

# Numerical Study on Wave Drift Loading on Slender Marine Structures

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#### MASTER THESIS IN MARINE TECHNOLOGY

Spring 2017

FOR

#### Julie Vadholm

#### Numerical study on wave drift loading on slender marine structures

(Numerisk studie av bølgedriftlaster på slanke marine konstruksjoner)

During the 1990s, some research showed that viscous effects contributed to a substantial increase in wave drift loads in high sea states for slender structures. Recently, the viscous contribution has been suggested as a possible explanation for some accidents with mooring line failures. In those cases, the sea states were within the operational limits for the mooring systems. This suggests that there exists a knowledge gap in connection with the wave loading. To account for viscous drift contributions, the state-of-the-art approach in the industry is to apply Morison equation to model drag loads on slender members. In the wave-drift problem, the most important contributions (both viscous and non-viscous) originate from the part of the structure in the free-surface zone. Here the uncertainties related to choice of drag coefficient is large, both due to nonlinear wave kinematics, local wave-body interactions leading to non-uniform water rise-up along the structure, and possibly due to variations in relevant flow parameters. Recently an empirical formula for the wave drift loads has been proposed within the "EXWAVE JIP" project for semis and other column-based floating platforms.

#### Objective

The present master thesis aims to contribute in enhancing the knowledge on the relevant effects for wave drift loading on slender marine structures. This will be done numerically, performing 2D and 3D simulations of the problem. As preliminary study, a project-thesis work examined relevant literature and identified an open-source CFD solver to carry out numerical investigations of the problem. The solver was verified for wave propagation and wave interaction against a fixed structure, assuming 2D flow conditions. Accuracy and numerical convergence were studied.

The master-thesis work should be carried out in steps as follows:

- 1. Summarize major findings/outcomes from the project thesis.
- 2. Complement the literature study initiated in the project thesis.
- 3. Apply the selected CFD solver to study a 2D problem with incident (regular and bichromatic) waves interacting with a fixed body mimicking a platform column. Examine the loads on the structure and discuss contributions from linear and nonlinear effects as a function of the incident-wave parameters and the body slenderness.
- 4. Compare results from selected cases of step 3 against a potential-flow fully nonlinear solution to assess importance of viscous effects.
- 5. Select relevant incident-wave parameters from the investigations in step 3 and 4 and apply the CFD solver to study a 3D problem with incident waves interacting with a fixed body mimicking a platform column. The body beam-to-draft ratio should be selected from the 2D body geometries used in step 3 and 4. Examine the loads on the structure and discuss importance of 3D effects.
- 6. Compare the drift-load results from the 3D simulations against the predictions from the available empirical formula recently proposed within the "EXWAVE JIP" project.
- 7. Summarize the results and provide suggestions for future research investigations.



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The work may show to be more extensive than anticipated. Some topics may therefore be left out after discussion with the supervisor without any negative influence on the grading.

The candidate should in his report give a personal contribution to the solution of the problem formulated in this text. All assumptions and conclusions must be supported by mathematical models and/or references to physical effects in a logical manner.

The candidate should apply all available sources to find relevant literature and information on the actual problem.

The thesis should be organised in a rational manner to give a clear presentation of the work in terms of exposition of results, assessments, and conclusions. It is important that the text is well written and that tables and figures are used to support the verbal presentation. The thesis should be complete, but still as short as possible. In particular, the text should be brief and to the point, with a clear language. Telegraphic language should be avoided.

The thesis must contain the following elements: the text defining the scope (i.e. this text), preface (outlining project-work steps and acknowledgements), abstract (providing the summary), table of contents, main body of thesis, conclusions with recommendations for further work, list of symbols and acronyms, references and (optional) appendices. All figures, tables and equations shall be numerated.

The supervisor may require that the candidate, in an early stage of the work, present a written plan for the completion of the work. The plan should include budget for the use of computer and laboratory resources that will be charged to the department. Overruns shall be reported to the supervisor.

From the thesis, it should be possible to identify the work carried out by the candidate and what has been found in the available literature. It is important to give references to the original source for theories and experimental results.

Supervisor: Marilena GrecoCo-supervisor: Finn-Christian W. HanssenCo-supervisor: Babak OmmaniDeadline: 28 August 2017

Marilena Greco Supervisor

# Preface

This is a master thesis within marine hydrodynamics concerning a numerical study on the wave drift loading on slender marine structures as a part of the study program Marine Technology at the Norwegian University of Science and Technology in Trondheim. The work has mainly been carried out from April to August 2017. The work is a continuation of the preliminary project study carried out from January to March 2017. The work has been performed in cooperation with Moss Maritime who initially suggested the topic and emphasized the importance of research within this particular area. Moss Maritime has provided a co-supervisor who has supported me throughout the work with this master thesis.

The master thesis covers a brief introduction to the relevant theory related to the problem, a description of the current methods used to calculate and predict the wave drift forces, the methodology behind the numerical study by use of openFOAM, a summary of the results, discussion of the main findings and recommendations for further work. A considerable amount of time has been used to modify the numerical wave tank in OpenFOAM developed during the project phase to include a damping area, by coupling the existing cases to the toolbox waves2foam. In addition, performing computational analyses with openFOAM is of course time consuming.

Trondheim, 2017-07-30

Julie Vadholm

# Acknowledgement

I would like to thank my supervisor, Marilena Greco, for the support while working with the master thesis and for discussing and answering all my questions during our weekely meetings. I would also like to thank my co-supervisors Babak Ommani from SINTEF Ocean and NTNU and Finn-Christian W. Hanssen from Moss Maritime. I would like to thank Babak for all the hours he has used to help me moving forward with OpenFOAM, guide me when I have had trouble with the simulations and answered all my questions. I would like to thank Finn-Christian for always being available and answering my questions regarding the relevant theory behind the thesis, providing me with relevant literature and post processing methods and valuable discussions concerning the results. I would also like to thank Moss Maritime for being positive and helpful and proposing a master topic of importance for the maritime industry. I would also like to thank Carl Trygve Stansberg for answering my questions regarding the new empirical formula proposed by MARINTEK and helping me find additional literature. At last I would like to thank Kjell Larsen and Statoil for providing me with some up-to-date model test results.

J.V.

# Summary

This master thesis presents the first step in the investigation of wave drift forces acting on slender structures through numerical simulations. Basic theory is presented along with an overview of how the problem is treated in the industry and a description of the most commonly used methods. The numerical study of how the wave drift forces act on a 2D geometry is performed using the computational fluid dynamics (CFD) program OpenFOAM. The effect of viscosity is also investigated. This is an in-depth study focusing on the methodology behind the problem and presents the initial part of the comprehensive study of wave drift forces on semi-submersible platforms. The study is a continuation of the preliminary project performed from January to March 2017.

Recent mooring line failures in the North Sea occurring at lower sea states than the design sea states indicate the need for an investigation of the existing load prediction methods. A full understanding of the problem is not yet obtained, even though many studies have been carried out during the recent years. Several studies suggest that the wave drift forces are higher than initially anticipated, especially the viscous contribution. Wave drift forces are of higher order and can be difficult to calculate accurately. Potential flow theory is often applied together with simplified viscous theory. However, these methods have shown to under-predict the drift forces. The viscous model is often based on the drag term of Morison's equation. The drag coefficient is case dependent and difficult to predict accurately without the use of model tests. Predictions of wave drift forces using model tests, CFD analysis and development of simplified empirical formulas are still an active research field with the purpose of understanding the underlying physics properly.

The analysis using OpenFOAM is performed for a 2D column of quadratic cross section mimicking one of the columns of a semi-submersible platform. The column is fixed in space, meaning that

the diffraction problem is investigated. The column is subjected to both regular and bi-chromatic waves with wave height 4.5, 6 and 7 m. The wave conditions have been chosen based on parameters presented in relevant literature. The ratio between wave length to body size is always such that the wave classifies as a long wave. The waves are generated in two ways; by a numerical wavemaker and by using the built-in feature of the toolbox waves2Foam, both based on first-order wave theory. An initial convergence study is performed in order to investigate the correct time step to be used. The choices made during the modelling and simulations are discussed along with the presented results. Selected inviscid cases have also been simulated in order to investigate the effect of viscosity.

The presented results show that the expected dominating frequencies are present in the spectra of the forces and wave elevation. For the bi-chromatic wave cases, the mean drift force in surge is proportional to more than the wave amplitude squared. The slow drift forces are also clearly increasing for higher wave amplitudes. Furthermore, the results show that the viscous contribution to the wave drift loads is small and that non-viscous flow separation is present at the sharp corners at the bottom of the column. The grid resolution might have been too low in order to fully capture some details of the viscous effects, and a finer mesh could have been applied in order to investigate this further. Future investigation of this problem would benefit from building up a 3D case to study the importance of three-dimensional effects with an enhanced grid refinement.

# Sammendrag

Denne masteroppgaven studerer bølgedriftskreftene på en slank konstruksjon gjennom numeriske simuleringer. Grunnleggende teori og en oversikt over hvordan problemet er behandlet i industrien og hvilke metoder som er mest brukt blir presentert. Det numeriske studiet av bølgedriftskrefter på en 2D geometri er gjennomført ved bruk av CFD-programmet OpenFOAM. Effekten av viskositet har også blitt undersøkt. Studiet har fokusert på metodikken bak problemet og presenterer det første steget i det omfattende studiet av bølgedriftskrefter på halvt nedsenkbare plattformer. Studiet bygger på resultatene fra det innledende prosjektet som ble gjennomført fra januar til mars 2017.

De siste årene har det oppstått flere brudd på ankerliner i Nordsjøen, selv ved lavere laster enn det de er designet for. Dette indikerer at det er et behov for å undersøke de gjeldende lastberegningsmetodene. En fullstendig forståelse av problemet er ennå ikke oppnådd, på tross av at det har blitt gjennomført mange studier på dette området de siste årene. De fleste studiene bemerker at bølgedriftskreftene er større enn tidligere antatt, særlig det viskøse bidraget. Bølgedriftskrefter er høyere ordens krefter og kan være vanskelig å beregne nøyaktig. Ofte er forenklet potensialteori brukt sammen med enkle viskøse metoder som er basert på Morisons likning. Drag-koeffisienten varierer ut fra tilfelle og er vanskelig å predikere uten sammenlikning opp mot modelltester. Prediksjon av bølgedriftskrefter ved hjelp av modelltester, CFD analyser og utvikling av empiriske formler er fortsatt interessante forskingsfelt i arbeidet mot å forstå den bakenforliggende fysikken korrekt.

Den numeriske analysen ved hjelp av OpenFOAM er utført for en 2D cylinder med kvadratisk tverrsnitt, som skal etterlikne en av søylene i en halvt nedsenkbar plattform. Søylen er fastholdt og diffraksjonsproblemet er studert. Søylen har blitt utsatt for både regulære og bikromatiske bølger med høyde 4.5, 6 og 7.5 m. Bølgeparametrene er valgt på grunnlag av det som er presentert i

relevant faglitteratur. Forholdet bølgelengde over søylebredde er valgt slik at bølgene er klassifisert som lange. Bølgene er generert på to forskjellige måter; med en numerisk bølgegenerator og ved å bruke den innebygde metoden i verktøyskassen waves2Foam. Begge er basert på førsteordens bølgeteori. En enkel konvergensstudie er gjennomført for å bestemme hvilket tidssteg som skal brukes i simuleringen. Noen simuleringer har også blitt gjennomført uten viskositet.

De presenterte resultatene viser at de forventede dominante frekvensene er synlige i kraft- og bølgehevingsspekteret. Koeffisientene til den gjennomsnittlige driftskraften i jag øker med mer enn kvadratet av bølgeamplituden, det samme gjelder for den saktevarierende kraften. Resultatene viser videre at det viskøse bidraget er lite og at ikke-viskøs separasjon oppstår ved søylens hjørner. Elementstørrelsen kan ha vært for stor til å kunne kartlegge de viskøse effektene korrekt og elementstørrelsen bør minkes for å studere dette videre. Videre arbeid med dette problemet vil dra nytte av å gå videre til 3D for å undersøke kreftene og effekten av viskositet nøyere, og å minke elementstørrelsen i et større område under søylen.

# Nomenclature

## Acronyms

2D Two-dimensional
3D Three-dimensional
3D Three-dimensional
CFD Computational Fluid Dynamics
QTF Quadratic Transfer Function
KC Keulegan-Carpenter number
BC Boundary Condition
FFT Fast Fourier Transform
DOF Degree Of Freedom
TLP Tension Leg Platform
JIP Joint Industry Project
VOF Volume Of Fluid

## **Symbols**

- B Bredth [m]
- L Length [m]
- $\omega$  Rotational frequency [rad/s]
- f Frequency [1/s]
- T Wave period [s]
- $\lambda$  Wave length [m]
- $C_D$  Drag coefficient [-]
- p Pressure [Pa]
- $\phi$  Velocity potential  $[m^2/s]$
- $\rho$  Density  $[kg/m^3]$
- *A<sub>j</sub>* Wave amplitude of wave component j [m]
- *A<sub>R</sub>* Reflected wave amplitude [m]
- *A<sub>T</sub>* Transmitted wave amplitude [m]
- $\zeta_a$  Incident wave amplitude [m]
- $g_i$  Gravitational acceleration  $[m/s^2]$
- $\epsilon$  Phase angle [rad]
- $T_{ij}^{is}$  Second order transfer functions connected to sine term
- $T_{ij}^{ic}$  Second order transfer functions connected to cosine term
- $F_i$  Force component i [N]
- $F_D$  Drift force [N]
- $F_V$  Viscous force [N]
- My Moment about y-axis [Nm]
- U Velocity [m/s]
- *u<sub>i</sub>* Velocity vector [m/s]
- $U_M$  Velocity amplitude of oscillatory flow [m/s]
- D Diameter [m]
- A Area  $[m^2]$
- Cu Current [m/s]
- H Wave height [m]

η	Wave elevation [m]
Hs	Significant wave height [m]
Тр	Peak period [s]
k	Wave number [rad/m]
$\mu$	Dynamic viscosity $[kg/(m \cdot s)]$
$\delta_{ij}$	Kronecker delta [-]
x	Displacement in x [m]
z	Displacement in z [m]
S(z)	Wave maker stroke [m]
$S(\omega_j)$	Wave spectrum $[m^2 s]$
Co	Courant number [-]
$e_{ij}$	Strain rate tensor [1/s]
h	Water depth [m]
$T_{oc}$	Oscillation period [s]
w	Relaxation weight [-]
α	Volume fraction [-]
m	exponent [-]
$a_n$	Coefficients in Fourier transformation [-]
$b_n$	Coefficients in Fourier transformation [-]

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## Chapter 1

# Introduction

#### **1.1 Background and Motivation**

Recent mooring line failures of semi-submersibles in the North Sea occurring at sea states lower than the design sea states, indicate that the predicted loads on these structures used in the design phase are too low in some wave consitions. This has emphasized a need for further investigation and improvement of the design predictions. Several studies have been carried out over the past years, but a full understanding of the problem is not yet obtained. However, several studies suggest that the wave forces, and especially the slow drift forces, are higher than anticipated. Wave drift forces are of higher order, which can be difficult to calculate accurately. Usually, potential flow theory is assumed to be applicable, which is further simplified by Newman's approximation using the first-order linear potential solutions. Comparison of model test results and results from potential-flow codes show that potential codes generally under-predict the loads for slender structures, particularly when the wave height increases. New empirical formulas are also developed in order to prevent underestimation of the mooring line loads, but the accuracy and validity of these formulas are not yet fully understood.

It has been shown that the viscous contribution to wave drift loading increases considerably with increased wave steepness, and is of particular interest in long waves where the viscous forces are dominating. Several studies have suggested that the viscous contribution to the wave drift loads is larger than previously anticipated. In the industry, the drag term of Morison's equation is used to

account for viscous effects in potential theory. Drag coefficients are found empirically by tuning against model tests. As stated by many studies, the most important contributions to the wave drift loads, both potential and viscous, originate in the free-surface zone. The free surface zone is the part of the structure which is in and out of water and there exist some uncertainties regarding the correct choice of drag coefficient in this region in particular. The free surface zone is associated with non-linearities both in the incoming and diffracted wave field, which make it difficult to obtain accurate results with a simplified drag formulation.

Ideally, wave loads should be calculated using the Navier Stokes equations as the governing equation when viscous effects are of importance. Due to the large processor capacity needed to perform these calculations, it can be very time consuming and expensive. However, due to the technological development within the area of computer performance, it becomes increasingly important to test and verify the Navier Stokes solvers for such problems.

### **1.2 Objective and Scope**

This master thesis is exploring the Navier Stokes solver OpenFOAM for calculation of the wave drift forces on a slender cylinder of quadratic cross section in 2D, fixed in space. The cylinder is subjected to selected wave conditions including regular and bi-chromatic waves. The work is a continuation of a preliminary study performed from January to March 2017.

The main objectives of this project have been to

- Apply the Navier Stokes solver OpenFOAM on a two-dimensional problem for a fixed cylinder in monochromatic and bi-chromatic waves
- Analyse the results in order to identify and discuss the wave drift forces
- Investigate the viscous contribution to the wave drift forces

## **1.3** Challenges With the Numerical Study

The simulations with OpenFOAM are partly done on a regular MacBook Pro computer and partly on the supercomputer Vilje, which is owned by both NTNU and the Norwegian Meteorological Institute. The processing capacity and storage possibilities of the personal computer are limited, and the number of elements and simulation time have therefore been kept low. In addition, OpenFOAM is Linux based, and has therefore been run using the program VirtualBox. VirtualBox makes it possible to run a different guest operative system than the existing operative system, but the program is only allowed to use a limited part of the computer's processing capacity. This may have increased the time needed to perform the analyses. In addition, VirtualBox makes the data handling challenging, as the only way of connecting the guest with the host operative system is via a shared folder. This often requires several loops of copying and transferring data back and forth, which is time consuming. Time limitations is always an issue when performing CFD analyses, however, having access to Vilje has made it possible to run several cases at the same time and decrease the time step.

A new toolbox for openFOAM called waves2Foam has been applied. This includes built-in methods to create wave damping areas and a new wave generation method, different from the wave maker used in the preliminary project. A considerable amount of time has been spent on getting familiar with these methods. Therefore, the thesis work has been limited to deal with the 2D problem. The project has been restricted to the diffraction problem, and the thesis represents the first step in the systematic approach in order to numerically study the wave drift forces. The focus has been on the methodology behind the study, and the simulated cases represent a small part of the real and more complex problem, where also 3D effects and wave radiation effects are anticipated to be important.

## **1.4 Outline of Project Thesis**

*Chapter 2* gives a brief introduction to the basic theory regarding wave drift forces, covers findings in available and relevant literature and summarises major findings from the initial project work.

Chapter 3 covers the general method used in the numerical study and the analysis of the results.

*Chapter 4* contains explanations and discussion of the choices made during the modelling of the domain and waves and throughout the simulations.

*Chapter 5* includes force and wave elevation results from the simulations and visual illustrations from selected time instants of the runs.

Chapter 6 discusses the results presented in Chapter 5.

*Chapter 7* includes conclusions from the performed work and the obtained results, and provides suggestions for further investigation and research of the problem.

## **Chapter 2**

## Introduction to wave drift forces

This chapter will cover the theory of wave drift forces, a brief summary of selected literature and results obtained during the preliminary study of this problem performed from January to March 2017.

#### 2.1 The Theory of Wave Drift Forces

Unlike linear wave forces with a zero mean value, wave drift forces are, in the framework of potential flow perturbation theory, second-order forces which generally have non-zero mean values. Wave drift forces can be divided into mean drift forces and difference-wave frequency forces, which are slowly varying loads. The slowly varying loads oscillate with low frequencies caused by sea states with more than one wave frequency component present (Greco; 2012). With two wave frequencies,  $\omega_j$  and  $\omega_i$ , the difference frequency is  $\Delta \omega = \omega_i - \omega_j$ . The second-order forces are often neglected because they might be small compared to the first-order force, but they are important for structures with low natural frequencies. Because the wave drift forces are of higher order and proportional to the wave amplitude squared,  $A^2$ , the importance of these forces typically increase with increasing wave height (Greco; 2012).

#### 2.1.1 Potential-Flow Theory Approach

Methods based on potential flow theory are widely used to calculate the wave drift forces because of their simplicity and due to the viscous contribution often being small enough to be neglected (Berthelsen et al. (2009) and Stansberg et al. (1998)). The higher order forces can be calculated by performing a perturbation analysis while keeping the terms up to the order of interest. For instance, starting from Bernoulli's equation and performing a perturbation analysis keeping the second-order terms, the pressure accurate to the second-order can be expressed as (Greco; 2012)

$$p = -\rho g z - \rho \frac{\partial \phi_1}{\partial t} - \rho \frac{\partial \phi_2}{\partial t} - \rho \frac{1}{2} \nabla \phi_1 \nabla \phi_1$$
(2.1)

where p is pressure,  $\rho$  is the density of water,  $\phi_1$  and  $\phi_2$  are the first and second-order velocity potentials, g is the gravitational acceleration and z is the vertical position. Bernoulli's equation, Equation 2.1, includes both first and second-order potentials and products between the gradient of the first-order potentials. As an example, the result of squaring the x-derivative of the first-order potential,  $\phi_{01} = A\cos(\omega t + \epsilon)$ , is

$$\left(\frac{\partial\phi}{\partial x}\right)^2 = \frac{A^2}{2} + \frac{A^2 \cos[2(\omega t + \epsilon)]}{2}$$
(2.2)

where A is the wave amplitude of the wave component and  $\epsilon$  is the phase angle. As shown in Equation 2.2 this second-order term contributes to a non-zero mean load (Greco; 2012).

#### **Mean Wave Drift Load**

In potential-flow theory, the mean wave drift forces originate from the diffraction and radiation of waves from the body. This implies that these loads are greater for large structures compared to the wave length, because of their capability of wave making. The mean wave drift forces can be estimated through either direct pressure integration or conservation of fluid momentum. Unlike for the slowly-varying loads, it can be shown that it is only the first-order velocity potential that contributes to the mean drift force. Hence, the mean drift force can be obtained from a linear potential-flow analysis. For a body in first-order incident waves, diffraction occurs because of the presence of the body and radiation because of the body motions. A simple way of calculating the mean wave drift of a 2D body is described by Maruo's formula, Equation 2.3, where the incident waves can split into the sum of transmitted and reflected waves with amplitudes  $A_T$  and  $A_R$ , respectively. The transmitted waves are the sum of incident waves, radiated waves and diffracted waves on the downstream side while the reflected waves include the radiated and diffracted waves on the upstream side. The largest mean wave drift forces using the formula occur when there are no transmitted waves or when  $A_R$  is very large. This can occur when waves are propagating towards a vertical fixed wall, if the waves are short compared to the encountering structure or if the body motions are large as a consequence of heave resonance (Faltinsen; 1990).

$$\overline{F} = \frac{\rho g}{2} A_R^2 \tag{2.3}$$

The mean wave drift force is proportional to the wave amplitude squared (Faltinsen; 1990), which means that the transfer function can be written as  $\overline{F_i(\omega)}/\zeta_a^2$ . The transfer functions are of importance, because they can be used to estimate the mean drift loads in irregular waves from the results in regular waves by dividing the irregular waves into N regular wave components and summing over all the components as given below (Faltinsen; 1990).

$$\overline{F_i} = \sum_{j=1}^{N} \frac{\overline{F_i(\omega_j)}}{\zeta_a^2} A_j^2$$
(2.4)

 $F_i$  is the force in the i-th direction,  $\zeta_a$  is the incident wave amplitude and  $A_j$  is the amplitude of the j-th wave component.

#### **Difference-Wave-Frequency Oscillations: Slow Drift Loads**

According to Faltinsen (1990), slow drift motions are resonance oscillations at low frequencies compared to the incoming wave frequency. The slow drift motions are caused by loads connected to second-order difference frequency effects. These non-linear loads can be critical for moored structures with large natural periods if resonance occurs. The long period and small frequency

imply that the linear wave radiation damping is small and that viscous contribution to damping may be important. This means that slow drift motions can be larger than linear motion in certain conditions.

Slow drift forces are caused by a sea state consisting of waves with at least two different frequencies,  $\omega_i$  and  $\omega_j$ . Both first and second-order velocity potentials contribute to slowly varying loads. An expression for the slow drift forces up to the second-order is given below (Faltinsen; 1990).

$$F_{i}^{SV} = \zeta_{a1}\zeta_{a2}T_{12}^{ic}\cos[(\omega_{2} - \omega_{1})t + (\epsilon_{2} - \epsilon_{1})] + \zeta_{a1}\zeta_{a2}T_{12}^{is}\sin[(\omega_{2} - \omega_{1})t + (\epsilon_{2} - \epsilon_{1})]$$
(2.5)

Where  $T_{12}^{is}$  and  $T_{12}^{ic}$  are the second-order quadratic transfer functions (QTF) for the slow drift loads which are independent of sea state, but functions of the frequencies  $\omega_i$  and  $\omega_j$  (Faltinsen; 1990).  $\epsilon$  is the phase angle and  $\zeta$  is the wave amplitude. This formula can be modified to be used for a wave spectrum  $S(\omega)$ , by splitting the spectrum into regular wave components with amplitudes  $A_j = \sqrt{2S(\omega_j)\Delta\omega}$  and performing a double summation over all the wave components. By using this method, both the mean and slowly varying forces are found. The diagonal of the QTF matrix constitutes the mean drift load and the off-diagonal values are related to the slow drift loads.

Generally, the second-order potential must be included in order to calculate the slow drift forces. However, through Newman approximation, given below, the calculations are simplified considerably, because the slowly varying loads then can be predicted based on the results from a linear analysis (Faltinsen; 1990).

$$T_{jk}^{ic} = T_{kj}^{ic} = \frac{1}{2} (T_{jj}^{ic} + T_{kk}^{ic})$$
(2.6)

$$T_{jk}^{is} = -T_{kj}^{is} = 0 (2.7)$$

The formulas above indicate the same as taking the values of the transfer functions along the line  $\omega_k \approx \omega_j$ , i.e. the mean drift force, given that the difference  $\omega_k - \omega_j$  is small. According to Faltinsen (1990), Newman approximation can be used when 1) the small difference frequencies and large oscillation periods are of interest and 2) when the transfer functions,  $T_{jk}^{ic}$  and  $T_{jk}^{is}$  does not change much with frequency around the main diagonal. Newman approximation should not be used when  $T_{ik}^{ic}$  has a pronounced maxima or minima close to the line  $\omega_j = \omega_k$ . If both the conditions are

satisfied, then

$$T_{jj}^{ic} \cong T_{kk}^{ic} \tag{2.8}$$

and

$$0.5(T_{jj}^{ic} + T_{kk}^{ic}) \cong \sqrt{T_{jj}^{ic} T_{kk}^{ic}}$$
(2.9)

Introducing the mentioned simplifications, the resulting force expression for the slowly varying force becomes (Faltinsen; 1990)

$$F_i^{SV} \cong \sum_{j=1}^N \sum_{k=1}^N A_j A_k \sqrt{T_{jj}^{ic} T_{kk}^{ic}} \cos[(\omega_k - \omega_j)t + (\epsilon_k - \epsilon_j)]$$
(2.10)

This expression depends on the transfer functions of the mean drift loads and the first-order solution in regular waves, and  $\phi_2$  does not need to be calculated. Potential-flow theory with Newman's approximation is widely used in industry (Berthelsen et al.; 2009).

#### 2.1.2 Viscous Effects

When the waves are long compared to the structure and the wave height increases, viscous effects become increasingly important relative to the potential-flow loads, because the viscous contribution is of higher order than second. Long waves are related to small wave generation causing less drift loads due to potential drift. In order to add the viscous effects to the potential-flow theory methods, the drag term of Morison's equation, Equation 2.11, based on strip theory is widely used in the industry (Berthelsen et al.; 2009). Morison's equation is a long wave approximation, valid when  $\lambda/D \gtrsim 5$  for calculation of wave loads on cylindrical structures (Faltinsen; 1990). The drag term of Morison's equation is given below.

$$F_V = -\int_L \frac{\rho}{2} C_D U |U| dl \qquad (2.11)$$

 $C_D$  is the drag coefficient, L is the length of the structure and U is the velocity perpendicular to the cylinder axis. The resulting force when the expression is integrated up to the instantaneous free surface is proportional to the wave amplitude cubed (Dev and Pinkster; 1995). This means that viscous load in the wave zone becomes very important for high waves that are long compared to

the structure. The drag coefficient must be empirically determined and is dependent on several flow variables (Faltinsen; 1990).

The choice of drag coefficient is not straightforward. The most accurate way of estimating  $C_D$  is by using results from model tests. However, the drag coefficient can also be estimated using the Keulegan-Carpenter number, KC-number, which is defined as (Faltinsen; 1990)

$$KC = \frac{U_M T}{D} \tag{2.12}$$

where  $U_M$  is the velocity of an oscillatory planar flow, T is the oscillation period and D is the characteristic cross-section dimension of the structure. From a defined KC-number,  $C_D$  can be found from relations and graphs based on previous experiments (Faltinsen; 1990).

## 2.2 Results and Methods Covered in Literature

A brief literature study has been performed, which covers some of the methods and results emphasized in existing studies of drift forces in cases where both potential and viscous loads matter. From the literature study, different methods used to calculate wave drift forces have been investigated. Research within this field are based on numerical, empirical and experimental methods in order to as accurately as possible analyse the wave drift forces and response. Model tests are often used to validate the results obtained using empirical formulas or numerical methods.

Most of the literature investigated agree that results from methods based on potential-flow theory show discrepancies when compared to experimental results regarding the wave drift forces on semi-submersibles (Berthelsen et al. (2009), Stansberg et al. (1998), Hu and Zhou (2015), Stansberg et al. (2016)). The potential-flow theory methods are most commonly based on Newman's approximation. Large deviations are observed in cases where the waves are long and for high sea states. When the viscous effects are added to the potential-flow theory solutions, the results show better agreement. Therefore, viscous effects are expected to be the main reason for the discrepancies towards results based on potential-flow theory. The studies also show that the relative
importace of the viscous contribution increases with increasing wave height and wave length. In order to calculate the viscous contribution, the most common approach is through Morison's equation. The results are most accurate if the drag term,  $C_D$ , is tuned against model tests and separate drag coefficients are introduced for the free surface zone (Berthelsen et al.; 2009).  $C_D$  is case dependent and choosing a large value may result in too large damping (Berthelsen et al. (2009), Stansberg et al. (1998)).

An empirical formula for calculation of the wave drift forces on slender structures has been proposed by a joint industry project (JIP) called EXWAVE (Stansberg and Nestegård; 2014). The formula includes the effect of current, and has been tested for different kinds of semi-sumbersible platforms and compared to experimental and numerical results. It shows good agreement in some cases and discrepancy in other (Stansberg et al.; 2016) without indicating a clear trend or pattern. The formula is given below.

$$F = \frac{F_D(\omega, U, H_s)}{A^2} = f_{d,pot}(1 + C_p U) + b(GU + Hs)$$
(2.13)  

$$b = b' D_{sum}$$
  

$$D_{sum} = \Sigma D_i$$
  

$$b' = k \cdot P$$
  

$$P = exp(-1.25(kD_0)^2)$$

The formula consists of two terms, the first one is related to the potential flow and the second accounts for the viscous effects.  $f_{d,pot}$  is the second-order mean wave drift coefficient calculated from first-order potential-flow theory. It can be found from potential-flow codes like WAMIT. WAMIT is a linear and second-order potential-flow program based on the panel method, where the output is in the frequency domain (WAMIT; 2017).  $C_p$  is the wave-current interaction coefficient from potential-flow theory and is set to be 0.25 in the EXWAVE study. b is the viscous drift,  $D_{sum}$  is the sum of the column diameters, b' accounts for the upper limit of frequency where the viscous effects are important, k is the wave number,  $D_0$  is the diameter of the main columns, U is the current velocity parallel to the waves, A is the wave amplitude and Hs is the significant wave height.

G, the viscous wave-current interaction term, is empirically determined and is equal to 10 in the EXWAVE study.

Model tests of semi-submersibles with the purpose of investigating the wave drift forces have recently been performed, and the results have been compared to the empirical formula (Fonseca (2016), Thys and Ambrahamsen (2016), Braaten (2016)). One of the semi-submersibles consists of columns with squared cross section with rounded corners (Fonseca; 2016), whereas the other two consist of circular columns (Thys and Ambrahamsen (2016), Braaten (2016)). The two semi-submersibles with circular columns differ by the size and number of columns. One of them consists of several small interconnecting cylinders in addition to the larger main columns. The reports show that there is a difference in the results for the three semi-submersibles. The large quadratic columns cause a considerable amount of wave radiation, which is visible from additional peaks in the foce coefficient plots. Another finding is that the force in the low frequency region is largest for the semi-submersible with the largest amount of columns. This is most likely due to the viscous effects as a consequence of large periods and small cylinder diameters of the interconnecting cylinders. For the two semi-submersibles with circular cross section, the interaction of the cylinders is visible in the results through less smooth force coefficient plots.

As for the geometry of the semi-submersibles investigated and covered in the literature, most are of squared section with rounded corners (Berthelsen et al. (2009), Stansberg et al. (1998), Hu and Zhou (2015), Fonseca (2016)). The draught is about 20 meters and the diameter of the columns are about 15 meters.

A 3D numerical study on the wave run-up and forces on a vertical fixed squared cylinder in regular waves has recently been published (Ali et al.; 2017). The main goal with the study was to validate and verify the computational fluid dynamics software OpenFOAM in prediction of wave run-up. The waves were validated against analytical and experimental results, and it was concluded that openFOAM can produce results in good agreement with experimental data and that the accuracy is satisfactory. The method used in this study is quite similar to the one used in this master thesis, although the parameters differ. The experimental results used for verification have been taken from Stansberg et al. (2005). This study also include an introductory CFD study on a circular vertical column in which significant non-linear contributions are observed in the free surface area.

## 2.3 Result from Preliminary Numerical Study

The project thesis, performed from January to March 2017, covered a preliminary analysis of the forces acting on a simple 2D geometry by use of the Navier Stokes solver OpenFOAM. For a simple 2D geometry, useful observations and experiences were made. The results from this CFD study were compared to model test results for the same body and wave parameters presented in the paper by Tanizawa et al. (1999).

The most important experience from the initial numerical study has been that the numerical wave tank needs to be made such that the resulting force time series is sufficiently long and not affected by reflected waves. Because the desired waves to be investigated are long compared to the structure, there will be a considerable amount of transmitted waves, which eventually will reflect back from the downstream boundary wall. Results show that the wave drift force for the particular case investigated is higher when the body is exposed to the reflected waves from the wall. In order to avoid this effect, the wave tank can be made very long (Stansberg et al.; 2005), as done in the preliminary project, or by introducing a damping zone in the downstream end of the wave tank. The last alternative is desirable because it involves a smaller domain, less elements and thus a shorter simulation time. The results from the analysis also show that that the wave drift force is need to be small enough to get accurate results. Another valuable experience is the quick increase in computational time associated with decreasing the element size and time step. However, the over-all numerical results match the experimental reference results quite well, which validates the use of the numerical method and the simulation variables for the particular case tested.

# **Chapter 3**

# **Description of Method**

This chapter covers the general methodology used in the nummerical study. It contains descriptions of the CFD software, solver, mesh generation, wave generation, wave damping method, the choice of time variables and post-processing methods.

## 3.1 Introduction to OpenFOAM

In order to numerically find the wave drift loads on a slender structure, the Navier Stokes solver OpenFOAM (Open Source Field Operation and Manipulation) has been used. OpenFOAM is an open source code to perform Computational Fluid Dynamics (CFD) simulations. The code is based on the programming language C + + and consists of different solvers and utilities designed to solve specific problems and perform tasks with data manipulation (CFD Direct; 2016). OpenFOAM comes with a post-processing utility based on ParaView, which is an open-source visualisation application (CFD Direct; 2016). The structure of the program can be found in Appendix B. In this thesis OpenFOAM is used to create a numerical wave tank in order to simulate wave drift loads on fixed slender surface-piercing structures in a viscous fluid. The Navier Stokes equations, conservation of momentum, and continuity equation, conservation of mass (see Equation 3.1 and 3.2 (Kundu and Cohen; 2007)), are solved for the defined elements in the domain for each time step.

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \rho g_i + \frac{\partial}{\partial x_j} \left[ 2\mu e_{ij} - \frac{2}{3}\mu(\nabla \cdot u_i)\delta_{ij} \right]$$
(3.1)

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho u) = 0 \tag{3.2}$$

 $u_i$  is the velocity,  $\mu$  is the dynamic viscosity,  $e_{ij}$  is the strain rate tensor and  $\delta_{ij}$  is the Kronecker delta function.

For an incompressible fluid, like water can be considered as in most cases, the conservation of momentum of the Navier Stokes equations reduces to Equation 3.3 (Kundu and Cohen; 2007).

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \rho g_i + \mu \nabla^2 u_i$$
(3.3)

A chosen set of cases with zero viscosity have also been run in order to investigate the effect of viscosity on the forces. By defining the viscosity to be zero,  $\mu = 0$ , the Navier Stokes equations reduces to the Euler equation as given in Equation 3.4 (Kundu and Cohen; 2007).

$$\rho \frac{Du_i}{Dt} = -\nabla p + \rho g_i \tag{3.4}$$

#### 3.2 Solver

OpenFOAM includes different solvers for different applications. In the preliminary project the solver interFoam was used, which is valid for two incompressible and isothermal fluids. The solver is based on VOF (Volume Of Fluid) phase-fraction to capture the interface between the fluid. The two fluids are air and water. The solver can be used even though air is not incompressible, because the force from the air is small compared to the force from the water. Using VOF method, fluids are specified by giving the desired elements a defined phase-fraction at the initial time step where 1 corresponds to 100 % water and 0 corresponds to 100 % air. Due to the issue of damping, described

in Chapter 2 Section 2.3, a new solver called waveFoam is used, which is a part of the toolkit waves2Foam. This solver is very similar to interFoam, but together with waves2foam it includes more features as will be covered later. The simulations have been run using a laminar turbulence model, which means that the flow behaviour is assumed to be laminar, and turbulence has not been modelled. The location of the separation points in a general boundary layer dependends on the flow being laminar or turbulent, but because the investigated geometry has sharp corners, the separation points are known.

#### **3.3** Computational Power

The simulations are partly run on a personal MacBook Pro computer via Ubuntu Linux on Virtual Box and partly on the super computer, Vilje, owned by both NTNU and the Norwegian Meteorological Institute. Vilje runs the opreative system SUSE Linux Enterprise Server 11 and have 1404 nodes with 16 cores per node (Hpc Group; 2017).

#### **3.4 Domain and Mesh Generation**

The domain of the numerical wave tank and the initial mesh generation has been made using blockMesh, a built-in utility of OpenFOAM. The coordinates of the tank boundaries are defined, and origo is placed upstream at the left bottom corner. For the 2D case, the thickness of the wave tank is set to be 0.1 meters. The overall dimensions of the wave tank are based on the wave parameters and will be presented in the next chapter. 70 % of the domain is water, while the rest is air. The utility snappyHexMesh has been used in order to generate a finer mesh around geometries and in the free surface zone, such that the surface elevation is defined as accurately as possible and such that there is no artificial numerical damping. In the free surface zone, the ratio between the wave height and the element size in the vertical direction,  $H/\Delta z$ , should ideally be larger than 10, and the ratio of wave length to horizontal element size,  $\lambda/\Delta x$ , should be larger than 20. In addition, the elements should have a shape close to a quadrat. The mesh size increases outside the

free surface region. SnappyHexMesh uses the existing blockMesh domain and refines the mesh according to specified refinement levels in desired regions and around defined geometries described in separate STL-files. The STL-files contains the information about the geometry and describes the surface of an object approximated by triangles (3D Systems; 2017).

In addition, the domain boundaries must be defined, and there exist a variety of possible boundary conditions that need to be chosen based on the problem and relevant theory. The boundaries of relevance in this thesis is the bottom of the wave tank, inlet and outlet wall, the sides of the wave tank, the cylinder boundaries and the boundary defining the vertical end of the wave tank.

#### **3.5 Wave Making**

During the first phase of this study the waves were generated in a similar way as in the preliminary project work. Because the results from the project were compared to experimental results, the waves were made using a numerical wavemaker based on first-order wavemaker theory. The wavemaker was made by imposing a special boundary condition at the upstream (left) wall, so that the wall acted as a wavemaker. The wave motion induced by a wavemaker can be determined well from linear wave theory (Dean and Dalrymple; 1991). The waves are generated by forced oscillation of the flap, or the upstream wall boundary. The flap type wavemaker is analogue with a vertical wall being able to rotate about a point. The wavemaker concept is described in figure 3.1. The wavemaker has a horizontal displacement and a surface respectively described by the following equations (Dean and Dalrymple; 1991).

$$x = S(z)\sin\omega t \tag{3.5}$$

S(z) is the wavemaker stroke. The ratio of wave height to stroke,  $\frac{H}{S}$ , satisfying the Laplace equation and the boundary conditions, is described by the following equation for a flap type wavemaker.

$$\frac{H}{S(0)} = 4\left(\frac{\sinh kh}{kh}\right)\frac{kh\sinh kh - \cosh kh + 1}{\sinh 2kh + 2kh}$$
(3.6)

*k* is the wave number for the progressive wave, h is the depth and z = 0 corresponds to the calm water level. The resulting wave elevation can be described by the following equation.



 $\eta = \frac{H}{S(0)} \cos(kx - \omega t) \tag{3.7}$ 

Figure 3.1: Flap-type wavemaker

The goal has been to create the numerical wave tank as identical as a real wave tank environment as possible. A flap type wave maker was initially of interest, because deep water is considered and the desired wave particle orbits are circular and not elliptical, as is the case for shallow water. This phenomenon is shown in figure 3.2 taken from Pettersen (2007). The flap type wave maker is more efficient in resembling deep-water wave conditions (Dean and Dalrymple; 1991).



Figure 3.2: Wave particle orbits (Pettersen; 2007)

However, a new method of generating the waves has been implemented through the work with this thesis. The method is a part of the toolkit waves2Foam. waves2Foam (Jacobsen; 2011) is a library made in order to handle the free surface wave generation and damping better than the basic openFOAM library. waves2Foam is built for the exact purpose of surface waves and is based on VOF (Volume of Fluid) method, which is also the base of the interFoam solver used in the project work. The library contains different wave theories, which means that different kinds of waves

easily can be made without having to modify the wave maker. Stokes first-order wave theory has been used, which generates linear waves described by the same equations as for the wavemaker generated waves, Equation 3.7.

The wave length, period and height have been decided upon by looking at the wave conditions used in the literature and former studies regarding wave drift forces on semi-submersibles.

In order to trace the surface elevation and analyse the waves, the actual free surface is measured at different locations in the wave tank during the simulations. This is done by saving the vertical position of the elements that have the volume fraction closest to the value of 0.5, which indicate that the element is in a fluid state between air and water. It is valuable to trace the wave elevation at different locations in the wave tank in order to investigate how the waves are propagating both with time and in space.

#### 3.6 Damping

A very long wave tank was used in the preliminary analysis done during the project work in order to generate a long enough steady state force time series without disturbance from the reflecting waves. The project report emphasizes the need of a damping area for further analysis of the problem in order to limit the number of elements that comes with a large domain. With a damping area present, the domain can be made much smaller while preventing reflection of waves from the boundaries and avoid interference with the force results.

There exist several methods of wave damping, for instance stretching the mesh in the end of the wave tank, create a beach and construct a sponge layer to absorb the energy. The simplest and most straight forward method is stretching the mesh. This has been tested, but because the waves investigated in this project are long, the method of stretching the mesh is not efficient. The method requires a smooth increase of the element size in order to avoid that the elements act as walls and reflect the waves. This requires a very long domain for the waves to be satisfactory damped. Due to the complexity of damping the waves properly, the toolkit waves2Foam has been introduced. It is important to note that the goal with this master project is to investigate the wave drift forces on a

slender body and not to make a perfect wave tank. Therefore, more time has been spent on trying to make a simple working wave tank that is good enough, rather than developing and validating different methods to make the perfect wave tank.

waves2Foam defines specific areas for damping and wave generation, called relaxation zones. Different geometries for the damping area are available, which makes it convenient to use for a variety of cases. As a default, damping areas at the inlet and at the outlet must be defined.

The damping zone in the waves2Foam toolkit is based on certain relaxation weights, w, included in the equations for volume fraction and velocity. These equations are updated and solved every time step in the relaxation zones. The equations are given below, Equation 3.8 and 3.9

$$u = (1 - w)u_{target} + wu_{computed}$$
(3.8)

$$\alpha = (1 - w)\alpha_{target} + w\alpha_{computed} \tag{3.9}$$

 $\alpha_{target}$  and  $u_{target}$  are the volume fraction and velocity at the outlet boundary and  $\alpha_{computed}$  and  $u_{computed}$  are the values calculated from the Navier Stokes Equations. There are different ways of calculating the relaxation weight, w. Anyhow, the weight is defined to be 1 at the start of the outlet damping area and decrease to zero at the end of the outlet damping area or at the outlet wall. For the inlet, the opposite is defined, the weight is 1 at the end of the inlet damping area and zero at the inlet boundary, corresponding to the beginning of the inlet damping area. The default way of defining the weight is exponential with an exponent, m, of 3.5. The equation is given below, Equation 3.10.

$$w = 1 - \frac{exp(\sigma^m) - 1}{exp(1) - 1}$$
(3.10)

## 3.7 Time Variables

There are several time variables that need to be defined in order for the simulation to run. An appropriate simulation time must be chosen. The simulation time is dependent on how much time it takes before the simulation reaches steady state, and how long the force time series need to be for the analysis and the Fourier transformation to become accurate enough.

Another variable to be decided upon is the time step. The time step is chosen based on the wave period, the element size and the Courant number. The Courant number is defined as  $Co = \frac{\delta t |U|}{\delta x}$ , which is the time step multiplied by the velocity through the cell divided by the element size in the direction of the velocity. According to CFD Direct (2016), temporal accuracy and numerical stability is obtained when the Courant number is less than one. Therefore, the final time step is chosen in accordance with the element size in order to assure a Courant number smaller than one.

The last time variable, writeInterval, defines how often the results are going to be saved. The force time series and the wave elevation time series are saved at each time step and are independent of the writeInterval variable. Therefore, the saved results are only used to visualize and investigate the flow characteristics in paraView, and there is no need to save the results too often.

#### 3.8 Post-Processing

paraView has been used to get a first impression of the results, visualise the waves on the body and to check the mesh generation. A picture of the user interface of paraView is shown in Figure 3.3. Matlab is used for further analysis, because it is faster and more convenient when interpreting the results.

In order to study the wave generation and find the actual wave height and period, the wave elevation time series are transformed into the frequency domain through the FFT (Fast Fourier Transformation) function in Matlab and the wave amplitudes and frequencies are collected. This method is also used for the force time series in order to investigate the frequency of the force peaks.



Figure 3.3: User interface of paraView

In order to further evaluate and interpret the force results from the time domain simulations, the time series are transformed into Fourier series. A force time series can be described by a Fourier series as shown below

$$F(t) = \frac{a_0}{2} + \sum_{n=1}^{\infty} (a_n \cos(n\omega t) + b_n \sin(n\omega t))$$

$$a_0 = \frac{1}{N \cdot T} \int_{t_1}^{t_1 + N \cdot T} f(t) dt$$

$$a_n = \frac{1}{N \cdot T} \int_{t_1}^{t_1 + N \cdot T} f(t) \sin(n\omega t) dt$$

$$b_n = \frac{1}{N \cdot T} \int_{t_1}^{t_1 + N \cdot T} f(t) \cos(n\omega t) dt$$
(3.11)

The first coefficient,  $a_0$ , corresponds to the mean force,  $a_n$  and  $b_n$  are the amplitude of the oscillating, harmonic force, n defines up to which order of harmonics the analysis is to be performed, T is the period, N is the number of periods and decides the size of the time series interval used in the Fourier analysis and  $t_1$  is the start of the interval. In order to perform a good and accurate Fourier Transformation, the interval of the time series used in the analysis has to be where steady-state is obtained. The interval should also be chosen such that it includes a complete number of wave

periods. The Fourier transformations of the force time series in this project have been performed in Matlab by use of a script made by one of the co-supervisors of this project, Finn-Christian W. Hanssen. The script outputs the mean wave drift force in x and z-direction, mean moment about the y-axis, the harmonic forces in x and z-direction up to the chosen order, the moment in y-direction up to the chosen order and the slow drift force in x and z direction and moment about y.

The mean force coefficients in x and z direction are made dimensionless by  $\rho gLA^2$  and the moment by  $\rho gLBA^2$ . The slow drift force in x and z-direction are made dimensionless by  $\rho gLA^2$  and the slow drift moment about y by  $\rho gBLA^2$ . The first to third-order coefficients in x direction are made dimensionless by  $\rho gLdA$ , in z direction by  $\rho gLBA$  and the moment about y by  $\rho gLBdA$ . L [m] is the length of the column in y-direction and is the same as the thickness of the wave tank in 2D, A [m] is the wave amplitude, B [m] is the breadth of the column in x-direction and d [m] is the draft of the column.

# **Chapter 4**

# **Modelling and Simulation**

This chapter is discussing and summarising the choices made during the pre-processing, simulation and post processing of the cases. All the case files have been generated through a MATLAB script based on a script made by on of the co-supervisors, Babak Ommani. This has made the change of parameters simple and clean. The chosen parameters are defined in or calculated by the script which further generates the OpenFOAM case files in the correct folder and directories. The MATLAB script can be found in Appendix C.

#### 4.1 Domain Geometry

The chosen geometry of the wave tank can be seen in figure 4.1. The wave tank length, L, is made 11 times the incident wave length,  $\lambda$ . To satisfy the deep water condition the depth has been set to  $0.5\lambda$ . As stated in Chapter 3, the water covers 70 % of the wave tank. The applied boundary conditions are presented and explained in Appendix A.2. As in the preliminary project, a flap-type wavemaker is first placed at the left, upstream boundary to generate waves. The damping area is placed at the downstream, right, end of the wave tank and can be seen in figure 4.1 by the striped region.

In order to trace the free surface as accurately as possible and due to the recommendations presented in the report from the preliminary project, the free surface is further refined. The initial blockMesh



Figure 4.1: Domain geometry

grid has 20 elements per wave length and a refinement level of 3 is defined in the free surface zone and close to the body using the snappyHexMesh utility. This causes the smallest element size to be approximately 1.4 m, which means that the ideal ratio of 10 elements per wave height is not fulfilled, as can be seen from the wave parameters presented i Section 4.3. Although it would be desirable to have a smaller element size in the vertical direction, the choice of element size is a matter of available time and nessesary accuracy. However, in the horizontal direction, the amount of elements more than fulfil the recommended amount discussed in Chapter 3. Because the simulation variables are connected, a smaller element size requires a smaller time step, which also increase the simulation time considerably. Figure 4.2 and 4.3 show the meshed wave tank domain as presented in paraView.

Due to the simplicity and convenience of the method, another way of generating waves using waves2Foam has been implemented and is used to create all the results to be presented in Chapter 5. The new cases using waves2Foam also contains a damping area at the inlet as shown in figure 4.1. This was not present in the domain for the cases generating waves with the wavemaker.

#### Figure 4.2: Domain with mesh

The locations of the probes for measuring the actual wave elevation during the simulations are shown in figure 4.4. Because the probes are placed at diffrent locations along the wave tank, it is possible to trace the wave propagation and validate that the generated waves are as expected. The



Figure 4.3: Close-up of domain with mesh

wave elevation results presented in Chapter 5 are collected from the probe located one cylinder diameter in front of the cylinder, the rest of the probes are only used for validation.



Figure 4.4: Location of probes

The full domain with the mesh and the volume fraction is shown in figure 4.5. Red corresponds to water and blue corresponds to air.



Figure 4.5: Domain with mesh and volume fraction

# 4.2 Body Geometry

The chosen body geometry is shown in figure 4.6. Due to the waves being long and that slender bodies will be the main focus in this thesis, only one body geometry is studied. It is assumed

that changing the geometry within the slender body condition in 2D will not affect the drift force considerably. In the 2D case, vortex shedding and separation will only occur from the bottom corners of the 2D cylinder. The chosen geometry is inspired by and chosen based on relevant geometry in litarature.



Figure 4.6: *Body geometry* 

## 4.3 Wave Parameters

Both regular and bi-chromatic waves have been simulated. The wave parameters are presented below.

#### 4.3.1 Regular Waves

First a set of regular wave runs with varying wave heights has been performed. The wave parameters of these runs are summarised in table 4.1. The wave period is long and is chosen based on literature and previous studies of the wave drift forces (Berthelsen et al. (2009), Stansberg et al. (1998), Stansberg and Nestegård (2014)). Results from these studies show that the viscous contribution to the wave drift forces for longer waves and larger wave heights. Even though it is interesting to investigate and study waves with a high steepness, the chosen cases have smaller

steepness than 1/30, because it can be difficult to model steeper waves in openFoam. In order to accurately model steep waves, the elements must be even smaller which requires a further reduction of the time step and the computational time will increase quickly.

Run	Wave lenght [m]	Wave height [m]	<b>Steepness</b> $H/\lambda$ [-]	$\lambda/D[m]$
1	224.8	4.5	1/50	18.7
2	224.8	6	1/37	18.7
3	224.8	7.5	1/30	18.7

**Table 4.1:** Wave parameters for regular wave runs with T = 12 s

The regular wave cases listed in Table 4.1 have been run for waves both generated with the wave maker created in the project phase and with the waves2Foam toolkit in order to test if the toolkit could be used for the remaining of the simulations. These simulations were run with a time step of 0.05 s. The resulting wave elevation from these simulations are shown in Table 4.2 and 4.3. The force time series are shown in Figure 4.7 and 4.8.

**Table 4.2:** *Result from wave maker generated waves of* T=12 s

	Run 1	Run 2	Run 3
Actual wave amplitude [m]	1.899	2.521	3.035

**Table 4.3:** *Result from waves2Foam generated waves of T=12 s* 

	Run 1	Run 2	Run 3
Actual wave amplitude [m]	1.950	2.595	3.067

Looking at the wave elevation results, it is clear that wave generation using waves2Foam creates a wave with an amplitude somewhat closer to the theoretical value. Regarding the force time series, the figures show similar results for the steady state region. Because the plots show quite similar results, waves2foam has been considered valid to use for the remaining of the simulations.

#### 4.3.2 **Bi-chromatic Waves**

The cases performed for bi-chromatic waves are summarised in the three tables below, Table 4.4, 4.5 and 4.6. The difference frequencies are chosen to be in the range of natural periods of



Figure 4.7: Force time series for simulation using a wavemaker

semi-submersibles covered in literature. This is of relevance because sea states containing these frequencies can be the most severe conditions regarding slow drift wave loads and excitation. Moored semi-submersibles generally have a natural period in surge of about 100 s. The two chosen difference frequencies correspond to slowly varying periods of 125.6 s and 89.8 s. Simulations are done for three different wave heights and two different difference frequencies in order to investigate how they are related to the resulting wave loads.

Run	$\omega_1 - \omega_2$ [rad/s]	<i>T</i> <sub>1</sub> [s]	T <sub>2</sub> [s]	$H/\lambda_1[-]$	$H/\lambda_2[-]$	$\lambda_1/D$ [-]	$\lambda_2/D[-]$
1	0.05	12	13.27	1/50	1/65	18.7	22.9
2	0.07	12	13.85	1/50	1/66	18.7	25.0

**Table 4.4:** *Wave parameters for bi-chromatic wave runs with* H = 4.5 m



(c) Run 3

Figure 4.8: Force time series for simulation using waves2Foam

Run	$\omega_1 - \omega_2$ [rad/s]	$T_1[\mathbf{s}]$	$T_2[\mathbf{s}]$	$H/\lambda_1[-]$	$H/\lambda_2[-]$	$\lambda_1/D$ [-]	$\lambda_2/D[-]$
1	0.05	12	13.27	1/37	1/46	18.7	22.9
2	0.07	12	13.85	1/37	1/50	18.7	25.0

**Table 4.5:** Wave parameters for bi-chromatic wave runs with H = 6 m

**Table 4.6:** Wave parameters for bi-chromatic wave runs with H = 7.5 m

Run	$\omega_1 - \omega_2$ [rad/s]	$T_1[\mathbf{s}]$	$T_2[\mathbf{s}]$	$H/\lambda_1[-]$	$H/\lambda_2[-]$	$\lambda_1/D$ [-]	$\lambda_2/D[-]$
1	0.05	12	13.27	1/30	1/37	18.7	22.9
2	0.07	12	13.85	1/30	1/40	18.7	25.0

The cases summarised in Table 4.4 to 4.6 are also simulated without the body present in order to only investigate and analyse the wave propagation. In addition, the cases with a difference

frequency of 0.05 rad/s have been simulated with zero viscosity,  $\mu = 0 \ kg/(m \cdot s)$ , in order to investigate the viscous contribution to the forces. The results from these simulations are solutions of the Euler equation.

Figure 4.9, 4.10 and 4.11 show details of the wave propagation for a wave height of 6 m and difference frequency of 0.05 rad/s.



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Figure 4.10: Waves at the inlet of the domain with mesh



Figure 4.11: Waves around the body

# 4.4 Damping Parameters

The relaxation weight has been defined defined through an exponential function, as explained in Chapter 3, where the exponent, m, in Equation 3.10 has been set to 3.5. The relaxation area is rectangular and is defined at the inlet and at the outlet of the domain.

## 4.5 Adjusting Time Variables

The initial time step used in the simulations was 0.05 s, and was chosen based on the smallest element size, which is found in the free surface area and is approximately 1.4 m, together with the Courant Number. With a time step of 0.05 s, element size of 1.4 m and a phase velocity of approximately 18 m/s, the resulting courant number is Co = 0.64 which is below 1. After studying the wave elevation results and getting access to Vilje, the time step has been reduced to 0.01 s. Comparison of the simulated and theoretical wave elevation, according to linear wave theory, for the time steps 0.05 s and 0.01 s can be seen in figure 4.12, 4.13 and 4.14, corresponding to the regular waves presented in Table 4.1 with wave height 4.5, 6 and 7.5 m.



(*a*)  $\Delta t = 0.05$ 



Figure 4.12: Comparison of time step and simulated and theoretical wave elevation for H=4.5 m



Figure 4.13: Comparison of time step and simulated and theoretical wave elevation for H=6 m



**Figure 4.14:** Comparison of time step and simulated and theoretical wave elevation for H=7 m

From the figures, it is clear that the accuracy of the wave elevation decreases as the steepness of the wave increases. Figure 4.14 a) shows that the wave elevation is decreasing with time, which indicates that steady state is not obtained. In addition, the wave frequency is varying, causing a phase difference with the theoretical wave. Looking at Figures 4.12 to 4.14 from the simulations with a time step of 0.01 s, the plots indicate that the simulated waves are closer to the desired waves and that steady state is more or less obtained. The wave elevation shows some discrepancies towards the wave through, which is most likely due to the element size being too big. These plots also show that the wave frequency is more constant. Therefore, a time step of 0.01 s has been used in the simulations producing the final results presented in Chapter 5.

The total simulation time for the cases where a wave maker is used to generate waves is about 550 s and is based on and limited by the waves' return time after reflecting from the body and traveling back to the body again. When the toolbox waves2foam is used, a damping area at the inlet of the wave tank is applied which means that reflections from the body will die out as it reaches the inlet. In this case, the simulation time is chosen to be 800 s for the cases where the cylinder is subjected to bi-chromatic waves. This result in about four steady state periods of the slowly varying wave in case of the smallest difference frequency. In regular waves a simulation time of 500 s is considered sufficient for the post processing of the results to be accurate enough. Regardless of which method is used to produce waves, a damping zone is placed at the downstream boundary, which means that there are no reflected waves from the domain boundary at the downstream side of the body.

# 4.6 Post Processing

Fonseca et al. (2011) is used as an inspiration for the post-processing methods and appearance of the results presented in Chapter 5. The post processing of the results from the simulations consist of mostly Fourier analyses of the force and wave elevation time series. For the analysis to be accurate enough, the data interval used must be taken from where steady state has been reached, as discussed in Chapter 3. For the regular waves, the analysed interval from the time series consists of 10 wave periods,  $10 \cdot T$  [s], and the interval starts at the time equal to 30 wave periods,  $30 \cdot T$  s. For the irregular waves, 4 periods of the envelope of the waves are used in the analysis and the start of the interval depends on the difference frequency of the bi-chromatic wave. With a difference frequency of 0.05 [rad/s] the interval starts at 2.22  $\cdot T$  s and for a difference frequency of 0.07 [rad/s] the interval starts at 4.05  $\cdot T$  s. This is descided upon through close investigation of the time series. The nominal amplitude has been used to non-dimensionalize the forces, because it is dificult to measure the exact actual wave amplitude of the wave components at the location of the body in case of bi-chromatic waves from the simulations.

A selection of the MATLAB scripts used for post processing of the results are included in Appendix C.

The main interest throughout this thesis has been towards the mean wave drift and slow drift forces, and the presented results are reflecting this.

# **Chapter 5**

# **Results from the Numerical Study**

## 5.1 Regular Waves

This section includes a selection of both the wave elevation results and the force results from the regular wave simulations. The results will be discussed in Chapter 6.

#### 5.1.1 Wave Elevation Results



**Figure 5.1:** *Regular waves: wave elevation analysis for* H = 6m *and* T = 12s

The time series of the wave elevation and the wave elevation spectrum from the regular wave simulations with wave amplitude 6 m and period 12 s are shown in figure 5.1. The vertical lines in the spectra show the most important existing frequencies in the wave condition.  $\omega$  has the unit [rad/s]. The wave elevation data is measured at the probe located one cylinder diameter upstream of the body. The results from the remaining simulations can be found in Appendix B.

#### 5.1.2 Force Results

Figure 5.2 show the force time series and Figure 5.3 show the force spectrum from the Fourier analysis of the force time series for wave height 6 m and period 12 s. The colored lines show the frequencies of the force peaks and the y-axis marks the peaks of the mean force. The results for the remaining simulations can be found in Appendix B.



**Figure 5.2:** *Regular waves: force time series for* H = 6m *and* T = 12s



**Figure 5.3:** *Regular waves: force spectrum for* H = 6m *and* T = 12s

The force coefficients from the regular wave simulations with wave period 12 s are shown in figure 5.4 to 5.7.



Figure 5.4: Regular waves: coefficients of mean drift force







Figure 5.6: Regular waves: coefficients of force in z direction



Figure 5.7: Regular waves: coefficients of moment about y axis

Table 5.2 summarises the percentage of change of the force coefficients presented in Figure 5.4 to 5.7.

Force Direction	Mean Drift	1. Order
Force in x	5.19 %	0.17 %
Force in z	-6.48 %	2.46 %
Moment about y	8.78 %	-2.00 %

 Table 5.1: Regular waves: percentage change of force coefficient from lowest to highest amplitude

# 5.2 Bi-Chromatic Waves

This section includes a selection of the wave elevation and force results from the bi-chromatic wave simulations. In addition, it contains pictures and illustrations from the simulations made in paraView.

#### 5.2.1 Results from paraView

The figures below are taken from post processing in paraView of results from the simulations of bi-chromatic waves with a difference frequency of 0.05 rad/s and where the two individual wave components have a wave height of 6 m.



Figure 5.8: Plot of the dynamic pressure distribution during wave propagation

Figure 5.9: Plot of fluid velocity during wave propagation



Figure 5.10: Plot of fluid velocity vectors around the body



Figure 5.11: Plot of fluid velocity vectors at the inlet



**Figure 5.12:** *Plot of velocity vectors around the body for* H=4.5 *m and*  $d\omega = 0.05$  *at* t=700 *s* 



**Figure 5.13:** *Plot of velocity vectors around the body for* H=6 *m and*  $d\omega = 0.05$  *at* t=700 *s* 



**Figure 5.14:** *Plot of velocity vectors around the body for H=7.5 m and*  $d\omega = 0.05$  *at t=700 s* 



**Figure 5.15:** *Plot of velocity around body for H=4.5 m and*  $d\omega = 0.05$  *at two different time instants* 

#### 5.2.2 Wave Elevation Results

Figures 5.16 to 5.19 show the time series of the wave elevation together with the spectra of the time series made from the interval marked on the plot. The wave elevation data is measured at the probe located one cylinder diameter upstream of the body. Selected results from the runs with bi-chromatic waves with a difference frequency of 0.05 rad/s and wave height 4.5, 6 and 7.5 m and difference frequency of 0.07 rad/s for wave height 6 m are included, the results for the remaining simulations can be found in Appendix B.



**Figure 5.16:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: wave elevation time series for* H = 4.5m



**Figure 5.17:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: wave elevation time series and spectrum for* H = 6m



**Figure 5.18:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: wave elevation time series and spectrum for* H = 6m



**Figure 5.19:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: wave elevation time series for* H = 7.5m

#### 5.2.3 Force Results

Figure 5.20 to 5.27 show the force time series and the spectra from the Fourier analysis of the force time series. The time series interval used in the Fourier analysis is marked in the plots of the time series . The colored lines in the plots of the spectra show the frequencies of the force peaks. The unit of  $\omega$  is rad/s.


**Figure 5.20:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: force time series for* H = 4.5m



**Figure 5.21:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: force spectrum for* H = 4.5m



**Figure 5.22:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: force time series for* H = 6m



**Figure 5.23:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: force spectrum for* H = 6m



**Figure 5.24:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: force time series for* H = 6m



**Figure 5.25:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: force spectrum for* H = 6m



**Figure 5.26:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: force time series for* H = 7.5m



**Figure 5.27:** *Bi-chromatic waves with*  $d\omega = 0.05$ *: force spectrum for* H = 7.5m

The mean drift force coefficients for the runs with bi-chromatic waves are shown in figure 5.28 to 5.30. The slow drift coefficients are presented in figure 5.31 to 5.33. The first-order force coefficients related to the single wave components' frequency for the bi-chromatic runs with a difference frequency of 0.05 rad/s and 0.07 rad/s are presented in figure 5.34 and 5.35, respectively. The third order force coefficients related to the frequencies  $2\omega_1 - \omega_2$  and  $2\omega_2 - \omega_1$  are plotted in figure 5.36.



Figure 5.28: Bi-chromatic waves: coefficients of mean drift force in x direction



Figure 5.29: Bi-chromatic waves: coefficients of mean drift force in z direction



Figure 5.30: Bi-chromatic waves: coefficients of mean drift moment in y direction



Figure 5.31: Bi-chromatic waves: slow drift force coefficient in x direction



Figure 5.32: Bi-chromatic waves: slow drift force coefficient in z direction



Figure 5.33: Bi-chromatic waves: slow drift moment coefficient about y



**Figure 5.34:** *Bi-chromatic waves: first-order force coefficients for*  $d\omega = 0.05$ 



**Figure 5.35:** *Bi-chromatic waves: first-order force coefficients for*  $d\omega = 0.07$ 



**Figure 5.36:** *Bi-chromatic waves: third order force coefficients for*  $d\omega = 0.05$ 

The following tables show the percentage of change of the presented force coefficients from smallest to largest amplitude. Table 5.2 shows the mean drift force, 5.3 shows the slow drift force and 5.4 and 5.5 show the first-order force.

Table 5.2: Percentage change of mean drift force coefficient from lowest to highest amplitude

Force Direction	$d\omega = 0.05$	$d\omega = 0.07$
Force in x	21.32 %	30.23 %
Force in z	-3.97 %	- 15.42 %
Moment about y	2.43 %	- 0.69 %

 Table 5.3: Percentage change of slow drift force coefficient from lowest to highest amplitude

Force Direction	$d\omega = 0.05$	$d\omega = 0.07$
Force in x	28.14 %	28.58 %
Force in z	6.86 %	7.17 %
Moment about y	11.40 %	18.26 %

**Table 5.4:** *Percentage change of first-order force coefficient from lowest to highest amplitude for*  $d\omega = 0.05$ 

Force Direction	$\omega_1 = 0.5236$	$\omega_2 = 0.4736$
Force in x	-15.94 %	5.46 %
Force in z	-7.86 %	13.56 %
Moment about y	-25.83 %	-3.00 %

Force Direction	$\omega_1 = 0.5236$	$\omega_2 = 0.4536$
Force in x	-21.07 %	5.59 %
Force in z	-8.81 %	15.50 %
Moment about y	-30.68 %	-1.29 %

**Table 5.5:** *Percentage change of first-order force coefficient from lowest to highest amplitude for*  $d\omega = 0.07$ 

## 5.3 Inviscid Flow Case

Selected results from the inviscid flow simulations for bi-chromatic waves with a difference frequency of 0.05 rad/s and zero viscosity,  $\mu = 0 kg/(m \cdot s)$ , are included in this section. The rest of the results and figures are included in Appendix B.

#### 5.3.1 Wave Elevation Results

The wave elevation results from the inviscid simulation with bi-chromatic waves of height H = 6m are presented in figure 5.37.



**Figure 5.37:** Inviscid flow case with  $d\omega = 0.05$ : wave elevation time series and spectrum for H = 6m

A plot of the velocity vectors around the column is shown in Figure 5.38.



**Figure 5.38:** Inviscid flow case with  $d\omega = 0.05$ : plot of velocity vectors around the column

#### 5.3.2 Force Results

Force results, force time series and force spectra, from the inviscid simulations for bi-chromatic waves of height H = 6m are presented in figure 5.39 to 5.43. In figure 5.41 to 5.43, the force coefficient results from the inviscid flow simulations are plotted together with the viscous flow results. Figure 5.41 shows the coefficients of the mean drift force, Figure 5.42 shows the first-order force coefficients and Figure 5.43 the slow drift force coefficients.



**Figure 5.39:** Inviscid flow case with  $d\omega = 0.05$ : force time series for H = 6m



**Figure 5.40:** *Inviscid flow case with*  $d\omega = 0.05$ *: force spectrum for* H = 6m



**Figure 5.41:** Inviscid and viscous flow results for  $d\omega = 0.05$ : mean drift force coefficients



**Figure 5.42:** Inviscid and viscous flow results for  $d\omega = 0.05$ : first-order force coefficients



**Figure 5.43:** Inviscid and viscous flow results for  $d\omega = 0.05$ : slow drift force coefficients

# **Chapter 6**

## **Discussion of Results from Numerical Study**

This section covers a discussion of the results and findings presented in Chapter 5. As discussed in Chapter 1, this study is intended as the first step of a comprehensive study of wave drift forces on semi-submersible platforms. The focus has been on the numerical methodology and the analysis and understanding of the results. More attention has been directed towards the surge force as a consequence of what has been focused on in the literature.

#### 6.1 Regular Waves

The wave elevation time series and the Fourier analysis of the time series shown in Figure 5.1 in Chapter 5 and B.1 and B.2 in Appendix B show three peaks at the wave frequencies which are present in the wave field,  $\omega = 0$ ,  $\omega_1$  and  $2\omega_1$ . These force peaks represents the mean force, first-order force and second-order force, respectively. The single wave frequency component, corresponding to  $\omega_1$ , is much larger than the double-frequency component. This is expected. From the time series it can be seen that some time is required for the wave field to stabilise. The figures of the interval of the wave elevation time series presented in Section 4.5 show that the difference between the actual and theoretical wave increases slightly with increasing wave amplitude. As the wave height increases, the wave becomes more non-linear, and the wave will deviate from linear wave theory. This means that the run with an amplitude of 4.5 m most likely will give a more accurate result compared to the run with an amplitude of 7.5 m in terms of linear wave theory.

In the Fourier analysis of the force time series for the regular wave with 6 m wave height, presented in Figure 5.3 in Chapter 5, three major peaks are visible. These peaks represent the expected harmonic components, corresponding to the frequencies of the peaks in the wave elevation spectrum. The same can be noted from the rest of the results presented in Appendix B. As expected, the first-order force is the largest for all the runs, and is followed by the mean force and the second-order force. The force time series in Figure 5.2 show the interval in which the force spectrum has been calculated. It shows that steady state is reached quite quickly after the waves meet the column.

From the plots of the mean drift force coefficients in Figure 5.4 it is clear that the magnitude of the mean drift moment about y is the largest. The graphs are more or less constant with wave amplitude, which means that the mean drift forces are proportional to the amplitude squared. The mean drift force coefficients in x and z direction, surge and pitch, have about the same magnitude, however the vertical force is negative. It is important to note that the hydrostatic force has been subtracted from the heave force.

Figure 5.5, 5.6 and 5.7 show the same as stated from the Fourier analysis, namely that the first-order force components are much bigger than the second and third order force components. In x-direction, the coefficients are more or less constant with wave amplitude, however in z-direction and for the moment coefficient about y, the second-order force coefficients are increasing with increasing wave amplitude.

#### 6.2 Bi-Chromatic Waves

The figures taken from paraView presented in Section 5.2.1, show the velocity and pressure distribution in the domain at fixed timesteps. Figure 5.8 shows the pressure distribution of the travelling wave when the hydrostatic pressure has been subtracted. High pressure, red, correspond to a wave crest and low pressure, blue, correspond to a wave trough. The modulated wave is clearly visible. Figure 5.9 shows the velocity distributon. The highest velocities are found close to the free surface and the highest wave amplitudes cause the air to flow faster. Figure 5.10 and 5.11 show the

velocity vectors around the column and at the inlet of the domain. The circular wave particle orbits are visible.

Figure 5.15 show the velocity distribution around the bottom of the column at two different time instants. High velocities are found close to the column and a pattern that look like vortices is present, although it is only slightly visible. Vortices are expected, because the flow separates at the sharp corners, but the element size might be too large in order to capture the vortex propagation clearly.

The effect of simulating bi-chromatic waves is clearly visible in the time series of the wave elevation, because of the modulated amplitudes. The wave spectra in Figure 5.16 to 5.19 show that non-linear effects from the wave-body interaction are present, because the wave spectra are broader. The sea states contains a wider range of excitation frequency content compared to the regular wave runs and all the wave frequencies are present. The same is visible in the result for the other cases included in Appendix B. As the wave height increases, the spectra becomes less clean and the energy is no longer limited and present only at the expected frequency peaks, additional force peaks appear. This might indicate that running cases with higher wave heights in openFoam are less accurate and appear more noisy, likely because of the difficulty of simulating steep waves accurately.

For the force time series and the associated Fourier analysis of the time series presented in Figure 5.20 to 5.27, the same harmonic components are present as in the wave spectra. The force time series show the same behavior as the wave elevation time series. The force spectra show that the first-order force components related to the single wave frequencies,  $\omega_1$  and  $\omega_2$ , are the largest components in all cases. However, for the heave force, in particular, the mean drift force is quite large relative to the first-order force components. The magnitude of the slow drift force peaks, for all the force components, are much smaller than the first-order and mean force peaks, but increase with increasing wave height.

An interesting observation from the wave amplitude and force spectra is the peaks at frequencies  $2\omega_1 - \omega_2$  and  $2\omega_2 - \omega_1$ . This is an effect caused by interaction of first and second-order harmonics Fonseca et al. (2011). The Fourier analysis show that this effect is larger than the second-order

effect related to  $2\omega_1$  and  $2\omega_2$ . The corresponding force coefficients are presented in Figure 5.36. The figure shows that the coefficients are increasing with wave amplitude.

The plots of the mean drift force coefficients in Figure 5.28 to 5.30 show that the mean drift force is increasing with increasing amplitude in surge and partly in pitch and decreasing in heave. From Table 5.2, it is clear that the change is most significant in surge and that the mean pitch moment is changing very little. The mean coefficient plot suggest that the surge coefficient is proportional to more than the amplitude squared. This may be explained by looking at Figure 5.12, 5.13 and 5.14. As shown in these figures, the asymmetry of the wave elevation on the upstream and downstream side of the column increases with growing wave height. This asymmetry can cause a considerable surge force and may explain the increasing coefficient. In addition, vortex shedding from the bottom of the column may become stronger with larger wave heights causing an increasing force coefficient.

Figure 5.29 shows that the mean force in heave is decreasing with increasing amplitude, especially when the difference frequency is 0.07 rad/s. A possible explanaition can be that the local hydrostatic pressure, as a consequence of the wave run-up along the column wall, increases for higher wave heights and counteracts the dynamic pressure such that the total force decreases. This, however, needs to be further validated.

Considering the mean drift force coefficients, the cases behave in a similar manner regardless of difference frequency, except for pitch. However, the cases with a difference frequency of 0.07 rad/s have somewhat smaller values compared to the cases with difference frequency 0.05 rad/s, except for the highest amplitude in heave.

Figure 5.31, 5.32 and 5.33, presenting the slow drift force coefficients, show that the slowly varying forces are increasing with increasing wave amplitude for all the force directions. Table 5.3 shows that the relative increase is largest for surge and for a difference frequency of 0.07 rad/s. In addition, the figures imply that the slow drift force coefficients are proportinal to more than the amplitude squared. The behaviour is similar regardless of difference frequency, but a difference frequency of 0.07 rad/s is connected to the largest force coefficients in surge and heave.

The first-order force coefficients with respect to  $\omega_1$  and  $\omega_2$  are shown in Figure 5.34 and 5.35. These plots together with Table 5.4 and 5.5 show that the first-order force coefficients connected to  $\omega_2$  increase while the coefficient related to  $\omega_1$  decrease with increasing wave amplitude. This is also shown in Table 5.4 and 5.5. However, the rate of change of the coefficients are varying.

### 6.3 Inviscid Flow Case

The results from the simulations with zero viscosity presented in Chapter 5 are very similar and almost equal to the results from the simulations with viscosity. This is clearly visible from the force coefficient plots in Figure 5.41 to 5.43. After defining the viscosity to be zero, the Euler equation is solved, as explained in Chapter 3. This means that the flow still is rotational and that vortices may exist in the fluid, and flow separation will occur at the sharp corners of the cylinder. The presented results indicate that non-viscous flow separation is more important than viscous flow separation. In addition, the viscous effects are most likely very small for the presented cases. This might be different in case of 3D simulation, when the flow of water around the cylinder can cause additional drag and frictional forces.

From Figure 5.38 it is clear that separation occurs at the two corners at the bottom of the column, and that vortces develops close to the column wall. However, the grid resolution is most likley too low to capture the possible vortices further downstream of the column. This means that the grid size may be too large in order to correctly capture the viscous effects in the visocus flow cases as well.

## 6.4 General Remarks

When discussing the force results it is important to keep the natural periods of the structure in mind. Even though this study has focused on a simple, fixed 2D geometry of quadratic shape, the overall goal is to eventually releate the results to a complete semi-submersible platform. The main

focus has been towards the mean and slowly varying drift forces in surge, which means that the sum-frequency components generally have been neglected in the discussion.

Although the first-order force components related to the frequencies of the two single waves are the largest, the slow drift force can become of great importance if the semi-submersible platform has a natural period close to the slowly varying wave period. The slowly varying wave periods are 125.6 s and 89.8 s which is in the range of the natural surge periods for semi-submersibles. In heave, the natural period is proportional to the water plane area and are over 20 s, corresponding to an angular frequency of about 0.3 rad/s. Looking at the wave elevation spectra for heave, there are no considerable energy located around this frequency, meaning that the heave excitation will be low. The surge excitation is therefore of greater interest.

However, there are limited connections to draw between these results and the total expected behaviour of a complete semi-submersible platform. A platform consists of several columns and pontoons that will cause coupled effects and interaction. In a real environment, the bottom of the column will not experience force directly from the wave, there will be forces on the pontoon. The behaviour and excitation of a structure depends on the system characteristics, and to be able to say anything definite and certain about a structures excitation or behaviour, the correct geometry must be investigated.

It is important to emphasize that this study has investigated the fundamental mechanisms for a simple geometry, which is important in order to understand the mechanisms in a real environment. It is difficult to understand the total picture when studying all the effects at the same time, and isolating and identifying the induvidual contributions, as done in this study, is valuable.

Due to the simple geometry, the presented force results can also be valuable for and related to other floating structures than semi-submersibles, for instance such as spread moored FPSO's (Floating Production, Storage and Offloading) subjected to incoming waves from the side and SPAR-buoys.

## Chapter 7

# **Conclusions and Recommendations for Further Work**

### 7.1 Conclusion

The presented results show that the expected dominating frequencies are present in the spectra of the forces and wave elevation. For the bi-chromatic wave cases, the mean drift force in surge is proportional to more than the wave amplitude squared. This is also the case for the slow drift forces, which clearly increase for higher wave amplitudes. Furthermore, the results show that the viscous contribution to the wave drift loads is small and that non-viscous flow separation is present at the sharp corners at the bottom of the column. However, the grid resolution might have been too large in order to capture the viscous effects correctly and the elements should be smaller, at least further below the column. The problem in this thesis has been limited to 2D, and 2D cases are quite far from a real situation. However, the findings and the methodology presented are valuable as a basis for a future study of the wave drift forces in three dimensions with additional features as for instance wave-current interaction.

## 7.2 **Recommendations for Further Work**

Further work on this problem is of great interest. This thesis has been more of an in-depth systematic study in order to clarify the general methodology of a numerical study of the wave drift

forces. This thesis represents the first step in the comprehensive numerical investigation, and it will be valuable to build the cases further. The grid resolution should be increased in order to be able to capture the viscous effects better, at least below the column. It would also be interesting to simulate steeper waves, which will require careful consideration and evaluation of the simultion and mesh parameters to ensure a stable solution.

Future investigation will benefit from building up the 3D case to study the 3D effects on the structure as well as the effect of vicosity. A 3D case allows for flow around the body which result in a larger wetted surface and probably larger drag forces. In addition, more model test results are available for 3D cases than for 2D cases. Furthermore, the empirical formula presented in Chapter 3 can be applied to investigate and compare against the simulation results.

This study has been focusing on a part of the over-all problem by concentrating on a single column representing a platform column. It will be valuable to eventually study and build up the case towards a complete semi-submersible, including the pontoons, in order to study how the different parts interact and influence the wave drift forces. Further, it will be valuable to include the radiation problem and not only the diffraction problem by creating a floating platform in order to simulate a more real environment. This, however, is a quite large task and it is assumed to be very time consuming to reach this step.

In general, CFD simulations should be performed together with model tests in order to verify the results against empirical data. In addition, the empirical formulas should be investigated and developed further, because they can be valuable for quick estimations of the forces. Comparing results from potential codes with CFD calculations can also be practical and efficient in order to investigate conditions where the potential and viscous theory differ.

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# Appendix A

# **OpenFOAM Details**

## A.1 Overview of OpenFOAM Case Set-Up



Figure A.1: Overview of initial programming set-up

Figure A.1 shows roughly how OpenFoam is organised. The boxes represent the different directories.

The folder *0.org* contains the initial conditions of the problem and the boundary conditions for the different parts of the domain boundaries.

The folder *constant* consists of several files that describe the constant properties of the problem, for instance the gravitational acceleration, the viscosity of the fluid, the structure geometry and mesh information. The folder also contains the files containing information about the waves and relaxation zones.

The *system* folder contains the information that controls how the problem is solved, for instance what kind of interpolation, integration and solver that are used. In addition, it contains the files that control the time step, the simulation time and how the meshing is performed.

After running the code and solving the problem, additional directories with results are saved. Figure A.2 shows some of these files.



Figure A.2: Overview of programming set-up with result folders

The wave elevation information and the forces on the structure at each time step are found in the *postProcessing* folder. The *writeInterval i* folders contain the field solutions at the defined times. How many of these directories that appear are controlled by the *writeControl* option, described in 3., defines how often the results are saved.

# A.2 Boundary Conditions and Explanation

Surface	Velocity BC	Pressure BC	Volume fraction BC
inlet	waveVelocity	zeroGradient	waveAlpha
outlet	fixedValue	zeroGradient	zeroGradient
bottom	fixedValue	zeroGradient	zeroGradient
atmosphere	pressureInletOutletVelocity	totalPressure	inletOutlet
front	empty	empty	empty
back	empty	empty	empty
hull	fixedValue	zeroGradient	zeroGradient

Table A.1: Boundary Conditions

Boundary condition	Description
waveVelocity	Special bc connected to waves2Foam toolbox
waveAlpha	Special bc connected to waves2Foam toolbox
fixedValue	Specified by a value
pressureInletOutletVelocity	Allow some inflow with unknown velocity
empty	For 2D cases
fixedFluxPressure	Adjusts pressure gradient to match velocity BC
zeroGradient	Normal gradient is zero
totalPressure	Fixed value condition from specified total pressure
inletOutlet	zeroGradient if fluid flows out of domain and fixedValue into domain

 Table A.2: Boundary Condition Description (CFD Direct; 2016)

# **Appendix B**

# **Additional Simulation Results**

## **B.1 Regular Waves**









**Figure B.2:** *Regular waves: wave elevation analysis for* H = 7.5m



**Figure B.3:** *Regular waves: force time series for* H = 4.5m

#### **B.1.2** Force Results



**Figure B.4:** *Regular waves: force spectrum for* H = 4.5m



**Figure B.5:** *Regular waves: force time series for* H = 7.5m



**Figure B.6:** *Regular waves: force spectrum for* H = 7.5m

# **B.2** Bi-Chromatic Waves

#### **B.2.1** Wave Elevation Results



**Figure B.7:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: wave elevation time series for* H = 4.5m



**Figure B.8:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: wave elevation time series for* H = 7.5m

### **B.2.2** Force Results



**Figure B.9:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: force time series for* H = 4.5m



**Figure B.10:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: force spectrum for* H = 4.5m


**Figure B.11:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: force time series for* H = 7.5m



**Figure B.12:** *Bi-chromatic waves with*  $d\omega = 0.07$ *: force spectrum for* H = 7.5m

### **B.3** Inviscid Flow Case

### **B.3.1** Wave Elevation Results



**Figure B.13:** Inviscid flow case with  $d\omega = 0.05$ : wave elevation time series and spectrum for H = 4.5m



**Figure B.14:** *Inviscid flow case with*  $d\omega = 0.05$ *: wave elevation time series and spectrum for* H = 7.5m

### **B.3.2** Force Results



**Figure B.15:** *Inviscid flow case with*  $d\omega = 0.05$ *: force time serie for* H = 4.5m



**Figure B.16:** *Inviscid flow case with*  $d\omega = 0.05$ *: force spectrum for* H = 4.5m



**Figure B.17:** *Inviscid flow case with*  $d\omega = 0.05$ *: force time series for* H = 7.5m



**Figure B.18:** *Inviscid flow case with*  $d\omega = 0.05$ *: force spectrum for* H = 7.5m

# **Appendix C**

## **MATLAB Scripts**

## C.1 Case Generation Script

**Bi-Chromatic Wave Case** 

```
1 \% Prepares bi-chromtic wave cases to be simulated in openFOAM
2 % Based on script made by Babak Ommani
3
4 function generateRun()
       workPath =...
5
  '/Users/julie/Documents/Marin5/Delt/runGeneration/MasterCases/Bichromatic';
6
       sourcePath = '/Users/julie/Documents/Marin5/Delt/runGeneration';
7
       templatePath = [sourcePath '/' 'projectfilesBodyBi'];
8
9
10 %% CASE PARAMETERS
11 T1 = 12; % Period 1
12 H=[4.5 6 7.5]; % Wave height
13 draft = 20;
              % Body breadth
14 B = 12;
15 dw = 0.05;
                 % Difference frequency
16
17 for i=1:3
18 omega1 = 2*pi/T1;
19 omega2 = omega1 - dw;
20 T2 = 2*pi/omega2; %Period 2
21 frequency1 = 1/T1;
22 frequency2 = 1/T2;
23 g = 9.81;
24 k1 = \text{omega1^2/g};
```

```
k2 = omega2^2/g;
26 \quad lambda1 = (2*pi)/k1;
27 lambda2 = (2*pi)/k2;
28 steep1_ka(i) = k1*(H(i)/2);
29 steep1_lam(i) = H(i)/lambda1;
30 steep2_ka(i) = k2*(H(i)/2);
31 steep2_lam(i) = H(i)/lambda2;
32 depth = 0.5*lambda2;
33 \text{ air} = (depth/0.7) - depth;
34 tankL = 11*lambda2;
35 sizeCells = lambda2/20;
36 nCellXs = round(tankL/sizeCells);
37 nCellZs = round((depth+air)/(tankL/nCellXs));
38 dampstart = 0.82*tankL; %DAMPING AREA
39 damp_Yn = -1;
40 damp_Yp = 1;
41 inletDamp = 400;
42
43 endtime = 800;
44 vel_g = g/(2*omega1);
                                             % Group speed
45 vel_ph = g/(omega1);
                                             % Phase velocity
46
47 %stl file
48 y N e g = -5;
49 yPos=5;
50 LeftX=0.7*tankL;
51 rightX=0.7*tankL+B;
52 midX = LeftX + (B/2);
53 lowZ = -20;
54 midZ=0;
55 highZ(i)=2*H(i)+H(i)/2;
56 rotmomX = midX;
57 \text{ rotmomY} = 0.15;
58 rotmomZ = 0;
59
60 %Snappyhexmesh
61 fsboxZlow = lowZ-4;
62 fsboxZhigh(i) = highZ(i)+2;
63 fsboxX = tankL+5;
64 fsrefine = 3;
65 brefone = 3;
66 breftwo = 3;
67 bodyXmin = LeftX-5;
68 bodyXmax = rightX+5;
69 bodyZmin = lowZ-2;
70 bodyZmax(i) = highZ(i)+1;
```

```
71
72 %Probe in front and back of body
   front=midX - (2*B);
73
   back=midX+(2*B);
74
75
   %Coords for porbes
76
77 one = 0.2 * tankL;
   two = 0.5 * tankL;
78
   three = 0.7 * \text{tankL};
79
80 four = LeftX - B;
   five = rightX + B;
81
   six = 0.79 * tankL;
82
   seven = 0.9 *tankL;
83
84
   end
85
86
   %%
        for ind = 1:3
87
            %input parameters
88
89
            inputs.nCellX = nCellXs;
            inputs.L = tankL;
90
91
            inputs.D = depth;
            inputs.f = frequency1;
92
            inputs.wH = H;
93
            inputs.domend = air;
94
            inputs.endt = endtime;
95
            inputs.yNeg = yNeg;
96
97
            inputs.yPos = yPos;
            inputs.LeftX = LeftX;
98
            inputs.rightX = rightX;
99
            inputs.midX = midX;
100
            inputs.lowZ = lowZ;
101
102
            inputs.midZ = midZ;
            inputs.highZ = highZ(ind);
103
            inputs.fsboxZlow = fsboxZlow;
104
            inputs.fsboxZhigh = fsboxZhigh(ind);
105
            inputs.fsboxX = fsboxX;
106
107
            inputs.fsrefine = fsrefine;
            inputs.brefone = brefone;
108
            inputs.breftwo = breftwo;
109
            inputs.bodyXmin = bodyXmin;
110
            inputs.bodyXmax = bodyXmax;
111
            inputs.bodyZmin = bodyZmin;
112
            inputs.bodyZmax = bodyZmax(ind);
113
            inputs.nCellZ = nCellZs;
114
            inputs.front = front;
115
            inputs.back = back;
116
```

```
inputs.rotmomX = rotmomX;
117
            inputs.rotmomY = rotmomY;
118
            inputs.rotmomZ = rotmomZ;
119
            inputs.dampstart = dampstart;
120
121
            inputs.dampYn = damp_Yn;
            inputs.dampYp = damp_Yp;
122
            inputs.one = one;
123
124
            inputs.two = two;
            inputs.three = three;
125
            inputs.four = four;
126
            inputs.five = five;
127
            inputs.six = six;
128
            inputs.seven = seven;
129
130
            inputs.T1 = T1;
            inputs.T2 = T2;
131
132
            inputs.inletDamp = inletDamp;
            inputs.H = H(ind);
133
134
135
            caseName = ['caseBi' num2str(ind),'Inviscid'];
136
137
            casePath = [workPath '/' caseName];
            mkdir(casePath);
138
            copyfile([templatePath '/*'], casePath);
139
140
            applyInput(casePath, inputs);
141
        end
142
143
    end
144
    function applyInput(casePath, inputs)
145
146
        %-----
147
        % block Mesh dic file
148
        file = fullfile(casePath,'constant','polyMesh/blockMeshDict');
149
        params={};
150
151
        param.name = '__nCellX__';
152
153
        param.value = inputs.nCellX;
154
        params{end+1} = param;
155
156
        param.name = '__L__';
        param.value = inputs.L;
157
        params{end+1} = param;
158
159
        param.name = '__domend__';
160
        param.value = inputs.domend;
161
        params{end+1} = param;
162
```

```
163
        param.name = '__D__';
164
        param.value = - inputs.D;
165
        params{end+1} = param;
166
167
        param.name = '__nCellZ__';
168
        param.value = inputs.nCellZ;
169
        params{end+1} = param;
170
171
172
        replaceParams(file,params);
        %-----
                                     ------
173
        %stl file
174
        file = fullfile(casePath,'constant','triSurface/hull.stl');
175
176
        params={};
177
178
        param.name = '__yPos__';
        param.value = inputs.yPos;
179
        params{end+1} = param;
180
181
        param.name = '__yNeg__';
182
183
        param.value = inputs.yNeg;
        params{end+1} = param;
184
185
186
        param.name = '__LeftX__';
        param.value = inputs.LeftX;
187
        params{end+1} = param;
188
189
190
        param.name = '__midX__';
        param.value = inputs.midX;
191
        params{end+1} = param;
192
193
        param.name = '__rightX__';
194
        param.value = inputs.rightX;
195
        params{end+1} = param;
196
197
        param.name = '__lowZ__';
198
199
        param.value = inputs.lowZ;
        params{end+1} = param;
200
201
        param.name = '__midZ__';
202
        param.value = inputs.midZ;
203
        params{end+1} = param;
204
205
        param.name = '__highZ__';
206
        param.value = inputs.highZ;
207
        params{end+1} = param;
208
```

#### APPENDIX C. MATLAB SCRIPTS

```
209
210
        replaceParams(file,params);
        %-----
211
212
        %snappyHexMesh file
213
        file = fullfile(casePath,'system','snappyHexMeshDict');
        params={};
214
215
216
        param.name = '__fsboxZlow__'; %for blocks
        param.value = inputs.fsboxZlow;
217
        params{end+1} = param;
218
219
        param.name = '__fsboxZhigh__'; %for blocks
220
        param.value = inputs.fsboxZhigh;
221
222
        params{end+1} = param;
223
224
        param.name = '__L__'; %for one block
        param.value = inputs.L;
225
        params{end+1} = param;
226
227
        param.name = '__fsrefine__';
228
229
        param.value = inputs.fsrefine;
        params{end+1} = param;
230
231
232
        param.name = '__brefine1__';
        param.value = inputs.brefone;
233
        params{end+1} = param;
234
235
        param.name = '__brefine2__';
236
        param.value = inputs.breftwo;
237
        params{end+1} = param;
238
239
        param.name = '__bodyXmin__';
240
        param.value = inputs.bodyXmin;
241
        params{end+1} = param;
242
243
        param.name = '__bodyZmin__';
244
245
        param.value = inputs.bodyZmin;
        params{end+1} = param;
246
247
248
        param.name = '__bodyXmax__';
        param.value = inputs.bodyXmax;
249
        params{end+1} = param;
250
251
        param.name = '__bodyZmax__';
252
        param.value = inputs.bodyZmax;
253
        params{end+1} = param;
254
```

```
255
       replaceParams(file,params);
256
       %-----
257
                                 -----
       % waveProperties.input
258
       file = fullfile(casePath,'constant','waveProperties.input');
259
       params={};
260
261
       param.name = '__D__';
262
       param.value = inputs.D;
263
       params{end+1} = param;
264
265
       param.name = '__damping__';
266
       param.value = inputs.dampstart;
267
268
       params{end+1} = param;
269
270
       param.name = '__dampYn__';
       param.value = inputs.dampYn;
271
       params{end+1} = param;
272
273
274
275
       param.name = '__dampYp__';
       param.value = inputs.dampYp;
276
       params{end+1} = param;
277
278
       param.name = '__L__';
279
       param.value = inputs.L;
280
       params{end+1} = param;
281
282
       param.name = '__T1__';
283
       param.value = inputs.T1;
284
       params{end+1} = param;
285
286
       param.name = '__T2__';
287
       param.value = inputs.T2;
288
       params{end+1} = param;
289
290
291
       param.name = '__inletDamp__';
292
       param.value = inputs.inletDamp;
       params{end+1} = param;
293
294
       param.name = '__H__';
295
       param.value = inputs.H;
296
       params{end+1} = param;
297
298
299
       replaceParams(file,params);
       %------
300
```

```
% controldict
301
        file = fullfile(casePath,'system','controlDict');
302
        params={};
303
304
305
        param.name = '__end__';
        param.value = inputs.endt;
306
        params{end+1} = param;
307
308
        param.name = '__midX__';
309
        param.value = inputs.midX;
310
        params{end+1} = param;
311
312
        param.name = '__midZ__';
313
314
        param.value = inputs.midZ;
        params{end+1} = param;
315
        replaceParams(file,params);
316
317
        param.name = '__rotmomX__';
318
319
        param.value = inputs.rotmomX;
        params{end+1} = param;
320
321
        param.name = '__rotmomY__';
322
        param.value = inputs.rotmomY;
323
324
        params{end+1} = param;
325
        param.name = '__rotmomZ__';
326
        param.value = inputs.rotmomZ;
327
        params{end+1} = param;
328
329
        replaceParams(file,params);
330
        %-----
331
        % coords
332
        file = fullfile(casePath, 'postProcessing', 'coords');
333
        params={};
334
335
        param.name = '__one__';
336
337
        param.value = inputs.one;
338
        params{end+1} = param;
339
340
        param.name = '__two__';
        param.value = inputs.two;
341
        params{end+1} = param;
342
343
        param.name = '__three__';
344
        param.value = inputs.three;
345
        params{end+1} = param;
346
```

```
347
        param.name = '__four__';
348
        param.value = inputs.four;
349
        params{end+1} = param;
350
351
        param.name = '__five__';
352
        param.value = inputs.five;
353
        params{end+1} = param;
354
355
        param.name = '__six__';
356
        param.value = inputs.six;
357
        params{end+1} = param;
358
359
360
        param.name = '__seven__';
        param.value = inputs.seven;
361
        params{end+1} = param;
362
363
        replaceParams(file,params);
364
365
    end
366
367
    function replaceParams(filePath, params)
368
369
370
        txt = fileread(filePath);
371
        for ind = 1:length(params)
372
373
             param = params{ind};
374
             txt = strrep(txt, param.name,num2str(param.value));
375
        end
376
        fid = fopen(filePath,'wt');
377
        fprintf(fid, '%s', txt);
378
        fclose(fid);
379
380
381
   end
```

### C.2 Post-Processing Scripts

### C.2.1 FFT of Wave Elevation

```
    % Script to perform FFT on the wave elevation time series and plot the
    % Time series together with the spectra
```

```
4 close all;
5 clear;
6 clc;
7 %SET PARAMETERS
8 wavemax=12;
9 wavemin=-12;
10 numProbes=7;
11 legend1={};
12 legend2={};
13 legend3={};
14
   %IMPORT WAVEDATA FROM FILES
15
16 for i=1:numProbes
17
           name=['probe',num2str(i)];
           pathname=['elevationVsTime', num2str(i-1)];
18
           data.(name) = importdata(pathname);
19
       depth(i)=mean(data.(name)(:,2));
20
   end
21
22
23 %SET WAVE PARAMETERS
24 \quad w = 0.05;
25 \ \% w = 0.07;
                                    % MUST CHANGE WITH CASE
26 \text{ omega} = [0.05 \ 0.5236 \ 0.4736];
27 %omega = [0.07 0.5236 0.4536]; % MUST CHANGE WITH CASE
28 Tw = 2 * pi / w;
                                    % MUST CHANGE WITH CASE
29 g =9.81;
30 numT = 4;
                                    % Number of periods to use in validation
31 Tstart = 2.22; %4.05;
                                    % Number periods at interval start
32 delT=0.001;
                                    % Timestep in interpolation
33 dt=0.01;
                                    % Time step used in OpenFOAM simulation
                                    % Time in sekunder
34 t_start_sec=Tstart*Tw;
35 t_start=t_start_sec*1/dt;
                                    % Time in "elements"
                                    % Delta time in seconds
36 t_delta_sec=numT*Tw;
37 t_delta=t_delta_sec*1/dt;
                                    % Delta time in elements
38 t_end_sec=round(t_start_sec+t_delta_sec,1); % End time seconds
39 t_end=t_start+t_delta;
                                    % End time in elements
40
41 %RUN AND PLOT
  for i=4
42
    name=['probe',num2str(i)];
43
    m=1/dt;
44
45
    t=data.(name)(t_start:t_end,1);
46
    y=data.(name)(t_start:t_end,2) - mean(data.(name)(t_start:t_end,2));
47
    newt = t_start_sec:delT:t_end_sec;
48
    newy = interp1(t,y,newt);
49
```

```
newy(isnan(newy)) = 0;
50
    newL=length(newt)*1;
51
    newFs=1/delT;
52
    freq=2*pi*((0:newL-1)*newFs/newL);
53
54
    x=fft(newy);
    mag=abs(x);
55
56
    if i = = 4
57
       fig = figure;
58
59
       hold on;
60
       set(gcf, 'PaperUnits', 'centimeters');
61
       paperSize = [18 1.4*18*sqrt(2)/3]*2;
62
63
       figureSize = [0 0 paperSize(1) paperSize(2)-0.01];
       axesfontSize = 14;
64
65
       subplot( 2, 1, 2 )
66
       plot(freq(1:newL/2),(2/newL*mag(1:newL/2)));
67
68
       title('Single Sided Amplitude Spectrum of \eta(t)')
69
70
       h = zeros(6, 1);
       h(1)=line([omega(1) omega(1)], [0 4], 'Color', 'm'); % line through freq
71
       h(2)=line([omega(2) omega(2) ], [0 4], 'Color', 'r');
72
       h(3)=line([omega(3) omega(3)], [0 4], 'Color', 'b');
73
       h(4)=line([2*omega(2) 2*omega(2)], [0 4], 'Color',...
74
            [204/255 \ 102/255 \ 0/255]);
75
       h(5)=line([2*omega(3) 2*omega(3)], [0 4], 'Color', 'c');
76
77
       h(6)=line([omega(2)+omega(3) omega(2)+omega(3)], [0 4], 'Color', 'k');
       h(7)=line([2*omega(2)-omega(3) 2*omega(2)-omega(3)], [0 4],...
78
            'Color','y');
79
       h(8)=line([2*omega(3)-omega(2) 2*omega(3)-omega(2)], [0 4],'Color',...
80
            [0/255 102/255 0/255]);
81
       legend1 = ['\omega = ', num2str(omega(1))];
82
       legend2 = ['\omega_1 = ', num2str(omega(2))];
83
       legend3 = ['\omega_2 = ', num2str(omega(3))];
84
       legend4 = ['2 \circ mega_1'];
85
       legend5 = ['2 \ omega_2'];
86
       legend6 = ['\omega_1+\omega_2'];
87
       legend7 = ['2 \otimes 2 \otimes 2'];
88
       legend8 = ['2\omega_2-\omega1'];
89
       legObj=legend(h,legend1,legend2,legend3,legend4,legend5,legend6,...
90
            legend7, legend8);
91
       legObj.FontSize=14;
92
       xlabel('Frequency [rad/s]')
93
       ylabel('|\eta|')
94
       xlim([0 1.4])
95
```

```
ylim([0 1.1*max(2/newL*mag(1:newL/2))])
96
        box on:
97
        set( gca, 'FontSize', axesfontSize );
98
99
        subplot( 2, 1, 1)
100
        plot(data.probe4(:,1),data.probe4(:,2))
101
        h = zeros(2, 1);
102
        h(1)=line([t_start_sec t_start_sec], [-10 10],'Color','g');
103
        h(2)=line([t_end_sec t_end_sec], [-10 10], 'Color', 'r');
104
        legend1 = ['Interval Start'];
105
        legend2 = ['Interval End'];
106
        legObj=legend(h,legend1,legend2);
107
        legObj.FontSize=14;
108
109
        title('Time Series of Wave Elevation')
        xlabel('Time [s]')
110
111
        ylabel('wave elevation, \eta')
        ylim([wavemin wavemax])
112
        xlim([0 data.probe1(end,1)])
113
114
        grid;
        box on;
115
116
        set( gca, 'FontSize', axesfontSize );
117
        plotname =...
118
119
        '/Users/julie/Documents/Marin5/Master/Master-set-up/fig/H7/dw005FFT';
        set(gcf,'PaperSize',paperSize*1.05)
120
        set(gcf,'PaperPosition',figureSize*1.05)
121
        set(gca, 'LooseInset', get(gca, 'TightInset'));
122
123
        print(gcf,'-dpng', plotname);
124
     end
125
     end
```

### C.2.2 FFT of Force Data

```
1 % Performs FFT on the force results and plots the firce time series and the
   % force spectra
2
3
4
  close all;
5 clear;
6 clc;
7 fourierData={};
8 \text{ hAll} = [4.5 6 7.5 8]; \%j
   zetaAll=[2.25 3 3.75 4];
9
   progressBar=0;
10
11
   output={};
12
```

```
13 for kk=1:2;
       if kk==1
14
            w = [0.05 0.5236 0.4736]; %i
15
        elseif kk==2
16
            w = [0.07 \ 0.5236 \ 0.4536];
17
       else
18
            disp('error1')
19
20
       end
       for j=1:3;
21
            H=hAll(j);
22
            for i=1;
23
                %close all;
24
25
26
                B = 12;
                d = 20;
27
                L = 0.1;
28
                zeta_A = zetaAll(j); % NOMINAL AMPLITUDE
29
                if w(1) == 0.05
30
                     nTStart=2.22;
31
                 elseif w(1) == 0.07
32
33
                     nTStart=4.05;
                 else
34
                     disp('error');
35
36
                 end
37
                numT = 4.00; %number of periods in fourier analysis
38
39
                rho = 1000.0;
40
                 g = 9.81;
41
                Tw = 2.0*pi/w(1);
42
                k = (w(1)^2.0)/g;
43
                 % Load data, wave loads
44
45
                 filename=['BiH', num2str(floor(H)), 'dw00', num2str(w(1)*100),...
46
                     't001'];
47
                 output= [output; filename num2str(i) num2str(j) num2str(kk)];
48
49
                run(filename);
50
51
                t0 = time(4:end);
52
                dt0 = t0(2) - t0(1);
53
                fx0 = fpx(4:end) + fvx(4:end);
54
                fz0 = fpz(4:end)+fvz(4:end) - rho*g*d*B*L;
55
                my0 = mpy(4:end) + mvy(4:end);
56
57
                % Harmonic components
58
```

59	% Analysis window
60	t_start= nTStart*Tw;
61	$t_end = t_start + numT*Tw; %+0.1*dt;$
62	<pre>t_end = round(t_end);</pre>
63	<pre>delT = 0.001; %timestep in interpolation</pre>
64	
65	% FORCE X (SURGE)
66	<pre>newt = t_start:delT:t_start+(numT*Tw);</pre>
67	<pre>newfx = interp1(t0,fx0,newt);</pre>
68	<pre>newLx = length(newt)*1;</pre>
69	<pre>newFsx = 1/delT;</pre>
70	<pre>freq = 2*pi*((0:newLx-1)*newFsx/newLx);</pre>
71	<pre>fftx = fft(newfx,length(freq));</pre>
72	<pre>magx = abs(fftx);</pre>
73	<pre>[ampx, I] = max(2/newLx*magx);</pre>
74	<pre>Fq = freq(I);</pre>
75	
76	% FORCE Z (HEAVE)
77	name=['fz'];
78	<pre>newt = t_start:delT:t_start+(numT*Tw);</pre>
79	<pre>newfz = interp1(t0,fz0,newt);</pre>
80	<pre>newLz=length(newt);</pre>
81	newFsz=1/delT;
82	freq=2*pi*((0:newLz-1)*newFsz/newLz);
83	<pre>fftz=fft(newfz,length(freq));</pre>
84	<pre>magz=abs(fftz);</pre>
85	<pre>[ampz , I]=max(2/newLz*magz);</pre>
86	Fq=freq(I);
87	
88	% MOMENT Y (PITCH)
89	name=['my'];
90	<pre>newt = t_start:delT:t_start+(numT*Tw);</pre>
91	<pre>newmy = interp1(t0,my0,newt);</pre>
92	<pre>newLy=length(newt)*1;</pre>
93	newFsy=1/delT;
94	<pre>freq=2*pi*((0:newLy-1)*newFsy/newLy);</pre>
95	<pre>fftmy=fft(newmy,length(freq));</pre>
96	<pre>magmy=abs(fftmy);</pre>
97	%magmy(1)=0;
98	<pre>[ampmy, I]=max(2/newLy*magmy);</pre>
99	Fq=freq(I);
100	
101	% FIGURES
102	<pre>fig=figure;</pre>
103	hold on;
104	<pre>set(gcf, 'PaperUnits', 'centimeters');</pre>

```
paperSize = [18 \ 1.4*18*sqrt(2)/3]*2;
105
                 figureSize = [0 0 paperSize(1) paperSize(2)-0.01];
106
                 axesfontSize = 14;
107
108
                 subplot( 3, 1, 1 )
109
                 plot(freq(1:floor(newLx/2)),(2/newLx*magx(1:floor(newLx/2))));
110
                 title('Surge Force')
111
                 h = zeros(6, 1);
112
                 h(1)=line([w(1) w(1)], [0 1.1*ampx], 'Color', 'm'); %plot line
113
                 h(2)=line([w(2) w(2)], [0 1.1*ampx], 'Color', 'r');
114
                 h(3) = line([w(3) w(3)], [0 1.1*ampx], 'Color', 'g');
115
                 h(4)=line([2*w(2) 2*w(2)], [0 1.1*ampx], 'Color',...
116
                     [204/255 102/255 0/255]); % []
117
118
                h(5)=line([2*w(3) 2*w(3)], [0 1.1*ampx], 'Color', 'c');
                 h(6)=line([w(2)+w(3) w(2)+w(3)], [0 1.1*ampx], 'Color', 'k');
119
                 h(7)=line([2*w(2)-w(3) 2*w(2)-w(3)], [0 1.1*ampx], 'Color', 'y');
120
                h(8)=line([2*w(3)-w(2) 2*w(3)-w(2)], [0 1.1*ampx], 'Color',...
121
                     [0/255 102/255 0/255]);
122
                 legend1 = ['\omega = ', num2str(w(1))];
123
                 legend2 = [' \circ mega_1 = ', num2str(w(2))];
124
125
                 legend3 = ['\omega_2 = ', num2str(w(3))];
                 legend4 = '2\omega_1';
126
                 legend5 = '2 \setminus omega_2';
127
128
                 legend6 = '\omega_1+\omega_2';
                 legend7 = ['2\omega_1-\omega_2'];
129
                 legend8 = ['2\omega_2-\omega1'];
130
                 legObj=legend(h,legend1,legend2,legend3,legend4,legend5,...
131
                     legend6,legend7,legend8);
132
                 legObj.FontSize=axesfontSize;
133
                 xlabel('Frequency [rad/s]', 'FontSize', axesfontSize)
134
                 ylabel( 'F_x [N]', 'FontSize', axesfontSize );
135
                 xlim([0 1.3])
136
137
                 vlim([0 1.1*ampx])
                 box on:
138
139
                 set( gca, 'FontSize', axesfontSize );
140
141
                 subplot( 3, 1, 2)
142
                 plot(freq(1:floor(newLz/2)),(2/newLz*magz(1:floor(newLz/2))));
143
                 title('Heave Force')
144
                 h = zeros(8, 1);
145
                 h(1)=line([w(1) w(1)], [0 1.1*ampz], 'Color', 'm'); %plot line
146
                 h(2)=line([w(2) w(2)], [0 1.1*ampz], 'Color', 'r');
147
                h(3)=line([w(3) w(3)], [0 1.1*ampz], 'Color', 'g');
148
                 h(4)=line([2*w(2) 2*w(2)], [0 1.1*ampz], 'Color',...
149
                     [204/255 102/255 0/255]);
150
```

```
h(5)=line([2*w(3) 2*w(3)], [0 1.1*ampz], 'Color', 'c');
151
                 h(6)=line([w(2)+w(3) w(2)+w(3)], [0 1.1*ampz], 'Color', 'k');
152
                 h(7)=line([2*w(2)-w(3) 2*w(2)-w(3)], [0 1.1*ampz], 'Color', 'y');
153
                 h(8)=line([2*w(3)-w(2) 2*w(3)-w(2)], [0 1.1*ampz], 'Color',...
154
                      [0/255 \ 102/255 \ 0/255]);
155
                 legend1 = ['\omega = ', num2str(w(1))];
156
                 legend2 = [' \circ num2str(w(2))];
157
                 legend3 = ['\mbox{omega}_2 = ', num2str(w(3))];
158
                 legend4 = ['2 \setminus omega_1'];
159
                 legend5 = ['2 \setminus omega_2'];
160
                 legend6 = [' \circ ega_1 + ega_2'];
161
                 legend7 = ['2 \setminus omega_1 - \setminus omega_2'];
162
                 legend8 = ['2\omega_2-\omega1'];
163
164
                 legObj=legend(h,legend1,legend2,legend3,legend4,legend5,...
165
                      legend6,legend7, legend8);
166
                 legObj.FontSize=axesfontSize;
167
                 xlabel('Frequency [rad/s]', 'FontSize', axesfontSize)
168
                 ylabel( 'F_z [N]', 'FontSize', axesfontSize );
169
                 xlim([0 1.3])
170
171
                 ylim([0 1.1*ampz])
                 box on;
172
173
174
                 set( gca, 'FontSize', axesfontSize );
175
                 subplot( 3, 1, 3)
176
                 plot(freq(1:floor(newLy/2)),(2/newLy*magmy(1:floor(newLy/2))));
177
                 title('Pitch Moment')
178
179
                 h = zeros(6, 1);
180
                 h(1)=line([w(1) w(1)], [0 1.1*ampmy], 'Color', 'm'); %plot line
181
                 h(2)=line([w(2) w(2)], [0 1.1*ampmy], 'Color', 'r');
182
183
                 h(3)=line([w(3) w(3)], [0 1.1*ampmy], 'Color', 'g');
                 h(4)=line([2*w(2) 2*w(2)], [0 1.1*ampmy], 'Color',...
184
                      [204/255 \ 102/255 \ 0/255]);
185
                 h(5)=line([2*w(3) 2*w(3)], [0 1.1*ampmy], 'Color', 'c');
186
                 h(6)=line([w(2)+w(3) w(2)+w(3)], [0 1.1*ampmy], 'Color', 'k');
187
                 h(7)=line([2*w(2)-w(3) 2*w(2)-w(3)], [0 1.1*ampmy], 'Color',...
188
                 'y');
189
190
                 h(8)=line([2*w(3)-w(2) 2*w(3)-w(2)], [0 1.1*ampmy], 'Color',...
                      [0/255 \ 102/255 \ 0/255]);
191
                 legend1 = ['\omega = ', num2str(w(1))];
192
                 legend2 = [' \setminus omega_1 = ', num2str(w(2))];
193
                 legend3 = ['\omega_2 = ', num2str(w(3))];
194
                 legend4 = ['2 \setminus omega_1'];
195
                 legend5 = ['2 \ omega_2'];
196
```

```
legend6 = ['\omega_1+\omega_2'];
197
                 legend7 = ['2 \land omega_1 - \land omega_2'];
198
                 legend8 = ['2\omega_2-\omega1'];
199
                 legObj=legend(h,legend1,legend2,legend3,legend4,legend5,...
200
                     legend6,legend7,legend8);
201
                 legObj.FontSize=axesfontSize;
202
                 xlabel('Frequency [rad/s]', 'FontSize', axesfontSize)
203
                 ylabel( 'My [Nm]', 'FontSize', axesfontSize );
204
                 xlim([0 1.3])
205
                 ylim([0 1.1*ampmy])
206
                 box on;
207
                 set( gca, 'FontSize', axesfontSize );
208
209
210
                 plotname=...
      ['/Users/julie/Documents/Marin5/Master/Master-set-up/fig/fftForceBiH',...
211
        num2str(floor(H)),'dw00',num2str(w(1)*100),'t001'];
212
                 set(gcf, 'PaperSize', paperSize*1.05)
213
                 set(gcf,'PaperPosition',figureSize*1.05)
214
                 set(gca, 'LooseInset', get(gca, 'TightInset'));
215
                 print(gcf,'-dpng', plotname);
216
217
                 progressBar=progressBar+1;
                 progress=round(progressBar/6*100);
218
                 clc; %clearing "chat"
219
                 loadingText=['Simulation is ',num2str(progress),'% done'];
220
                 disp(loadingText);
221
                 hold off;
222
                 close all;
223
            end
224
225
        end
   end
226
   clc;
227
   disp('Complete')
228
```

#### C.2.3 Fourier Analysis of Force Data

```
1 % Script to perform Fourier analysis on the force results
2 % Based on the script made by Finn-Christian Hanssen
3 % Uses the function intergral values.m and fourier_comp.m
4 % Outputs a matrix with all the dimensionless force coefficients
5
6 clear all; clc; close all;
7
8 %Basic data
9 fourierData={};
10 fourierDataThird={};
```

```
11 hAll = [4.5 \ 6 \ 7.5];
                             %Wave heigt, index j
   zetaAll=[2.25 3 3.75]; %Wave Amp
12
   progressBar=0;
13
14
15
  for kk=1:2;
        if kk == 1
16
            w = [0.05 \ 0.5236 \ 0.4736]; \% frequencies, index i
17
        elseif kk==2
18
            w = [0.07 \ 0.5236 \ 0.4536];
19
20
        else
            disp('error1')
21
       end
22
       for j=1:3;
23
24
            H=hAll(j);
            for i=1:3;
25
26
                                      % Breadth of column
                B = 12;
27
                d = 20;
                                      % Draft of column
28
29
                L = 0.1;
                                      % Domain size in y - thickness of body
                zeta_A = zetaAll(j);% NOMINAL AMPLITUDE
30
31
                if w(1) == 0.05
                     nTStart=2.22;
                                      %Start of interval for diff. freq 0.05
32
                elseif w(1) = = 0.07
33
34
                     nTStart=4.05; %Start of interval for diff. freq 0.07
                else
35
                     disp('error');
36
                end
37
38
                n_Tw = 4.00;
                                      %Number of periods in fourier analysis
39
40
                rho = 1000.0;
41
                g = 9.81;
42
                Tw = 2.0*pi/w(1);
43
                k = (w(i)^2.0)/g;
44
45
                % Load data, wave loads
46
47
                filename=...
                ['BiH', num2str(floor(H)), 'dw00', num2str(w(1)*100), 't001.m'];
48
49
                run(filename); %Runs script collecting force data from file
50
51
                t0 = time(4:end);
52
                dt0 = t0(2) - t0(1);
53
                fx0 = fpx(4:end) + fvx(4:end);
54
                fz0 = fpz(4:end)+fvz(4:end);
55
                my0 = mpy(4:end) + mvy(4:end);
56
```

```
57
58
                % Resample data . interpolate
                t = t0(1):(dt0/10):t0(end);
59
                dt = t(2) - t(1);
60
                t = t';
61
                fx = interp1(t0, fx0, t);
62
                fz = interp1(t0, fz0, t) - rho*g*d*B*L;
63
                my = interp1(t0, my0, t);
64
65
                % Harmonic components
66
                % Analysis window
67
                t1 = nTStart * Tw;
68
                t2 = t1 + n_Tw * Tw + 0.1 * dt;
69
70
                % Zeroth order
71
                n = 0;
72
                [ a0.fx, b0.fx ] = fourier_comp( n, t, fx, w(i), t1, t2 );
73
                [ a0.fz, b0.fz ] = fourier_comp( n, t, fz, w(i), t1, t2 );
74
75
                [ a0.my, b0.my ] = fourier_comp( n, t, my, w(i), t1, t2 );
                a0.fx = a0.fx/2.0;
76
77
                a0.fz = a0.fz/2.0;
                a0.my = a0.my/2.0;
78
79
80
                % First order
                n = 1:
81
                [ a1.fx, b1.fx ] = fourier_comp( n, t, fx, w(i), t1, t2 );
82
                [ a1.fz, b1.fz ] = fourier_comp( n, t, fz, w(i), t1, t2 );
83
                [ a1.my, b1.my ] = fourier_comp( n, t, my, w(i), t1, t2 );
84
85
                % Second order
86
                n = 2;
87
                [ a2.fx, b2.fx ] = fourier_comp( n, t, fx, w(i), t1, t2 );
88
                [ a2.fz, b2.fz ] = fourier_comp( n, t, fz, w(i), t1, t2 );
89
                [ a2.my, b2.my ] = fourier_comp( n, t, my, w(i), t1, t2 );
90
91
                % Third order
92
                n = 3;
93
                [ a3.fx, b3.fx ] = fourier_comp( n, t, fx, w(i), t1, t2 );
94
                [ a3.fz, b3.fz ] = fourier_comp( n, t, fz, w(i), t1, t2 );
95
96
                [ a3.my, b3.my ] = fourier_comp( n, t, my, w(i), t1, t2 );
                nn = 1;
97
                w1 = 2 * w(2) - w(3); %2 w1 - w2
98
                w^2 = 2 * w(3) - w(2); \% 2 w^2 - w^1
99
                [ a3w1.fx, b3w1.fx ] = fourier_comp( nn, t, fx, w1, t1, t2 );
100
                [ a3w1.fz, b3w1.fz ] = fourier_comp( nn, t, fz, w1, t1, t2 );
101
                [ a3w1.my, b3w1.my ] = fourier_comp( nn, t, my, w1, t1, t2 );
102
```

```
103
                [ a3w2.fx, b3w2.fx ] = fourier_comp( nn, t, fx, w2, t1, t2 );
104
                [ a3w2.fz, b3w2.fz ] = fourier_comp( nn, t, fz, w2, t1, t2 );
105
                [ a3w2.my, b3w2.my ] = fourier_comp( nn, t, my, w2, t1, t2 );
106
107
108
                % Nondimensional forces
109
                fd_x = a0.fx/(0.5*rho*g*L*zeta_A^2.0);
110
                fd_z = a0.fz/(0.5*rho*g*L*zeta_A^2.0);
111
112
                fd_my = a0.my/(0.5*rho*g*L*B*zeta_A^2.0);
113
                fx1 = sqrt(a1.fx^2 + b1.fx^2)/(rho*g*L*d*zeta_A);
114
                fz1 = sqrt( a1.fz^2 + b1.fz^2 )/( rho*g*L*B*zeta_A );
115
116
                my1 = sqrt(a1.my^2 + b1.my^2)/(rho*g*L*B*d*zeta_A);
                fxSlow = sqrt(a1.fx^2 + b1.fx^2)/(rho*g*L*zeta_A^2);
117
                fzSlow = sqrt(a1.fz^2 + b1.fz^2)/(rho*g*L*zeta_A^2);
118
                mySlow = sqrt(a1.my^2 + b1.my^2)/(rho*g*L*B*zeta_A^2);
119
120
121
                fx2 = sqrt(a2.fx^2 + b2.fx^2)/(rho*L*d*g*zeta_A);
                fz2 = sqrt(a2.fz^2 + b2.fz^2)/(rho*L*B*g*zeta_A);
122
                my2 = sqrt(a2.my^2 + b2.my^2)/(rho*g*L*B*d*zeta_A);
123
124
                fx3 = sqrt(a3.fx^2 + b3.fx^2)/(rho*L*d*g*zeta_A);
125
126
                fz3 = sqrt( a3.fz^2 + b3.fz^2 )/( rho*L*B*g*zeta_A );
                my3 = sqrt( a3.my^2 + b3.my^2 )/( rho*g*L*B*d*zeta_A );
127
                fx3w1 = sqrt( a3w1.fx^2 + b3w1.fx^2 )/( rho*L*d*g*zeta_A );
128
                fz3w1 = sqrt( a3w1.fz^2 + b3w1.fz^2 )/( rho*L*B*g*zeta_A );
129
                my3w1 = sqrt( a3w1.my^2 + b3w1.my^2 )/( rho*g*L*B*d*zeta_A );
130
                fx3w2 = sqrt(a3w2.fx^2 + b3w2.fx^2)/(rho*L*d*g*zeta_A);
131
                fz3w2 = sqrt(a3w2.fz^2 + b3w2.fz^2)/(rho*L*B*g*zeta_A);
132
                my3w2 = sqrt( a3w2.my^2 + b3w2.my^2 )/( rho*g*L*B*d*zeta_A );
133
134
135
                fourierData=[fourierData; fd_x fx1 fz1 my1 fx2 fz2 my2 fx3...
                    fz3 my3 fd_z fd_my fxSlow fzSlow mySlow i j kk];
136
                fourierDataThird=[fourierDataThird;...
137
                    fx3w1 fz3w1 my3w1 fx3w2 fz3w2 my3w2 i j kk];
138
                progressBar=progressBar+1;
139
                progress=round(progressBar/18*100);
140
141
                clc;
142
                loadingText=['Simulation is ',num2str(progress),'% done'];
                disp(loadingText);
143
144
            end
145
       end
   end
146
147
   fourierData = cell2mat(fourierData);
   fourierDataThird = cell2mat(fourierDataThird);
148
```

```
149 clc;
150 disp('Done');
```

1

### C.2.4 Input Function in fourier\_analysist001.m

```
1 % Made by Finn-Christian Hanssen
2 % Used by the script fourier_analysist001.m
3 % Uses the function integral_values.m
4
5 function [ an, bn ] = fourier_comp( n, t, y, w, t1, t2 )
6
7 % Determine vectors for integration
8 ind = find( ( t >= t1 ) & ( t <= t2 ) );</pre>
9 ti = t( ind );
10 yi = y( ind );
11 T = ti( end ) - ti( 1 );
12
13 for i = 1:length( ti )
      tt = ti( i );
14
      yy = yi(i);
15
       [ ian( i ), ibn( i ) ] = integral_values( n, w, tt, yy );
16
  end
17
18
19 % Perform trapeziodal integration
20 an = (2/T) * trapz(ti, ian);
21 bn = (2/T) * trapz(ti, ibn);
22
23 end
```

### C.2.5 Input Function in fourier\_comp.m

```
2 % Made by Finn-Christian Hanssen
3 % Input function to fourier_comp.m
4
5 function [ ia, ib ] = integral_values( n, w, t, y )
6 ia = y*cos( n*w*t );
7 ib = y*sin( n*w*t );
8 end
```

#### C.2.6 Script Used to Plot Mean Drift Force Coefficients

```
1 % Script to plot the mean drift force coefficients
2 % Uses Matrix output from fourier_analysis.m
3
   run fourier_analysist001.m
4
5
   close all;
6
  figure
7
8
   grid on
9
10 hold on
   colorVector=['b' 'r'];
11
12 lineVector={ '- ', '-.', ': '};
   legendString={};
13
14
  for kk=1:2;
15
       if kk == 1
16
           w = [0.05 \ 0.5236 \ 0.4736]; \%i
17
       elseif kk==2
18
19
           w = [0.07 \ 0.5236 \ 0.4536];
       else
20
21
           disp('error1')
22
       end
23 for i=1
24
   plotY=fourierData(9*(kk-1)+((i-1)+1:3:(i-1)+9),1);
   plotXentry=fourierData(9*(kk-1)+((i-1)+1:3:(i-1)+9),17); %track wave height
25
   heightVector=[2.25 3 3.75];
26
   plotX=[heightVector(plotXentry(1)) heightVector(plotXentry(2))...
27
28
       heightVector(plotXentry(3))];
29
   set(gcf, 'PaperUnits', 'centimeters')
30
   paperSize = [18 1.1*18*sqrt(2)/3]*2;
31
  figureSize = [0 0 paperSize(1) paperSize(2)-0.01];
32
33
  axesfontSize = 16;
34
   plotOption = [cellstr(lineVector(i)), colorVector(kk)];
35
   plotOption = strjoin(plotOption);
36
   plot(plotX,plotY,plotOption,'LineWidth',1.5)
37
   legendString=[legendString ['d\omega = ' num2str(w(1))]];
38
   legend(legendString,'Location', 'NorthEast');
39
40
41 xlim([2 4]);
42 %ylim([0.75 1]);
43
44 set(gca,'GridLineStyle',':')
45 set(gcf, 'PaperSize', paperSize*1.05)
46 set(gcf,'PaperPosition',figureSize*1.05)
```

```
47 set(gca, 'LooseInset', get(gca, 'TightInset'));
48 set(gca,'FontSize',axesfontSize)
49 prcentChangeMeanForceX(:,kk) = ((plotY(3)-plotY(1))/plotY(1))*100;
50 end
51 end
52
53 plotname =...
       '/Users/julie/Documents/Marin5/Master/Master-set-up/fig/MeanCoeffX';
54
55 legend(legendString);
56 xlabel('Wave Amplitude [m]');
57 ylabel('Fx_{mean}/0.5\rhogLA^2 [-]');
58 title('Mean Drift Force Coefficients in x-Direction');
59 box on;
60 print(gcf,'-dpng', plotname)
61 hold off
```