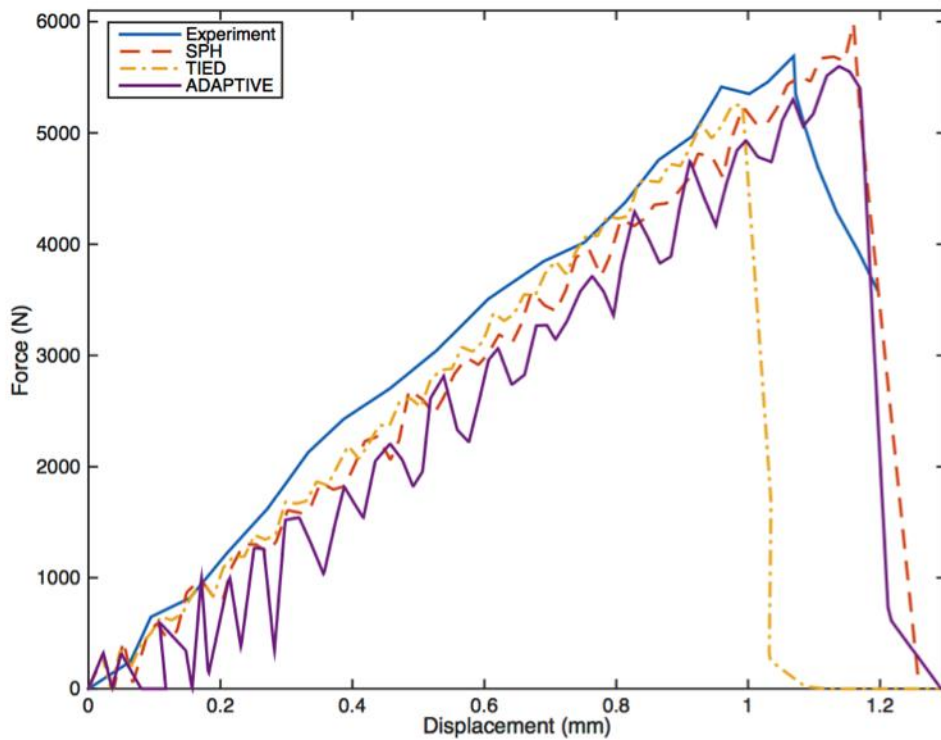
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628

629

630 *Application of SPH-FEM coupling to simulate four-point bending failure of ice*

631 The different coupling models were then used to simulate flexural failure of an ice beam under four-
 632 point bending. The outcomes from the numerical experiments are compared in Table 10. Both the
 633 adaptive and the SPH models agreed well with experimental results, particularly with respect to failure
 634 load, deflection and failure time. The force-displacement plots are compared in Figure 11. It can be
 635 observed that the tied contact model reached the tensile failure stress earlier than other numerical
 636 models. This is understandable because a tied contact connects two different domains and a discontinuity
 637 exists at the SPH-FEM boundary. The response in a tied contact was quite accurate initially. As the

638 deformation progressed, the stiffness, displacement, pressure and other key functionals started to be
 639 different in the SPH and FEM parts leading to a dissimilar distribution around the tied interface. Prior to
 640 failure, the distribution of tensile pressure in a few SPH elements at the tied interface was about twice the
 641 surrounding solid elements and consequently led to premature failure there. Hence, a suitable solution
 642 was to use the adaptive transformation of failed finite elements into SPH particles. For adaptive coupling,
 643 such premature failure was not observed, and the failure results were found to be in good agreement with
 644 experimental observations.



645
 646 Figure 11: Comparison of force-displacement plots for SPH and SPH-FEM models

647 Table 10: Comparison of numerical results with experiment

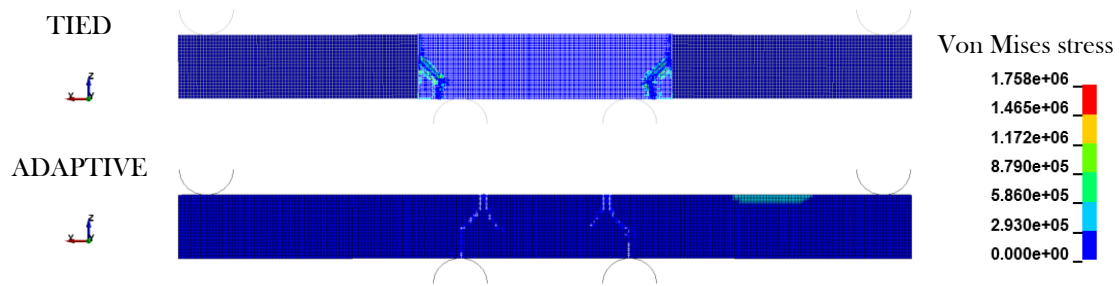
METHOD	Deviations in output at failure (%)		Failure time (sec)	CPU time (hrs)
	Failure load	Displacement		
SPH-40	5.10	8.41	0.34	27.10
TIED	7.73	0.93	0.32	13.77
ADAPTIVE	2.46	8.41	0.34	4.730

648 1 (Kajala et al. 1999)

649 The time requirement for the bending simulation can be substantially reduced if SPH-FEM coupling is
 650 used instead of a pure SPH simulation. In fact, it can be seen from Table 10 that the tied model was two
 651 times faster than SPH-40. However, variations in pressure distribution at the SPH-FEM boundary
 652 initiated failure around the tied interface (Figure 12). The adaptive coupling model was almost six times

653 faster than a pure SPH simulation and the simulated failure load, time and displacement agreed well with
 654 published experimental data.

655



656 Figure 12: Breaking location and fracture pattern for SPH-FEM models

657

658 Table 11 reviews the different techniques studied in this paper and quantifies the reduction in
 659 computational resource requirement and the associated error percentage compared to a pure SPH
 660 model. The adaptive transformation of failed solid elements into SPH particles was computationally
 661 most efficient. A combination of the described techniques can also be investigated to study the feasibility
 662 of using them to improve the efficiency even more.

663 Table 11: Summary of results from domain decomposition, scaling techniques and SPH-FEM models

METHODS		Deviations in output at failure (%)		CPU time (hrs)
		Failure load	Failure displacement	
Domain Decomposition	SLICE X	5.10	8.41	27.10
	SLICE Y	5.10	8.41	31.88
	SLICE Z	5.10	8.41	30.45
Time scaling	TSF = 2.5	4.39	7.48	11.24
	TSF = 5.0	2.64	5.61	5.080
Mass scaling	MSF = 1.05	4.22	1.87	24.84
	MSF = 1.10	9.84	2.80	22.97
SPH-FEM	TIED	7.73	0.93	13.77
	ADAPTIVE	2.46	8.41	4.730

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668 **CONCLUSION**

669 The SPH method, since its inception to simulate astrophysical phenomena, has gradually developed and
670 diversified to become one of the best numerical methods to simulate fracture in brittle solids (Monaghan
671 2005). However, the computational requirements associated with SPH are significantly higher than
672 conventional grid-based methods. Thus, several key ways to improve the computation time were
673 investigated and their corresponding accuracy, ability to simulate failure, and limitations were compared.
674 Of the key techniques investigated, domain decomposition, time scaling and mass scaling were found to
675 be significantly useful to compute the results within a desired time limit. For quasi-static cases, these
676 methods could be effectively used to reduce the CPU requirements by at least 59% for time scaling and
677 8% for mass scaling. At the same time, the corresponding deviations in output were less than 5% for
678 failure load and 7.5% for failure displacement, when compared to experiments. These methods could
679 provide valuable insights into the overall behaviour of a simulation as long as the kinetic energy was
680 constantly monitored to ensure that inertial effects were negligible. The SPH-FEM models were also
681 identified as viable alternative to significantly reduce the computational demand. If a tied contact was
682 used to couple the SPH and FEM parts, the CPU requirements were reduced by almost 50%. However,
683 the breaking location could not be simulated accurately in tied contacts as the discontinuity between the
684 SPH and FEM parts resulted in premature failure at the contact interface. A suitable solution was to use
685 adaptive transformation of failed finite elements into SPH particles. These new particles assumed the
686 properties of the finite elements at failure, and not only improved the computational time by 83% but
687 also the overall accuracy by 3%. The satisfactory correspondence of the simulation results with the
688 experimental outcomes, especially with respect to failure load and displacement, confirm the accuracy
689 and credibility of the different approaches presented in this paper. Future research should explore the
690 application of these approaches in the context of combined bending and crushing failure of ice during
691 ice-structure interactions.

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