Finite Element Analysis of Thermoelastic Fiber-Reinforced Anisotropic Hollow Cylinder with Dual-Phase-Lag Model

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In the present paper, we have constructed the equations for generalized thermoelasticity of a fiber-reinforced anisotropic hollow cylinder. The formulation is applied in the context of dualphase-lag model. An application of hollow cylinder is investigated for the outer surface is traction free and thermally isolated, while the inner surface is traction free and subjected to thermal shock. The problem is solved numerically using a finite element method. The results of displacement, temperature and radial and hoop stress are obtained and then presented graphically. Finally, the comparisons are made between the results predicted by the coupled theory, Lord and Shulman theory and dual-phase-lag model in presence and absence of reinforcement.

Introduction. Materials such as resins reinforced by strong aligned fibers exhibit highly anisotropic elastic behavior in the sense that their elastic moduli for extension in the fiber direction are frequently of the order of 50 or more times greater than their elastic moduli in transverse extension or in shear. Due to their low weight and high strength, the fiber-reinforced composites are used in a variety of structures. The mechanical behavior of many fiber-reinforced composite materials is adequately modeled by the theory of linear elasticity for transversely isotropic materials, with the preferred direction coinciding with the fiber direction. The theory of strongly anisotropic materials has been widely discussed in the literature, Belfield et al. [1] investigated the stress in plates reinforced by fibers lying in concentric circles. Hashin and Rosen [2] studied the elastic moduli for fiber-reinforced materials.

The first of such modeling is the extended thermoelasticity theory of Lord and Shulman [3], who introduced the concept of thermal relaxation time into the classical Fourier law of heat conduction. Subsequently, modifying the stress versus strain relationship as well as the entropy relationship with relaxation time, Green and Lindsay [4] proposed the temperature ratedependent thermoelasticity (GL) theory. The theory was extended for anisotropic body by Dhaliwal and Sherief [5]. Tzou [6, 7] proposed the dualphase-lag (DPL) model, which describes the interactions between phonons and electrons on the microscopic level as retarding sources causing a delayed response on the macroscopic scale. The DPL model proposed by Tzou [8] is such a modification of the classical thermoelastic model in which the Fourier law is replaced by an approximation to a modified Fourier law with two different time translations: a phase-lag of the heat flux t_q and a phase-lag of temperature gradient t_{θ} . Abouelregal [9] studied a problem of a semi-infinite medium subjected to exponential heating using a dual-phase-lag thermoelastic model. Verma [10] studied the shear waves in self-reinforced bodies. Singh [11] discussed the wave propagation in thermally conducting linear fiber-reinforced composite materials with one relaxation time. Othman and Abbas [12] studied the effect of rotation on plane waves at the free surface of a fiber-reinforced thermoelastic halfspace. Abbas [13] investigated the effect of magnetic field on thermoelastic interaction in a fiber-reinforced anisotropic hollow cylinder. Chattopadhyay and Choudhury [14] investigated the propagation, reflection and transmission of magnetoelastic shear waves in a self-reinforced media. Chattopadhyay and Choudhury [15] studied the propagation of magnetoelastic shear waves in an infinite self-reinforced plate. Tian et al. [16], Abbas et. al [17, 18, 19, 20, 21, 22, 23], applied the finite element method in different generalized thermoelastic problems.

In the present paper, we have considered a problem of dual-phase-lag model on generalized thermoelasticity of a fiber-reinforced anisotropic hollow cylinder. The problem has been solved numerically using a finite element method (FEM). Numerical results for the temperature distribution, displacement, radial stress and hoop stress are represented graphically. The results indicate that the different between the coupled theory (CT), Lord and Shulman (LS) theory, and DPL model are very pronounced.

Basic Equations and Formulation of the Problem. For a fiberreinforced linearly thermoelastic anisotropic medium, the constitutive equations preferred to whose direction is that of a unit vector **a** [<u>11</u>]:

(1)

(2)

(3)

The equation of heat conduction under DPL model [<u>9</u>] The equation of motion

(4)

Three cases arise:

(i) classical dynamical coupled theory

(ii) LS theory

(iii) DPL model

where u_i are the displacement vector components, ρ is the mass density, e_{ij} is the strain tensor, T is the temperature change of a material particle, τ_{ij} is the stress tensor, β_{ij} is the thermal elastic coupling tensor, c_e is the specific heat at

constant strain, T_0 is the reference uniform temperature of the body, t_q is a phase-lag of heat flux, t_{θ} is a phase-lag of temperature gradient, K_{ij} is the thermal conductivity, α , β , ($\mu_L - \mu_T$) are reinforced anisotropic elastic parameters, and λ and μ_T are elastic parameters and the component of the vector **a** are (a_1, a_2, a_3) , where a21+a22+a23=1.a12+a22+a32=1. Let us consider a fiber-reinforced hollow cylinder with an external radius b and internal radius a. By using the cylindrical system of coordinates (r, θ, z) with the z-axis lying along the axis of the cylinder. Due to symmetry, the displacement vector has the components (5)For circumferential reinforcement, the equation of motion in the absence of body forces is given by (6)The energy equation without heat sources has the form (7)with (8)(9)where and α_{11} and α_{22} are coefficients of linear thermal expansion. It is convenient to change the preceding equations into the dimensionless forms. To do this, the dimensionless parameters are introduced as (10)From Eqs. (10) into Eqs. (6)–(9) one may obtain (after dropping the superscript ' for convenience) (11)(12)(13)(14)where From preceding description, the initial and boundary conditions may be expressed as (15)(16)where *a* and *b* are inner and outer radii of the hollow cylinder, respectively,

and H(t) is the Heaviside unit step function.

Finite Element Method. The FEM is a powerful technique originally developed for the numerical solution of complex problems in structural mechanics, and it remains the method of choice for complex systems. In this section, the governing equations of generalized thermoelasticity with dual-phase-lag are summarized, using the corresponding finite element equations. In the FEM, the three isoperimetric, quadrilateral element is used for displacement and temperature. Thus, the displacement component *u* and temperature *T* are related to the corresponding nodal values by

where *m* denotes the number of nodes per element, and *N* the shape functions. In the framework of standard Galerkin procedure, the weighting functions and the shape functions coincide:

(18)

(19)

(20)

(21)

hus, Eqs. $(\underline{11})$ and $(\underline{12})$ corresponding to the finite element equations can be written as

where me is the total number of elements. Appendix presented the coefficients of Eq. (21).

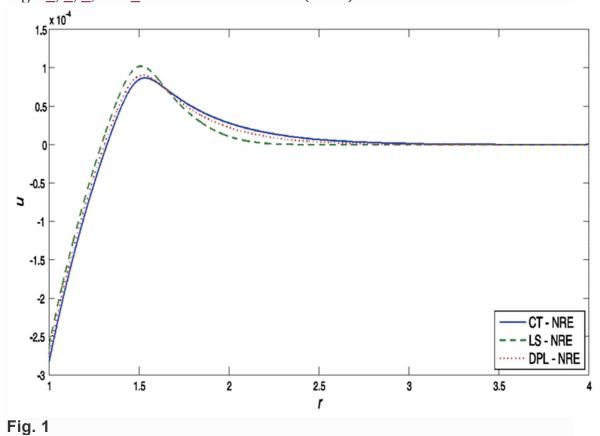
The matrix form of Eq. (<u>21</u>) can be written as Md["]+Cd["]+Kd=Fext,Md["]+Cd["]+Kd=Fext,

(22)

where $d = \begin{bmatrix} u & T \end{bmatrix}^T$, F_{ext} , M, C, and K represent external force vectors, the mass, damping, and stiffness matrices, respectively. Finally, the Newmark time integration method or other methods have to be used to determine the time derivatives of the unknown variables (see [24]).

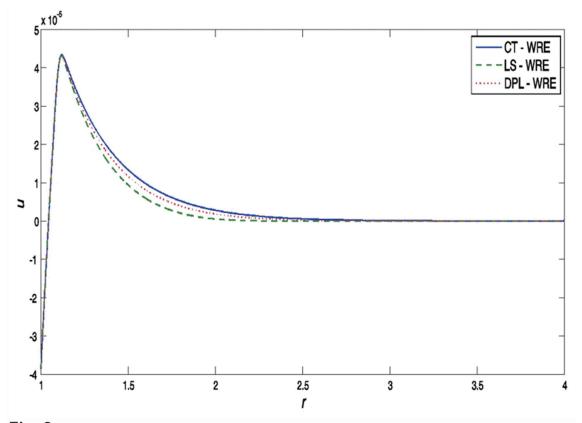
Numerical Example. To study the effect of reinforcement on wave propagation, we use the following physical constants for generalized fiber-reinforced thermoelastic materials [<u>11</u>]: $\rho = 2660 \text{ kg/m}_3$, $\lambda = 5.65 \cdot 10^{10} \cdot \text{N/m}_2$, $\mu_T = 2.46 \cdot 10^{10} \cdot \text{N/m}_2$, $\mu_L = 5.66 \cdot 10^{10} \cdot \text{N/m}_2$, $\alpha = -1.28 \cdot 10^{10} \cdot \text{N/m}_2$, $\beta = 220.9 \cdot 10^{10} \cdot \text{N/m}_2$, $\alpha_{11} = 0.017 \cdot 10^{-4} \text{ deg}^{-1}$, $T_1 = 1$, $\alpha_{22} = 0.015 \cdot 10^{-4} \text{ deg}^{-1}$, $ce = 0.787 \cdot 10^3 \text{ J/(kg \cdot deg)}$, $T_0 = 293 \text{ K}$, $t_q = 0.2$, $t_0 = 0.1$, $K_{11} = 0.0921 \cdot 10^3 \text{ J/(m \cdot s \cdot deg)}$, $K_{22} = 0.0963 \cdot 10^3 \text{ J/(m \cdot s \cdot deg)}$, and t = 0.5.

These physical quantities are represented and plotted in Figs. <u>1</u>–<u>8</u> with respect to radial distance for $T_1 = 1$ and t = 0.5. Furthermore, all the variables and parameters are taken in non-dimensional forms. In Figs. <u>1</u>, <u>3</u>, <u>5</u>,

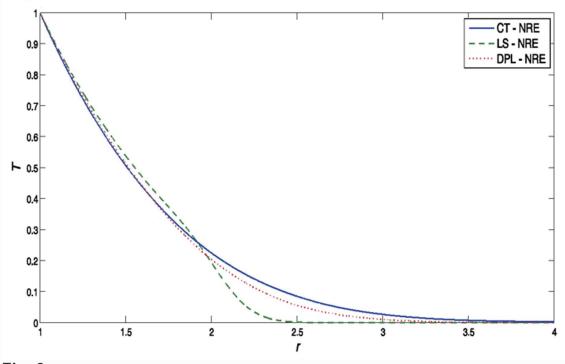


and $\underline{7}$ refer to thermoelastic solid without reinforcement (NRE), while in Figs. $\underline{2}$, $\underline{4}$, $\underline{6}$, and $\underline{8}$ – with reinforcement (WRE).

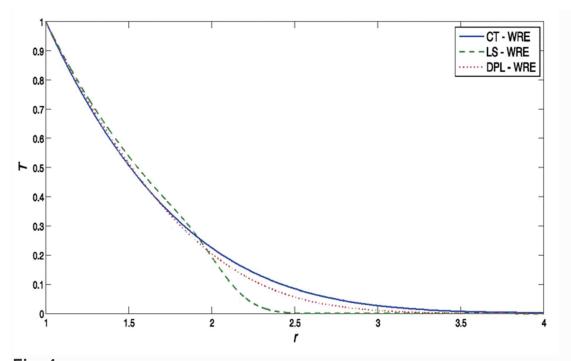
Displacement distribution for different theories without reinforcement.



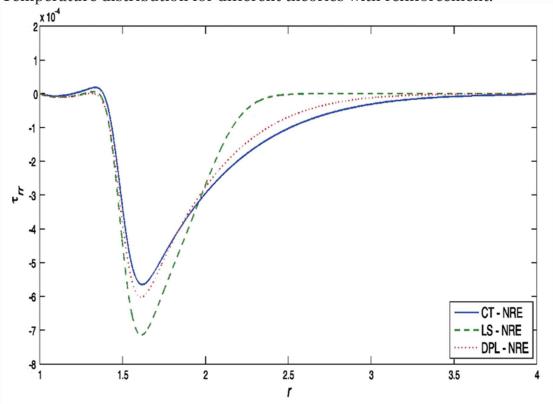




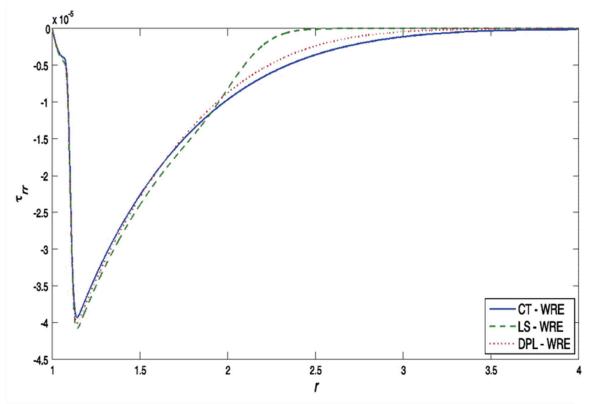




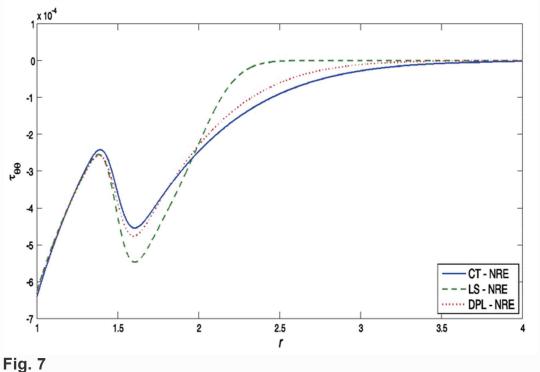














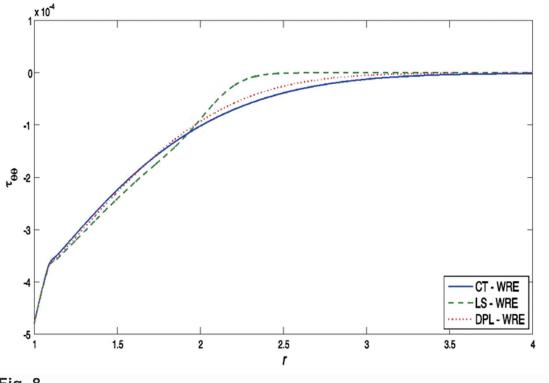


Fig. 8

Hoop stress distribution for different theories with reinforcement.

From Figs. 1-8 is seen that, there is no significant difference in the value of temperature for WRE and NRE as in Figs. <u>3</u> and <u>4</u>. Figures <u>1</u> and <u>2</u> show the variation of displacement for NRE and WRE. It is noticed that the displacement is continuous and the displacement gradually decreases with *r* and is zero at r = b. This is also in agreement with the theoretical result where beyond the thermal wave front displacement vanishes. Figures 5 and 6 represent the variation of stress with respect to distance, which we observed that, the stress, always starts from the zero value and terminates at the zero value to obey the boundary conditions. Figures 7 and 8 gives the variation of hoop stress versus r. Also, for each theory the hoop stress have a maximum magnitude at the boundary. Figures 9, 10, 11, and 12 display the distribution of the displacement, temperature, radial and hoop stresses for a wide range of r ($1 \le r \le 4$) and for a wide range of dimensionless time t ($0 \le t \le 0.5$). The reinforcement has a great effect on the distribution of displacement and stresses. Finally, it is obvious that the phase-lag parameters t_{θ} and t_{q} have very pronounced effect on the temperature and displacement, radial and hoop stresses.

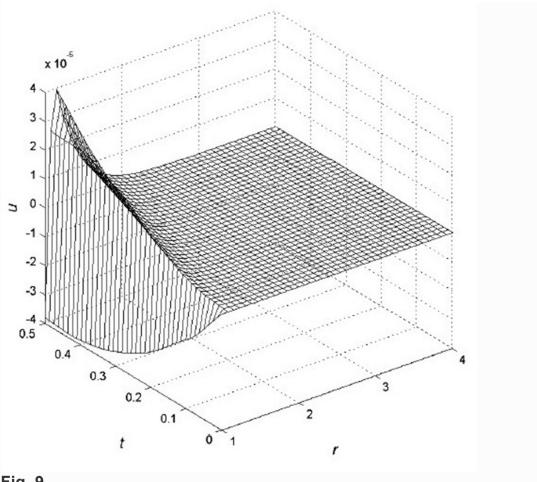


Fig. 9 Displacement distribution with reinforcement.

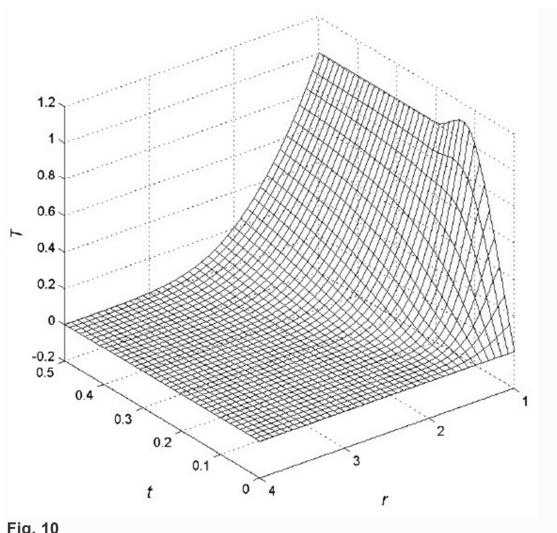


Fig. 10 Temperature distribution with reinforcement.

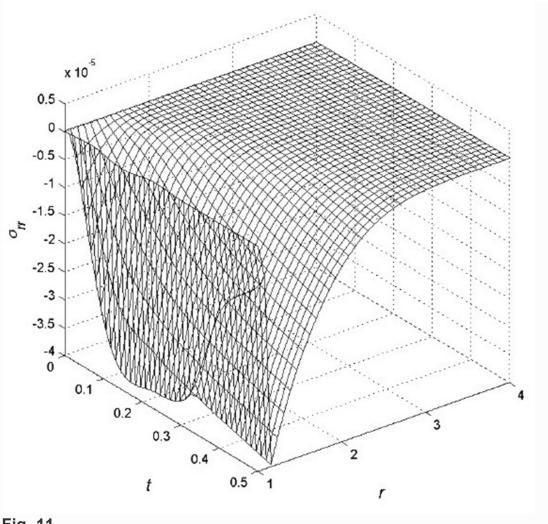


Fig. 11 Radial stress distribution with reinforcement.

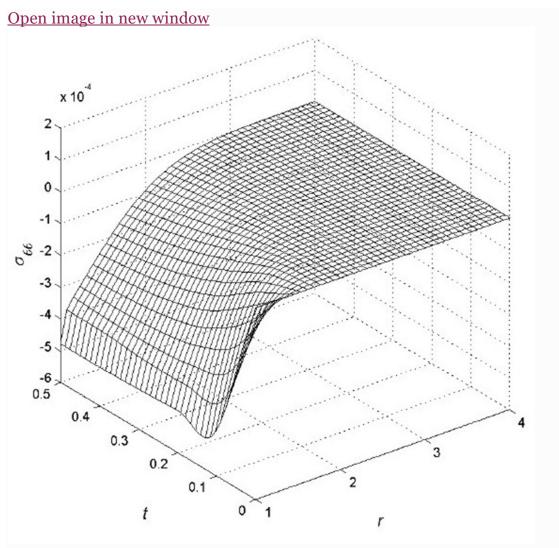


Fig. 12 Hoop stress distribution with reinforcement.