



Norwegian University of
Science and Technology

Possibilities and Implications by Designing an Aluminium Integrated Template Structure.

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Subsea Technology

Submission date: June 2017

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Preface

This master thesis is submitted to the Norwegian University of Science and Technology (NTNU) as a part of the master's degree program within Subsea Technology with specialization in operation and maintenance. The thesis was written at the Department of Geoscience and Petroleum, in the spring semester of 2017.

Acknowledgement

We would like to thank our supervisor Tor Berge Gjersvik for his engagement. And a special thanks to our co-supervisor Roy Johnson and Ole Terje Midling in Marine Aluminium AS for their interest in the project.

Trondheim, June 9, 2017



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Summary

Steel has historically been the governing material for use in subsea applications. This master thesis looks into the possibility of using aluminium rather than steel in subsea structures. A four-slots integrated template structure installed on the Gjøa field in the Norwegian Sea is used as design basis. The use of aluminium represents a lighter structure with a potential in reducing life cycle cost. The aluminium alloys 5083 and 6082 are the only ones approved for structural seawater applications by the NORSOK M-121 standard, where H116 and T6 are the strongest approved tempers, respectively. However, material properties has to be fully understood in order to utilize aluminium in a subsea environment. Especially with respect to corrosion behaviour, mechanical properties and joining methods. Bolting, metal inert gas and friction stir welding are identified as preferred joining methods.

The structure is redesigned utilizing appropriate alloys and joining methods. Load simulations are subsequently performed using finite element method, where the structure withstands design loads defined by industry standards with acceptable results. The main load types are drilling, trawling, tie-in and weight of equipment installed on the integrated template structure.

The utilization of aluminium in subsea structures looks very promising for the integrated template structure, although there is a need for more research, especially with respect to corrosion of aluminium buried in soil. A total weight reduction of 36% is achieved by utilizing aluminium. Total cost for the aluminium made structure is competitive to steel, where aluminium fabrication cost is the decisive factor.

Sammendrag

Stål har historisk vært det mest brukte materialet for bruk i undervannskonstruksjoner. Denne masteroppgaven ser på mulighetene for å bruke aluminium i stedet for stål i undervannsstrukturer. En fire-slots bunnramme installert på Gjøa feltet i Norskehavet er brukt som et design grunnlag. Bruk av aluminium resulterer i en lettere struktur med potensial i å redusere livssyklus kostnadene. Aluminiumslegeringene 5083 og 6082 er de eneste som er godkjent for strukturelt bruk i undervannsmiljøet ifølge NORSOK M-121 standarden, hvor H116 og T6 er de sterkeste temperamentene. En grundig forståelse for materialegenskapene må ligge til grunn for å kunne anvende aluminium i undervannsmiljøet. Spesielt med hensyn til korrosjonsadferd, mekaniske egenskaper og sammenføyningsmetoder. Bolting, metall inert gass og friksjons sveising er identifisert som foretrukne sammenføyningsmetoder.

Foretrukne sammenføyningsmetoder og de godkjente aluminiumslegeringene er benyttet for å redesigne strukturen. Last simulasjoner er gjennomført ved å anvende elementmetoden. Strukturen tåler påsatte belastninger definert av industri-standarder med akseptable resultater. Evaluerte last typer er boring, tråling, opp-kobling og vekt av installerte komponenter.

Å anvende aluminium i undervannsstrukturer ser lovende ut for den evaluerte bunnrammen, selv om det er behov for mer forskning, spesielt med hensyn til korrosjon av aluminium som er begravet i sedimenter. En total vektreduksjon på 36% er oppnådd ved å bruke aluminium i stedet for stål. Samlet sett er kostnaden for aluminiumskonstruksjonen konkurransedyktig med stål, hvor fabrikasjonskostnadene er avgjørende for hvorvidt stål eller aluminium fremstår som mest kostnadseffektivt.

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Abbreviations

AS Advancing Side

BM Base Material

CAD Computer Aided Design

CAPEX Capital Expenditure

CASA Center for Advanced Structural Analyses

CP Cathodic Protection

DAF Dynamic Amplification Factor

DT Drawn Tube

DNV Det Norske Veritas

DVS Deutscher Verband für Schweißtechnik

Eq Equation

ET Extruded Tube

FCC Face Centered Cubic

FEM Finite Element Method

FSW Friction Stir Welding

HAZ Heat Affected Zone

HB Brinell Hardness number

HE Hydrogen Embrittlement

HYB Hybrid metal extrusion and Bonding

HT Heat Treatable

HV Vickers Hardness

ISO International Organization for Standardization

ITS Integrated Template Structure

LW Laser Welding

MIG Metal Inert Gas

MOI Moment Of Inertia

NCS Norwegian Continental Shelf

Nd:YAG Neodymium: Doped Yttrium Aluminium Garnet

NHT Non Heat Treatable

NORSOK Norsk Søkels Konkurransesposisjon

NTNU Norges Teknisk-Naturvitenskapelige Universitet

OPEX Operating Expenditure

RFW Rotary Friction Welding

PL Plate

PTFE Poly Tetra Fluoro Ethylene

RS Retreating Side

SCE Saturated Calomel Electrode

Shell Royal Dutch Shell plc

SINTEF Stiftelsen for Industriell og Teknisk Forskning ved Norges Tekniske Høgskole

SKL Skew Load factor

Subsea 7 Subsea 7 SA

SZ Stir Zone

Te Metric Ton

TEC Thermal Expansion Coefficient

TIG Tungsten Inert Gas

TMAZ Thermal-Mechanically Affected Zone

TS Tensile Strength

TSA Thermal Sprayed Aluminium

VAT Value Added Tax

YS Yield Strength

X-mas Subsea christmas tree

Chapter 1

Introduction

1.1 Background

Oil companies are constantly looking for ways to reduce CAPEX (capital expenditure including installation cost) on field developments. Especially in the current market as the oil price has dropped significantly over the last couple of years (since 2014), putting an extra pressure on the field developments to reduce costs. This master thesis is based on the idea that the extensive use of aluminium in subsea structures may be beneficial to steel in terms of capital expenditures. Steel has been the governing material for use in subsea structures since the early days of subsea engineering, as it possess good and well-known mechanical properties and joining methods making it an easy choice. This study will focus on aluminium as a competitive alternative to steel.

1.2 Objective

The main objective of this study is to map aluminium properties related to subsea applications and to study design implications. Design possibilities will be based on one of the most extensively used integrated template structure (ITS) designs on the Norwegian Continental Shelf (NCS). A similar model of an aluminium integrated template structure will be build using appropriate software design tools. The main objectives are described in bullet points below.

- Map essential aluminium properties for subsea applications.

- Identify and describe possible joining methods for aluminium in a subsea environment.
- Identify the geometry and scale of an extensively used integrated template structure design.
- Redesign integrated template structure to suit aluminium's properties.
- Perform a cost comparison between steel and aluminium as construction materials for an integrated template structure.
- Identify research areas and areas with a need for further documentation.

1.3 Historical overview

Aluminium was introduced into the marine environment in the 1890's, and was mainly used in shipbuilding as an alternative to more conventional steel ships. Alfred Nobel's yacht "Mignon" was among the first ships to utilize an aluminium made hull structure [19]. Many of the earliest ships utilizing aluminium corroded heavily as the corrosion mechanisms for aluminium were not fully understood at the time [19]. Today, these technical difficulties have been solved or mitigated, and aluminium is now widely used in ship building [19].

1.3.1 Use of aluminium in the subsea environment

To date, most of the aluminium used in subsea equipment is only meant for short term exposure to the subsea environment, with a few exceptions. Aluminium made protection structures (hatches) were permanently installed on a satellite template structure on the Lille-Frigg field from 1991-2001. These hatches were protected by the use of sacrificial anodes and the aluminium structures were not in galvanic contact with the surrounding steel structures. According to Ole Terje Midling, a contact in Marine Aluminium, there was not observed any more degradation than expected after 10 years in service, meaning the design worked as expected [34]. Another example is the protection structures on the Gullfaks field installed in 2000 [35].

1.3.2 Selection of structure

An integrated template structure has been selected for all case studies in this thesis for several reasons. It is normally the heaviest structure to be installed, thus reducing the weight of the ITS provides the possibility to use smaller installation vessels with lower crane capacity. Another reason is the load scenario of an ITS, where the ITS is mainly statically loaded. Aluminium structures have lower fatigue properties compared to steel, which is the reason for choosing a statically loaded structure. More on aluminium's fatigue properties are presented in section 3.12. In addition, the operating temperature for aluminium alloys should generally not be higher than 80-100 °C [2], as described in section 3.2.1. Exceedance of the operating temperature may lead to unwanted material strength loss. The production fluids in the manifold and christmas tree (X-mas) may have a higher temperature than those mentioned above which is one of the reasons for not studying the use of aluminium on these structures in this thesis. The manifold and X-mas trees are important interfaces to the ITS, and will therefore be described in detail in chapter 2.

Chapter 2

Subsea Equipment

2.1 Integrated template structure

The ITS is a subsea structure with several important functions. The structure protects petroleum processing equipment from trawling (fishing gear) and dropped objects/impacts. It is also the base/foundation for the wells it is hosting. The ITS is kept in place using suction anchors to "suck" itself into the soil, or mud mats on hard subsea surfaces. The ITS model used in this study has 7.0 meter tall suction anchors with a diameter of ca. 5 meters. A figure of a suction anchor can be seen in figure 2.1. The length and diameter of the suction anchors depends on soil properties and the amount of weight to support.

This study will focus on a four-well slots ITS, as this is the most common size on the NCS. Larger template structures exist and are in operation, e.g. the eight-well slots ITS on the Ormen Lange field operated by Shell. A X-mas tree will be installed on each well slot prior to petroleum production, as well as a manifold located in the center of the structure, as seen in figure 2.2. The manifold is installed prior to the X-mas trees.

The selected ITS is installed on the Gjøa field, located approximately 65 km southwest from Florø. The ITS commenced start-up of production in 2010 [36]. The original weight of the structure is 270 metric ton (Te) [37]. It should be noted that the weight is dependent upon the dimensions of the suction anchors.



Figure 2.1: Several suction anchors lined up, each with a height of ca. 6.5 meters [16].

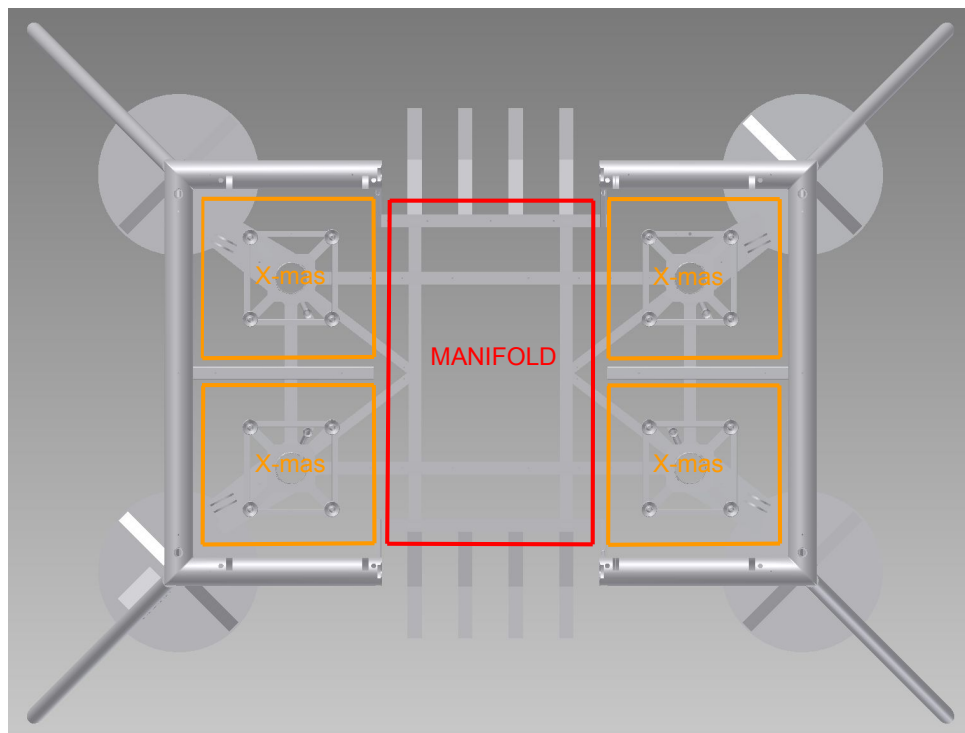


Figure 2.2: Figure of X-mas and manifold placement on the ITS seen from above, orange is space for X-mas while the red is space for manifold.

One of the purposes of this study is to reduce the weight of the ITS, and thereby reduce installation cost as lighter construction vessels with smaller crane capacity can be used to install the structure. There are several ways to install an ITS, where the most common ones are:

- Installation from subsea construction vessel, utilizing the vessels crane capacity to lift the structure overboard and install it.
- Perform a submerged tow from land to offshore, and then install it. This method has been developed more recently, and Subsea 7 had it patent pending in 2011 according to reference [38].
- Use a barge to transport the structure from shore and install it using an offshore crane vessel.

This thesis will focus on the first mentioned installation method, installation from subsea construction vessel, as this installation method is widely used by the industry [39].

The weight reduction will be achieved by implementing new construction materials for the ITS, where aluminium is believed to be the most promising material because of it's light weight compared to mechanical properties and corrosion resistance in seawater. Many of the same joining methods can be used for both steel and aluminium which makes the introduction of aluminium realistic in an industrial perspective. The fabrication cost of aluminium made structures are competitive with steel, exemplified by the extensive use of aluminium in topside structures [9].

A model of the selected ITS including markups has been received from Subsea 7, making it possible to rebuild a realistically sized ITS in a computer aided design program (CAD-program). In order to perform structural analysis on the structure, using finite element method (FEM) and conventional stress formulas. Most parts of the structure will have to be redesigned to utilize aluminium's properties in a positive way, especially with respect to difference in yield strength, joining and fabrication methods.

2.1.1 X-mas tree

The X-mas tree is the connection between the well and subsea production system, and it is connected to the wellhead. The X-mas tree acts as an essential integrity barrier, and it is therefore designed to be in a functional state under demanding operating conditions (all X-mas trees are pressure rated). The weight of a subsea X-mas tree ranges from 30-70Te [17]. The interface between the ITS and the X-mas tree can be seen in figure 2.3.

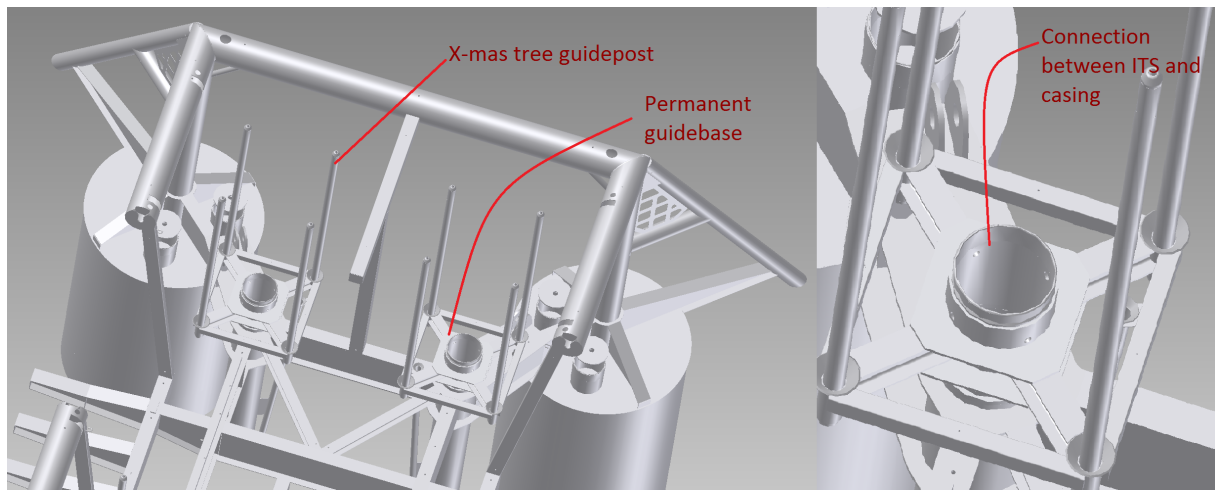


Figure 2.3: Position of X-mas tree on permanent guide-base on the ITS, only two out of four wells are shown on the left figure.

2.1.2 Manifold

The purpose of the manifold is to co-mingle the production fluids from several wells and transport it to a hosting facility in one pipeline (in some cases several pipelines). This method of commingling the production minimizes the use of expensive pipelines and risers. A manifold uses several independently operated valves to control the production from each well. The weight of a manifold normally ranges from 50 to 400Te [17]. The manifold installed on the selected ITS weights approximately 80Te [40]. There is no need for a weight reduction of the manifold at this stage as the weight of the concerned manifold is substantially lower than the weight of the concerned ITS. Figure 2.4 shows the type and size of a manifold installed on the ITS.



Figure 2.4: Picture of a manifold manufactured by Agility Group [17].

Chapter 3

Aluminium

Aluminium alloys offers many positive properties that have been proven to be successful in the aerospace [41], automobile [42] and naval [43] industry. An article [43] published in 2008 on the use of aluminium in naval applications estimated the life cycle cost of aluminium to be more cost efficient than steel. These findings resulted in a recommendation to the U.S Navy to increase the use of aluminium in ship structures [43].

This chapter describes properties of aluminium that are considered to be of importance in order to utilize aluminium in subsea structures.

3.1 Aluminium designation

Aluminium alloys are normally either wrought or casted, where they are further designated in series ranging from 1xxx-9xxx for wrought and 1xx.x-9xx.x for casted alloys. Table 3.1 presents the principal alloy in each series, and which series that is susceptible to heat treatment [1]. Extruded and rolled products are examples of wrought products, which generally possess higher strength than casted products [1].

The wrought aluminium alloys 5083 and 6082 are the only ones approved for immersed seawater applications in NORSOK M-121 [44], where the latter is more suited for extrusion while the first is often made in plated product forms due to strengthening by cold work. The strongest acceptable tempers for the mentioned alloys are H116 and T6 respectively [44].

Table 3.1: Wrought and cast series [1].

Wrought alloys ¹⁾		Casted alloys ¹⁾	
Series	Principal alloy	Series	Alloy family
1xxx (Al)	Unalloyed. NHT	1xx.0	Unalloyed. NHT
2xxx (AlCu)	Copper. HT	2xx.0	Copper. HT
3xxx (AlMn)	Manganese. NHT	3xx.0	Silicon and/or copper. HT
4xxx (AlSi)	Silicon. Primarily NHT	4xx.0	Silicon. HT
5xxx (AlMg)	Magnesium. NHT	5xx.0	Magnesium. NHT
6xxx (AlMgSi)	Magnesium and Silicon. HT	6xx.0	Not used
7xxx (AlMgZn)	Zinc. HT	7xx.0	Zinc. HT
8xxx (Other)	(e.g. Iron or Tin)	8xx.0	Tin. HT
9xxx	Unassigned	9xx.0	Other

1) HT= Heat treatable. NHT= Non heat treatable.

The alloy designation is always followed by a capital letter indicating the hardening method applied as illustrated in table 3.2. The strongest approved temper designations for immersed seawater applications are also presented.

Table 3.2: Temper designations [2].

Temper designation	Detailed temper designation
F As fabricated	
O Annealed	
H Strain hardened	H116 ¹⁾ Treated against exfoliation and intergranular corrosion and optimum resistance against stress corrosion [44].
W Solution heat treated	
T Thermally treated	T6 Solution heat treated followed by artificially aging.

1) Equivalent mechanical properties as H32 stated in EN-1999-1-1 [2, 44].

3.2 Material properties

General properties of aluminium alloy 5083-H116 and 6083-T6 are presented in table 3.3, and the chemical composition is tabulated in table 3.4. Note that the modulus of elasticity for aluminium is ca. 33% of steel's modulus of elasticity, which may lead to higher displacement under loading. The density of aluminium is 35% compared to steel, and aluminium has a higher absorption of energy per unit weight during impact loading compared to steel [45]. Yield strength (YS) for aluminium is defined at 0.2% elongation offset the modulus of elasticity line.

Table 3.3: Properties of aluminium alloys 5083-H116 and 6083-T6 [2].

	Density [kg/m ³]	Young's modulus [GPa]	YS [MPa]	TS ¹⁾ [MPa]	Brinell Hardness [HB]	TEC ²⁾ [1/°C]	Thermal conductivity [W/(m·K)]	Durability ³⁾ rating
5083-H116	2660	70	215	305	85	$23,8 \cdot 10^{-6}$	117	A
6082-T6	2700	70	255	300	91	$23,4 \cdot 10^{-6}$	170	B

1) Tensile Strength.

2) Thermal Expansion Coefficient.

3) A= The need for protection for freely exposed members in seawater depends on inspection availability and planned maintenance [2].

B= Protection of freely exposed members generally required for seawater immersion [2].

Table 3.4: Chemical composition of aluminium alloy 5083-H116 and 6082-T6 [wt%] [3].

Alloy	Si	Fe	Cu	Mn	Mg	Cr	Zn	Ti
5083-H116	0.4	0.4	0.1	0.4-1.0	4.0-4.9	0.05-0.25	0.25	0.15
6082-T6	0.7-1.3	0.5	0.1	0.4-1.0	0.6-1.2	0.25	0.2	0.1

3.2.1 Operating temperature

Operating temperature for aluminium alloys should in general not be higher than 80 - 100 °C [2]. However, the 5083 alloy is more sensitive to high temperatures, hence the operating temperature for this alloy shall not be higher than 65 °C [44]. Exceedance of operating temperature may lead to unwanted strength losses.

3.2.2 Cryogenic toughness

Aluminium alloys have the ability to maintain ductility and improve toughness while the strength is increasing when subjected to very low temperatures due to the face centered cubic crystal structure (FCC) [46]. The 5xxx and 6xxx series are particularly noteworthy for their abilities in extremely low temperatures where their properties improves when exposed to temperatures close to absolute zero compared to room temperature [1]. For arctic temperatures which subsea structures may be subjected to during transport the ductility is maintained or increased [1]. The comparative steel material may develop brittle properties in these environments.

3.2.3 Hydrogen embrittlement

Hydrogen may be produced from cathodic protection (CP) at the cathode (see section 3.5.8), which can diffuse into susceptible materials and cause embrittlement. Three factors need to be present in order to be in danger of hydrogen embrittlement (HE): susceptible material, hydrogen and stress. A typical example of a susceptible material is duplex stainless steels [47]. Aluminium alloys are however considered immune to HE according to DNV-RP-B401 [47]. A study [48] on the effect of CP and susceptibility to HE in aluminium alloys concluded that the alloys 5083-H321 and 6082-T6 are not particularly prone to HE. Composition and thermal treatment did not effect the results. However, an alloy within the 7xxx series showed a decrease in strength due to HE [48].

3.3 Extrusion

The ability to extrude aluminium alloys enables advantages in structures; complex shapes and profiles readily forms through a die in order to obtain the optimal and desired design. The cross-section can be designed in such a way that a low weight is obtained while at the same time possess an excellent structural strength efficiency. For instance, where installation of stiffeners allows for an enhanced steel design, the extrusion process have the potential to achieve improved properties simply by adjusting the cross-sectional die shape of the extruded components. Extruded profiles can in theory be manufactured in any two di-

mensional profile where the only limitation is the press capabilities. Table 3.5 illustrates the extrusion limits for the three largest identified extrusion companies in Europe, and one from Russia (KUMZ). An illustration of the circumference limits at STEP-G and SAPA can be seen in figure 3.1 and appendix F.1, respectively.

Table 3.5: Extrusion limits at STEP-G [4], SAPA [5], Constellium [6] and KUMZ [7].

	STEP-G	SAPA	Constellium	KUMZ
Max. Press force	90 MN	65 MN	100 MN	-
Weight pr. meter	190 kg/m	65 kg/m	80 ³⁾ kg/m	-
Total weight	535 kg	-	-	-
Max. Length	30 m	26 m	30 m	6 m
Max. diameter	500 ¹⁾ mm	320 mm	520 ²⁾ mm	553 mm
Max. width	750 mm	620 mm	750 mm	480 mm

1) Max. thickness is 30 mm for tubes.

2) Alloy 6082, Max. \varnothing 500x20 mm with length 6 m.

3) For alloy 6082.

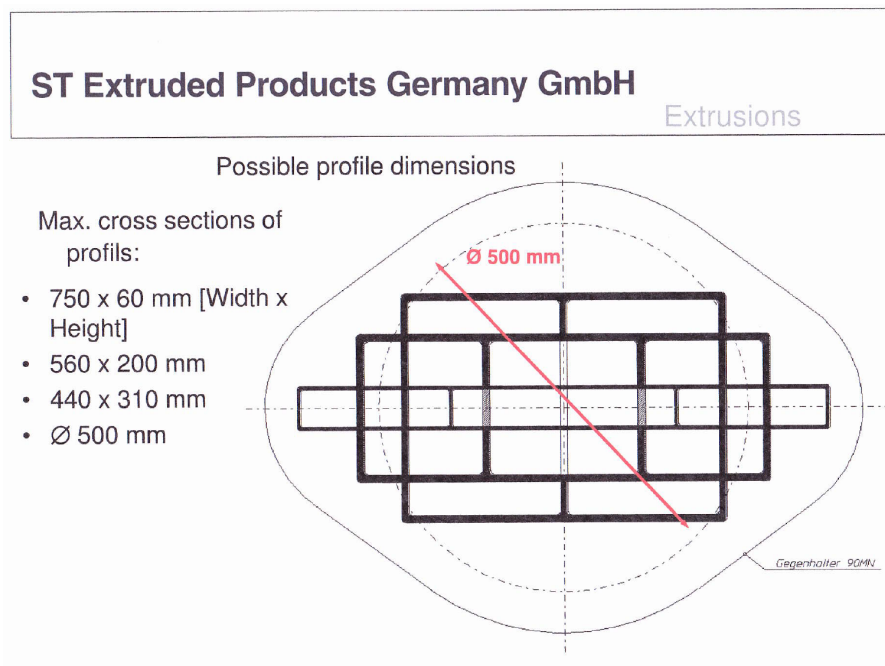


Figure 3.1: Circumference limits for the extrusion press at STEP-G [4].

A disadvantage by extrusion is the fact that each profile requires a unique die, if the die is not readily obtainable through standards the die must be custom made causing increased costs. Therefore, the designer should limit both the amount and complexity of the profiles needed and establish contact with competent fabricators.

3.4 Recycling

Aluminium can be recycled very efficient, requiring only a fraction (ca. 5 percent) of the energy consumed producing primary aluminium for the remelting process. About 75 percent of all extracted aluminium remains in use, whereas some are recycled through multiple cycles. However, Only 20-25 percent of the total aluminum demand is covered from recycled products due to the extended lifetime of many aluminium products. Basically all products made of aluminium are possible sources of recycled aluminium [49]. Recycling has a potential in being environmentally friendly by lowering emissions and saving energy.

There is a special concern with aluminium recycling. As aluminium is recycled over and over again impurities will build up. The accumulation of impurities/contamination may result in a chemical composition, which will influence the aluminium's properties. This is a growing problem due to the increased amount of recycled aluminium but it can be handled in the remelting process by using several technologies. Further description on applicable technologies can be seen in reference [50].

3.5 Corrosion

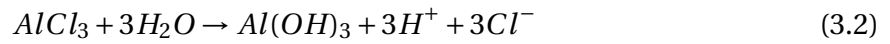
The corrosion properties of aluminium alloys in seawater have to be assessed in detail to ensure there are no problems by utilizing aluminium alloys subsea. A protection strategy for aluminium in the concerned environments will be described based on existing literature. Several types of corrosion are relevant to the two types of environments, aluminium in seawater and aluminium buried in soil. Relevant corrosion types are crevice corrosion, intercrystalline corrosion, pitting corrosion, galvanic corrosion and uniform corrosion [19].

3.5.1 Crevice corrosion

Crevice corrosion is localized corrosion happening in crevices, i.e. around overlapping zones such as bolting, riveting and welding. These crevices are difficult to enter for seawater, and when an electrochemical reaction occurs in the crevice the local environment will change as a result of the difficult access. The potential becomes more electronegative, and aluminium in the crevice will be oxidized as described by reaction 3.1 [19].



While the outside of the crevice will have an oxygen reduction by the reaction occurring in the crevice. The aluminium chloride will hydrolyze by the chemical reaction 3.2.



The aluminium hydrolyze is caused by the local geometry limiting diffusion rates causing depletion of oxygen, and an excess of Al^{3+} ions which leads to an inflow of Cl^{-} ions. Figure 3.2 illustrates the reactions and where they take place in relation to a crevice [19].

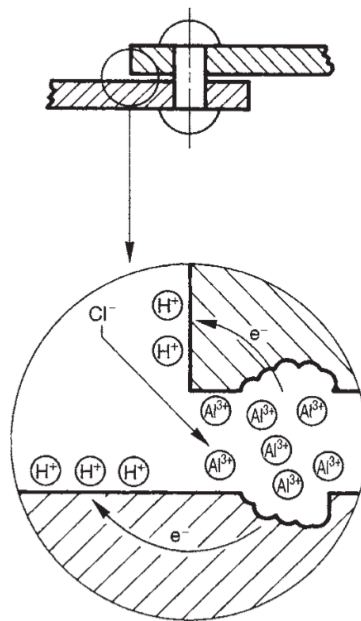


Figure 3.2: Crevice corrosion in an overlapping joint [18].

Aluminium's susceptibility towards crevice corrosion is rather low compared to some other alloys, e.g. stainless steel. Aluminium crevices are often sealed by the corrosion products of aluminium, which is one of the reasons for the low susceptibility compared to stainless steel. However, crevice corrosion can and should be avoided by design. Either by using gaskets or sealing compounds to avoid liquid from penetrating into crevices. Or by designing in such a way that crevices are avoided, a figure of poor and good design can be seen in figure 3.3.

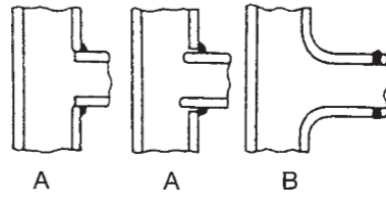


Figure 3.3: Poor design in A and good design in B with respect to crevices [19].

3.5.2 Pitting corrosion

Localized corrosion of aluminium in the passive pH range is usually subjected to a formation of pits. See figure 3.15 for illustration of the pitting potential, (E_P) and (E_{CC}), at various pH values for alloy 5086. Aluminium alloys polarized to a potential greater than the E_P potential and lower than E_{CC} in the passive region are in danger of encountering pitting corrosion. When polarized and kept in the passive area the protective oxide layer can readily maintain its integrity (anodic polarization) and thereby avoid initiation of pits [51]. At lower or higher potentials than the passivity area the oxide layer is no longer able to maintain an adequate integrity, and the oxide layer breaks at weak spots where it cannot repair itself. Pitting is often produced by chloride ions (Cl^-), in for instance seawater [51].

The localized corrosion of aluminium alloys is primarily determined by the properties of the intermetallic particles [20], as seen in figure 3.4. It is secondarily determined by the alloy composition, as long as copper is absent. Pitting initiates around the intermetallic particles, which is the "weak spot" for commercial alloys. Figure 3.4 illustrates this scenario, and also illustrates how the local environment changes, and the pH value increases locally [20]. For further reading about this phenomenon, see reference [20].

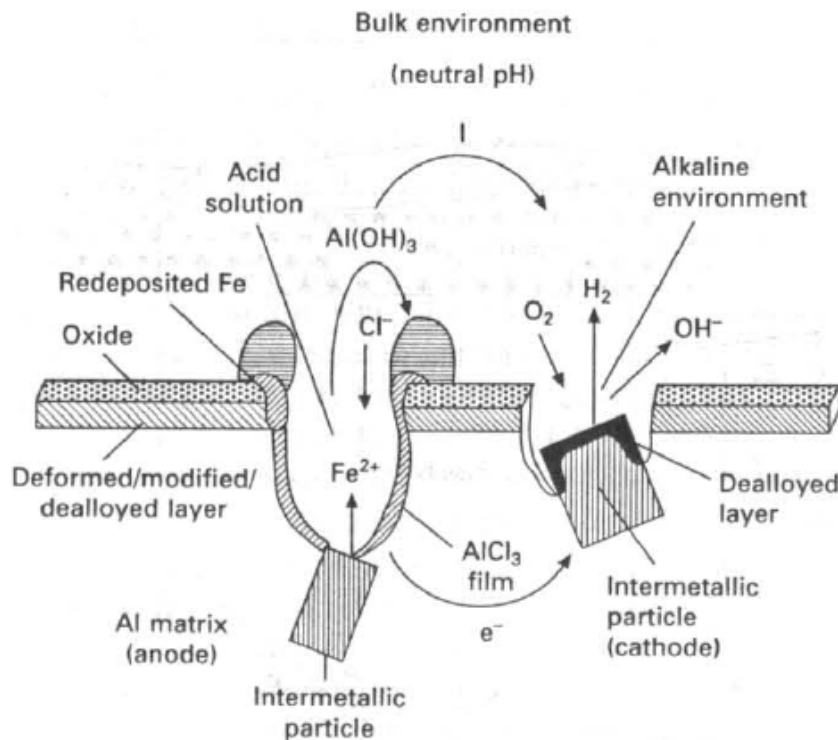


Figure 3.4: Illustration of localized corrosion around intermetallic particles in the acidic pit and adjacent alkaline cathodic site [20].

3.5.3 Intercrystalline corrosion

Intercrystalline corrosion may propagate within the grains or through grain boundaries, referred to as transgranular or intergranular corrosion respectively [19]. This type of corrosion initiates at pits described in section 3.5.2. It propagates due to an electrochemical potential between the grain boundaries (intermetallic phases) and the grains (bulk), see figure 3.5. The intermetallic compound can have a more negative and a more positive potential, therefore the grain boundaries can act as an anode or as a cathode with respect to the solid solution (bulk of the grains). Aluminium alloys are in danger of encountering intercrystalline corrosion if a corrosive environment, a potential difference of more than 100 mV, and a continuous precipitation of intermetallics are present. The intensity of intercrystalline corrosion depends on the amount of grain layers attacked. If there is no more than three to four influenced layers, it is considered as superficial. Superficial intercrystalline corrosion occurs in alloys of the 6xxx series without copper content. 5xxx and 6xxx series with magnesium content more than 3.5%, may be susceptible to intercrystalline corrosion [19]. However, alloy 5083 with temper H116 should not be susceptible to this type of corrosion [19, 44].

rate is high for low resistivities, as seen in figure 3.6 [19].

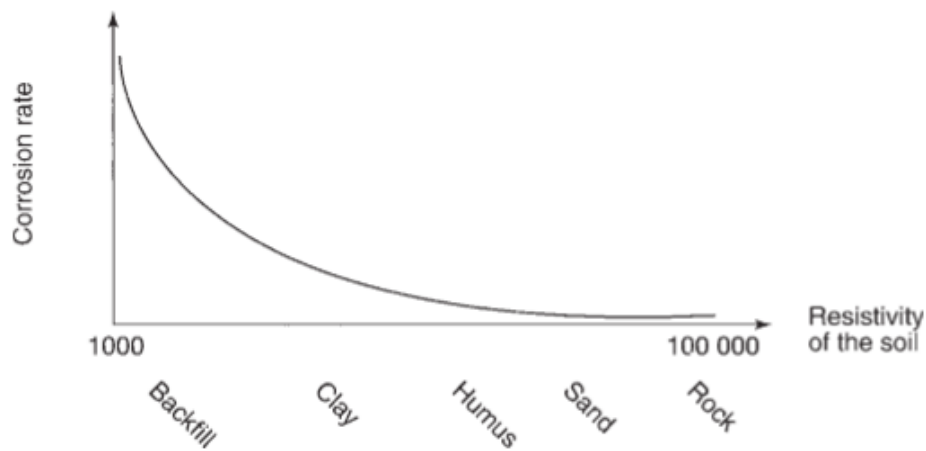


Figure 3.6: Resistivity versus aggressiveness for soils [19].

Experiments on cathodically polarized thermally sprayed aluminium (TSA) in soil has been carried out [21], the set-up can be seen in figure 1, page 562 in the reference [21]. The experiment was aimed at higher temperatures, ranging from ambient temperature up to 95 °C. As seen in the figure 3.7, the cathodic current density decreased from -50 mA/m^2 to ca. -10 mA/m^2 . The cathodic current is negative because the TSA is acting as an anode, not a cathode [21].

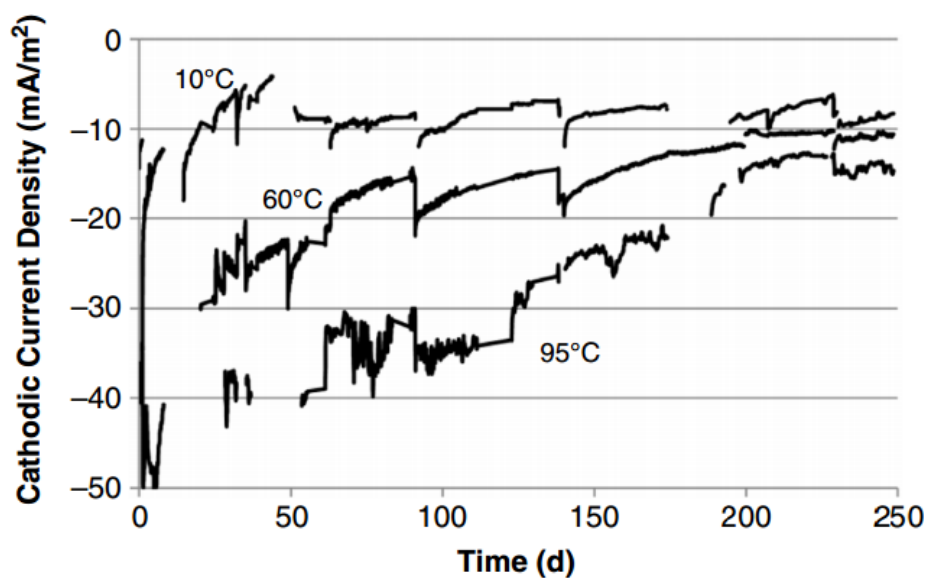


Figure 3.7: Cathodic current density in soil at -1100 mV vs Ag/AgCl for different temperatures [21].

The corrosion rate of the cathodically polarized TSA buried in soil decreases with time due to passivation of the TSA surface, as seen in figure 3.8. After 250 days the corrosion rate is below $10 \mu\text{m}/\text{year}$. The pH value at the TSA/soil interface is kept between 8.2 (pH of soil before the tests started) and 6.5 (pH at 95 degrees Celsius) [21]. TSA is typically applied in a 200 to 400 micrometer thick layer, resulting in a lifetime of ca. 20 years for a 200 micrometer thick layer of TSA.

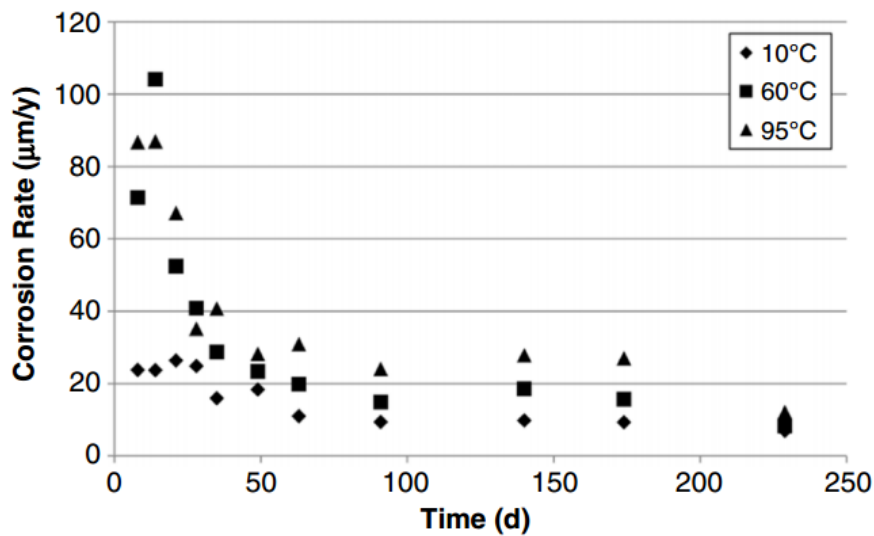


Figure 3.8: Corrosion rate in soil polarized to -1100 mV vs Ag/AgCl for several temperatures [21].

The corrosion properties of TSA and alloy 5083 or alloy 6082 are not the same. Alloy 5083 and alloy 6082 shows better corrosion resistance against uniform corrosion according to email correspondence with Ole Øystein Knudsen, seen in appendix B.3. Corrosion of aluminium in soil is considered to be a relatively undiscovered area with ongoing research. The literature within this topic is therefore very limited, and there is a need for more research in order to state a valid conclusion regarding the use of aluminium components buried permanently in soil.

3.5.5 Seawater flow rates influence on corrosion rates

As a result of the integrated template structure's geometry it is expected to be areas with limited seawater flow rate, so called stagnant conditions. This condition is especially expected inside circular tubes with small holes to the outside as seen in figure 3.9.

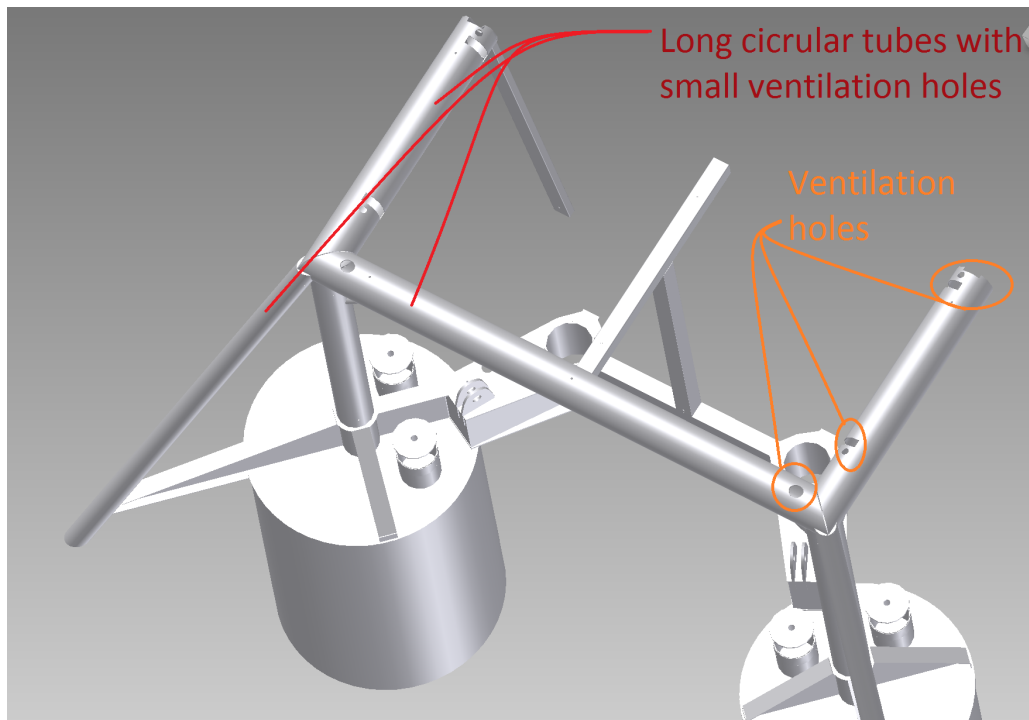


Figure 3.9: Example of long circular tubes on the structure with small ventilation holes to the outside seawater.

Kemal Nisancigolu and Torgeir Wenn conducted an experiment [23] on corrosion protection of aluminium in flowing seawater, where aluminium alloys were exposed to seawater at various seawater flow rates (0 - 100 cm/s). The result did not show any critical pitting depth after two months of exposure under open-circuit conditions for low velocities, as seen in figure 3.11. 2.5 cm/s flow rate seems to be more critical with respect to pitting corrosion depth than 0 cm/s (stagnant conditions) for AlMgSi1 (alloy 6082) [23].

Stagnant seawater conditions are not a corrosive problem according to the reviewed literature, and there has not been found any evidence to support the case of stagnant conditions being a corrosive problem, which is different from the same condition for steel (unless it is a closed compartment).

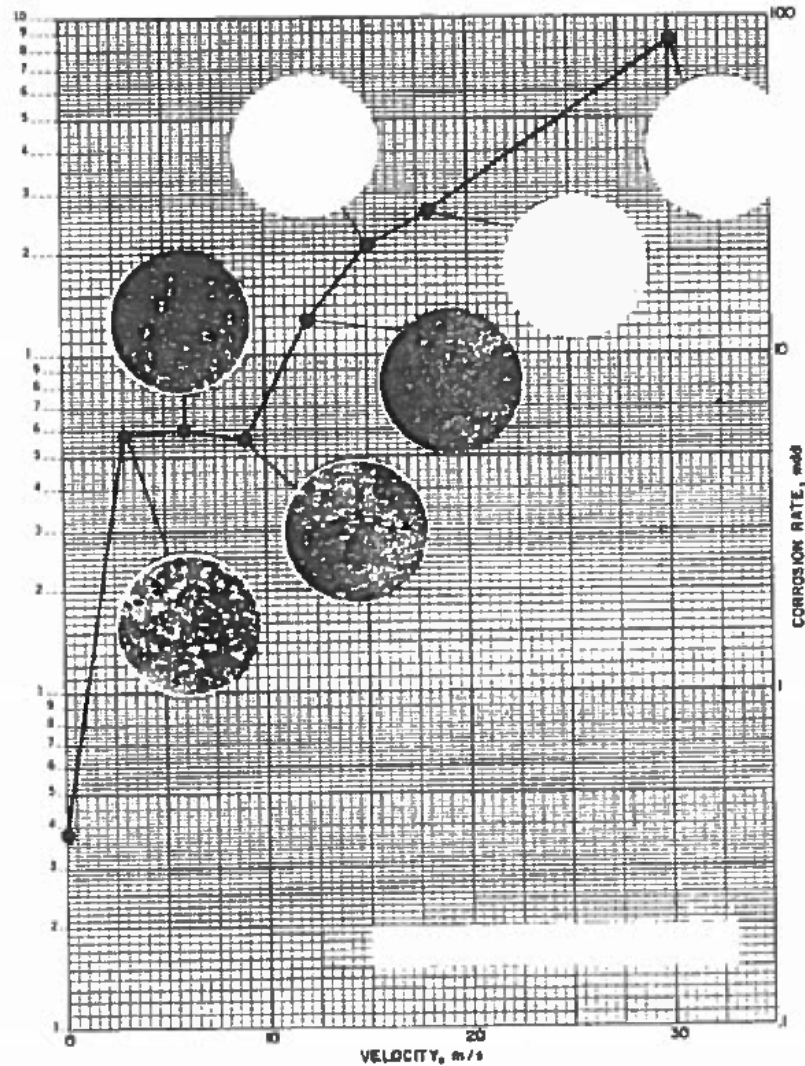


Figure 3.10: Velocities influence on corrosion rate for alloy 5456 connected to CP, 1 mdd = $13.5 \mu m$ [22].

For steel structures the internal tubing is treated as closed compartments by using "split rubber grommets" to close the compartment [53]. All internal tubing has to be filled with seawater in order to avoid a high differential pressure which can lead to structural collapse at high pressure differences [53]. NORSOK M-001 [54], section 4.3.7 can be used for corrosion protection of closed compartments, where no internal corrosion protection is needed for completely closed seawater filled compartments in carbon steel. This is assuming there is no seawater exchange between internal and outer part of the structure, achieved by using split rubber grommets as seen in figure 3.12.

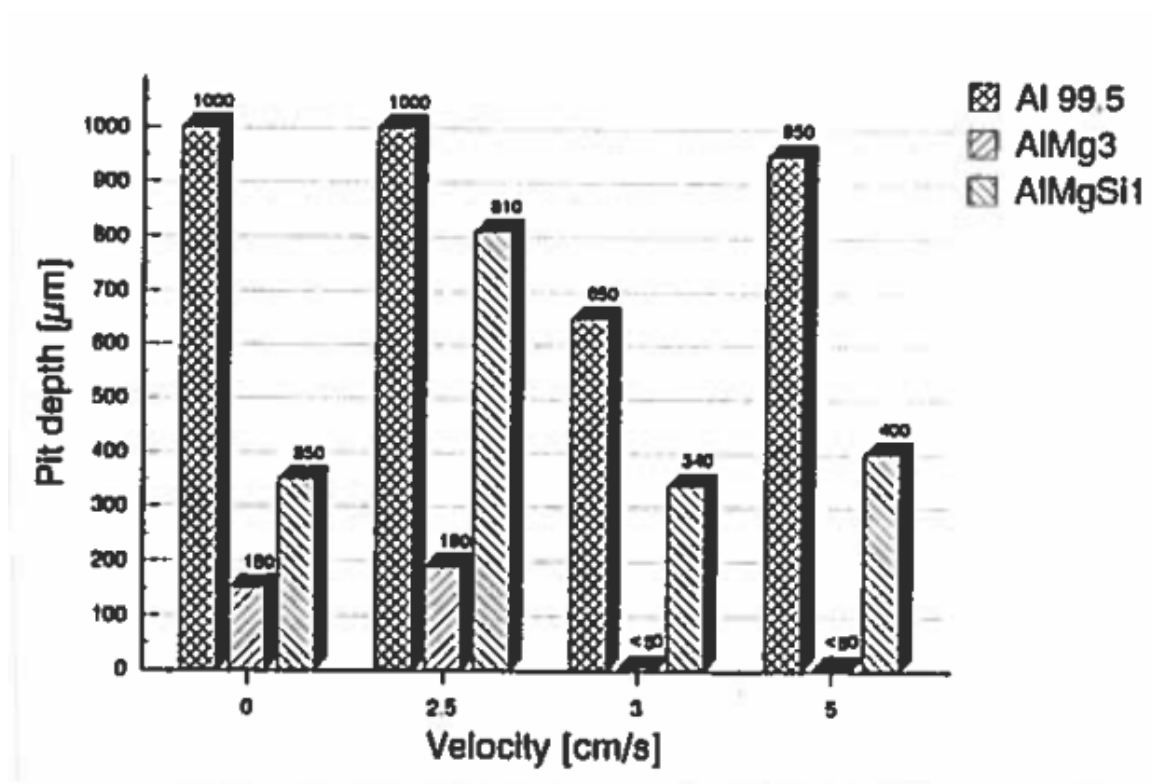


Figure 3.11: Maximum pit depth for Al 99,5, AlMg3 and AlMgSi1 (alloy 6082) after two months of exposure under open-circuit potential [23].



Figure 3.12: A split rubber grommet to restrict seawater flow in/out of compartments [24].

A SINTEF report from 1986 [22], concerning corrosion of aluminium in special environments relevant to applications in the petroleum industry studied flowing seawater velocities influence on corrosion rate. Figure 3.10 shows the corrosion rate as a function of flowing seawater velocity, low seawater velocity results in low corrosion rate. The corrosion rate at the plateau, from 3 to 9 m/s is about 0.08 mm/year [22], the corrosion rate is almost independent of the

seawater velocity in this range [22]. The nearly constant corrosion rate in this region is related to oxygen transport and the corrosion reactions at the anodic sites [22]. Above 9 m/s, the corrosion rate increases rapidly due to erosion-corrosion [22]. Flow induced corrosion cannot be mitigated by CP, it may actually increase the corrosion rate [20]. The transition velocity between pitting at low velocities (stagnant conditions) and uniform corrosion due to flowing seawater were estimated to occur around 10 cm/s for alloy 6082 [20]. A transition velocity for alloy 5754, also tested in the experiment, was not identified [20].

3.5.6 Galvanic corrosion

Structural members surrounded by an electrolyte (seawater) are in danger of encountering galvanic corrosion if there is a metallic contact between dissimilar metals possessing different corrosion potentials. Coupling between steel and aluminium is an example of a galvanic coupling, where the aluminium is acting as anode and steel as a cathode, resulting in corrosion of aluminium and CP of steel. The galvanic series are illustrated in figure 3.6, where the more noble materials possesses more positive potentials. Cathodic protection that utilizes sacrificial anodes with lower corrosion potentials are therefore an effective method to mitigate galvanic corrosion, which is confirmed by a recent study seen in reference [55]. See section 3.5.8 and 3.5.1 for more information about CP and crevice corrosion of aluminium alloys in seawater.

A galvanic coupling may also occur between aluminium alloys and filler material used in welding or brazing for instance, as these possess varying corrosion potentials due to their respective chemical composition [56]. Hence, it is important to ensure proper CP of these interconnections. Corrosion potentials for 5xxx and 6xxx alloys are tabulated in table 3.6. Figure 3.14 illustrates an example of a galvanic interconnection between alloy 5083, 6082 and filler material. Considering such a welded coupling without CP, one may assume that the alloys in the 5xxx series would suffer more from corrosion due to possession of a more negative corrosion potential.

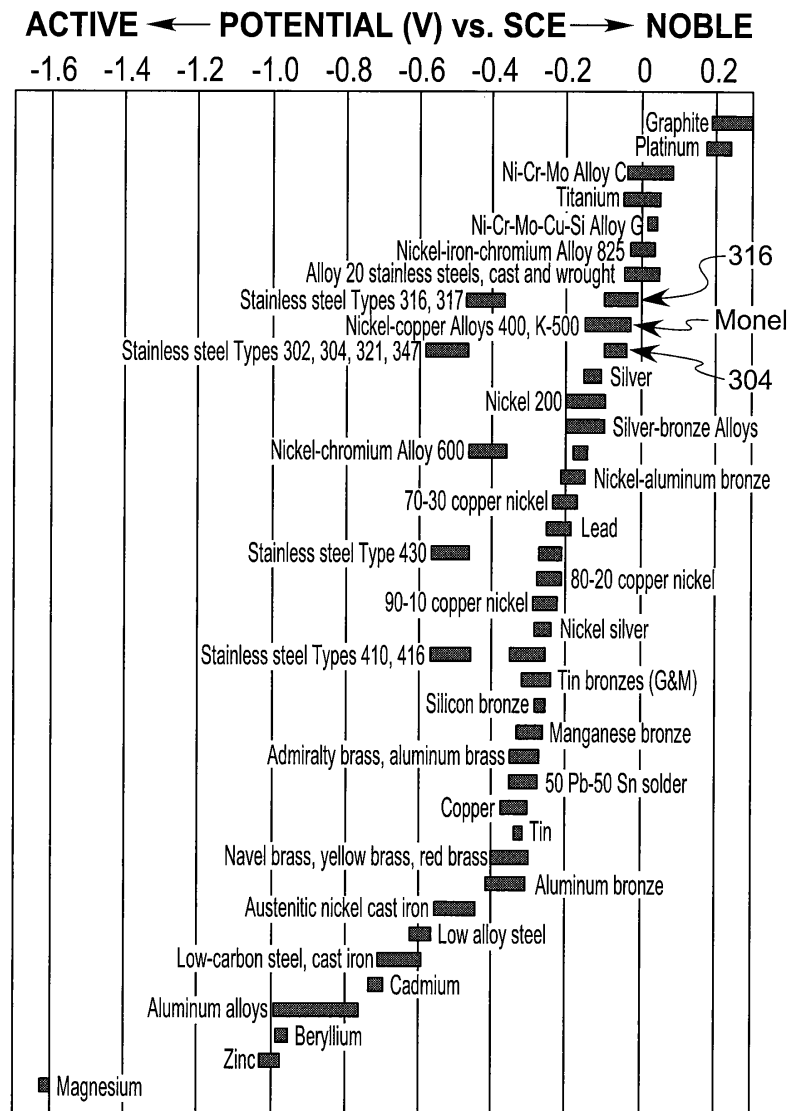


Figure 3.13: Galvanic series in seawater [25].

For bolted joints one may introduce extra material that are different from the main structure, such as carbon or stainless steel bolt, nut and washer that can contribute to galvanic corrosion. Bolted joints are in addition externally sealed to avoid mitigation of crevice corrosion. All components in a bolted connection must be cathodically protected and ensured to possess the correct protection potential. Wiring can be used to ensure an electrical contact of isolated parts. Corrosion issues with the use of bolts are further discussed in section 3.8.

Table 3.6: Corrosion potentials for 5xxx and 6xxx alloys [8].

Material	Corrosion potential [Volt vs SCE]
5xxx alloys	-0.78 to -0.76
6xxx alloys	-0.73 to -0.70

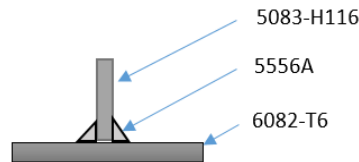


Figure 3.14: Galvanic interconnection between aluminium alloy 5083-H116, 6082-T6 and filler metal 5556A.

3.5.7 Coating

The selected corrosion strategy does not make coating of the aluminium surface necessary, as the structure will be protected by a passive oxide layer and cathodic protection. This is in line with the NORSOK M-501 [57] standard stating:

"The following shall not be coated unless otherwise specified:

- *Aluminium, titanium, uninsulated stainless steel, insulated stainless steel heating ventilation/air conditioning ducts, chrome plated, nickel plated, copper, brass, lead, plastic or similar;*
- *Jacketing materials on insulated surfaces."*

3.5.8 Corrosion and protection of aluminium in seawater

The passivation of aluminium in seawater is dependent upon the aluminium alloy composition and pH. The presence of manganese and magnesium as alloying components will increase the aluminium alloys resistance to high pH, as a result of this alloy 5086 in figure 3.15 can achieve passivation at pH values similar to seawater (seawater pH dependency of depth can be seen in appendix F.2) [20]. According to a recent study [58] it was found that the Pourbaix diagram in figure 3.15 is a good estimate for alloy 5083 and 6082 at pH 8.2 (seawater condition). At pH 3 it was found to be a rough estimate, and at pH 10 it was not valid [58].

Cathodic protection of aluminium differs from CP of other metals e.g. steel in two ways. The aluminium oxide layer is not stable in alkaline or acid environments where the aluminium

will start corroding actively [26]. The local environment is influenced by the seawater flow speed, further discussed in section 3.5.5. The second difference is the low current demand compared to the current demand for steel. The difference in current demand is because the current is confined by the cathodic intermetallic areas for aluminium. These areas are small compared to the exposed area of aluminium in contrast to steel where the whole exposed area demands current [26]. How aluminium compares to steel can be seen in figure 3.16, showing cathodic polarization curves, where plates sized $10 \cdot 19$ cm are inserted into a flow channel at a flow rate of 8 cm/s. The data were measured before the onset of calcareous deposition [26].

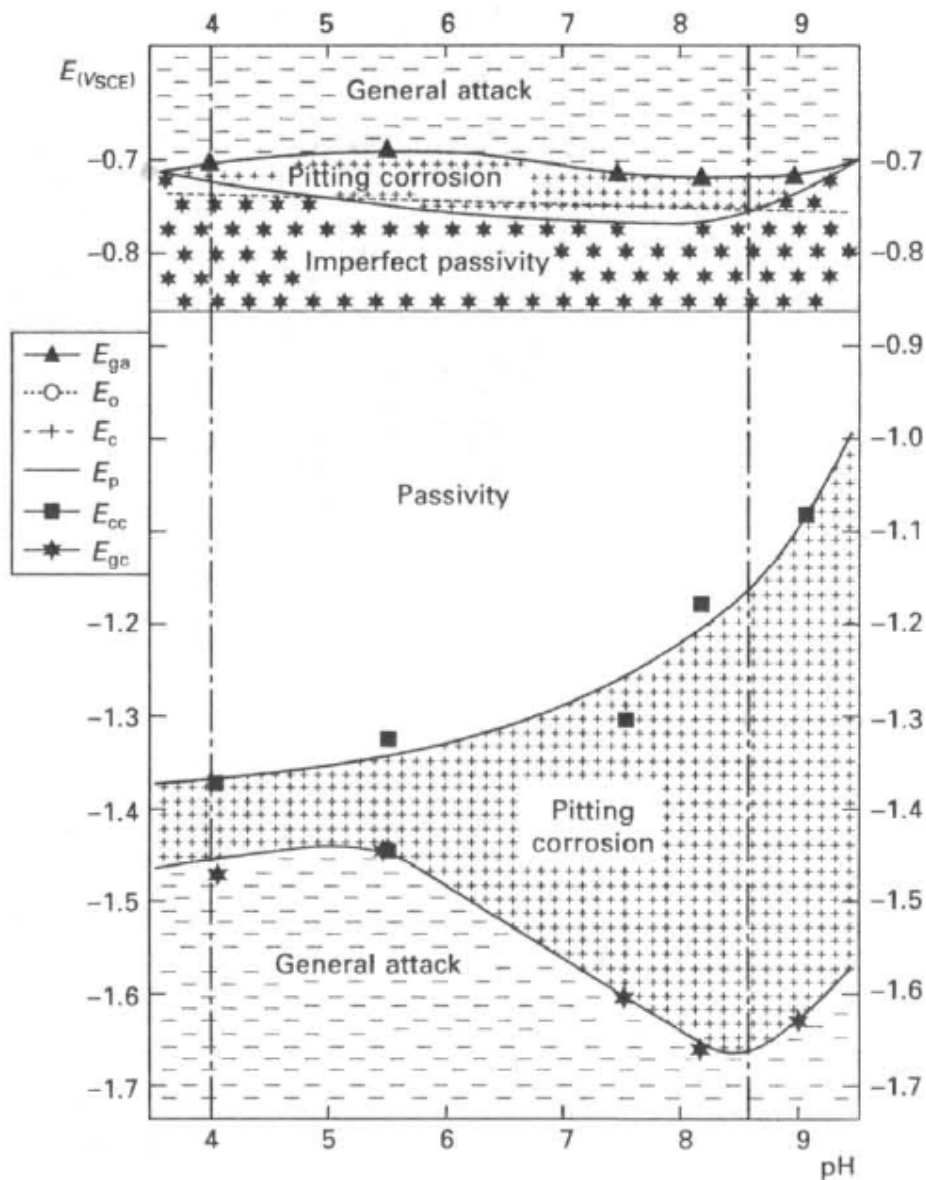


Figure 3.15: Pourbaix diagram for alloy 5086 [20].

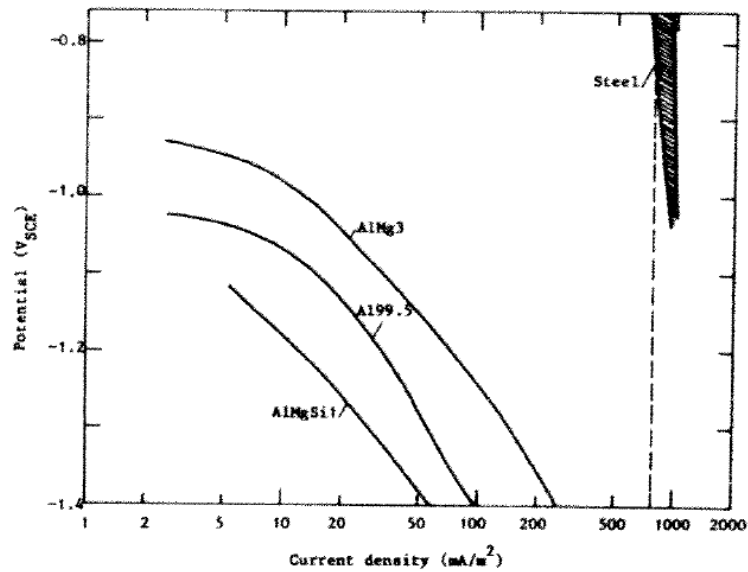


Figure 3.16: Cathodic polarization curve for steel and aluminium at 8 cm/s flow rate [26]. AlMgSi1 is equivalent to 6082 and AlMg3 is equivalent to 5754.

Cathodic protection strategy

Cathodic protection is selected to act as the main barrier against pitting corrosion, uniform corrosion on coated steel components and to mitigate the consequences of a galvanic coupling. Cathodic protection is achieved by placing the material to be protected in the situation of an cathode. This is performed by electrically linking it to a metal with lower corrosion potential where the two materials are placed in the same electrolyte.

Protection of aluminium is achieved by keeping the aluminium surface in a passive zone according to a Pourbaix diagram, seen for alloy 5086 in figure 3.15. And not by bringing the potential into an immunity range, which is the case for steel [19]. The cathodic protection potential should therefore be kept more negative than the critical pitting potential to achieve satisfying protection. Figure 3.17 shows schematically how the pits surrounding the intermetallic particles will develop when it is connected to a CP system [20].

The current demand for CP of aluminium will typically decline by one order of magnitude with time compared to freshly exposed areas. It is because of the passive layer (oxide layer) combined with the detachment of intermetallic particles, which are removed as shown in 3.17. Figure 3.18 shows how the current demand decreases with time [20].

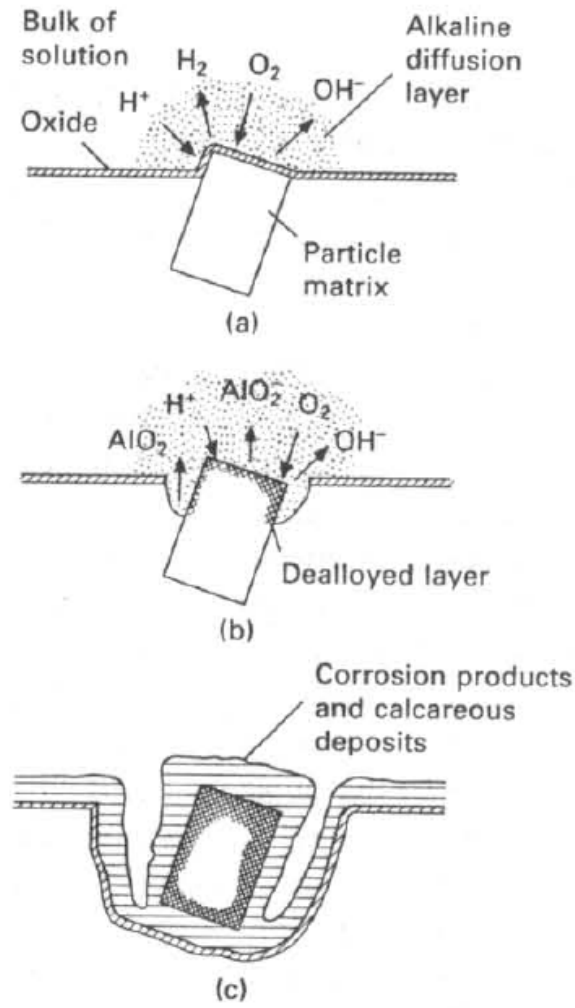


Figure 3.17: Mechanism of CP on aluminium in seawater [20].

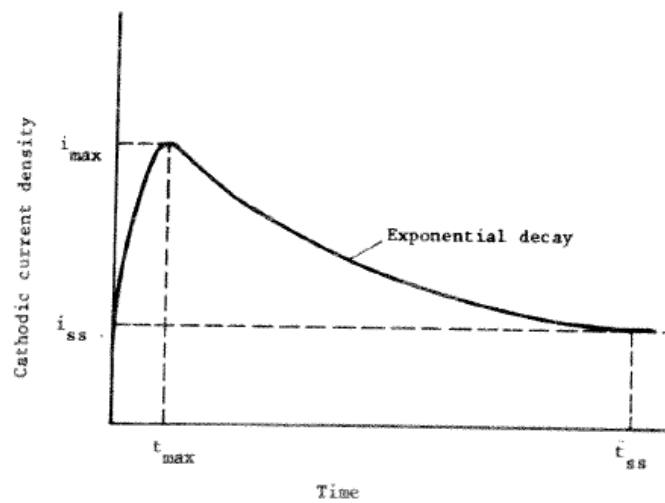


Figure 3.18: Typical current density vs time for a cathodically protected aluminium alloy in seawater [20].

Cathodic protection is less or ineffective if the seawater current reach a certain level, ca. 10 cm/s, as described in section 3.5.5. Uniform corrosion becomes critical when the seawater flow is above ca. 9 m/s because the oxide layer is mechanically removed continuously, as seen from figure 3.10.

3.6 Welding

Aluminium may be welded by several different methods. The relevant welding techniques for aluminium applications are: metal inert gas (MIG) welding, tungsten inert gas (TIG) welding, friction stir welding (FSW), and laser welding (LW). Each method has their individual advantages and drawbacks. MIG and TIG are well documented in Eurocode 9 [2], FSW and LW are however mentioned as alternative methods that has to be approved appropriately through testing before structural use [2].

Generally, structural members shall only be welded by workers with proper certification, and at an appropriate location, preferable in a welding workshop. An approval should be at hand if the welding has to be done elsewhere [9]. Additionally, proper closing of welds are important to avoid cracks and crevices that may lead to crevice corrosion in the welded zones [9].

3.6.1 Arc welding

MIG and TIG welding are the most common welding methods for structural aluminium [9]. MIG welds aluminium members by subjecting the filler metal to a current, hence, the supplied filler material is acting as the electrode. TIG welding is equipped with a non-consumable tungsten electrode, the filler must therefore be added separately during welding. An argon and/or helium inert gas is applied during welding to protect against oxide development in the molten metal. TIG and MIG provides high quality aluminium welds [59], and the welds may be produced with minimal distortions. An advantage with this type of welding is it does not require any protective flux that may affect the corrosion resistance, and the actual welding operation can be performed in challenging positions [60].

MIG welding are extensively used in the industry due to its user friendliness and low cost compared to TIG welding. MIG welding is also known to be a faster process than TIG [59]. However, TIG produces more precise welds of higher quality, which should be regarded as an advantage for members that undergoes very strict design requirements. TIG welding is a more environmental friendly process than MIG, since there are hardly any air contaminants produced in the process. On the other hand, TIG welding requires highly skilled welders [59]. For welding of aluminium, alternating current is normally preferred for TIG and direct current for MIG, where the consuming electrode is positively charged [27]. See figure 3.19 for illustrations of the two arc welding methods.

Thick members must be preheated in order to ensure a proper TIG and MIG joining, due to a requirement of possessing sufficient local heat [2]. It is important to perform any necessary preheating in a controlled manner, where particular concern should be directed to maximum allowable temperature and exposure time that may alter the material properties if exceeded. Table 3.7, presents maximum temperature and exposure time for alloy 5083 and 6082 at certain thicknesses. Additionally, temperature restrictions of the base metal when adding additional layers for multi-layer welds are tabulated [9].

Table 3.7: Preheating limits for arc welding of thick members [9].

Alloy	Thickness [mm]	Max. Temperature [$^{\circ}$ C]	Max. Exposure time [min]	Temperature limit multi-layer welds [$^{\circ}$ C]
5083	6-12	200	10	120
6082	above 10	200	30	100

The pistol used for MIG welding has a significant size, which may cause problems in less accessible areas. The diameter of the mouthpiece is ca. 30 mm and the filler wire extend 10-15 mm beyond the mouthpiece. This geometry on the end of the MIG pistol imposes difficulties in areas with too sharp angles between two profiles for example. The angle between structural members should therefore not be less than 35-40 $^{\circ}$ [9], and there must be enough space for the pistol itself in order to weld in the correct positioning of the pistol. Welding accessibility should be an important factor in design of structures as poor access often results in poor welds [9].

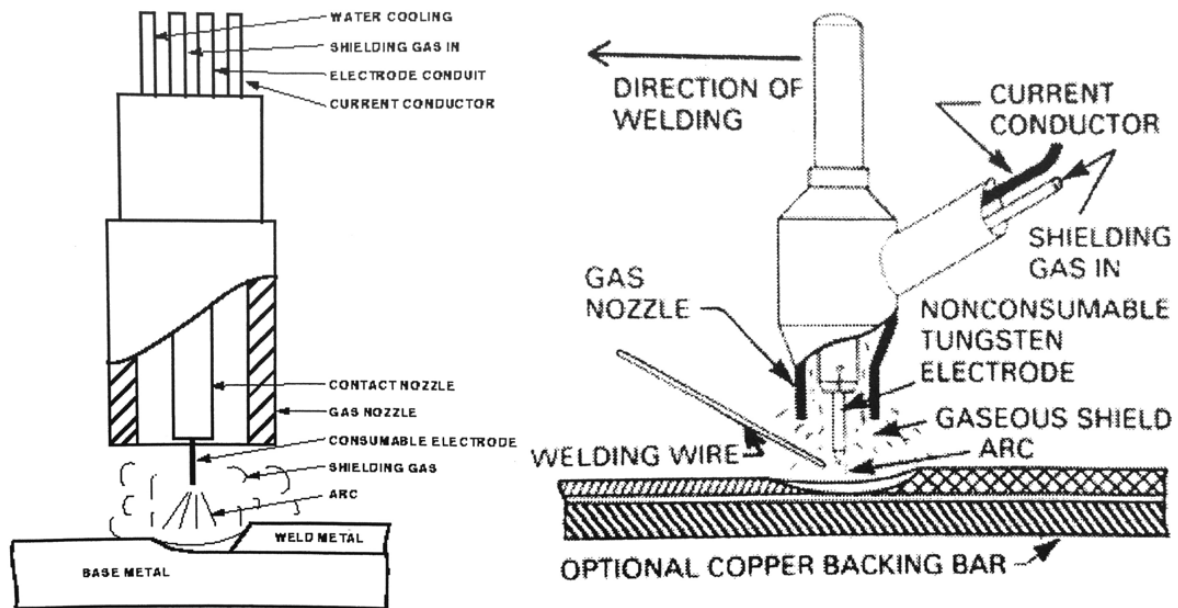


Figure 3.19: MIG (left) and TIG (right) welding [27].

In Eurocode 9 [2], the strength of the heat affected zone (HAZ) for several common aluminium alloys and product forms are listed. The HAZ values are only valid for MIG and TIG welds for thicknesses up to 15 mm, however according to a recent study [61] of HAZ values for larger thicknesses it is found that the HAZ strength will not be negatively influenced for thicknesses up to 30 mm [61].

The characteristic strength reductions and product forms for the NORSOK approved alloys are tabulated in table 3.9. The strength reductions varies with product form and welding method, different products are therefore tabulated along the with the respective strength reduction. Strength of the HAZ in TIG welds may also be influenced by thickness. The strength values for the 6xxx series are valid after 3 days due to the occurrence of natural ageing. The welded materials must be held above 10 °C to fulfill the ageing phenomenon [2].

It is noteworthy that the strength reduction for the tensile strength (TS) is greater than the loss of YS in the aluminum HAZ, and the strength characteristics after MIG and TIG (up to 6mm) for the 5xxx series are equivalent. The difference in strength between plate (PL) and the two other products (extruded tube and drawn tube) may be regarded as negligible, an exception is however observed when comparing the TS of the products for the alloy 5083-H116.

There are also listed strength characteristics of weld metal in the Eurocode [2]. The strength of the weld metal is normally lower than the base metal, but higher when compared to the HAZ. The strength of weld metal, which is tabulated in table 3.8, are applicable given that the recommendations presented in table 3.11 are followed [2].

Table 3.8: Strength of weld metal for alloy 5083 and 6082 [2].

Filler material	5083	6082
Type 4	-	190 MPa
Type 5	250 MPa	210 MPa

As mentioned, the strength in the HAZ is for most cases lower than the weld metal, one should therefore be more concerned about designing structures to accommodate the lower HAZ strength, however both strength characteristics must be accounted for in structural design [54].

Table 3.9: Strength reductions in HAZ [2].

		MIG ¹⁾		TIG ²⁾		TIG ³⁾	
		YS [%]	TS [%]	YS [%]	TS [%]	YS [%]	TS [%]
5083-H116	PL ⁴⁾	28	10	28	10	35.1	18.9
	DT ⁴⁾	32.5	3.6	32.5	3.6	39.3	13.2
6082-T6	PL ⁴⁾	51	38.3	60.7	50.6	68.6	60.7
	ET ⁴⁾	52	40.3	61.5	52.3	69.2	61.9

1) Up to 15 mm thickness.

2) Up to 6 mm thickness.

3) Thickness between 6-15 mm.

4) PL = Plate, ET = Extruded tube, DT = Drawn tube.

The extent of the HAZ for different shapes and welds are measured as shown in figure 3.20. The HAZ is larger and more severe in TIG welds due to greater heat input. Furthermore, values of the HAZ wideness varies with thickness; increasing thickness leads to a more severe HAZ [2]. The extension of HAZ is tabulated in table 3.10 for both MIG and TIG welding. The values given in table 3.10, are coinciding with figure 3.20.

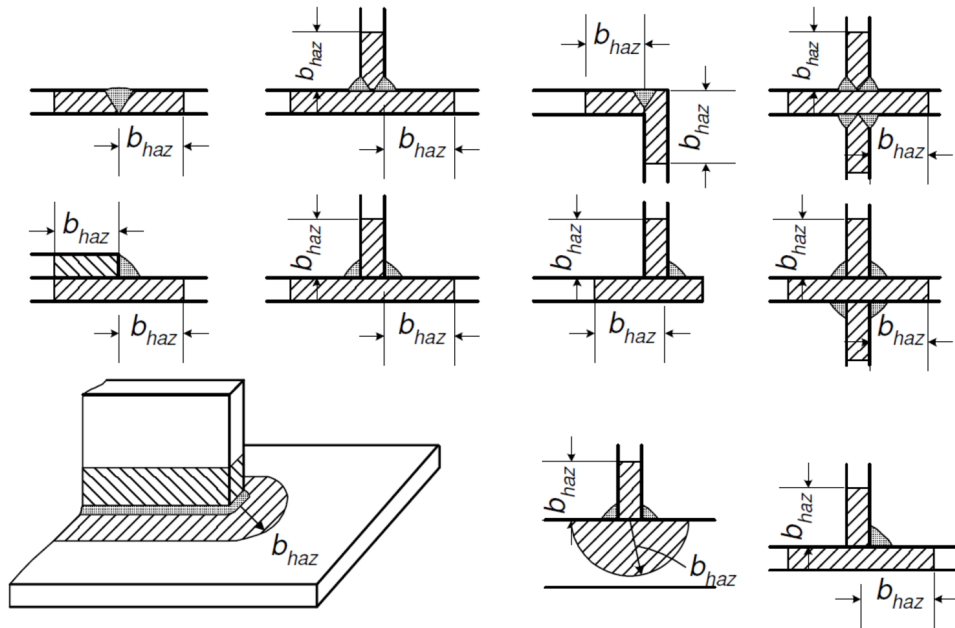


Figure 3.20: Extent of HAZ for different welding geometries [2].

Table 3.10: Extent of HAZ for different base metal thicknesses [2].

Thickness	0-6mm	6-12mm	12-25mm	above 25mm
MIG	$b_{HAZ} = 20\text{mm}$	$b_{HAZ} = 30\text{mm}$	$b_{HAZ} = 35\text{mm}$	$b_{HAZ} = 40\text{mm}$
TIG	$b_{HAZ} = 30\text{mm}$	-	-	-

According to Eurocode 9 [2], the filler metal may be chosen based on weld strength, corrosion resistance or weld cracking [2]. Table 3.11, summarizes the recommended filler materials for different base metal combinations and demands. Primary filler material selection should be alloy 5183 for both 5xxx and 6xxx series according to NORSOK M-121 [44]. The consumption of filler material should be frugal, meaning that one should not overfill welding grooves as filler material is quite expensive (the same apply for cover gas) [9].

Table 3.11: Filler materials [2].

		5083	6082
5083	Weld strength	5556A	Type 5
	Corrosion resistance	Type 5	Type 5
	Weld porosity	Type 5	type 5
6082	Weld strength		Type 5
	Corrosion resistance		Type 4
	Weld porosity		Type 4
Type 5:	5056A, 5356/5356A, 5556A/5556B, 5183/5183A		
Type 4:	4043A, 4047A		

There will always exist weld deformations to some extent, one should therefore be aware of precautionary actions. Firstly, it is important to make sure that the bonding components are sufficiently fastened (e.g. clamped) to reduce welding stress that may cause cracking, however in some rare cases where it is of interest to perform corrective measures it is preferable to avoid any constraints in order to reduce residual stresses [9]. Tack welding should also be commenced sufficiently to make sure the welding process takes place at the desired position. Welding speed should be as high as possible and prior bending may neutralize weld deformations in some cases [9]. For load bearing structural members, butt welds should be fully penetrated [2].

3.6.2 Friction stir welding

Friction stir welding is a new version of the conventional rotary friction welding. It was developed at The Welding Institute, Cambridge in 1991. The method was originally aimed for solid-state joining of aluminium but has now been successfully applied on other harder metals and plastic as well [29].

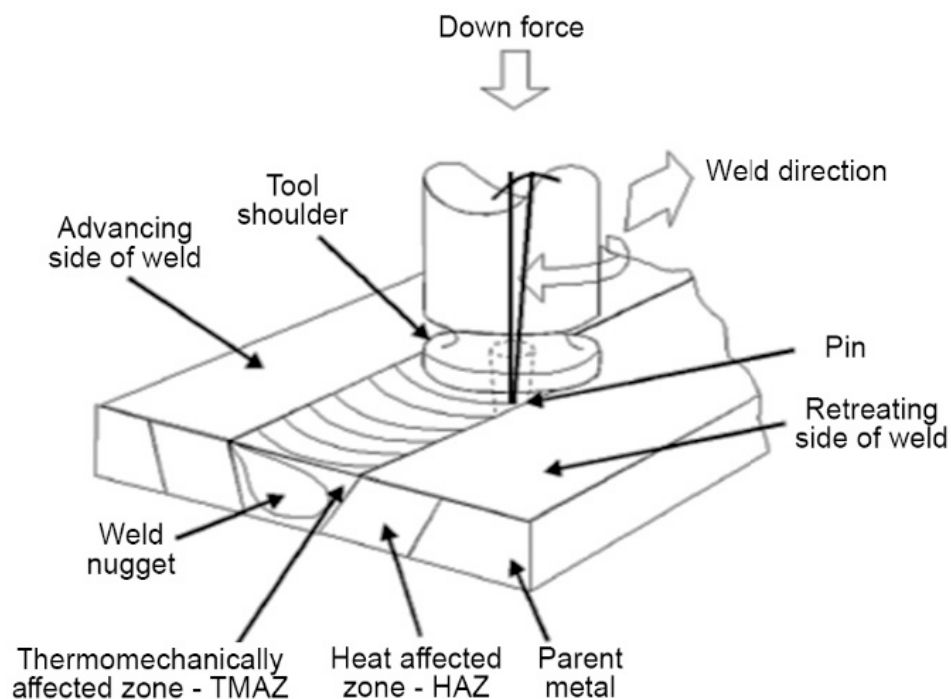


Figure 3.21: Friction stir welding process [28].

The principle of FSW is to use a rotating pin tool with a shoulder to generate heat from friction and plastic deformation of the substrate [29]. The substrates have to be rigidly clamped to avoid misalignment. The rotating pin generates a material flow from the front to the tailing end where the metal is forged into a joint, this material flow causes a plastic deformed area around the weld, as shown in figure 3.22. Notice the plastic zone surrounding the pin is uneven, caused by the rotating direction of the pin, where one side is called advancing side (AS) and the other is called retreating side (RS) [29], figure 3.21 shows the principle of FSW.

Microstructure

The friction stir welded material can be divided into four different microstructural zones according to how they are affected by the welding [29].

1. Base material, BM in figure 3.22.
2. Heat affected zone, HAZ in figure 3.22.
3. Thermo-mechanically affected zone, TMAZ in figure 3.22.
4. Stir zone, also called weld nugget, SZ in figure 3.22.

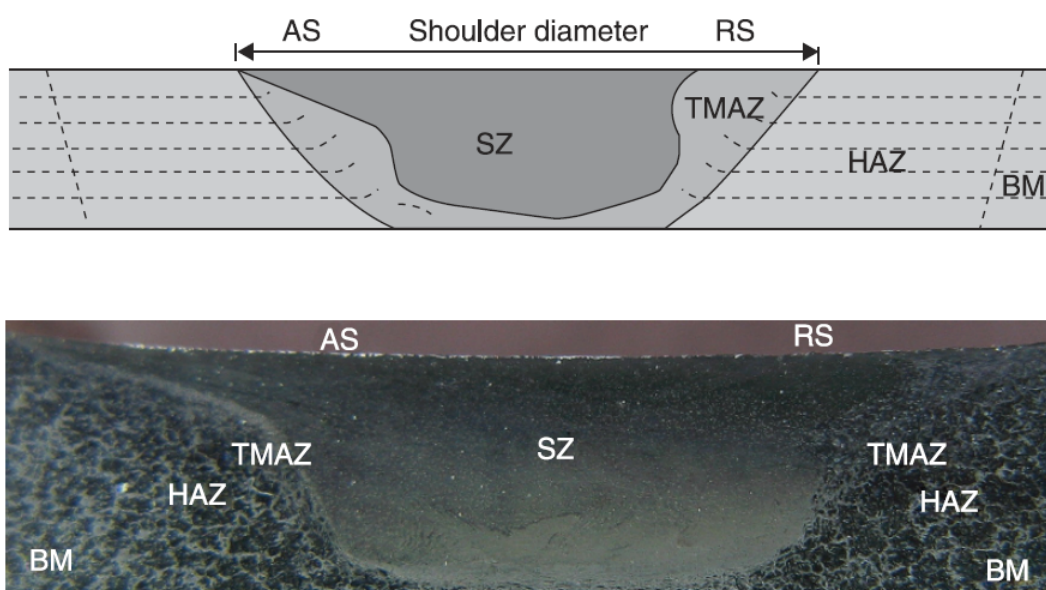


Figure 3.22: Microstructure after friction stir welding [29].

The base material is not affected by the weld, hence there is no significant change in the materials microstructure in this zone [29]. The HAZ is only heat affected by the welding process, and there is no mechanical deformation. The thermo-mechanically affected zone (TMAZ) is affected by mechanical plastic deformation, without undergoing recrystallization. There is no recrystallization due to insufficient temperature and insufficient mechanical strain. The stir zone (SZ) has undergone recrystallization due to high temperature caused by frictional heat and high plastic deformation. The transition from stir zone to base material is relatively smooth and continuous on the RS side, while the transition is sharper on the AS side of the weld [29].

Mechanical properties are changed and some of the most important properties will be significantly reduced. Especially with respect to yield strength, ca. 50% decreased for alloy 6082-T6. Tensile strength is reduced by ca. 30% for alloy 6082-T6. The FSW effect on mechanical properties for alloy 5083-O and 6082-T6 can be seen in table 3.12. The material hardness is reduced as seen in figure 3.23, showing the Vickers hardness (HV) against the distance from weld centre for alloy 6082-T6 and alloy 6061-T6 [29].

Table 3.12: Mechanical properties of FSW.

Material		TS [MPa]	Efficiency [%]	YS [MPa]	Reference
5083-O	BM	285-298	95-119	-	[62]
	FSW	271-344		-	
6082-T6	BM	323	68	276	[63]
	FSW	224		134	

Friction stir welding has been introduced into several industries working with aluminium. For example, Marine Aluminium has a FSW machine capable of welding 15 meters long large sized panels [64] and SpaceX has a special made FSW machine to weld their Falcon 9 rockets [65].

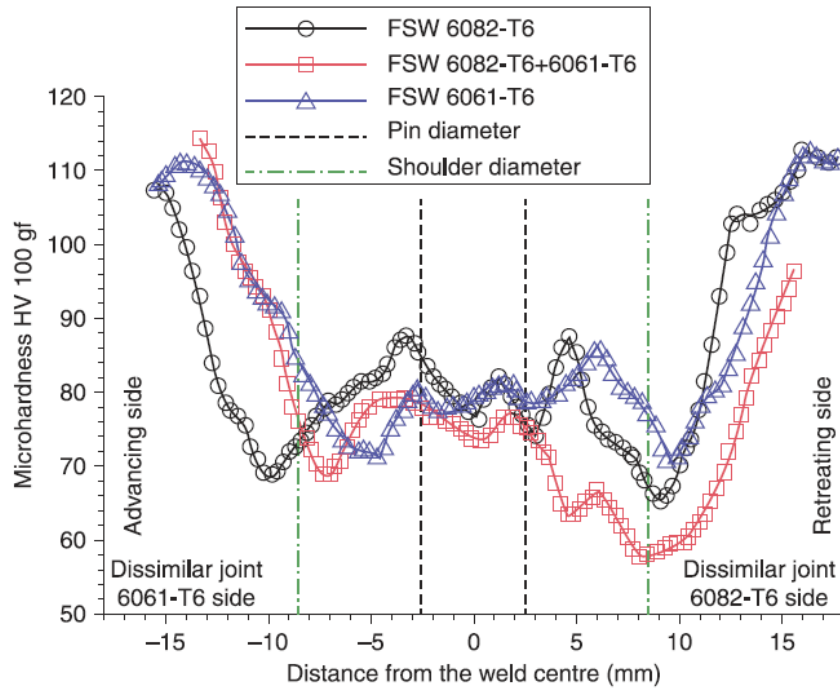


Figure 3.23: Comparison of Vickers hardness after friction stir welding for alloy 6082-T6 and alloy 6061-T6 [29].

Rotary friction welding

In principle, rotary friction welding (RFW) bonds two members by rotating one member against the other with a specified rpm and pressure [66]. The heat generated from the friction softens both surfaces, which leads to material plasticity. After sufficient friction heat and plasticity are reached, the rotating components are forced together by applying hydraulic pressure. The pressure may be held for a period after the rotation stops, sometimes referred to as forging force. The interface material is consequently extruded outwards, which leaves behind clean and pure metal in the whole cross-section of the joint. The base metal is not melted during this process, which means the weld is formed in a solid state. A typical application for this bonding method is joining of steel axles in vehicles [66].

It is noteworthy that the microstructure for a RFW is very similar to the one caused by FSW. The resulting strength properties of RFW are dependent on variables such as heat characteristics and welding time etc. Based on such process variables, strength values can be obtained for the HAZ by using appropriate process models [66].

3.6.3 Hybrid metal extrusion and bonding

A new patented process called hybrid metal extrusion and bonding (HYB) is an innovative low temperature solid state method that allows joining without any strength reductions in the HAZ [30]. The technology developed at NTNU is still in an early phase. However, in 2014 Statoil and Cardo Partners became part of the owner group in order to implement the new technology in fabrication of the new Johan Sverdrup oil field [30].

The principle of the HYB method is illustrated in figure 3.24. The filler metal is plasticized where it interacts with the base metal [30]. The oxide layer on the base metals to be bonded are subsequently removed during the process. Removal of the oxide layer is beneficial in terms of interatomic bonding between the atoms. The process temperature is below 300 °C and the filler metal possess the same properties as the base metal. HYB do not require heavy clamping, as required for FSW due to large reaction forces during welding. Low reaction forces is advantageous in terms of robotic automation [30].

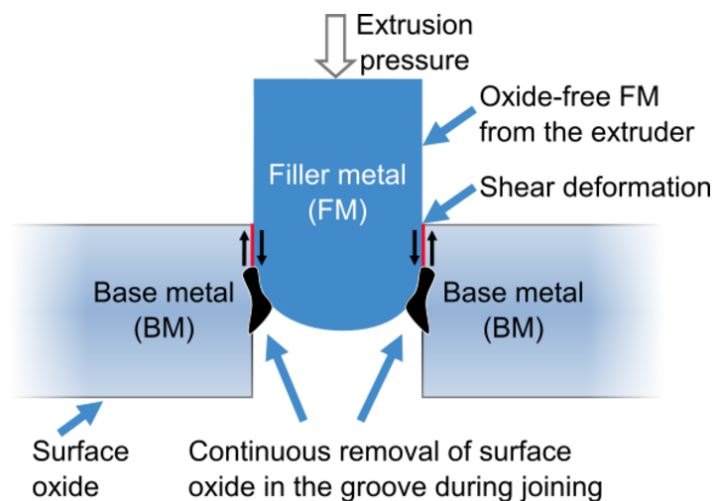


Figure 3.24: Hybrid metal and extrusion [30].

3.6.4 Laser welding

Laser welding (LW) is a process that requires no contact between the bonding members, it is a so called non-contact process [27]. The heat created by the concentrated laser beam causes coalescence of the joining materials. The type of lasers that are mostly seen in industries are the Nd:YAG (solid state laser) and CO₂ (gas laser) lasers where the YAG possesses a wavelength 10% of the latter [27]. The Nd:YAG laser light may be transmitted through fiber optics instead of copper used for the CO₂ laser light [31]. Gas shielding (i.e. helium and argon) is usually applied to protect the melted material from oxidation. The fact that this type of welding only require access from one side of the joint impose low-heat input that subsequently cools rapidly [10], and bonds without filler material, should be regarded as advantages. The laser joining process may be conducted with a pulsating or continuous beam [67].

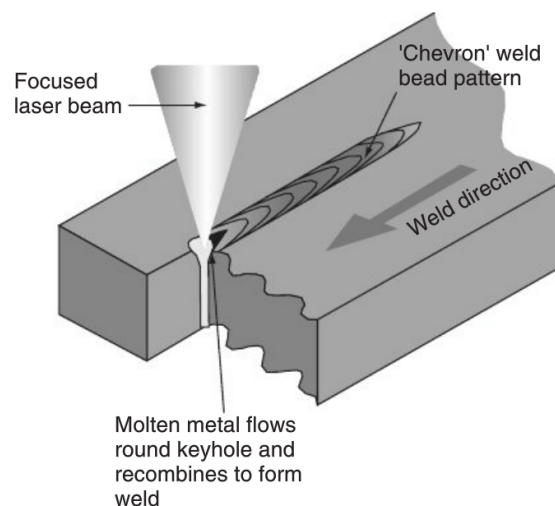


Figure 3.25: Laser keyhole welding principle [31].

There are some important limitations concerning the use of LW that should be mentioned. Magnesium and zinc, that are often seen in aluminium alloys with significant amounts, are quite easily vaporized from the heat of the laser, which may lead to formation of a beam blocking plasma layer [68]. Entrapped vapour pores and hydrogen may also occur during LW. Gas shielding are usually implemented to withstand the hydrogen diffusion, where helium is preferred for high quality welds at maximum depth applications [68]. Pores that causes porosity often seen in LW [31] may also arise from vaporization of alloying elements, for instance magnesium [67].

Pre-surface cleaning is very important for LW as a preventive measure for porosity [10]. Removal of deteriorating contaminants that may alter characteristics of the weld, and the fact that the LW process does not have the ability to remove oxide during welding. Aluminium is in addition one of the best light reflectors, which complicates the LW process due to a low power absorption. The reflection effect together with the relatively high temperature expansion coefficient of aluminium leads to a requirement for using much more powerful lasers for welding aluminium than for steel, although the absorption slightly increases when aluminium liquefies [67]. The Nd:YAG laser are preferable to use in order to overcome the reflection issues [68]. A disadvantage is the high fit-up tolerances required. The strict fit-up requirements are tabulated in the ISO 13919-2 standard [69] for multiple cases. Moreover, LW generally requires a significant investment [67].

With the use of high energy density lasers (above 40 kW/mm²) one is able to produce welds in the so called keyhole regime, which improves the absorption due to a positive reflection effect within the cavity [31] seen in figure 3.25. The deep keyhole penetration reduces the width and strength losses of HAZ, minimizes distortion and the loss of alloying elements containing a lower boiling point are reduced (e.g. magnesium) [31]. It is important to have control over the power in order to create a stable keyhole. Another difficulty is the relatively low viscosity of the liquefied weld metal that may lead to drop-through of the molten metal. One can overcome this problem by welding in the horizontal-vertical position [31].

Strength in the as welded condition and in the base metal together with the respective strength reduction of laser welding are tabulated in table 3.13.

Table 3.13: Laser weld strength characteristics¹⁾ [10].

Alloy	Strength reduction	
	YS [%]	TS [%]
5083-H21	7	11
6061-T6	25	34

1) CO₂ laser welding on 2-3 mm thick plates (butt weld).

3.6.5 Hybrid laser

Laser and the conventional arc welding can be combined to form a hybrid laser. The most popular combination is Nd:YAG laser implemented with MIG. This type of hybrid laser enables improved fit-up tolerances i.e. positioning of the components to be welded and the conventional welding speed possessed in arc welding is improved [27]. The Nd:YAG laser creates a keyhole ahead of the MIG that is depositing the weld metal, as illustrated in figure 3.26. It is claimed that the characteristic shape of the weld pool from a hybrid laser enables hydrogen to escape the joint, which reduces incidents of hydrogen entrapment and thereby reduces porosity defects [31].

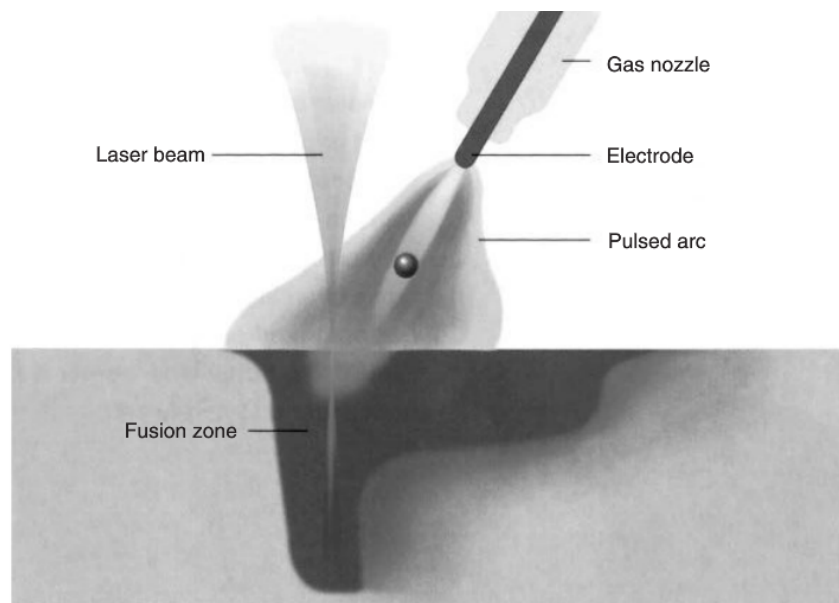


Figure 3.26: Hybrid laser principle [31].

3.6.6 Summary of welding methods

A brief summary of the advantages and disadvantages for the welding methods described in section 3.6 are presented in table 3.14. Brazing is excluded from table 3.14, as it is not a welding technique.

Table 3.14: Summary of welding methods.

Method	Advantages	Disadvantages
MIG	Commercial and cost efficient. Hand held apparatus.	30-50% strength reduction.
TIG	High quality welds.	Requires highly skilled welders. Larger HAZ than MIG.
FSW	High speed welding Brilliant strength and ductility characteristics.	Only applicable to butt welds. Bonding members must be rigidly clamped. Not a hand held apparatus
LW	Good strength characteristics.	Costly method to implement. Light reflection complicates the use of laser.
RFW	Brilliant strength and ductility characteristics.	Only applicable to joining of circular members. Not a hand held apparatus.
Hybrid laser	Improved fit up tolerances, and the welding speed of the conventional MIG method is improved.	Costly method to implement.
HYB	Joining without any strength reductions.	Method is recently developed. Limited literature on the technology.

3.7 Brazing

Joining aluminium alloys by brazing is a method that does not involve melting of the structural components, the parts are instead joined by a filler material.

Brazing differs from soldering by the fact that brazing filler material for aluminium alloys are always aluminium-based, and the heat required for the brazing process is considerably higher. The brazing filler material typically exhibit a composition that is just below the melting temperature of the base metal. By definition from DVS (Deutscher Verband für Schweißtechnik), the brazing and soldering process is distinguished according to an applied

temperature of 450 °C [56].

The filler metal distributes throughout the joint surfaces at brazing temperature. During this process the filler and base metal will be combined by inter-atomic attraction, which result in a permanent metallic bonding [56]. The bonding may also partly be a result of an atomic interchange between the base and filler material due to diffusion at the working temperature. Diffusion may also lead to a reduced mechanical strength as the intermetallic phases are precipitating [56].

The heat required for the brazing process of aluminium alloys leads to a strength reduction of the base metal surrounding the brazed joint, in the magnitude of 50-60% reduction of its initial YS [56].

As for welding, the oxide layer on aluminium alloys complicates the brazing process. Removal of the oxide layer must therefore be commenced before initiation, and the bare surface needs to be protected against post oxide formation. Thick oxide layers has to be removed either chemically or mechanically. However, thinner layers may be displaced by brazing flux [56].

Aluminium alloys are generally restricted to filler material based on aluminium together with silicon. Hence, the 2xxx and 7xxx aluminium series are not suitable for brazing, as the melting temperature for these alloys are too low. The filler metal used in brazing may lead to a more corrosive behavior in seawater as it may possess a more noble potential in the galvanic series than the base metal. Furthermore, alloys with a raised amount of magnesium, more or less above 1-2% (e.g. 5083-H116), are more challenging to braze due to a more aggressive oxide development that is more complicated to handle by common removal methods.

3.8 Bolting

Bolted sections enables joining without any significant strength reduction in the base metal of bonding sections, as is the case for joining methods (e.g. welding) that imposes enough heat to degrade strength properties of aluminium alloys. No reduction in strength is an ad-

vantage, however bolting imposes some drawbacks.

Bolted joints in subsea applications are in danger of encountering crevice corrosion and/or galvanic corrosion, as described in section 3.5.1 and 3.5.6, respectively. Protective measures must therefore be applied to all connections that possess a crevice (bolt heads, nuts, washers) and/or galvanic coupling between various materials used in the joint. Sealing compound, coating and CP are considered effective protection methods [2, 20]. The sealing compound shall function as an external seal to prevent water ingress in openings and cracks of the assembly. All contact surfaces of the bolt assembly shall be free of coating to ensure adequate friction and electrical continuity throughout the bolt assembly [11]. See figure 3.27 for illustration of two bolted plates made of steel and aluminium with sealing compound. The components in the bolt assembly should have an electric resistance of less than 0.1 ohm with respect to an anode to ensure proper CP of the joint members. The continuity shall be verified by actual measurements [11].

Submerged carbon steel bolts should have a protective layer of phosphating or coated with poly tetra fluoro ethylene (PTFE) as long as electrical continuity is verified [54]. Phosphating is therefore advantageous in terms of electrical continuity. Hot dip galvanizing should not be implemented on structural bolts, as it may cause loss of pretension in the bolt due to dissolution of zinc layer [54]. Stainless steel bolts do not require any coating [54]. See appendix B.3 for e-mail correspondence with Ole Øystein Knudsen regarding CP of phosphated bolts.

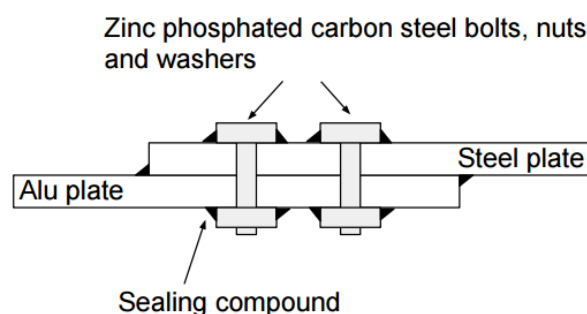


Figure 3.27: Bolt assembly with CP between aluminium and steel with zinc phosphated carbon steel bolts and sealing compound.

If elements in the bolt assembly (or structure) do not possess an adequate continuity (<0.1 ohm) one can use copper wires that can be fastened either by welding or bracing at each end,

or with braced cable shoes fitted to desired bolt size. Minimum cross section of the copper wire is 16 mm^2 [11].

It must be decided if bolted joints shall be slip resistant or not if they are subjected to shear loading, and if the bolt(s) shall be preloaded or not for tension loading. A slip resistant and preloaded coupling requires more bolt tightening, which further implies the bolt type to be used. For the case of slip resistant connection the yield strength of the bonded material (except bolts) shall not be lower than 200 MPa [2]. According to NORSOK M-001 [54], bolts for subsea applications shall not have a higher hardness than 300 Brinell hardness number (HB) and strength class 8.8. Mechanical properties of grade 8.8 carbon steel bolts are tabulated in table 3.15. The hardness of the carbon steel bolt is ranging from 270-331 HB, it must therefore be ensured that the bolts are manufactured with a hardness below 300 HB. The hardness of stainless steel 316 is below 300 HB [70]. Properties for stainless steel bolts grade A4 (stainless steel 316) are also presented in table 3.15. Note, for subsea structural applications the bolts should generally be made of carbon steel [54].

Table 3.15: Properties of 316 stainless and 8.8 carbon steel bolts.

Material	Grade	Property class	Yield strength	Tensile strength	Elongation after fracture
Carbon steel ¹⁾ [71]	8.8	-	640 MPa	800 MPa	12 %
Stainless steel 316 [72]	A4 ³⁾	70	450 MPa	700 MPa	$0.4d^{2)}$ mm

1) Valid for bolt diameter larger than 16 mm.

2) d = Bolt size.

3) Marine grade [72].

Bolting requires a higher number of tolerances and alignments than welding. Holes must be machined in an exact position, both for structural members and the bonding components between them. Clearance between bolt and machined holes should not be more than 3 mm for high strength slip resistant connections [73].

The recommended preloading per bolt in slip resistant joints can be calculated according to equation 3.3 [2], and the resulting slip resistance per bolt (F_{sr}) are found from equation 3.4. If the bolted connection are subjected to additional tensile force, equation 3.5 should be used as the tensile force lowers the slip resistance. The 1.25 fraction represents the safety

factor at ultimate limit state [2]. For total joint thicknesses larger than 30 mm, the slip factor can be set to 0.4 [2].

$$F_p = 0.7\sigma_{ub}A_s \quad (3.3)$$

Where:

F_p = Preloading force.

σ_{ub} = Ultimate strength of bolt material.

A_s = Bolt tensile stress area.

$$F_{sr} = \frac{n\mu F_p}{1.25} \quad (3.4)$$

$$F_{sr} = \frac{n \cdot \mu (F_p - 0.8F_t)}{1.25} \quad (3.5)$$

Where:

n = Number of friction interfaces.

μ = Slip factor.

F_t = Required tensile force per bolt for ultimate limit state.

The required shear ($F_{s,i}$) and tensile force ($F_{t,i}$) (per bolt) for a bolt group subjected to combined shear and moment are calculated from equation 3.6 and 3.7 respectively, where the first are valid for in-plane loading and the latter for out-of-plane loading [32]. See illustration of the two load cases in figure 3.28 and 3.29.

$$F_{s,i} = \sqrt{\left(\frac{P_x}{n} + \frac{M \cdot y_i}{\sum (x_i^2 + y_i^2)}\right)^2 + \left(\frac{P_y}{n} + \frac{M \cdot x_i}{\sum (x_i^2 + y_i^2)}\right)^2} \quad (3.6)$$

P_x and P_y are the vertical and horizontal shear loads acting on the bolt group. Number of bolts are denoted n , and M is the moment around the centre of the bolt group. x_i and y_i

represents the orientation of each bolt relative to the centre of rotation in figure 3.28. The bolts must be orientated with positive or negative values as appropriate. Use of equation 3.6 implies three assumptions: Plate deformations are negligible, total shear is distributed equally, and shear from applied moment is proportional to the distance between bolt and center of rotation [32].

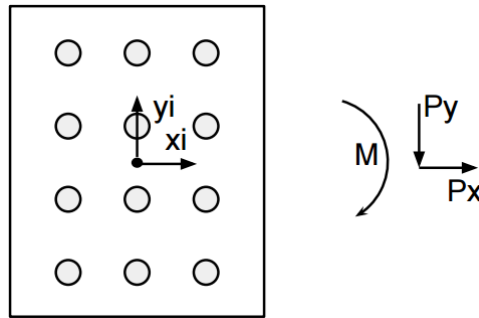


Figure 3.28: In-plane moment.

$$F_{t,i} = \frac{M \cdot l_i}{\sum(l_i \cdot L_i)} \quad (3.7)$$

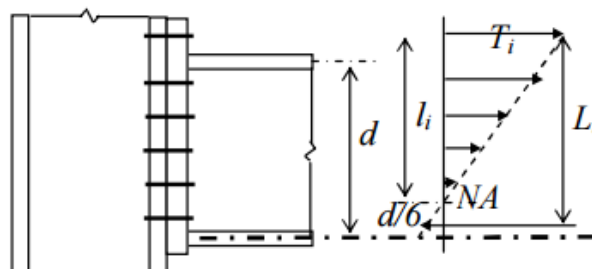


Figure 3.29: Out-of-plane moment ($T_i = F_{t,i}$) [32].

In the case of an out-of-plane loading, the point of rotation is assumed to be one-sixth of the beam height above the bottom of the beam, where $F_{t,i}$ is assumed to be proportional to the distance from point of rotation. However, if there is a hard spot in the lower area of the beam such as a flange, the point of rotation may be assumed to act in the middle of that spot. In equation 3.7, the distance between bolt "i" and center of rotation is denoted l_i and the distance between the lowest bolt and bolt "i" is denoted L_i . The equation assumes proportional relation between $F_{t,i}$ and l_i , as illustrated in figure 3.29 [32].

For structural members limited by available lengths due to for example fabrication or transport restrictions, one may combine them with the use of beam splices illustrated in figure 3.30. Beam splices are known to resist large shear forces and bending moment. If a beam splice joint is oriented at a point with somewhat lower moment, one can make an assumption where the flange splice takes up the moment and the shear force is carried by the web splice [32]. However, if that assumption is not accurate, the moment should be split between the flange and web in accordance with the stress distribution between them. In the case of shared moment, the web splice is subjected to both moment and shear loading.

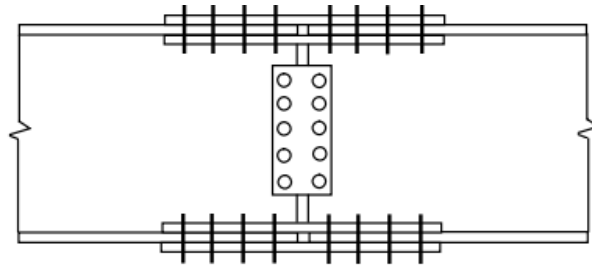


Figure 3.30: Beam splices [32].

The respective design resistance for bolts subjected to shearing (σ_{sr}) and/or tension (σ_{tr}) are found from equation 3.8 and 3.9. $\alpha_s = 0.6$ for class 8.8 steel bolts and 0.5 for stainless steel bolts. For the case of combined shearing and tension, the criteria in equation 3.10 shall be fulfilled [2].

$$\sigma_{sr} = \frac{\alpha_s \sigma_{ub}}{1.25} \quad (3.8)$$

$$\sigma_{tr} = \frac{0.9 \sigma_{ub}}{1.25} \quad (3.9)$$

$$\frac{F_{s,i}}{\sigma_{sr} A_s} + \frac{F_{t,i}}{1.4 \sigma_{tr} A_s} \leq 1 \quad (3.10)$$

For bolted connections there are specified minimum, regular and maximum spacing be-

tween bolts and edges. The maximum values are strongly dependent on exposed environment, where corrosive environments lowers the maximum value due to a need for proper closing of the connection, which hinders the occurrence of crevice corrosion. See figure 3.31 and table 3.16 for approved spacing values [2]. The maximum values are also influenced by the susceptibility of encountering buckling during compression, as described in section 3.11. Local buckling between fasteners are considered to be satisfactory if the ratio p_1/t is lower than 9ϵ , see eq. 3.15 in section 3.11 for calculation of ϵ . The edge distance e_2 is tolerable as long as it is not exceeding the maximum value. However, the distance e_1 do not affect the occurrence of local buckling [2], and placement of fasteners should enable enough space for tightening tools.

Table 3.16: Minimum, regular and maximum values for spacing between bolts and edges [2].

Spacing	Minimum ¹	Regular ¹	Maximum ² (corrosive environment)
e_1	$1.2d_0$	$2.0d_0$	$4t + 40 \text{ mm}$
e_2	$1.2d_0$	$1.5d_0$	$4t + 40 \text{ mm}$
p_1	$2.2d_0$	$2.5d_0$	Smallest of $14t$ or 200 mm
p_2	$2.4d_0$	$3.0d_0$	Smallest of $14t$ or 200 mm

1) d_0 = Diameter of hole.
2) t = Thinnest thickness of the outer connected member.

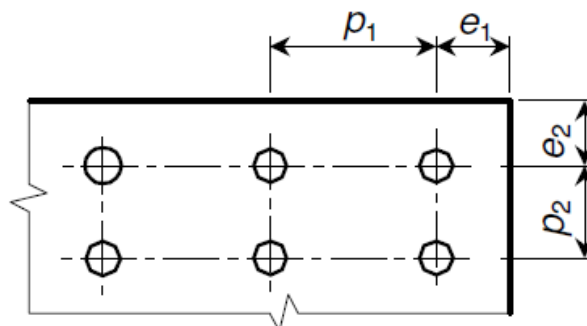


Figure 3.31: Fastener spacing [2].

Tightened bolts shall be re-tightened after 72 hours due to effects like creep, relaxation and settlement [73]. The applied tightening torque shall be continuously and smooth [74].

3.9 Adhesives

As for bolting, adhesive bonding does not negatively alter the mechanical properties of jointed structural members. The use of adhesives also enables joining of different materials with a wide range of properties. Furthermore, adhesive joints possess an excellent uniform stress distribution. A drawback is adhesive's sensitivity to different environments, and the comprehensive validation required before any use of adhesives in structures [27]. Figure 3.32, illustrates the structure of an adhesive bonded joint.

According to Eurocode 9 [2], structural joints may be connected with the use of adhesives. The bonding requires an expert technique and it should only be used with care. For main structural joints comprehensive environmental and fatigue testing must be approved in order to establish its validity. A combination of plate and stiffeners may be more suitable for adhesive joining. In general, adhesives should only be subjected to shear stress [2].

The strength of the adhesive itself may be measured according to tests described in the standard ISO 11003-2 [75]. However, it is not enough to only have insight in the specific strength of the adhesive to ensure the strength of the joint, it has to be tested thoroughly in terms of strength through laboratory tests, where pretreatment, adhesive, environment and ageing are accounted for. If an adhesive joint passes the laboratory tests, it has to be validated even further in real tests conditions. The results obtained at the laboratory should only be treated as advisory [2].

High strength adhesives recommended by Eurocode 9 [2] for structural applications are illustrated in table 3.17. These values are based on extended research. Higher values for strength of adhesives may be utilized if the respective adhesive is tested and approved according to ISO 11003-2 [2, 75]. A safety factor of 3.0 is imposed [2]. Adhesives combined with riveting or bolting may increase the joints robustness [27].

Table 3.17: Strength values of recommended adhesives [2].

Adhesive types	Strength [Mpa]
Modified epoxide, heat cured, 1 component	35
Modified epoxide, cold cured, 2 components	25
Modified acrylic, cold cured, 2 components	20

Pre-treatment is very important for an adhesive joint, especially if it has to perform over extended periods, and the durability in environments that may degrade the adhesive is highly dependent on the chosen pretreatment. Examples of environments that may degrade the quality and/or contribute to ageing of the adhesives are waterish environments (saltwater), humid atmosphere and other aggressive environments [2]. Delamination process in the interface of an adhesive bonded joint are schematically illustrated in figure 3.32. Adhesive failure occurs in the metal-adhesive interface due to environmental degradation and stress, which can ultimately lead to crevice corrosion in the interface [27].

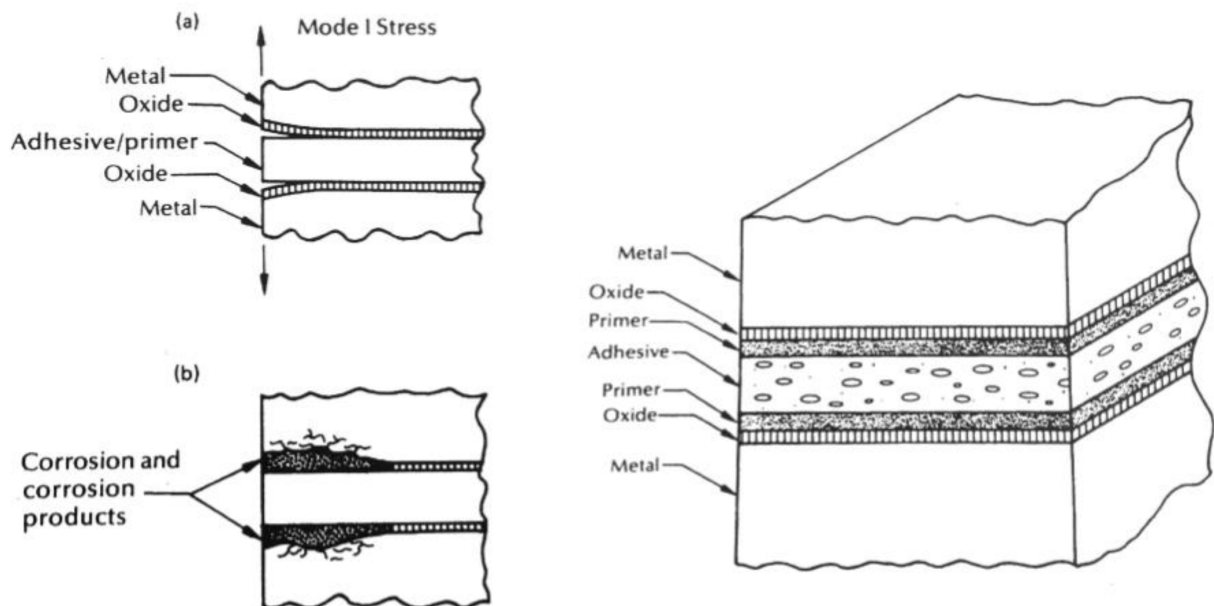


Figure 3.32: Right figure: Structure of an adhesive metal to metal joint. Left figure: Delamination mechanism of an adhesive bonded joint: a) The oxide /primer interface is weakened due to stress and water. b) Crevice corrosion products accelerates the delamination [27].

3.10 Formability

Aluminium is generally easy to form and machine, especially compared to steel. Bending of aluminium are roughly performed the same way as for steel, however one must be careful when it comes to the relatively soft aluminium surface that may be accumulated with metal particles during the bending process, which may lead to local corrosion [9]. Preventive actions are proper cleaning and use of lubricants.

Plates are normally formed when cold. Generally when bending aluminium it is of importance to make sure the bending radius is not too small. The radius requirement is dependent upon thickness and alloy [9]. Too small bending radius may cause fracture, especially for high strength alloys with low ductility and plastic necking (decreased thickness) will always occur in the bend zone. Aluminium must be bent more than the desired angle due to an elastic return effect. Minimum bending radius for a 12.5 mm 5083-H116 alloy plate with a 90° angle is 4 four times thickness [76].

Tubes and profiles may be bent when cold, which is preferable considering loss of mechanical properties when heated. As for plates, the radius should not be too small for profiles and tubes, however the radius is also dependent on the profile shape. The minimum bending profile and tubing radius, R , for alloy 6082-T6 can be calculated according to formula 3.11 and table 3.18 [9]. z is the external diameter or cross sectional profile height. If the relation z /wall-thickness is greater than 20, the table values are not valid. All components to be bent should be bent uniformly. Profiles can be made with stiffeners in order to avoid buckling of the profile that is exposed to compression. Tubes with thin walls may be filled with sand or alloys with low melting point. Generally, tube benders and jigs are the recommended bending processes for both profiles and tubes [9].

$$R = \sqrt{\text{Bend factor} \cdot z^2} \quad (3.11)$$

Table 3.18: Minimum bend factors for cold bending of alloy 6082-T6 tubes and profiles [9].

Tube	Profile
2.0-2.5	2.5-3.0

3.10.1 Machining

Aluminium is readily machined by commercial methods such as sawing, plasma cutting and water jet cutting. However, torch cutting should not be applied to aluminium, due to the HAZ created by the torch [9]. Water jet cutting has the advantage that it does not create any HAZ, as the process does not involve any heating of the material. Water jet cutting of aluminium alloys can be used on all components on the concerned ITS.

3.11 Buckling

The cross section buckling resistance for structural members can be assessed by identifying the appropriate buckling class described in Eurocode 9 [2]. The cross sectional classification assess if the resistance and rotational capacity is confined by its resistance to local buckling. The buckling class are found by calculating the width and thickness ratio of all the parts that may be subjected to compression under considered load combinations.

There are four buckling classes described in Eurocode 9, where the first one has the best buckling resistance. The classes are defined as follows:

- Class 1 posses the required rotation capacity for plastic analysis, due to formation of plastic hinge without reduction of resistance ($\beta \leq \beta_1$).
- Class 2 can reach their plastic moment resistance, but local buckling limits the rotational capacity ($\beta_1 < \beta \leq \beta_2$).
- Class 3 may reach its yield strength in the most compressed parts, but local buckling restricts full development of plastic moment resistance ($\beta_2 < \beta \leq \beta_3$).
- Class 4 develop local buckling before yield stress is reached in one or more compressed parts ($\beta > \beta_3$).

For class 4 cross sections, an effective thickness may be used to accommodate the low buckling resistance. The different compression parts of a cross section (e.g. web and flange) can in theory possess different classes, however the least favourable class apply for the whole

cross section [2].

The buckling calculation process is related to the member's slenderness, which is dependent upon the cross sectional profile, shape of the compressed parts, and whether it is internal or outstanding. The parts may also be reinforced (e.g. ribs, edge lips or bulbs), which further affect the calculation procedure. The web and flange of a symmetrical unreinforced I-beam can be calculated by finding the individual slenderness parameter, β , according to eq. 3.12 and 3.13, respectively [2]. The web is regarded as an internal part while the flange is an outstanding part. h is the height of the web, and b is half the flange width excluding the web width. Slenderness parameter of a tube can be found from eq. 3.14 (internal part), Where D is the mid-thickness diameter and t is the respective thicknesses. By comparing obtained slenderness parameter with the tabulated values in table 3.19, one will be able to find the right class for each compressed part of a cross section. See the class description for the respective slenderness criteria, and use equation 3.15 to find the required value for ϵ by inserting the yield strength of considered material.

$$\beta_w = 0.4h/t \quad (3.12)$$

$$\beta_f = b/t \quad (3.13)$$

$$\beta_t = 3\sqrt{\frac{D}{t}} \quad (3.14)$$

$$\epsilon = \sqrt{250/YS} \quad (3.15)$$

Reinforcement has potential in improving buckling resistance. Slenderness parameter for a reinforced flange (β_r) fitted with single sided rib or lip possessing the same thickness as the flange throughout an I-beam, can be calculated according to equation 3.16 [2]. c is the length of the rib or lip, measured from the internal edge of the flange. The other parameters are the same as described in equation 3.13. Table 3.19 also apply for equation 3.16.

$$\beta_r = b/(t\sqrt{1+0,1(c/t-1)^2}) \quad (3.16)$$

Table 3.19: Slenderness parameters [2].

Material classification	Internal parts			Outstand parts		
	β_1/ϵ	β_2/ϵ	β_3/ϵ	β_1/ϵ	β_2/ϵ	β_3/ϵ
Class A ¹	11	16	22	3	4.5	6
Class A welded	9	13	18	2.5	4	5
Class B ²	13	16.5	18	3.5	4.5	5
Class B welded	10	13.5	15	3	3.5	4

1) 6082-T6
1) 5083-H116

If a part happens to be classified as class four and it is not practical to enhance the local buckling by design it is possible to introduce a reduction factor (eq. 3.17), which gives an effective thickness. C_1 and C_2 varies with material class, if the part is welded or not and whether it is internal or not. For internal parts, $C_1=25$ and $C_2=150$ apply for alloy 5083-H116 with welds, and $C_1=29$ and $C_2=198$ for alloy 6082-T6 with welds [2].

$$\rho_r = \frac{C_1}{(\beta/\epsilon)} - \frac{C_2}{(\beta/\epsilon)^2} \quad (3.17)$$

3.12 Fatigue

The fatigue resistance compared to static resistance is significant for aluminium alloys, meaning that the fatigue resistance is significantly lower than static resistance. Fracture due to fatigue may occur without necessarily exceeding its strength capabilities if subjected to intolerable load cycles [52]. Typical initiation spots are welds, cracks, holes (pitting), notches, recesses, etc. Varying compressive stress cycles are usually not a problem with regards to encountering fatigue failure, however tensile and compressive-tensile load cycles are [52].

Design against fatigue failure may be performed according to one of the three following methods [77]:

- Safe life design
- Damage tolerant design
- Design assisted by testing

Safe life design means that the structure shall perform throughout its estimated lifetime without any required inspections. This method is a conservative fatigue life approach, which may be used on structures where it is difficult or not economically beneficial to perform inspections. The second method accepts small cracks, which is justified by fatigue crack monitoring in order to set up an appropriate inspection program. It is intended that the first and second method shall have the same reliability [77]. Damage tolerant design may be used where fatigue significantly affects design cost and in structures where it is permissible with somewhat more fatigue cracking than for safe life design, however inspections of the structure must be possible. If there is not obtained enough fatigue data for any relevant structure, one can design the structure against fatigue by performing tests in accordance with standards [77].

S-N design curves, stress range and load cycle must be established in order to assess the fatigue capacity of a structure. The S-N design curves are determined based on many factors such as: material quality, direction of extrusion, type of weld and its orientation relative to direction of stress, transitions in cross section, and other geometric factors [77]. The S-N curves which are constructed based on test, are plotted logarithmically as a function of stress range and load cycles, and can be found for many structural details. If the appropriate detail is nowhere to be found in approved standards, one can conduct full-scale fatigue testing on the missing details [77]. It is noteworthy that the S-N curves are not sensitive to different alloy compositions within the 5xxx and 6xxx series according to Eurocode 9 [77], however there may be different curves between the series itself due to lower durability rating in seawater for the 6xxx series. The curves do not distinguish between different temper designations of the alloys (e.g. T4 and T6) [9].

Structures may encounter resonant effects, which can cause magnification of dynamic stress. Slender structures with low natural frequency are particularly susceptible to this effect. In addition, structures supporting vibrating equipment (e.g. X-mas tree) should be carefully assessed for resonance [77]. If the structure is subjected to transport loads prior to its final destination, the integrity of the structure should be validated appropriately.

Fatigue damage during sea transport from wind and wave loads should be assessed if the structure is exposed to significant cyclic loading according to DNV-OS-H102 [78], which shall be based on specified environmental load history for weather restricted operations.

Fabrication quality of aluminium is very important for the fatigue capacity, and the relatively soft aluminium surface should be protected against any damage during fabrication, transport and assembly, as uniform surfaces possess better fatigue characteristics. Welds are often the most critical locations of encountering fractures due to fatigue (even when exposed to low loading), it is therefore very important that welding is performed appropriately in excess of inspections before installation [77].

Corrosion of aluminium alloys in seawater can affect fatigue crack growth, due to anodic dissolution at the apex of a crack (e.g. pit). Anodic dissolution is a relatively slow process, it is therefore most effective for small crack growth rates, meaning it effectively reduces the threshold region [52]. More on fatigue challenges are discussed in section 6.3.

Chapter 4

Integrated Template Structure Analysis

This chapter will focus on the practical design of the structure, where design possibilities by the use of aluminium alloys will be provided. The aluminium design will be based on the literature in chapter 3. The new size and geometry of the reviewed components are based on mechanical integrity, where the moment of inertia and yield strength are used as the most important design parameters, seen in detail in section 4.1. This approach is checked by modelling the structure in Autodesk Inventor and applying realistic design loads in a finite element method (FEM) analysis to check if the mechanical integrity is maintained and acceptable.

4.1 Design of beams

The mechanical integrity of the ITS has to be maintained in order to make a comparable analysis between steel and aluminium. All the beams are originally made from steel type S355, the aluminium redesign is therefore mainly based on increasing the moment of inertia (MOI) to accommodate the difference in yield strength. However, the final design is a result of an iterative optimization procedure between moment of inertia criteria, buckling, extrusion limits and weight savings. Applicable joining methods further affects the final outcome of the beam design. See section 3.2 for more information about mechanical properties of considered aluminium alloys.

Emphasis has been directed to the number of different extrusion dies needed for the whole structure to lower fabrication costs. The plate thicknesses are always divisible by five in order to ease plate procurement, as standard plate thicknesses can be used.

Each beam is given its own designation to organize the structure, an overview of all the beams designations can be seen in figure 4.1.

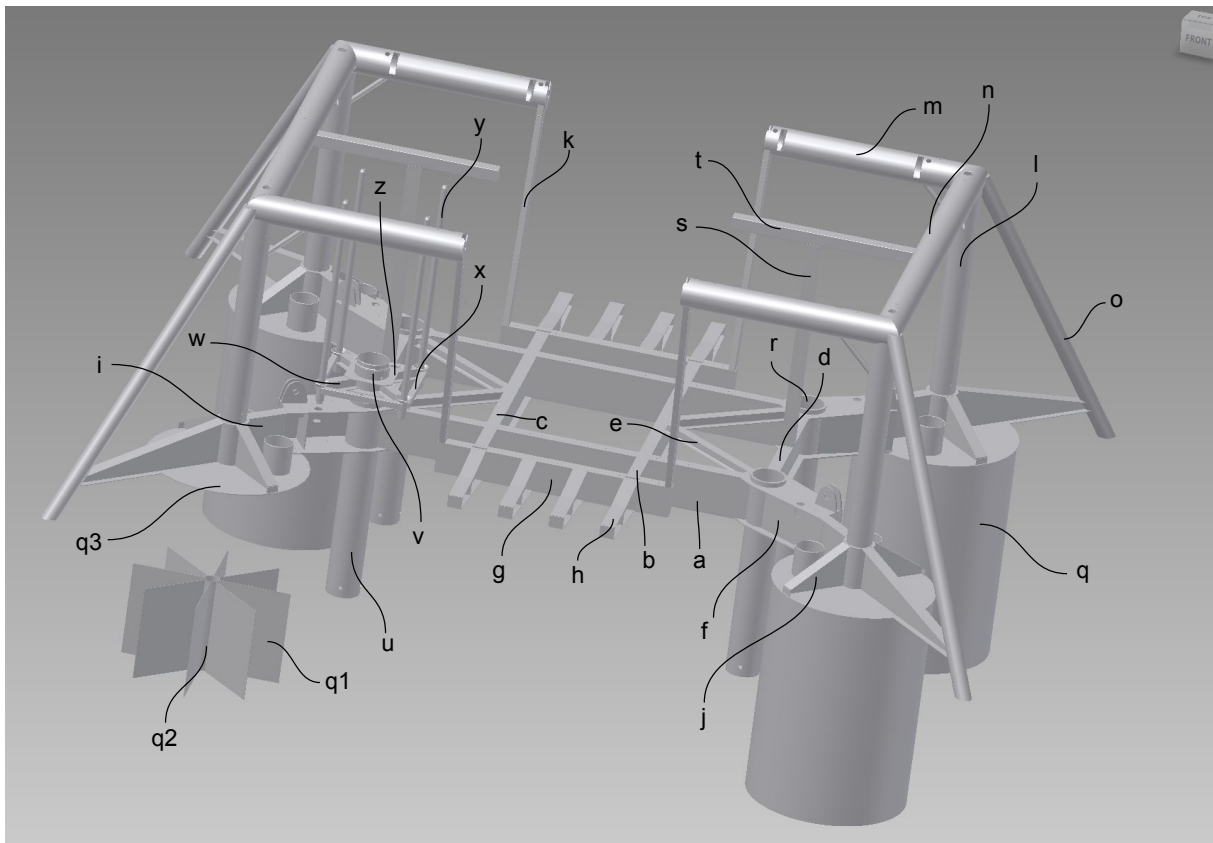


Figure 4.1: The ITS with identification letters, where beam 'a' refers to the beam marked with an 'a' etc.

Both original and new MOI for all beams along with the respective ratios in addition to buckling factor of the redesigned beams can be seen in table 4.1 . The calculations for all beams are attached in appendix D. The results in table 4.1 are based on a formula for mechanical bending stress (eq. 4.1), and the buckling assessment is based on Eurocode 9 [2]. More detailed information regarding local buckling resistance can be seen in section 3.11. The same method is used on beam number 'b' to 'y' as seen in table 4.1.

Table 4.1: MOI of beams, MOI ratios and buckling classification.

Beam ¹⁾	$I_{xx,old}$ 10^6 [mm ⁴]	$I_{xx,new}$ 10^6 [mm ⁴]	$I_{yy,old}$ 10^6 [mm ⁴]	$I_{yy,new}$ 10^6 [mm ⁴]	$I_{xx,ratio}$ [-]	$I_{yy,ratio}$ [-]	$J_{co,ratio}^{3)}$ [-]	A_{ratio} [-]	Buckling ⁴⁾ [-]
a	30134	40011	3862	5687	0.753	0.679	0.744	1.027	Class 2
b, c, d	10762	19407	999	1411	0.555	0.708	0.565	0.519	Class 3
b*	4948	7081	768	983	0.699	0.781	0.709	0.577	Class 3
e	357	526	183	264	0.679	0.693	0.684	0.679	Class 2
f	43559	60618	19258	28359	0.719	0.679	0.706	0.679	Class 2
g	6468	9648	1125	1605	0.670	0.701	0.675	0.514	Class 3
h	4948	9050	768	1288	0.547	0.596	0.553	0.469	Class 2
h*	636	1020	421	635	0.624	0.664	0.639	0.525	Class 2
i	11810	24215	1248	2006	0.488	0.622	0.498	0.579	Class 3
i*	486	822	449	727	0.591	0.618	0.603	0.554	Class 2
j	11811	24215	1248	2006	0.488	0.622	0.498	0.579	Class 3
j*	137	213	335	560	0.642	0.598	0.610	0.539	Class 3
k	298	486	99	226	0.613	0.420	0.550	0.781	Class 4
l	5357	12652	5357	12652	0.423	0.423	0.423	0.595	Class 3
m, n	4088	7040	4088	7040	0.581	0.581	0.581	0.575	Class 1
m ₂ , n ₂ ⁵⁾	4090	5806	4090	1231	0.704	3.323	1.163	0.500	Class 4
o	1183	2070	1183	2070	0.572	0.572	0.572	0.591	Class 1
q	$1155 \cdot 10^3$	$1907 \cdot 10^3$	$1155 \cdot 10^3$	$19057 \cdot 10^3$	0.606	0.606	0.606	0.602	Class 1
q ₁	$108 \cdot 10^3$	$180 \cdot 10^3$	1.7	8.0	0.600	0.216	0.600	0.600	Class 4
q ₂	575	1109	575	1109	0.519	0.519	0.519	0.480	Class 1
q ₃	6.6	39	$276 \cdot 10^3$	$497 \cdot 10^3$	0.171	0.556	0.556	0.556	Class 4
r, u, v, z ²⁾	-	-	-	-	-	-	-	-	-
s, t	485	1042	485	710	0.466	0.684	0.554	0.538	Class 3
w	418	597	186	280	0.699	0.666	0.689	0.651	Class 2
x	46	64	46	64	0.719	0.719	0.719	0.685	Class 1
y	73	103	74	103	0.711	0.711	0.711	0.504	Class 1

1) Slim end of beam is marked with *.

2) The beams are made of steel, and hence no change in MOI.

3) Torsional MOI ($J_{co} = I_{xx} + I_{yy}$).

4) See section 3.11.

5) Not in accordance with NORSOK U-002, MOI in y-direction is not within criteria.

Equation 4.2 calculating the necessary moment of inertia to maintain the same bending stress in relation to yield strength with the same applied load is based on bending stress being the only significant form of stress, as the resulting cross sectional area withstanding the moment is large enough to carry the resulting shear and axial stresses. A maximum moment of inertia ratio of 0.718 between original steel and new aluminium design is necessary when considering alloy 6082-T6, and 0.606 for alloy 5083-H116. This conservative beam design approach has been selected since the exact magnitude and location of the loads acting on the beams are uncertain.

$$\sigma = \left(\frac{M \cdot y}{I} \right) \quad [Pa] \quad (4.1)$$

Where σ is the bending stress [MPa], M is the moment about the neutral axis [Nmm], y is the perpendicular distance to the neutral axis and I is the second moment of inertia about the neutral axis [mm⁴].

By dividing equation 4.1 for aluminium alloy 6082 with 255 MPa in yield strength by the same equation for steel S355 with 355 MPa in yield strength the following is obtained:

$$\frac{\sigma_{Al}}{\sigma_{St}} = \frac{\left(\frac{M*y}{I_{Al}}\right)}{\left(\frac{M*y}{I_{St}}\right)} \Rightarrow \frac{255}{355} = \left(\frac{I_{St}}{I_{Al}}\right) = 0.718 \quad (4.2)$$

The height has been changed for some of the beams ('a', 'b', 'c', 'd', 'f', 'i' and 'j') erecting the base frame in order to increase the second moment of inertia, meaning the "y" in eq. 4.2 is different for aluminium and original design. To exemplify, the height change from 1.4 meters to 1.5 meters results in a "y-factor" of $0.75/0.7 = 1.07$. This will have an amplification on the identified ratio of 0.718 by $0.718 \cdot 1.07 = 0.769$ for the mentioned beams.

Beam 'a', 'f', 'k', 's' and 't' originally designed as squared pipes have been changed to an I-beam design. I-beams have greater potential in reducing weight since the I-beams possess a more effective design in terms of moment resistance. Closed compartments followed by squared pipes are in addition avoided.

For large I-beams that are impossible to extrude as one profile ('a', 'b', 'c', 'd', 'f', 'g', 'h', 'i' and 'j'), the top and bottom flange shall be extruded in alloy 6082-T6 with a built-in guide way for the web, and a 5083-H116 alloy web plate. The web and flanges shall be joined with two fillet welds each oriented at the guide way by MIG welding, as illustrated in figure 4.2. The guide way is designed to avoid a HAZ in the extruded part of the web, and to orient the weld at an area with lower bending stress. The length of the web extrusion is based on a linear distribution of stress from the neutral axis towards the top or bottom of the beam where bending stress is peaking. The length is estimated such that the bending stress in the HAZ will not exceed the HAZ strength of alloy 5083-H116 (150 MPa). See section 3.6.1 for relevant theory of the HAZ effect. For beams erected with an extruded flange, it is appropriate to use 0.718 as the design criteria between original steel beam and new aluminium beam since the moment is known to be highest at the location farthest away from the neutral axis.

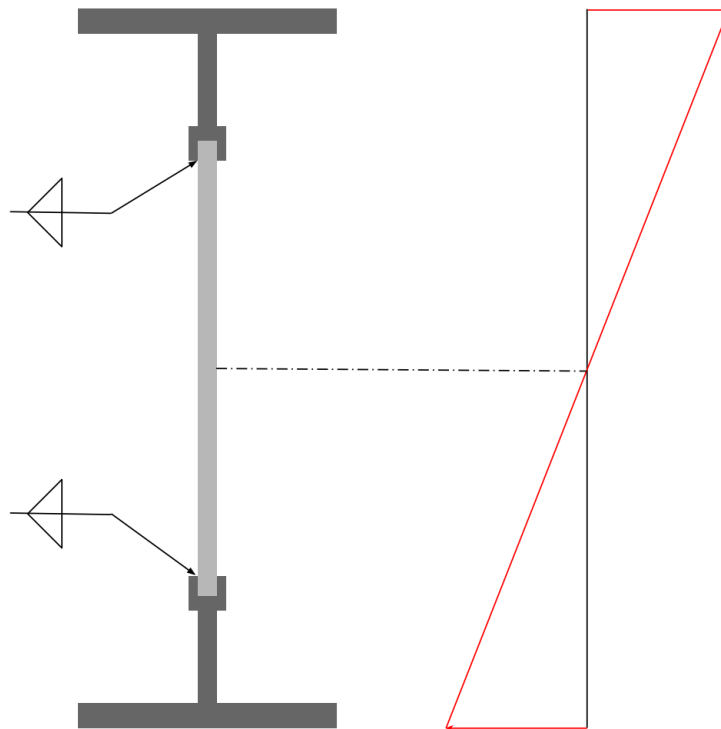


Figure 4.2: I-beam with a built-in guideway for the 5083-H116 plate in the 6082-T6 flange profile. Along with illustration of linear distribution of bending stress and orientation of fillet welds.

The beams ('l', 'm', 'n' and 'o') erecting the overtrawlable grid, is intended to be manufactured by bending a 5083-H116 plate into a circular tube closed by the use of FSW or MIG welding. The longitudinal welds shall be oriented as close to 90° to the direction of overtrawlable loads as possible in order to locate the HAZ away from maximum bending stress. The weld in for instance beam 'n' should therefore be located down and inwards as shown in figure 4.3. The HAZ magnitude is only 2.8% of total cross sectional area, loss of shear and axial resistance is thereby negligible. The 0.606 MOI criteria is used for tubes made of alloy 5083-H116.

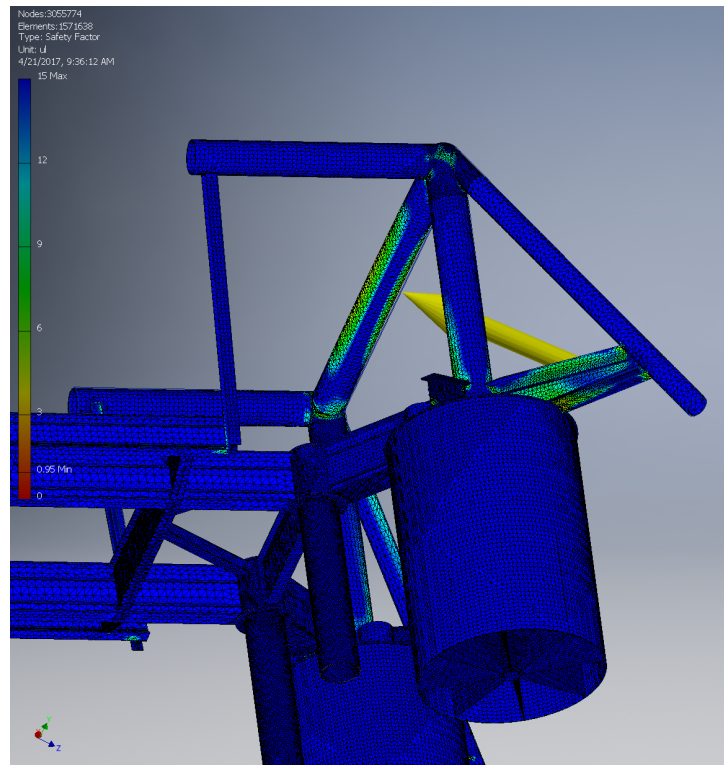


Figure 4.3: Stress distribution in beam 'n' when loaded with 1000 kN (yellow vector).

Two manufacturing methods for the cylindrical suction anchor (beam 'q') is evaluated. These two methods are based on plate size restrictions, 3.05 m width and 36 m length [79], that must be fulfilled in shaping a 15.675 m cylindrical perimeter. One possibility is to bend two plates into a tube, and bond them with one vertical and one horizontal friction stir weld. Another possibility is to friction stir weld six plates and subsequently bend them into a cylinder. The latter is regarded as the optimal alternative in consideration of preserving full strength from top to bottom of the anchor when a horizontal weld is avoided. However, the first method mentioned offers less welding but not more than the equivalence of three vertical welds as one horizontal weld approximately equals the length of two verticals (14 m). The top plate is too big to be made from one plate due to the large diameter (4.992 m) exceeding the plate size restriction, it should therefore be shaped by bonding two plates with the use of FSW. One plate is intended to be larger than the other, where the largest has a length and width of $4.992 \cdot 3.05$ m due to larger stresses towards the center. See section 4.2.2 for welding methods applicable to each joint on the structure. An illustration of the suction anchor with welds can be seen in appendix F.3.

Beam 'q₁' and 'q₃' are denoted with buckling class 4 as seen in table 4.1, the buckling class for the beams is however not considered to be problematic. Beam 'q₃' is supported by the crossing beams 'i' and 'j', which are assumed to be enough reinforcement against local buckling. In addition, the only design change for these beams from the original steel design is an increase in thickness, meaning that the buckling resistance in new design should not be any lower than the original.

The insufficient buckling class of beam 'k' seen in table 4.1 is assessed by calculating the MOI's with a lowered effective thickness, 15 mm instead of 20 mm. The effective thickness is found from eq. 3.17 in section 3.11. The obtained effective MOI is thereby lower than the actual, which results in a reduction of maximum design moment. The relation between design moment and MOI can be seen from eq. 4.1.

The moment of inertia ratio for beam 'b' (small end of beam) in y-direction (along the web) is 0.781, meaning it is violating the 0.718 criteria. That is because beam 'c' is designed continuously through beam 'b', which hinders an individual iterative procedure on the slanting beam 'b' as the dimensions are set for beam 'c'. However, the extruded flange oriented at the bottom of beam 'b' as illustrated in figure 4.4 is designed individually meaning it is subject to change to accommodate the moment criteria. Optimization of the bottom 'b' flange is neglected as it is believed to not significantly influence the outcome of the complete design.

The slanting beams 'h', 'i' and 'j' are heightened in the small end to create enough space for the built-in guide way extrusions and web plate, see figure 4.5 for illustration of beam 'h'. Otherwise it would cause unnecessary difficulties erecting the beams. For beam 'h', only the bottom slant is changed due to sealine interaction on top of beam.

Dimensions of I-beam 'k' and 'e' are within the extrusion limits listed in table 3.5, and are therefore intended to be fully erected by extrusion. I-beam 's' and 't' are designed to be made of two extruded flanges with large enough web length to accommodate for the beam height, where they can be subsequently bonded by FSW or MIG welding at the neutral axes of bending moment.

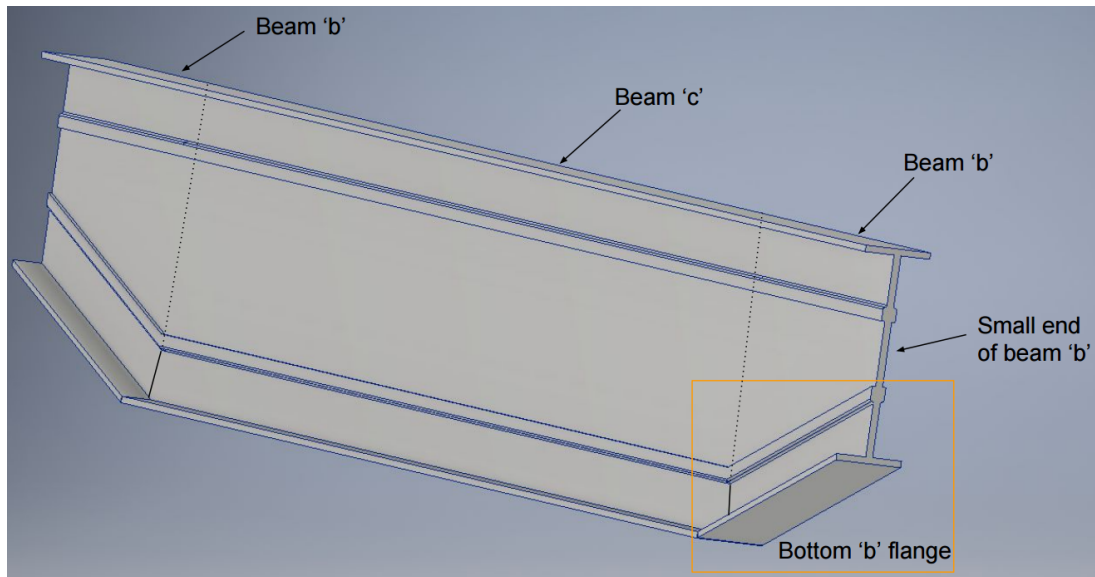


Figure 4.4: Merged 'b' and 'c' beam.

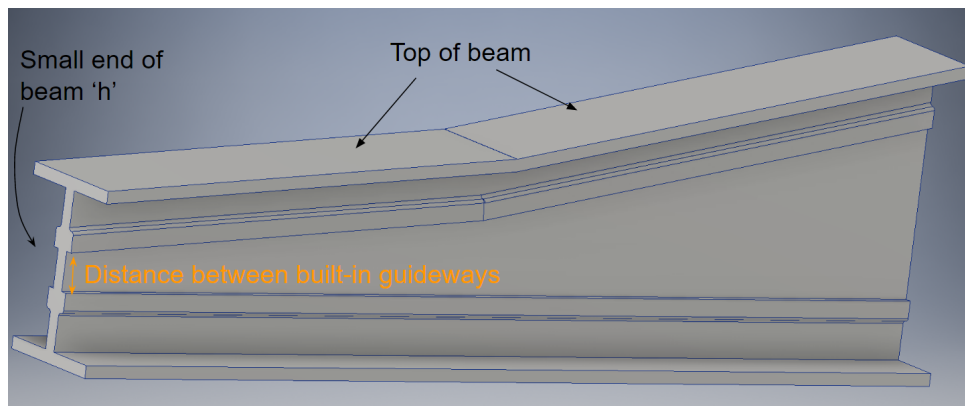


Figure 4.5: Heightened Beam 'h'.

Beam 'f' is intended to be erected by two I-beams in order to match the MOI from the original square steel beam, see figure 4.6. The I-beams are intended to be connected by the use of FSW on the top and bottom flange facing each other. The moment resistance in the y-direction seen in figure 4.6 should not be decreased significantly as the weld is oriented at the center of rotation (low stress area). Generally, the HAZ caused by welding the I-beams in figure 4.6 does not affect the beam's strength significantly, as the HAZ area is small compared to the affected area.

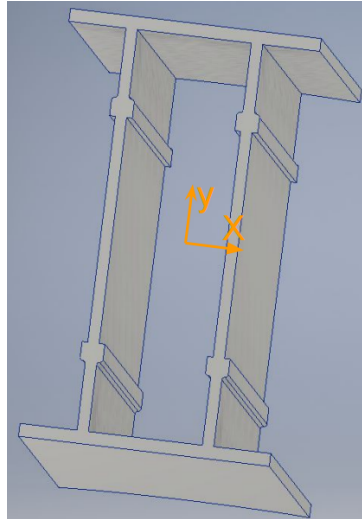


Figure 4.6: Beam 'f' with respective x and y MOI directions.

Beam 'm' and 'n' cannot be made by extrusion, as their dimensions are way too large for any of the identified extrusion presses given in table 3.5. Two types of beam design have been reviewed, and both can be seen in figure 4.7 and table 4.1. In table 4.1 the beams 'm₂' and 'n₂' has an unfavourable I_{yy} ratio, due to the change in geometry as seen in figure 4.7. This design is not in accordance with NORSOK U-002 [14], as the requirement for external edges/members shall have a minimum radius of 250 mm, which is not fulfilled. However, placing the beam in such a way that the circular edge is facing the external loads (trawl loads) could be sufficient, dependent upon the interpretation of the NORSOK U-002 standard [14]. Emphasize has been on producing a realistic design which can be introduced into the industry, thus the NORSOK approved design is therefore selected as a basis for all further design analyses.

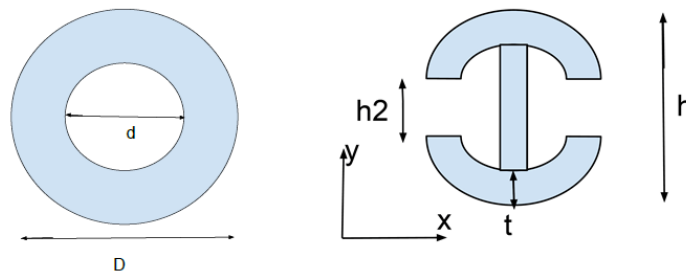


Figure 4.7: Two alternatives for beam 'm' and 'n', where the left one is in accordance with NORSOK U-002 for $D \geq 250\text{mm}$.

4.2 Design of joints

Some of the structure's joints will have to be redesigned to accommodate the use of aluminium as construction material. This is especially so with respect to welds as aluminium MIG welds result in a reduced mechanical strength of approximately 30-50% in the HAZ zone, dependent upon the aluminium alloy as described in section 3.6.1. Bolting has also been introduced as an alternative joining method to welding, as bolted sections enables joining without any significant strength reduction in the base metal of bonding section. A more detailed review of bolting with its advantages and drawbacks has been conducted in section 3.8.

The X-mas tree supports (beam 'x', 'y' and 'z') are regarded as negligible in terms of structural integrity of the base frame connected to suction anchors, since the X-mas support is resting on the steel tube (beam 'r'), which is designed to be jointed with the intersecting base frame beams ('a', 'c', 'd' and 'e'). The X-mas tree support is however important in itself supporting the X-mas, but an aluminium joint design of the support is left out of this study, as it is known to be achievable.

All joints on the concerned ITS is designated with roman numerals in figure 4.8. The joints are further designated with a subscript specifying the beams to be bonded (e.g. V_{a-a}), which should ease the understanding of orientating for each joint when described below.

4.2.1 Bolted joints

An analysis of all the bolted connections have been conducted in accordance with Eurocode 9 [2]. Forces acting on each joint are presented in table 4.2, and dimension of bolt plating, arrangement and number of stainless steel bolts are tabulated in table 4.3. The results given in table 4.3 are only valid for stainless steel bolts as this is the most conservative approach comparing the use of stainless and carbon steel bolts. Only difference in calculation input between the two bolt materials in this analysis is the TS entered in eq. 3.3. The difference in plate dimensions and total number of bolts, 1468 vs 1336, used in weight and cost assessments are therefore regarded as negligible. The numerical calculation sheets created in the

analysis can be found in appendix C. The bolts are designed as slip resistant, eq. 3.3 and 3.4 are thereby used to find the number of bolts required for different bolt types ranging from M20 - M39. M36 are found to be suitable for all joints, considering the number of bolts and joint size. All bolt joints designed as spliced beam connections are assumed to support all shear forces in the web while the flange supports all tension and compressive forces, which is mainly obtained from moment. An ultimate limit safety factor of 1.25 is used in the bolt calculations and the slip factor is set to 0.4 in accordance with standards [2].

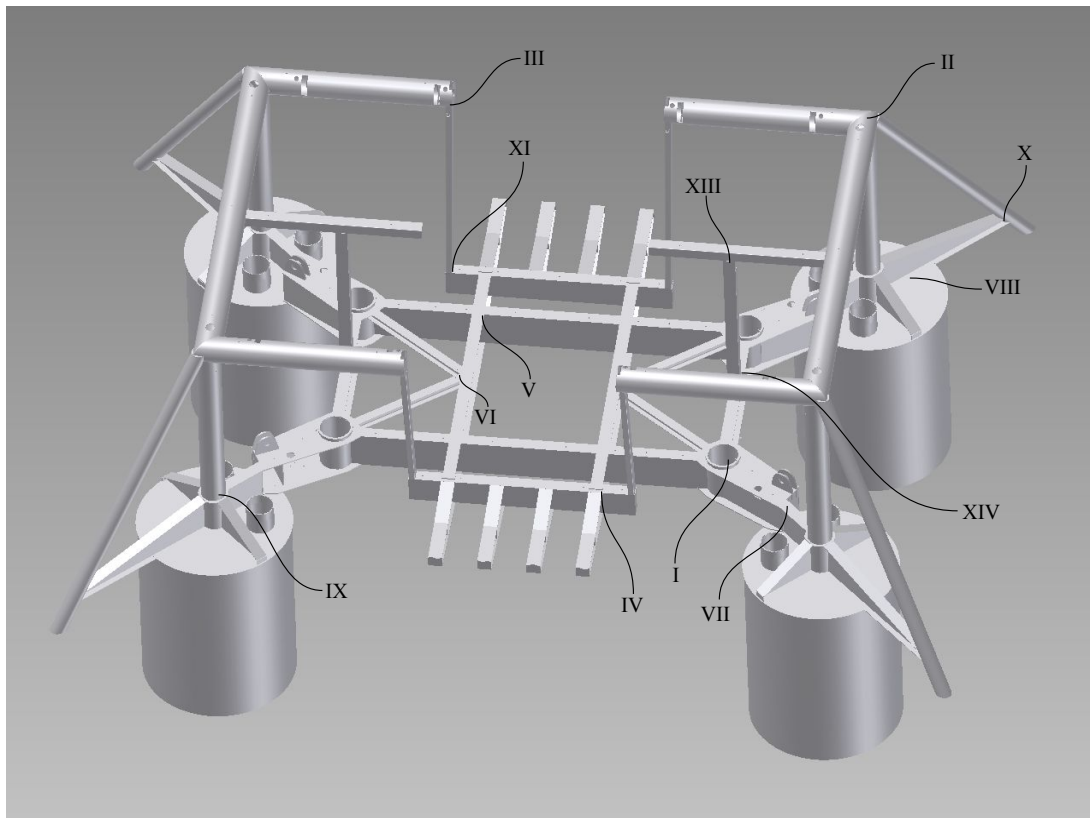


Figure 4.8: Joints designated with roman numerals.

There are uncertainties related to which structural joints the loads are applied on for the base frame connected to suction anchors, the maximum moment in beam 'a' is therefore calculated by using the moment of inertia and design strength of the original steel beam, where the design strength is obtained by dividing the yield strength with a safety factor for steel, 1.15, stated in NORSOK N-001 [80]. By re-arranging eq. 4.1, 1300 kNm is obtained as the maximum moment. The resulting moment distribution along the beam is further assumed linear between the mid point (maximum load) and the suction anchor support, where the moment is assumed to be zero as outlined in figure 4.9. Table 4.2 presents the calculated val-

ues of the respective moments and forces in joint V_{a-a} , I_a and I_f . Since the beam is designed to accommodate maximum moment it will be overly conservative to find shear forces from geometry of the beam. Maximum shear force in beam 'a' is found by dividing the maximum moment by half the beam length, resulting in 1625 kN. Considering the amount of weight supported by the base frame, the magnitude of the calculated shear force is coherent. Beam 'f' is however calculated with 2000 kN shear force as this beam is assumed to be subjected to the highest shear forces.

Table 4.2: Forces acting on the bolted connections.

Connection	Moment [kNm]	Flange force [kN]	Web force [kN]	Comment
I_a	4998	3452 ⁴⁾	1625	Moment is 3/8 of maximum moment (1300 kNm).
I_d	1662	1131 ⁴⁾	100	Moment is 1/8 of maximum moment ¹⁾ .
I_f	3326	2301 ⁴⁾	1625	Moment is 2/8 of maximum moment.
I_e	0	200	0	Assume 200 kN tensile force ²⁾ .
V_{a-a}	9977	6904 ⁴⁾	1625	Moment is 6/8 of maximum moment. 5180 mm moment arm ³⁾ .
V_{b-c}	518	652 ⁴⁾	100	Only one internal bottom flange.
IV_{h-g}	340	650 ⁴⁾	100	3400 mm moment arm ³⁾ .
IV_{b-g}	380	692 ⁴⁾	100	3800 mm moment arm ³⁾ .
VI_{c-e}	0	200	0	Assume 200 kN tensile force ²⁾ .
XI_{g-k}	460	66	-	Calculated from geometry of beam 'k'.
XIV_{d-s}	750	111	-	Calculated from geometry of beam 's'.
$XIII_{s-t}$	0	111	-	Calculated from geometry of beam 's'. No moment acting on beam 't'.

1) Loads on beam 'd' are uncertain. Conservative to use 1/8 of maximum moment.

2) Joint force is not known. An assumption is made.

3) Arises from pull in forces (section 4.3.2).

4) Flange force = Moment/Height of beam.

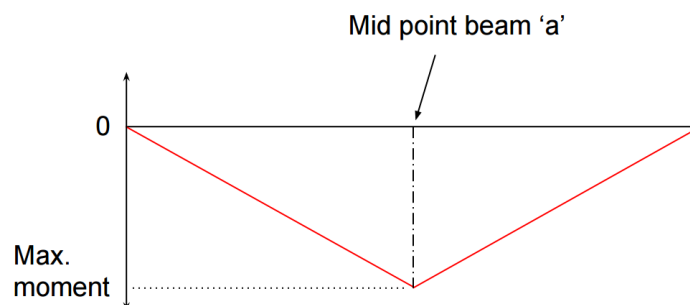


Figure 4.9: Assumed bending moment distribution in beam 'a'.

Connection 'V_{b-c}' and 'IV' are dimensioned for maximum axial and vertical sealine loads. Maximum tensile sealine load (between 0°-15° horizontally) on one beam is 300 kN, and maximum loading on all four sealine supports are 600 kN, giving rise to various scenarios of sealine loads described in section 4.3.2. However, 2 · 300 kN is considered in calculations. Some extra loading is added vertically in bolt calculations due to an unknown weight of sealine. Vertical load used in bolt calculations on one sealine support is then 100 kN. These loads creates moment, tensile and shear forces in joint 'IV' and 'V', the resulting shearing loads are negligible and the connections are therefore designed with only spliced flanges that have capacity to support all forces.

The 'XIV_{d-s}' connection is designed to accommodate maximum moment and resulting shear force calculated from geometry of beam 's' as seen in table 4.2. The connection between beam 'd' and 's' is retrofitted with a 90° angled plate on the top and bottom flange of beam 'd'. Connection 'XI_{g-k}' is jointed in the same way with an angled plate that is fitted to the angle between beam 'g' and 'k'. Plates on each flange is used in connection 'XIII_{s-t}' to carry the calculated shear force in table 4.2. See figure 4.10 for illustration of joint 'XIII_{s-t}' and 'XIV_{d-s}'.

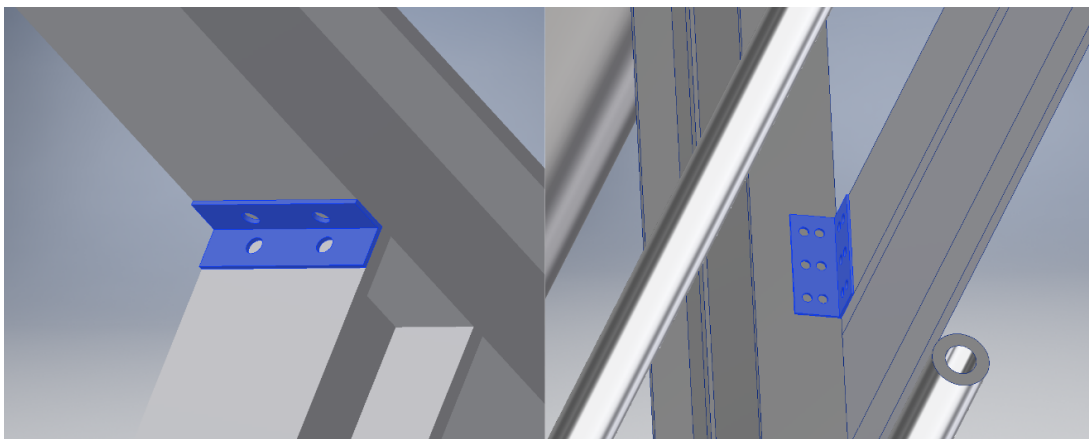


Figure 4.10: Angled bolt plating on joint 'XIII_{s-t}' (left) and 'XIV_{d-s}' (right).

Table 4.3: Dimensions of bolt plating and number of bolts required.

Connection	Flange plate dimension external/internal ($l-w_e/w_i-t_e/t_i$) [mm]	Web plate dimension (h-w-t) [mm]	Length of cross section flange/web [mm]	Number of flange/web bolts for all connections ¹⁾	Sum M36 stainless steel bolts
I_a	610-500-25	600-200-10	80/70	208 (4x7)/ 48 (2x6)	256
I_d	438-500-10	200-150-10	40/55	72 (2x5)/ 8 (1x2)	80
I_e	188-300/ 0-10/0	0	42/0	16 (2x1)/0	16
I_f	400-880-25	600-200-20	70/60	208 (8x4)/ 60 (2x4)	268
V_{a-a}	1220-500/ 420-40/40	600-150-10	135/80	416 (4x9)/ 48 (1x6)	464
V_{b-c}	532-0/ 550-0/15	0	45/0	48 (2x3)/0	48
IV_{h-g}	360-330/ 265-10/10	0	50/0	128 (2x2)/0	128
IV_{b-g}	360-330/ 265-10/10	0	50/0	64 (2x2)/0	64
VI_{c-e}	188-300/ 0-10/0	0	42/0	16 (2x1)/0	16
XI_{g-k}	180-300-10	-	50/-	64 (2x2)/-	64
XIV_{d-s}	180-420-10	-	40/-	48 (3x2)/-	48
$XIII_{s-t}$	100-420-10	-	45/-	16 (2x1)/-	16
Sum					1468

1) Bolt arrangements are given in parentheses.

Flange and web plating used in the bolted joints must be designed to withstand the transmitting forces, meaning that stress should not exceed YS reduced by the appropriate safety factor for aluminium, 1.2, [80], and dimensioned according to bolt arrangement and required spacing as seen in figure 3.16. Plate dimensions and number of bolts together with the cross-sectional bolt length for each joint are tabulated in table 4.3.

Joint 'I' is designed as a steel tube with welded steel plates that will function as flange and web for each connected beam ('a', 'd', 'e' and 'f'). The extensive use of steel in joint 'I' is due to difficulties bonding four interconnecting beams, the joint is critical, and it is assumed to be beneficial to have a steel surface interacting with drilling equipment.

Beam 'f' is made up of two I-beams as seen in figure 4.6, the inside of the webs and flanges will thereby possess a limited access for bolt assemblies. The web and flange bolts must therefore be fastened to the steel plates bonding joint 'I' and beam 'f' before welding the plates onto the steel tube. Inspection should be possible either through the opposite side of the beam or through retrofitted holes.

4.2.2 Welded joints

Some of the joints shall be welded as it is believed to be more beneficial than bolting in terms of fabrication cost, availability and uniform beam surface, which is required for the overtrawlable tubes on top of the structure. All welds are to be placed away from stress concentrations. Welded areas are checked for stresses with the use of FEM analysis in Autodesk Inventor. Stress in welded areas can be assessed in Autodesk Inventor using contour plot for the stress results.

MIG welding is a well established and commercial cost efficient method that is very well documented in standards as seen in section 3.6.1, and it is the recommended welding method by Marine Aluminium [34]. It should therefore be used on all fillet welds required (e.g. joint 'I', 'III', 'VII', 'VIII', 'IX' and 'X'). For butt welds (e.g. beam 'q'), FSW is believed to be the preferable method as long it is possible to weld without a hand-held apparatus as seen in section 3.6.2. Quality of welds is critical and shall therefore be assured appropriately as outlined in standards [2].

Beam 'i' is designed continuously through joint 'VII' where the intersecting 'j' beams are welded onto beam 'i'. Beam 'i' and 'j' are additionally intended to be welded to the suction anchor. Bottom end of beam 'l' is further designed with four tracks welded on the intersecting beams, with additional welds on the suction anchor at the bottom. The larger beam 'f' is functioning as an extension for beam 'i' towards joint 'I'. Beam 'f' should be welded onto beam 'i' and top of suction anchor, see figure 4.11 for illustration.

Joint 'II' is designed with an insert embedded in all four tubes intersecting the joint. An insert is used in order to accommodate strength losses from welding, which enables stress

resistance obtained from moment. All tubes shall be circumferential FSW or MIG welded individually as outlined in figure 4.12.

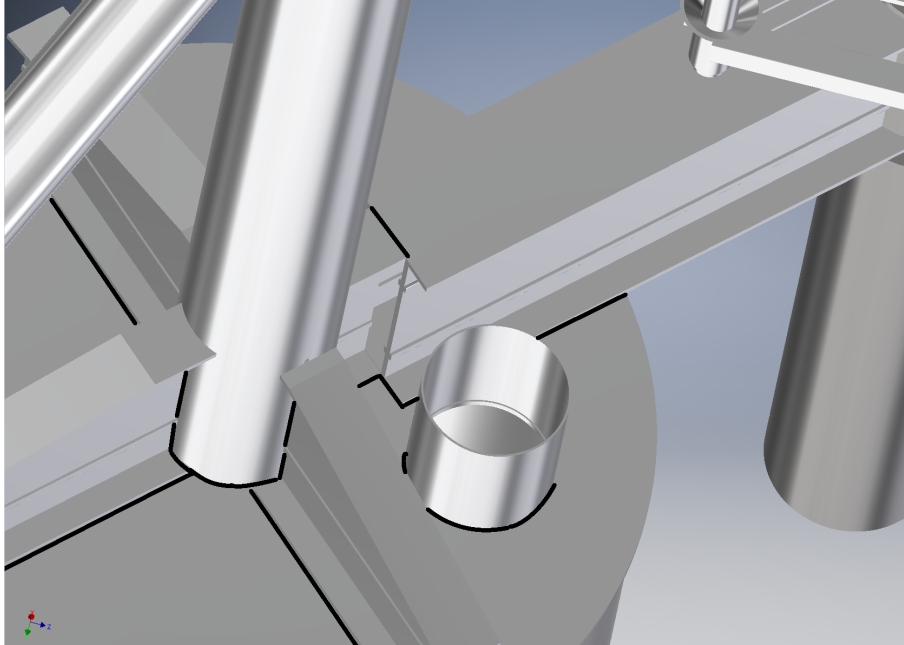


Figure 4.11: Weldments on suction anchor and beam 'i', dark lines are weldments.

Manufacturing of the insert seen in figure 4.12 has proven to be challenging, due to the relative large size that must be machined as one unit from one of the NORSOK approved alloys. See section 6.4.2 for further discussion about this challenge.

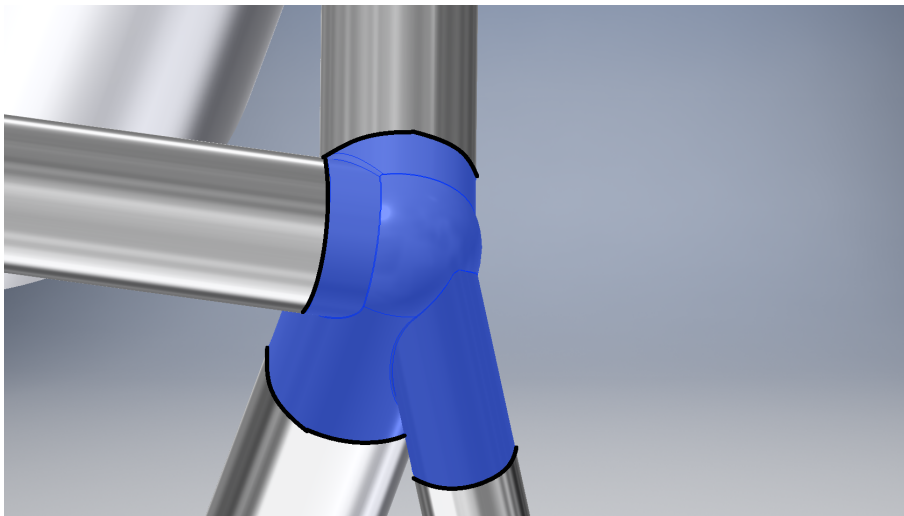


Figure 4.12: Weldments on insert, dark lines are weldments and the insert is colored blue for better vision.

All connections on the suction anchor are designed to be welded, because it is not an optimal choice to bolt through the top plate of the suction anchor. As the availability to assemble the bolts is restricted and it would be very difficult to inspect bolts inside the anchor after installation. The relatively large bonding area on the suction anchor is advantageous in terms of joining since the joint forces are spread over a large area, and stresses generally lowers with larger area. The internal stiffeners ('q1', 'q2') located at the bottom of the suction anchor is also designed intently to be fastened by welding.

Joint 'I' shall be made up of a steel tube with welded steel plates as mentioned in section 4.2.1. The effects that may mitigate from steel welding has not been analyzed, as it is out of scope for this study. However, the steel welds should not introduce any problems of concern.

4.3 Loads

The ITS is subjected to loads during installation and operation. Lift off, splash zone and tie-in loads are considered installation loads. Dropped objects, drilling and trawling loads are treated as operational loads. Loading from weight of X-mas trees and manifold supported by the ITS are defined as static loads. A broad explanation of relevant load scenarios are given in this chapter, while numerical summation of load values are found in appendix A. Loads that may occur during barge transport is not included in this study.

4.3.1 Lift off and splash zone loads

The lifting operation interval from the ITS is lifted from deck until it is completely immersed in seawater is the most critical process during installation. The ITS is subjected to added mass from seawater, slamming forces from seawater/waves and fluctuating loads when airborne. There are four dynamic forces and one static force to be considered in the splash zone [81], all of them are further described in appendix A.1. Note that the loads occurring before and during splash zone penetration mainly concern the crane capacity, and not the integrity of the ITS. However, care must be taken into account during hydration of air filled compartments (e.g. suction anchor) due to the airflow leaving the compartments and the

ITS could be buoyant enough to cause slack of lifting wire. Dynamic loads during the critical process can be accounted for by a dynamic amplification factor (DAF). A DAF value of 1.25 should be used offshore when lifting 100-300Te heavy structures [82]. The weight of the lifting rig (connecting crane wire and lifting points) on the ITS itself should in addition be added to the structural weight, the lifting rig weight is tabulated in table 4.10. For deeper waters (more than 500 m) the weight of the wire may be significant, especially for an ITS made of steel. However, an ITS made of aluminium as in this case are more readily immersed to deeper waters since the buoyancy factor is significantly lower for aluminium (0.63) than for steel (0.87). The weight of the wire is neglected in this study, in order to ease calculations.

The crane capacity must be greater than structural weight (incl. lifting rig) multiplied by the DAF. The lifting force on each of the four lifting point can not be found by simply dividing the main lifting force by four. The angle on the wire sling connecting the lifting rig and lifting points on the structure must be accounted for. A global skew load factor (SKL) of 1.25 as defined in DNV-OS-H205 [82] may be added to the force (included DAF) on each lifting point in order to calculate the resulting wire force. The SKL is only valid for angles less than 60° horizontally [82]. Maximum design force on the main lifting wire is then 2453 kN, and the resulting force on each lifting point is 766 kN. See e-mail correspondence with Asbjørn Wathne (Subsea 7) in appendix B.5 for some details on ITS lifting and installation operation.

4.3.2 Pull in of umbilicals and pipelines

Pull in loads are related to hydraulic forces used to pull the pipeline or umbilical into desired position. In the case of an ITS used in this study, the maximum pull in load for all the sealine supports (four support arms) combined is 600 kN. Giving rise to multiple load cases, for example $4 \cdot 150$ kN or $2 \cdot 300$ kN [15]. Note that the maximum pull in load on one sealine support is 300 kN. The load direction may be in an inclination range of 0 to 15 degrees horizontally. Maximum vertical design load is therefore 77 kN. Scenarios of sealine loads are defined in appendix A.4. The weight of the sealines and umbilicals are uncertain, however it should not lead to a load of any significant size, as it is known to be much lower than pull in load. Pull in shall only be performed after the template structure is lowered into final vertical position (final setting of suction anchor).

4.3.3 Drilling and trawling loads

The drilling loads are found from NORSOK U-002 [14] that describes loads for different activities. There are two cases outlined in the standard, one for water depths up to 750 m and one for depths up to 1500 m. Only the first case is considered in this study to limit the scope of work. The "750 m case" is defined in appendix A.5. The largest loading occurring during drilling is when the 30" conductor is lowered. The temporary design load from the conductor weight is 600 kN in the vertical direction.

All subsea installations in Norway have to withstand overtrawling from fishing gear. Load types to be considered are trawl net friction, trawl board overpull, trawl board impact and trawling snag. See appendix A.2 for tabulation of the different trawl load scenarios. The most severe load case is snagging of trawl ground rope, which is in the magnitude of 1000 kN. However, a subsea structure may not be designed against snag loads if the structure is documented to be snagfree/overtrawlable. The most severe load case for a snagfree/overtrawlable design is $2 \cdot 200$ kN in trawl net friction. The original steel ITS design is known to possess a snagfree design, see appendix B.4 for e-mail correspondence concerning overtrawlablity. Nevertheless, 1000 kN are used in section 4.4 as a conservative approach.

4.3.4 Dropped objects

Impact energy from dropped objects can be seen in appendix A.3. These objects are normally accidentally dropped from topside vessel, but can also come from fishing vessels. The protective hatches fitted on top of the ITS to protect the X-mas trees and manifold from dropped objects are neither modelled or designed, as it is left out of scope in this study. It should be noted that aluminium made protection structures have been successfully used on the NCS on the Lille-Frigg field [34].

4.3.5 Static loads

Static loads from support of X-mas trees and manifold used in calculations can be seen in appendix A.2. Note that the buoyancy factor for steel in seawater will lower the loading from components to be supported by the ITS by an estimated factor of 0.87, as the components

are installed after final installation of the ITS. Buoyancy factor is not included in analysis, as the factor of 0.87 is only an estimate for the manifold and X-mas trees and excluding the buoyancy factor is the most conservative approach.

4.4 Model

An assembly model containing all the redesigned components in section 4.1 has been built to run global stress analyses to validate the new design. The computer aided design (CAD) software Autodesk Inventor has been used to design, assemble and analyse the structure. A figure of the assembled and meshed structure can be seen in figure 4.13. Bolt assemblies are not included in the model, however stress concentrations of concern, especially for joints and welds will be identified on the stress contour plots. All analyses are based on the finite element method, hence the results are only valid within the materials elastic area, where the materials have a linear stress-strain relation.

All the components in the model have to be designated a material before the simulation can be performed. Aluminium alloy 6082-T6, aluminium alloy 5083-H116 and steel S355 are the three materials used in the model. The different components designated material can be found in section 4.1.

The mesh is generated with an average element size of 0.100, as it is the recommended value by the software developer [83]. A smaller value would result in a more accurate and time-consuming analysis. The number of nodes and elements may vary from each load case and evaluation as there may be a need to edit the mesh set-up for each simulation, due to changes in load or fixed objects/surfaces. Approximate numbers for both element size and nodes can be seen in table 4.4. The general approach for the simulations are the following:

Create contact points → Mesh set-up → 1. Simulation → Result interpretation → Fix mesh set-up → 2. Simulation → Result interpretation → ...

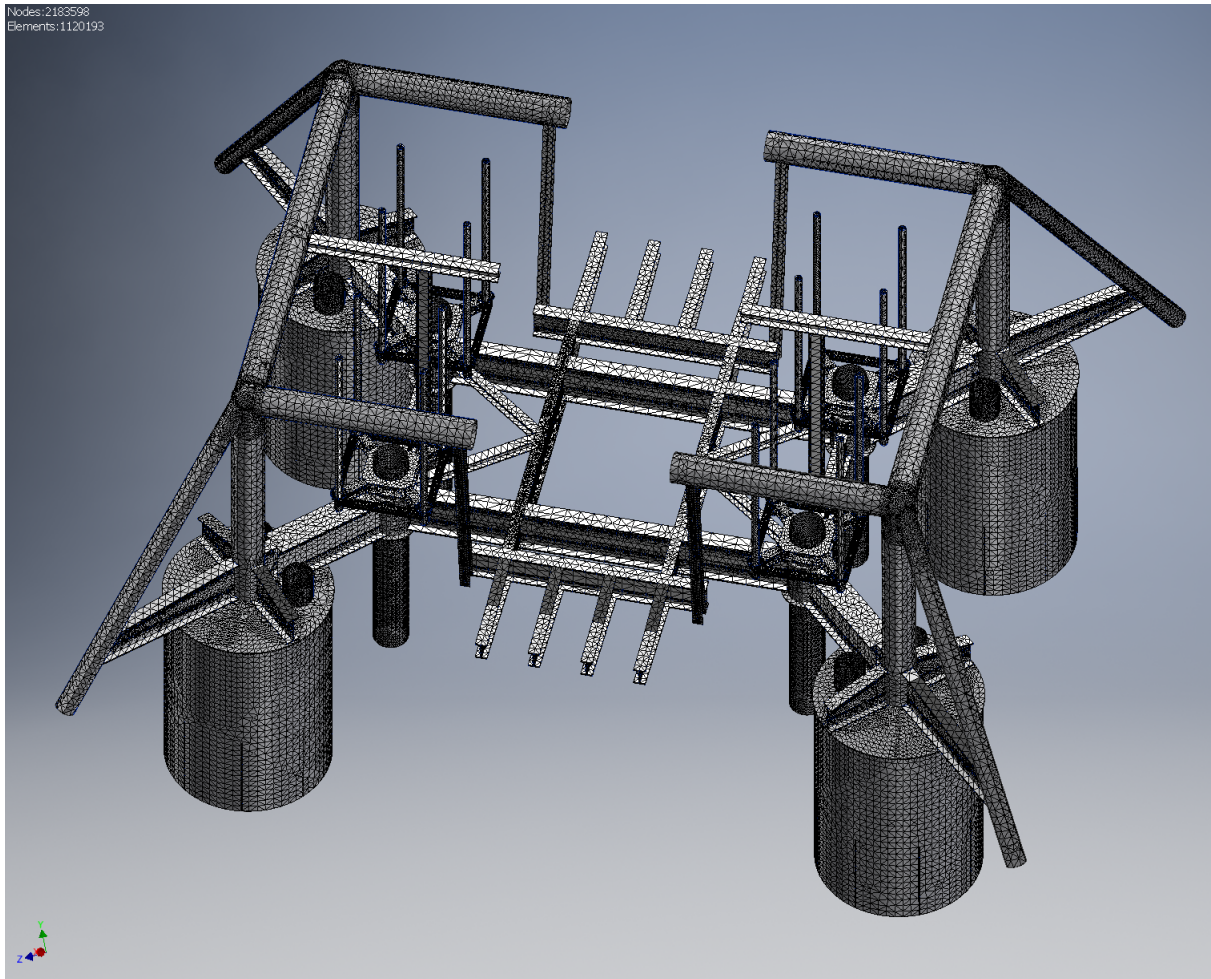


Figure 4.13: Assembly model of the redesigned integrated template structure.

Table 4.4: Input numbers for mesh and resulting number of elements and nodes.

Number of elements	Number of nodes	Average element size ¹⁾	Minimum element size ¹⁾
ca. 1 150 000	ca. 2 225 000	0.100	0.200

1) Input value for mesh-generation.

This is an iterative process where step five (Fix mesh set-up) and forward are repetitive processes. The reason for selecting this approach is to identify stress singularities that may be inaccurate near stress concentrations. The theoretical stress around such stress singularities are infinite, and should therefore be left out of the evaluation. Stress singularities are normally found surrounding sharp edges or sharp corners. A convergence plot can be made in the simulation to interpret the results and decide if there is a stress singularity present, or if the stress is caused by a stress concentration. A convergence plot is created in Autodesk Inventor by increasing mesh density on the structure locally around the peak stress several times to see how the stress changes with smaller element sizes. If the plot converges towards

a limit the stress is caused by a stress concentration, if the stress is divergent (stress does not converge towards a single value) there is a stress singularity present [84].

The applied boundary conditions for the structure is not entirely realistic as a solid clay model has been applied for the suction anchors. Meaning the skirt of the suction anchor is set to fixed position (no vertical or horizontal displacement). A more realistic approach would take into account the undrained shear strength of the soil and the displacement of the suction anchor skirts. The boundary condition for each stress analysis can be seen in appendix I.

4.4.1 Load cases

Multiple load set-ups are necessary to accommodate all the possible load scenarios. Table 4.5 lists all modelled set-ups, the loads are taken from section 4.3 containing load definitions and combined as described in NORSOK U-002 [14] for up to 750 meters seawater depth. All events described in the NORSOK standard are covered in the cases seen in table 4.5, except for the impact loads.

The impact/dropped object loads are excluded from simulations due to uncertainties regarding modelling of impact loads and to limit the workload. Some qualitative comments regarding dropped object loads can be provided based on material properties and the geometrical design of the structure. Dropped object loads are according to NORSOK U-002 [14] designed as PLS loads, hence plastic deformation of the structure is accepted, which is positive for aluminium as it possess better ductile properties compared to steel. The aluminium structure's increased moment of inertia described in section 4.1 have to compensate for aluminium's reduction in energy absorption compared to steel. Steel has better energy absorption properties due to higher yield strength and higher E-modulus, however if energy absorption is measured per weight unit, aluminium is favourable to steel as described in reference from the automobile industry [45]. Any further assessment regarding impact/dropped object loads are left out of scope in this study.

The applicable loads for trawling and tie-in are defined with a horizontal angle, as seen in

appendix A. These angles ranging from 0 to 20 degrees with the horizontal plane are not considered in FEM analyses, as it would increase the number of cases to simulate substantially, and each case is very time consuming to simulate.

Table 4.5: ITS load cases.

Case #	Load [kN]	Load type	Reference
A	$1000-H^{(6)}$	Trawl ground rope snag	Section 4.3
B ¹⁾	$1000-H^{(6)} + 800$	Trawl ground rope snag + Manifold weight	Section 4.3
C ²⁾	$1000-H^{(6)} + 800$ $+ 2 \cdot 300$	Trawl ground rope snag + Manifold weight + Tie-in	Section 4.3
D ⁴⁾	$4 \cdot 600 + (450 + 200-H^{(6)})$ $+ 800 + 2 \cdot 300$	X-mas trees + Drilling loads + Manifold weight + Tie-in	Section 4.3
E ⁴⁾	$1000-H^{(6)} + 600 + 800$ $+ 2 \cdot 300$	Trawl ground rope snag + Drilling loads + Manifold weight + Tie-in	Section 4.3
F ⁵⁾	$1000-H^{(6)} + 4 \cdot 600$ $+ 800 + 2 \cdot 300$	Trawl ground rope snag + 4* X-mas trees + Manifold weight + Tie-in	Section 4.3
G ⁷⁾	$4 \cdot 766.5$	Offshore installation load	Section 4.3

1) Weight of manifold is approximately 800 kN.

2) Tie-in loads are $2 \cdot 300$ kN divided on to tie-in ports.

3) 600 kN during installation of 30" conductor (temporary drilling load).

4) Unrealistic to happen as trawling cannot happen while drilling is ongoing, but provides information regarding integrity and conservatism in design.

5) 4 installed X-mas trees, each weighting approximately 600 kN.

6) -H means the load is to be applied in horizontal direction.

7) Lifting the ITS-structure offshore, with an angle of 60° the load is measured in dry weight as described in section 4.3.

4.4.2 Results

Results from all the simulations will be presented in this section with necessary comments concerning the results. All results can be seen in appendix I.

Case-A

Case-A represents a trawl ground snag resulting in a maximum von-Mises stress of 189.7 MPa. This is a relatively high stress for the aluminium alloy, resulting in a safety factor of 1.13. This is a lower value than the 1.2 limit stated in NORSOK N-001 for ultimate level state [80]. However stress figures in appendix I.1 show the peak stress to be very localized and a convergence plot has been made to decide whether the peak stress is a result of a stress singularity or not. As seen in figure 4.14, the stress convergence plot suggest a divergent stress, and the von-Mises peak stress at 189.7 MPa is therefore believed to be a stress singularity which should be disregarded as a valid result. The displacement results in appendix I.1 suggest the same, as the location of the highest stress is not exposed to high displacements, while the upper cross beam has the highest displacement without any critically high stress, approximately 50 MPa von-Mises stress at the location of highest displacement. The aluminium structure is therefore believed to be well dimensioned for case-A.

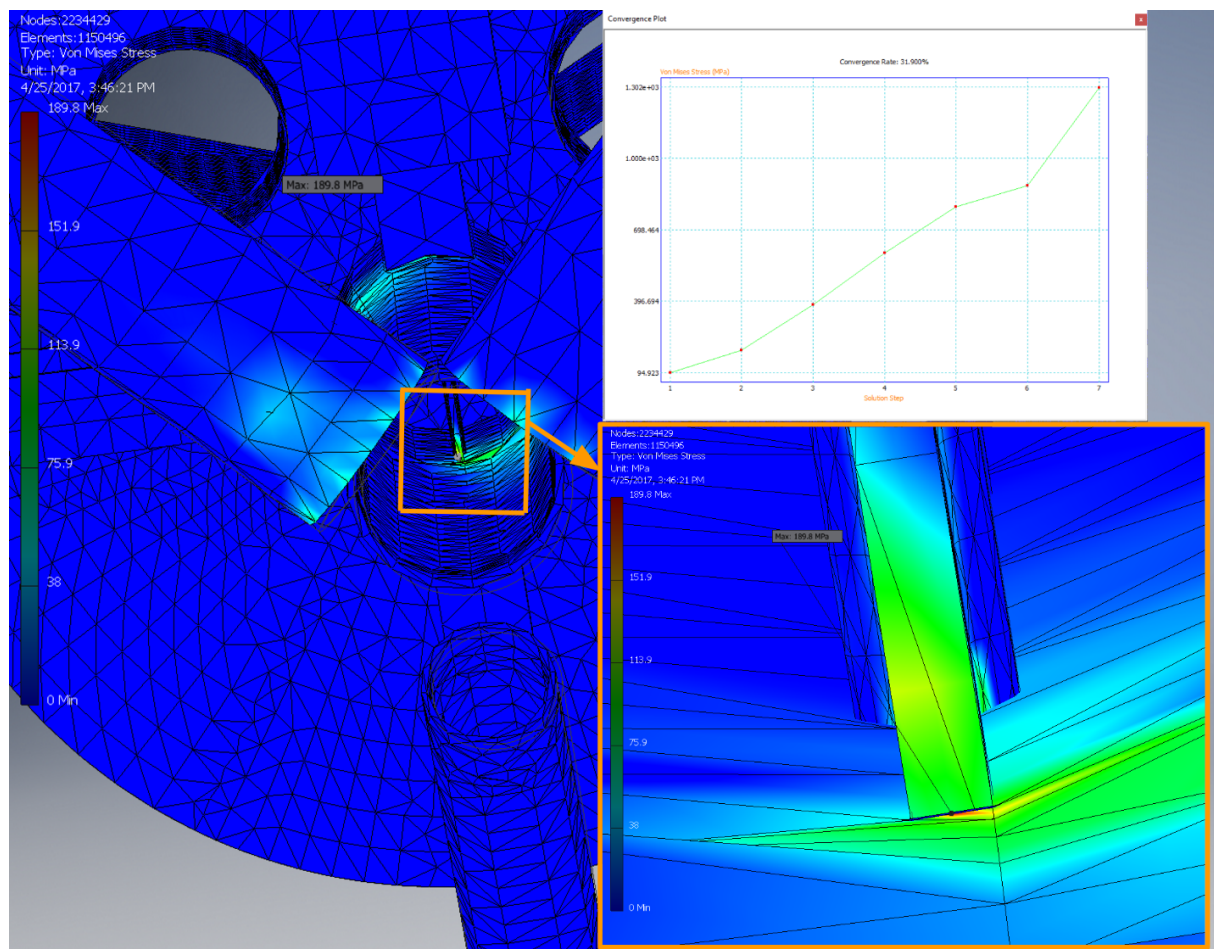


Figure 4.14: Stress singularity for case-A, with convergence plot.

Case-B

Case-B represent a combination of trawl ground snag (1000 kN) and the weight of the manifold in the center of the structure, 800 kN vertical load. The maximum resulting von-Mises stress on the structure is 194.7 MPa as seen in appendix I.2.1. As for case-A this stress is believed to be a stress singularity as the convergence plot in figure 4.15 suggests. The stress is limited to approximately 80 MPa, expect for the mentioned stress singularity of 194.7 MPa. The aluminium structure is therefore well dimensioned for case-B.

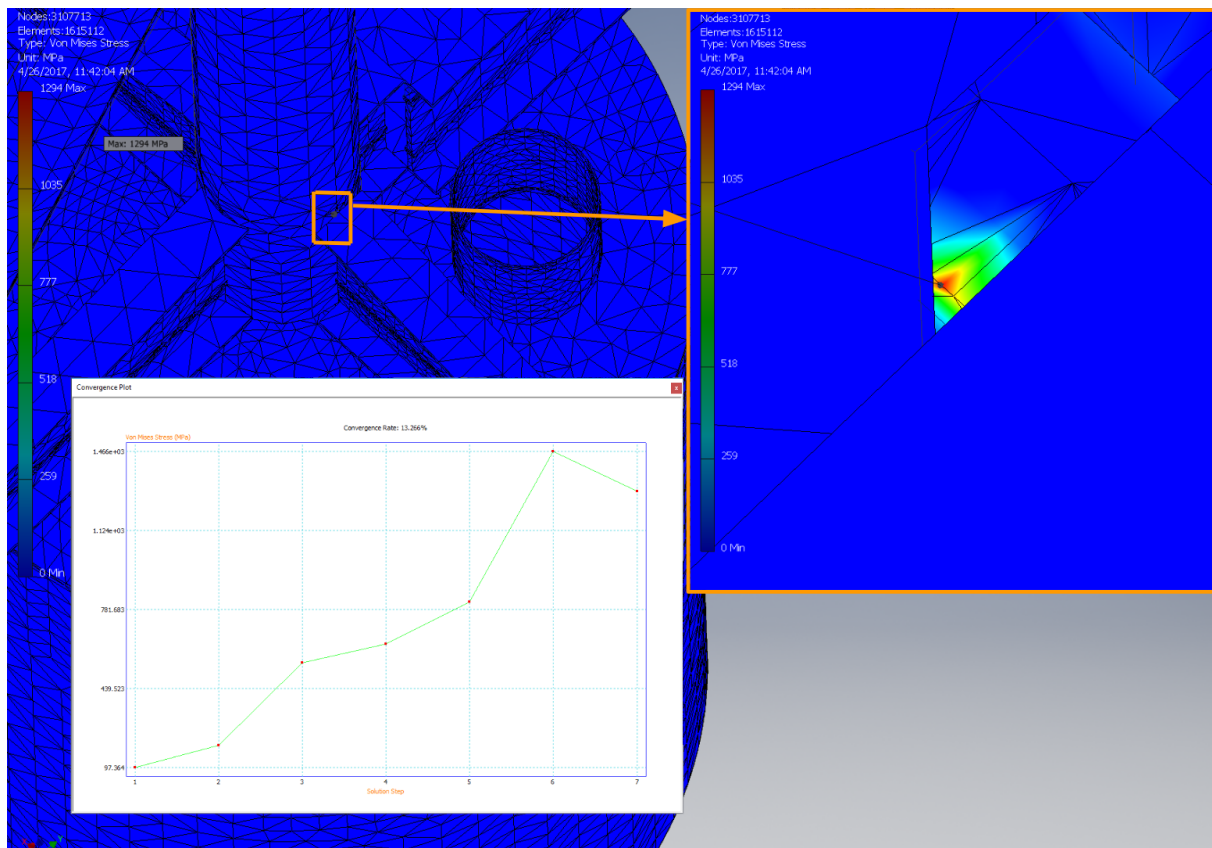


Figure 4.15: Stress singularity for case-B, with convergence plot.

Case-C

Case-C represent a combination of trawl ground rope snag, manifold weight and tie-in loads. The simulation set-up can be seen in appendix I.3, the resulting von-Mises peak stress is 368 MPa at a sharp corner on the beam subjected to trawl ground rope snag. A convergence plot has been created for the local stress of 368 MPa and the result as well as stress distribution and name of critical locations can be seen in figure 4.16. The convergence plot is not converging towards a final value and the stress is very localized around a sharp edge. The resulting stress of 368 MPa is therefore disregarded as a valid result.

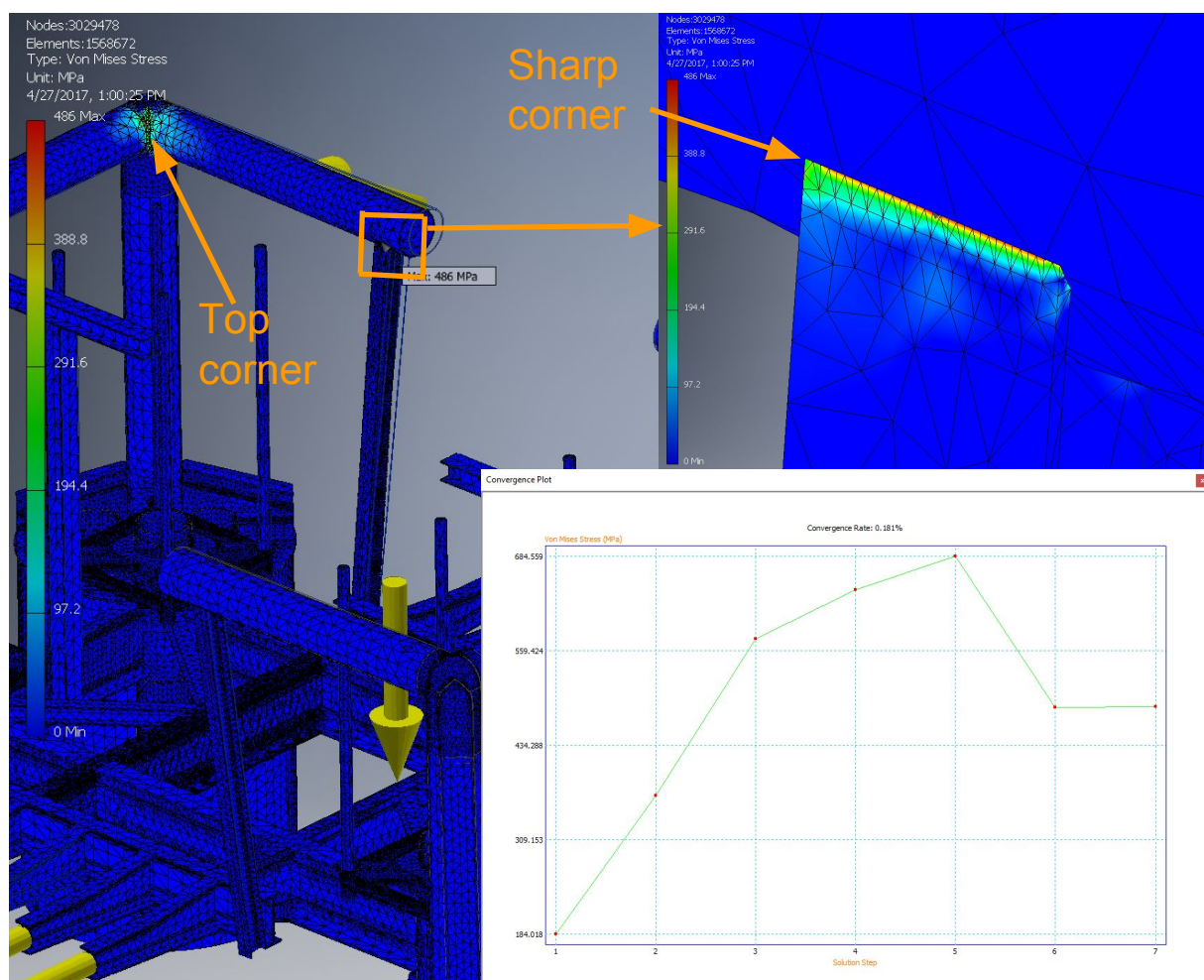


Figure 4.16: Stress singularity for case-C, with convergence plot.

The same case shows high stress in the top corner exposed to trawl ground rope snag. The local stress in the corner is approximately 300 MPa as seen in appendix I.3.1. A convergence plot has been created for this location as well, seen in figure 4.17. The convergence plot is converging and a von-Mises peak stress of 483.6 MPa can be observed in appendix I.3.2.

The modelled structure is not designed to withstand these loads without any improvements, however the displacement is very low at the location of the peak stress.

As described in section 4.3.3, the 1000 kN load is very conservative, a new simulation with 400 kN ($2 \cdot 200$ kN) instead of 1000 kN is therefore performed to see if the top corner can withstand trawlnet friction. The resulting stress is 230.4 MPa at the location of stress singularity (sharp corner) and approximately 140 MPa at the critical location in the top corner, as seen from von-Mises stress contour plot in appendix I.3.4. The difference between the load scenario of 1000 kN (not overtrawable structure) and 400 kN (overtrawable/snagfree structure) is therefore significant, as the structure is designed to withstand the loads applicable to an overtrawable/snagfree structure. While it is not designed to withstand the trawling snag load of 1000 kN at the given location.

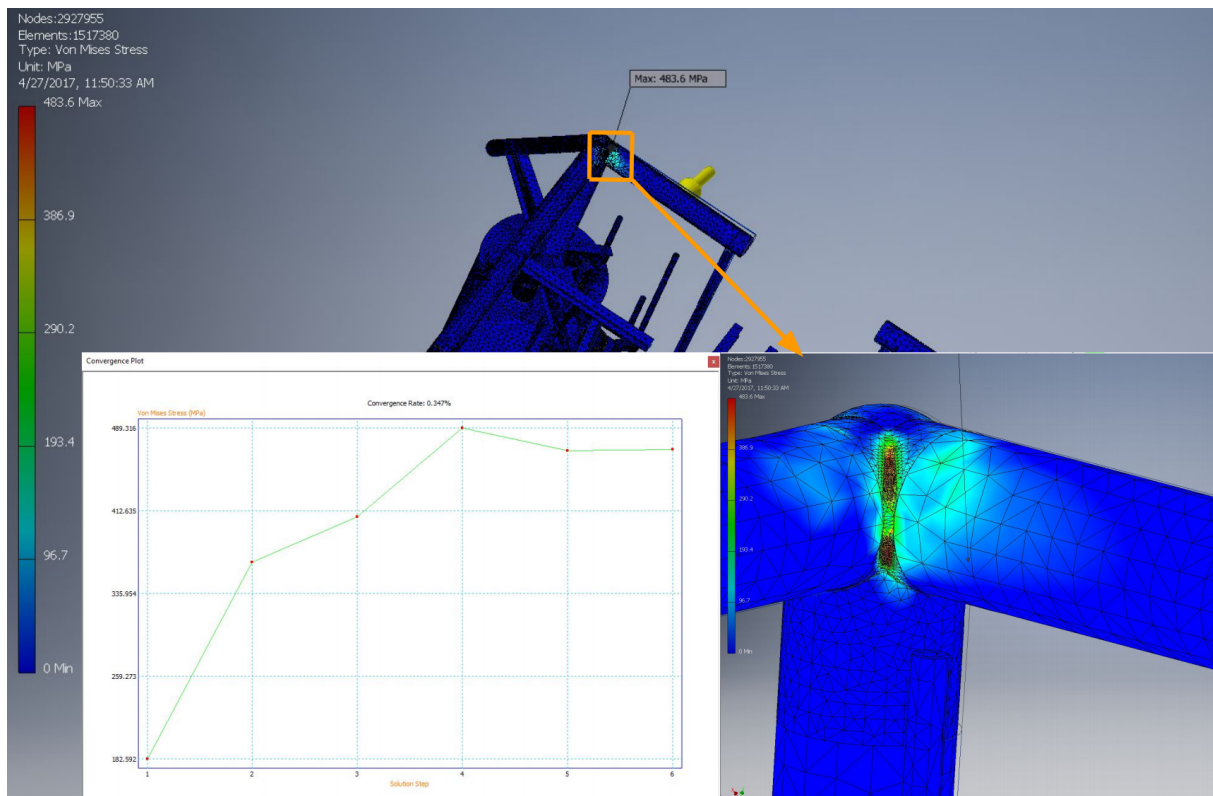


Figure 4.17: Stress in top corner for case-C, with convergence plot.

Case-D

Case-D represents a combination of X-mas tree weights, drilling loads, manifold weight and tie-in loads as shown in table 4.5. The set-up showing each of the forces and constraints can be seen in appendix I.4. The resulting von-Mises peak stress is 127 MPa at a location close to a well exposed to both drilling loads (450 kN vertical and 200 kN horizontal) and X-mas tree load (600 kN vertical). The resulting minimum safety factor is 2.71 which is acceptable. This design case shows how conservatively the structure is designed with respect to drilling and subsea equipment loads. The highest displacement is located where the structure is subjected to tie-in loads and the highest displacement value is 17.7 mm. All results for this load case are acceptable.

Case-E

Case-E represents a combination of trawl ground rope snag, drilling load, manifold weight and tie-in loads as shown in table 4.5. The case set-up and results can be viewed in appendix I.5. The resulting von-Mises peak stress of 283.3 MPa is located at the edge of a sharp corner, as seen in figure 4.18. The same location show high displacement and a convergence plot has been made to decide whether this is caused by a stress singularity or not. As seen from figure 4.18 the high stress is clearly caused by a stress singularity and is therefore disregarded as a valid result. The convergence plot shows an exponentially increase in stress with increased mesh density. The rest of the structure is limited to approximately 60 MPa in von-Mises stress, seen on stress contour plots in appendix I.5. The results for this load case are acceptable when the stress singularity is disregarded as a valid result.

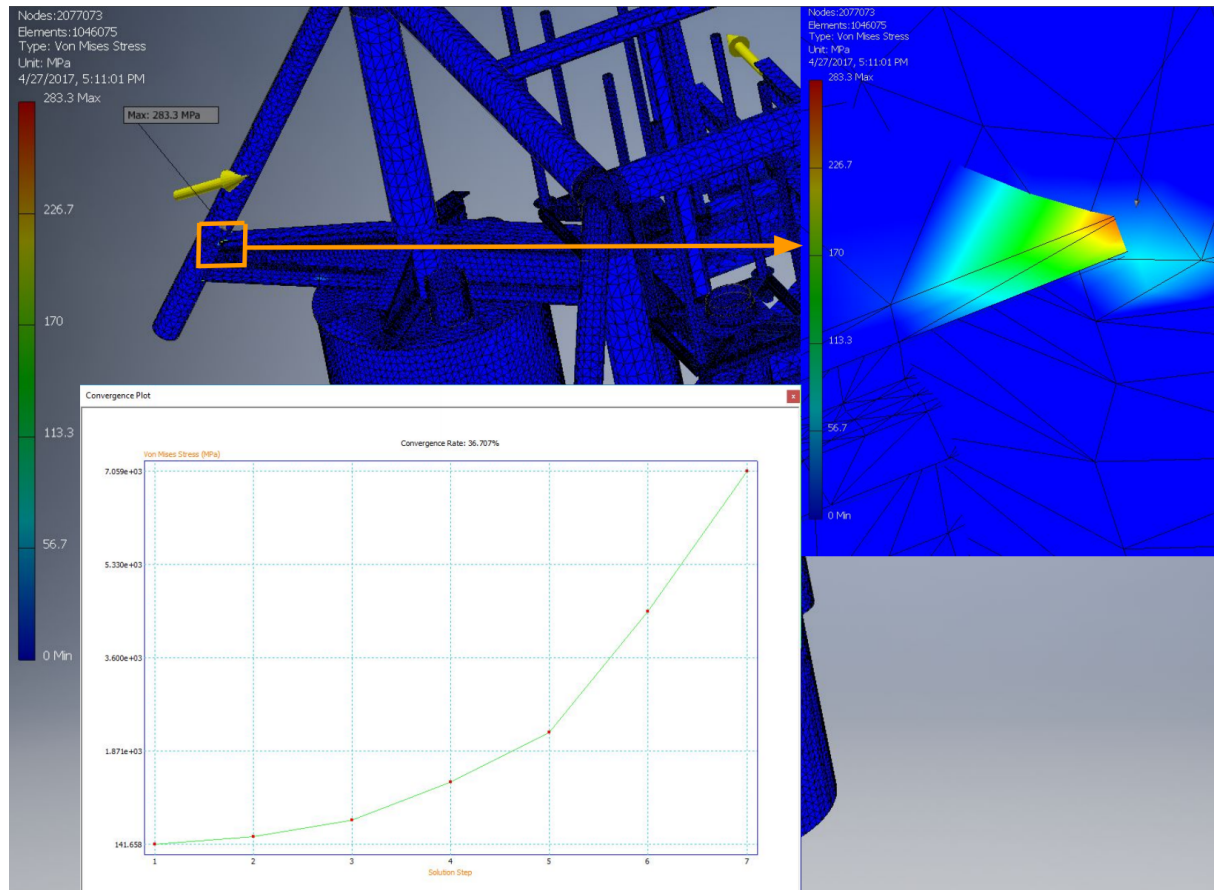


Figure 4.18: Local stress for case-E, with convergence plot.

Case-F

Case-F represents a combination of trawl ground rope snag, X-mas tree weights, manifold weight and tie-in loads. The trawling load is applied in the same way as in case-E. The set-up and results can be seen in appendix I.6. The resulting von-Mises maximum stress is 344 MPa at a sharp edge. A convergence plot has been made to decide if the high stress is caused by a stress singularity or not, as seen in figure 4.19 the peak stress is clearly caused by a stress singularity and is therefore disregarded as a valid result. The rest of the structure is limited to approximately 100 MPa in von-Mises stress. The maximum displacement is caused by the trawling load and results in a displacement of 65.1 mm, as seen in appendix I.6.2. The results for this load case are acceptable when the stress singularity is disregarded as a valid result.

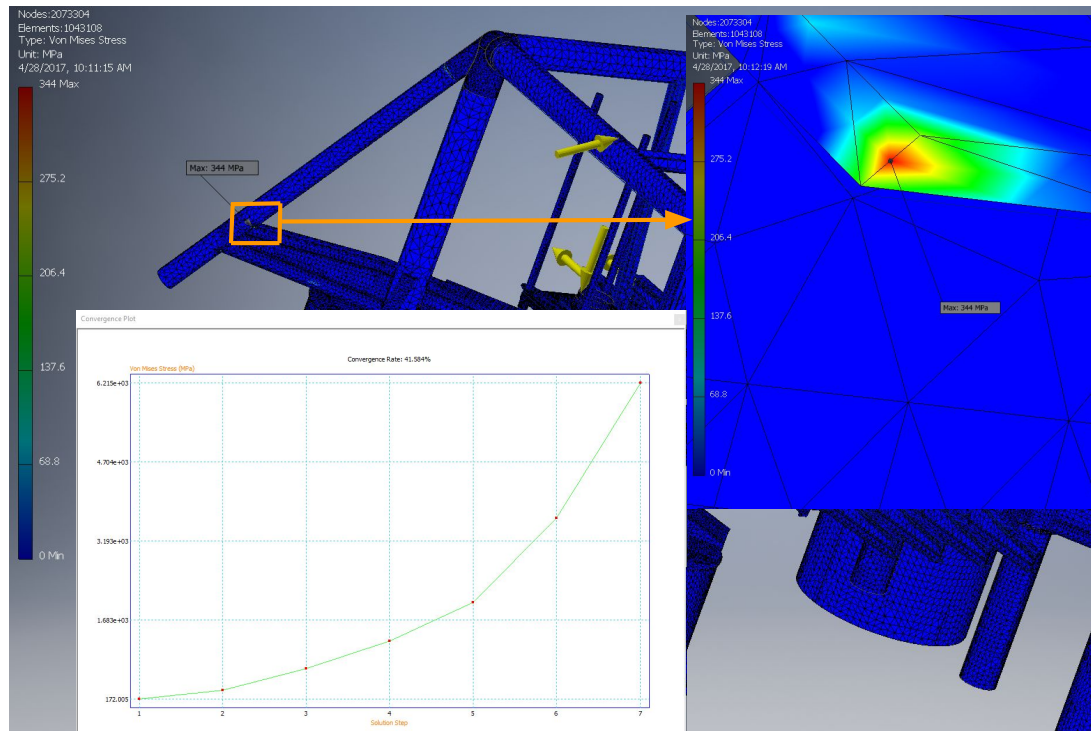


Figure 4.19: Local stress for case-F, with convergence plot.

Case-G

Case-G represents an offshore installation case where the load in each corner is calculated to 766.5 kN in section 4.3.1. There are two lifting arrangements in each corner resulting in a bearing force of 383.3 kN modelled as a bearing load. The case set-up and results can be seen in appendix I.7. The resulting von-Mises peak stress is 65.5 MPa resulting in a safety factor of 4.2, seen on stress contour plots in appendix I.7. This safety factor is based on the yield strength of aluminium, while the location of peak stress is also subjected to MIG welding reducing the minimum safety factor to 2.3 (150 MPa/ 65.5 MPa) when welding effects are considered, as described in section 3.6. The highest displacement is located on the lifting arrangement and limited to 2.5 mm. All results for this load case are acceptable, and the structure seems to be conservatively designed with respect to offshore lifting operations.

Summary

Table 4.6 contains a summary from all the cases with the most important findings. Case-C is the only problematic case as the stress is very high, 483.6 MPa. Possible methods to mitigate the high stress level is discussed in section 6.5.1. All the other cases are acceptable, as shown in table 4.6.

Table 4.6: Summary from analyses for all the selected cases.

Case #	Stress ¹⁾ [MPa]	Displacement [mm]	Safety factor ²⁾ [-]	Singularity stress ³⁾ [MPa]
A	50	93.3	4.30	189.7
B	80	94.75	2.69	194.7
C ⁴⁾	483.6 (140)	244.4 (101.8)	0.44 (1.54)	368 (230.4)
D	127	17.7	1.69	N/A
E	60	19.6	3.58	283.3
F	100	65.1	2.15	344
G	65.6	2.5	2.3	N/A

1) The approximate maximum stress when stress singularities are disregarded.

2) Safety factor as a ratio between yield strength and maximum valid stress.

3) The measured stress at a stress singularity.

4) 1000 kN trawl load case with a high stress location, 483.6 MPa and a stress singularity of 368 MPa. Numbers in parenthesis are for the 400 kN load case.

4.5 Anode mass estimation

An estimation of the total anode mass is necessary to conclude whether the weight of anodes will have a significant impact on the total mass and to perform cost estimations for anodes in chapter 5. The estimation is based on NORSOK-M-CR-503 [11], and DNV RP-B401[47]. Cathodic protection of the relevant aluminium alloys in seawater can be obtained for potentials ranging from -830 mV to -1130 mV vs SCE with low risk of corrosion [19], which is acceptable for protection of steel in seawater.

Aluminium anodes are selected due to their high capacity compared to zinc anodes, the technical data for the different anodes can be seen in table 4.7. According to DNV-B401 [47], the following should be applied for cathodic protection of aluminium in seawater:

"A design current density of 0.01 A/m² is recommended for initial, final and mean value ... the design current density shall be increased by 0.0002 A/m² for each °C that the metal/seawater is assumed to exceed 25 °C."

Table 4.7: Anode specifications according to section 6.9.1 in NORSOK-M-CR-503 [11].

	Aluminium based anode	Zinc based anode
Capacity, ϵ [Ah/kg]	2000	780
Potential [mV] vs Ag/AgCl/ Seawater	-1050	-1030

The temperature is not expected to exceed 25 °C and 0.01 A/m² will therefore be used in calculations. A design lifetime of 25 years will be used, as subsea structures are normally designed to last between 20 and 30 years. An anode utilization factor of 0.9 is imposed, as shown in appendix E and described in the DNV standard [47]. The current demand will be calculated in accordance with section 7.4, "Current Demand Calculations", in the DNV standard [47]. The surface area to be protected is calculated using the CAD-model in Autodesk Inventor, where a surface area of 1521 m² is estimated. The anode mass calculation is based on section 7.7, "Anode Mass Calculations", in the DNV standard [47]. All the input numbers can be seen in table 4.8 with reference to where the input numbers are found. The current demand calculation and the anode mass calculation can be seen in table 4.9. The number 8760 refers to the number of hours in one year.

The total anode mass must also provide current to suction anchors and well-slots, as described in NORSOK M-CR-503 [11]. Mud mats, suction piles and skirts made of steel exposed to sediments shall be designed with a current drainage of 20 mA/m², the same number is used for aluminium in calculations, as no recommendations have been found for aluminium in the literature. The limited amount of literature on the topic suggests 20 mA/m² as a conservative number, the earlier mentioned TSA in soil, outlined in section 3.5.4, had a current drainage of 10 mA/m² after ca. 250 days [21]. A subsea well shall be designed with a current drainage of 8 amps per well [11]. The additional drainage from anchors and wells will have a large impact on the total anode mass as the current raises from 15.21 amps for seawater exposed aluminium surfaces to 56.21 amps for the structure accounting for submerged anchors and all well-slots made from steel, as seen in table 4.9 and appendix E.

Table 4.8: Input numbers used in estimation of anode mass.

$i_c^{1)}$ [A/m ²]	0.01	Section 6.3.11 in DNV-RP-B401 [47].
$u^{2)}$ [-]	0.9	Appendix A- Table 10-8 in DNV-RP-B401 [47].
$A_c^{3)}$ [m ²]	1521	Surface area from CAD-model.
$I_{well}^{4)}$ [A]	8	8 amps per well, from 6.5 in NORSOK M-CR-503 [11].
$i_{sa}^{5)}$ [mA/m ²]	20	20 mA/m ² for steel suction anchors, from 6.4 in NORSOK M-CR-503 [11].
$A_{sa}^{6)}$ [m ²]	112.5	Outer surface area of one suction anchor.
$tf^{7)}$ [years]	25	

1) i_c : Design current density for aluminium components, recommended for initial/final as well as mean value.

2) u : Recommended anode utilization factor for CP design calculations.

3) A_c : Total surface of all aluminium components on the structure.

4) I_{well} : Current addition of 8 amps per well.

5) i_{sa} : Current drainage for mud mats, skirts and piles.

6) A_{sa} : Surface area pr. suction anchor.

7) tf : Design life.

Table 4.9: Current demand and anode mass calculations.

	Formula	Result
Current demand aluminium, I_{Al}	$I_{Al}=A_c \cdot i_c$ [A]	15.21 A
Current demand wells, I_{wells}	$I_{wells} = 4 \cdot I_{well}$ [A]	32.0 A
Current demand suction anchors, I_{sa}	$I_{sa} = i_{sa} \cdot A_{sa} \cdot 4$ [A]	9.0 A
Total current demand, I_c	$I_c=(A_c \cdot i_c) + I_{wells} + I_{sa}$ [A]	56.21 A
Anode mass calculation, M_a	$M_a=(I_c \cdot tf \cdot 8760)/(u \cdot \epsilon)$ [kg]	6840 kg

The weight of anodes are significant compared to the total weight of the structure, it is therefore included in calculation of the structures mass in section 4.6. From table 4.9 it is clear that wells and suction anchors are the main source for current drainage, responsible for approximately 73% of the current. The aluminium surfaces exposed to seawater are responsible for ca. 27% of the current drainage. The use of coating on these surfaces as a strategy to reduce the anode mass requirement would therefore have a very limited effect, as the suction anchors and wells would have the same current drainage. Current demand for coated aluminium is not included in the reviewed standards, NORSOK and DNV standard on cathodic protection. Further calculations on the impact of coating aluminium is therefore left out of scope.

4.6 Weight saving

The overall weight has been decreased significantly due to the introduction of aluminium. Table 4.10 shows all changes for all components, except for lifting eyes, suction anchor ventilation holes and coating weight. These weights are assumed to be insignificant (estimated to be maximum 1,5Te).

Table 4.10: Weight changes for each component, and total weight comparison.

Component	Weight of original component [kg]	Weight of new component [kg]	Number in assembly	Weight of original components [kg]	Weight of original components [kg]	Weight ratio ²⁾ [%]
a	12439	4 193	2	24879	8387	0.663
b	397	265	4	1590	1061	0.332
c	1590	1213	2	3180	2426	0.237
d	1390	928	2	2780	1856	0.332
e	575	293	4	2302	1174	0.490
f	3861	1555	4	15444	6221	0.597
g	2452	1273	2	4905	2547	0.481
h	948	483	8	7582	3866	0.490
i	2656	1136	4	10623	4544	0.572
j	605	288	8	4837	2301	0.524
k	790	417	4	3160	1668	0.472
l	4486	2610	4	17946	10442	0.418
m	2147	1292	4	8418	4872	0.421
n	4209	2436	2	8418	4872	0.421
o	2447	1434	4	9789	5738	0.414
q	20442	11755	4	81766	47021	0.425
q1	626	361	32	20038	11560	0.423
q2	438	307	4	1750	1229	0.298
q3	3984	2482	4	15934	9928	0.377
r ³⁾	2021	2021	4	8082	8082	0.000
s	1023	804	2	2046	1607	0.214
t	883	568	2	1767	1136	0.357
u ³⁾	1918	1918	4	7672	7672	0.000
v ³⁾	1085	1085	4	4341	4341	0.000
w	179	95	16	2856	1520	0.468
x	153	78	16	2453	1240	0.494
y	802	481	16	12836	7688	0.401
z ³⁾	321	321	8	2569	2569	0.000
Anode weight	-	-	-	7513 ¹⁾	6840	0.090
Bolt weight	-	-	-	0	14000	-
Bolting plates	-	-	-	0	2600	-
Lifting slings	-	-	-	12500	12500	-
Sum:	72326	40035	174	310142	202718	0.346

1) Anode mass calculations is based on [47, 11]. Calculations can be seen in appendix E.

2) Weight ratio is defined as the weight reduction between original and new weight.

3) Made of steel.

The calculated weight is reduced from 298Te to 191Te by utilizing aluminium, which is a weight reduction of 36%. 18.1% of the new weight is steel. Anode weight is included in the mentioned weights but weight of lifting slings are excluded. As a result of this weight saving lighter subsea construction vessels can be used (250Te crane vessel class instead of 400Te crane vessel class), reducing the cost of installation. The cost implications are described in detail in chapter 5.

Chapter 5

Cost Estimations and Comparison

An introduction of aluminium in subsea structures are not likely to happen if it is not cost efficient. It is therefore necessary to perform a cost estimation comparison study. The study will only focus on capital expenditure (CAPEX), operating expenditures (OPEX) will not be a part of the study to limit the scope of work.

5.1 Installation cost

The calculated steel structure weight is 298Te, while the calculated aluminium structure weight is 191Te. The weight difference of 107Te results in a larger number of vessels capable of installing the structure. To estimate an installation cost for the two alternatives, it is necessary to estimate an installation time. The installation time in days is set to 3, 6 and 8 days, as a reference in Subsea 7 estimated 3 days for such installations [85], a source in Seabrokers estimated 5-7 days [86], while the software program Que\$tor used for cost estimations estimated 8 days for the installation of an ITS as seen in appendix G. The number Que\$tor estimated were identified by setting up two cases, a case with one ITS and a case with two integrated template structures, and 8 days were the installation time difference as seen in appendix G.2 and G.3. All sources have been used in the cost estimation to point out the uncertainties related to installation time. The difference in installation cost can be seen in table 5.1. The daily rate is the total cost of renting operational vessel (including onboard personnel) [39], the e-mail correspondence with Asbjørn Wathne in Subsea 7 can be seen in appendix B.5. It should be noted that the daily vessel rate changes a lot with the market situation as pointed out in e-mails from both Seabrokers [39] and Subsea 7 [85]. Vessel rates

can therefore differ from the numbers used in the table 5.1, dependent upon the market situation. The daily vessel rates numbers should therefore not be used for any cost estimations in the future, as they reflect the current market.

Table 5.1: Installation vessel size influence on installation cost.

Vessel crane size	Daily rate [NOK/day]	Installation time [days]	Installation cost [NOK]
250Te	1 500 000	3	4 500 000
		6	9 000 000
		8	12 000 000
	1 750 000	3	5 250 000
		6	10 500 000
		8	14 000 000
	2 000 000	3	6 000 000
		6	12 000 000
		8	16 000 000
400Te	2 000 000	3	6 000 000
		6	12 000 000
		8	16 000 000
	2 250 000	3	6 750 000
		6	13 500 000
		8	18 000 000
	2 500 000	3	7 500 000
		6	15 000 000
		8	20 000 000

5.2 Fabrication cost

Fabrication cost for both steel and aluminium have been estimated to conclude whether aluminium is a cost efficient solution or not. Estimated costs for both alternatives are presented in this section.

5.2.1 Steel structure cost

The cost estimation program Que\$tor from IHS Markit [87] has been used to find the cost for an steel made ITS. The program is based on historical development costs in the oil & gas industry. A Gjøa field was generated in the software by selecting structures and field specifications according to references [88, 89]. The field layout can be seen in figure 5.1, where the left figure shows the overall layout while the right figure shows the subsea layout with three integrated template structures.

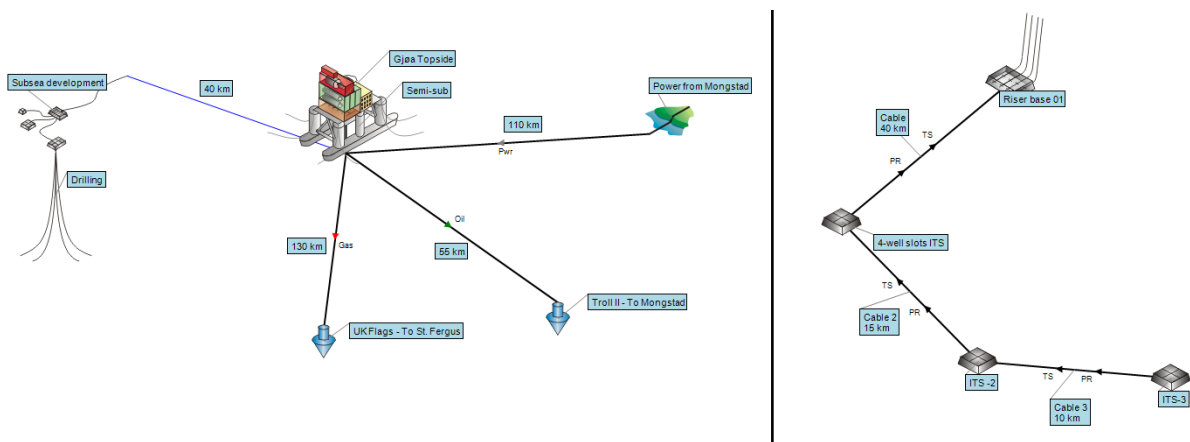


Figure 5.1: Gjøa south field layout, and subsea layout with three integrated template structures.

The cost estimation found in Que\$tor for a 4-well slots ITS can be seen in table 5.5. The total cost of the main structure is 6 361 000 USD, equal to ca. 53 854 000 NOK (dependent upon the exchange current between USD and NOK). All the information extracted from the Que\$tor software can be seen in appendix G.

5.2.2 Aluminium structure cost

The aluminium fabrication cost is not possible to identify using the Que\$tor software, as an aluminium made ITS does not exist. Fabrication cost for the aluminium alternative is developed by contacting industrial companies and asking for price offers on all items. The aluminium cost has therefore been divided into the following sub-groups: fabrication cost, extrusion cost, plate cost, bolt cost and anode cost.

Aluminium fabrication cost

The fabrication cost is the cost of constructing the structure when the extrusion profiles, plates, bolts and anodes have been purchased. Based on numbers from Marine Aluminium, the cost of fabricating such an aluminium structure is estimated to be in the range of 200 to 300 NOK/kg, as seen from e-mail correspondence in appendix B.6. A cost estimation can be carried out based on these numbers combined with the total weight of the aluminium structure. The cost estimations can be seen in table 5.2, where the calculated structural weight is combined with several options for NOK/kg to accommodate the uncertainties regarding these values. The cost per kg (NOK/kg) numbers presented in table 5.2 are rough estimates, more details regarding cost uncertainties are presented in section 6.8.

Table 5.2: Cost estimation for aluminium based on numbers in e-mail correspondence seen in appendix B.6.

Total weight [Te]	Cost per kg [NOK/kg]	Fabrication cost [NOK]	Fabrication cost [MNOK]
191	200	38 200 000	38.20
191	250	47 750 000	47.75
191	300	57 300 000	57.30

Extrusion cost

The extrusion cost is estimated from numbers received from STEP-G. STEP-G's price offer is based on the part-list for extrusion profiles in appendix H.1 which covers all extrusion profiles used on the aluminium made ITS. The received price offer from STEP-G can be seen in appendix H.2. The cost of extruded profiles is estimated to 3 532 000 NOK, using 9.45 as exchange current between NOK and EUR. 1/3 of the extrusion cost is directed to manufacturing of the dies needed. The estimated cost is excluding VAT (Value Added Tax) and including European Standard certificate.

Plate cost

Plate cost is based on price offer received from Constellium, the price offer can be seen in appendix H.1 covering all plates included in the ITS design. The total cost for plates including DNV certificate and transport to Norway is approximately 550 300 EUR. Using 9.45 as

exchange current between NOK and EUR results in 5 200 340 NOK.

Bolt cost

Costs for stainless steel type 316 and carbon steel obtained from AccuGroup [12], and additional phosphating costs received from Odda Coating Technology AS [13], are presented in table 5.3. Phosphating is chosen as the optimal coating method mainly due to the possession of electrical continuity, see section 3.8 for bolt coating alternatives and bolt properties. Required bolt length and respective quantities are estimated from results provided in table 4.3.

Table 5.3: Cost comparison between stainless, carbon and phosphated carbon M36 steel bolts [12, 13].

Bolt length [mm]	Quantity ²⁾	NOK/ stainless bolt	NOK/ stainless nut	NOK/ stainless washer	NOK/ carbon steel bolt ¹⁾	NOK/ carbon steel nut ¹⁾	NOK/ carbon steel washer ¹⁾
110	120	188	67	21	69 (219)	9 (69)	7 (62)
120	352	200	67	21	73 (223)	9 (69)	7 (62)
130	68	207	67	21	76 (226)	9 (69)	7 (62)
140	256	211	67	21	80 (230)	9 (69)	7 (62)
150	256	222	67	21	84 (234)	9 (69)	7 (62)
210	416	352	67	21	111 (261)	9 (69)	7 (62)
Sum	1468	364 442	98 826	30 883	126 345 (347 180)	13 451 (101 531)	20 314 (181 794)
Sum stainless steel bolts, nuts and washers		494 151 NOK					
Sum carbon steel bolts, nuts and washers ¹⁾		160 764 NOK (630 506 NOK)					

1) Cost for phosphated carbon steel bolts are given in parentheses.

2) Double amount of washers (2936).

5.2.3 Anode material cost

The anode cost estimations are based on prices received from Skarpenord, e-mail correspondence can be seen in appendix B.7. The estimated anode cost is 44 NOK/kg. Table 5.4 shows the estimated cost for 6840 kg of anodes as calculated in section 4.5.

Table 5.4: Anode cost estimation.

Anode cost [NOK/kg]	Anode mass [kg]	Anode cost [NOK]
44	6840	300 960

5.2.4 Total cost estimates comparison

The installation of an aluminium made ITS involves a 250Te crane vessel instead of a 400Te crane vessel which is the case for steel. Table 5.5 identifies the lowest cost case for aluminium as a cost competitive alternative to steel, while the upper cost case identify steel to be the most cost efficient choice. Aluminium fabrication cost is the decisive factor. It should be noted that these numbers are estimates and will differ from one field to another (as the distances changes, depth changes, installation time changes, etc.).

Table 5.5: Cost comparison for the two alternatives (aluminium and steel).

	Aluminium	Steel
Fabrication cost [NOK]	38 200 000 to 57 300 000	53 854 000
+ Extrusion cost [NOK]	3 532 000	N/A
+ Plate cost [NOK]	5 200 340	N/A
+ Bolt cost [NOK]	630 506	N/A
+ Anode cost [NOK]	300 960	N/A
+ Installation cost [NOK]	4 500 000 to 16 000 000	6 000 000 to 20 000 000
= Total cost [NOK]	52 363 806 to 82 963 806	59 854 000 to 73 854 000

Chapter 6

Discussion

6.1 Connection methods

A number of different joining methods have been presented in chapter 3, it is therefore necessary to summarize and discuss relevant findings.

Joining by adhesives, bracing and laser are not used in the aluminium design. As adhesives require a comprehensive documentation and testing, which is not beneficial in terms of costs and ultimately NORSOK approval. Bracing of alloy 5083-H116 is difficult due to the relatively large magnesium content and the resulting strength reduction is high. Laser welding is believed to be a possible welding method, but the experience and knowledge within aluminium fabrication companies are limited, and costs by implementing laser could be severe.

MIG and FSW are on the other hand identified as promising and applicable joining methods for joints that needs to be welded, and are therefore implemented in design and cost analyses. MIG is an extensively used method and likely to be the most cost efficient method [9]. FSW possess a high ductility with favorable strength characteristics, but it is limited to butt welds. TIG is also a possibility, but the strength reductions are significantly higher for larger thicknesses and it is found to be a more costly process compared to MIG.

Bolting is regarded as the optimal joining method in terms of retaining strength throughout the joints, although some extra corrosion concerns must be expected.

The new HYB method shows very interesting and promising joint characteristics. The innovative joining method is however in an early stage, relevant literature is therefore very limited. HYB is thereby not implemented in the ITS redesign.

6.2 Corrosion protection strategy

Based on the provided evidence in section 3.5.8, cathodic protection of aluminium is regarded as an efficient and simple solution to protect the aluminium structure from corrosion attacks. At higher flow rates some degradation on the aluminium surface must be accepted. Flow induced corrosion is not found to be problematic as the resulting corrosion rate is approximately 0.08 mm/year for velocities between 3 - 9 m/s, which would result in 2 mm uniform thickness loss after 25 years in service. Flow rates higher than 9 m/s in a subsea environment is regarded as highly unlikely.

Corrosion protection strategy for the redesigned ITS is mainly to use sacrificial anodes. Total anode weight is calculated in section 4.5, but anode distribution is not analyzed in this study, to limit the scope of work. However, anode distribution requirements should be clarified. The anodes must be distributed to assure that the structural steel is polarized to the immune area of the Pourbaix diagram and the aluminium alloys shall be polarized to the passive area in order to produce a protective oxide layer, which is obtained for both materials if a potential range of -830 mV to -1130 mV vs SCE is valid throughout the structure. Extra concerns should be directed to the relatively large current drainage towards wells and suction anchors.

6.2.1 Corrosion challenges

A complete understanding of corrosion of aluminium in closed compartments and soil is necessary before an aluminium ITS can be considered. There are indications outlined in section 3.5.4 and 3.5.5 that the approved NORSOK alloys are resistant under these conditions, but the literature found in this study is not enough to state a valid conclusion at this point.

Sealing compounds required for bolting applications must have a durability rating that satisfies desired lifetime and subsea requirements. The use of sealing compounds as an external seal is defined as the appropriate measure against water ingress in Eurocode 9 [2] for submerged applications. Suppliers of sealing compounds have not been identified, and subsequently the durability of the mentioned compound could not be confirmed during this study.

6.3 Fatigue concerns

Fatigue damage that may occur from cyclic loading during the integrated template structures lifetime has not been assessed, to limit the scope of work. It should be emphasized in an aluminium redesign that steel possess a greater fatigue resistance. Possible fatigue scenario are cyclic loading during sea transport. Trawling, drilling and unstable production through X-mas may also be a source of fatigue on the ITS, but that has not been confirmed. The ITS is however known to be subjected dominantly by static loading.

A safe life fatigue design described in section 3.12 is believed to be appropriate, since it is probably not cost efficient to perform the inspections required subsea for the damage tolerant design.

6.4 Design implications

The design of the ITS has been a time consuming iterative process with a number of considerations, especially with respect to aluminium extrusion limits outlined in section 3.3. Note that each extrusion is not only restricted by circumferential limits but also by a maximum weight per extrusion of 535 kg [4], which ultimately restricts the extrusion length. The redesign must in addition be within acceptable values for MOI, buckling and weight reduction.

6.4.1 Deflection

Larger beam deflection due to lower E-modulus for aluminium compared to steel has not been regarded as a design issue for the redesigned beams, although thickness among other dimensions is increased in order to meet MOI criteria. Considering the overtrawlable grid's allowance to experience plastic deformation (larger displacement) according to NORSOK U-002 [14] and DNV-RP-C204 [90] from accidental loads defined as a PLS condition (e.g. trawl snagging). Deformation of the overtrawlable grid should not lead to difficulties opening and closing the protective hatches covering the manifold and X-mas trees. In addition, the resulting elastic deformations in the FEM model load cases seen in section 4.4 is not of any concern. Only load case-C has a stress concentration beyond the elastic region, which is further discussed in section 6.5.1.

6.4.2 Insert

The joint in the top corners of the overtrawlable tubes is solved by using an insert to connect all four tubes. Manufacturing of the insert has proven to be challenging. The aluminium block required to machine the insert in one piece is too big for any billets identified in this study, which means that it is not found a proper manufacturing method for the insert in the approved NORSOK M-121 [44] alloys for submerged applications. It is further difficult to cast the insert in the 5xxx or 6xxx series, since the 6xxx series is not used for casting and the 5xxx series would be very difficult to temper to an adequate strength, as it is NHT and strengthening by cold work is not feasible on a casted block. To limit the work, manufacturing of the insert has not been analyzed in detail in this study. However, two possible solutions have been evaluated; the insert could be casted in an unapproved NORSOK alloy, for example in the 7xxx series and heat treated to desired strength, another possibility is to conduct further attempts to optimize the joint design by welding the tubes together. Note that the overtrawling requirement (smoothness) makes the joint design more difficult. A price estimate for the insert has not been identified, as details concerning the insert fabrication method is unknown.

6.4.3 Bolt calculations

Some of the loads and moments used as inputs in section 4.2.1 are established based on some simplified judgments, and should therefore be discussed. The joint loads along beam 'a' is based on design moment calculated from original geometry of beam 'a', where it is assumed a maximum moment at mid point of the beam, which is reasonable for a beam supported in both ends with a symmetric distributed load. However, the maximum original design moment found in the bolt analysis could in theory be spread evenly over a larger area than just the "mid point", but that is highly unlikely due to the resulting shear force in each beam support, which would be unreasonably large compared to known vertical loads acting on the base frame. The assumptions made in section 4.2.1 is supported by results in the FEM analysis performed in section 4.4 in the sense that it is conservative. It is not found any stress concentrations of concern in any of the bolted joint locations, meaning the geometry in these joints are well within desired safety factor.

Hydrogen embrittlement effects on the alternative stainless steel bolts connected to a potential source of hydrogen from CP, has not been analysed in this study. It is observed that stainless steel bolts shall only be used with a diameter of 10 mm or less according to NORSOK M-001 [54]. Nevertheless it is of interest to compare both carbon and stainless steel bolts in terms of required corrosion protection, amount of bolts and costs, as the utilization of large aluminium structures permanently installed subsea is still in an early stage.

6.4.4 Manifold support

The extensive use of bolting assemblies may interfere with the manifolds supporting points on the ITS, due to a non uniform beam surface on the assembly area. The manifolds interaction with the integrated template structures beams have not been studied in detail.

6.4.5 Formability

Bending of 5083-H116 plates are required in order to erect some of the parts, especially tubes and the plates intended to shape the suction anchor. As mentioned in section 3.10, one must reassure that the minimum bending radius is not too sharp. It has not been identified any challenges with respect to the radius limit. A company with sufficient equipment for the required manufacturing of the overtrawlable tubes have in addition been identified [79].

6.5 FEM Analysis

The structural analyses performed in section 4.4 is based on some simplifications to limit the amount of work. These simplifications will be a source of error in the analysis, and should therefore be discussed. Welding spots are not included in the model, welded locations on the structure will have reduced mechanical properties as described in section 3.6. It is therefore necessary to study the stress results with respect to stress in welded areas in order to obtain appropriate safety factors. Dropped object loads are excluded from the analysis as impact simulations required to perform such analysis has been left out of scope to limit the work load.

6.5.1 Validation of modelled cases

All design load cases in section 4.4.1 are acceptable except for the original case-C, where the trawling load of 1000 kN is too large for the modelled structure while the 400 kN load case is acceptable. The model does not take into account the protection covers and connection of the whole long side. These components are believed to contribute to distribute the stress over a larger part of the structure. Figure 6.1 shows the mentioned components and their locations. The stress distribution caused by these components are not further discussed in this study, to limit the scope of work.

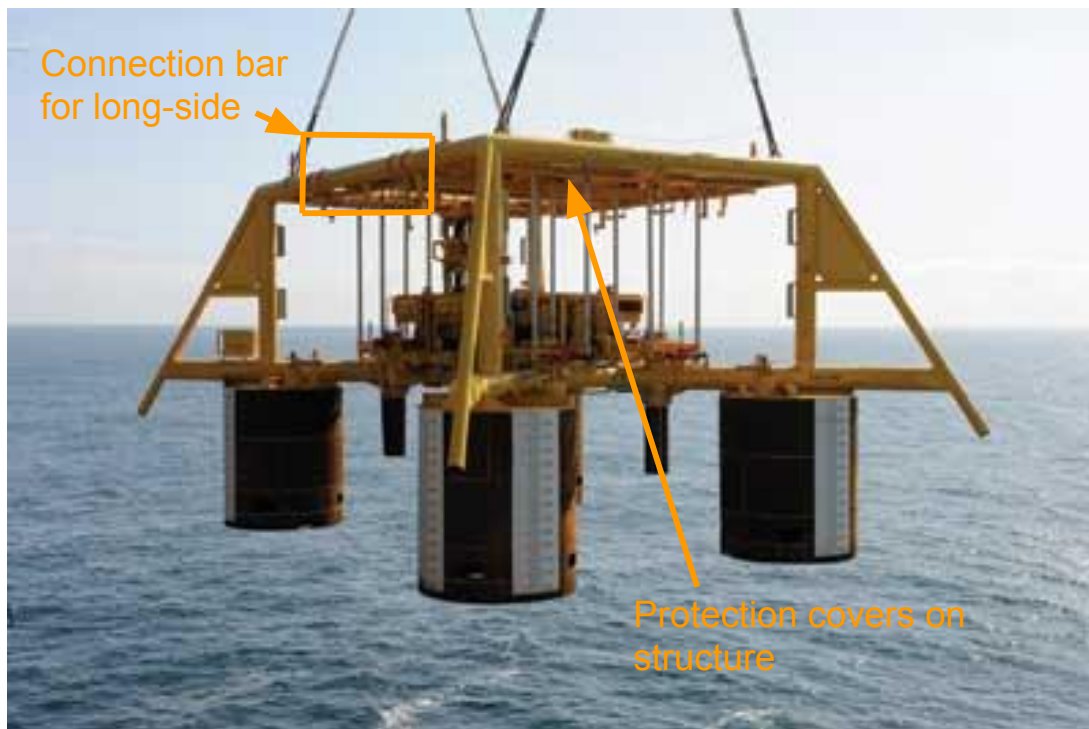


Figure 6.1: Integrated template structure with protection structures on top and connection of the whole side top beam [33].

In the set-up of case-G, the suction anchors has been set as "fixed" items meaning they do not have any displacement. This is not correct, as case-G is ment to describe a lifting scenario. Only the lowest tip of the suction anchors should have been set as "fixed" rather than the whole suction anchor. It is not known to which extend this influences the results, but there is no reason to believe it would change the outcome of the design as the identified safety factor is relatively high, 2.3 as shown in table 4.6 in section 4.4.2.

6.5.2 Dropped objects

Analyses of dropped objects has been left out of scope, but should be discussed. Protective hatches are excluded in this study, a realistic model of the different impact scenarios on the ITS can therefore not be conducted. Impact properties are however presented for aluminium in section 4.4.1. Center for advanced structural analyses (CASA) at NTNU has a lot of experience with impact testing on aluminium, which makes it a suitable organization for further impact research on subsea aluminium structures and hatches.

6.6 Weight saving

Total weight saving of 36% between original steel design and new aluminium design is achieved, which is regarded as an adequate reduction considering the beams that have not been changed to an aluminium design and amount of steel bolts. The new obtained weight (including lifting slings) is 202.7Te, which is just within reach for 250Te installation vessels, which saves 500 000 NOK/day compared to 400Te vessels.

6.7 Coating

Operating companies responsible for field developments may specify a coating or colour for aluminium as it is currently for steel, based on the yellow coating's high visibility and extensive use in today's subsea structures. This possible requirement from the field developer has not been studied any further, but it should be mentioned that an extra layer of coating would increase the fabrication cost and time.

6.8 Cost uncertainties

The cost estimations are not accurate for future use, as the material cost of both aluminium and steel are changing continuously. As mentioned in section 5.1, the daily vessel rates are based on the current market, the presented numbers are therefore not applicable for future market changes. There are some uncertainties related to the installation time, as several references reported different installation times. An upper and lower installation cost were calculated based on the difference in estimated installation times to accommodate the uncertainties. This approach will impact the accuracy of the total cost estimation, as the total cost estimation is given as a range.

Freight cost for extrusions from STEP-G is excluded in the cost estimates, they are however expected to be insignificant since the parts submitted to STEP-G, located in Germany, fits

in large commercial trailers. The VAT (Value Added Tax) is excluded from the cost estimates from STEP-G and Constellium, the VAT for import goods to Norway is ca. 25 %, dependent upon the goods [91]. The plate cost estimate from Constellium includes freight cost, where the destination for delivery is set to Trondheim/Norway.

Management costs related to design and manufacturing are not included in the cost analyses, which can have a significant influence on the total cost estimation. The cost of fabricating the inserts mentioned in section 6.4.2 has also been excluded as an estimate could not be identified because of the uncertainties regarding fabrication method. It is not clear whether inserts will have a significant impact on the total cost.

The fabrication cost numbers presented in section 5.2.2 is based on total weight, overall dimensions, experience and comparable structures. Length of welds, number of alignments and number of bolted connections are thereby not used as input in the fabrication cost estimates.

Chapter 7

Conclusion

Aluminium properties

Aluminium's properties with respect to subsea applications have been reviewed and the following remarks can be concluded:

- The aluminium alloys 5083-H116 and 6082-T6 have excellent corrosion properties in seawater. Coating of the aluminium surfaces is not necessary, as cathodic protection and passivation of the aluminium surfaces are sufficient.
- Cathodic protection mitigates galvanic corrosion between aluminium and welds and between steel and aluminium.
- Cathodic protection of aluminium surfaces is not effective at high flow rates.
- It is not fully understood how aluminium will degrade when buried in soil.
- Several joining methods can be applied to an aluminium subsea structure, where bolting and welding are the preferred methods.
- MIG welding of alloy 5083-H116 and 6082-T6 reduces the yield strength by 30% and 50% respectively. FSW welding reduces the tensile strength by ca. 30% for alloy 6082 and 5% for alloy 5083, FSW can also increase the tensile strength for alloy 5083 by up to 19%.
- Adhesives, laser welding, friction welding, bracing and HYB are joining methods that have not been used in the redesign.

Cost estimates

The following remarks can be concluded regarding cost implications by using aluminium rather than steel as construction material for an integrated template structure:

- The reduced structure weight enables installation to be performed by the use of smaller installation vessels, with lower rates (NOK/day) reducing the installation cost.
- The assembly process is the main cost for the aluminium made integrated template structure.
- Total cost for the aluminium made structure is cost competitive to steel, where the aluminium fabrication cost is the decisive factor.

Redesigned aluminium made structure

The integrated template structure has been rebuild using aluminium as the main construction material, and simulated for several load cases in accordance with industry standards. The following can be concluded concerning the redesign and performed simulations:

- The redesign results in a structure with thicker members and optimized beam profiles in order to maintain the mechanical integrity of the structure. All important offsets and main dimensions remain the same, e.g. the well centre distance, height, width and length of the structure.
- The weight of the integrated template structure is reduced from 298Te to 191Te, a weight reduction of 107Te or 36%.
- Simulations performed on the aluminium designed structure showed acceptable stress, displacement and safety factor results for the applied load cases, simulating trawling, tie-in and structural weights of manifold and X-mas trees.

Chapter 8

Future Work

- Research on aluminium in soil with respect to corrosiveness and degradation mechanisms.
- Research on aluminium in closed compartments with respect to corrosiveness and degradation mechanisms.
- Conduct more experiments on various seawater flow rates influence on corrosion rate and cathodic protection.
- Identify other relevant subsea structures where it is possible to introduce aluminium as construction material. Structures with repetitive installation and retrieval will have a larger impact on installation cost.
- Develop Pourbaix diagrams for 5083-H116, 6082-T6 and 7020-T6.
- Perform impact (dropped object) loads analysis.
- Find suitable sealing compounds to be used on steel-aluminium interfaces.
- Develop more detailed standards concerning aluminium in a subsea environment. E.g. there has not been identified any standards concerning aluminium buried in mud, as there is for steel.

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Appendix A

Loads

A.1 Splash zone loads

There are four types of hydrodynamic (dynamic) forces and one static load to consider during splash zone penetration according to DNV-RP-H103 [81], The following hydrodynamic forces have to be included:

F_m : Characteristic hydrodynamic mass force [N], dependent upon object mass in air, heave added mass caused by water flooding of the structure, vertical acceleration of crane tip, the wave displaced volume of water and vertical water acceleration due to wave motion.

F_d : Characteristic hydrodynamic drag force [N], dependent upon drag coefficient, the objects area of horizontal directed surfaces and vertical velocity for object relative to water.

F_{slam} : Characteristic slamming impact force [N]. The magnitude of this load is dependent upon impact velocity (known as slamming velocity [m/s]) and the slamming area (size of structure projected to the slamming loads).

F_{ro} : Characteristic varying buoyancy force [N], change in buoyancy as the wave height changes. Dependent upon the volume difference between water volume for stilled water and wave.

The static load is determined according to DNV-RP-H103 [21], section 4.2.2:

$$F_{static,min} = M_{min} \cdot g - r_o \cdot V \cdot g \text{ [N]}$$

$$F_{static,max} = M_{max} \cdot g - r_o \cdot V \cdot g \text{ [N]}$$

Where:

V : represents the displaced water in both cases.

M_{min} : The minimum mass is equal the mass of object in air (i.e. the structure is submerged but the flooding has not yet started) [kg].

M_{max} : The maximum mass is equal the mass of object in air including the full weight of the water that floods the structure (i.e. the structure is fully flooded after submergence) [kg].

A.2 Static loads

Table A.1: Manifold and christmas tree weights with resulting forces.

Component	Dry weight	Wet weight
Manifold	80Te (800 kN)	70Te (700 kN)
Christmas tree	60Te (600 kN)	53Te (530 kN)

A.3 Trawling loads

Table A.2: Trawling loads [14].

Design load type	Design load figure		
Trawl net friction	2x200 kN	0-20 deg. ⁴⁾	ULS ¹⁾
Trawl board overpull	300 kN	0-20 deg. ⁴⁾	ULS
Trawl board impact	13 kJ	-	ULS
Trawl board snag ³⁾	600 kN	0-20 deg. ⁴⁾	PLS ²⁾
Trawl ground rope snag ³⁾	1000 kN	0-20 deg. ⁴⁾	PLS
Trawl board snag on sealine ³⁾	600 kN	-	PLS

1) ULS means it is regarded as normal operation, something that can occur during normal operation.

2) PLS means it is regarded as an abnormal operation.

3) Negligible if the subsea structure is documented to be snagfree.

4) With respect to horizontal direction.

A.4 Dropped objects impact energy

Table A.3: Dropped objects impact energy [14].

Group	Impact energy	Impact area	Object diameter
Multi well structures	50 kJ	Point load	700 mm
	5 kJ	Point load	100 mm
Other structures	20 kJ	Point load	500 mm
	5 kJ	Point load	100 mm

A.5 Sealine loads

Table A.4: Sealine loads scenarios [15].

Sealine supports in use	Maximum load on one support	Total maximum combined load
1	300 kN	300 kN
2	300 kN	600 kN
3	200 kN	600 kN
4	150 kN	600 kN

A.6 Drilling loads

Table A.5: Drilling loads for depths up to 750 m [14].

Activity	Load case	Design load
Lowering of 30" conductor.	Weight carried by template.	Temporary 600 kN vertical.
Drilling 24", lowering and cementing of 18 5/8" casing.	Partly 30" and 18 5/8" casing will be transferred to soil via the cement, assuming settling of the structure. Normal pull off stucked drillstring (2000 kN) and rig offset 4.5 degrees, including 1.5 degrees misalignment. Vertical load will be carried by conductor. Horizontal load will be carried by template and conductor	Permanent 450 kN vertical. Vertical 0 kN, Horizontal 160 kN.
Drilling of subsequent sections.	A BOP stack with riser attached landing on template at 0.5 m/s. Impact load mainly taken by the conductor casing. Normal pull of stucked drill string (2000 kN) and rig offset 4.5 deg. Including 1.5 deg. Misalignment. Vertical load carried by conductor. Horizontal load to be carried by template. Tension from riser (300 kN) will be taken up by conductor casing. Horizontal component to be carried by template and conductor. Guideline tension, max 200 kN. Vertical load will be taken by template. Horizontal comp. from 4 off lines at 4.5 deg. to be carried by template and conductor.	Vertical 31 kJ impact, assuming 40 % on template. Vertical 0 kN, Horizontal 160 kN. Vertical 0 kN, Horizontal 25 kN. Vertical 0 kN, Horizontal 15 kN.

Appendix B

E-mail Correspondences

B.1 E-mail correspondence with Keld Reimer Hansen regarding production capabilities at STEP-G.

Bonn pressen - tekniske data

Keld Reimer Hansen <keld.reimer.hansen@step-g.com>

ma 06.03, 12:41

Hej Kjell

Nej, I kan ikke udregne max gods ud fra max-vægt pr meter.

Vi kan godt presse $\varnothing 500 \times 30$ mm mm rør som max størrelse, men vores sav kan ikke save det. Ny sav kræver en investering på 100.000 €.

$\varnothing 400 \times 30$ mm er største rør vi kan tilbyde og kun I 6060 T6, Lmax = 5000 mm

Alternativt kan vi producer halv skaller 500×60 og svejse sammen til et rør.

Flangen til den store I-bjælke vil komme til at veje ca 100 kg/m og da presbolten max kan veje 535 kg, kan der "kun" producer L=5000 mm.

Send gerne en tegning af flangen så skal vi kigge på optimeringer.

Mit freundlichen Grüßen / best regards

Keld Reimer Hansen

ST Extruded Products Germany GmbH

Devillestrasse 2 | 06749 Bitterfeld, Deutschland | www.step-g.com

T: | F: | M: +45 20643306 | keld.reimer.hansen@step-g.com

Sitz: Bergstraße 17, 88267 Vogt, Germany | Geschäftsführer: Michael Zint | Handelsregister Ulm | HRB 550822

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Kjell Petter Løkling Lunde

ma 06.03, 11:05

Hei,

Takk for svar. Finner man maksimal tykkelse på de oppgitte profilene ved å bruke vekt per. m (190kg/m for pressen i Bonn)?

For et $\varnothing 500$ mm rør, blir maksimal tykkelse da 50 mm? Og hva blir maksimal lengde på et slikt rør?

Vi ønsker å lage en I-profil slik som vist i vedlagt bilde. Der tanken er å ekstrudere flensen. Kan flensen ekstruderes med, bredde: 700 mm, tykkelse: 50 mm, og lengde: 12 000 mm (dimensjonen på "built-in guideway" er ikke helt bestemt enda)?

Med vennlig hilsen,
Kjell og Henrik

B.2 E-mail correspondence with Andreas Fellhauer regarding production capabilities at Constellium.

WG: Large Extrusion

Fellhauer, Andreas <andreas.fellhauer@constellium.com>

ma 27.02, 14:01

Kjell Petter Løklung Lunde;

martin@jnssweden.com;

Renner Juergen <juergen.renner@constellium.com>

Hi Kjell

I'm Andreas from the technical department.

Our leaflet is giving very general information. The thickness you mentioned below is the minimum thickness which is very often of interest for lightweight structures.

The diameter 600mm is valid for alloys 6060 and 6063 only, but more for profiles in rectangular shapes.

In the case of tubes we are generally limited to sizes of diameter 520 mm and looking at your requirement concerning the wall-thickness we have to speak as well about other restrictions:

- Choice of the alloy: in case of 6082 we should not exceed very much the weight per meter of 80 kg/m which is the case e.g. for 500x20 or 350x30
- Max delivery length of the tube/profile is limited according to the weight per meter and choice of alloy – e.g. tube 500x20 in 6082 is only up to 6m
- More lightweight profiles/tubes can be delivered in longer lengths

I hope this answers your question sufficiently. If you like you can send me further information so that I can support you on the design of profiles.

Mit freundlichen Grüßen / Kind regards

Andreas FELLHAUER, Dipl.-Ing.
Head of Technical Customer Service Industry
Large Profiles
Automotive Structures and Industry
Constellium Singen GmbH
Alusingen-Platz 1
78224 Singen, Germany
Office phone : +49 7731 80 3523
Mobile phone : +49 172 6397355
Fax : +49 7731 80 2436
Mail : andreas.fellhauer@constellium.com

Visit our website: www.constellium.com

Zertifiziert nach ISO 9001, ISO/TS 16949, ISO 14001, ISO 50001, BS OHSAS 18001

Sitz der Gesellschaft: Singen eingetragen in das Handelsregister des Amtsgerichts Freiburg im Breisgau, B 540034
USt.-ID-Nummer DE 811178046
Vorsitzender des Aufsichtsrates: Paul Warton

Geschäftsführung: Rolf Schencking (Vorsitzender), Hans-Joachim Chwalisz
Bank: Deutsche Bank, Singen

Von: Kjell Petter Løkling Lunde [<mailto:kplunde@stud.ntnu.no>]

Gesendet: Donnerstag, 23. Februar 2017 15:29

An: Renner Juergen

Cc: Henrik Westersjø Nesheim

Betreff: Large Extrusion

Hi,

We are working on a project, where we are modelling a large subsea structure in aluminium. We are trying to find out how large profiles (and what shape) it is possible to extrude. Do Constellium have a product catalog of large extruded profiles, or are you able to help us?

Regards,
Kjell

B.3 E-mail correspondence with Ole Øystein Knudsen regarding aluminium subsea properties.

Thermal Sprayed Aluminium i mud

Ole Øystein Knudsen <Ole.Knudsen@sintef.no>

ti 28.02, 13:56

Henrik Westersjø Nesheim

Innboks

Hei

I mud blir jo betingelsene ganske stagnant, siden muden ikke er flytende. Det rant friskt sjøvann over muden hele tiden så denne ikke skulle tørke inn. I sjøvann ble vannet skifta ut hele tida. Må det, ellers vil bakterier gjøre at vannet råtner, og da er ikke betingelsene sammenlignbare med naturlig sjøvann lenger.

TSA korroderer mer enn valsa og ekstrudert Al, trolig fordi det har en annen mikrostruktur og en geometri med mye sprekker og spalter. Kan se ut som det vil angripes av generell korrosjon til en viss grad. Vi finner korrosjonsprodukter over hele overflata på TSA etter eksponering. Anodiske polarisasjonskurver til TSA viser alltid høyere anodisk reaksjonshastighet enn valsa og ekstrudert materiale. Men TSA viser også et "pittingpotensiale" på omkring -700 mV, så helt ulikt er det jo ikke. Disse resultatene er ikke publisert enda.

-oøk

Henrik Westersjø Nesheim

ti 28.02, 10:33

Hei,

Takk for paper på korrosjon i sjøvann. Har to spørsmål angående paperet:

- Er sjøvannet stillestående i begge beholderene (både for sjøvannsprøvene og for prøvene som er nedsenket i mud) ?
- Og om du vet noe om korrosjonsegenskapene til TSA i forhold til AA-5083 og AA-6082 ?

Resultatene og konklusjonen i paperet er veldig nyttig i vår masteroppgave ettersom at det viser at TSA har akseptable korrosjonsegenskaper nedsenket i mud.

Hilsen,
Henrik og Kjell

Ole Øystein Knudsen <Ole.Knudsen@sintef.no>

on 22.02, 11:30

Hei

Vedlagt finner dere et paper på dette.

Hvis dere ikke har CP på TSA er det fare for MIC, som kan akselerere korrosjon vesentlig. Den undersøkelsen blir ikke publisert før i september.

Hilsen Ole Øystein

Henrik Westersjø Nesheim

on 22.02, 10:48

Hei,

Vi er to studenter som skriver oppgave ved PTS (petroleum) om bruken av aluminium i subsea strukturer. I den forbindelse har vi identifisert et mulig problem når en begraver aluminium i mud, vi er usikre på hva som vil skje med aluminiumet.

I dag hadde vi ett møte med Professor Roy Johnsen, hvor han nevnte at du hadde sett på hva som skjer med TSA (thermal sprayed aluminium) i mud under sjøvann.

Vi lurer derfor på om du har noe informasjon om emnet som du kan dele med oss ?

Hilsen,

Henrik og Kjell

RE: PTFE Coating subsea bolter

Ole Øystein Knudsen <Ole.Knudsen@sintef.no>

fr 28.04, 11:35

Kjell Petter Løkling Lunde

Innboks

Siden det er CP vil bolten være beskyttet. Det er kun i kontaktflata det er viktig med ekstra beskyttelse, for å hindre at sinken går i oppløsning og bolten mister forspenninga. Ser ut til at NORSOK godkjenner bare sinkfosfatering, uten ekstra lag med belegg. Med sinkfosfatering antar jeg at sinken er katodisk. Sinken kan jo strengt tatt korrodere selv med CP. Fosfatering vil passivere den, slik at den ikke korroderer.

-oøk

From: Kjell Petter Løkling Lunde [mailto:kplunde@stud.ntnu.no]

Sent: 28. april 2017 09:12

To: Ole Øystein Knudsen <Ole.Knudsen@sintef.no>

Subject: SV: PTFE Coating subsea bolter

Hei igjen,

Er sinkfosfatering alene tilstrekkelig (med CP) for å beskytte boltene? Eller må det påføres et ekstra lag med coating?

Med vennlig hilsen,

Kjell

Fra: Ole Øystein Knudsen <Ole.Knudsen@sintef.no>

Sendt: 26. april 2017 10:46:30

Til: Kjell Petter Løkling Lunde; Henrik Westersjø Nesheim

Emne: RE: PTFE Coating subsea bolter

Hei

Jeg tror man kjøper ferdig PTFE belagte bolter. Belegget er svært tynt, ~20 µm bare, så det er sannsynlig at det skades i monteringa og elektrisk kontakt oppnås på den måten. Eller at de skraper det av på et område. Det står at fosfatering av sinkbelegget også er et alternativ, hvilket ikke introduserer dette problemet. Tror jeg ville gått for fosfatering

-oøk

From: Ole Øystein Knudsen [mailto:ole.oystein.knudsen@ntnu.no]

Sent: 26. april 2017 10:28

To: Ole Øystein Knudsen <Ole.Knudsen@sintef.no>

Subject: Vs: PTFE Coating subsea bolter

Fra: Kjell Petter Løkling Lunde

Sendt: 26. april 2017 10:28:04 (UTC+01.00) Amsterdam, Berlin, Bern, Roma, Stockholm, Wien

Til: Ole Øystein Knudsen

Kopi: Henrik Westersjø Nesheim

Subjekt: PTFE Coating subsea bolter

Hei,

Jobber med en master angående bruk av aluminium subsea (ITS-Integrated template structure), med Roy Johnsen som veileder. Noen av de strukturelle bjelkene skal sammenføres ved bolting, både alu-alu plater mot hverandre og alu-stål. Karbonstål bolter skal brukes (NORSOK).

Boltene skal coates i henhold til følgene (NORSOK):

"Carbon steel and/or low alloy bolting material shall be hot dip galvanised to ASTM A153 or have similar corrosion protection. For submerged applications, where dissolution of a thick zinc layer may cause loss of bolt pretension, phosphating shall be used. For sub-sea installations the use of poly-tetra-fluoro-ethylene (PTFE) based coatings can be used provided electrical continuity is verified by measurements. ."

PTFE er som kjent et ikke ledende materialet. Hvordan kan man da sørge for elektrisk kontinuitet til bolten (fra CP på strukturen)? Er det meningen at PTFE skal påføres etter at bolten er montert, slik at det er en ubehandlet kontaktflate mellom bolt og plate for å skape elektrisk kontinuitet?

Mvh,

Kjell og Henrik

B.4 E-mail correspondence with Tor Berge Gjersvik.

Tor Berge Gjersvik <tor.b.gjersvik@ntnu.no>

From: Harald Thomander Neerland [mailto:haraldt.neerland@technipfmc.com]

Sent: fredag 17. februar 2017 10.08

To: Tor Berge Gjersvik <tor.b.gjersvik@ntnu.no>

Subject: RE: Gjøa ITS

Hei Tor,

Gjøa er designet for å være Snag Free» når manifold og sealine protection er installert.

Designet er model testet (Tror faktisk det var i Trondheim).

Slik bildet viser er ikke Gjøa «Snag free», men temporary roof og sealine protection ble (så vidt jeg vet) instalert i samme kampanje.

Harald

PS. Uttrykket «Snag free» er i hovedsak relatert til snag av trålbord. For oss er faktisk friksjon fra trål et viktigere design kriteria; dvs. friksjon når trål blir dratt over strukturen.

From: Tor Berge Gjersvik [mailto:tor.b.gjersvik@ntnu.no]

Sent: 16. februar 2017 13:29

To: HaraldT.Neerland@technipfmc.com

Subject: Gjøa ITS

Harald,

Jeg har noen studenter som sitter å ser på noen fine bilder og figurere de har fått fra Subsea 7 og fra installasjon av template på Gjøa. Se her:

Skandi Acergy has completed her 2008 installations



StatoilHydro

Skandi Acergy Integrated Template Structures installed



StatoilHydro

Dette er jo slik det gikk i sjøen uten luker og sea-line protection.

Så til spørsmålet: Kan du si eller finne svar på om denne typen er designet for å være «snag free»?

Hilsen,

Tor Berge

Tor Berge S. Gjersvik, siv.ing, dr.ing

Professor

Norwegian University of Science and Technology - NTNU

Faculty of Engineering Science and Technology,

Department of Geoscience and Petroleum

Spørsmål og svar

Tor Berge Gjersvik <tor.b.gjersvik@ntnu.no>

fr 17.03, 13:26

Henrik Westersjø Nesheim;

Kjell Petter Løkling Lunde;

Roy Johnsen;

Patrick Reurink

Innboks

Hei,

Har hatt kontakt med Harald T. Neerland i TechnipFMC nå på spørsmål som kom opp i dagens møte:

- Hva er de typiske stålqualitetene nyttet i struktur (ramme + sugeankre) og beskyttelsesstruktur for en ITS?
Topp ramme: S355 G15+N Norsok Y 2t
Skørt: S355 J2+N Norsok Y05
- I trålbekyttelsen benyttes et rammeverk av «rør». Er disse påført noen form for coating innvendig?
Nei. Man antar at O2 nivået blir utarmet inne i lukkede rom. For å unngå kolaps fra ytre krefter blir alle hullrom punktert, og diskusjonen har gått på hvorvidt vår filosofi /antagelse om utarming av O2 er riktig, eller om vi får utskifting av innestengt volum. For å forhindre dette krever Statoil (og enkelte andre) at vi installerer «Splitt Rubber Grommets) i de hull som punkterer strukturen (personlig ikke sikker på om de er nødvendig. Jeg vet vi har mistet noen under installasjon uten at man har ropt ulv).

Tor Berge

Tor Berge S. Gjersvik, siv.ing, dr.ing

Professor

Norwegian University of Science and Technology - NTNU

Faculty of Engineering Science and Technology,
Department of Geoscience and Petroleum

B.5 E-mail correspondence with Asbjørn Wathne in Subsea 7 regarding ITS installation details.

Gjøa ITS installasjon

Asbjørn Wathne <asbjorn.wathne@subsea7.com>

ti 28.03, 15:18

Heisann

Bare hyggelig å være til hjelp. Dere finner svarene mine rødt nedenfor. Det er en herlig blanding av engelsk og norsk, men dere er jo smarte studenter, så det finner dere nok ut av. ☺ Lykke til!

Regards,

Asbjørn Wathne
Project Engineering Manager

eMail Asbjorn.Wathne@subsea7.com

Tel +47 51725184

Mobile +47 91685582

Switchboard +47 51725000

Website www.subsea7.com

From: Henrik Westersjø Nesheim [mailto:henrikwn@stud.ntnu.no]

Sent: 27. mars 2017 15:00

To: Asbjørn Wathne

Cc: Kjell Petter Løkling Lunde

Subject: Gjøa ITS installasjon

Hei,

Takk for hyggelig telefon tidligere i dag, setter stor pris på at du tar deg tid til å svare på henvendelse.

Nedenfor ser du en oversikt over hva vil lurer på:

- Er vekt på wire med i beregning av vekt/løftekapasitet, slik at en må ta hensyn til vekt av wire på dypere vann ? **Normalt ikke nødvendig på grunt vann, men for dypt vann, si 500m og dypere, er det viktig å ta med vekten av kranwirer. Dette kan ofte bli en showstopper på veldig dypt vann. Noen fartøy har egen fibertau winch for bruk på dypt vann. Strukturen løftes da ut med vanlig krane og overføres til fibertau winchen for videre nedsenking til sjøbunn. Husk og også legge til vekt av løfterigging på selve ITSen. Denne kan utgjøre 10-15 Te dersom det er stålslings, en del mindre om dere bruker fiberslings.**

- Tidsbruk under installasjon, hvor lenge leies ett vessel for å installere en ITS, med utgangspunkt i Gjøa ITS'er og gjerne noe om variasjon i installasjonstider? Her er det store forskjeller fra ITS til ITS. Dette avhenger av flere forskjellige årsaker, som f.eks størrelse på sugeankerene, sjøbunnsforhold og tiden det tar å suge dem ned (6-18 timer). Her ville jeg sagt at selve ITS installasjonsjobben tar ca 24 timer (inkludert 4 stk XT luker) fra ankomst til feltet, til fartøyet kan returnere til land. Legg til 6-12 timer i transit hver vei, avhengig av hvor feltet ligger og hvor du mobiliserer fartøyet. Dere kan også regne 24 timer i mobilisering av fartøyet (løfte om bord utstyr, ITS, luker, sjøsikring, risikomøter, crew familiarisation etc.). Så grovt sett kan en regne ca 3 dager på en ITS installasjon. Jeg har ikke inkludert for sealine protection cover. Dere kan regne 2-3 timer per cover. Normalt er det kun en SPS på en ITS som står alene. Venting på vær er ikke inkludert.
- Logg over installasjonsforløpet dersom det eksisterer. Har lagt til en røff breakdown av hvordan vi installerer en typisk FMC ITS nedenfor:

The bullets below explains the main steps for the deployment of the ITS:

Set up on DP

Preparation for launching of the ITS

Survey of the installation area

Connect crane to ITS rigging at deck level.

Lift ITS clear of deck

Slew the load to outboard until the ITS is clear of the vessel.

Lower through splash zone to a water depth of 50m to allow for flooding of tubular members and disconnection of tugger wires.

Continue lower the ITS to approx 10m above seabed

The bullets below explains the main steps for the positioning of the ITS:

Adjust the position of the vessel and rotate the ITS with the ROV as required until the ITS is within the allowable tolerances

Land the ITS in target position

Gradually reduce tension and complete self-weight penetration

Pay out slack on the main crane

ROV to inspect the actual penetration and verify correct positioning and heading from transponder/gyro

Disconnect rigging and recover crane and rigging to deck

The bullets below explains the main steps for the levelling of the ITS:

Clean and lock the vent hatches on the suction cans

Level and suck down ITS by use of suction pump and by operating the valves on the levelling panel

Confirm correct inclination and depth by inclinometers, bullseye and digiquartz

Close valves on the levelling panel and undock the ROV

The bullets below explains the main steps for the XT hatches installation:

Offset vessel to a safe distance from existing subsea structures/lines.

Connect crane to the XT hatch rigging

Upend the XT hatch on deck and overboard

Lower to approx 20m above seabed

Relocate vessel to the ITS and position hatch over the dedicated position.

Lower down XT hatch and land onto the ITS until the hinges engage

Continue to pay out on crane wire and assist with ROV to further lower the hatch to closed position

Lock XT hatch in closed position.

Repeat for remaining 3 off XT hatches

Perform as-left survey of complete ITS

- Lurer også på hva som er utfordringer/begrensningene under installasjon ? (Eksempelvis waiting on weather). Den største utfordringen er selve avløftet fra dekk. Altså løft i luft. Større fartøy greier normalt disse løftene i 2-3m Hs. Farene er svingende last under avløft, samt økte dynamiske laster i rigging/kranwire. Selve deployment har også noen begrensninger, men normal mindre kritisk enn løft i luft. Her er det krefter i plaskesonen som er utfordringen. Har ITS f.eks. store sugeanker, så kan dette gi oppdrift (om ITS senkes raskere enn luften i sugeankerne kan evakueres) som igjen kan gi slakke slings og rykkaster. Lokalt, så kan f.eks. luker på ITS ødelegges/rives av pga. slamming/drag krefter. Waiting on weather er en direkte konsekvens av å ikke ha rette værforhold når ITS'en skal løftes. Mye tid blir brukt på deployment analyser under forberedelsene til jobben. Vi ser da på statistiske værdata, som vind, bølgehøyde, bølgeperiode og strøm, og ser på hvilken innvirkning de har på fartøybevegelse (heave, roll, pitch etc.) og konkluderer så hvilke kombinasjoner/scenarioer vi kan operere i.
- Hvor tidkrevende (eventuelt ekstra kostnader) ved å løfte på plass beskyttelsesstruktur (hatches og sealine protection). Vil det være muligheter for å redusere antall løft ved å inkludere hatches og sealine protection i samme løft som ITS'en når ITS'en veier ca. 200 ton. Sealine protection covers må normalt installeres etter ITS er landet. På FMC sine ITSer må de 4 XMT lukene også installeres separat, da disse har for mye drag til å installeres med ITS'en (de risikerer å rives av i plaskesonen). Men vi har sett ITSer med luker som har større perforering og således ikke er noe problem å inkludere i ITS løftet, gitt at totalvekten er håndterbar.
- Dagrater på relevant vessel (Ett vessel i 400Te klassen og ett i 250Te klassen). Målet med oppgaven vår er å redusere installasjonsvekten til en ITS og således bruke mindre installasjonsfartøy. Dette er konfidensiell informasjon, og variere veldig med markedsituasjonen. Her vil jeg anbefale å ta kontakt med f.eks. Seabrokers, da disse nok sitter på gjennomsnittspriser i markedet. Men husk at i tillegg til en kraftig kran, så må også kranen ha løfteradius nok til å plukke opp ITS'en fra dekk. Fartøyet må også ha nok dekksplass til ITS'en.
- Vet du noe om hva hvert enkelt løft veide (tenker da spesielt på løftene for hatches og sealine protection) ? Fra Gjøa prosjektet så var vektene omtrent som følger:

Gjøa ITS	275 Te (+ 10 Te for vekt av løfterigging)
XT luker	3.5 Te per luke (installert separat)
Sealine Protection Cover	15 Te (installert separat)

Om dere ikke allerede er klar over det, så var det en gruppe studenter som gjorde modelltester av Gjøa ITS for oss på Marintek i 2008. Dere kan sikkert finne noe god info der også.

Vi lurer også på om noe av informasjonen du sender oss er konfidensielt ?, i så fall må vi referere til dette som konfidensielt, og unnlate det fra oppgaven ettersom at vår oppgave ikke er konfidensiell.

Informasjonen jeg har gitt dere her er ikke konfidensiell. Men vi setter alltid pris på at dere refererer til oss når dere bruke informasjonen. 😊

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Henrik Westersjø Nesheim

ma 27.03, 14:59

Asbjorn.Wathne@subsea7.com <asbjorn.wathne@subsea7.com>;

Kjell Petter Løkling Lunde

Hei,

Takk for hyggelig telefon tidligere i dag, setter stor pris på at du tar deg tid til å svare på henvendelse.

Nedenfor ser du en oversikt over hva vil lurer på:

- Er vekt på wire med i beregning av vekt/løftekapasitet, slik at en må ta hensyn til vekt av wire på dypere vann ?
- Tidsbruk under installasjon, hvor lenge leies ett vessel for å installere en ITS, med utgangspunkt i Gjøa ITS'er og gjerne noe om variasjon i installasjonstider?
- Logg over installasjonsforløpet dersom det eksisterer.
- Lurer også på hva som er utfordringer/begrensningene under installasjon ? (Eksempelvis waiting on weather).
- Hvor tidkrevende (eventuelt ekstra kostnader) ved å løfte på plass beskyttelsesstruktur (hatches og sealine protection). Vil det være muligheter for å redusere antall løft ved å inkludere hatches og sealine protection i samme løft som ITS'en når ITS'en veier ca. 200 ton.
- Dagrater på relevant vessel (Ett vessel i 400Te klassen og ett i 250Te klassen). Målet med oppgaven vår er å redusere installasjonsvekten til en ITS og således bruke mindre installasjonsfartøy.

- Vet du noe om hva hvert enkelt løft veide (tenker da spesielt på løftene for hatches og sealine protection) ?

Vi lurte også på om noe av informasjonen du sender oss er konfidensielt ?, i så fall må vi referere til dette som konfidensielt, og unnlate det fra oppgaven ettersom at vår oppgave ikke er konfidensiell.

Vi kommer begge fra Stavanger-området og besøker dere gjerne på Forus dersom det passer. Vi er hjemme i påsken fra og med Lørdag den 06. April til og med Tirsdag den 18. April.

Hilsen,
Henrik og Kjell

B.6 E-mail correspondence with Ole Terje Midling in Marine Aluminium.

RE: Prisoverslag - Fabrikasjon ITS

Ole Terje Midling <ole.terje.midling@m-a.no>

Svar alle ti
16.05, 16:12

Hei,

Som sagt er det en utfordring å prise denne jobben da vi ikke vet omfanget av sveisingen (annet en antall meter sveis, og MIG eller FSW prosessen som er valgt) Spekteret for fabrikasjon vil være i størrelsesorden 300-400NOK/kg avhengig av kompleksitet^{*)}.

Mvh
Ole Terje

Ole Terje Midling
Chief Fabrication Officer PhD , Marine Aluminium AS
Mobile+4793208952
[Websit](#) [Email](#)

From: Kjell Petter Løkling Lunde [mailto:kplunde@stud.ntnu.no]
Sent: 16. mai 2017 13:04
To: Ole Terje Midling <ole.terje.midling@m-a.no>; Henrik Westersjø Nesheim <henrikwn@stud.ntnu.no>
Subject: SV: Prisoverslag - Fabrikasjon ITS

Hei Ole,

Har dere en ca. fabrikasjonstid (timer/tonn) og fabrikasjonskostnad (kr/time) eller andre veiledende tall vi kan bruke? Det kan gjerne være i form av et spekter.

Vi har fått prisoverslag fra både STEP-G og Constellium. Ekstrusjon hos STEP-G og plater hos Constellium.

Med vennlig hilsen,
Kjell og Henrik

^{*)} Etter diskusjon per telefon den 05.31.2017, er det avtalt å bruke spekteret 200 – 300 NOK/kg som et realistisk estimat i kostnadsanalysen.

B.7 E-mail correspondence with Skarpenord regarding aluminium anodes

Anode kostnader

Jørn Voje <joern.voje@skarpenord-corrosion.no>

to 27.04, 13:27

Hei,

Som et slags budsjettall foreslår jeg at dere benytter NOK 44.00 pr. kg netto legering.

Vennlig hilsen,
for Skarpenord Corrosion a.s.
Jørn Voje

*Jørn Voje
Skarpenord Corrosion a.s.
P.O. Box 46
N-3993 Langesund
Norway
Telephone: +47 35967941
Handphone: +47 91614902
Email: jv@scas.no
Web: <http://www.skarpenord-corrosion.no>*

Henrik Westersjø Nesheim

to 27.04, 12:50

post@skarpenord-corrosion.no;
Kjell Petter Løkling Lund

Hei,

Vi er to studenter ved NTNU som skriver en masteroppgave innen subsea. Vi tar for oss en stor struktur som trenger anode-beskyttelse mot korrosjon.

Har du/dere en enhetskostnad (NOK/Ton eller NOK/kg) på anoder av typen "Coral 'A' High Grade" og typen "Coral 'A' Special Grade" ?

Strukturen vi ser på vil grovt regnet trenge ca. 7,5 ton med anoder.

Hilsen,
Henrik og Kjell

B.8 E-mail correspondence with Halvard Torget Eriksen regarding cost of phosphated bolts

RE: Sinkfosfatering bolter - prisoverslag

Halvard Torget Eriksen <halvard@oddacoating.no>

fr 28.04.2017 08:59

Til:

Kjell Petter Løkling Lunde

Kopi:

Bjarne Sørum <bjarne@oddacoating.no>

Innboks

Hei og god morgen!

Et kjapt prisoverslag på dette:

Vask, blåserensing, zink-fosfatering:

- Bolter	Kr. 145.-
- Mutre	Kr. 60.-
- Skiver	Kr. 50.-

Prisene forutsetter antallet dere antydte i forespørselen under.

Vi vil anbefale at delene får påført et tynt lag olje som er egnet for formålet rett etter fosfatering. Om dere ønsker dette vil det medføre en kostnad på Kr. 5.- per del.

Bare ta kontakt om dere ønsker utfyllende opplysninger eller om det er andre prosjekter dere ønsker hjelp til. 😊

Beste hilsen / Best Regards

Halvard Torget Eriksen

Project Coordinator

☎ +47 906 44 100 📞 +47 992 06 228

www.oddacoating.no

From: Kjell Petter Løkling Lunde [mailto:kplunde@stud.ntnu.no]

Sent: 27. april 2017 14:08

To: Odda Coating Technology AS <post@oddacoating.no>

Cc: Henrik Westersjø Nesheim <henrikwn@stud.ntnu.no>

Subject: Sinkfosfatering bolter - prisoverslag

Hei,

Vi jobber med en master angående bruk av aluminium subsea ved NTNU. Noen av de strukturelle bjelkene vi jobber med må sammenføres med bolter.

Har dere mulighet til å gi oss et prisoverslag på å sinkfosfatere 1468 stk karbonstålbolter i tillegg til tilhørende muttere (1468 stk) og skiver ($1468 \cdot 2 = 2936$ stk)? Dimensjonen på boltene er M36 med varierende lengder mellom 110-210 mm.

Har dere en budsjetterende enhetspris pr. komponent?

Med vennlig hilsen,
Kjell og Henrik

Appendix C

Bolt Calculations

C.1 Design sheets exported from excel

Joint V, a-a	Accounts for both web and flange		Moment 6/8 of maximum		Height of beam [mm]
	Safety factor for ultimate state	Number of friction interfaces	Slip factor (0,3-0,4?pakning?)	UTS [Mpa]	
	1.25	2	0.4	700	9,976,568,229
Bolt Type	As [mm^2]	Preloading (F_p) [N]	Slip resistance per bolt [N]	Shear resistance per bolt [N]	Number of flange bolts required (one flange, one side)
M20	333	163,170	104,429	93,240	67
M24	384	188,160	120,422	107,520	58
M30	621	304,290	194,746	173,880	36
M36	865	423,850	271,264	242,200	26
M39	1030	504,700	323,008	288,400	22
Bolt size [mm]	do [mm] +3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 420 mm space to bolt one side)
	20	23	28	51	56
	24	27	33	60	65
	30	33	40	73	80
	36	39	47	86	94
	39	42	51	93	101
Flange fixturing characteristics	Flange thickness [mm]	Flange force [N]	Web thickness [mm]	Web force [N]	Web force [N]
Width of top flange plate [mm]	500	55	6,904,199	60	1,625,000
Thickness of top flange plate [mm]	40				
Width of bottom flange plate (one side of web) [mm]	210	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint
Thickness of bottom flange plate [mm]	40	268	16	64	332
Axial stress in top plate (one friction interface) [Mpa]	345	232	14	56	288
Axial stress in both plate (two friction interfaces) [Mpa]	188	144	9	36	180
Length through cross-section (one friction interface) [mm]	95	104	6	12	116
Length through cross-section (two friction interfaces) [mm]	135	88	6	24	112
Web fixturing characteristics	Number of bolts along the flange	Min. Length of coupling (one side) [mm]	Min. Length of flange plate (both sides) [mm]		
Width of web plate [mm]	150	10,00	1030		
Height of middle plate [mm]	600	10,00	606	1212	
Height of top or bottom plate [mm]	0	8,00	591	1182	
Thickness of web plate [mm]	10	7,00	610	1220	
Shear stress in web plate [Mpa]	67.70833333	6,00	567	1134	
Length through cross-section [mm]	80				
Cubic meter of flange plating [mm^3]	89792000				
Cubic meter of web plating [mm^3]	3600000				
Total Plate weight [kg] 4 connections	1008.6336				
Total number of flange bolts	416	cross section length [mm]			
Total number of web bolts	48	135			
		80			

Joint Ia	Accounts for both web and flange				moment 3/8 of maximum Moment [N*mm]	Height of beam [mm]	1500
	Safety factor for ultimate state	Number of friction interfaces	Slip factor (0,3-0,4-palming?)	UTS [Mpa]			
1.25	1	0.4	700	4,988,284,115			
Bolt Type							
M20	333	163,170	52,214	93,240	Number of flange bolts required (one flange, one side)	67	
M24	384	188,160	60,211	107,520		58	
M30	621	304,290	97,373	173,880		36	
M36	865	423,850	135,632	242,200		26	
M39	1030	504,700	161,504	288,400		22	
Bolt size [mm]							
20	23	28	51	56	Number of bolts across the flange (ca. 420 mm space to bolt one side)	7.00	
24	27	33	60	65		6.00	
30	33	40	73	80		5.00	
36	39	47	86	94		4.00	
39	42	51	93	101		4.00	
Flange fixturing characteristics							
Width of top flange plate [mm] Steel plate	500	Flange thickness [mm]	Flange force [N]	Web thickness [mm]	Web force [N]		
Thickness of top flange plate [mm]	25	55	3,452,100	60	1,625,000		
Axial stress in top plate (one friction interface) [Mpa]							
Length through cross-section (one friction interface) [mm]	276	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint		
	80	134	116	32	166		
Web fixturing characteristics							
Width of web plate [mm]	200	52	44	12	12	64	
Height of middle plate [mm]	600	44	44	11	11	55	
Height of top or bottom plate [mm]	0	Number of bolts along the flange	Min. Length of coupling (one side) [mm]				
Thickness of web plate [mm]	10	10.00	515				
Shear stress in web plate [Mpa]	270.8333333	10.00	606				
Length through cross-section [mm]	70	8.00	591				
		7.00	610				
		6.00	567				
Cubic meter of flange plating [mm^3] Steel							
	15250000						
Cubic meter of web plating [mm^3] Steel							
	1200000						
Total steel plate weight [kg, 4* connections							
	513.24	cross section length [mm]					
Total number of flange bolts							
	208	80					
Total number of web bolts							
	48	70					

Joint I, d	Accounts for both web and flange (flange only one interface due to steel plating)				Moment (1/8 of maximum)	
	Safety factor for ultimate state	Number of friction interfaces	Slip factor (0.3-0.4?pakring?)	UTS [Mpa]	Moment [N*mm]	Height of beam [mm]
	1.25	1	0.4		700	1,662,761,372
Bolt Type	As [mm^2]	Preloading (F_p) [N]	Slip resistance per bolt [N]	Shear resistance per bolt [N]	Number of flange bolts required (one flange, one side)	
M20	333	163,170		93,240		22.00
M24	384	188,160		107,520		19.00
M30	621	304,290		173,880		12.00
M36	865	423,850		242,200		9.00
M39	1030	504,700		288,400		8.00
Bolt size [mm]	do [mm] +3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 250 mm space to bolt one side)	
	20	28	33	51	56	4.00
	24	33	40	60	65	3.00
	30	40	47	73	80	3.00
	36	47	51	86	94	2.00
	39	51		93	101	2.00
Flange fixturing characteristics	Flange thickness [mm]	Flange force [N]	Web thickness [mm]	Web force [N]	Web force [N]	
	500	30	1,131,130	45		100,000
Thickness of top flange plate [mm]	Steel plate					
	10					
Axial stress in top plate (one friction interface) [Mpa]						
	226	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint	
	40	44.00	2.00	2.00	2.00	46.00
		38.00	2.00	2.00	2.00	40.00
		24.00	2.00	2.00	2.00	26.00
Web fixturing characteristics						
	150	18.00	2.00	2.00	2.00	20.00
Width of web plate [mm]		16.00	1.00	1.00	1.00	17.00
Height of top or bottom plate [mm]	0	Number of bolts along the flange	Min. Length of coupling (one side) [mm]			
	10	6.00	311			
Thickness of web plate [mm]		7.00	426			
Shear stress in web plate [Mpa]	55	4.00	299			
Length through cross-section [mm]		5.00	438			
		4.00	381			
Cubic meter of flange plating [mm^3] Steel	4380000					
Cubic meter of web plating [mm^3] Steel	300000					
Total steel plate weight [kg]	146.016	cross section length [mm]				
		72	40			
Total number of flange bolts	8	55				
Total number of web bolts						

Joint 1, f		Accounts for both web and flange		Moment 3/8 of max	
Safety factor for ultimate state		Number of friction interfaces		Moment [N*mm]	
1.25		1		700	
Slip factor (0,3-0,4?packing?)		0.4		4,988,284,115	
UTS [Mpa]		0.4		1500	
Bolt Type	As [mmx2]	Preloading (F_p) [N]	Slip resistance per bolt [N]	Shear resistance per bolt [N]	Number of flange bolts required (one flange, one side)
M20	333	163,170	52,214	93,240	66
M24	384	188,160	60,211	107,520	57
M30	621	304,290	97,373	173,880	36
M36	865	423,850	135,632	242,200	26
M39	1030	504,700	161,504	288,400	22
Bolt size [mm]	do [mm] *3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 760 mm space to bolt one side)
20	23	28	51	56	13.00
24	27	33	60	65	11.00
30	33	40	73	80	9.00
36	39	47	86	94	8.00
39	42	51	93	101	7.00
Flange fixturing characteristics	Flange thickness [mm]	Flange force [N]	Flange force [N]	Web thickness [mm]	Web force [N]
Width of top flange plate [mm]	880	45	3,428,374	40	2,000,000
Thickness of top flange plate [mm]	25				
Axial stress in top plate (one friction interface) [Mpa]	156	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint
Length through cross-section (one friction interface) [mm]	70	132	39	78	210
		114	34	68	182
		72	21	42	114
Web fixturing characteristics		52	15	15	67
Width of web plate [mm]	200	44	13	26	70
Height of middle plate [mm]	600				
Height of top or bottom plate [mm]	0	Number of bolts along the flange	Min. Length of coupling (one side) [mm]		
Thickness of web plate [mm]	20	6.00	311		
Shear stress in web plate [Mpa]	167	6.00	366		
Length through cross-section [mm]	60	4.00	299		
		4.00	352		
		4.00	381		
Cubic meter of flange plating [mm^3] Steel	15488000				
Cubic meter of web plating [mm^3] Steel	24000000				
Total steel plate weight [kg]	558.1056	cross section length [mm]			
Total number of flange bolts	208	70			
Total number of web bolts	60	60			

Joint IV, b-8		Accounts for both web and flange		Safety factor for ultimate state		Number of friction interfaces		Slip factor (0.3-0.4?packing?)		UTS [Mpa]		Moment [N*mm]		Height of beam [mm]	
		1.25	2	0.3	700	380,000,000	1000								
Bolt size [mm]		As [mm^2]		Preloading (F_p) [N]		Slip resistance per bolt [N]		Shear resistance per bolt [N]		Number of bolts across the flange (ca. 2/5 mm space to bolt one side, beam h)					
M20	20	333	163,170	78,322	93,240	9									
M24	24	384	188,160	90,317	107,520	8									
M30	30	621	304,290	146,059	173,880	5									
M36	36	865	423,850	203,448	242,200	4									
M39	39	1030	504,700	242,256	288,400	3									
Bolt size [mm]		do [mm] *3		Min. e1 and e2 [mm]		Min. p1 [mm]		Min. p2 [mm]		Number of bolts across the flange (ca. 2/5 mm space to bolt one side, beam h)					
20		23		28		51		56		5.00					
24		27		33		60		65		4.00					
30		33		40		73		80		3.00					
36		39		47		86		94		2.00					
39		42		51		93		101		2.00					
Flange fixturing characteristics		Flange thickness [mm]		Flange force [N]		Web thickness [mm]		Web force [N]							
Width of top flange plate [mm]		330		30		691,753		35		100,000					
Thickness of top flange plate [mm]		10													
Width of bottom flange plate (one side of web) [mm]		132.5		Total number of flange bolts		Number of web bolts (one web)		Total number of web bolts		Total number of bolts in the joint					
Thickness of bottom flange plate [mm]		10		36		2		4		40					
Axial stress in top plate (one friction interface) [Mpa]		210		32		2		2		36					
Axial stress in both plate (two friction interfaces) [Mpa]		116		20		1		2		22					
Length through cross-section (one friction interface) [mm]		40		16		0		0		16					
Length through cross-section (two friction interfaces) [mm]		50		12		1		2		14					
Cubic meter of flange plating [mm^3]		4284000		Number of bolts along the flange		Min. length of coupling (one side) [mm]		Min. length of flange plate (both sides) [mm]							
Cubic meter of web plating [mm^3]		0		2.00		107		214							
Total weight of plating [kg] 4 connections		46.2672		2.00		126		252							
				2.00		153		306							
				2.00		180		360							
				2.00		195		390							
Total number of flange bolts		64		cross section length [mm]											
Total number of web bolts		0		50		0									

Joint V, b-c Safety factor for ultimate state	Accounts for both web and flange		Slip factor (0.3-0.4?packing?)		UTS [Mpa]	tie in loads		Height of beam [mm]	1500
	1.25	Number of friction interfaces	Slip factor	Number of friction interfaces		Moment [N*mm]	518,000,000		
Bolt Type	As [mm²]	Preloading [F_p] [N]	Min. p1 [mm]	Min. p2 [mm]	Shear resistance per bolt [N]	Number of flange bolts required (one flange, one side)			
M20	333	163,170	28	51	52,214	93,240	13		
M24	384	188,160	33	60	60,211	107,520	11		
M30	621	304,290	40	73	97,373	173,880	7		
M36	865	423,850	47	86	135,652	242,200	5		
M39	1030	504,700	51	93	161,504	288,400	5		
Bolt size [mm]	do [mm] +3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 275 mm space to bolt one side, beam h)				
20	23	28	28	51	56	5.00			
24	27	33	33	60	65	4.00			
30	33	40	40	73	80	3.00			
36	39	47	47	86	94	2.00			
39	42	51	51	93	101	2.00			
Flange fixturing characteristics	Flange thickness [mm]	Flange force [N]	Web thickness [mm]		Web force [N]				
Width of top flange plate [mm]	0	652,381	35		100,000				
Thickness of top flange plate [mm]	0								
Width of bottom flange plate (one side of web) [mm]	275	Total number of flange bolts	Number of web bolts (one web)		Total number of web bolts		Total number of bolts in the joint		
Thickness of bottom flange plate [mm]	15	26	2		4		30		
Axial stress in top plate (one friction interface) [Mpa]	0	22	2		4		26		
Axial stress in both plate (two friction interfaces) [Mpa]	79	14	2		4		18		
Length through cross-section (one friction interface) [mm]	30	10	0		0		10		
Length through cross-section (two friction interfaces) [mm]	45	10	1		2		12		
Cubic meter of flange plating [mm³]	4389000	Number of bolts along the flange	Min. Length of coupling (one side) [mm]		Min. length of flange plate (both sides) [mm]				
Cubic meter of web plating [mm³]	0	3.00	158		316				
Total weight of plating [kg] 4 connections	47,4012	3.00	186		372				
		3.00	226		452				
		3.00	266		532				
		3.00	288		576				
Total number of flange bolts	48	cross section length [mm]							
Total number of web bolts	0	0							

Accounts for both web and flange													
Joint J_e		Safety factor for ultimate state		Number of friction interfaces		Slip factor (0.3-0.47pakning?)		UTS [Mpa]		Moment [N*mm]		Height of beam [mm]	
		1.25		1		0.3		700		0		0	
Boit Type	As [mm*2]	Preloading (F_p) [N]	Slip resistance per bolt [N]	Slip resistance per bolt [N]	Shear resistance per bolt [N]	Number of flange bolts required (one flange, one side)							
M20	333	163,170	188,160	188,160	93,240	6							
M24	384	188,160	304,290	304,290	107,520	5							
M30	621	304,290	423,850	423,850	173,880	3							
M36	865	423,850	504,700	504,700	242,200	2							
M39	1030	504,700			288,400	2							
Boit size [mm]	do [mm] +3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 260 mm space to bolt one side, beam h)								
	20	23	28	51	56	5.00							
	24	27	33	60	65	4.00							
	30	33	40	73	80	3.00							
	36	39	47	86	94	2.00							
	39	42	51	93	101	2.00							
Flange fixturing characteristics	Flange thickness [mm]	Flange force [N]	Web thickness [mm]									Web force [N]	
	300	32	200,000									20	
Width of top flange plate [mm] Steel plate	5												
Thickness of top flange plate [mm]	133												
Axial stress in top plate (one friction interface) [Mpa]	133												
Length through cross-section (one friction interface) [mm]	37												
	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts		Total number of bolts in the joint								
	37	12	0	0	0								
		10	0	0	0								
		6	0	0	0								
		4	0	0	0								
Cubic meter of flange plating [mm*3]	282000												
Cubic meter of web plating [mm*3]	0												
Total weight of plating [kg] 4 connections, steel	8.7984												
	Number of bolts along the flange	Min. Length of coupling (one side) [mm]											
	2,00	107											
	2,00	126											
	1,00	80											
	1,00	94											
	1,00	102											
Total number of flange bolts	cross section length [mm]												
Total number of web bolts	16												
	37												
	0												

Joint XI, g-k	Number of friction interfaces	Slip factor (0,3-0,4?pakning?)	UTS [Mpa]	Maximum moment from geometry of beam k	Height of beam [mm]
Safety factor for ultimate state	1.25	0.4		700	460,000,000
					1000
Bolt Type	As [mm ²]	Preloading (F _p) [N]	Slip resistance per bolt [N]	Shear resistance per bolt [N]	Number of flange bolts required (one flange, one side)
M20	333	163,170		52,214	93,240
M24	384	188,160		60,211	107,520
M30	621	304,290		97,373	173,880
M36	865	423,850		135,632	242,200
M39	1030	504,700		161,504	288,400
Bolt size [mm]	do [mm] *3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 180 mm space to bolt one side)
	20	28	28	51	56
	24	27	33	60	65
	30	33	40	73	80
	36	39	47	86	94
	39	42	51	93	101
Flange fixturing characteristics	Flange thickness [mm]	Flange force [N]	Web thickness [mm]	Web force [N]	
Width of top flange plate [mm]	500	40	460,000	20	0
Thickness of top flange plate [mm]	10				
Axial stress in top plate (one friction interface) [Mpa]	92	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint
Length through cross-section (one friction interface) [mm]	50	18	18	0	18
		16	16	0	16
cubic of one angle plate [mm ³]	1750000	10	10	0	10
Weight of all 8 plates [kg]	37.8	16	16	0	16
		6	6	0	6
		Number of bolts along the flange	Min. Length of coupling (one side) [mm]	Min. length of flange plate (both sides) [mm]	
		2.00	107	214	
		2.00	126	252	
		1.00	80	160	
		2.00	180	360	
		1.00	102	204	
		cross section length [mm]			
Total number of flange bolts	64	50			
Total number of web bolts	0	0			

Accounts for both web and flange									
Joint XIV, d-s									
Safety factor for ultimate state									
1.25	Number of friction interfaces	Slip factor (0.3-0.4?packing?)	UTS [Mpa]	Moment [N*mm]	750,000,000	Height of beam [mm]	1500		
1	1	0.4		700					
Bolt Type									
M20	333	163,170	Slip resistance per bolt [N]	52,214	93,240	Number of flange bolts required (one flange, one side)	10		
M24	384	188,160		60,211	107,520		9		
M30	621	304,290		97,373	173,880		6		
M36	865	423,850		135,632	242,200		6		
M39	1030	504,700		161,504	288,400		4		
Bolt size [mm]									
20	23	28	Min. p1 [mm]	51	56	Number of bolts across the flange (ca. 375 mm space to bolt one side, beam h)	6.00		
24	27	33		60	65		5.00		
30	33	40		73	80		4.00		
36	39	47		86	94		3.00		
39	42	51		93	101		3.00		
Flange fixturing characteristics									
Width of flange plate [mm]	420		Flange thickness [mm] beam d	Flange force [N]	Web thickness [mm]	Web force [N]			
Thickness of flange plate [mm]	10	30		500,000	20		0		
Axial stress in top plate (one friction interface) [Mpa]	119	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint		20		
Length through cross-section (one friction interface) [mm]	40	20		0	0		0		
cubic of one angle plate [mm^3]	2192400	18		0	0		18		
Weight of all 4 plates [kg]	23.67792	12		0	0		12		
		24		0	0		24		
		8		0	0		8		
Number of bolts along the flange									
		2.00	Min. length of coupling (one side) [mm]	107					
		2.00		126					
		2.00		153					
		2.00		266					
		2.00		195					
cross section length [mm]									
Total number of flange bolts	48	40							
Total number of web bolts	0	0							

Joint XII, s-t		Accounts for both web and flange		Safety factor for ultimate state		Slip factor (0,3-0,4?packing?)		UTS [Mpa]		Moment [N*mm]		Height of beam [mm]		400	
		Number of friction interfaces		1				0.3		700					
1.25		As [mm^2]		Preloading (F_p) [N]		Slip resistance per bolt [N]		Shear resistance per bolt [N]		Number of flange bolts required (one flange, one side)					
Bolt Type															
M20		333	163,170	39,161	93,240										
M24		384	188,160	45,158	107,520										
M30		621	304,290	73,030	173,880										
M36		865	423,850	101,724	242,200										
M39		1030	504,700	121,128	288,400										
Bolt size [mm]		do [mm] *3	Min. e1 and e2 [mm]	Min. p1 [mm]	Min. p2 [mm]	Number of bolts across the flange (ca. 365 mm space to bolt one side, beam h)									
		20	23	28	51										
		24	27	33	60										
		30	33	40	73										
		36	39	47	86										
		39	42	51	93										
Flange fixing characteristics				Flange thickness [mm] beam d	Flange force [N]	Web thickness [mm]	Web force [N]								
Width of flange plate [mm]			420		111,000	20									
Thickness of flange plate [mm]			10												
Axial stress in top plate (one friction interface) [Mpa]			26	Total number of flange bolts	Number of web bolts (one web)	Total number of web bolts	Total number of bolts in the joint								
Length through cross-section (one friction interface) [mm]			45		6	0	6								
cubic of one angle plate [mm^3]			747600		6	0	6								
Weight of all 4 plates [kg]			8.07408		4	0	4								
					8	0	8								
					2	0	2								
				Number of bolts along the flange	Min. Length of coupling (one side) [mm]										
					1.00	56									
					1.00	66									
					1.00	80									
					1.00	94									
					1.00	102									
				cross section length [mm]											
Total number of flange bolts			16		45										
Total number of web bolts			0		0										

Appendix D

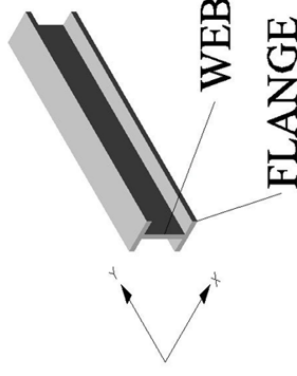
Beam Design

D.1 Design sheets exported from excel

Beam-A

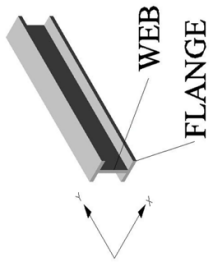
	Current square beam	New I beam (6082 extrusion and 5083 plate)	L_current/L_new	A_current/A_new	Density	Yield	As-welded
Height	1400	1500			Steel	7800	355
Width	400	500		500	Aluminium	2700	255
Flange thickness		50		55			
Web thickness	40	60		60			
Length	11590	11590					
I_xx	30,134,186,667	40,011,666,667		0.753	0.769	Limit for A_current/A_new	
I_yy	3,862,186,667	5,686,866,667		0.679	0.718	Limit for I_current/I_new	
J_co	33,996,373,333	45,698,533,333		0.744	0.718	Limit for I_current/I_new	
				weight save			
Vekt [kg/m]	1073.28	361.8		0.66 %			
Vekt [kg]	12,439	4,193					
Weight extrusion [kg/m]		67.5					
Weight extrusion [kg]		782.325					
Length extrusion (max 535kg) [m]		7.93					
Length of T-flange (vertical)		308.8235294					
Weight of extruded T-flange [kg/m]		117.5294118					
LengthT-flange extrusion (max 535kg) [m]		4.552052052					
Flange bending		4.4	3.5--4.5--5				
Web bending		9.333333333	9--13--18				
Flange bending with stiffener		3.751040559					
Length of stiffener		5					
		138400	mm^2 Tverrsnitt				

I-BEAM



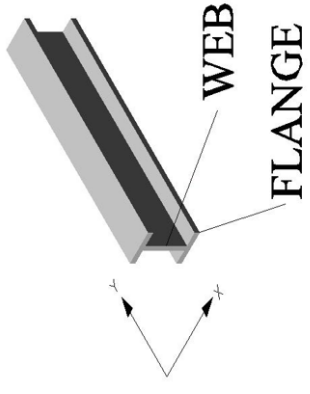
Beam-b	Current I beam	New I beam (6082 extrusion and 5083 plate)	L_current/L_new	A_current/A_new	SMYS Ratio	Small end of beam	New I beam (6082 e)	L_current/A_current	SMYS Ratio	Density	Yield
Height	1400	1500				Height	1000	1000	Steel	7800	355
Width	400	330				Width	400	330	Aluminium	2700	215
Flange thickness	20	30				Flange thickness	20	30			255
Web thickness	15	35				Web thickness	15	35			
Length	1400	1400				Length (flange-flange)	1400	1400			
L_xx	10,762,453,333	19,407,060,000	0.555	0.519	0.606	L_xx	4,948,053,333	7,081,476,667	0.699	0.537	0.718
L_yy	999,115,833	1,411,440,000	0.708	0.718	0.718	L_yy	788,003,333	983,747,292	0.781	0.718	0.718
J_co	11,761,569,167	20,818,500,000	0.565	0.606	0.606	J_co	5,716,056,667	8,065,223,958	0.709	0.718	0.718
Vekt [kg/m]	283.92	189.54	0.33	%		Weight [kg/m]	237.12	142.29	0.40	%	
Vekt [kg]	397.488	265.356				Weight [kg]		26.73			
Weight extrusion [kg/m]		37.422				Weight extrusion [kg/m]					
Weight extrusion [kg] (Flange Only)		20.01									
Length of T-flange (max 535kg) [m]		308,823,594									
Length of T-flange (vertical)		55,913,823,53									
Weight of extruded T-flange [kg/m], reell vekt		9.57	3								
Length T-flange extrusion (max 535kg) [m]											
Flange bending		4.916666667	3.5-4.5-5								
Web bending		16.45714286	9-13-18								
Flange bending with stiffener		4.270861709									
Length of stiffener		5									

I-BEAM



beam-c	Current I beam	New I beam (6082 extrusion and 5083 plate)	I_current/I_new	A_current/A_new	SMYS Ratio	Density	Yield
Height	1400	1500				Steel	355
Width	400	330		310		Aluminium	255
Flange thickness	20	30		30			
Web thickness	15	35		45			
Length (flange-flange)	5600	6400					
I_xx	10,762,453,333	19,407,060,000	0.555	0.519	0.718		
I_yy	999,115,833	1,411,440,000	0.708		0.718		
J_co	11,761,569,167	20,818,500,000	0.565		0.718		
Vekt [kg/m]	283.92	189.54				6082 as welded	130 Mpa
Vekt [kg]	1,590	1,213				5083 as welded	150 Mpa
Weight extrusion [kg/m]		26.73					
Weight extrusion [kg] (Flange Only)		171.072					
Length extrusion (max 535kg) [m]		20.01					
Length of T-flange (vertical)		308.8235294					
Weight of extruded T-flange [kg/m], reell vekt		55.91382353					
Length T-flange extrusion (max 535kg) [m]		9.57					
Flange bending		4.916666667	3.5-4.5-5				
Web bending		16.45714286	9-13-18				
Flange bending with stiffener		4.270861709					
Length of stiffener		5					

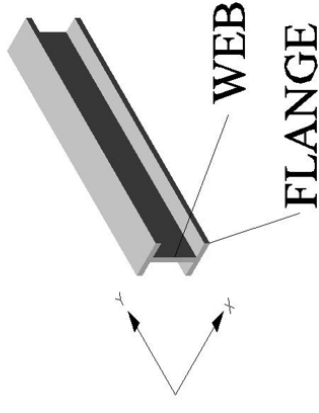
I-BEAM



Beam-d	Current I beam	New I beam (6082 extrusion and 5083 plate)	I_current/I_new	A_current/A-new	Density	Yield
Height	1400	1500			Steel	7800
Width	400	330			Aluminium	2700
Flange thickness	20	30				
Web thickness	15	35				
Length (tube surface-tube surface)	4895	4895				
I_xx	10,762,453,333	19,407,060,000	0.555	0.519		0.718
I_yy	999,115,833	1,411,440,000	0.708			0.718
J_co	11,761,569,167	20,818,500,000	0.565			0.718
Vekt [kg/m]	283.92	189.54	0.33 %			
Vekt [kg]	1,390	928				

Beam-e	Current I beam	New I beam (6082 extrusion)	I_current/I_new	A_current/A-new	Density	Yield
Height	340	350			Steel	7800
Width	300	300			Aluminium	2700
Flange thickness	21.5	32				255
Web thickness	13.52	20				
Length (tube surface-end)	4361.77	4361.77				
I_xx	357,164,671	526,023,027	0.679	0.679		0.718
I_yy	183,086,909	264,310,667	0.693	0.693		0.718
J_co	540,251,580	790,333,693	0.684	0.684		0.718
Vekt [kg/m]	131.94	67.28		weight saving		
Vekt [kg]	575	293		0.49 %		
Weight extrusion [kg/m]		25.92				
Weight extrusion [kg]		113.0570784				
Length extrusion (max 535kg) [m]		20.64				
Length of T-flange (vertical)		0				
Weight of extruded T-flange [kg/m]		25.92				
LengthT-flange extrusion (max 535kg) [m]		20.64		3		
Flange bending		4.375	3.5-4,5-5			
Web bending		5.72	9-13-18			
Flange bending with stiffener		3.74402006				
Length of stiffener		5				

I-BEAM

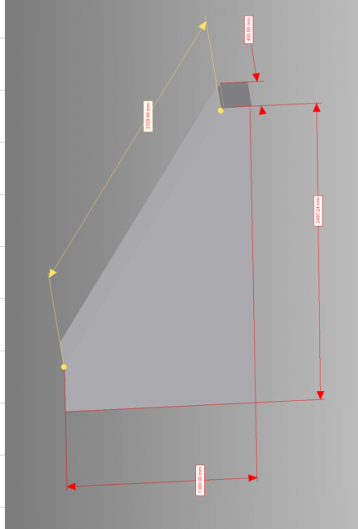


Beam-f	small_end	For 2 I-beams connected together, legg inn for 1 beam				Density	Yield
	Current square beam	New I beam (6082 extrusion and 5083 plate)	I_current/I_new	A_current/A_new			
Height	1400	1500			Steel	7800	355
Width	970	440			Aluminium	2700	255
Flange thickness (Horizontal)	40	45					
Web thickness (vertical)	20	40					
Length (ca) [mm]	3450	3000					
I_xx [mm^4]	43,559,146,667	60,618,600,000	0.719	0.679		0.718	
I_yy [mm^4]	19,258,953,333	28,359,040,000	0.679			0.718	
J_co	62,818,100,000	88,977,640,000	0.706			0.718	
Vekt [kg/m]	Vekt fra Inventor	For en bjeilke	Weight save for 2 beams				
Vekt [kg]	1017.12	259.2	0.60 %				
Weight extrusion [kg/m]	3861	777.6					
Weight extrusion [kg]		53.46					
Length of T-flange (vertical)		160.38					
Length of extruded T-flange [m]		10.01					
Weight of extruded T-flange [kg/m]		308.8235294					
LengthT-flange extrusion (max 535kg) [m]		86.81294118					
		6.16					

Beam-g	Current square beam	New I beam (6082 extrusion and 5083 plate)	I_current/_new	A_current/A_new	Density	Yield
Height	1000	1000			Steel	7800
Width	400	400			Aluminium	2700
Flange thickness (Horizontal)	25	40	40			355
Web thickness (vertical)	12	35	35			255
Length	7346	7346				150
I_xx	6,468,916,667	9,648,240,000	0.670	0.514		0.718
I_yy	1,125,041,067	1,605,253,750	0.701			0.718
J_co	7,593,957,733	11,253,493,750	0.675			0.718
Vekt [kg/m]	333.84	173.34	Weight saving			
Vekt [kg]	2452.39	1273.36	0.48 %			
Weight extrusion [kg/m]		43.2				
Weight extrusion [kg]		317.3472				
Length extrusion (max 535kg) [m]		12.38				
Length of T-flange (vertical)		205.8823529				
Weight of extruded T-flange [kg/m]		62.65588235				
Length T-flange extrusion (max 535kg) [m]		8.54	4.4			
Flange bending		4.5625	3.5--4.5--5			
Web bending		10.51428571	9--13--18			
Flange bending with stiffener		3.915378485				
Length of stiffener		5				

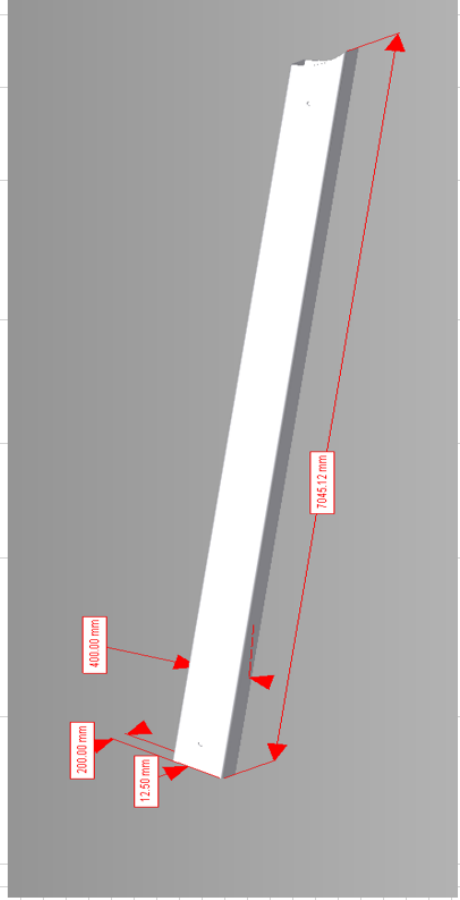
Beam-h	Current I beam	New I beam (6082 extrusion and 5083 plate)	L_current/_new	A_current/A-new	SMYS Ratio	Small end of beam	Current I beam	New I beam (6082 e.	L_current/_new	A_current/A-new	SMYS Ratio	Density	Yield
Height	1000	1000				Height	400	400	400			Steel	7800
Width	400	350				Width	400	350	350			Aluminium	2700
Flange thickness	20	40				Flange thickness	20	40	40				355
Web thickness	15	40				Web thickness	15	40	40				215
Length	1755	1755	Control length			Length	1600	1600	Control length	1625			150
L_xx	4,948,053,333	9,050,560,000	0.547	0.469	0.606	L_xx	636,653,333	1,020,160,000	0.624	0.525	0.606		
L_yy	766,003,333	1,286,940,000	0.596		0.606	L_yy	421,334,583	634,740,000	0.664		0.606		
J_co	5,716,056,667	10,339,500,000	0.553		0.606	J_co	1,057,787,917	1,654,900,000	0.639		0.606		
Vekt [kg/m]	349.44	174.96	weight saving			Vekt [kg/m]	209.04	110.16	weight saving				
Vekt [kg]	613.27	307.05	0.50			Vekt [kg]	334.46	176.26	0.47				
Weight extrusion [kg/m]						Weight extrusion [kg/m]							
Weight extrusion [kg]		66.339				Weight extrusion [kg]		60.48					
Length extrusion (max 535kg) [m]		14.15				Length extrusion (max 535kg) [m]		14.15					
Length of T-flange (vertical)		151,162,7907				Length of T-flange (vertical)		60,465,11628					
Weight of extruded T-flange [kg/m]		54,125,814	120	120, max heide		Weight of extruded T-flange [kg/m]		44,330,3256					
Length T-flange extrusion (max 535kg) [m]		9.88	4.4			Length T-flange extrusion (max 535kg) [m]		12.07	4.4				
Flange bending		3.875	3.5-4.5-5			Flange bending		3.875	3.5-4.5-5				
Web bending		0.061538462	9-13-18			Web bending		3.2	9-13-18				
Flange bending with stiffener		-21.14304382				Flange bending with stiffener		3.252775972					
Length of stiffener		5				Length of stiffener		5					

Beam-J	Current I beam	New I beam (6082 extrusion and 5083 plate)	L_current/I_new	A_current/A_new	SMYS Ratio	Small end of beam	Current I beam	New I beam (6082 ext)	L_current/A_current	A_current/SMYS Ratio	Density	Yield
Height	1400	1500				Height	200	200	200	Steel	7800	355
Width	400	400				Width	400	400	400	Aluminium	2700	255
Flange thickness	20	40				Flange thickness	20	40	40			150
Web thickness	20	30				Web thickness	20	20	20			
Length	448.45	450				Length (Diagonal)	2050.82	2050	2050			
L_x	11,810,560,000	24,215,786,667	0.468	0.579	0.718	L_x	136,960,000	213,386,667	0.642	0.539	0.718	
L_y	1,247,840,000	2,095,061,667	0.622		0.718	L_y	335,040,000	560,136,667	0.598		0.718	
L_co	13,058,400,000	26,321,348,333	0.688		0.718	L_co	472,080,000	779,533,333	0.610		0.718	
Weight [kg/m]	549.12	201.42	Weight saving			Weight [kg/m]	174.72	96.12	Weight saving			
Weight [kg]	246.25	90.64	0.66			Weight [kg]	358.32	187.05	0.45			
Weight extrusion [kg/m]		49.2				Weight extrusion [kg/m]		42.2				
Weight extrusion (max.355kg) [m]		19.44				Weight extrusion (max.355kg) [m]		88.96				
Length of I-flange (vertical)		308.8235294				Length of I-flange (vertical)		12.38				
Weight of extruded I-flange [kg/m]		68.21470588				Weight of extruded I-flange [kg/m]		41.17647059				
Length-I-flange extrusion (max.355kg) [m]		7.94	4.4			Length-I-flange extrusion (max.355kg) [m]		11.50	4.4			
Flange bending		4.625 3.5--4.5-5				Flange bending		4.025 3.5--4.5-5				
Web bending		18.93333333 9--13--18				Web bending		1.6 9--13--18				
Flange bending with stiffener		3.975615077				Flange bending with stiffener		3.975615077				
Length of stiffener		5				Length of stiffener		5				



Beam-k	Current square beam	New I beam (6082 extrusion and 5083 plate)	I _{current} /I _{new}	A _{current} /A _{new}	SMYS ratio
Height	400	400			
Width	200	300			
Flange thickness	12.5	20			
Web thickness	7045	8045			
I _{xx}	297,623,698	511,360,000	0.582	0.749	0.718
I _{yy}	99,186,198	241,440,000	0.411		0.718
J _{co}	396,809,896	752,800,000	0.527		0.718
			Weight saving		
Vekt [kg/m]	112.125	51.84		0.54 %	
Vekt [kg]	789.920625	417.0528			
Lengde [m]		10.32021605			
epsilon	0.990147543				
Flange bending					
Web bending		7.069653457	2.5--4.5--5		
Flange bending with stiffener		7.129062309	9--13--18		
Length of stiffener		6.32455532			
		5			
					Class 4, må bruke 18.6 mm tykkelse i utregninger
Rho_c, flange	0.934304413				
Rho_c, web	0.930486797				

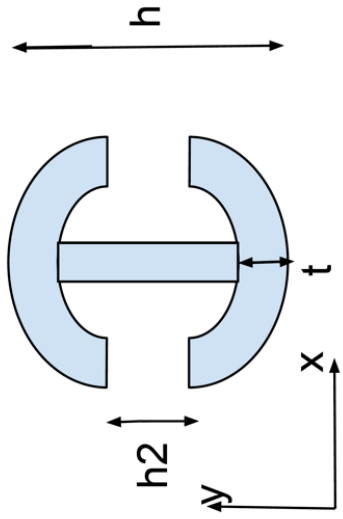
Density	Yield	Yield	Yield
Steel	7800	355	
Aluminium	2700	215	255
		6082 Extruded	Welded
			150



Beam-I	Current Tube	New Tube, 6082 extrusion	I_current/I_new	A_current/A_new	SMYS Ratio	Density	Yield
Diameter	813	970			Steel	7800	355
Thickness	28.2	40			Aluminium	2700	215
Length	8277	8277	7140				150
I_xx	5,357,055,364	12,651,766,500	0.423	0.595	0.423	0.606	
I_yy	5,357,055,364	12,651,766,500	0.423		0.423	0.606	
J_co	10,714,110,727	25,303,533,000	0.423		0.423	0.606	
Vekt [kg/m]	542	315	Weight saving				
Vekt [kg]	4,486	2,610	0.42 %				
Tetta_c				New T: 41.2032492			
		1.03008123					
beta		14.46547614			beta	$3*((D-t)/t)^{0.5}$	
epsilon		1.078327732			epsilon	$(250^{0.5})/(YS)$	
Betta/epsilon		13.41473071			Betta/epsilon < 15 -->		
					Klasse 1	< 10	
					Klasse 2	10 - 13,5	
					Klasse 3	13,5 - 15	
					Klasse 4	15 <	

Beam-m	Current Tube	New Tube, 6082 extrusion	I_current/I_new	A_current/A_new	Density	Yield
Diameter	914	920			Steel	355
Thickness	14.3	25	25		Aluminium	215
Length	6812	6812				150
I_xx	4,088,637,051	7,040,240,609	0.581	0.575	0.606	0.423
I_yy	4,088,637,051	7,040,240,609	0.581		0.606	0.423
J_co	8,177,274,102	14,080,481,219	0.581		0.606	0.423
Vekt [kg/m]	315	190	Weight saving			
Vekt [kg]	2,147	1,292	0.40 %			
Epsilon	1.078327732					
Buckling	3.056400306					
Beam m designed according to NORSOK.						

Beam-m2	Current square beam	New I beam (6082 extrus	I_current/I_new	A_current/A_new	Density	Yield
Height	914	1280			Steel	7800
Diameter		500			Aluminium	2700
Width						355
Flange thickness (Horizontal)	14.3	30				255
Web thickness (vertical)		30				
Length	6812	6812				
I_xx	4,090,710,867	5,806,563,575	0.704	0.500		0.718
I_yy	4,090,710,867	1,230,864,254	3.323			0.718
J_co	8,181,421,734	7,037,427,829	1.163			0.718
				Weight saving		
Vekt [kg/m]	315.2667961	218.4204323		0.31 %		
Vekt [kg]	2147.60	1487.88				
Weight extrusion [kg/m]		119.6004323				
Weight extrusion [kg]		814.718145				
Length extrusion (max 535kg) [m]		4.47				
Length of T-flange (vertical)		263.5294118				
Weight of extruded T-flange [kg/m]		140.9463147				
Length T-flange extrusion (max 535kg) [m]		3.80		6.812		
Flange bending		7.833333333	3.5--4,5--5			
Web bending		16.266666667	9--13--18			
Flange bending with stiffener		-0.966987557				
Length of stiffener		5				



Beam-n	Current Tube	New Tube, 6082 extrusion	I_current/I_new	A_current/A_new	Density	Yield
Diameter	914	920			Steel	7800
Thickness	14.3	24.01	25		Aluminium	2700
Length	13357	13357	12499-500 = 11999			150
I_xx	4,088,637,051	6,783,487,966	0.603	0.598		0.606
I_yy	4,088,637,051	6,783,487,966	0.603			0.606
J_co	8,177,274,102	13,566,975,932	0.603			0.606
			Weight saving			
Vekt [kg/m]	315.11	182.38	0.42	%		
Vekt [kg]	4208.88	2436.11				
Tetta_c			New T:	22.84982499		
		0.951679508				
beta		18.32638289			beta	$3*((D-t)/t)^{0.5}$
epsilon		1.078327732			epsilon	$(250^{0.5})/(YS)$
Betta/epsilon		16.99518833			Betta/epsilon	$< 15 \rightarrow$
					Klasse 1	< 10
					Klasse 2	10 - 13,5
					Klasse 3	13,5 - 15
					Klasse 4	15 <

Beam-o	Current Tube	New Tube, 6082 extrusion	I_current/I_new	A_current/A_new	Density	Yield
Diameter	600.65	620			Steel	7800
Thickness	15	25			Aluminium	2700
Length	11374	11374	11374-1000= 10374			
I_xx	1,183,396,194	2,070,601,859	0.572	0.591		0.606
I_yy	1,183,396,194	2,070,601,859	0.572			0.606
J_co	2,366,792,389	4,141,203,719	0.572			0.606
			Weight saving			
Vekt [kg/m]	215.16	126.11	0.41 %			
Vekt [kg]	2447.19	1434.38				
Epsilon	1.078327732					
Buckling	2.667523237	ok, Class 1				

Beam-q	Current Tube	New Tube, 5083 plates *3	I_current/I_new	A_current/A_new	Density	Yield
Diameter	4992	4992			Steel	7800
Thickness	24	40			Aluminium	2700
Length	7000	7000				215
I_xx	1,155,063,338,496	1,906,645,249,946	0.606	0.602		
I_yy	1,155,063,338,496	1,906,645,249,946	0.606			
J_co	2,310,126,676,992	3,813,290,499,891	0.606			
			Weight saving			
Vekt [kg/m]	2920.23	1679.32	0.42 %			
Vekt [kg]	20441.61	11755.26				
Epsilon	1.078327732					
Buckling	4.983948525	Class 1				

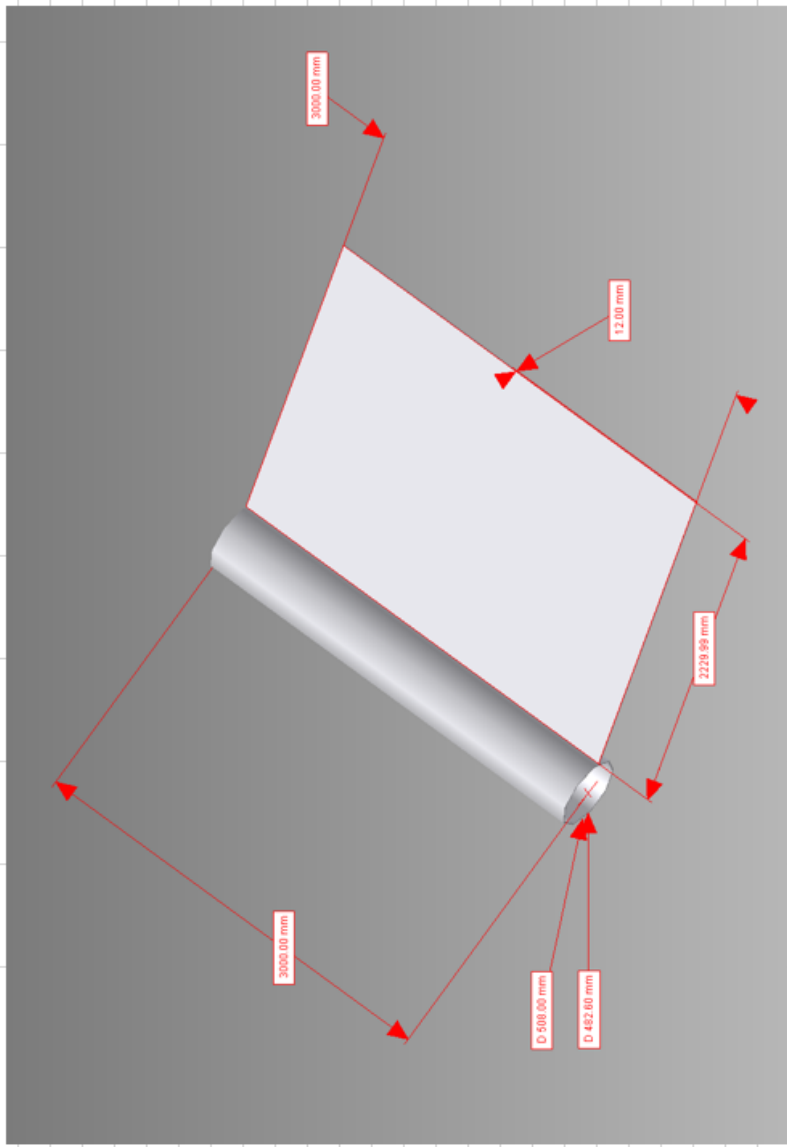
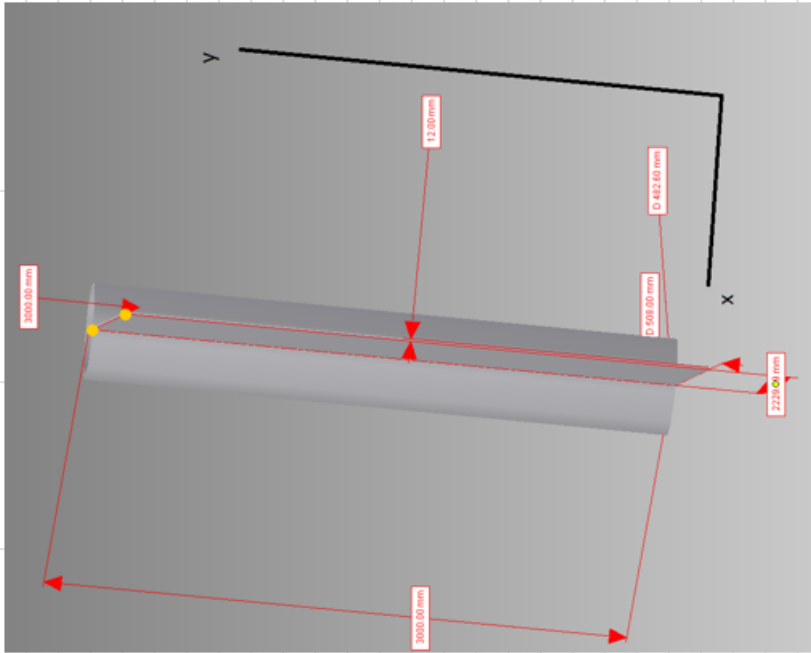
Beam-q1

	Current plate	New plate	I _{current} /I _{new}	A _{current} /A _{new}	Density	Yield
Height	3,000	3,000			Steel 7800	355
Width	2,230	2,230			Aluminium 2700	215
Thickness	12	20				

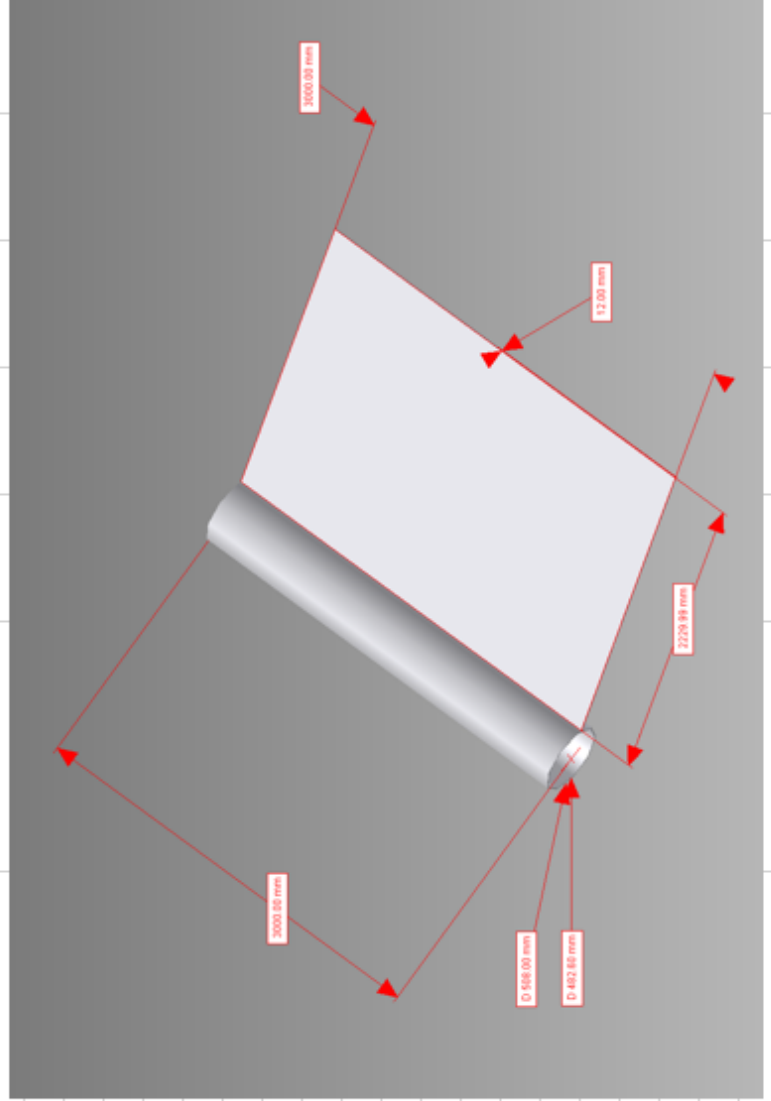
I _{xx}	108,000,000,000	180,000,000,000	0.600	0.600	0.606
I _{yy}	1,728,000	8,000,000	0.216		0.606
J _{co}	108,001,728,000	180,008,000,000	0.600		0.606

Weight saving 0.42 %

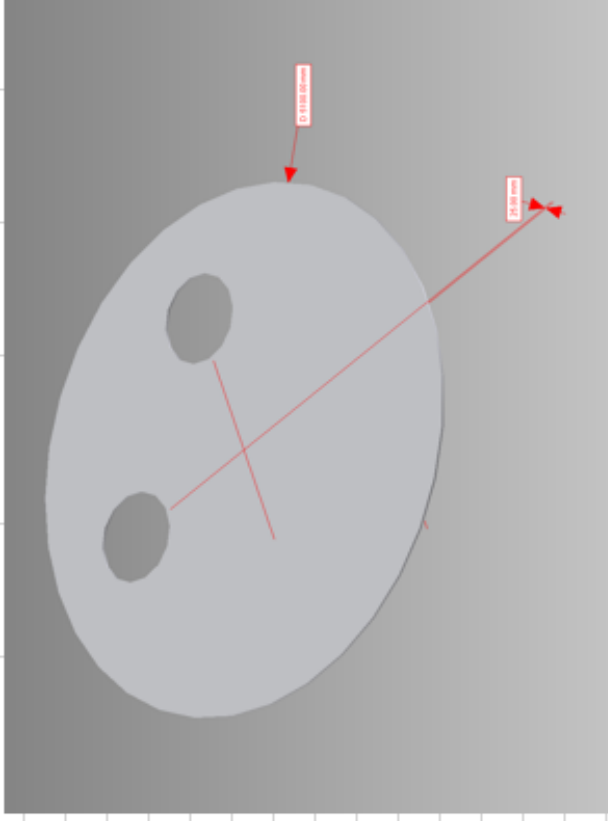
Vekt [kg/m]	-	626	Epsilon 1.078328
Vekt [kg]	361		Buckling height 55.64171 <5
			Bukling width 41.36015 <5



Beam-q2	Current cylinder	New cylinder	I_current/I_new	A_current/A_new	Density	Yield
Diameter	508	508			Steel	355
Length	3,000	3,000			Aluminium	215
Thickness						
I_xx						
I_yy	12	25				
J_co	575,360,829	1,109,183,092	0.519	0.480		0.606
	575,360,829	1,109,183,092	0.519			0.606
Vekt [kg/m]	1,150,721,658	2,218,366,184	0.519			0.606
Vekt [kg]	-	-	Weight saving			
			0.30 %			
Epsilon	1.078327732	438				
Buckling	2.683298826	307				
	Class 1					



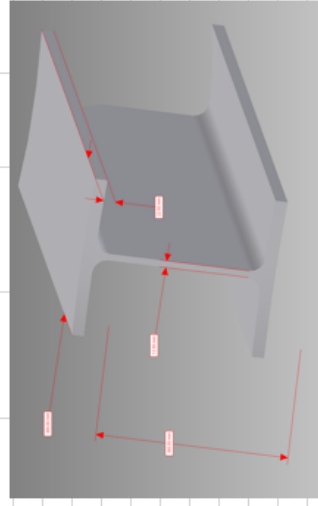
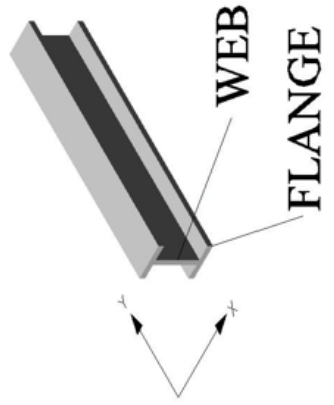
Beam-q3	Current cylinder	New cylinder	I_current/I_new	A_current/A_new	Density	Yield
Diameter	5,100	5,100			Steel	7800
Length					Aluminium	2700
Thickness	25	45				215
I_xx	6,640,625	38,728,125	0.171	0.556		
I_yy	276,356,250,000	497,441,250,000	0.556			
J_co	276,362,890,625	497,479,978,125	0.556			
Vekt [kg/r]	-	-	Weight saving			
Vekt [kg]	3,984	2,482	0.38	%		
Epsilon	1.078327732					
buckling	21.02020192					



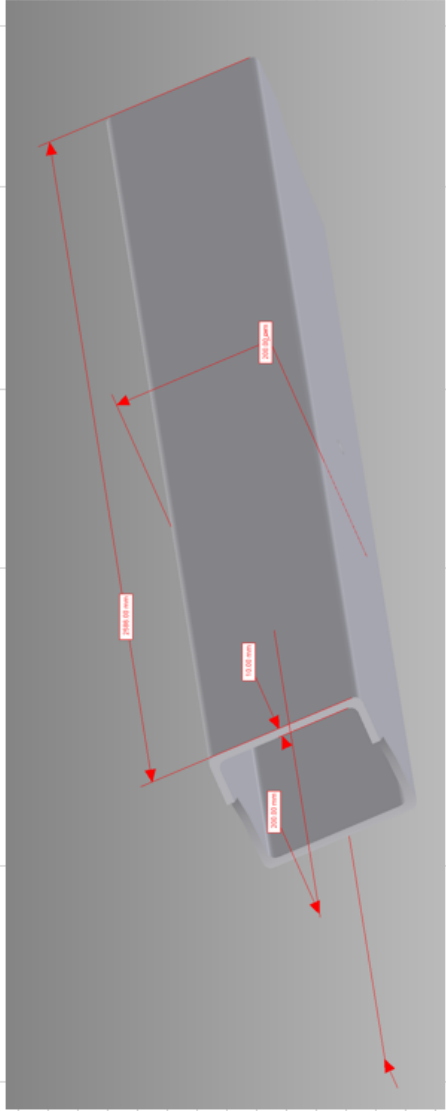
Beam-s								Density	Yield
	Current square beam		New I beam (6082 extrusion and 5083 plate)					Steel	355
Height	400		400					Aluminium	215
Width	400		420						
Flange thickness			35						
Web thickness	12.5		20						
Length	6768		8268	6768+1500					
I_xx	485,384,115		1,042,100,000	0.466	0.538	0.718			
I_yy	485,384,115		709,600,000	0.684		0.718			
J_co	970,768,229		1,751,700,000	0.554		0.718			
				Weight saving					
Vekt [kg/m]	151		97	0.21 %					
Vekt [kg]	1,023		804						
Lengde [m]			5.504115226	8.2					
			Square beam						
Heigt			400						
Width			400						
Thickness			25						
Length			8268						
I_xx			882,812,500	0.550	0.517	0.606			
I_yy			882,812,500	0.550		0.606			
J_co			1,765,625,000	0.550		0.606			
				Weight saving					
Vekt [kg/m]	151.125		101.25	0.33 %					
Vekt [kg]	1022814		837135						
Buckling in flange			5.714285714	3-6 for 6082					
Buckling in web			6.6						

Beam-w	Current I beam	New I beam (6082 extr I_current/I_new	A_current/A_new	Density	Yield
Height	360	360		Steel	355
Width	300	300		Aluminium	255
Flange thickness	22.5	35			150
Web thickness	12.5	20			
Length (tube surface-end)	1312.5	1312.5			
I_xx	417,561,328	597,323,333	0.699		0.718
I_yy	186,203,613	279,493,333	0.666		0.718
J_co	603,764,941	876,816,667	0.689		0.718
Vekt [kg/m]	136.01	72.36	Weight saving		
Vekt [kg]	179	95	0.47 %		
Weight extrusion [kg/m]		28.35			
Weight extrusion [kg]		37.209375			
Length extrusion (max 535kg) [m]		18.87			
Length of T-flange (vertical)		74.11764706			
Weight of extruded T-flange [kg/m]		32.35235294			
LengthT-flange extrusion (max 535kg) [m]		16.54			3
Flange bending		4	3.5--4.5--5		
Web bending		5.8	9--13--18		
Flange bending with stiffener		3.378105528			
Length of stiffener		5			
Alt i en ekstrusjon:		72.36	kg/m i 1 ekstrusjon		
Lengde [m]		7.39	m		

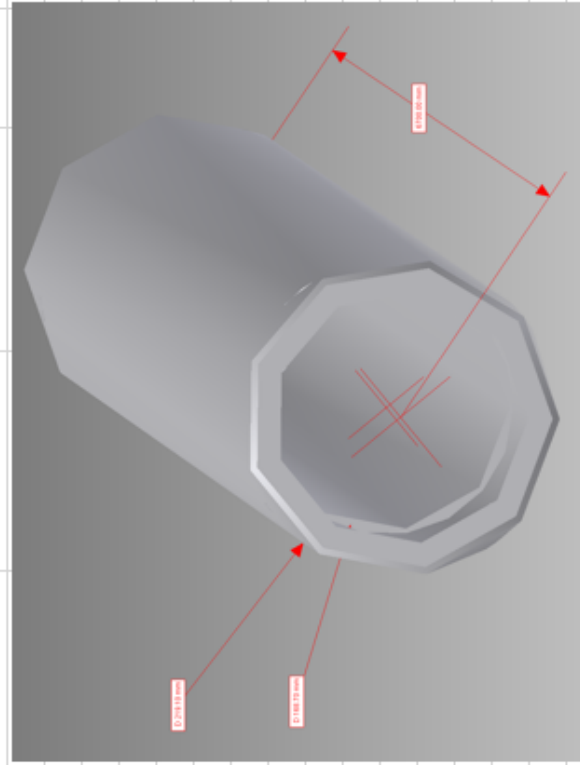
I-BEAM



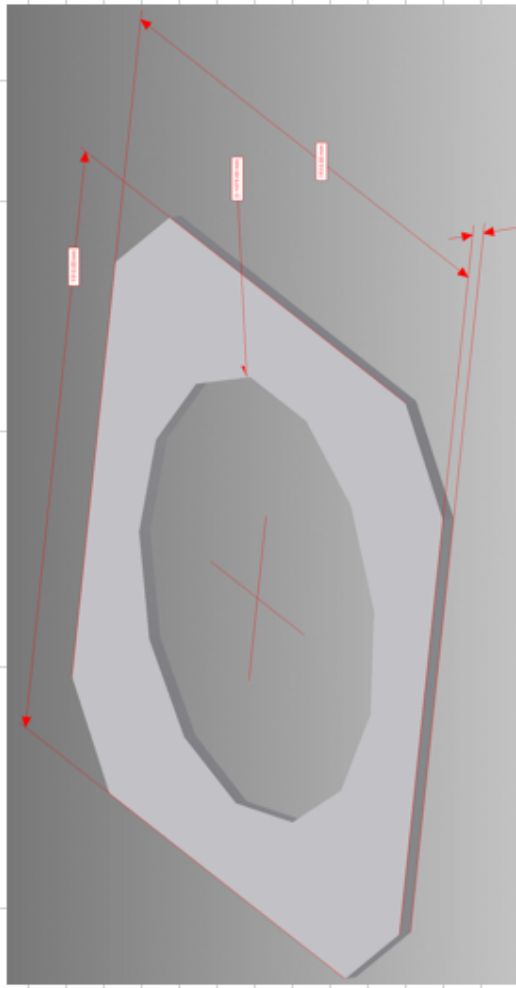
Beam-x	Current square beam	New I beam (6082 extrusion and I_current/I_new	A_current/A_new	SMYS Ratio	Density	Yield
Height	200	200			Steel	7800
Width	200	200			Aluminium	2700
Flange thickness	10	15				355
Thickness	2586	2586				255
I_xx	45,853,333	63,732,500	0.719	0.685	epsilon	0.990148
I_yy	45,853,333	63,732,500	0.719	0.718	Buckling	11.22167 Class 2
J_co	91,706,667	127,465,000	0.719	0.718		
			Weight saving			
Vekt [kg/m]	59.28	29.97	0.49 %			
Vekt [kg]	153.29808	77.50242				



Beam-y	Current cylinder	New cylinder	I_current/I_new	A_current/A_new	Density	Yield
Cylinder 1	219	219			Steel	7800
Diameter	6,700	6,700			Aluminium	2700
Length	25	50				255 Extruded
Thickness	73,361,481	103,243,188	0.711	0.504		
I_xx	73,361,481	103,243,188	0.711			
I_yy	146,722,963	206,486,375	0.711			
J_co						
			Weight saving			
Vekt [kg/m]	120	72	0.40 %		Epsilon	0.990147543
Vekt [kg]	802	481			Buckling	1.515972374 Class 1



beam-z	Current plate	New plate	I_current/I_new	A_current/A_new	Density	Yield
Plate	1,510	1,510			Steel	355
Height	1,510	1,510			Aluminium	2700
Width	30	50				215 5083 Plates
Thickness						
Minus Hole						
Diameter	1,075	1,075				
I_xx	5,501,635,313	9,169,392,188	0.600	0.600		
I_yy	978,750	4,531,250	0.216			
J_co	5,502,614,063	9,173,923,438	0.600			
			Weight saving			
Vekt [kg/m]	10,705	3,706	0.42 %			
Vekt [kg]	321	185				



Appendix E

Anode Calculation

Anode-Calculation			
	Aluminium anode	Zinc anode	Accodring to Norsok-M-CR-503 for
Cap. [Ah/kg]	2000	780	Capacity for temperatures up to 30.deg Celsius
CCP	-1050	-1030	[mV vs Ag/AgCl]

6.3.11 - DnV -	For aluminium components, or those coated with either aluminium or zinc, a design current density of 0.010 A/m ² is recommended for initial/final as well as mean values		
ic=	0.01	A/m ²	From DnV when Aluminium are to be protected
u=	0.9	Utilization factor, Norsok-6.9.2 and from Table 10-8 in Annex A in DnV.	
Ac=	1521	m ²	Exposed area of structure in aluminium

ALUMINIUM CP

NORSOK-M-CR503			
4 Wells	8 Amps/well	I _{well}	32 A

20mA/m ² for suction anchor	i= 20mA/m ²		
Area pr. suction anchor	A=pi*D*H	D=5100	H= 7025
	112555410.8	mm ²	112.5554
4 suction anchors	I=4*A*i	9004.43286	mA
	I _{sa}	9.00443286	A

	Ac*ic	15.21	
Current Demand Calculation	(Ac*ic)+I _{well} +I _{sa}	56.2144329	A

Anode Mass Calculation		Ma=(Icm*tf*8760)/(u*CCP)	
	20 yrs	5471.538132	kg
	25 yrs	6839.422665	kg
	30 yrs	8207.307198	kg

STEEL CP

8 amps per well:	I _{well}	32	A
	i _{sa}	20	mA/m ²
4 SA	A	484	m ²
Surface structure:	A _{st}	1521	m ²
Current surface:	I _{st} = A _{st} *i _{st} *fc		

where, fc is coating breakdown factor, i_{st} is current density and A_{st} is steel surface area (initially coated by a 2 layer epoxy, total DFT = 350 µm.

DnV:	ic and fc are then to be selected according to (6.3) and (6.4), respectively.
	2 layer epoxit, 350 µm DFT --> a=0.02 and b = 0.008
	$fc = a + b*t$

initial	mean	final	
0.22	0.11	0.17	A/m ² for seawater exposed bare metal

Depth more than 300m (assumption) and Temperature range 7-11 degrees celsius:

$$Ma = (l_{cm} * t_f * 8760) / (u * CCP)$$

$$l_{cm} = I_{st} + I_{well} + I_{sa} = (A_{st} * i_{mean} * f_{cm}) + (I_{well}) + (i_{sa} * A)$$

Current calc:

years	fc	I _{st} [A]	I _{well} [A]	I _{sa} [A]	I _{tot} [A]
20	0.1	16.731	32	9.68	58.411
25	0.12	20.0772	32	9.68	61.7572
30	0.14	23.4234	32	9.68	65.1034

Anode Mass Calculat	$Ma = (l_{cm} * t_f * 8760) / (u * CCP)$	
20 yrs	5685.337333	kg
25 yrs	7513.792667	kg
30 yrs	9505.0964	kg

Appendix F

Additional Figures

F.1 Extrusion limits

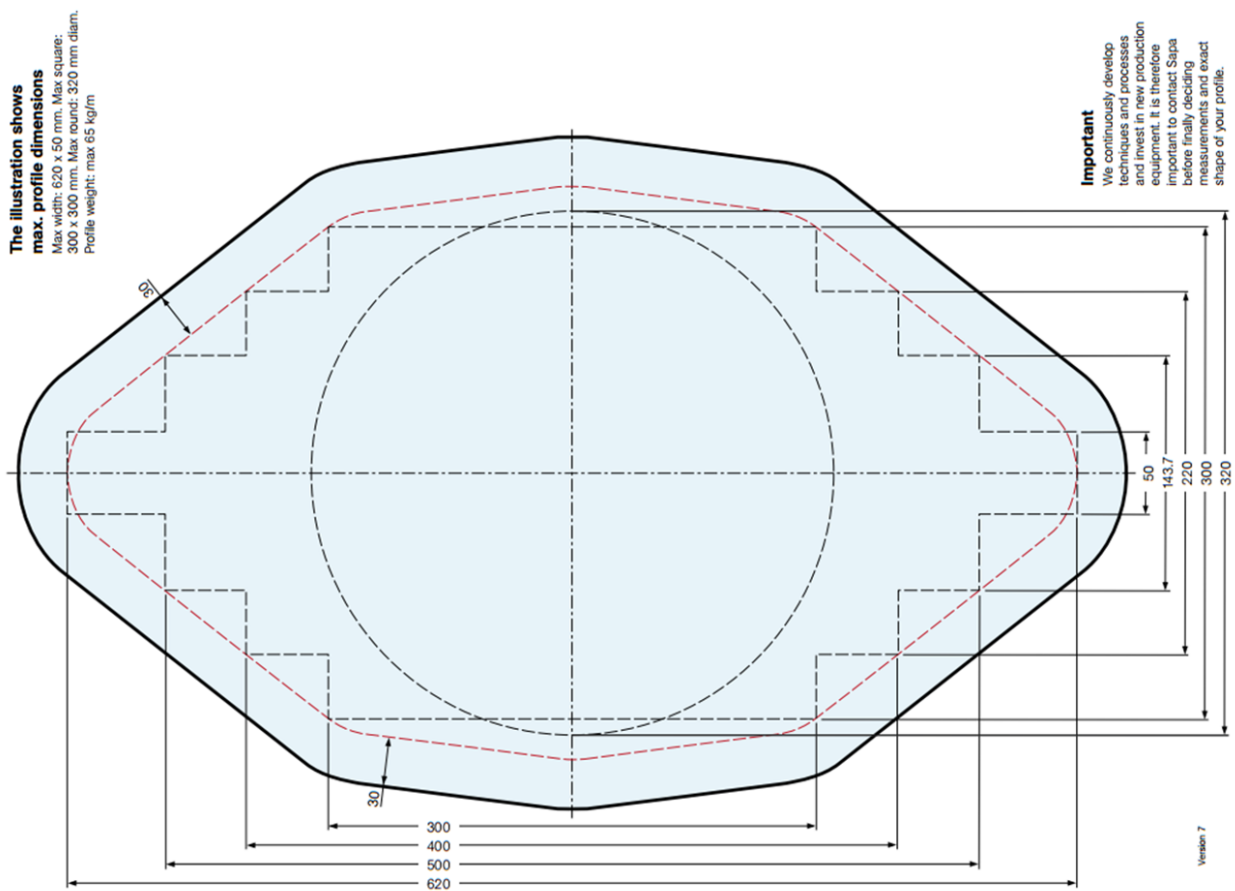


Figure F.1: Circumference limits for the extrusion press at SAPA [5]).

F.2 seawater properties as a function of depth

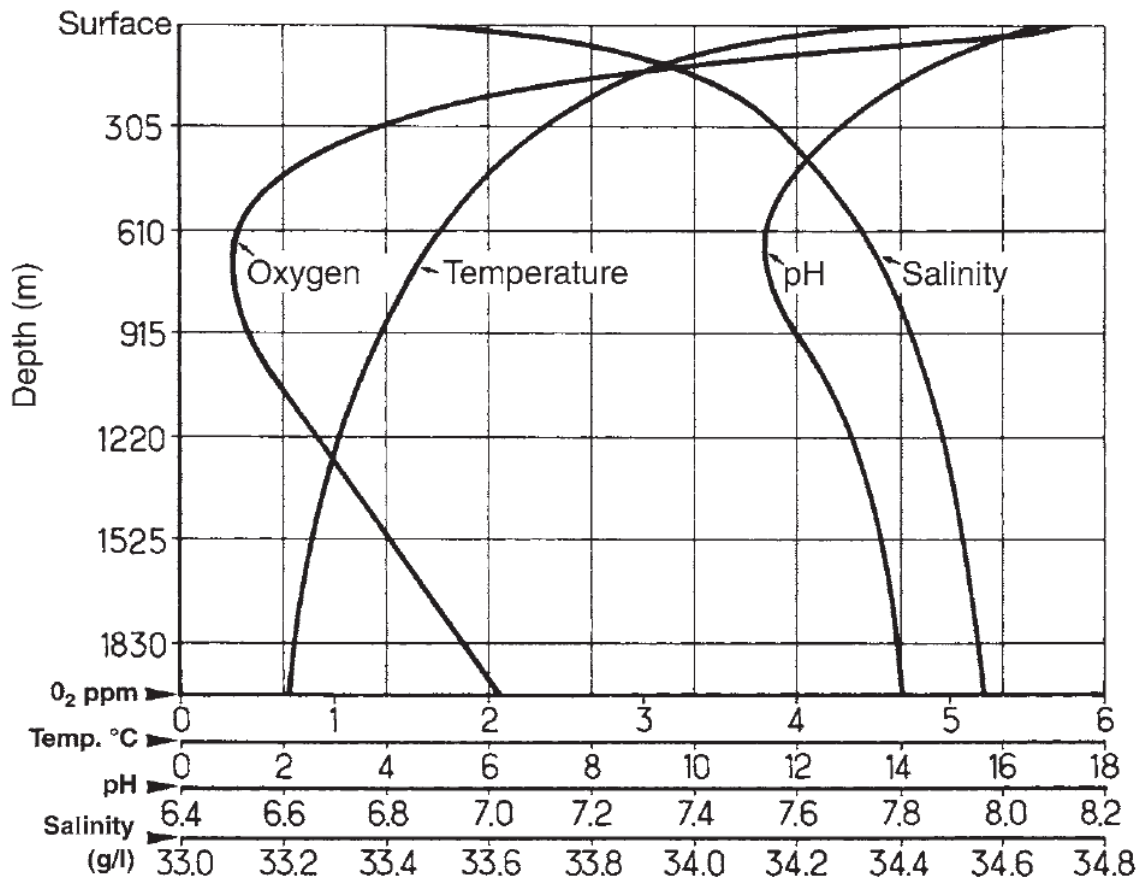
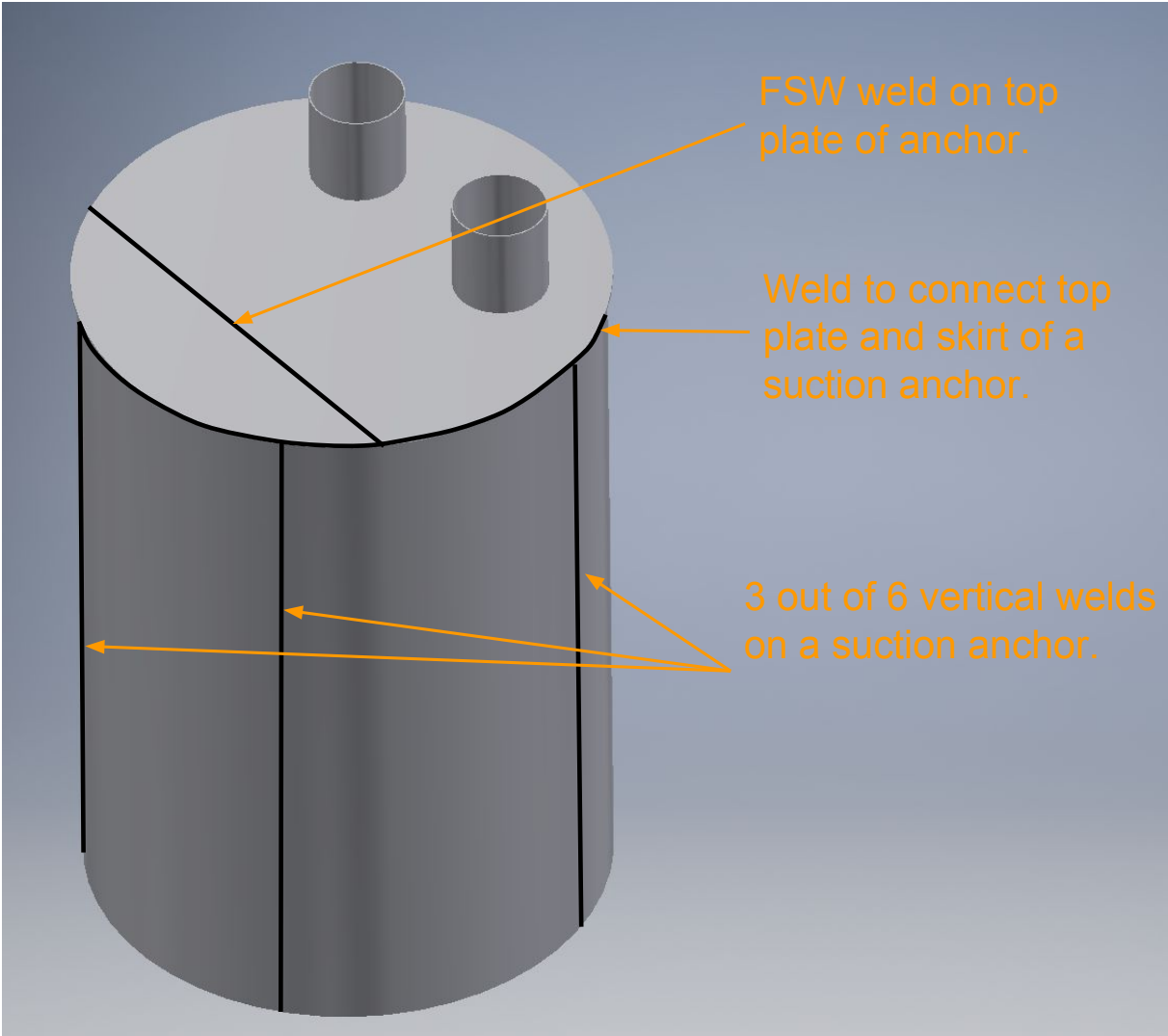


Figure F2: Seawater's change in properties by depth [19]).

F.3 Illustration of suction anchor with welds.



Appendix G

Que\$tor Cost Estimates

G.1 Full subsea layout.

OFFSHORE PROJECT SUMMARY

Project name	ITS-Project
Region	Europe
Country	Norway (North)
Basin	Trondelag Platform

Procurement strategy		Currency	Rate/\$
Offshore	ITS-test	\$	1.00
Contingency	N. North Sea (Norway)	\$	8.17
Equipment	N. North Sea (Norway)	\$	8.17
Materials	N. North Sea (Norway)	\$	8.17
Fabrication	N. North Sea (Norway)	\$	8.17
Linepipe	N. North Sea (Norway)	\$	8.17
Installation	N. North Sea (Norway)	\$	8.17
Design & PM	N. North Sea (Norway)	\$	8.17
Opex	N. North Sea (Norway)	\$	8.17
Certification	N. North Sea (Norway)	\$	8.17
Freight	N. North Sea (Norway)	\$	8.17

Technical database

N. North Sea (Norway)

Unit set	Oilfield
Development type	Oil
Development concept	Semi-submersible + Subsea tie-back

Overall input					
Design oil production flowrate	52.80	Mbbl/day	Reserves	120.00	MMbbl
Design associated gas flowrate	118.00	MMscf/day	Water depth	269.00	m
Design gross liquids flowrate	58.70	Mbbl/day	Reservoir depth	2310.00	m
Water injection	Yes		Reservoir pressure	246.00	bara
Water injection capacity factor	1.10		Reservoir temperature	71.20	°C
Design water injection flowrate	64.50	Mbbl/day	Reservoir length	4.89	km
Gas injection	Yes		Reservoir width	2.45	km
Design gas injection rate	118.00	MMscf/day			
Gas oil ratio	2230.00	scf/bbl			
Design factor	1.10				

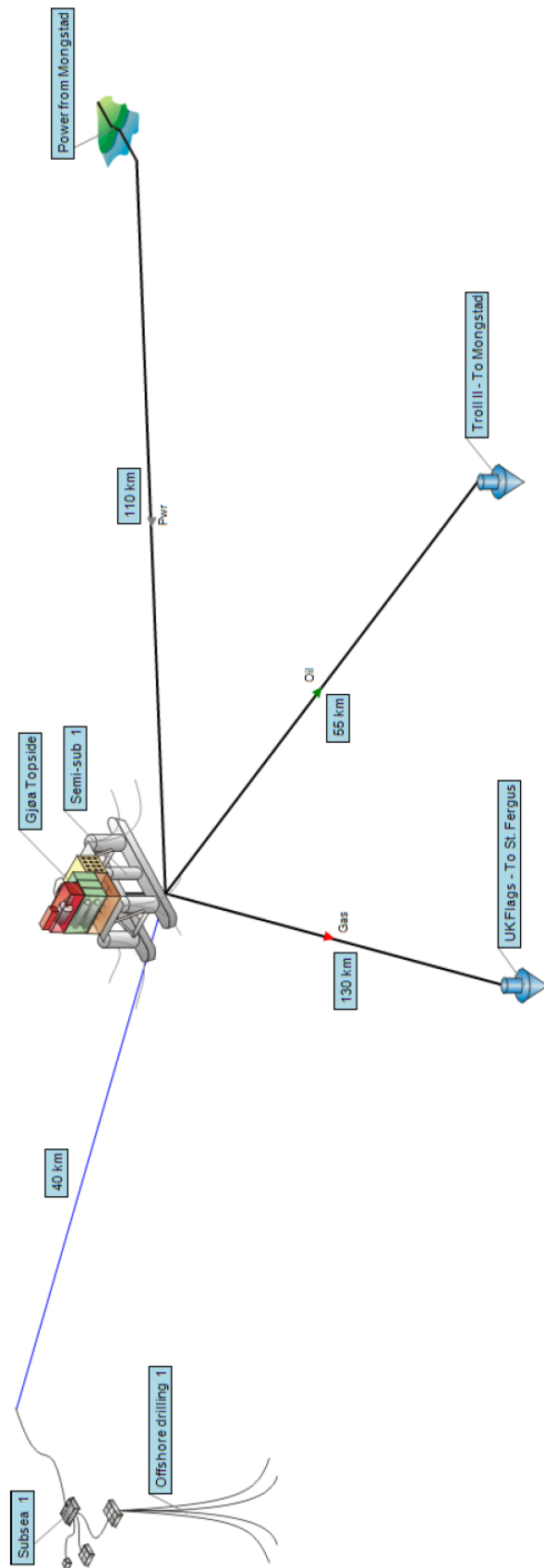
Fluid characteristics					
Oil density @ STP	40.20	°API	H2S content	0.00	ppm
CO2 content	0.30	%	Gas molecular weight	30.00	
Initial water cut	10.00	%			

Production profile characteristics					
Plateau rate	48.00	Mbbl/day	Years to plateau	2.00	year
Productivity	16.00	MMbbl/well	Plateau duration	3.00	year
Peak well flow	6.00	Mbbl/day	Field life	11.00	year
Maximum drilling stepout	3.00	km	Onstream days	350.00	day
Concurrent drilling operations	1		Wells per year per operation	4.90	

Export methods					
Oil export method	via existing pipeline		Gas export method	via existing pipeline	
Distance to delivery / tie-back point	55.00	km	Distance to delivery / tie-back point	130.00	km

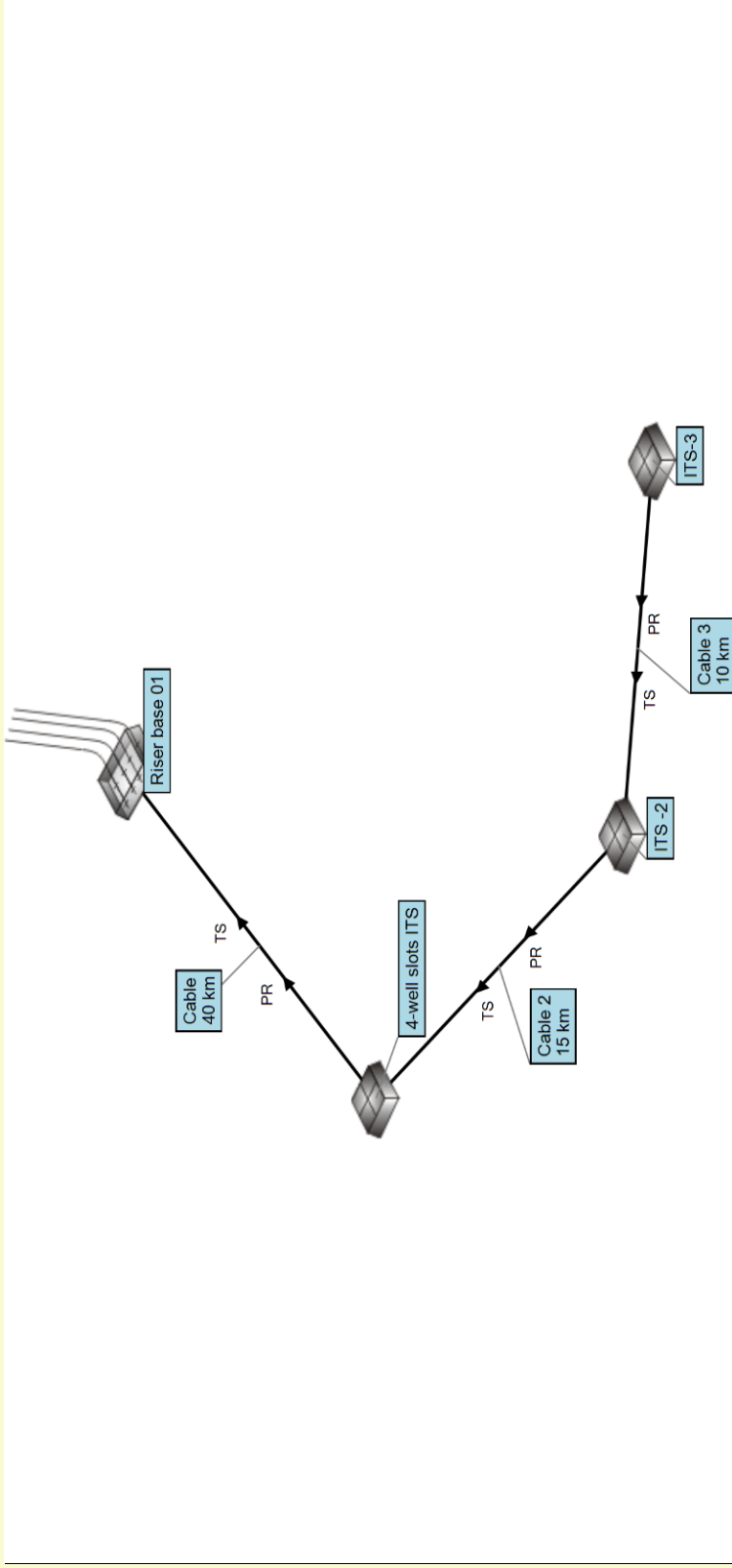
Number of wells		
Production wells	12	
Water injection wells	5	
Gas injection wells	4	

Field level miscellaneous data					
Distance to operations base	120.00	km	BOE equivalent values		
Distance to delivery point	120.00	km	BOE oil	1.00	BOE/bbl
Maximum drilling stepout	3.00	km	BOE condensate	0.94	BOE/bbl
Maximum ambient temperature	22.00	°C	BOE gas	0.17	BOE/Mscf
Average seawater temperature	10.00	°C			



Subsea input report
Component name

Subsea 1



	Well count	Flow	Type
Production	12	52.8 Mbbl/day	
Water injection	0	64.5 Mbbl/day	
Gas injection	0	118 MMscf/day	
Spare slots	0	-	

	Flow rate
Multiphase metering	No
Test	Yes
Gas lift	No
Chemical injection	No

Schematic edited	Yes
------------------	-----

Layout

Development type	Cluster manifold	Maximum wells per item	4
Infield flowline length	1.63 km	Tie-back	No
Tie-back length	-		

Features			
Water depth	350 m	Wellhead shut in pressure	203 bara
Pressure rating	345 barg	Wellhead temperature	64.7 °C
Soil conditions	Average	Acid gas	No
HIPPS	No	HIPPS minimum flowline length	-
Through pigging	No	Retrievable subsea	No
Trawler protection	No	Intervention tools	No
Diverless system	Yes		

Tie-in point			
Production delivery pressure	35 bara	Water injection pressure	136 bara
Gas injection pressure	220 bara	Test service delivery pressure	35 bara
Gas lift pressure	205 bara	Production / test service design pressure	202 barg
Production temperature	10.4 °C		

Flowlines	
Lay vessel	S-lay without DP
Flowline type	Steel
Buried lines	No

Umbilicals	
Control system	Electro-hydraulic
Tube material	Duplex
Inhibitor chemicals	Yes
Power cable	No

Flowline fluid	Production	Water Injection	Gas injection	Test service	Gas lift	Chemical injection
Flowline material	Carbon steel X60	Carbon steel X60	Carbon steel X60	Carbon steel X60	Carbon steel X60	Carbon steel X60
Insulation material	None	None	None	None	None	None
Insulation U value	-	-	-	-	-	-
PLET selected	Yes	Yes	Yes	Yes	Yes	Yes

Distance to supply base	120 km
Weather downtime (small vessels)	15 %
Weather downtime (large vessels)	10 %

Vessel durations (days)	Pipelay spread	DSV	SSCV	SSDV	Trench vessel	Survey vessel	Dredge vessel	Rock install vessel	Supply vessel
Template	0.0	18.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Satellite	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Cluster	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Manifold	0.0	2.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Equipment	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Risers	10.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Flowline links	67.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
PLETs	8.8	14.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Umbilical links	9.7	8.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Trenching	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Surveying	0.0	0.0	0.0	0.0	0.0	13.0	0.0	0.0	0.0
Dredging	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Rock installation	0.0	0.0	0.0	0.0	0.0	13.0	0.0	0.0	0.0
Transit loadout	1.9	3.8	0.0	0.0	0.0	1.3	0.0	0.0	26.0
Weather downtime	9.8	7.2	0.0	0.0	0.0	2.1	0.0	0.0	3.9
Mob/demob	8.0	8.0	0.0	0.0	0.0	8.0	0.0	0.0	8.0
Total	116.1	63.0	0.0	0.0	0.0	24.4	0.0	0.0	37.9

Subsea components	Type	Water depth	Production wells	Production flow per well	Water injection wells	Water injection flow per	Gas injection wells	Gas injection flow per well	Test service
4-well slots ITS	Template manifold	350 m	4	4.4 Mbbl/day	0	0 Mbbl/day	0	0 MMscf/day	Yes
ITS-2	Template manifold	350 m	4	4.4 Mbbl/day	0	0 Mbbl/day	0	0 MMscf/day	Yes
ITS-3	Template manifold	350 m	4	4.4 Mbbl/day	0	0 Mbbl/day	0	0 MMscf/day	Yes

Subsea components	Gas lift	Chemical injection	HIPPS	MFM on test service	MFM on production	MFM on wellheads	Spare slots	Spare slots type	Total slots count
4-well slots ITS	No	No	No	No	No	No	0	-	4
ITS-2	No	No	No	No	No	No	0	-	4
ITS-3	No	No	No	No	No	No	0	-	4

Subsea components	SDU selected	SDU wells serviced	SDU hydraulic flying	SDU electrical flying leads	UTA wells serviced	UTA hydraulic flying leads	UTA electrical flying leads
4-well slots ITS	Yes	4	0	0	4	1	2
ITS-2	Yes	4	0	0	4	1	2
ITS-3	Yes	4	0	0	4	1	2

Riser base manifold	Water depth	Termination type	Sub-type	Riser systems	Riser length
Riser base 01	350 m	Riser	Flexible lazy S	1	562 m

Cable	Length	Well end name	Well end water depth	Tie-back end name	Tie-back end water depth	
	40 km	4-well slots ITS	350 m	Riser base 01	350 m	
		Production oil flow	Production water flow	Production gas flow	Water injection flow	Gas injection flow
		52.8 Mbbl/day	5.87 Mbbl/day	118 MMscf/day	0 Mbbl/day	0 MMscf/day

Production flowline	Number of lines	Material	Carbon steel X60	Oil flow per line	52.8 Mbbl/day	Water flow per line	5.87 Mbbl/day
	1	Fixed pressure	Outlet pressure	Pressure In	63.9 bara	Pressure out	45.1 bara
	118 MMscf/day	Buckle arrestors	Yes	Inlet temperature	32.3 °C	Outlet temperature	10.5 °C
	203 bara	Wall thickness	22 mm	Corrosion allowance	3 mm	Cladding thickness	0 mm
	20 in	Weight coating	No	Cathodic protection	Yes	Insulation material	None
	Yes	Insulation U value	-				

PLETs	PLET required	Valve	Soil conditions	Pressure rating	Trawler protection	Jumper type
Well end	Yes	Yes	Average	345 barg	No	Rigid
Tie-back	Yes	Yes	Average	345 barg	No	Rigid

Test service flowline	Number of lines	Material	Carbon steel X60	Oil flow per line	8.8 Mbbl/day	Water flow per line	0.978 Mbbl/day
	1	Fixed pressure	Outlet pressure	Pressure In	84.5 bara	Pressure out	45.9 bara
	19.6 MMscf/day	Buckle arrestors	Yes	Inlet temperature	47 °C	Outlet temperature	10 °C
	203 bara	Wall thickness	11.2 mm	Corrosion allowance	3 mm	Cladding thickness	0 mm
	8 in	Weight coating	No	Cathodic protection	Yes	Insulation material	None
	Yes	Insulation U value	-				

PLETs	PLET required	Valve	Soil conditions	Pressure rating	Trawler protection	Jumper type
Well end	Yes	Yes	Average	345 barg	No	Rigid
Tie-back	Yes	Yes	Average	345 barg	No	Rigid

Umbilicals	Control and chemical tube material	Primary control tubes	Secondary control tubes	Primary chemical tubes	Secondary chemical tubes				
		Number of	Size	Number of	Size	Number of	Size	Number of	Size
	Duplex	6	25.4 mm	0	9.52 mm	6	25.4 mm	0	9.52 mm

Cable 2	Length	Well end name	Well end water depth	Tie-back end name	Tie-back end water depth
	15 km	ITS-2	350 m	4-well slots ITS	350 m

Production oil flow	Production water flow	Production gas flow	Water injection flow	Gas injection flow
35.2 Mbbl/day	3.91 Mbbl/day	78.5 MMscf/day	0 Mbbl/day	0 MMscf/day

Production flowline	Number of lines	1	Material	Carbon steel X60	Oil flow per line	35.2 Mbbl/day	Water flow per line	3.91 Mbbl/day
	Gas flow per line	78.5 MMscf/day	Fixed pressure	Outlet pressure	Pressure In	72.7 bara	Pressure out	63.9 bara
	Design pressure	203 bara	Buckle arrestors	Yes	Inlet temperature	44.1 °C	Outlet temperature	16.1 °C
	Nominal diameter	16 in	Wall thickness	18.2 mm	Corrosion allowance	3 mm	Cladding thickness	0 mm
	Coating	Yes	Weight coating	No	Cathodic protection	Yes	Insulation material	None
	Insulation U value	-						

PLETs	PLET required	Valve	Soil conditions	Pressure rating	Trawler protection	Jumper type
Well end	Yes	Yes	Average	345 barg	No	Rigid
Tie-back	Yes	Yes	Average	345 barg	No	Rigid

Test service flowline	Number of lines	1	Material	Carbon steel X60	Oil flow per line	8.8 Mbbl/day	Water flow per line	0.978 Mbbl/day
	Gas flow per line	19.6 MMscf/day	Fixed pressure	Outlet pressure	Pressure In	96.4 bara	Pressure out	84.5 bara
	Design pressure	203 bara	Buckle arrestors	Yes	Inlet temperature	48.5 °C	Outlet temperature	11.5 °C
	Nominal diameter	8 in	Wall thickness	11.2 mm	Corrosion allowance	3 mm	Cladding thickness	0 mm
	Coating	Yes	Weight coating	No	Cathodic protection	Yes	Insulation material	None
	Insulation U value	-						

PLETs	PLET required	Valve	Soil conditions	Pressure rating	Trawler protection	Jumper type
Well end	Yes	Yes	Average	345 barg	No	Rigid
Tie-back	Yes	Yes	Average	345 barg	No	Rigid

Umbilicals	Control and chemical tube material	Primary control tubes		Secondary control tubes		Primary chemical tubes		Secondary chemical tubes	
	Duplex	Number of	Size	Number of	Size	Number of	Size	Number of	Size
		6	19 mm	0	9.52 mm	6	19 mm	0	9.52 mm

Cable 3	Length	Well end name		Tie-back end name		Tie-back end water depth	
	10 km	ITS-3		ITS-2		350 m	
	Production oil flow	Production water flow	Production gas flow	Water injection flow	Gas injection flow		
	17.6 Mbbl/day	1.96 Mbbl/day	39.2 MMscf/day	0 Mbbl/day	0 MMscf/day		

Production flowline	Number of lines	1	Material	Carbon steel X60	Oil flow per line	17.6 Mbbl/day	Water flow per line	1.96 Mbbl/day
	Gas flow per line	39.2 MMscf/day	Fixed pressure	Outlet pressure	Pressure In	84.9 bara	Pressure out	72.7 bara
	Design pressure	203 bara	Buckle arrestors	Yes	Inlet temperature	64.7 °C	Outlet temperature	23.4 °C
	Nominal diameter	10 in	Wall thickness	13.2 mm	Corrosion allowance	3 mm	Cladding thickness	0 mm
	Coating	Yes	Weight coating	No	Cathodic protection	Yes	Insulation material	None
	Insulation U value	-						

PLETs	PLET required	Valve	Soil conditions	Pressure rating	Trawler protection	Jumper type
Well end	Yes	Yes	Average	345 barg	No	Rigid
Tie-back	Yes	Yes	Average	345 barg	No	Rigid

Test service flowline	Number of lines	1	Material	Carbon steel X60	Oil flow per line	8.8 Mbbl/day	Water flow per line	0.978 Mbbl/day
	Gas flow per line	19.6 MMscf/day	Fixed pressure	Outlet pressure	Pressure In	104 bara	Pressure out	96.4 bara
	Design pressure	203 bara	Buckle arrestors	Yes	Inlet temperature	64.7 °C	Outlet temperature	16.2 °C
	Nominal diameter	8 in	Wall thickness	11.2 mm	Corrosion allowance	3 mm	Cladding thickness	0 mm
	Coating	Yes	Weight coating	No	Cathodic protection	Yes	Insulation material	None
	Insulation U value	-						

PLETs	PLET required	Valve	Soil conditions	Pressure rating	Trawler protection	Jumper type
Well end	Yes	Yes	Average	345 barg	No	Rigid
Tie-back	Yes	Yes	Average	345 barg	No	Rigid

Umbilicals	Control and chemical tube material	Primary control tubes		Secondary control tubes		Primary chemical tubes		Secondary chemical tubes	
		Number of	Size	Number of	Size	Number of	Size	Number of	Size
	Duplex	4	19 mm	0	9.52 mm	4	19 mm	0	9.52 mm

Subsea 1

Name	Subsea 1
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TOTAL COST	US Dollars	385,208,000
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EQUIPMENT			Procured from: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
4-well slots ITS			34,374,000
ITS -2			33,696,000
ITS-3			32,183,000
Riser base 01			2,263,000
Platform controls - main	1	471,000	471,000
Platform controls - additional	12	68,700	824,000
Sub Total			103,811,000
Freight	3.00%		3,114,000
Total Equipment			\$ 106,925,000

MATERIALS			Procured from: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Cable			66,800,000
Cable 2			25,359,000
Cable 3			14,480,000
Riser 01			8,661,000
Riser systems (arch/buoy)	1	410,087	410,000
Sub Total			115,710,000
Freight	2.00%		2,314,000
Total Materials			\$ 118,024,000

INSTALLATION			Location: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Pipelay spread (S-lay without DP)	117 day	293,916	34,388,000
Diving support vessel tie ins	63 day	249,725	15,733,000
Diving support vessel test & commissioning	14 day	249,725	3,496,000
Testing & commissioning equipment	26 day	19,097	497,000
Semi-submersible crane vessel	0 day	355,001	0
Semi-submersible drilling vessel	0 day	141,021	0
Trench vessel	0 day	147,019	0
Survey vessel	25 day	99,155	2,479,000
Dredge vessel	0 day	226,466	0
Rock install vessel	0 day	97,931	0
Supply vessel	38 day	41,743	1,586,000
Total Installation			\$ 58,179,000

DESIGN & PROJECT MANAGEMENT			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Design	79,000 mhr	176	13,904,000
Project management	26,900 mhr	323	8,689,000
Total Design & Project management			\$ 22,593,000

INSURANCE & CERTIFICATION			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Certification	1.00%		3,057,000
Insurance	4.00%		12,229,000
Total Insurance & Certification			\$ 15,286,000

CONTINGENCY			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST

Contingency	20.00%		64,201,000
Total Contingency		\$	64,201,000

4-well slots ITS

TOTAL COST	US Dollars	34,374,000
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MAIN STRUCTURE			
	QUANTITY	UNIT RATE	COST
Structure	260 te	17,897	4,653,000
Guide base	4	360,911	1,444,000
Protection structure	0 te	15,393	0
Piles	55 te	4,807	264,000
Total Main structure			\$ 6,361,000

XMAS TREES			
	QUANTITY	UNIT RATE	COST
Production	4	3,478,800	13,915,000
Water injection	0	3,261,100	0
Gas injection	0	3,261,100	0
Total Xmas trees			\$ 13,915,000

MANIFOLDING (PIPING & VALVES)			
	QUANTITY	UNIT RATE	COST
Production	18 te	91,653	1,650,000
Test	8 te	91,653	733,000
Total Manifolding (piping & valves)			\$ 2,383,000

MULTIPHASE METERING			
	QUANTITY	UNIT RATE	COST
Multiphase meters (0-8 Mbbbl/day)	0	758,967	0
Multiphase meters (8-30 Mbbbl/day)	0	1,065,002	0
Multiphase meters (30-75 Mbbbl/day)	0	1,517,934	0
Total Multiphase metering			\$ 0

CONNECTORS / HUBS			
	QUANTITY	UNIT RATE	COST
8 in connectors	2	287,673	575,000
16 in connectors	1	465,173	465,000
20 in connectors	1	465,173	465,000
Umbilical connectors	2	119,966	240,000
Hydraulic connectors	2	42,845	86,000
Electrical connectors	4	21,178	85,000
Total Connectors / hubs			\$ 1,916,000

FLYING LEADS			
	QUANTITY	UNIT RATE	COST
Hydraulic (x 1)	50 m	2,399	120,000
Electrical (x 2)	100 m	66	7,000
Total Flying leads			\$ 127,000

CONTROL AND TESTING			
	QUANTITY	UNIT RATE	COST
Subsea distribution unit	1	3,427,592	3,428,000
Subsea controls	4	1,481,209	5,925,000
System testing	4	79,820	319,000
Total Control and testing			\$ 9,672,000

ITS -2

TOTAL COST	US Dollars	33,696,000
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MAIN STRUCTURE			
	QUANTITY	UNIT RATE	COST
Structure	260 te	17,897	4,653,000
Guide base	4	360,911	1,444,000
Protection structure	0 te	15,393	0
Piles	55 te	4,807	264,000
Total Main structure			\$ 6,361,000

XMAS TREES			
	QUANTITY	UNIT RATE	COST
Production	4	3,478,800	13,915,000
Water injection	0	3,261,100	0
Gas injection	0	3,261,100	0
Total Xmas trees			\$ 13,915,000

MANIFOLDING (PIPING & VALVES)			
	QUANTITY	UNIT RATE	COST
Production	14 te	91,653	1,283,000
Test	6 te	91,653	550,000
Total Manifolding (piping & valves)			\$ 1,833,000

MULTIPHASE METERING			
	QUANTITY	UNIT RATE	COST
Multiphase meters (0-8 Mbbbl/day)	0	758,967	0
Multiphase meters (8-30 Mbbbl/day)	0	1,065,002	0
Multiphase meters (30-75 Mbbbl/day)	0	1,517,934	0
Total Multiphase metering			\$ 0

CONNECTORS / HUBS			
	QUANTITY	UNIT RATE	COST
8 in connectors	2	287,673	575,000
10 in connectors	1	336,639	337,000
16 in connectors	1	465,173	465,000
Umbilical connectors	2	119,966	240,000
Hydraulic connectors	2	42,845	86,000
Electrical connectors	4	21,178	85,000
Total Connectors / hubs			\$ 1,788,000

FLYING LEADS			
	QUANTITY	UNIT RATE	COST
Hydraulic (x 1)	50 m	2,399	120,000
Electrical (x 2)	100 m	66	7,000
Total Flying leads			\$ 127,000

CONTROL AND TESTING			
	QUANTITY	UNIT RATE	COST
Subsea distribution unit	1	3,427,592	3,428,000
Subsea controls	4	1,481,209	5,925,000
System testing	4	79,820	319,000
Total Control and testing			\$ 9,672,000

ITS-3

TOTAL COST	US Dollars	32,183,000
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MAIN STRUCTURE			
	QUANTITY	UNIT RATE	COST
Structure	260 te	17,897	4,653,000
Guide base	4	360,911	1,444,000
Protection structure	0 te	15,393	0
Piles	55 te	4,807	264,000
Total Main structure			\$ 6,361,000

XMAS TREES			
	QUANTITY	UNIT RATE	COST
Production	4	3,478,800	13,915,000
Water injection	0	3,261,100	0
Gas injection	0	3,261,100	0
Total Xmas trees			\$ 13,915,000

MANIFOLDING (PIPING & VALVES)			
	QUANTITY	UNIT RATE	COST
Production	9 te	91,653	825,000
Test	4 te	91,653	367,000
Total Manifolding (piping & valves)			\$ 1,192,000

MULTIPHASE METERING			
	QUANTITY	UNIT RATE	COST
Multiphase meters (0-8 Mbbbl/day)	0	758,967	0
Multiphase meters (8-30 Mbbbl/day)	0	1,065,002	0
Multiphase meters (30-75 Mbbbl/day)	0	1,517,934	0
Total Multiphase metering			\$ 0

CONNECTORS / HUBS			
	QUANTITY	UNIT RATE	COST
8 in connectors	1	287,673	288,000
10 in connectors	1	336,639	337,000
Umbilical connectors	1	119,966	120,000
Hydraulic connectors	2	42,845	86,000
Electrical connectors	4	21,178	85,000
Total Connectors / hubs			\$ 916,000

FLYING LEADS			
	QUANTITY	UNIT RATE	COST
Hydraulic (x 1)	50 m	2,399	120,000
Electrical (x 2)	100 m	66	7,000
Total Flying leads			\$ 127,000

CONTROL AND TESTING			
	QUANTITY	UNIT RATE	COST
Subsea distribution unit	1	3,427,592	3,428,000
Subsea controls	4	1,481,209	5,925,000
System testing	4	79,820	319,000
Total Control and testing			\$ 9,672,000

Cable

TOTAL COST	US Dollars	66,800,000
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PRODUCTION FLOWLINE			
	QUANTITY	UNIT RATE	COST
Linepipe - 1 x (D = 20 in, t = 22 mm, Carbon steel X60)	40.00 km	356,800	14,272,000
Coating	40.00 km	47,700	1,908,000
Anodes	63.80 te	8,600	549,000
Subsea crossings	0	563,100	0
PLETs	2		4,294,000
Total Production flowline			\$ 21,023,000

TEST SERVICE FLOWLINE			
	QUANTITY	UNIT RATE	COST
Linepipe - 1 x (D = 8 in, t = 11.2 mm, Carbon steel X60)	40.00 km	100,700	4,028,000
Coating	40.00 km	23,700	948,000
Anodes	27.50 te	8,600	237,000
Subsea crossings	0	563,100	0
PLETs	2		1,760,000
Total Test service flowline			\$ 6,973,000

UMBILICALS			
	QUANTITY	UNIT RATE	COST
Control tubes			
6 x D = 25.4 mm	240.00 km	61,210	14,690,000
Chemical tubes			
6 x D = 25.4 mm	240.00 km	61,210	14,690,000
Electrical signal cable			
4 x XSA = 2.5 mm ²	160.00 km	7,720	1,235,000
Power cable			
2 x XSA = 25 mm ²	80.00 km	88,140	7,051,000
UTA	1	1,138,450	1,138,000
Total Umbilicals			\$ 38,804,000

G.2 One ITS and riser base.

Subsea development

	Well count	Flow	Type
Production	0	52.8 Mbbl/day	
Water injection	0	64.5 Mbbl/day	
Gas injection	0	118 MMscf/day	
Spare slots	0		-

	Present	Flow rate
Multiphase metering	No	
Test	Yes	
Gas lift	No	-
Chemical injection	No	

Subsea input report
Component name

Weather downtime	0.0	1.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.3
Mob/demob	0.0	8.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	8.0
Total	0.0	20.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0

Subsea components	Type	Water depth	Production wells	Production flow per well	Water injection wells	Water injection wells	Water injection flow per	Gas injection wells	Gas injection flow per well	Test service
4-well slots ITS	Template manifold	350 m	4	0 Mbb/day	0	0	0 Mbb/day	0	0	0 MMsct/day

Subsea components	Gas lift	Chemical injection	HIPPS	MFM on test service	MFM on production	MFM on wellheads	Spare slots	Spare slots type	Total slots count
4-well slots ITS	No	No	No	No	No	No	0	-	4

Subsea components	SDU selected	SDU wells serviced	SDU hydraulic flying	SDU electrical flying leads	UTA wells serviced	UTA hydraulic flying leads	UTA electrical flying leads
4-well slots ITS	Yes	4	0	0	4	1	2

Riser base manifold	Water depth	Termination type	Riser systems	Riser length
Riser base 01	350 m	Flexible lazy S	0	0 m

Subsea development

Name	Subsea development
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TOTAL COST	US Dollars	59,875,000
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EQUIPMENT			Procured from: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
4-well slots ITS			29,423,000
Riser base 01			213,000
Platform controls - main	1	471,000	471,000
Platform controls - additional	0	68,700	0
Sub Total			30,107,000
Freight	3.00%		903,000
Total Equipment			\$ 31,010,000

INSTALLATION			Location: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Pipelay spread (reel-lay)	0 day	195,862	0
Diving support vessel tie ins	21 day	249,725	5,244,000
Diving support vessel test & commissioning	0 day	249,725	0
Testing & commissioning equipment	0 day	19,097	0
Semi-submersible crane vessel	0 day	355,001	0
Semi-submersible drilling vessel	0 day	141,021	0
Trench vessel	0 day	147,019	0
Survey vessel	0 day	99,155	0
Dredge vessel	0 day	226,466	0
Rock install vessel	0 day	97,931	0
Supply vessel	10 day	41,743	417,000
Total Installation			\$ 5,661,000

DESIGN & PROJECT MANAGEMENT			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Design	37,600 mhr	176	6,618,000
Project management	13,100 mhr	323	4,231,000
Total Design & Project management			\$ 10,849,000

INSURANCE & CERTIFICATION			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Certification	1.00%		475,000
Insurance	4.00%		1,901,000
Total Insurance & Certification			\$ 2,376,000

CONTINGENCY			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Contingency	20.00%		9,979,000
Total Contingency			\$ 9,979,000

4-well slots ITS

TOTAL COST	US Dollars	29,423,000
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MAIN STRUCTURE			
	QUANTITY	UNIT RATE	COST
Structure	260 te	17,897	4,653,000
Guide base	4	360,911	1,444,000
Protection structure	0 te	15,393	0
Piles	55 te	4,807	264,000
Total Main structure			\$ 6,361,000

XMAS TREES			
	QUANTITY	UNIT RATE	COST
Production	4	3,135,600	12,542,000
Water injection	0	3,261,100	0
Gas injection	0	3,261,100	0
Total Xmas trees			\$ 12,542,000

MANIFOLDING (PIPING & VALVES)			
	QUANTITY	UNIT RATE	COST
Production	4 te	91,653	367,000
Test	2 te	91,653	183,000
Total Manifolding (piping & valves)			\$ 550,000

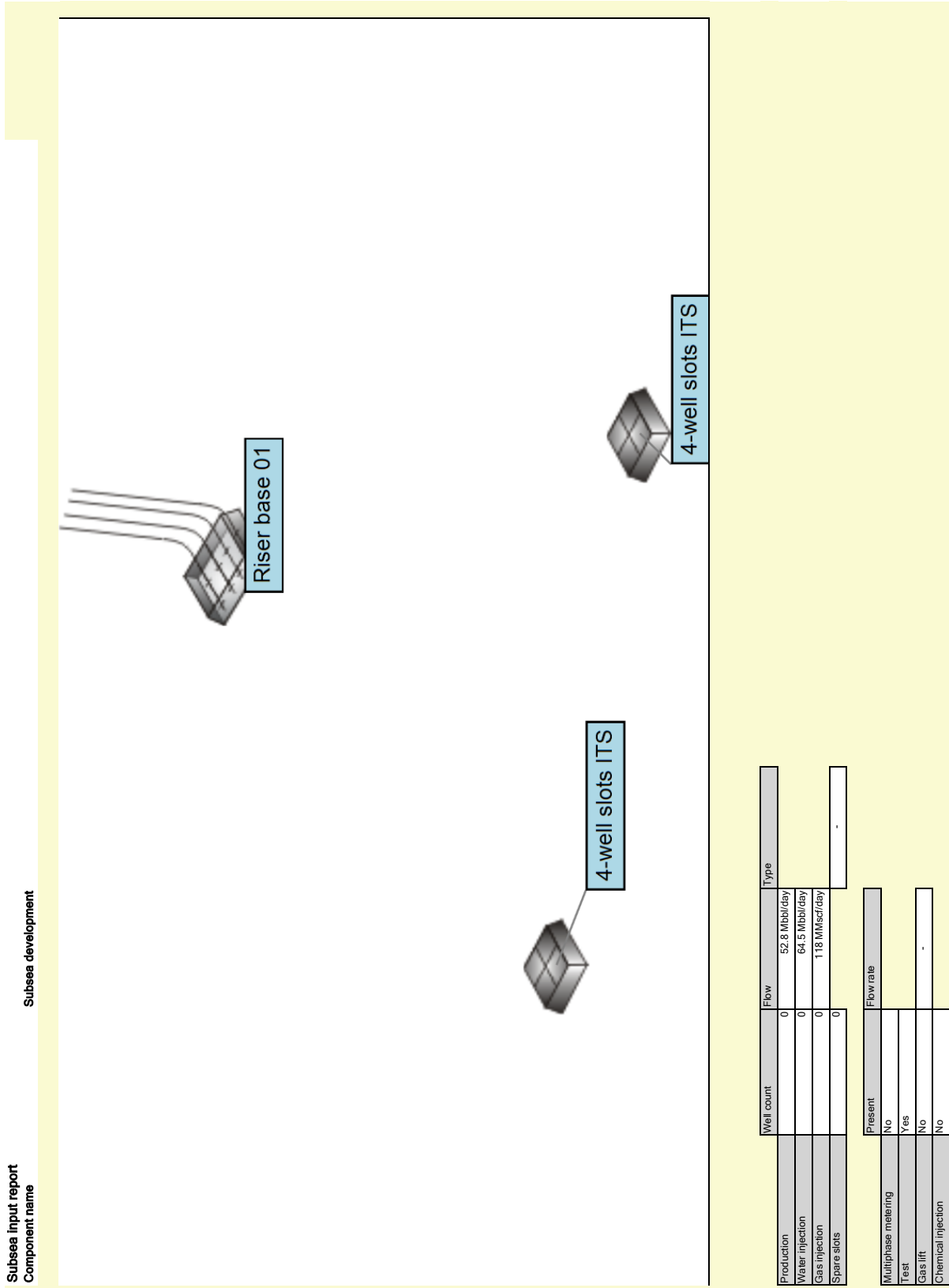
MULTIPHASE METERING			
	QUANTITY	UNIT RATE	COST
Multiphase meters (0-8 Mbbbl/day)	0	758,967	0
Multiphase meters (8-30 Mbbbl/day)	0	1,065,002	0
Multiphase meters (30-75 Mbbbl/day)	0	1,517,934	0
Total Multiphase metering			\$ 0

CONNECTORS / HUBS			
	QUANTITY	UNIT RATE	COST
Umbilical connectors	0	119,966	0
Hydraulic connectors	2	42,845	86,000
Electrical connectors	4	21,178	85,000
Total Connectors / hubs			\$ 171,000

FLYING LEADS			
	QUANTITY	UNIT RATE	COST
Hydraulic (x 1)	50 m	2,399	120,000
Electrical (x 2)	100 m	66	7,000
Total Flying leads			\$ 127,000

CONTROL AND TESTING			
	QUANTITY	UNIT RATE	COST
Subsea distribution unit	1	3,427,592	3,428,000
Subsea controls	4	1,481,209	5,925,000
System testing	4	79,820	319,000
Total Control and testing			\$ 9,672,000

G.3 Two integrated template structures and riser base.



Subsea input report
Component name

Weather downtime	0.0	2.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.5
Mob/demob	0.0	8.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	8.0
Total	0.0	28.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	11.8

Subsea components	Type	Water depth	Production wells	Production flow per well	Water injection wells	Water injection flow per well	Gas injection wells	Gas injection flow per well	Test service
4-well slots ITS	Template manifold	350 m	4	0 Mbb/day	0	0 Mbb/day	0	0 Mbb/day	Yes
4-well slots ITS	Template manifold	350 m	4	0 Mbb/day	0	0 Mbb/day	0	0 Mbb/day	Yes

Subsea components	Gas lift	Chemical injection	HIPPS	MFM on test service	MFM on production	MFM on wellheads	Spare slots	Spare slots type	Total slots count
4-well slots ITS	No	No	No	No	No	No	0	-	4
4-well slots ITS	No	No	No	No	No	No	0	-	4

Subsea components	SDU selected	SDU wells serviced	SDU hydraulic flying	SDU electrical flying leads	UTA wells serviced	UTA hydraulic flying leads	UTA electrical flying leads
4-well slots ITS	Yes	4	0	0	4	1	2
4-well slots ITS	Yes	4	0	0	4	1	2

Riser base manifold	Water depth	Termination type	Sub-type	Riser systems	Riser length
Riser base 01	350 m	Riser	Flexible lazy S	0	0 m

Subsea development

Name	Subsea development
------	--------------------

TOTAL COST	US Dollars	104,542,000
------------	------------	--------------------

EQUIPMENT			Procured from: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
4-well slots ITS			29,423,000
4-well slots ITS			29,423,000
Riser base 01			213,000
Platform controls - main	1	471,000	471,000
Platform controls - additional	0	68,700	0
Sub Total			59,530,000
Freight	3.00%		1,786,000
Total Equipment			\$ 61,316,000

INSTALLATION			Location: N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Pipelay spread (reel-lay)	0 day	195,862	0
Diving support vessel tie ins	29 day	249,725	7,242,000
Diving support vessel test & commissioning	0 day	249,725	0
Testing & commissioning equipment	0 day	19,097	0
Semi-submersible crane vessel	0 day	355,001	0
Semi-submersible drilling vessel	0 day	141,021	0
Trench vessel	0 day	147,019	0
Survey vessel	0 day	99,155	0
Dredge vessel	0 day	226,466	0
Rock install vessel	0 day	97,931	0
Supply vessel	12 day	41,743	501,000
Total Installation			\$ 7,743,000

DESIGN & PROJECT MANAGEMENT			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Design	49,300 mhr	176	8,677,000
Project management	16,200 mhr	323	5,233,000
Total Design & Project management			\$ 13,910,000

INSURANCE & CERTIFICATION			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Certification	1.00%		830,000
Insurance	4.00%		3,319,000
Total Insurance & Certification			\$ 4,149,000

CONTINGENCY			N. North Sea (Norway)
	QUANTITY	UNIT RATE	COST
Contingency	20.00%		17,424,000
Total Contingency			\$ 17,424,000

4-well slots ITS

TOTAL COST	US Dollars	29,423,000
------------	------------	------------

MAIN STRUCTURE			
	QUANTITY	UNIT RATE	COST
Structure	260 te	17,897	4,653,000
Guide base	4	360,911	1,444,000
Protection structure	0 te	15,393	0
Piles	55 te	4,807	264,000
Total Main structure			\$ 6,361,000

XMAS TREES			
	QUANTITY	UNIT RATE	COST
Production	4	3,135,600	12,542,000
Water injection	0	3,261,100	0
Gas injection	0	3,261,100	0
Total Xmas trees			\$ 12,542,000

MANIFOLDING (PIPING & VALVES)			
	QUANTITY	UNIT RATE	COST
Production	4 te	91,653	367,000
Test	2 te	91,653	183,000
Total Manifolding (piping & valves)			\$ 550,000

MULTIPHASE METERING			
	QUANTITY	UNIT RATE	COST
Multiphase meters (0-8 Mbbbl/day)	0	758,967	0
Multiphase meters (8-30 Mbbbl/day)	0	1,065,002	0
Multiphase meters (30-75 Mbbbl/day)	0	1,517,934	0
Total Multiphase metering			\$ 0

CONNECTORS / HUBS			
	QUANTITY	UNIT RATE	COST
Umbilical connectors	0	119,966	0
Hydraulic connectors	2	42,845	86,000
Electrical connectors	4	21,178	85,000
Total Connectors / hubs			\$ 171,000

FLYING LEADS			
	QUANTITY	UNIT RATE	COST
Hydraulic (x 1)	50 m	2,399	120,000
Electrical (x 2)	100 m	66	7,000
Total Flying leads			\$ 127,000

CONTROL AND TESTING			
	QUANTITY	UNIT RATE	COST
Subsea distribution unit	1	3,427,592	3,428,000
Subsea controls	4	1,481,209	5,925,000
System testing	4	79,820	319,000
Total Control and testing			\$ 9,672,000

4-well slots ITS

TOTAL COST	US Dollars	29,423,000
------------	------------	------------

MAIN STRUCTURE			
	QUANTITY	UNIT RATE	COST
Structure	260 te	17,897	4,653,000
Guide base	4	360,911	1,444,000
Protection structure	0 te	15,393	0
Piles	55 te	4,807	264,000
Total Main structure			\$ 6,361,000

XMAS TREES			
	QUANTITY	UNIT RATE	COST
Production	4	3,135,600	12,542,000
Water injection	0	3,261,100	0
Gas injection	0	3,261,100	0
Total Xmas trees			\$ 12,542,000

MANIFOLDING (PIPING & VALVES)			
	QUANTITY	UNIT RATE	COST
Production	4 te	91,653	367,000
Test	2 te	91,653	183,000
Total Manifolding (piping & valves)			\$ 550,000

MULTIPHASE METERING			
	QUANTITY	UNIT RATE	COST
Multiphase meters (0-8 Mbbbl/day)	0	758,967	0
Multiphase meters (8-30 Mbbbl/day)	0	1,065,002	0
Multiphase meters (30-75 Mbbbl/day)	0	1,517,934	0
Total Multiphase metering			\$ 0

CONNECTORS / HUBS			
	QUANTITY	UNIT RATE	COST
Umbilical connectors	0	119,966	0
Hydraulic connectors	2	42,845	86,000
Electrical connectors	4	21,178	85,000
Total Connectors / hubs			\$ 171,000

FLYING LEADS			
	QUANTITY	UNIT RATE	COST
Hydraulic (x 1)	50 m	2,399	120,000
Electrical (x 2)	100 m	66	7,000
Total Flying leads			\$ 127,000

CONTROL AND TESTING			
	QUANTITY	UNIT RATE	COST
Subsea distribution unit	1	3,427,592	3,428,000
Subsea controls	4	1,481,209	5,925,000
System testing	4	79,820	319,000
Total Control and testing			\$ 9,672,000

Appendix H

Cost Estimates

H.1 Part list sent to STEP-G

Profiles for extrusion:

Profile type 1:

Flange thickness: T_f [mm]

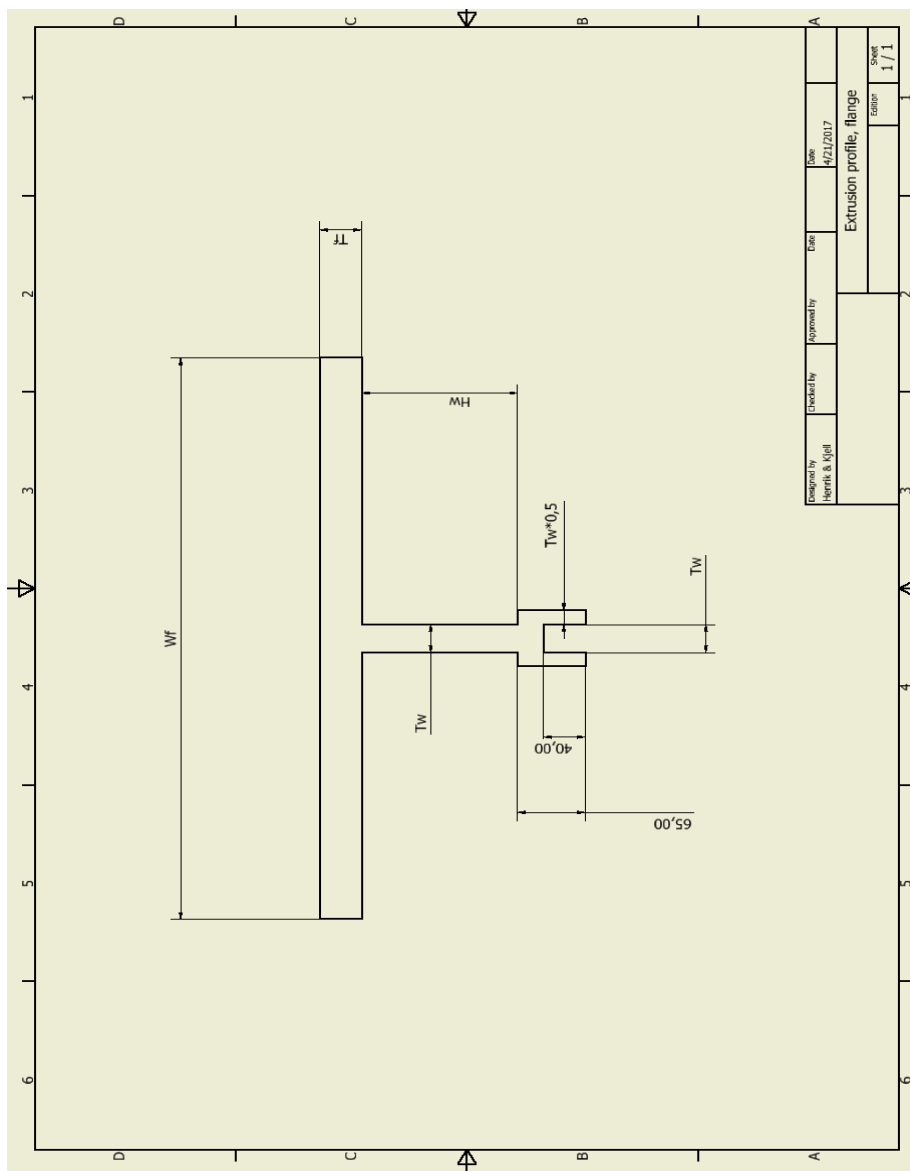
Web thickness: T_w [mm]

Flange width: W_f [mm]

Web height: H_w [mm]

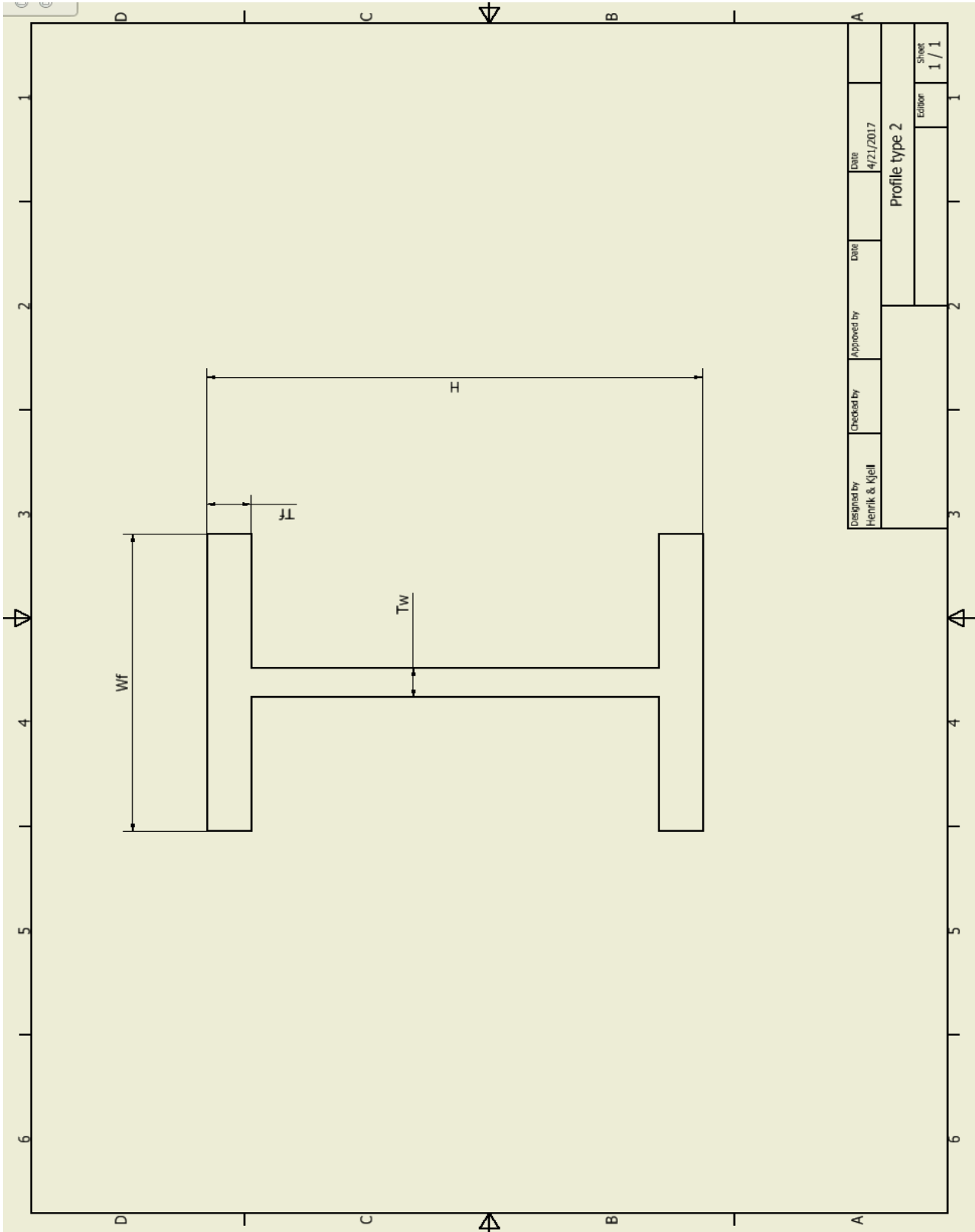
Length: L [mm]

	Beam #	T_f	T_w	W_f	H_w	L
a	1	50	60	400	310	46360
(b-c)+d	2	30	35	330	310	56400
f	3	45	40	440	310	60800
g	4	40	35	400	210	29600
h	5	40	40	350	120	54400
i+j	6	40	30	400	310	88000
s+t	7	35	20	420	200	113600



Profile type 2

	Beam #	Tf	Tw	Wf	H	L
e	8	32	20	300	350	17600
k	9	15	20	300	400	32200
w	10	35	20	300	360	21600



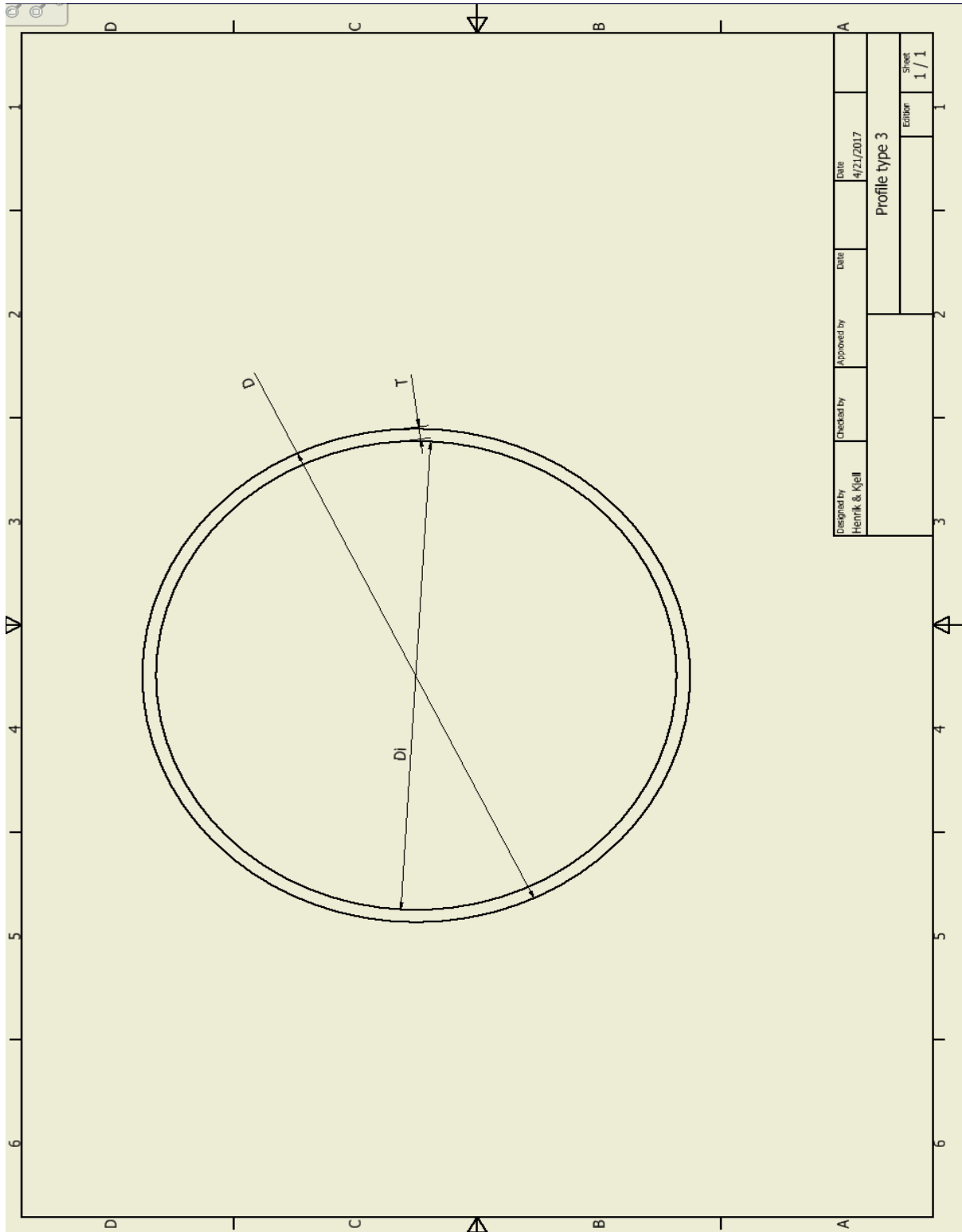
Profile type 3

Outer diameter: D [mm]

Thickness: T [mm]

Length: L [mm]

	Tube #	D	T	L
q2	1	500	25	12000
y	2	219	50	107200



Profile type 4

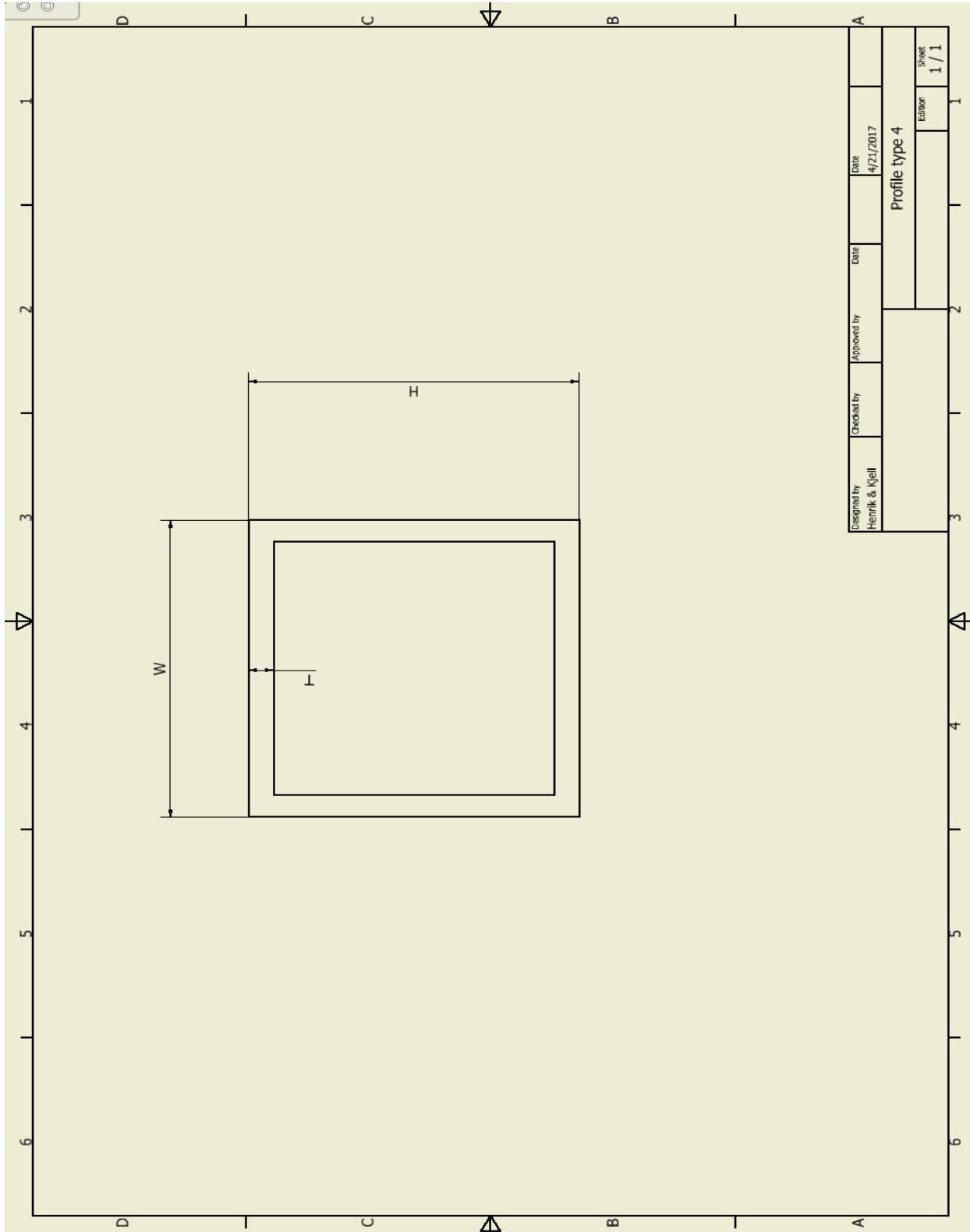
Width: W [mm]

Height: H [mm]

Thickness: T [mm]

Length: L [mm]

	Tube #	W	H	T	L
x	3	200	200	15	41600

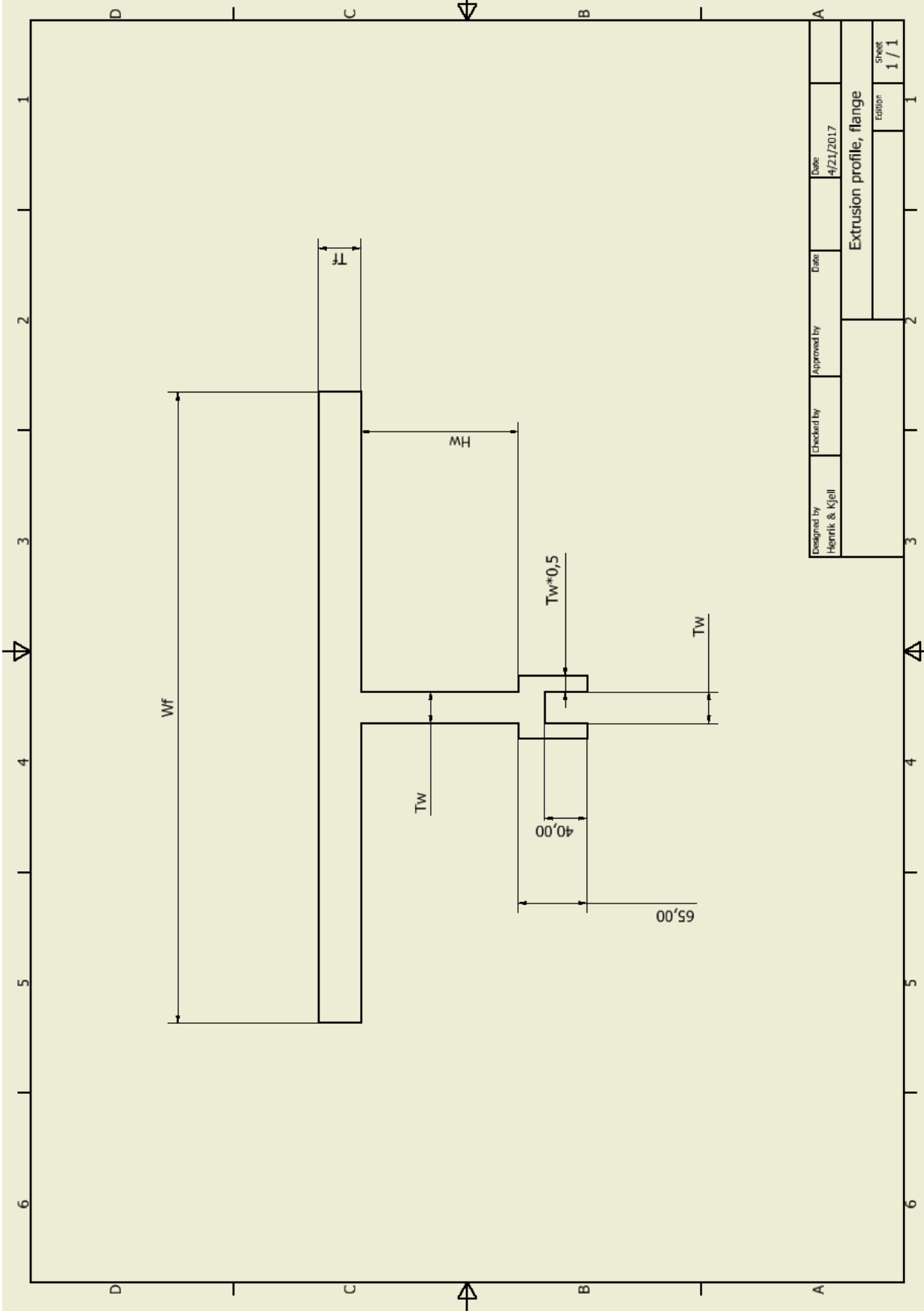


H.2 Price offer recieved from STEP-G

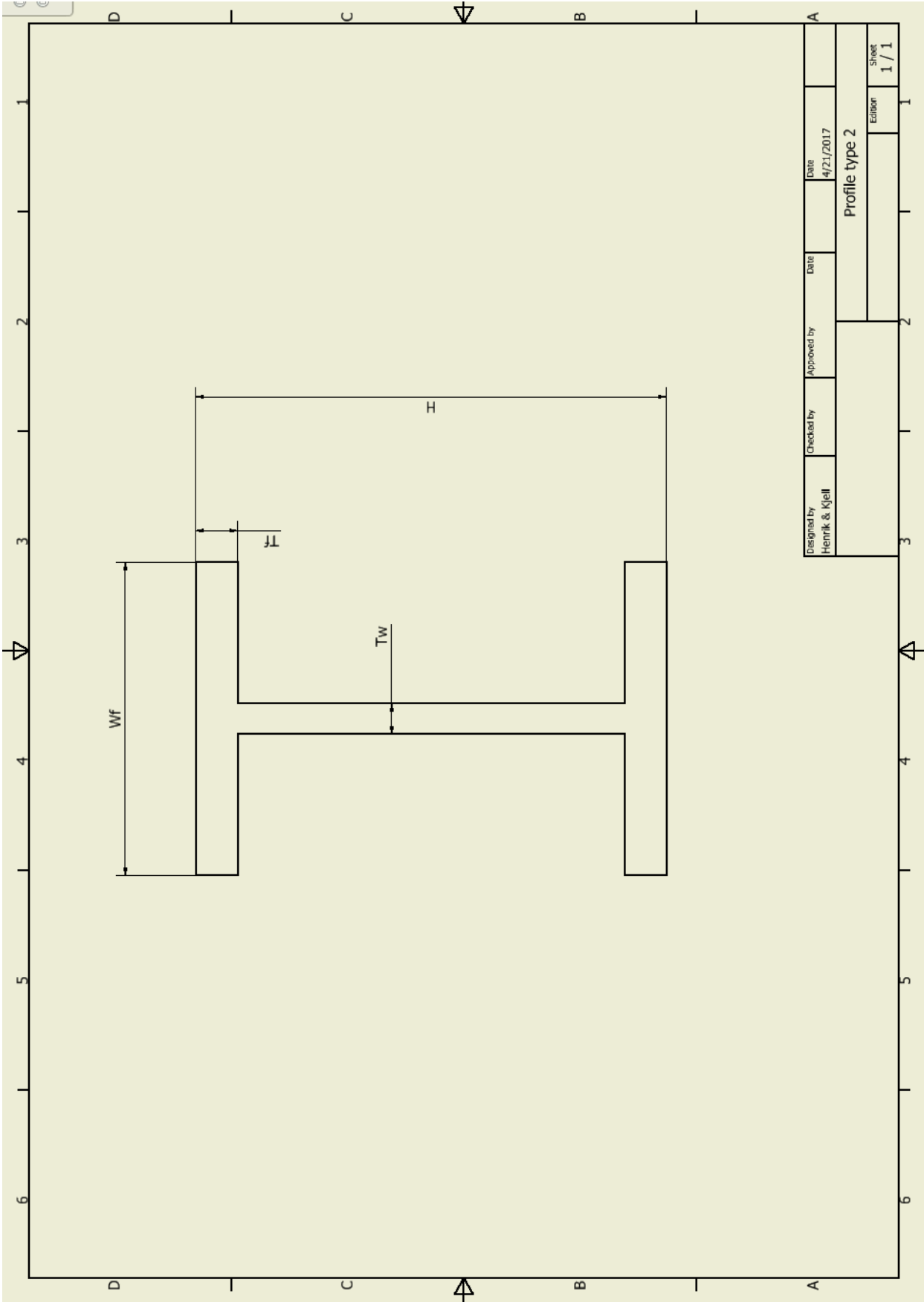
Profiles for extrusion (alloy 6082-T6):		Total cost =		385742 Euro		3,645,258 NOK		With 9,45 NOK/EUR		Price excludes VAT tax		Remarks	
Feasibility check STEP-G		Klaus Funken		5/4/2017								Standard packaging min lot/sizes to be respected Tolerances acc. EN 755-9 Alloy 6082 T6 according EN 755-2 & EN 573 max/min delivery length to be agreed just indicational offer, as details open Delivery ex works	
Profile type 1:		Flange thickness: Tf [mm]		Web thickness: Tw [mm]		Flange width: Wf [mm]		Web height: Hw [mm]		Length: L [mm]			
Beam #	Tf	Tw	Wf	Hw	L	kg/m	total weight kg	€/kg	die cost	du	comment		
a	50	60	118.8	400	46360	118.8	5508	4.61	13,000 €	548			
(b-c)+d	30	35	64.53	330	56400	64.53	3639	4.61	12,000 €	500			
f	45	40	96.66	440	60800	96.66	5877	4.61	15,000 €	578			
g	40	35	71.55	400	29600	71.55	2118	4.61	12,000 €	485			
h	40	40	60.48	350	54400	60.48	3290	4.61	6,000 €	396			
i+)	6	30	75.6	400	88000	75.6	6653	4.61	13,000 €	548			
s+t	35	20	55.35	420	113600	55.35	6288	4.61	12,000 €	497			
										with LME 1,76 and Billet prime 0,48			
										33373		83,000 €	
										236847 Euro			
										Profile 1 cost			



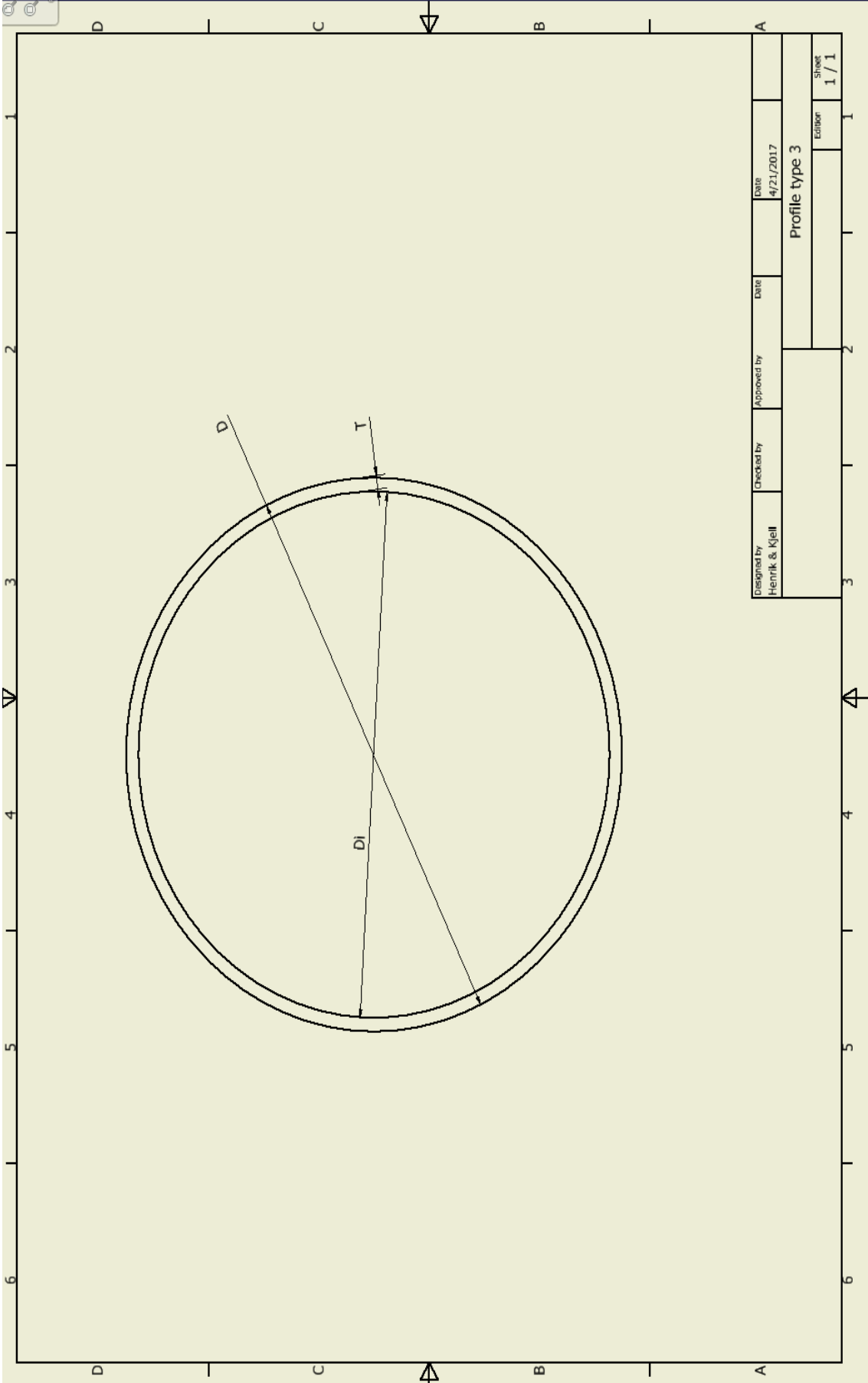
Profile-1:



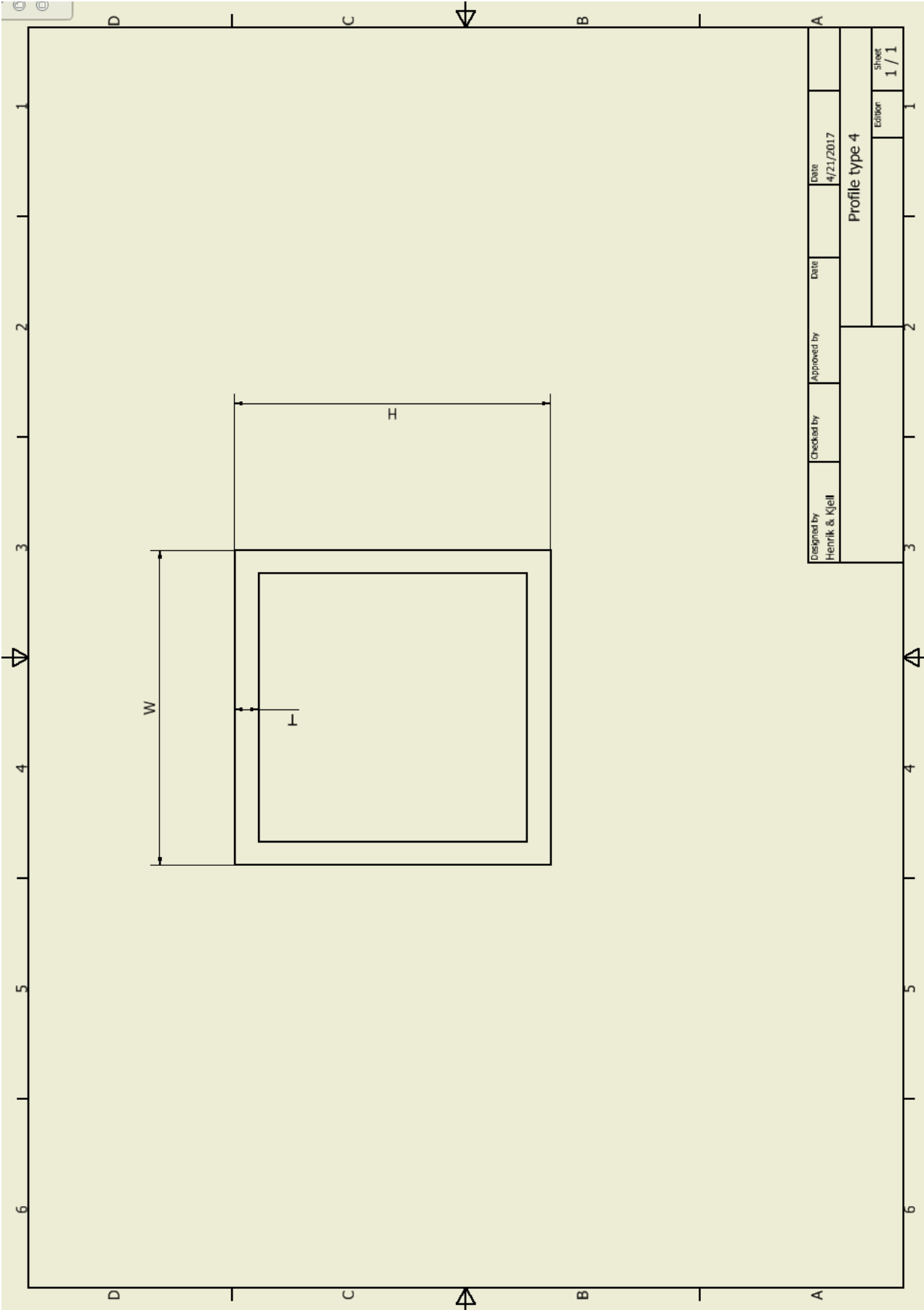
Profile-2:



Profile-3:



Profile-4:



Designed by Henrik & Kjell	Checked by	Approved by	Date 4/21/2017	Date	1
Profile type 4				Editor	1 / 1
				Sheet	1 / 1

H.3 Constellium price offer

Alloy	Temper	Norm	Certificate	Thickness	Width (mm)	Length (mm)	MOQ (nbr of pcs)	Full price with DNV certificate included (EUR/T) DAP Trondheim/Norway
5083	H116	ASTM B928	DNV	20	2500 max	7000	1 pc	3850
5083	H116	ASTM B928	DNV	25	2000	11400	1 pc	3650
5083	H116	ASTM B928	DNV	25	2500 max	13400	1 pc	3850
5083	H116	ASTM B928	DNV	25	2500 max	13400	1 pc	3850
5083	H116	ASTM B928	DNV	30	800	1100	20 pcs	3650
5083	H116	ASTM B928	DNV	35	500	7350	6 pcs	3650
5083	H116	ASTM B928	DNV	35	850	4900	4 pcs	3650
5083	H116	ASTM B928	DNV	35	850	9200	2 pcs	3650
5083	H116	ASTM B928	DNV	40	680	3360	6 pcs	3750
5083	H116	ASTM B928	DNV	40	790	3800	6 pcs	3750
5083	H116	ASTM B928	DNV	40	2500 max	10000 max	1 pc	3950
5083	H116	ASTM B928	DNV	40	2500 max	8300	1 pc	3950
5083	H116	ASTM B928	DNV	45	2050	5100	1 pc	3750
5083	H116	ASTM B928	DNV	45	2500	5100	1 pc	3950
5083	H116	ASTM B928	DNV	50	1510	1510	7 pcs	3750
5083	H116	ASTM B928	DNV	60	780	11600	1 pc	3850

5083 Plates											
Density		2700 kg/m ³									
							Full price				
	Plate #	Number of plates	H	W	T	MOQ	[EUR/T]	Limit	MOQ	Weight [kg]	Price [EUR]
a	1	2	780	11600	60	1	3850		Y	2932	11286.4752
bc	2	2	850	9200	35	2	3650		Y	1478	5394.627
d	3	2	850	4900	35	4	3650		N	787	2873.22525
f	4	8	790	3800	40	6	3750		Y	2594	9726.48
g	5	2	500	7350	35	6	3650		N	695	2535.19875
h	6	8	680	3360	40	6	3750		Y	1974	7402.752
i+j	7	4	800	11000	30	20	3650		N	2851	10406.88
l	8	4	8300	3140	40	1	3950	W=2500	Y	11259	44472.1968
m	9	4	6820	2900	25	1	3850	W=2500	Y	5340	20559.231
n	10	2	13400	2900	25	1	3850	W=2500	Y	5246	20197.485
o	11	4	2000	11400	25	1	3850		Y	6156	23700.6
q	12	24	2700	7000	40	1	3950	W=2500		48989	193505.76
q1	13	32	3000	7000	20	1	3850	W=2500	Y	36288	139708.8
q3	14	4	3800	5100	45	1	3950	W=2500	Y	9419	37203.786
q3-2	15	4	1300	5100	45	1	3750		Y	3222	12083.175
z	16	8	1510	1510	50	7	3750		Y	2463	9234.405
									Total	141691	550291
		EUR to NOK	9.45								
		Total price [NOK]	5200251								
		Cost includes DNV certificate and delivery to Trondheim/Norway									

Figure H.1: First table with bold letters received from Constellium, cost estimates in the second table are based on numbers in the first table (received from Constellium).

Appendix I

FEM Analysis

I.1 Case-A

I.1.1 Without convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/26/2017, 9:26 AM
Study Author:	Henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	Henrikwn

☐ Project

Part Number	Assembly
Designer	Henrikwn

☐ Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-a

General objective and settings:

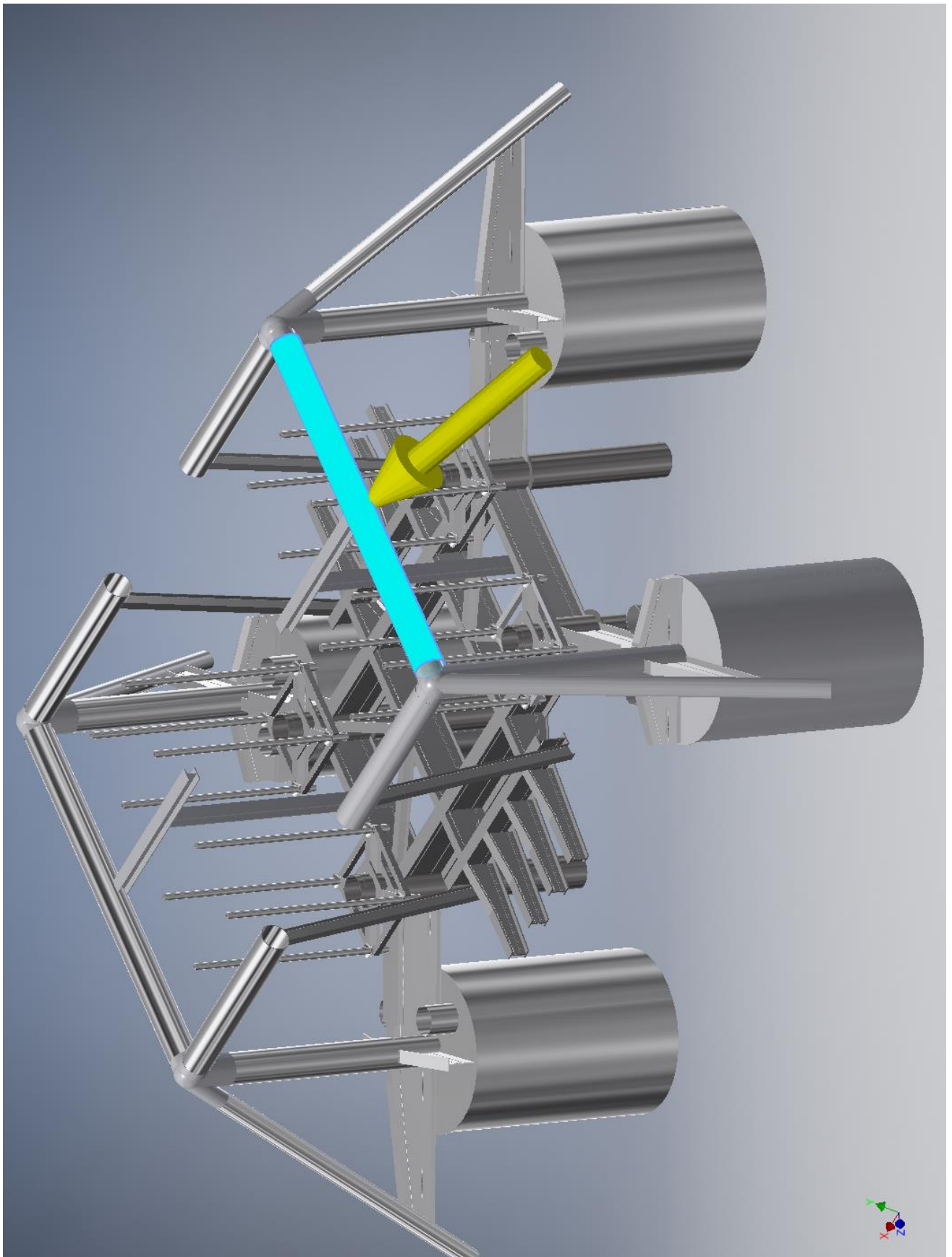
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/26/2017, 9:17 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.000 N
Vector X	481601.661 N
Vector Y	554956.253 N
Vector Z	678294.477 N

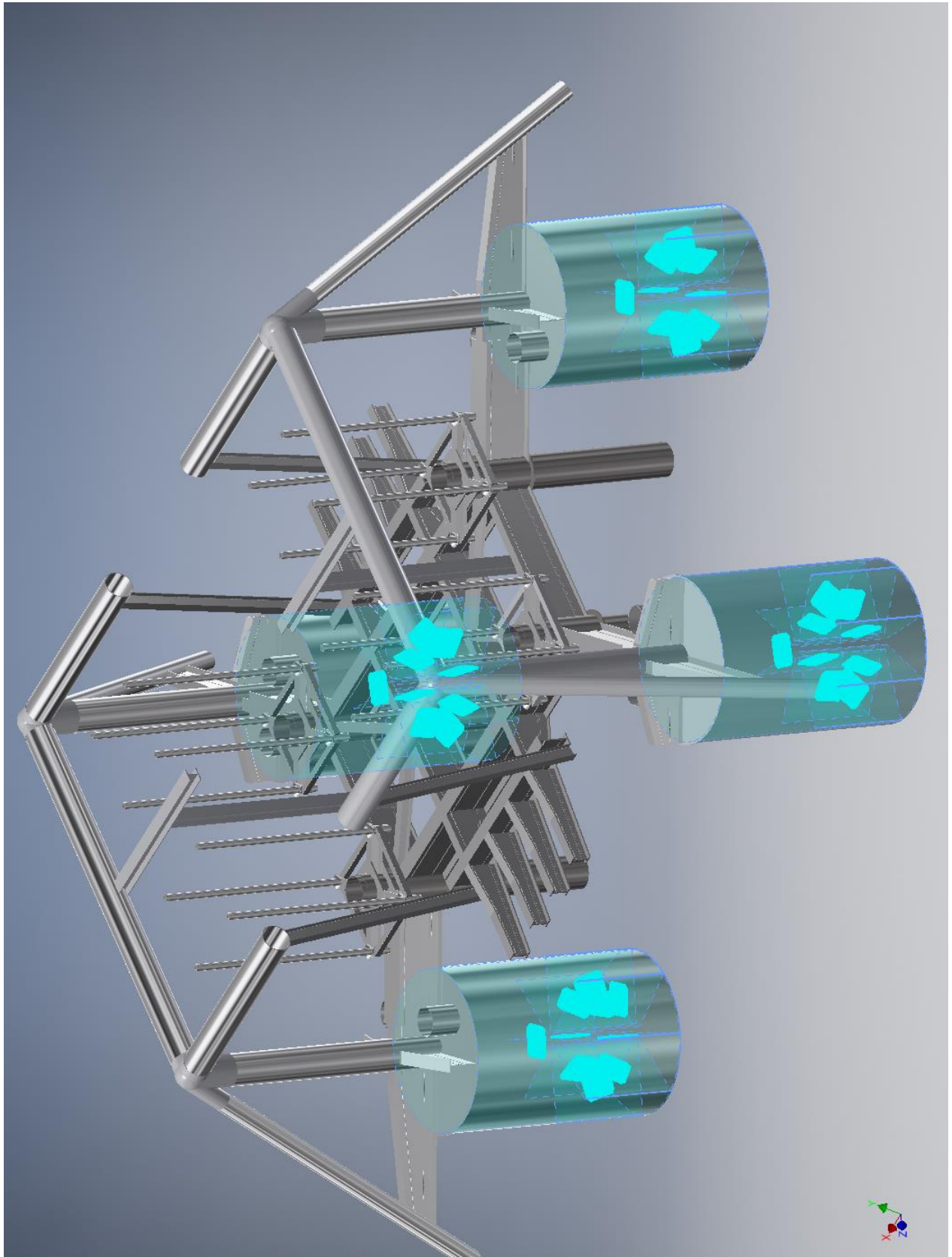
☐ Selected Face(s)



☐ **Fixed Constraint:1**

Constraint Type Fixed Constraint

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

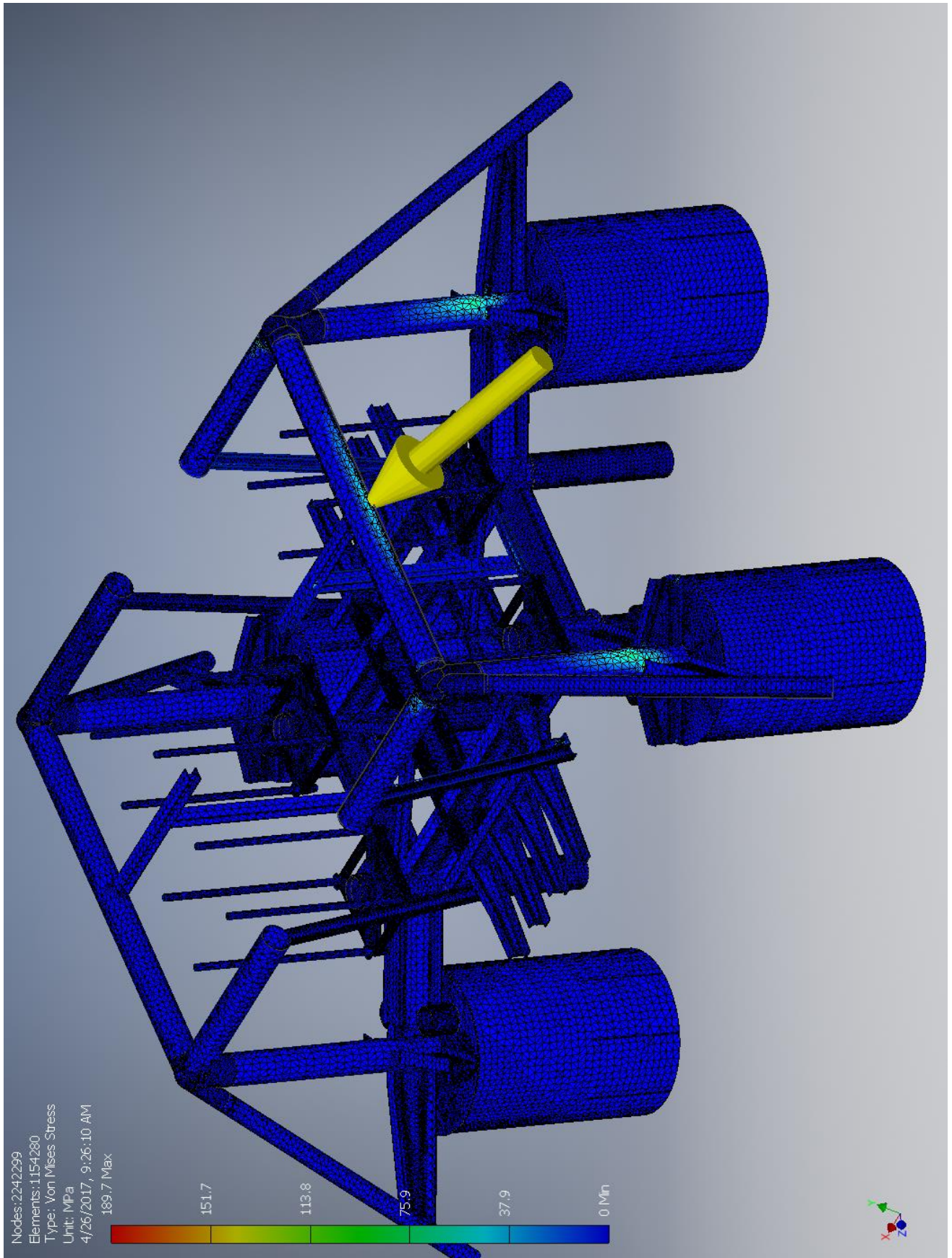
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1000000 N	-481602 N	15329500 N m	-10128600 N m
		-554956 N		11318400 N m
		-678295 N		-2072920 N m

☐ Result Summary

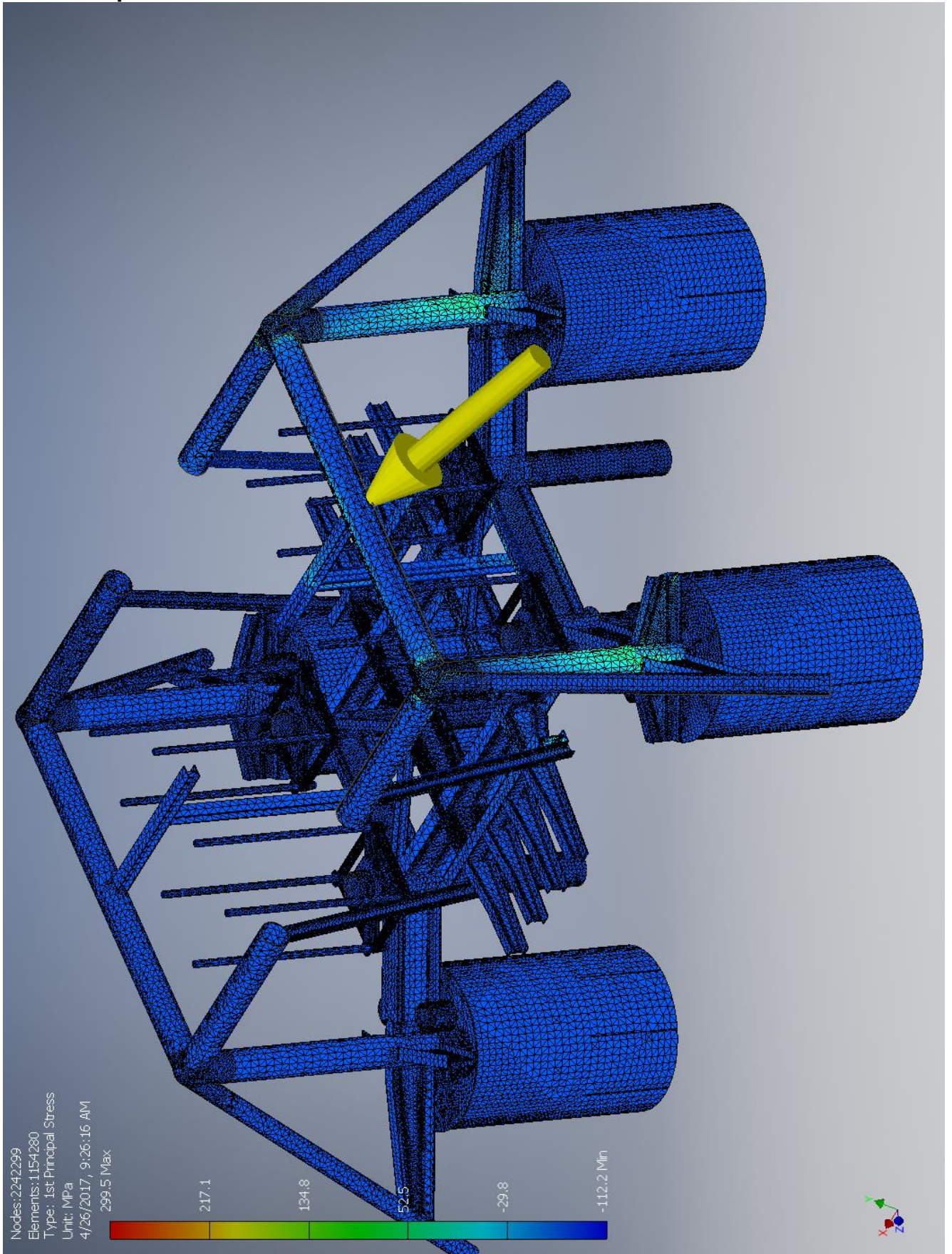
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	189.657 MPa
1st Principal Stress	-112.152 MPa	299.466 MPa
3rd Principal Stress	-264.441 MPa	119.904 MPa
Displacement	0 mm	93.2757 mm
Safety Factor	1.13362 ul	15 ul

☐ Figures

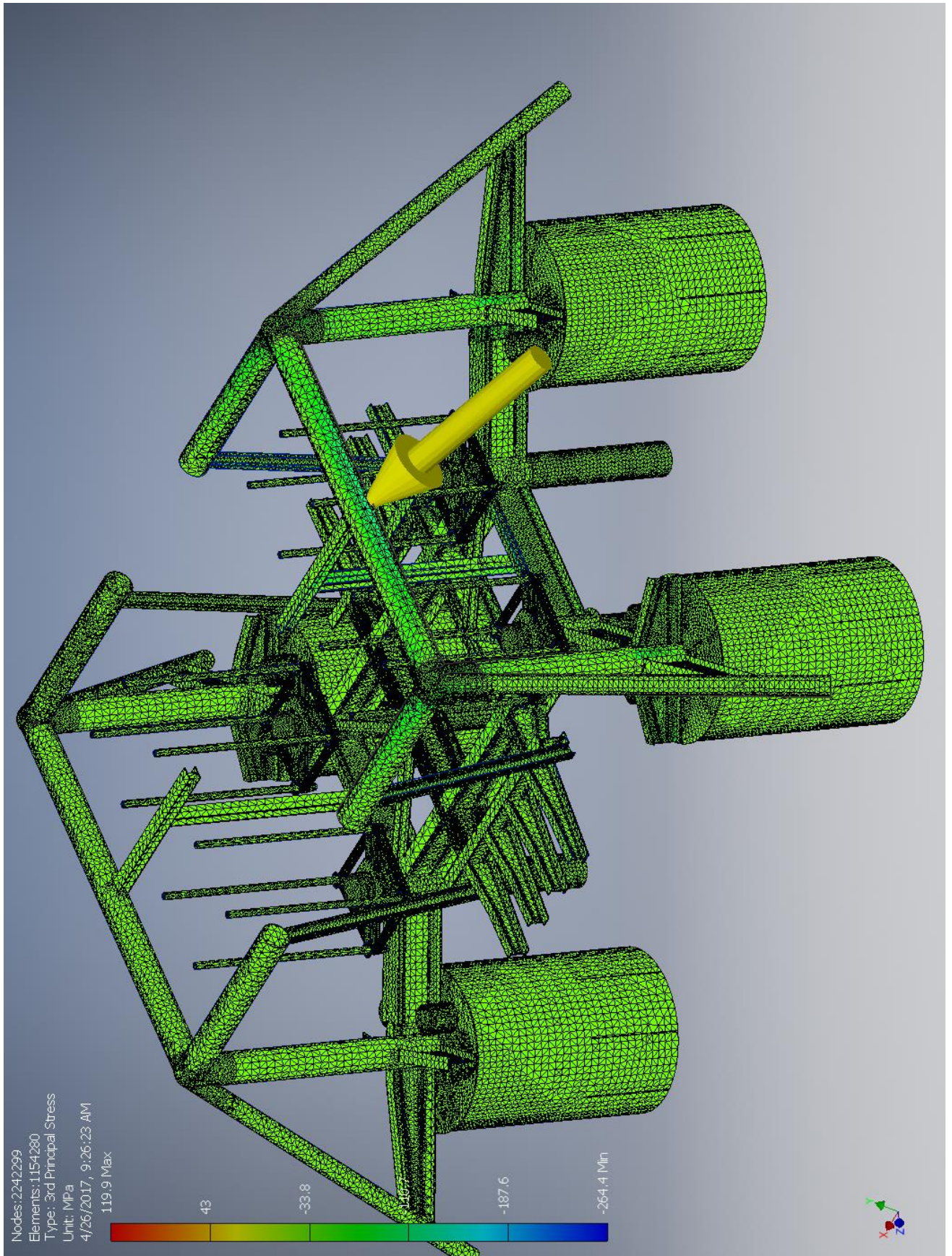
☐ Von Mises Stress



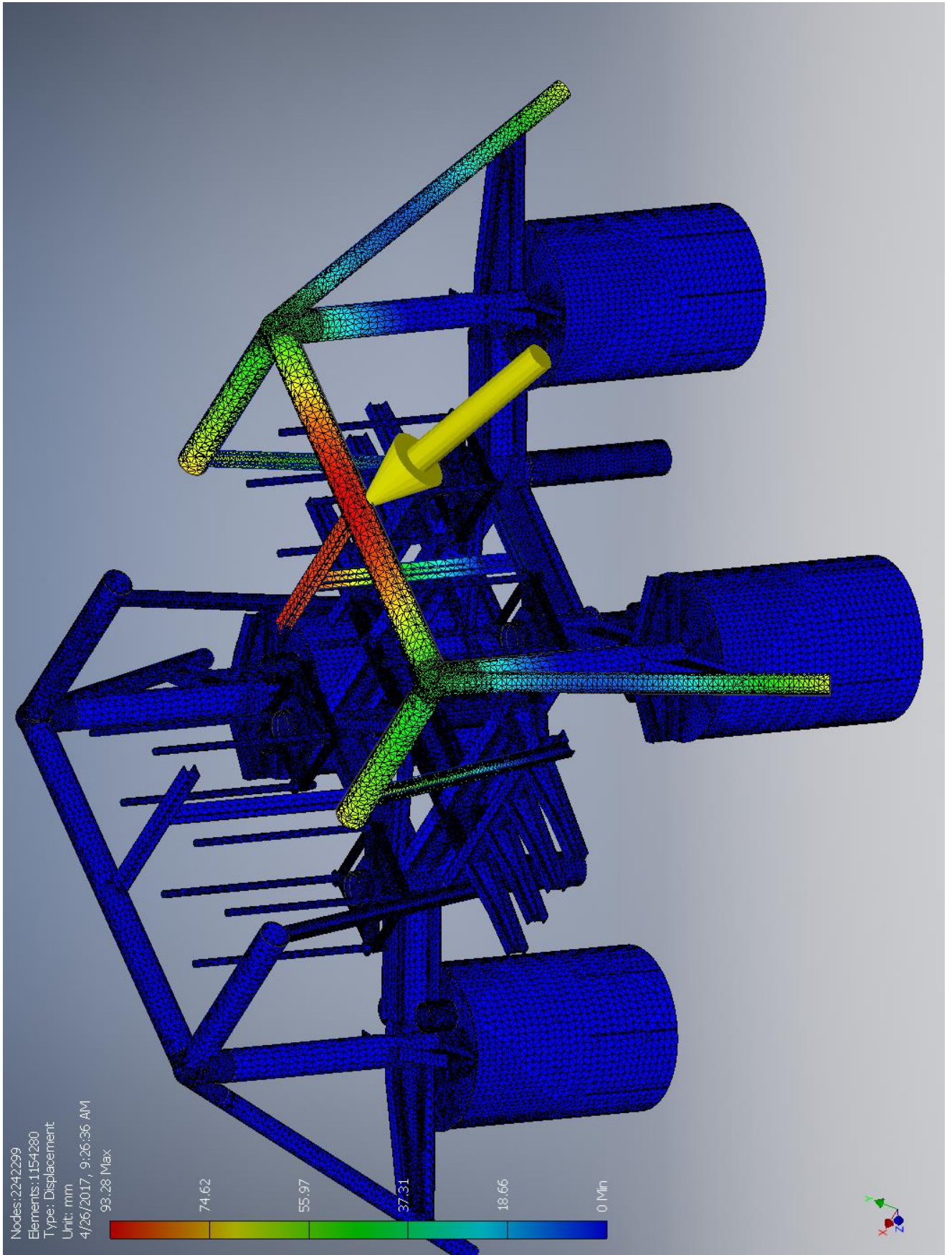
☐ 1st Principal Stress



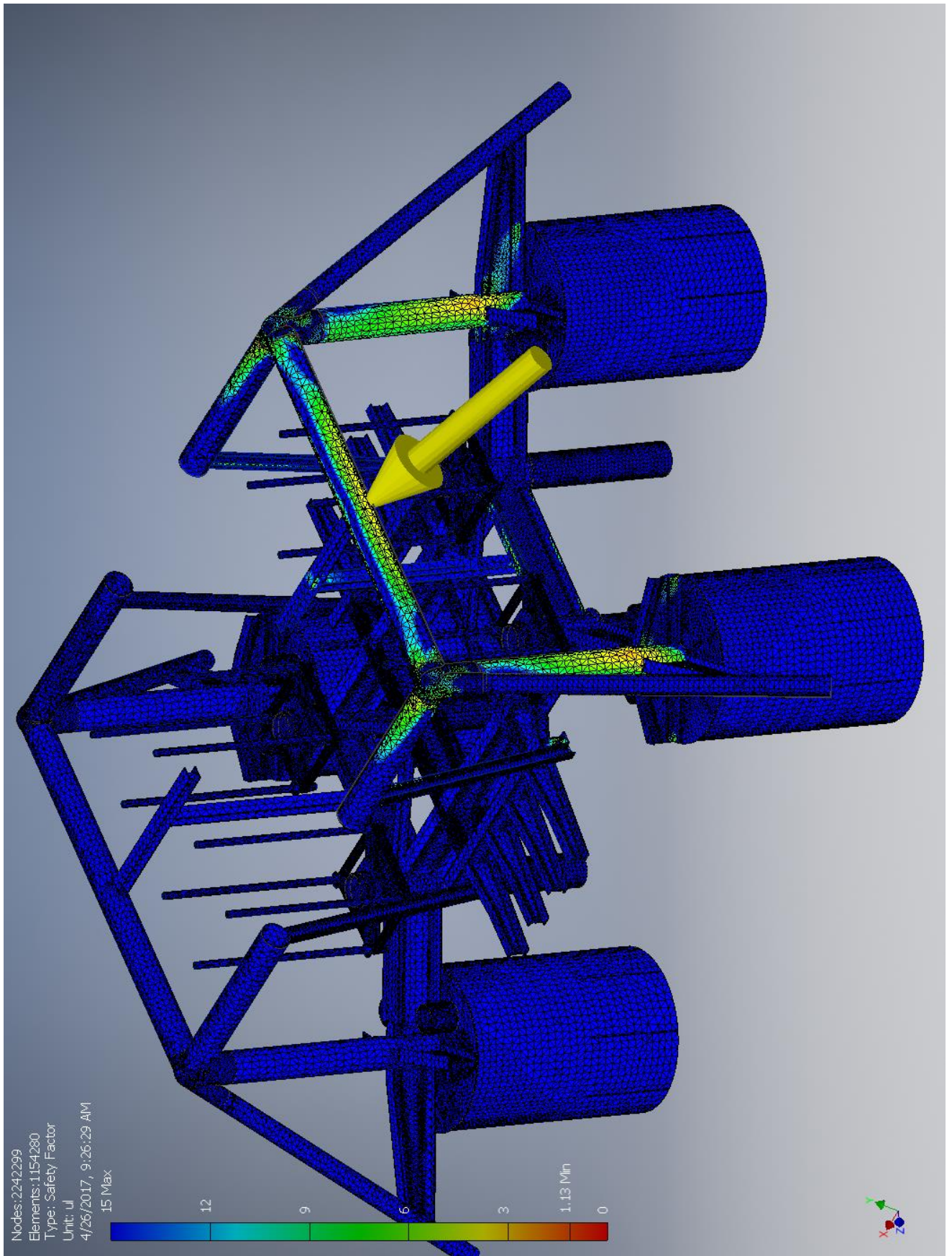
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.1.2 With convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/25/2017, 4:50 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Part Number	Assembly
Designer	henrikwn

☐ Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-a

General objective and settings:

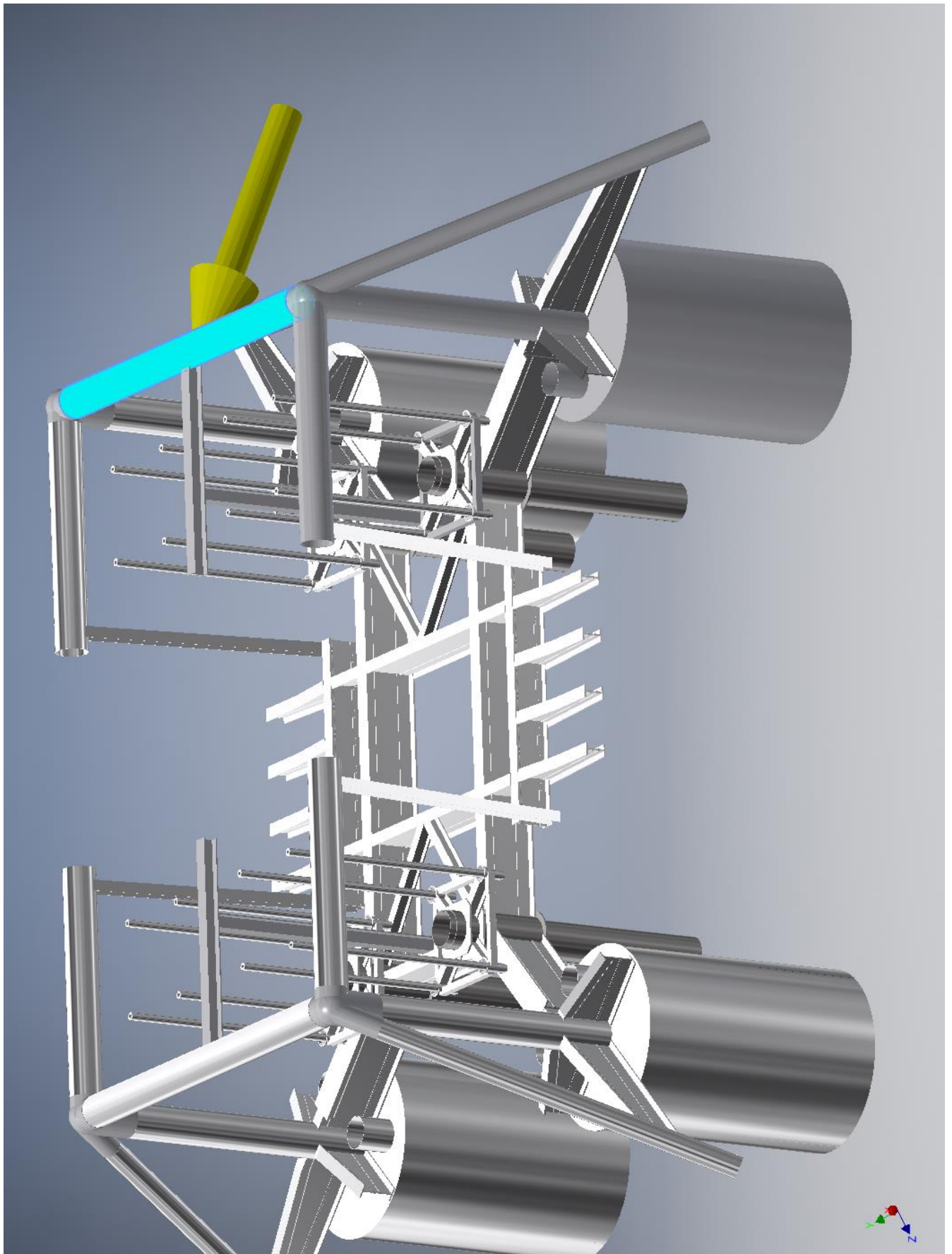
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/25/2017, 4:47 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.000 N
Vector X	481601.661 N
Vector Y	554956.253 N
Vector Z	678294.477 N

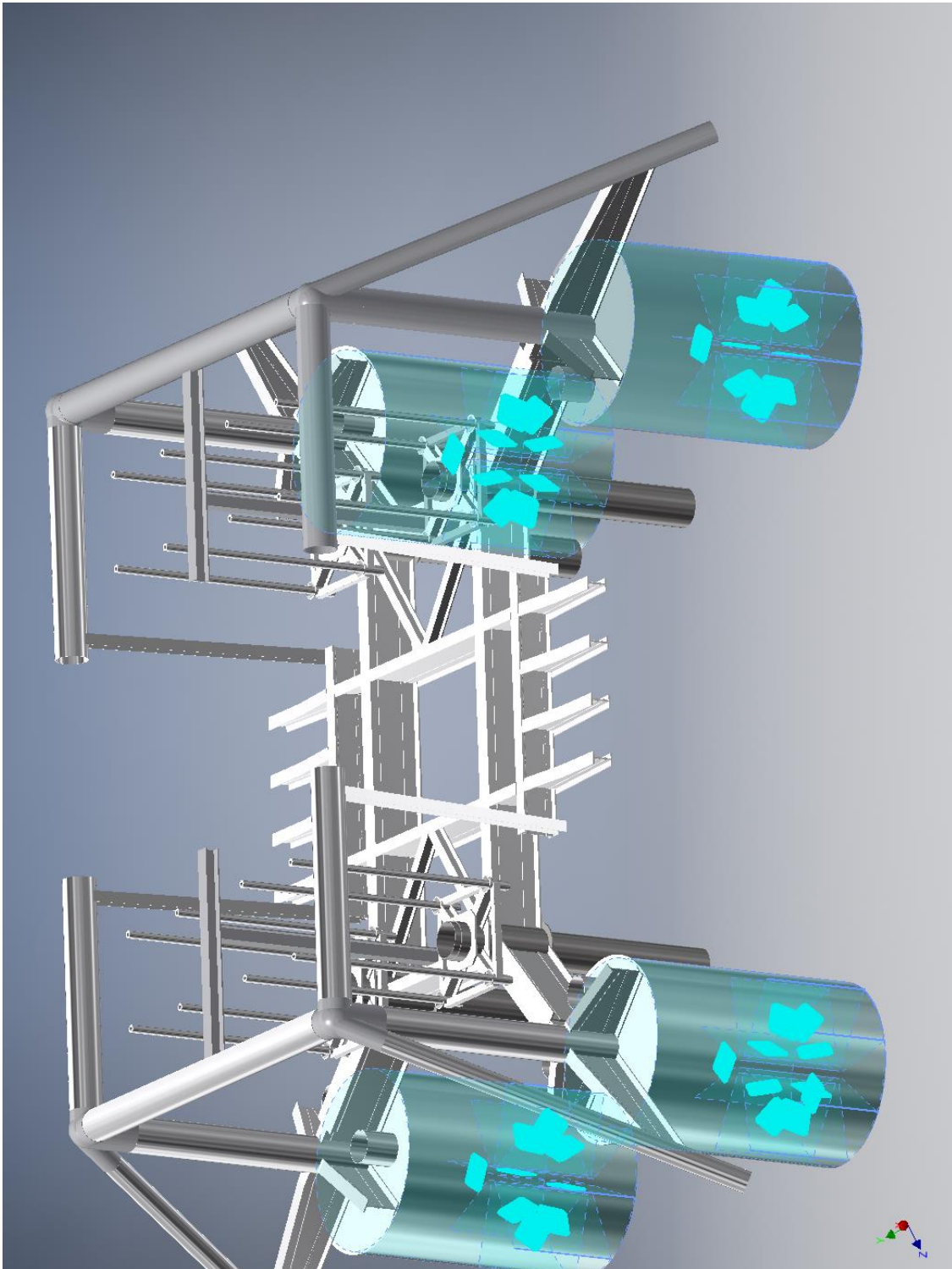
☐ Selected Face(s)



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

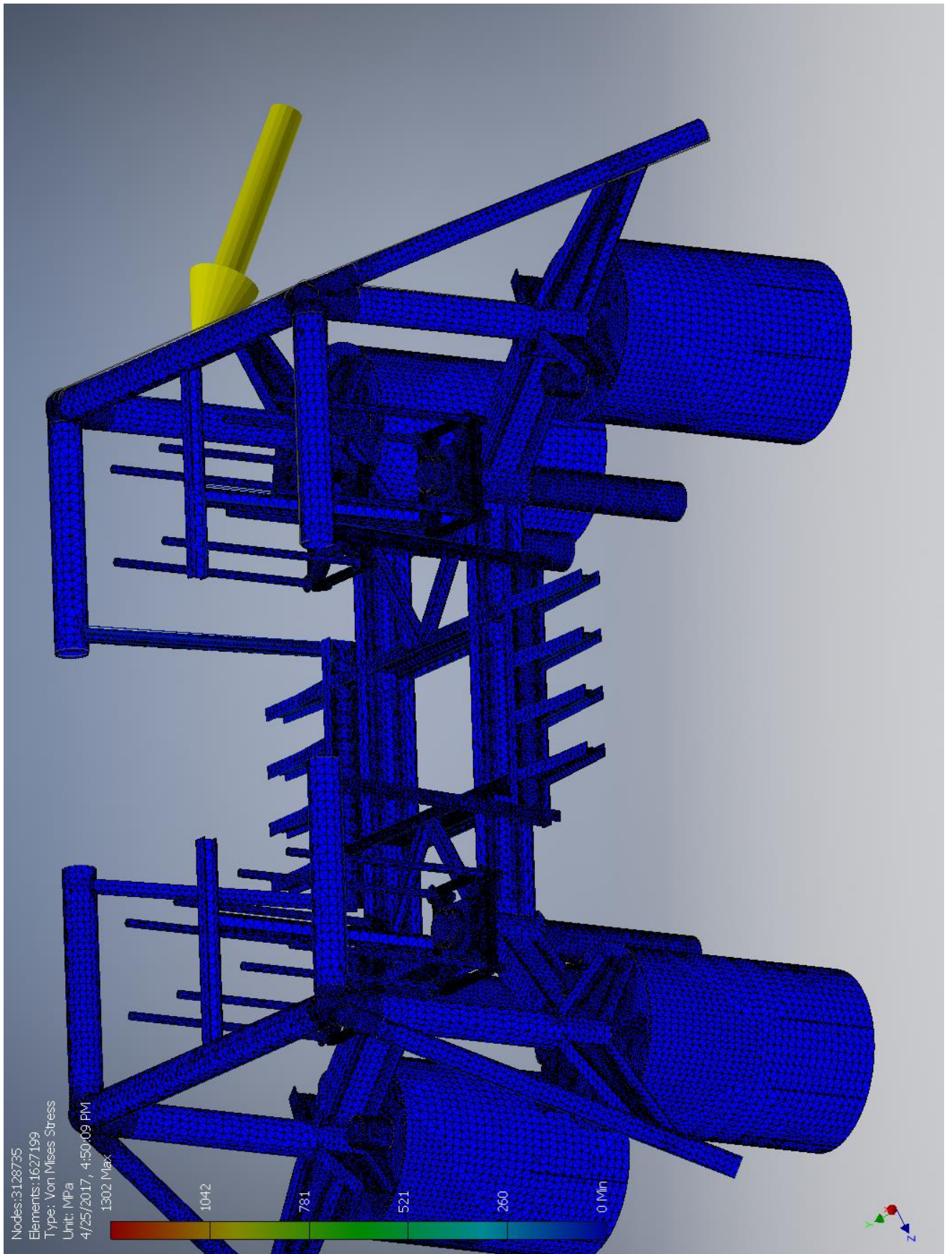
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1000000 N	-481602 N	15257000 N m	-10055100 N m
		-554956 N		11286400 N m
		-678295 N		-2070320 N m

☐ Result Summary

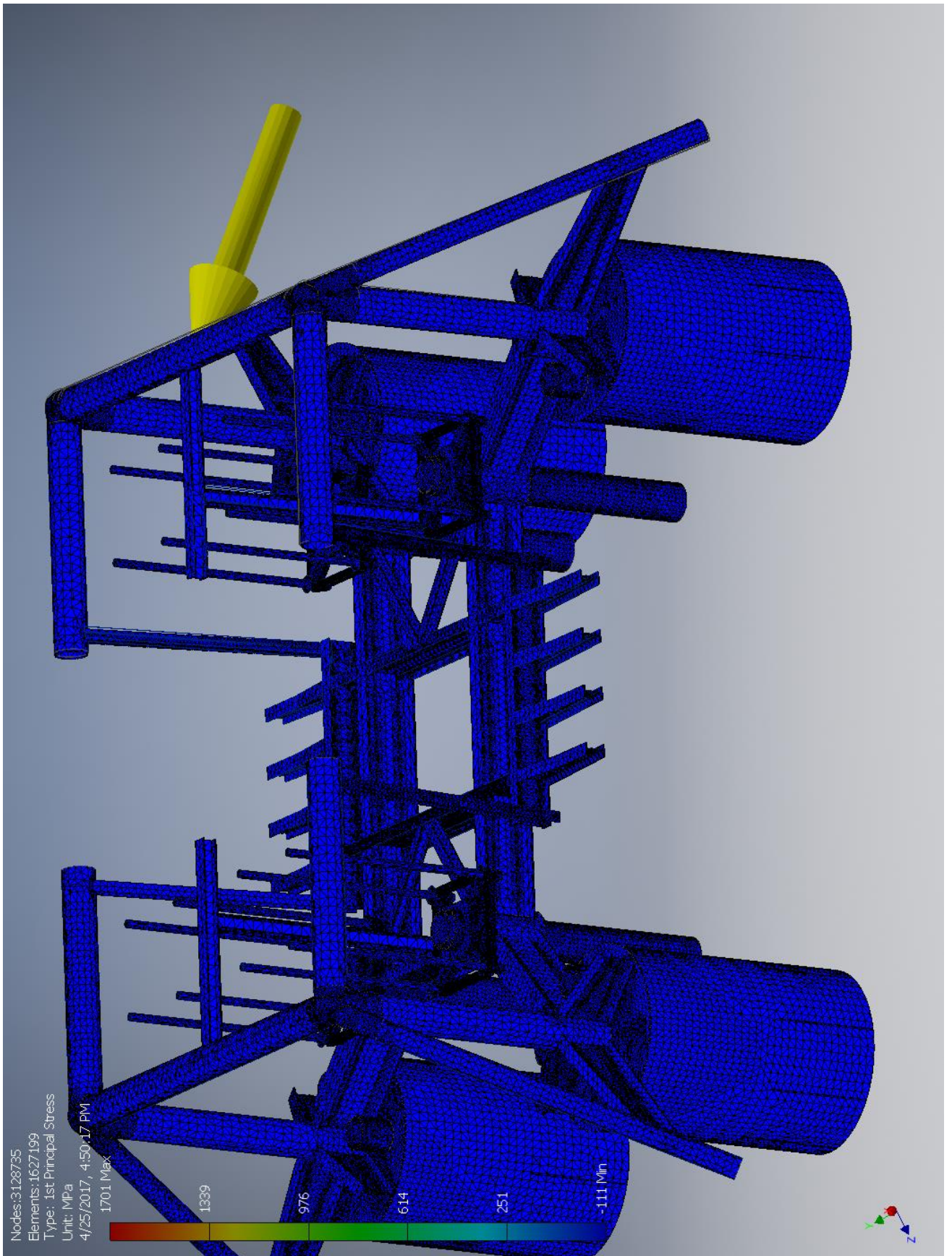
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	1302 MPa
1st Principal Stress	-111.021 MPa	1701.07 MPa
3rd Principal Stress	-369.69 MPa	330.557 MPa
Displacement	0 mm	92.7268 mm
Safety Factor	0.211213 ul	15 ul

☐ Figures

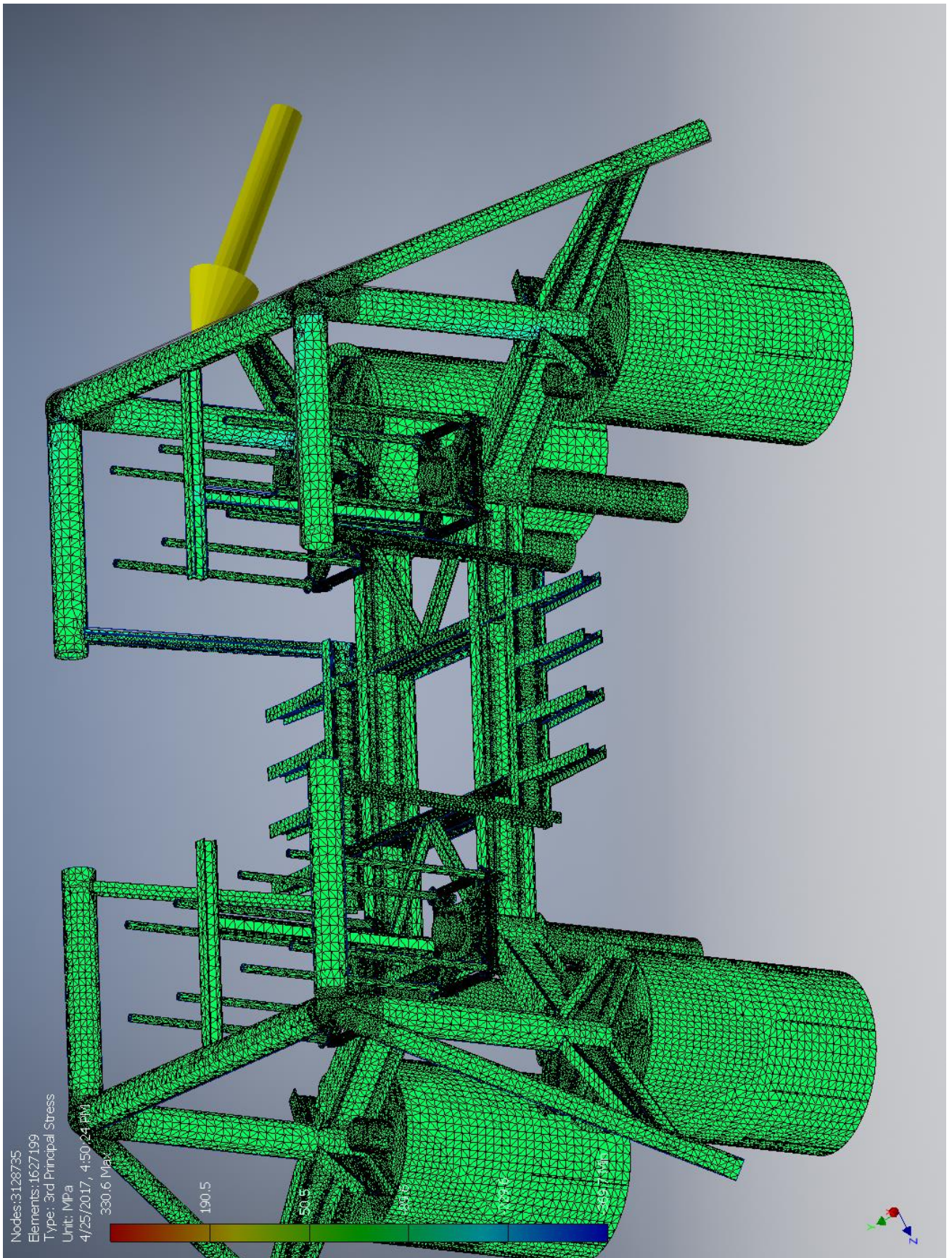
☐ Von Mises Stress



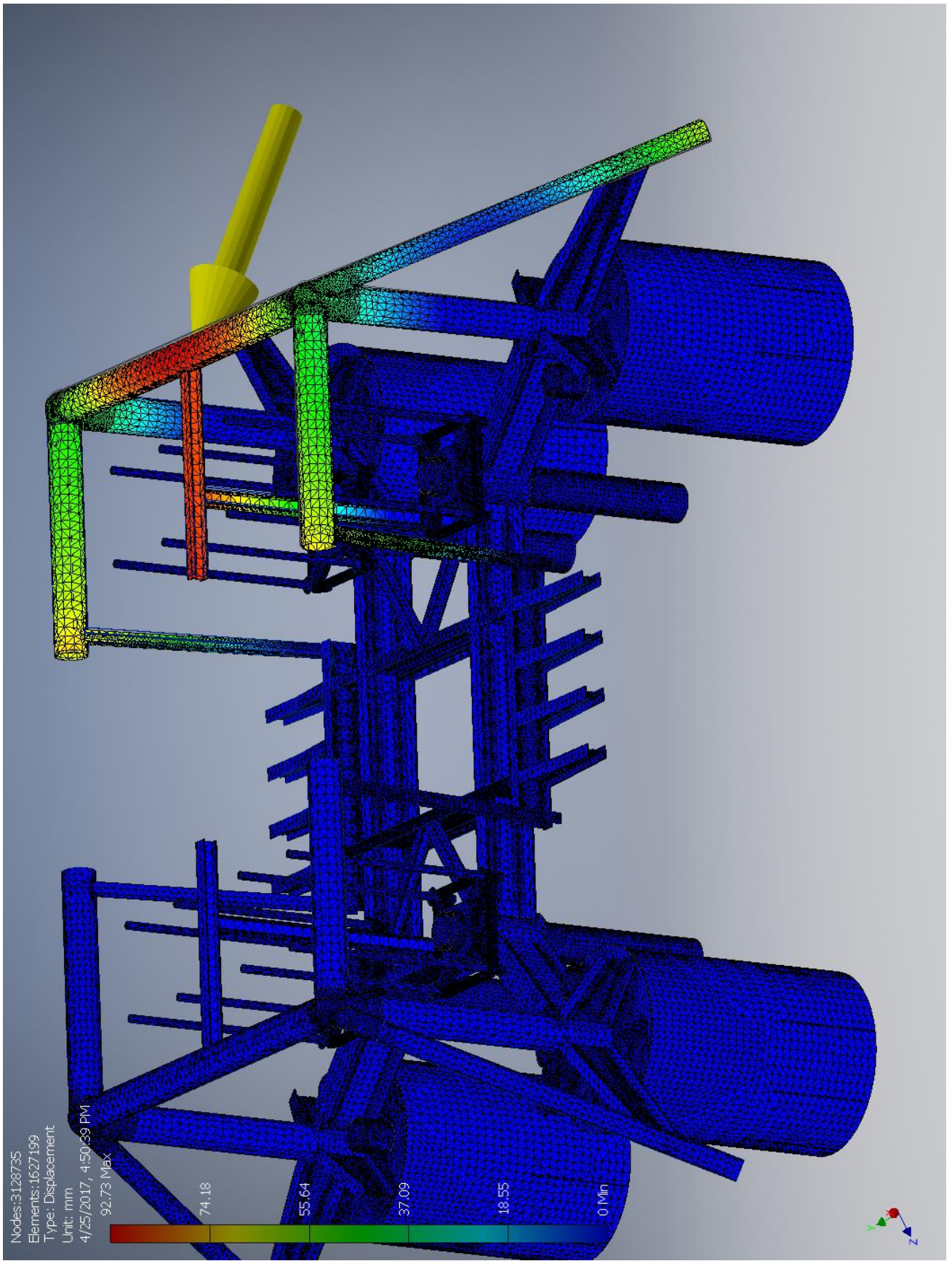
☐ 1st Principal Stress



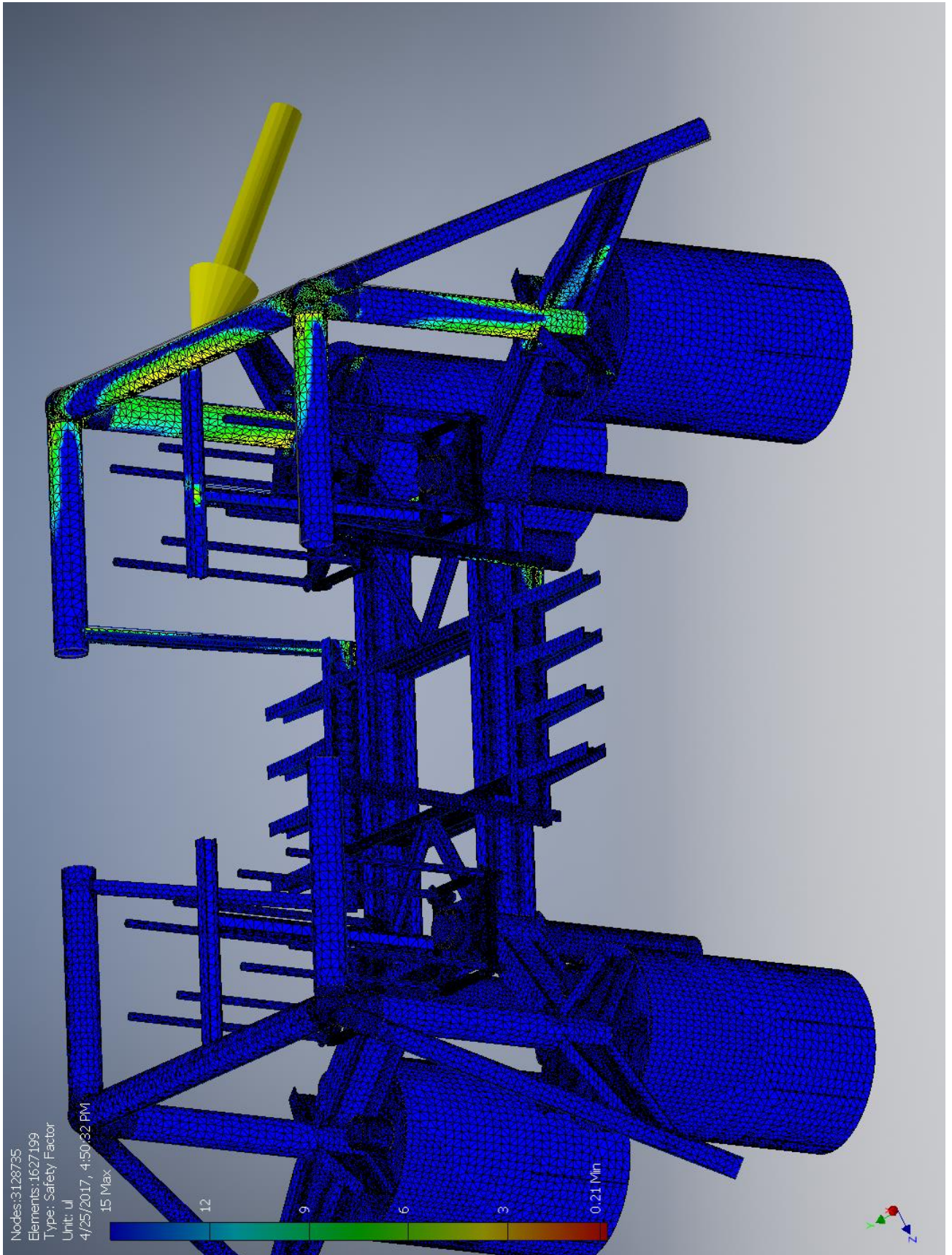
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.2 Case-B

I.2.1 Without convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/26/2017, 12:13 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
----------	----------

☐ Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-B

General objective and settings:

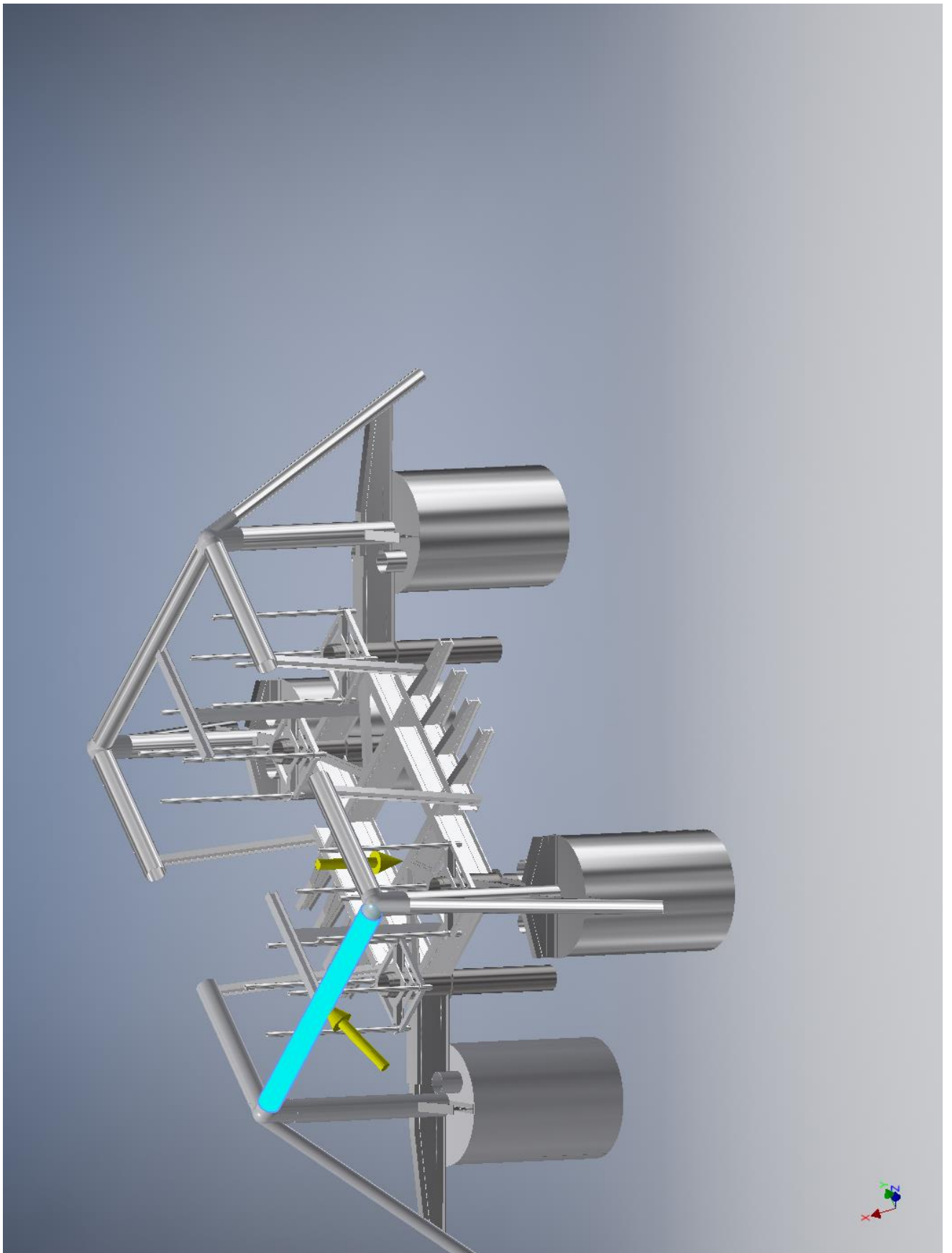
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/26/2017, 12:09 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.000 N
Vector X	481601.661 N
Vector Y	554956.253 N
Vector Z	678294.477 N

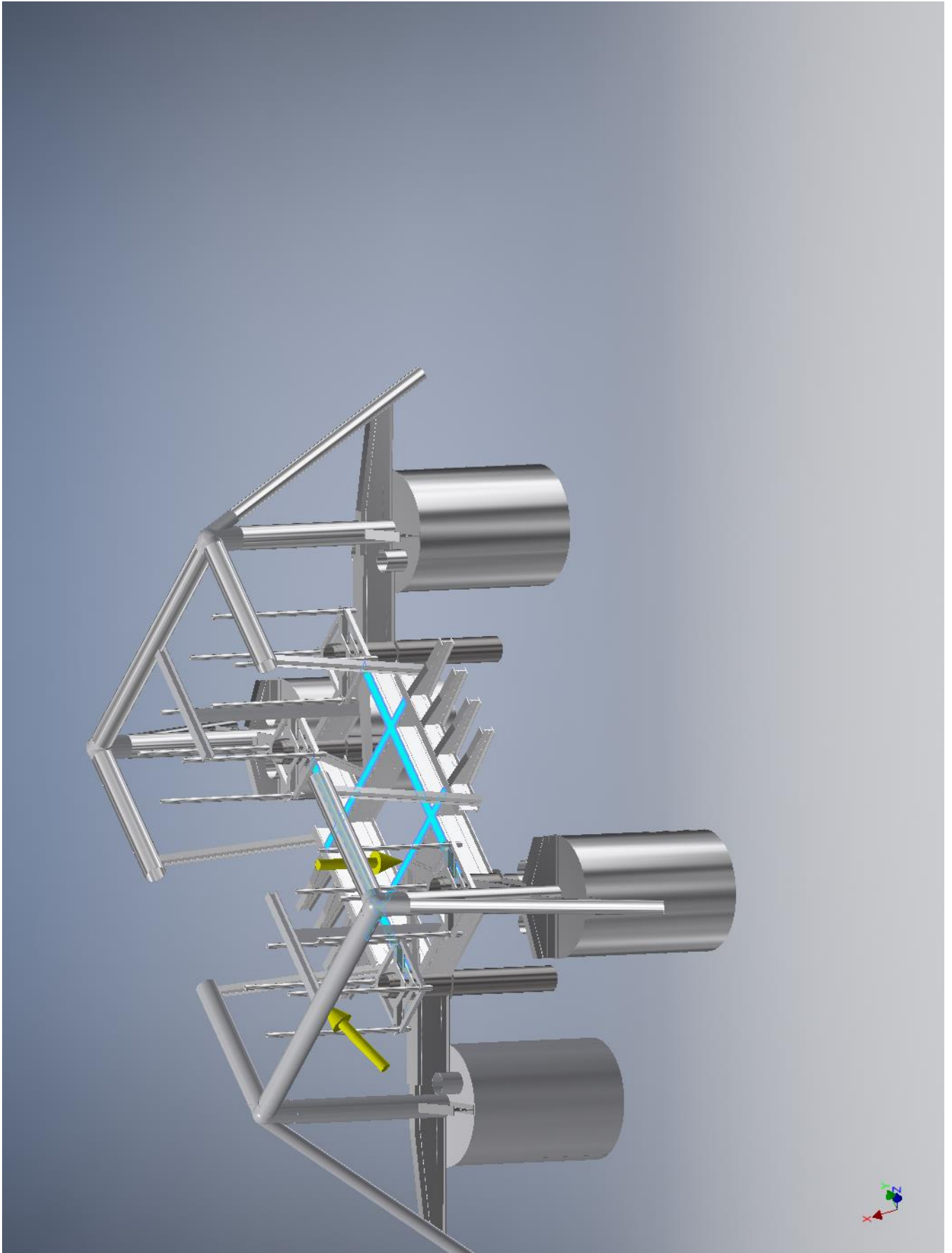
☐ Selected Face(s)



☐ **Force:5**

Load Type	Force
Magnitude	800000.000 N
Vector X	-575687.094 N
Vector Y	-456689.119 N
Vector Z	316258.468 N

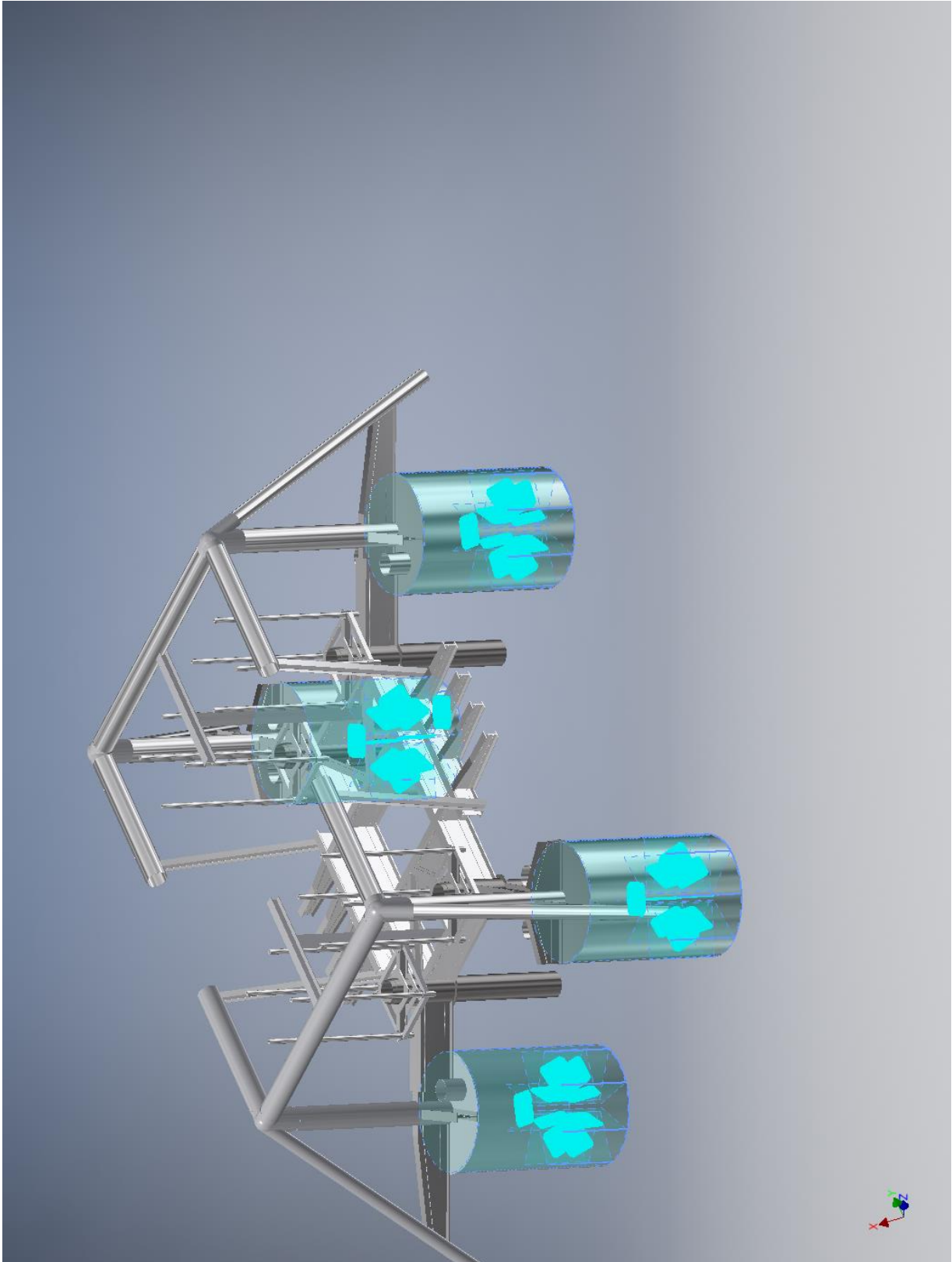
☐ Selected Face(s)



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

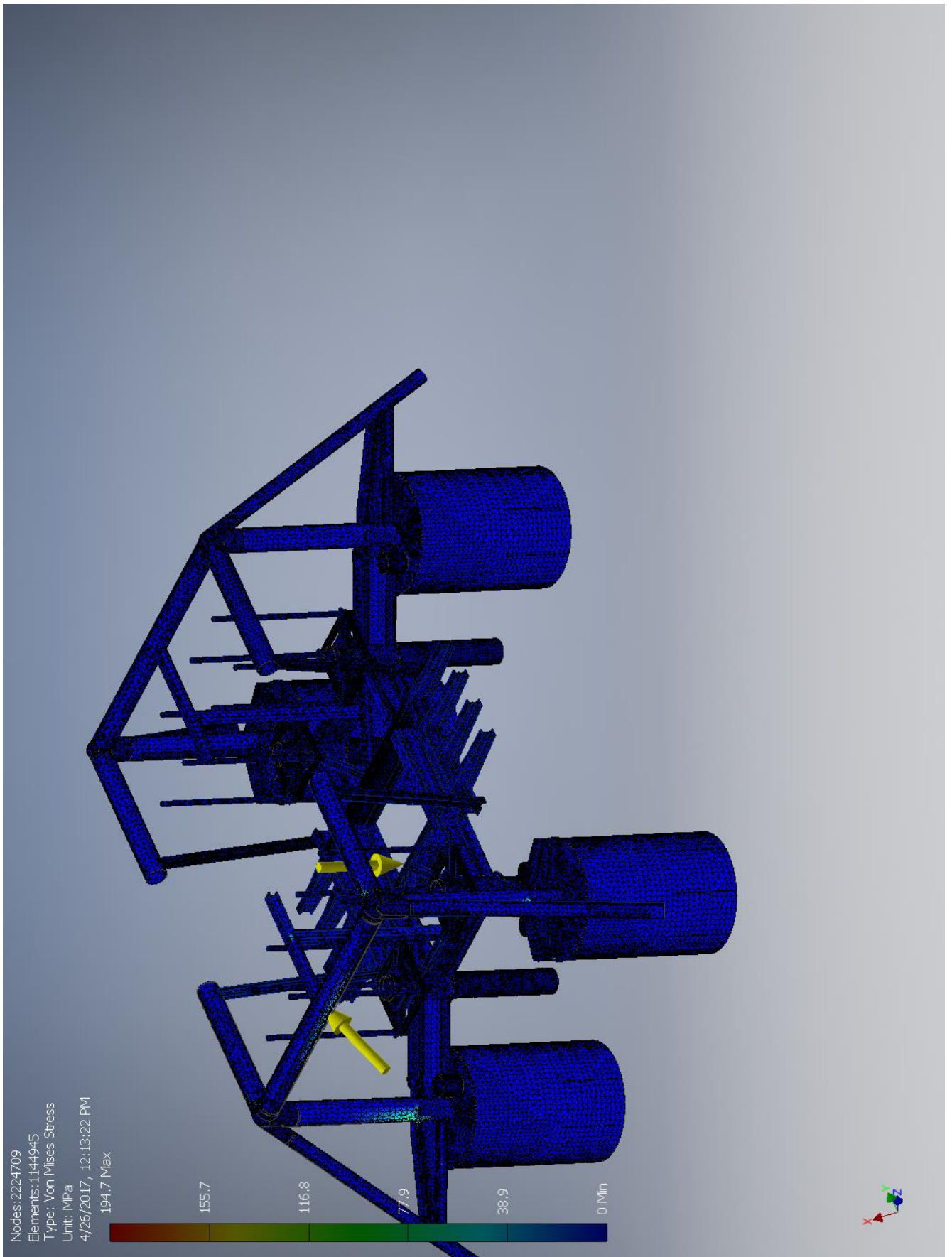
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1003810 N	94085.4 N	15253600 N m	-10070300 N m
		-98267.1 N		11269000 N m
		-994553 N		-2066750 N m

☐ Result Summary

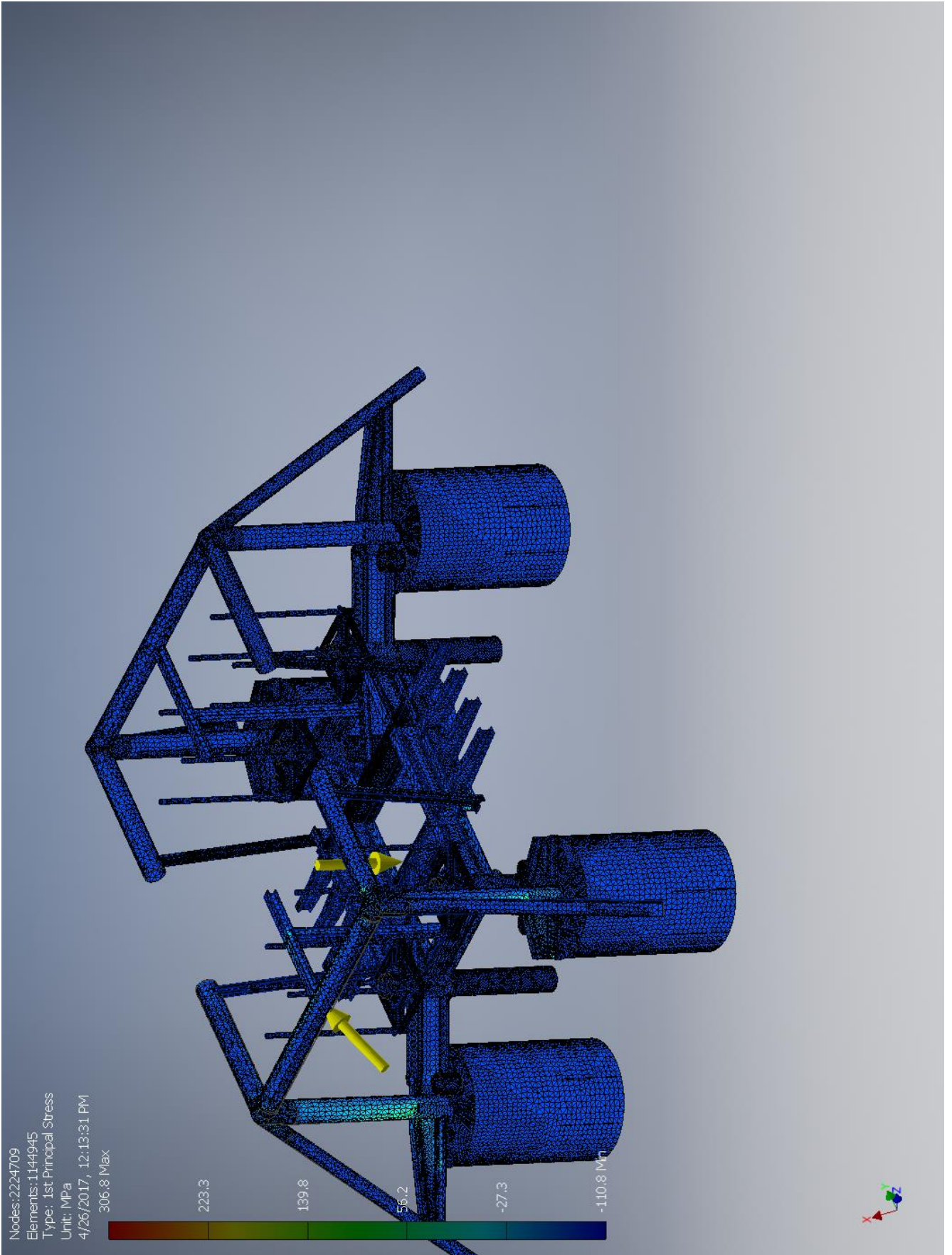
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	194.682 MPa
1st Principal Stress	-110.804 MPa	306.814 MPa
3rd Principal Stress	-262.644 MPa	125.266 MPa
Displacement	0 mm	93.9621 mm
Safety Factor	1.10436 ul	15 ul

☐ Figures

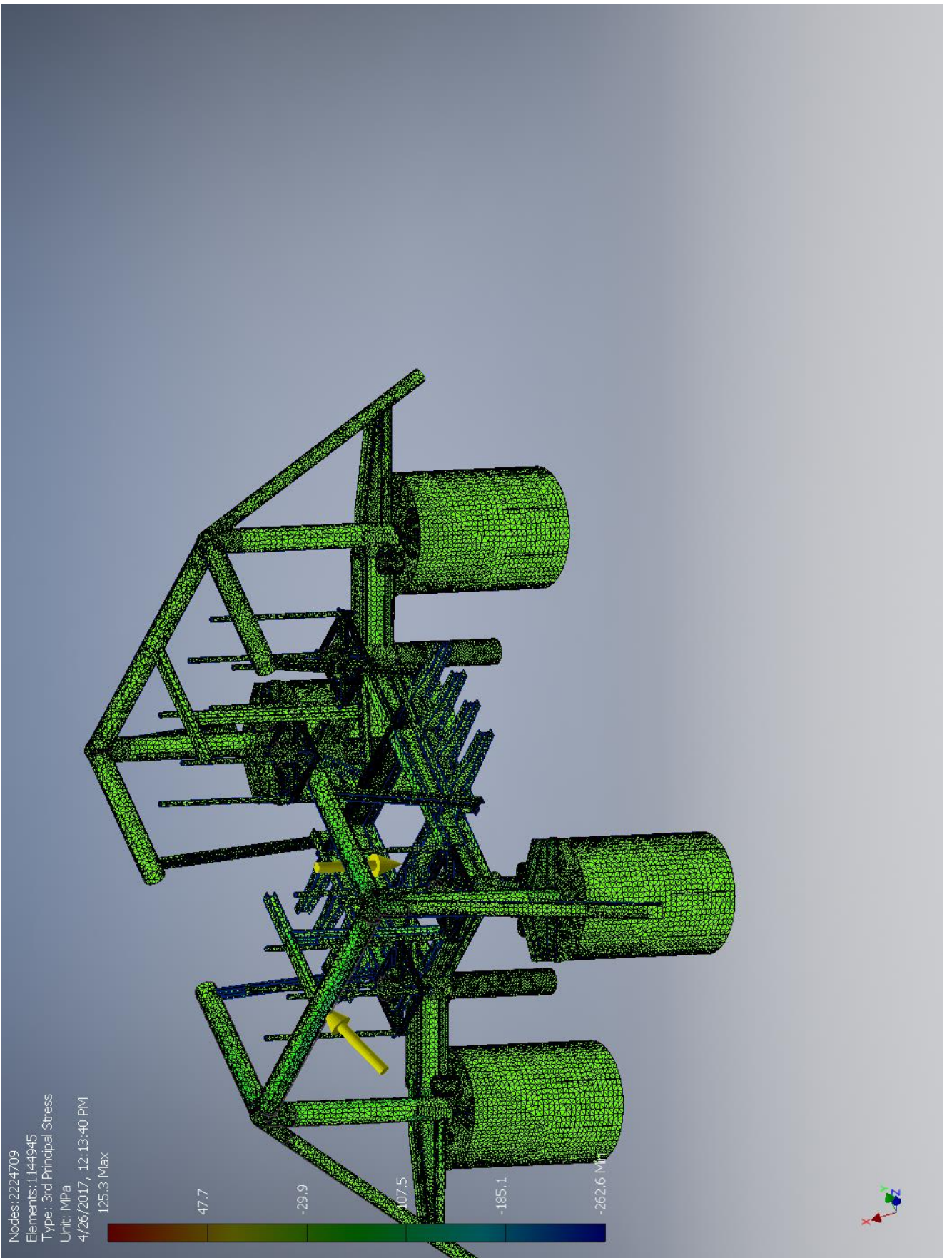
☐ Von Mises Stress



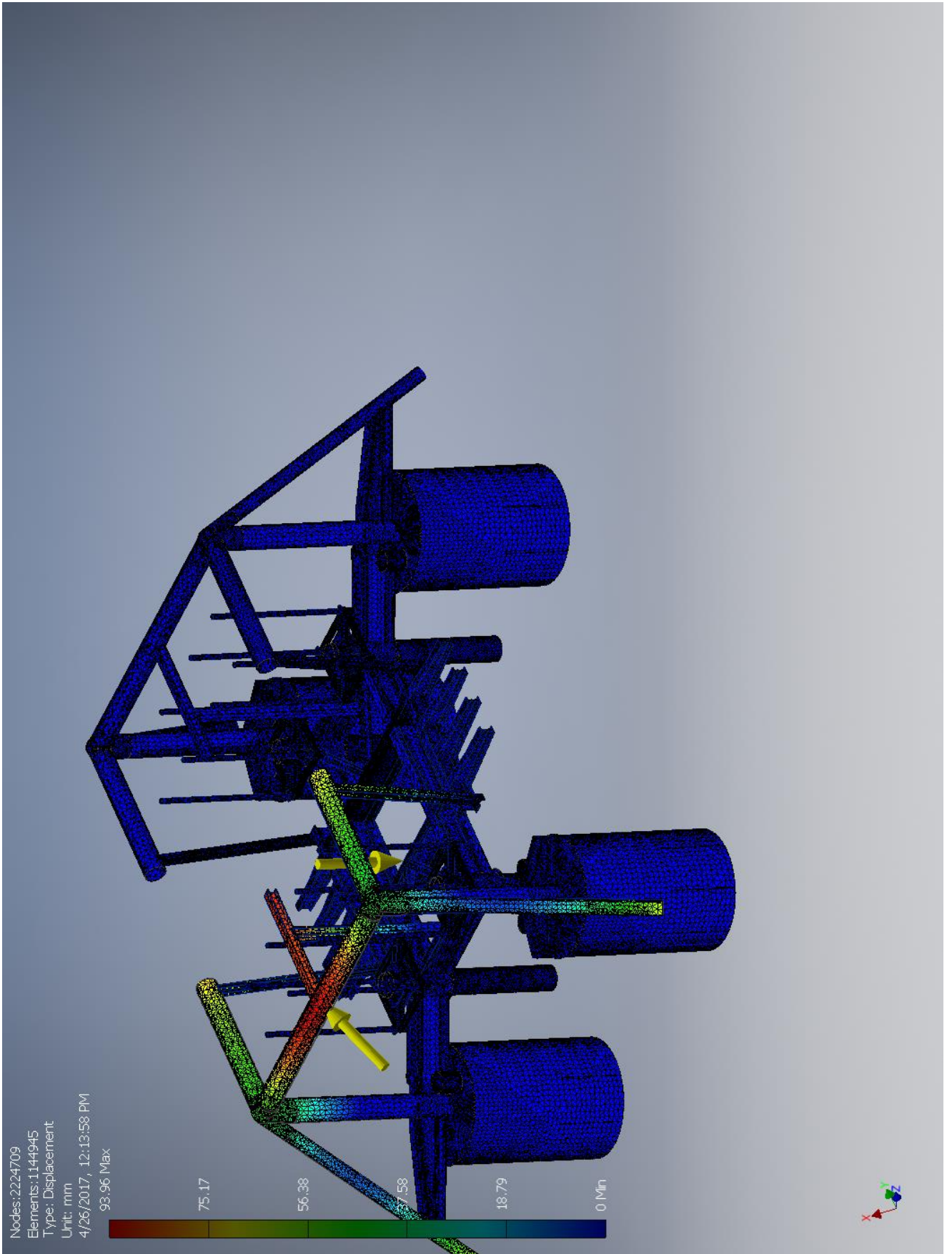
1st Principal Stress



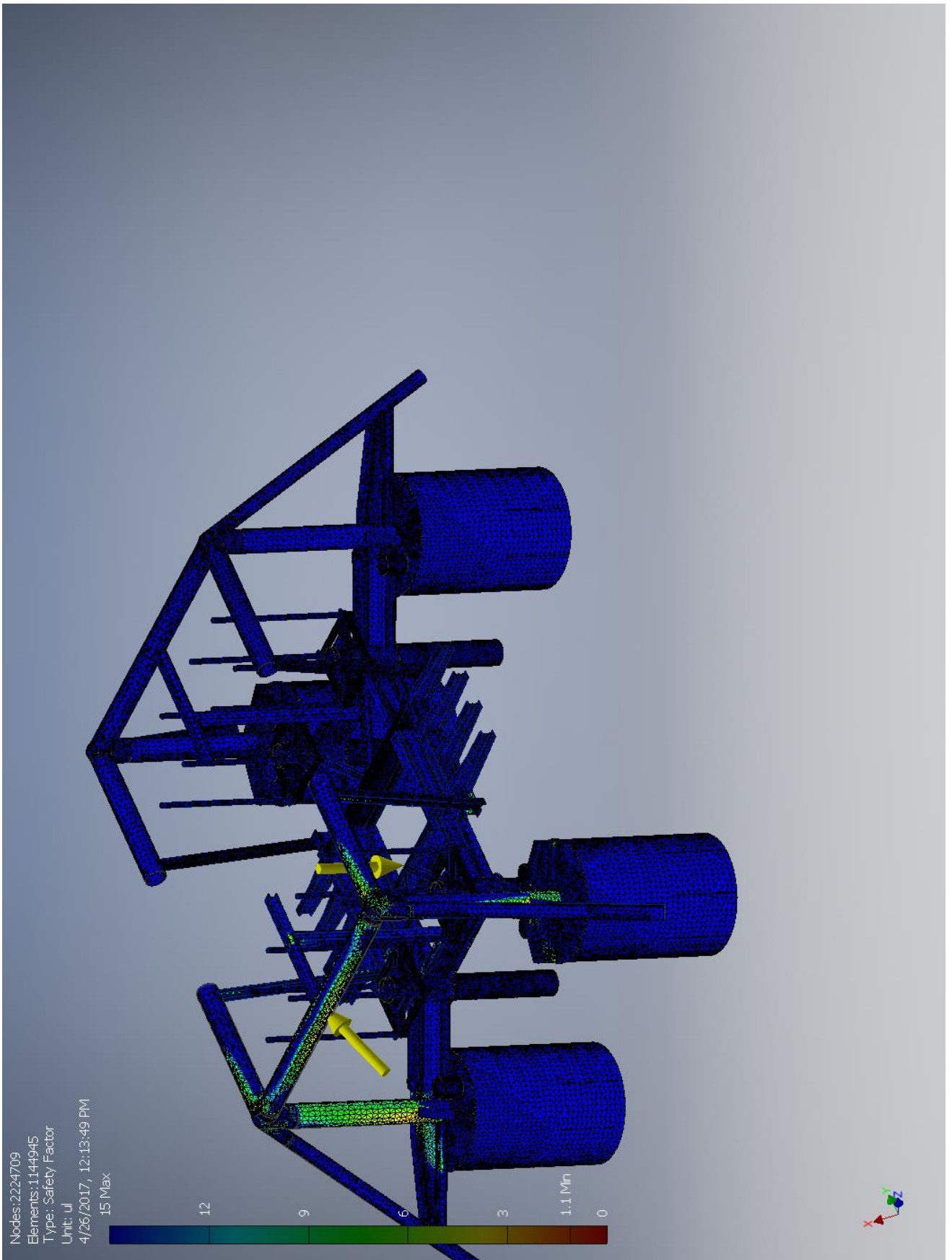
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.2.2 With convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/26/2017, 11:56 AM
Study Author:	henrikwn
Summary:	

Project Info (iProperties)

Summary

Title	
Author	henrikwn

Project

Part Number	Assembly
Designer	henrikwn

Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

Case-B

General objective and settings:

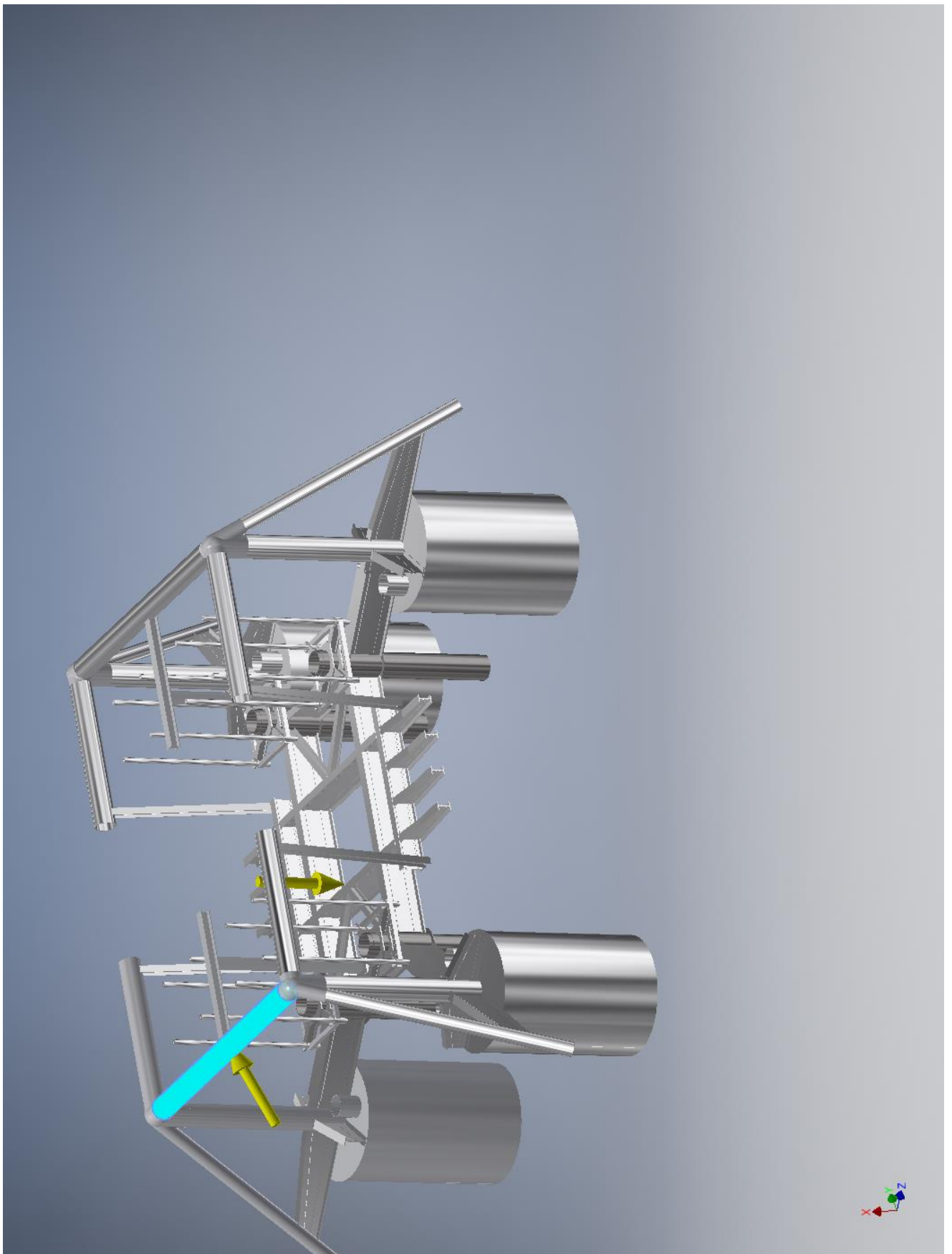
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/26/2017, 11:41 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

Operating conditions

Force:1

Load Type	Force
Magnitude	1000000.000 N
Vector X	481601.661 N
Vector Y	554956.253 N
Vector Z	678294.477 N

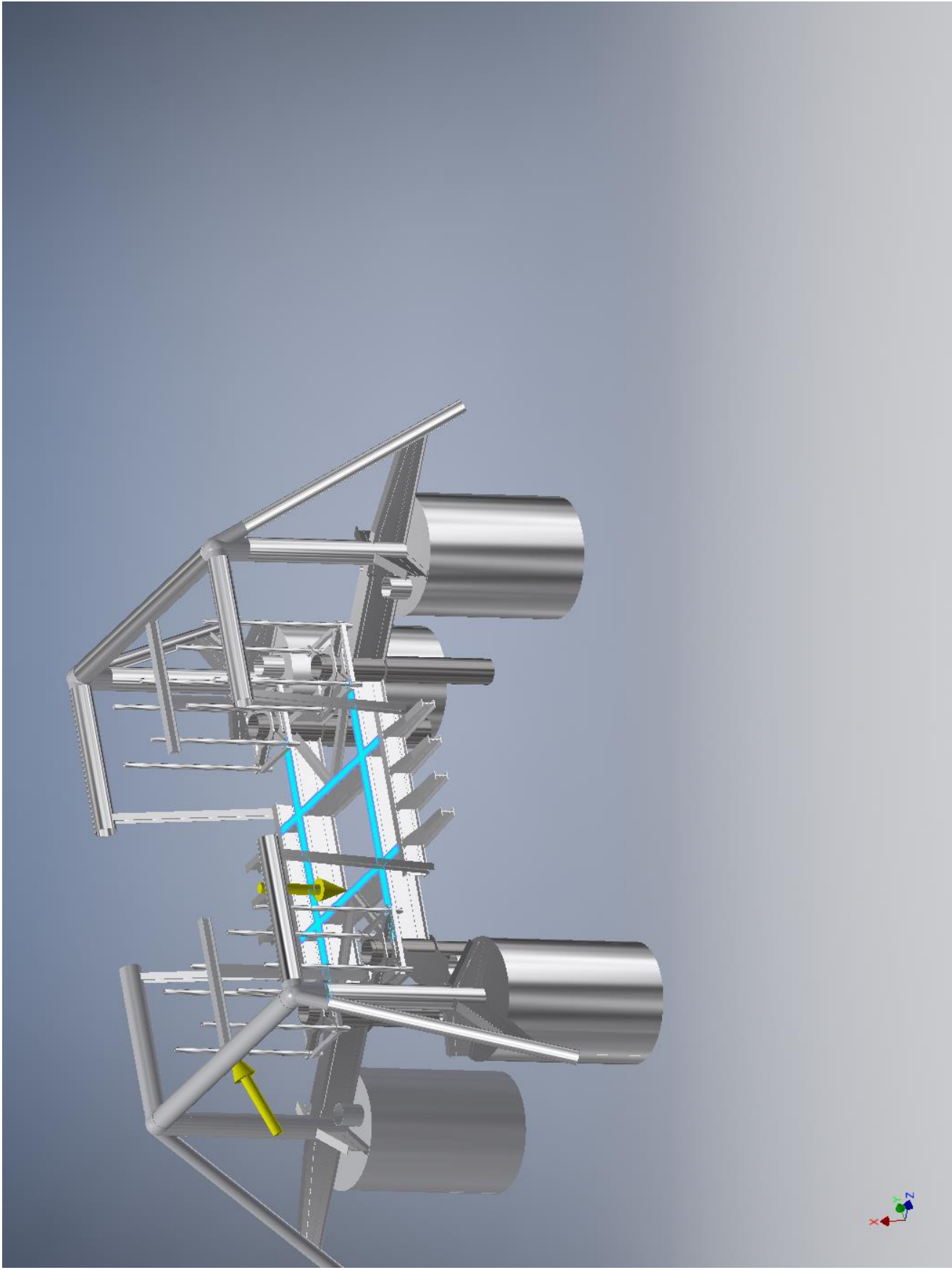
Selected Face(s)



Force:5

Load Type	Force
Magnitude	800000.000 N
Vector X	-575687.094 N
Vector Y	-456689.119 N
Vector Z	316258.468 N

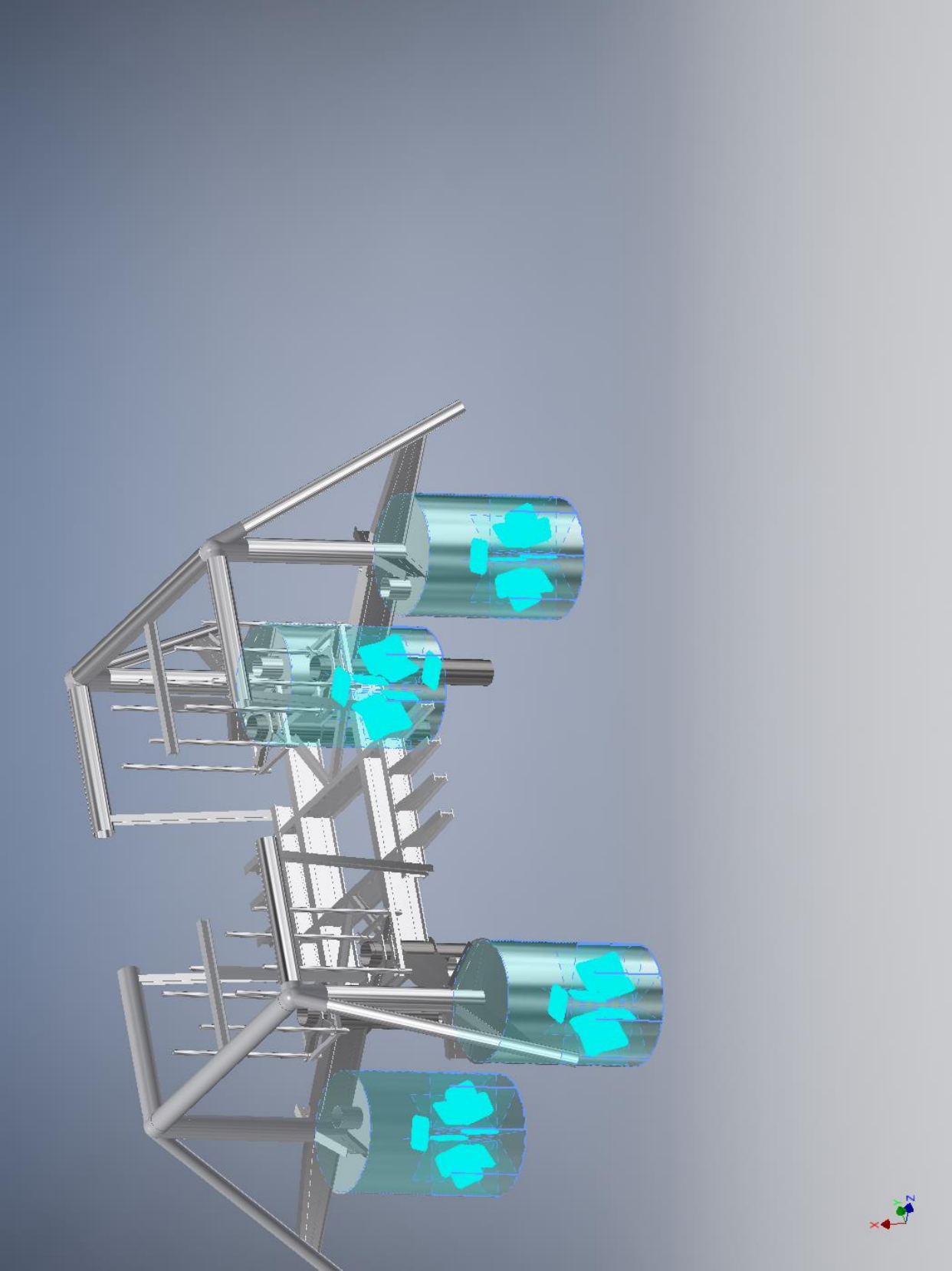
Selected Face(s)



Fixed Constraint:1

Constraint Type	Fixed Constraint
-----------------	------------------

Selected Face(s)



Results

Reaction Force and Moment on Constraints

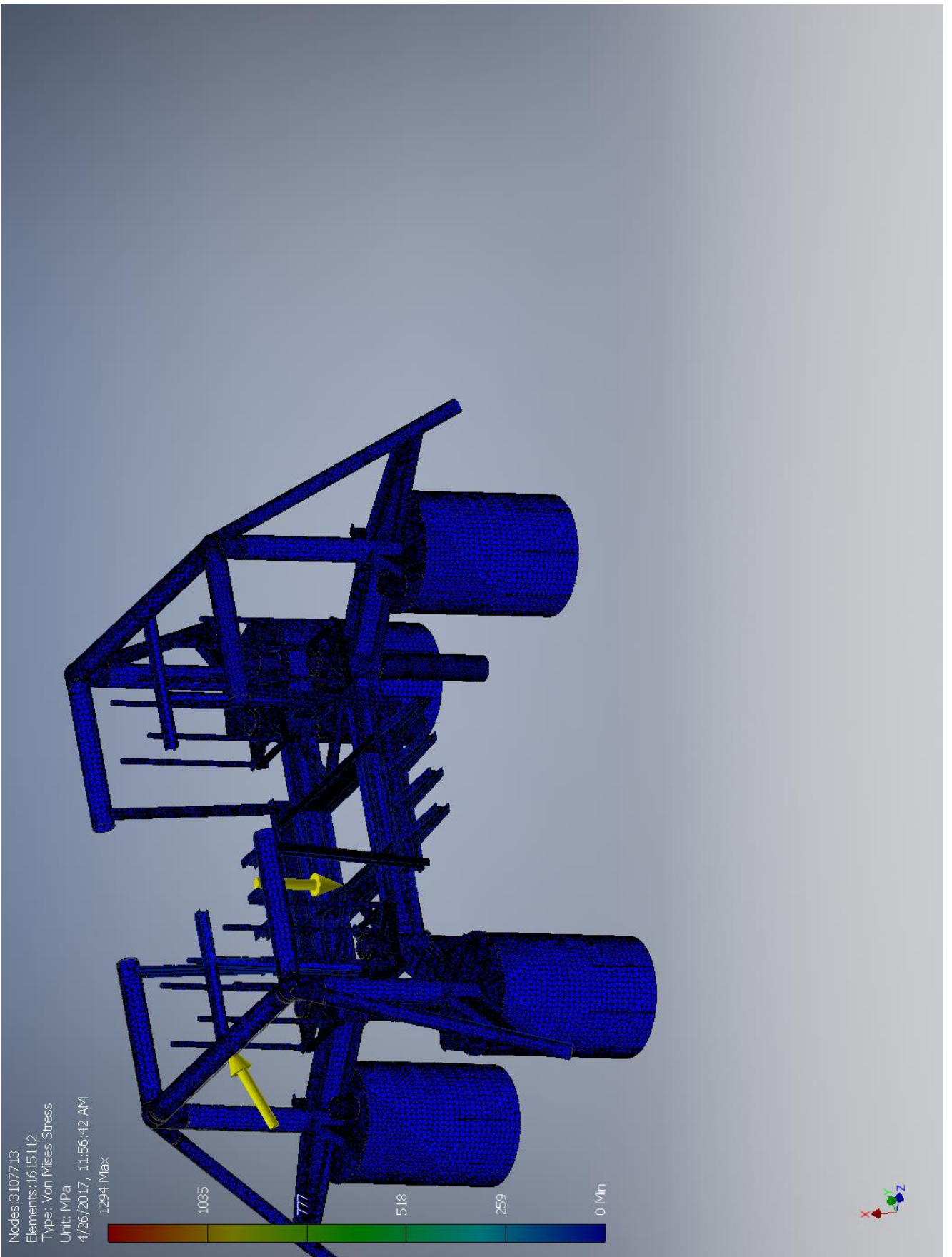
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1003810 N	94085.4 N	15255000 N m	-10061700 N m
		-98267.1 N		11278500 N m
		-994553 N		-2067110 N m

Result Summary

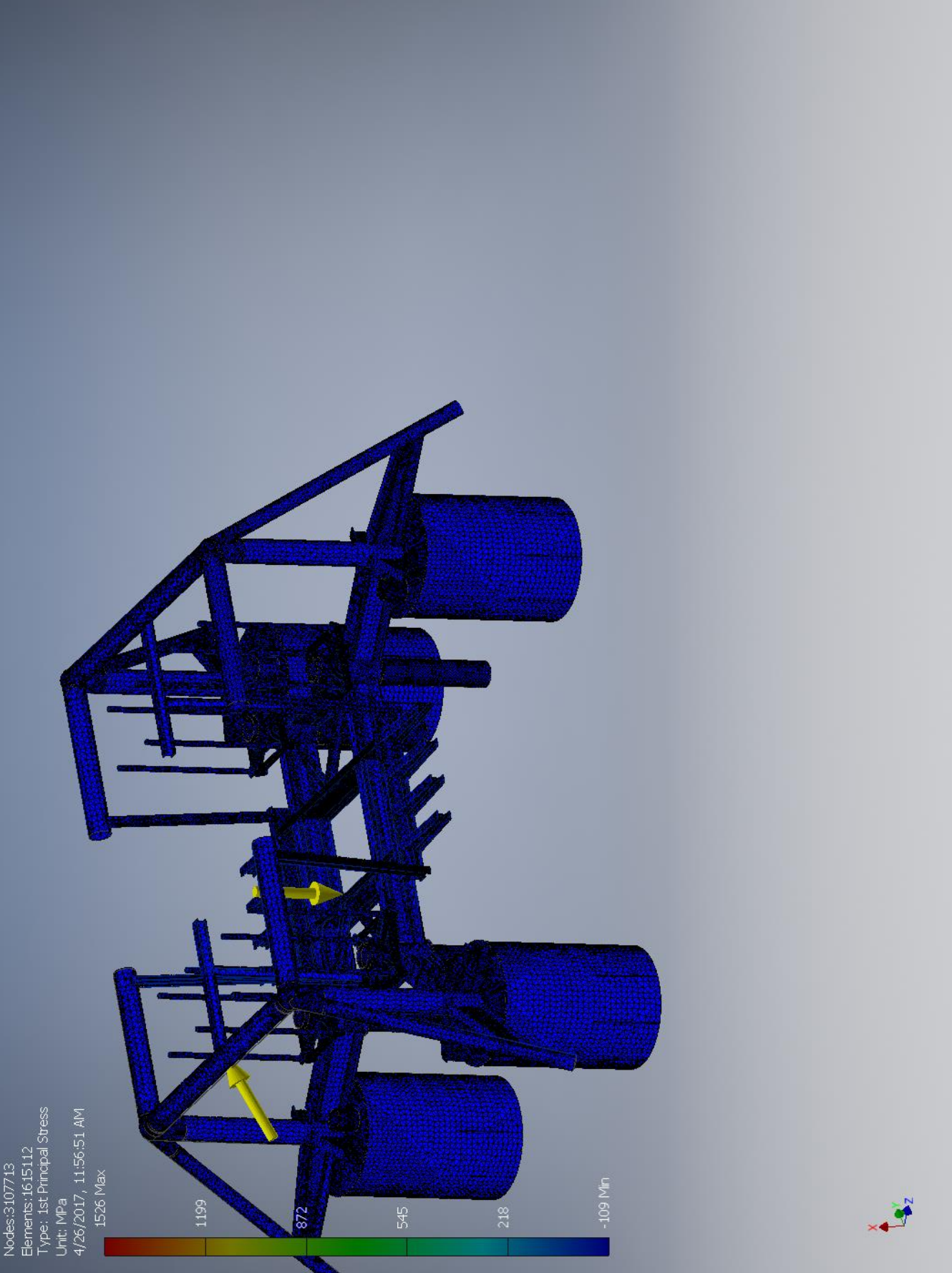
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	1294.3 MPa
1st Principal Stress	-108.756 MPa	1526 MPa
3rd Principal Stress	-320.304 MPa	169.333 MPa
Displacement	0 mm	94.1343 mm
Safety Factor	0.21247 ul	15 ul

Figures

Von Mises Stress



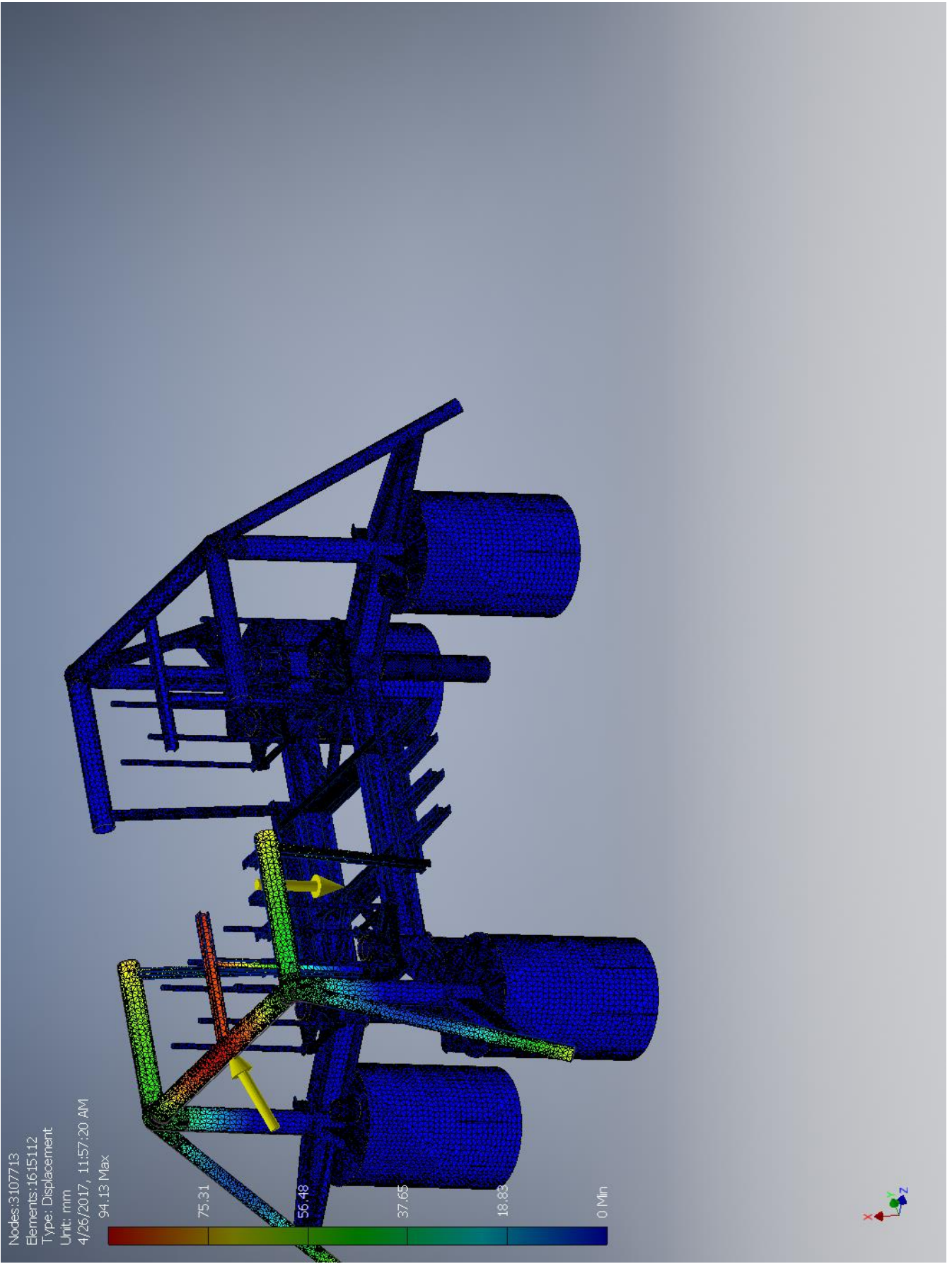
1st Principal Stress



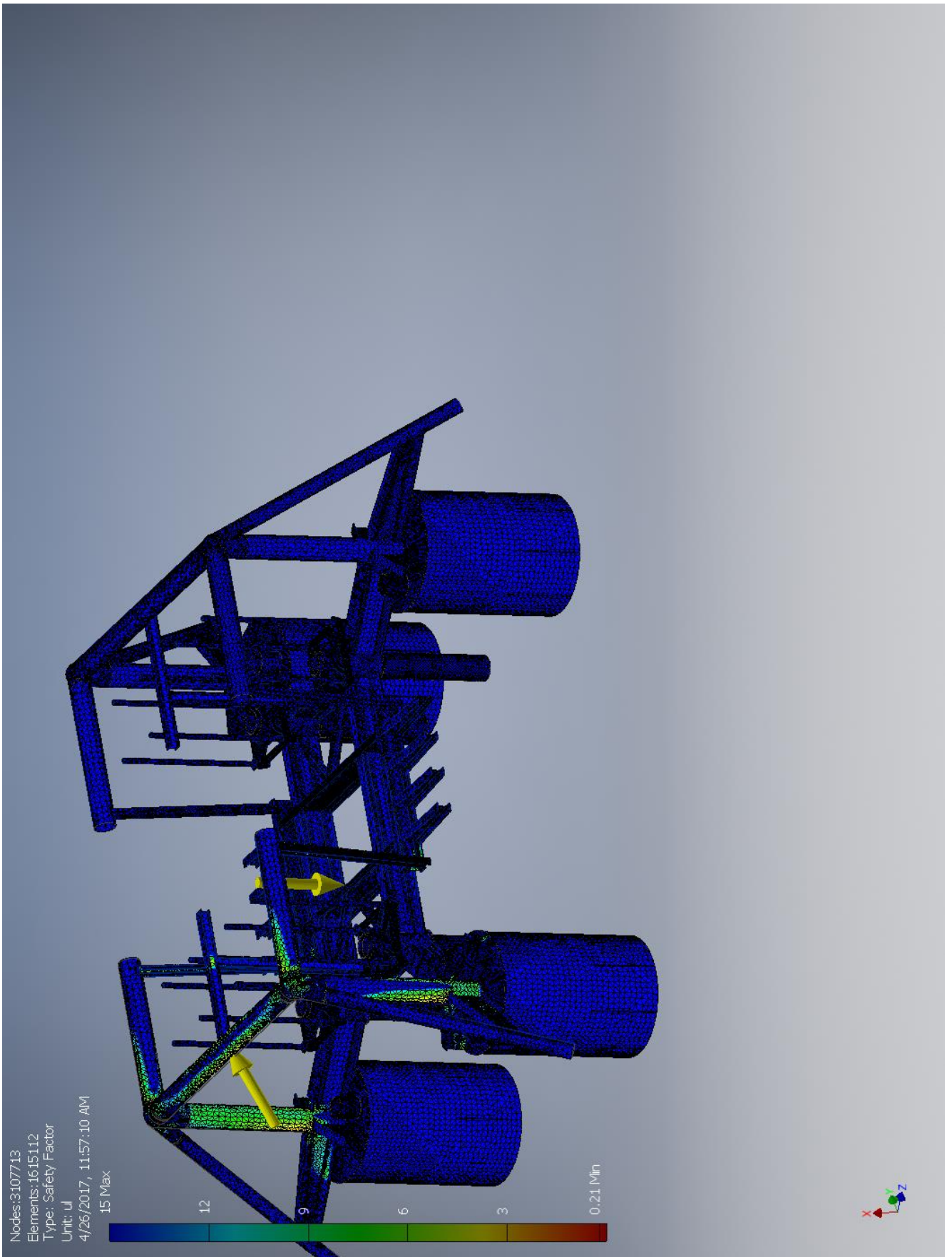
3rd Principal Stress



Displacement



Safety Factor



I.3 Case-C

I.3.1 Without convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/27/2017, 12:23 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-C

General objective and settings:

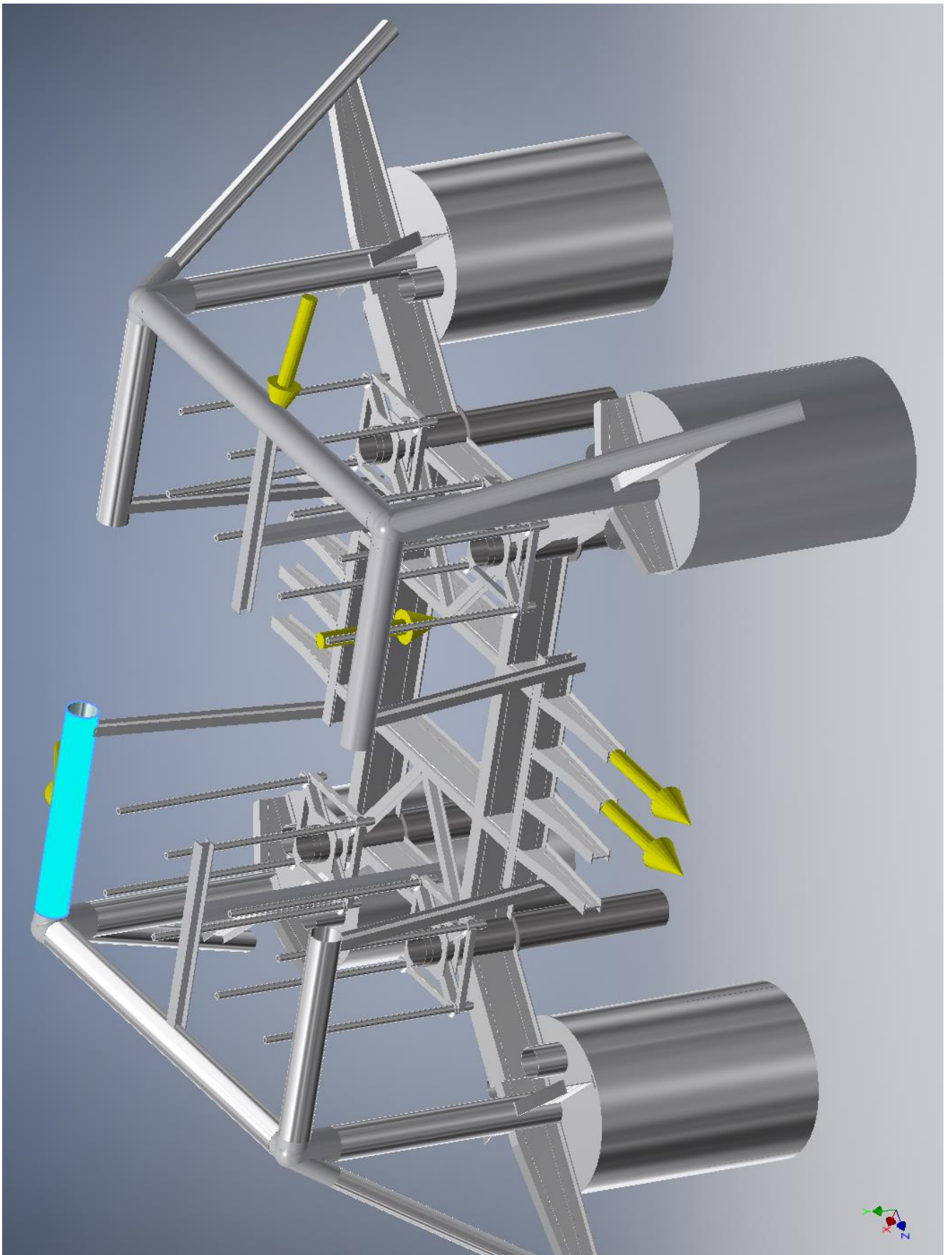
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/27/2017, 12:15 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.000 N
Vector X	936001.935 N
Vector Y	-341194.991 N
Vector Z	-86523.732 N

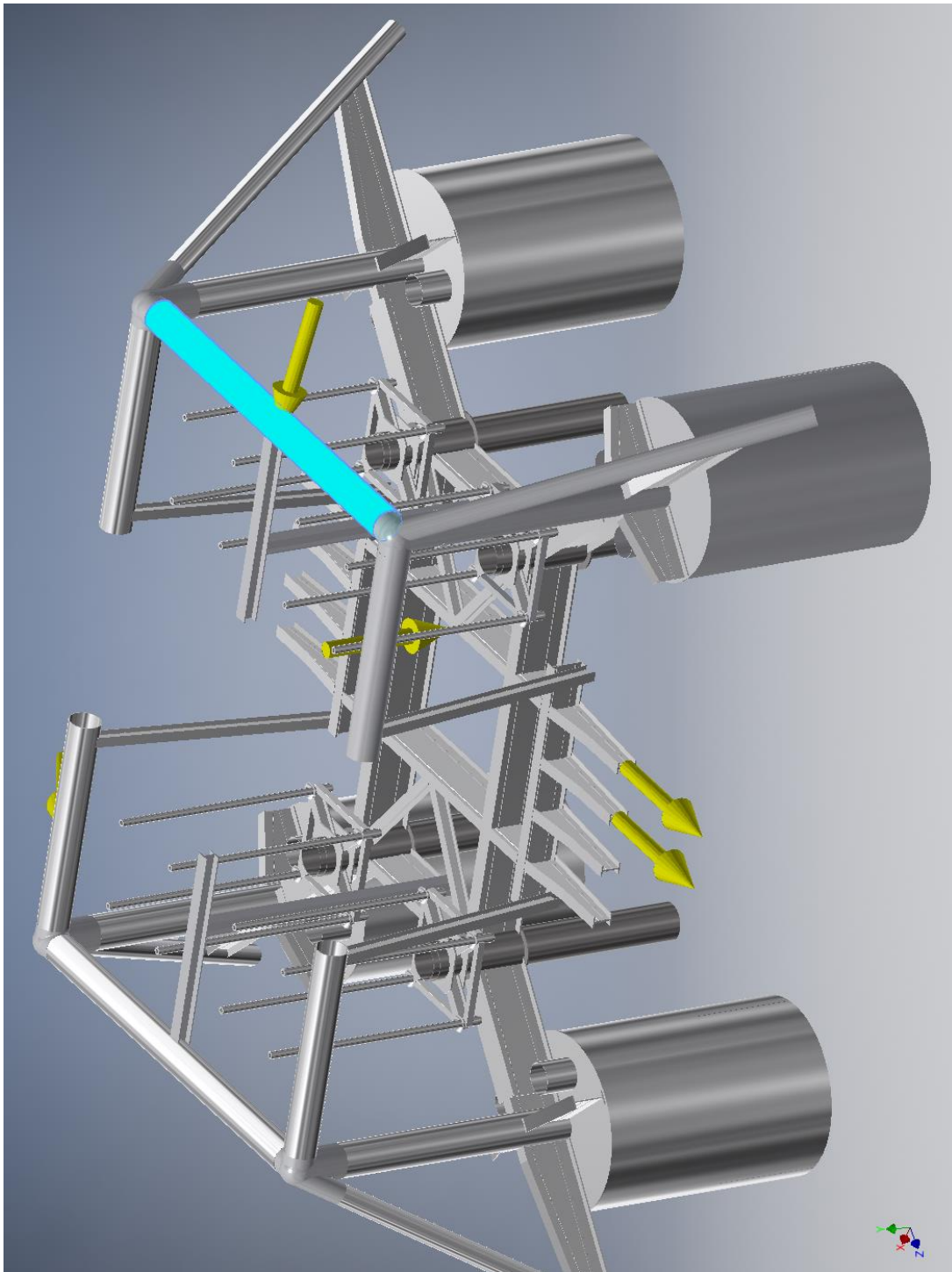
☐ Selected Face(s)



☐ **Force:2**

Load Type	Force
Magnitude	1000000.000 N
Vector X	319421.899 N
Vector Y	439526.628 N
Vector Z	839515.333 N

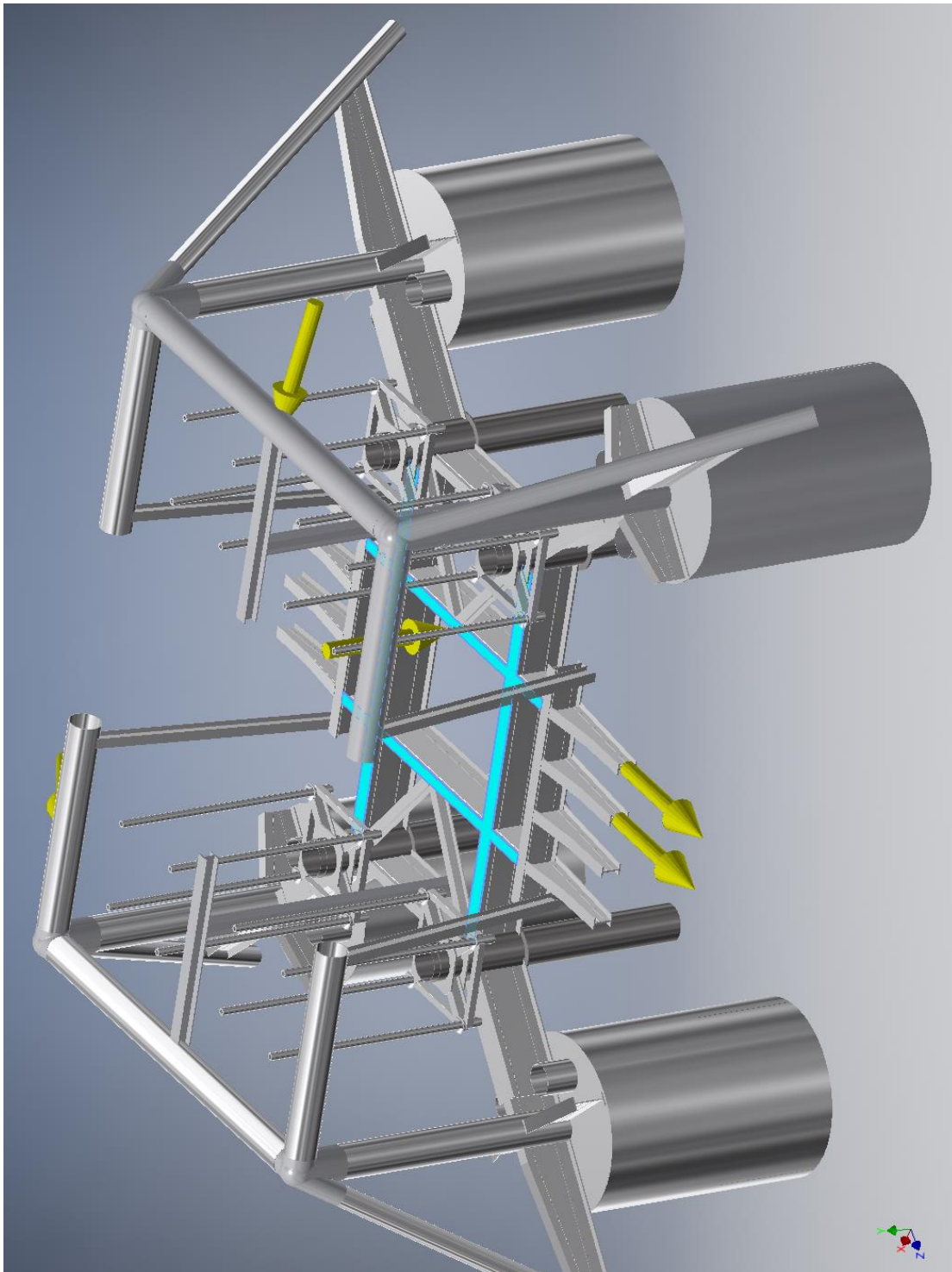
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	80000.000 N
Vector X	-57568.709 N
Vector Y	-45668.912 N
Vector Z	31625.847 N

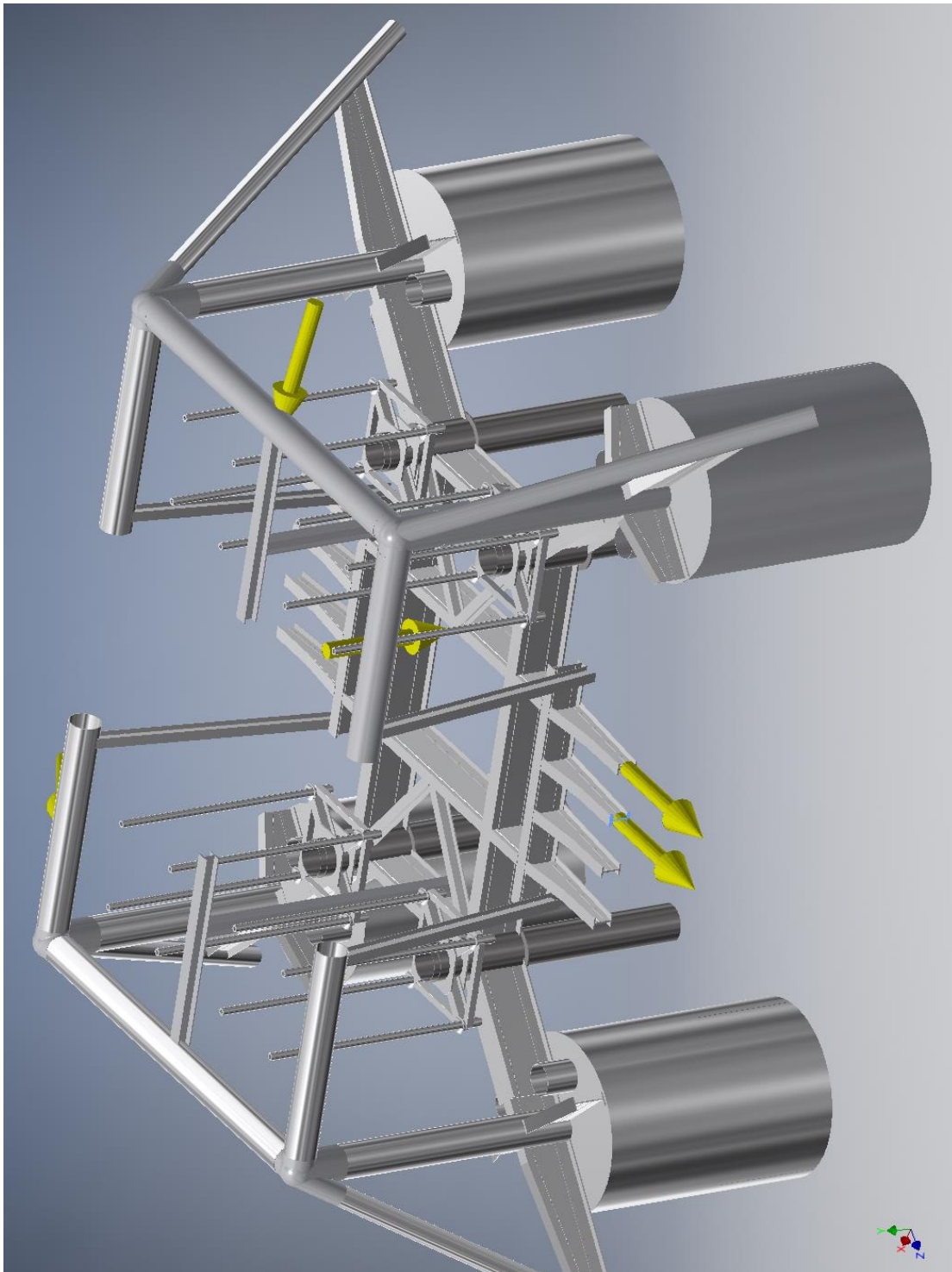
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	300000.000 N
Vector X	198108.677 N
Vector Y	-221589.070 N
Vector Z	40635.406 N

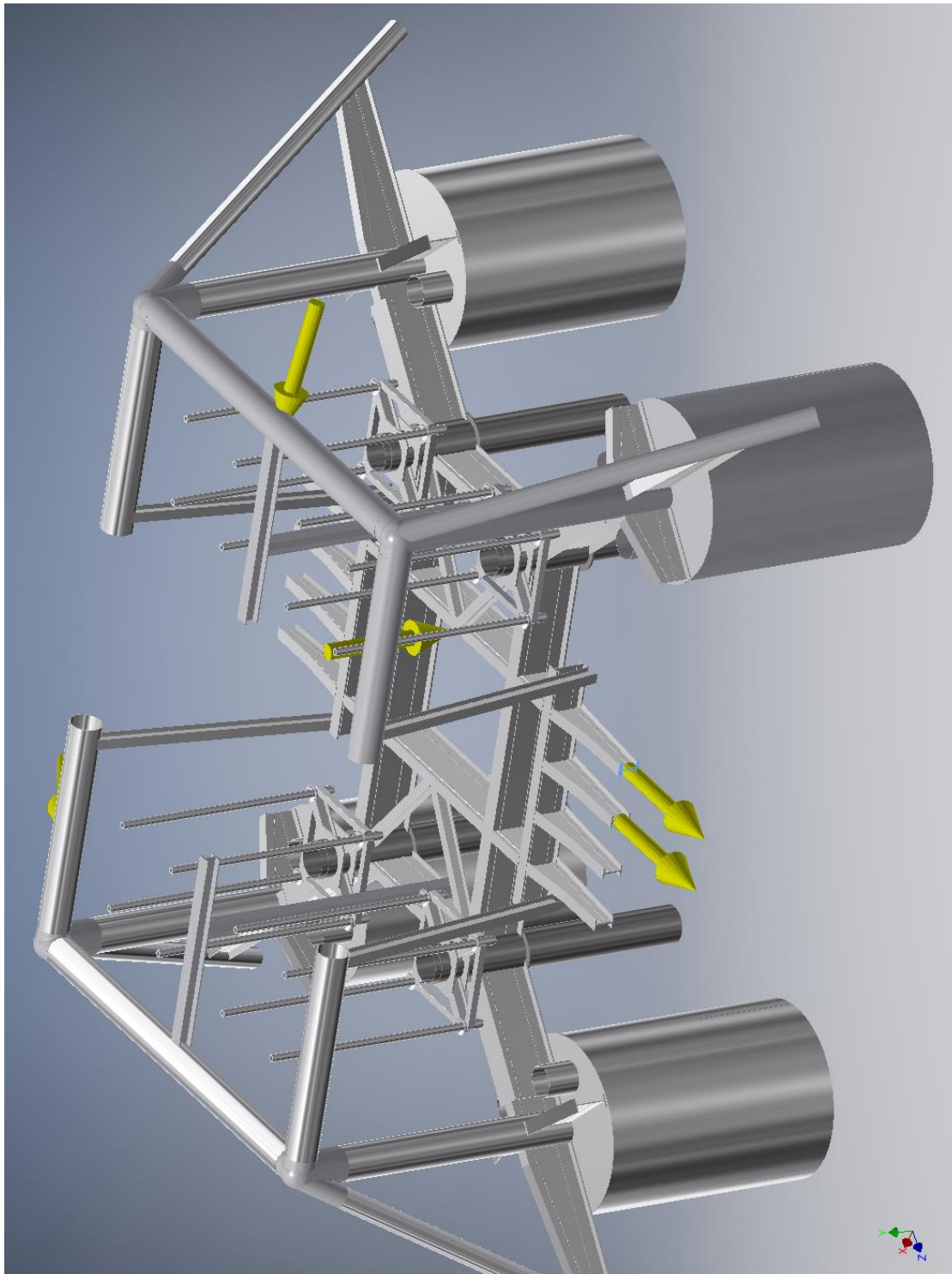
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	300000.000 N
Vector X	198108.677 N
Vector Y	-221589.070 N
Vector Z	40635.406 N

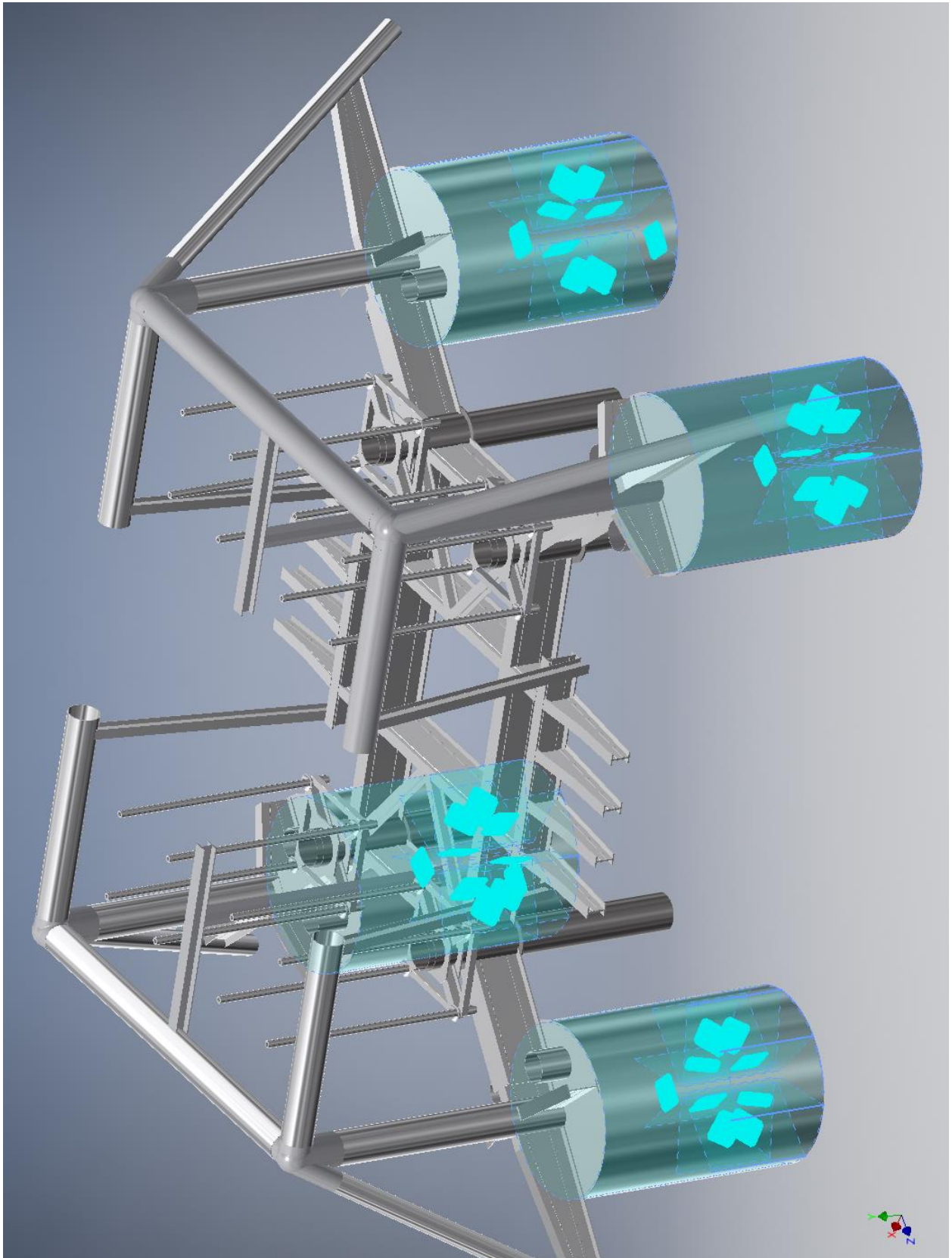
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

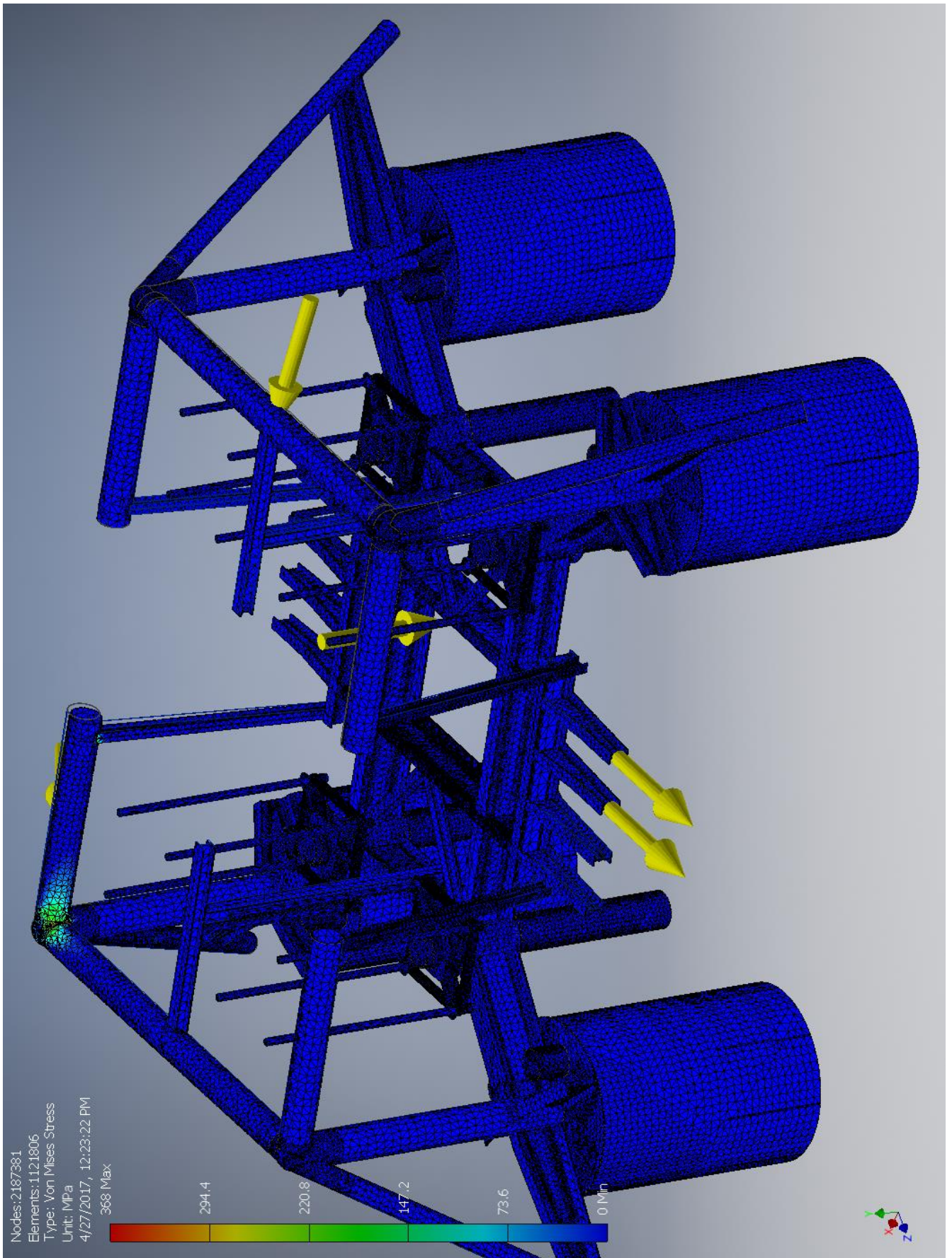
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1855620 N	-1594070 N	20742400 N m	-6426080 N m
		390515 N		11683200 N m
		-865888 N		15888900 N m

☐ Result Summary

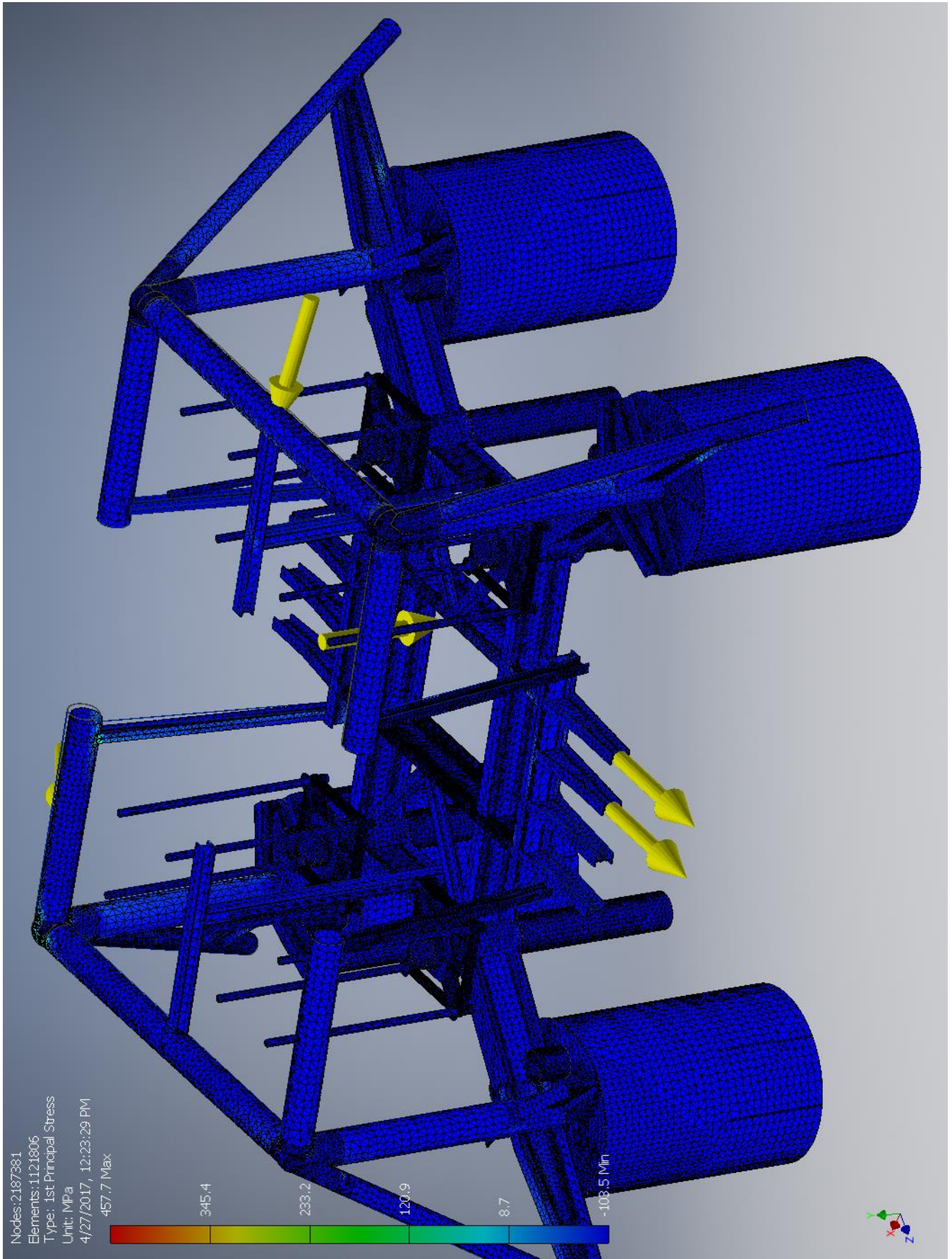
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	368.035 MPa
1st Principal Stress	-103.538 MPa	457.664 MPa
3rd Principal Stress	-464.726 MPa	113.157 MPa
Displacement	0 mm	244.38 mm
Safety Factor	0.588644 ul	15 ul

☐ Figures

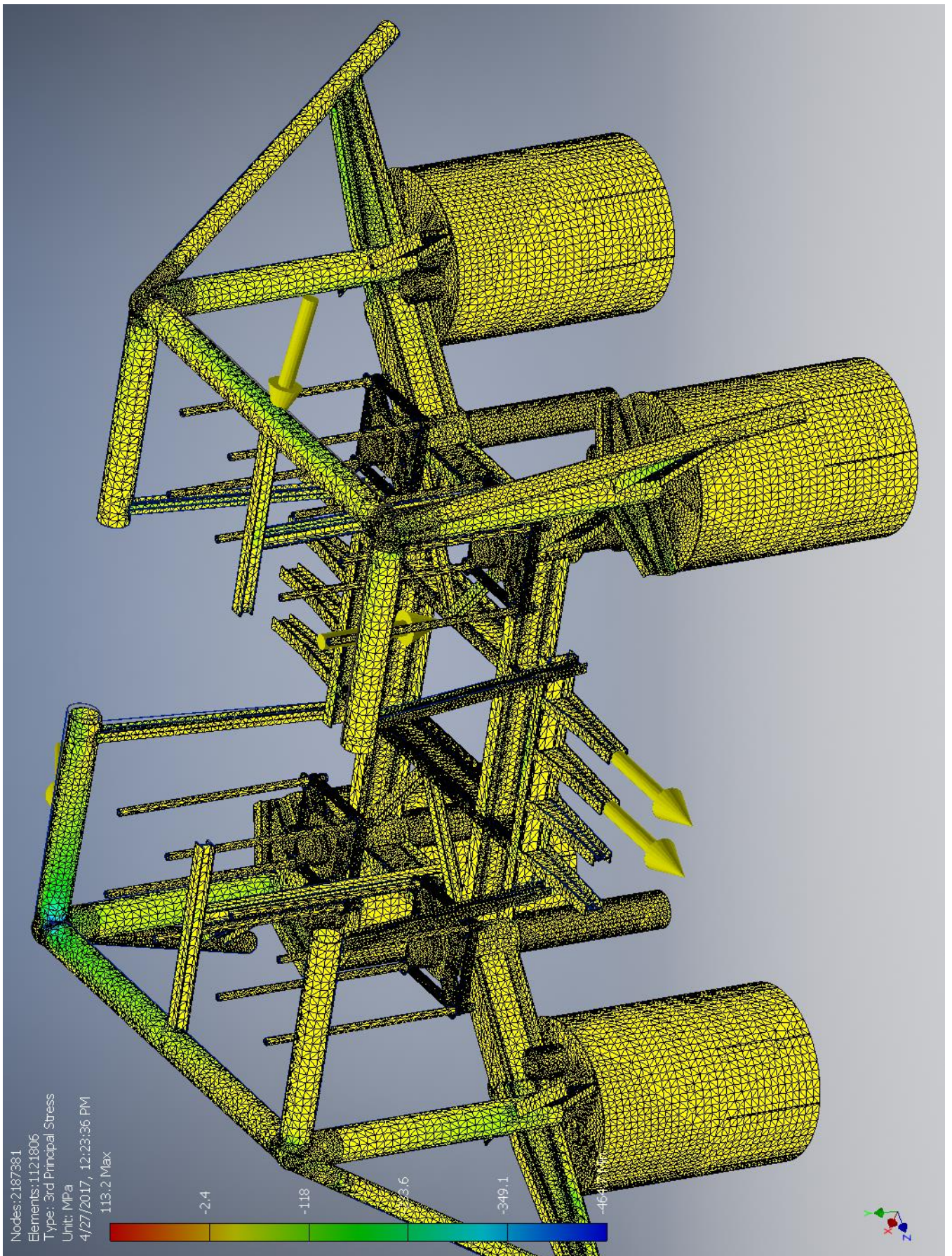
☐ Von Mises Stress



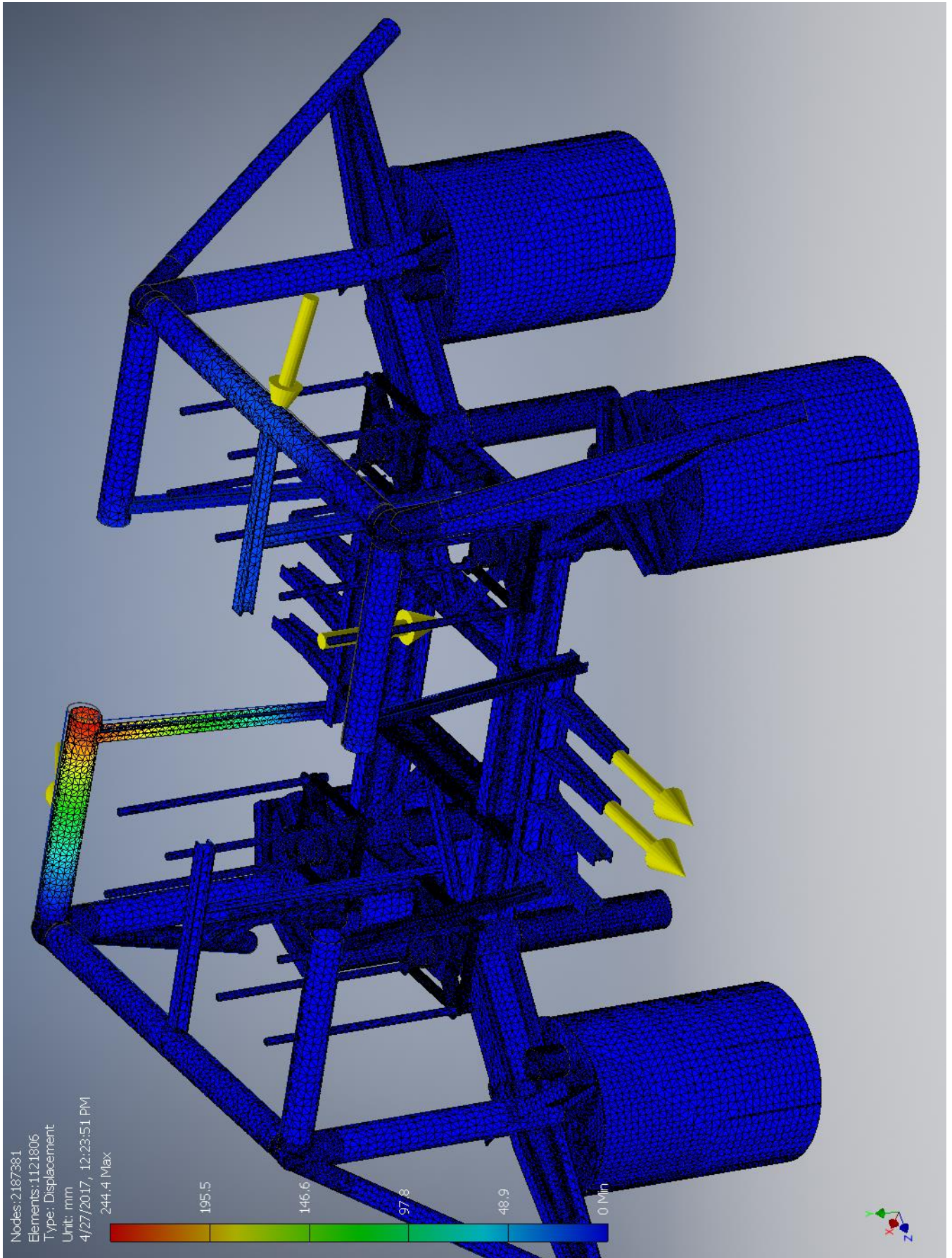
☐ 1st Principal Stress



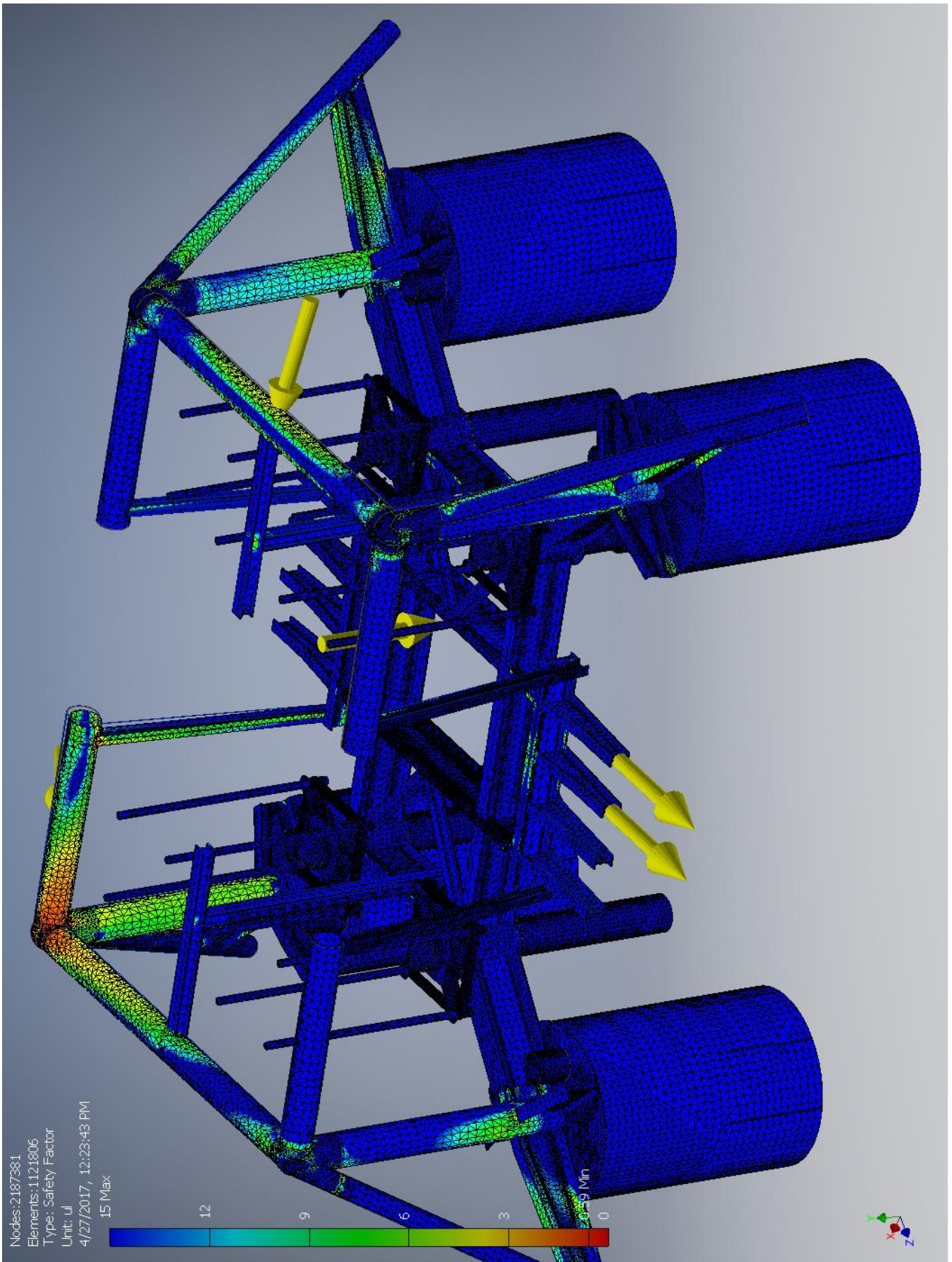
☐ 3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.3.2 With convergence on top corner

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/27/2017, 12:03 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-C

General objective and settings:

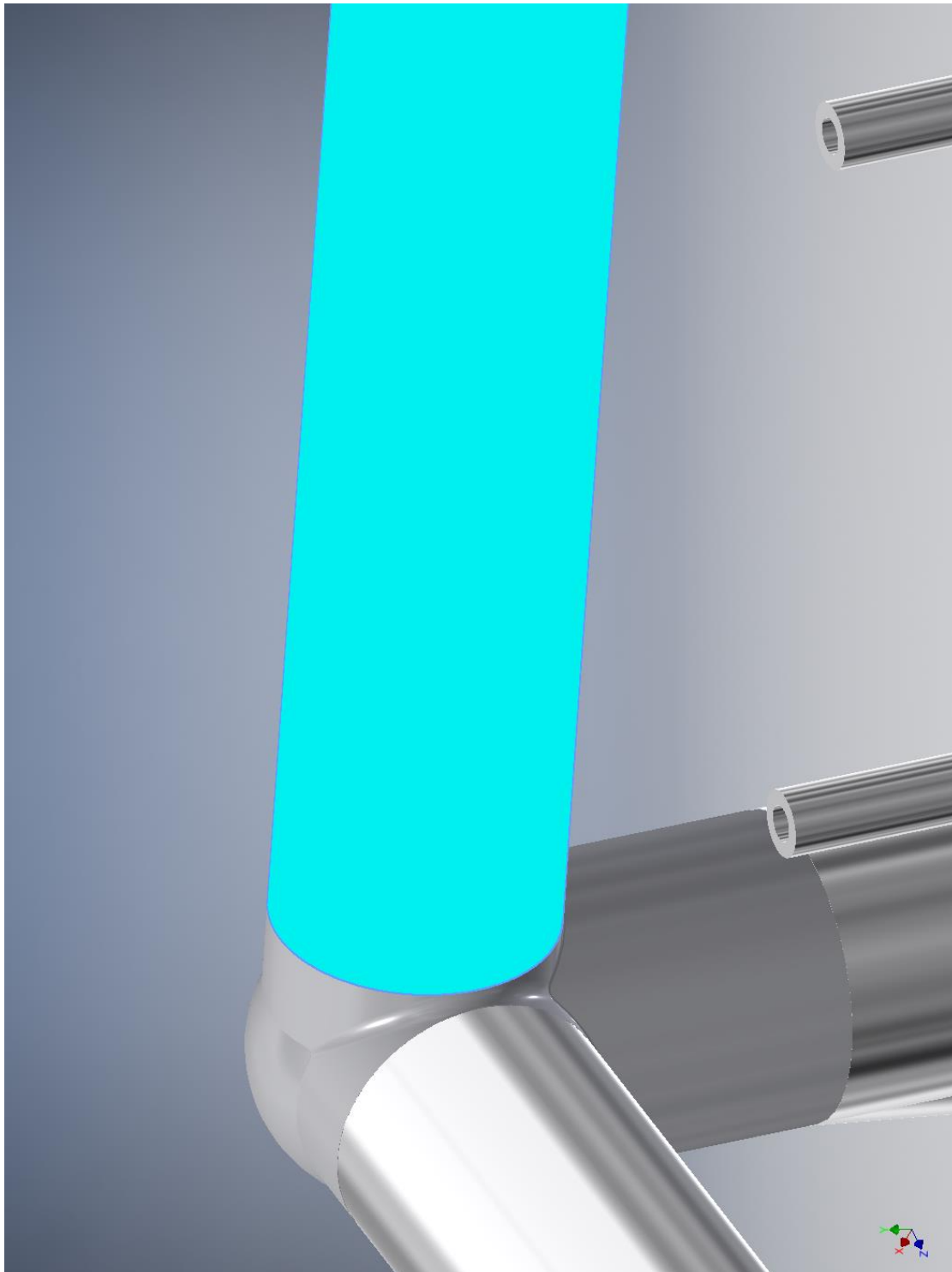
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/27/2017, 11:50 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.000 N
Vector X	936001.935 N
Vector Y	-341194.991 N
Vector Z	-86523.732 N

☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	1000000.000 N
Vector X	319421.899 N
Vector Y	439526.628 N
Vector Z	839515.333 N

☐ **Selected Face(s)**

☐ **Force:3**

Load Type	Force
Magnitude	80000.000 N
Vector X	-57568.709 N
Vector Y	-45668.912 N
Vector Z	31625.847 N

☐ **Selected Face(s)**

☐ **Force:4**

Load Type	Force
Magnitude	300000.000 N
Vector X	198108.677 N
Vector Y	-221589.070 N
Vector Z	40635.406 N

☐ **Selected Face(s)**

☐ **Force:5**

Load Type	Force
Magnitude	300000.000 N
Vector X	198108.677 N
Vector Y	-221589.070 N
Vector Z	40635.406 N

☐ **Selected Face(s)**

☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**

☐ **Results**

☐ **Reaction Force and Moment on Constraints**

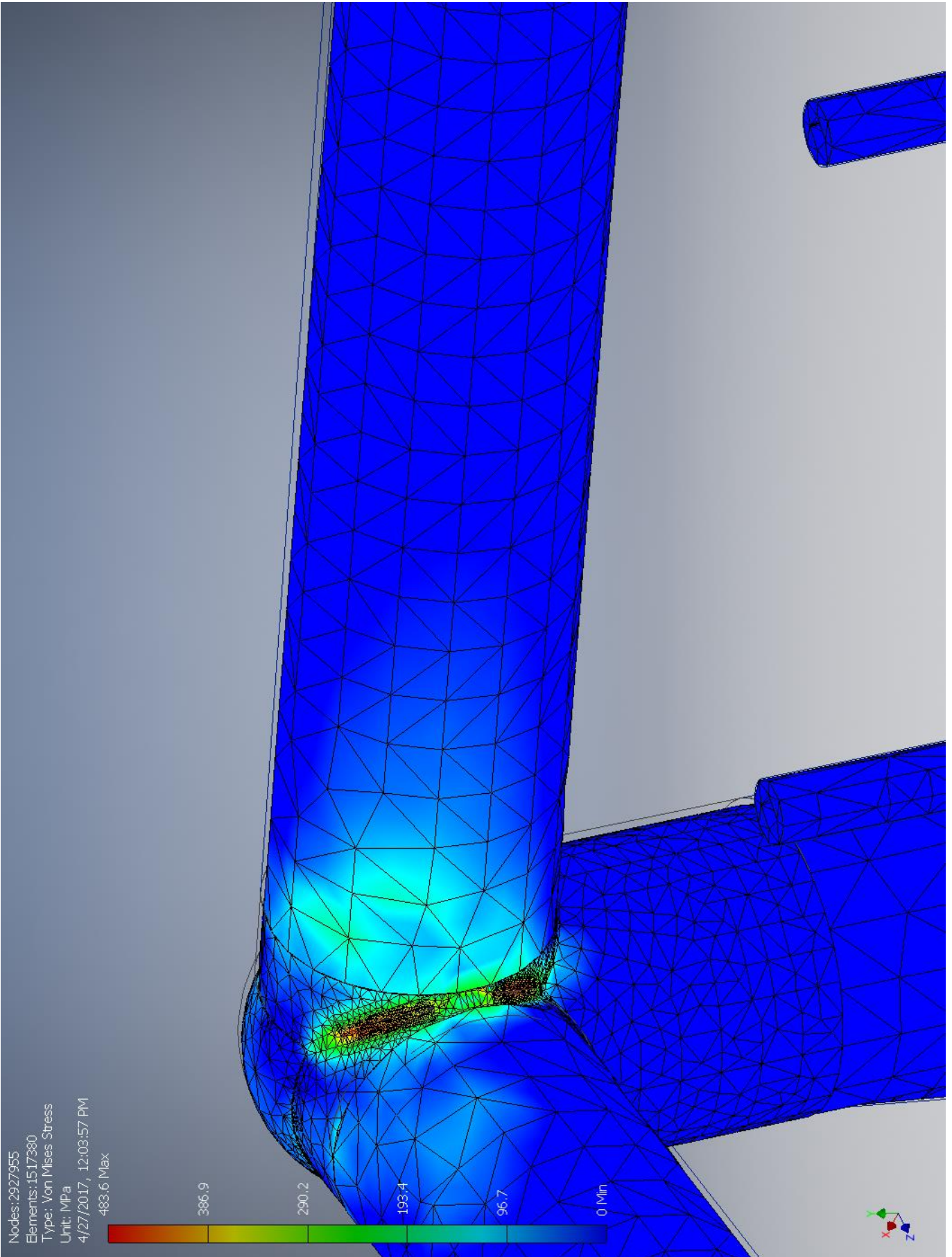
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1855620 N	-1594070 N	20792300 N m	-6465650 N m
		390515 N		11709100 N m
		-865888 N		15918900 N m

☐ **Result Summary**

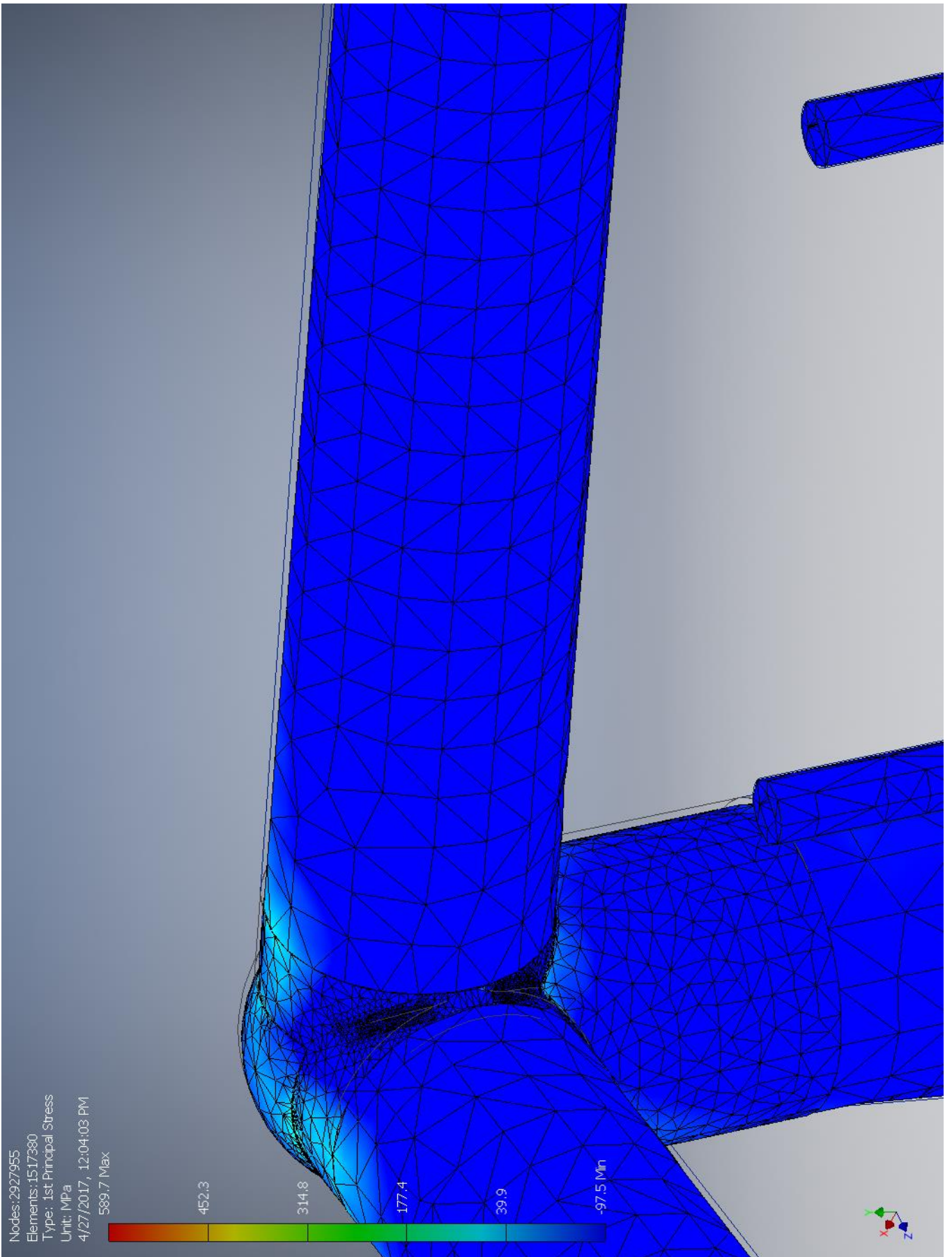
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	483.624 MPa
1st Principal Stress	-97.5239 MPa	589.698 MPa
3rd Principal Stress	-575.036 MPa	133.04 MPa
Displacement	0 mm	245.483 mm
Safety Factor	0.444559 ul	15 ul

☐ **Figures**

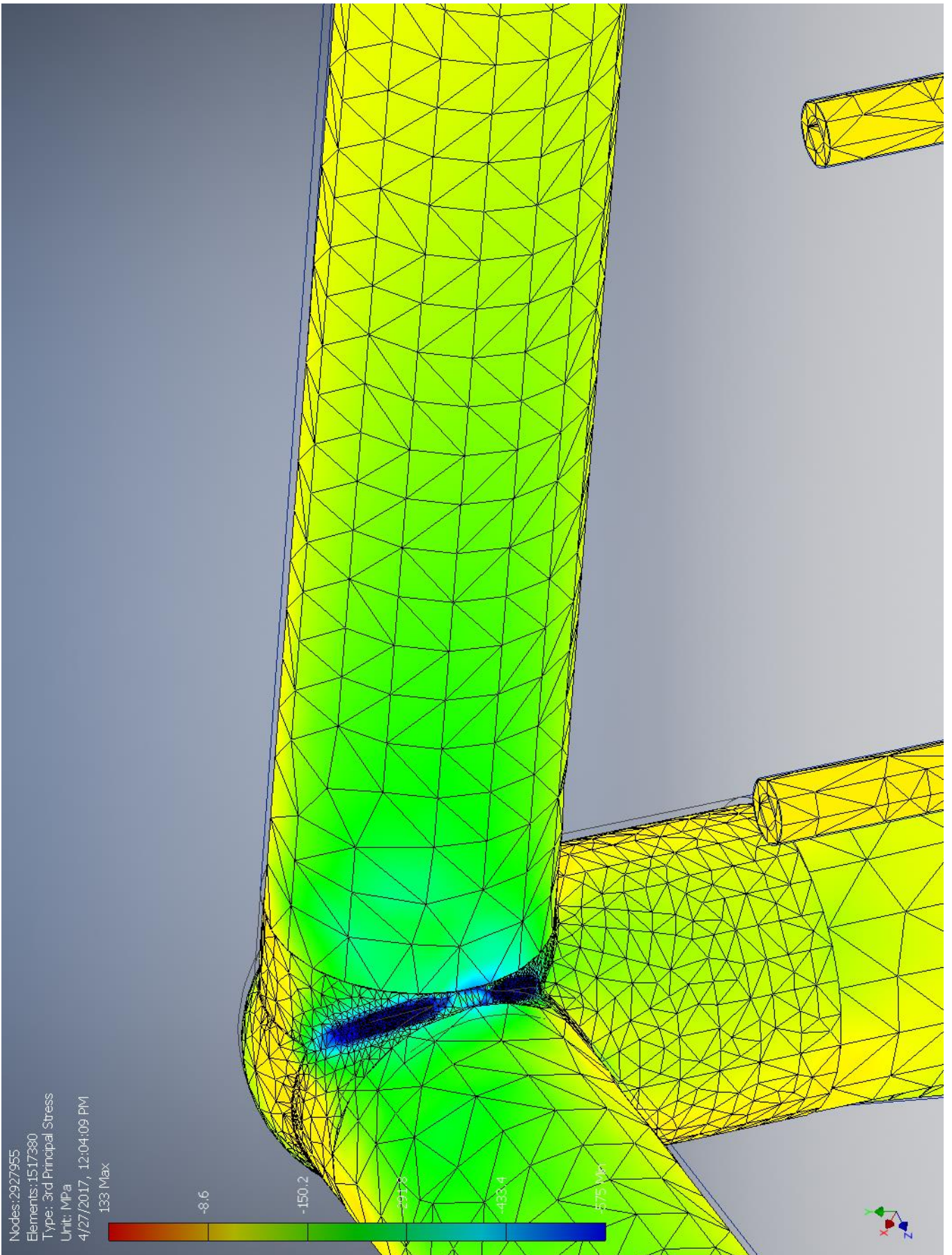
☐ Von Mises Stress



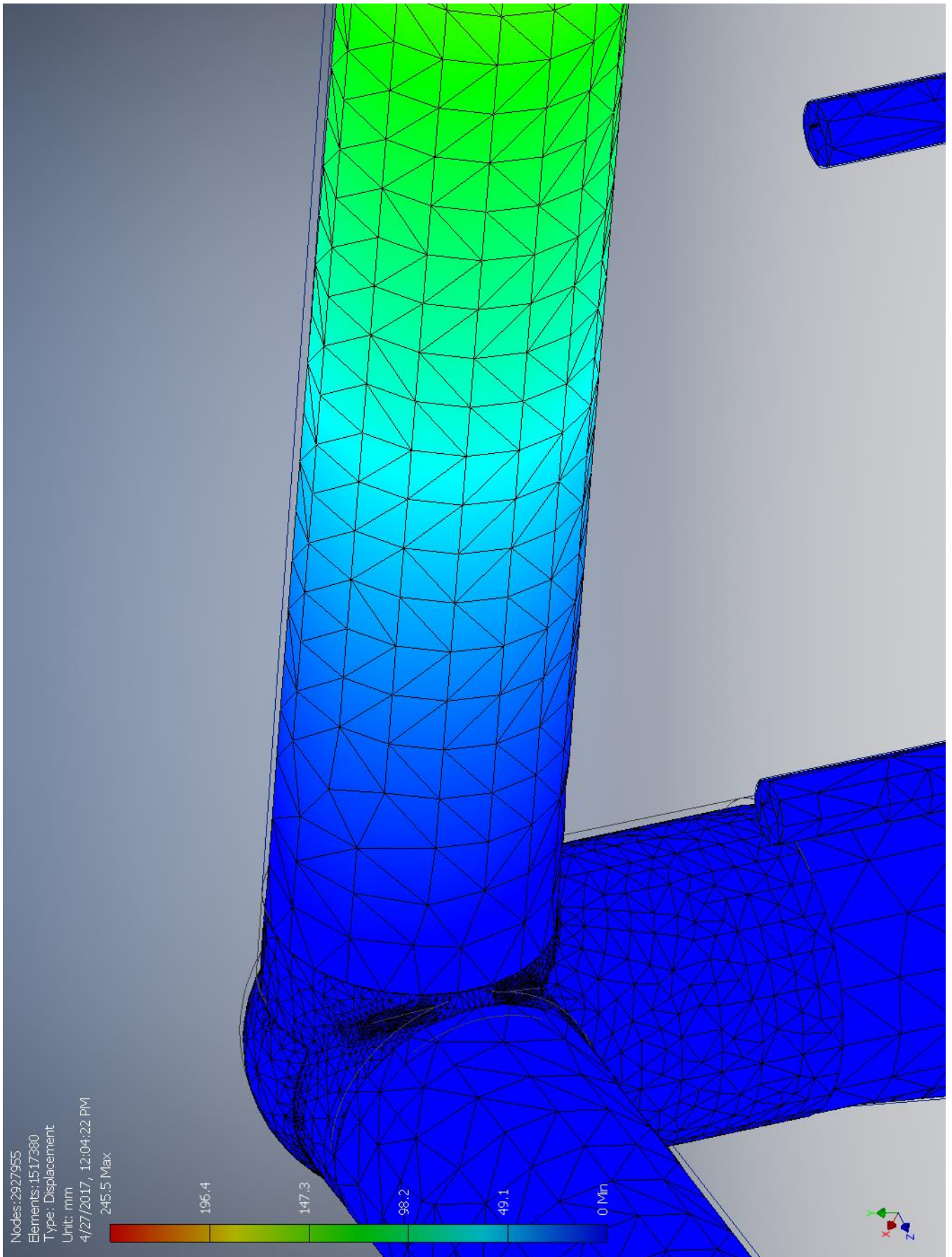
1st Principal Stress



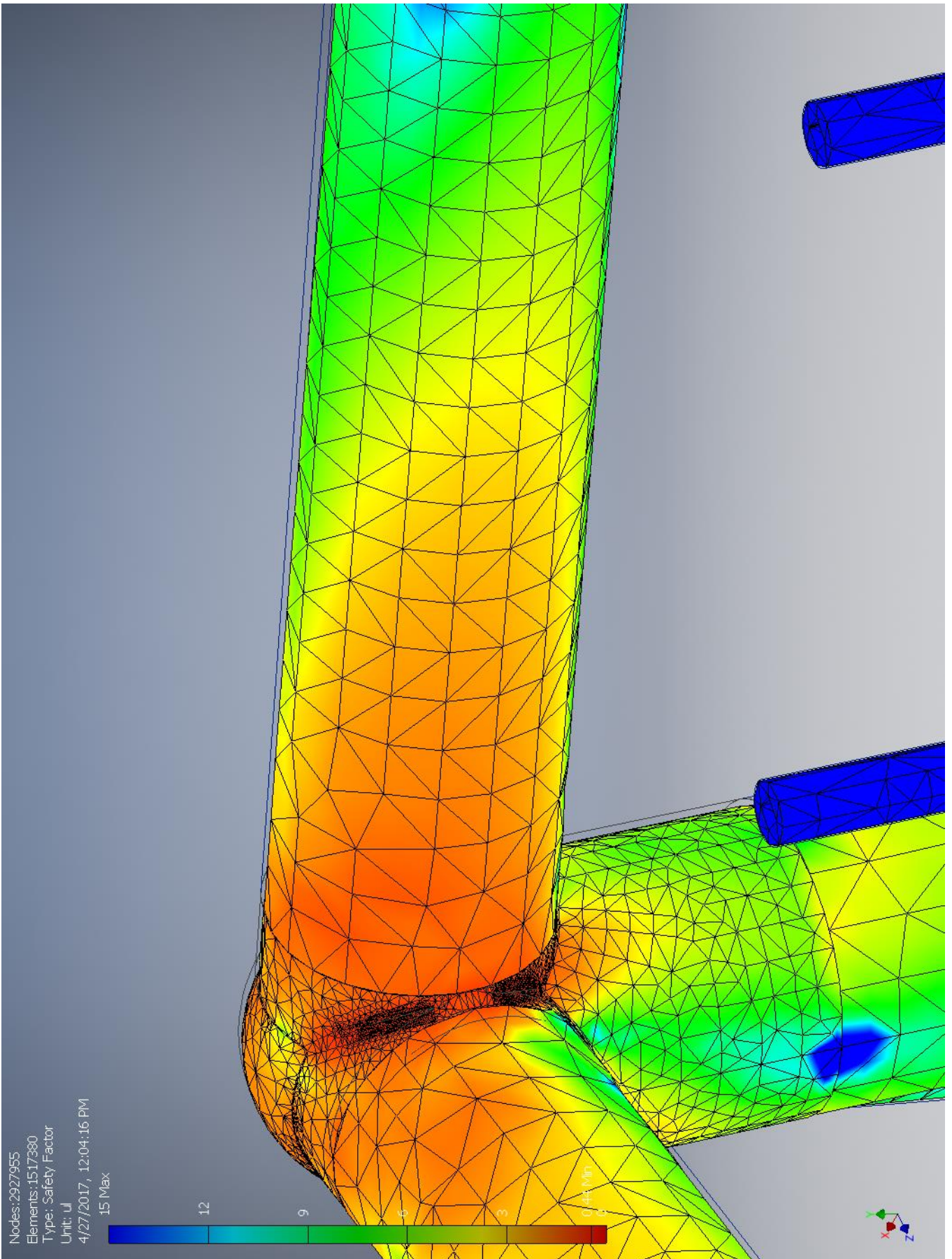
3d Principal Stress



☐ Displacement



☐ Safety Factor



I.3.3 With convergence on support beam

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/27/2017, 11:57 AM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
----------	----------

☐ Physical

Mass	182983 kg
Area	3.56846E+09 mm ²
Volume	6.09185E+10 mm ³
Center of Gravity	x=-41359.7 mm y=-19499.7 mm z=32758.2 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-C

General objective and settings:

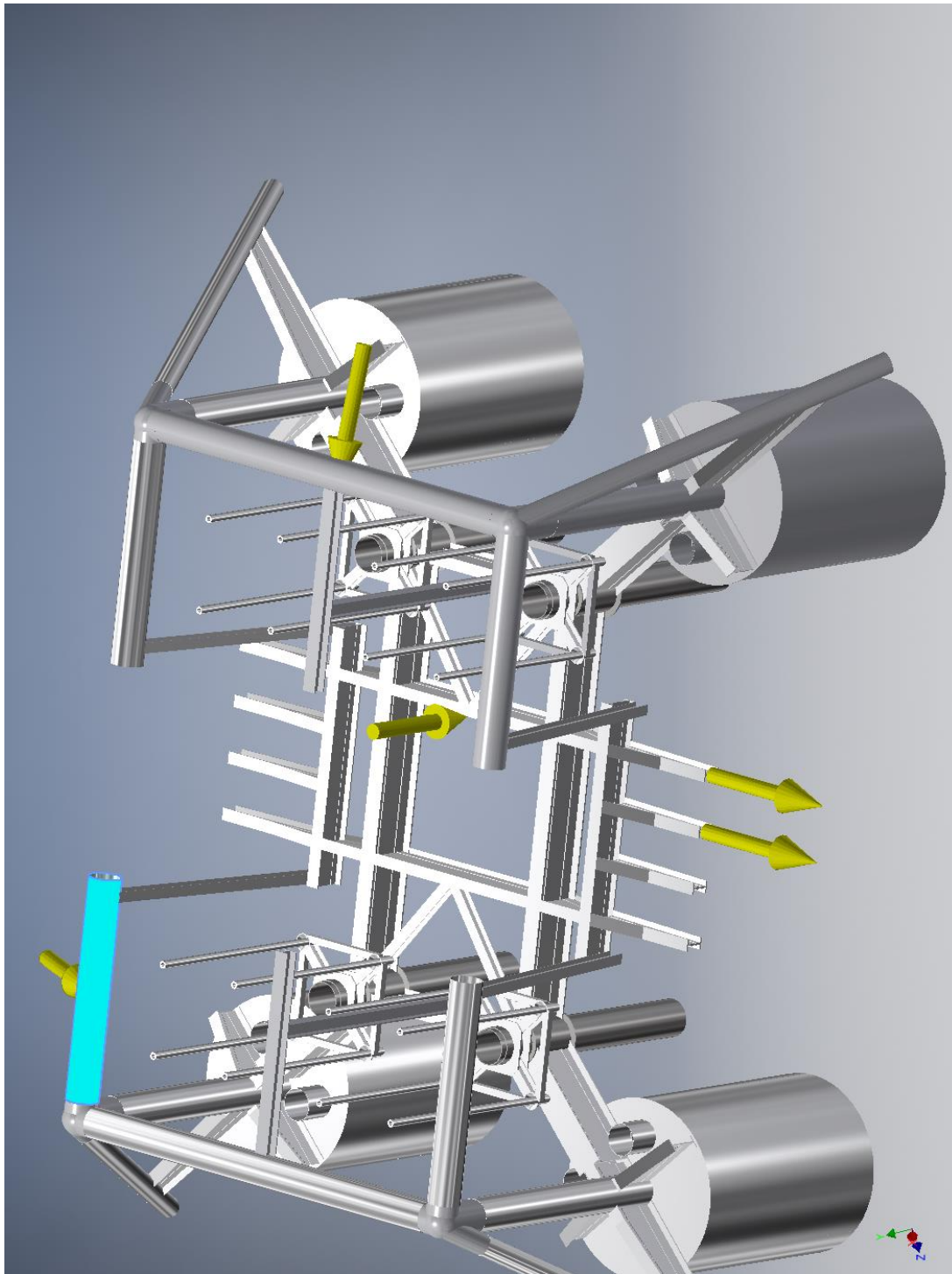
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/27/2017, 11:50 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.000 N
Vector X	936001.935 N
Vector Y	-341194.991 N
Vector Z	-86523.732 N

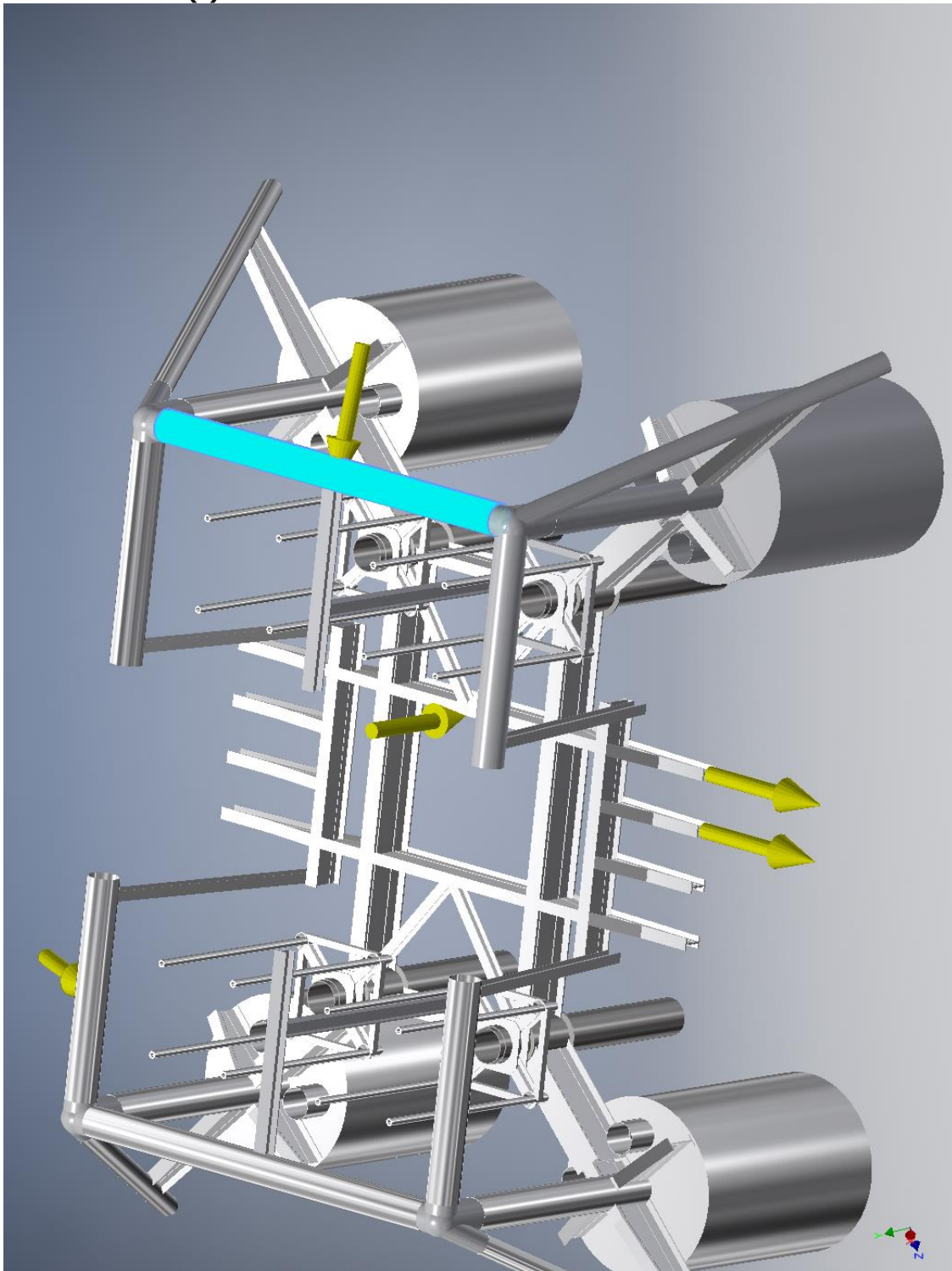
☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	1000000.000 N
Vector X	319421.899 N
Vector Y	439526.628 N
Vector Z	839515.333 N

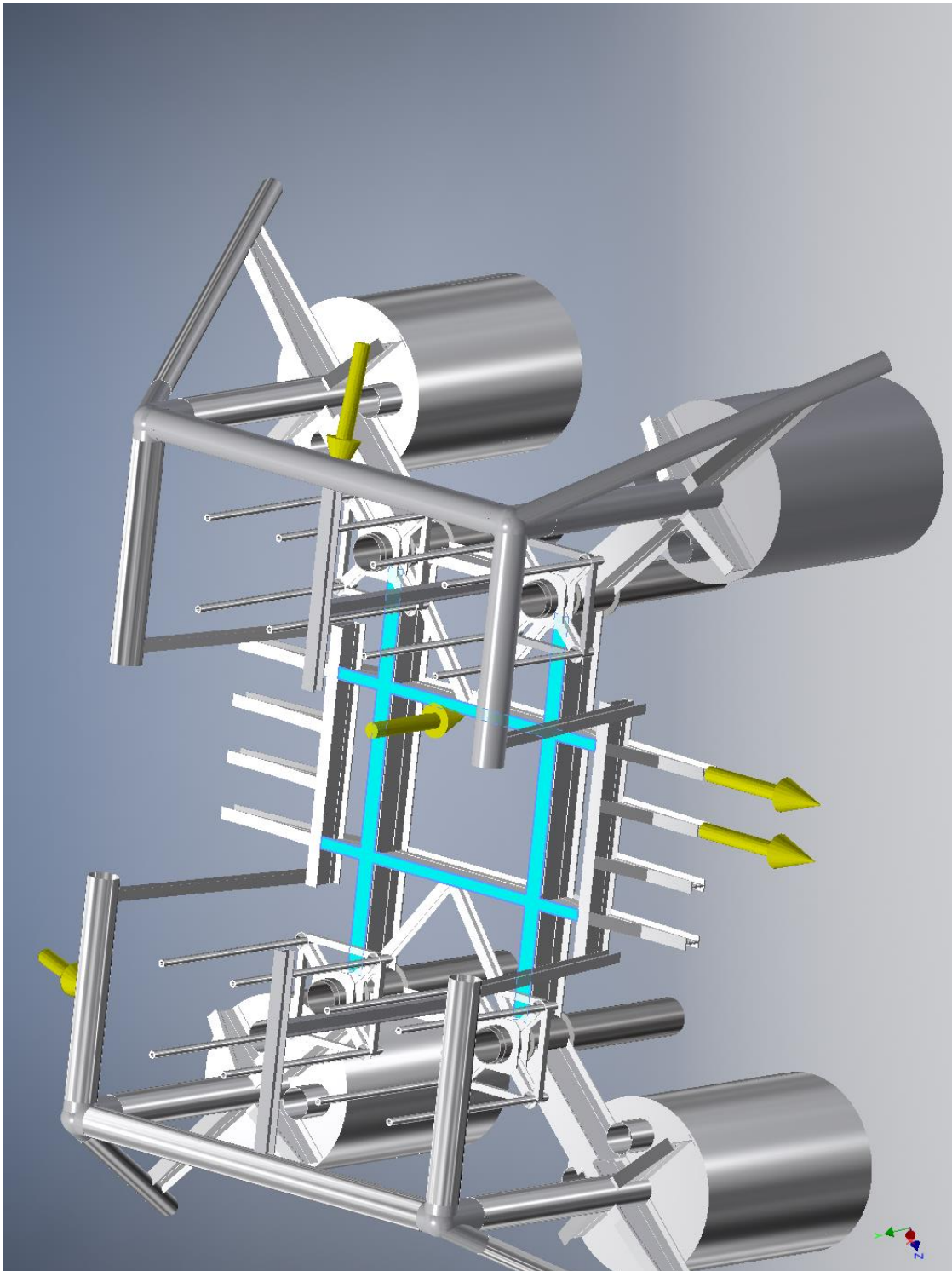
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	80000.000 N
Vector X	-57568.709 N
Vector Y	-45668.912 N
Vector Z	31625.847 N

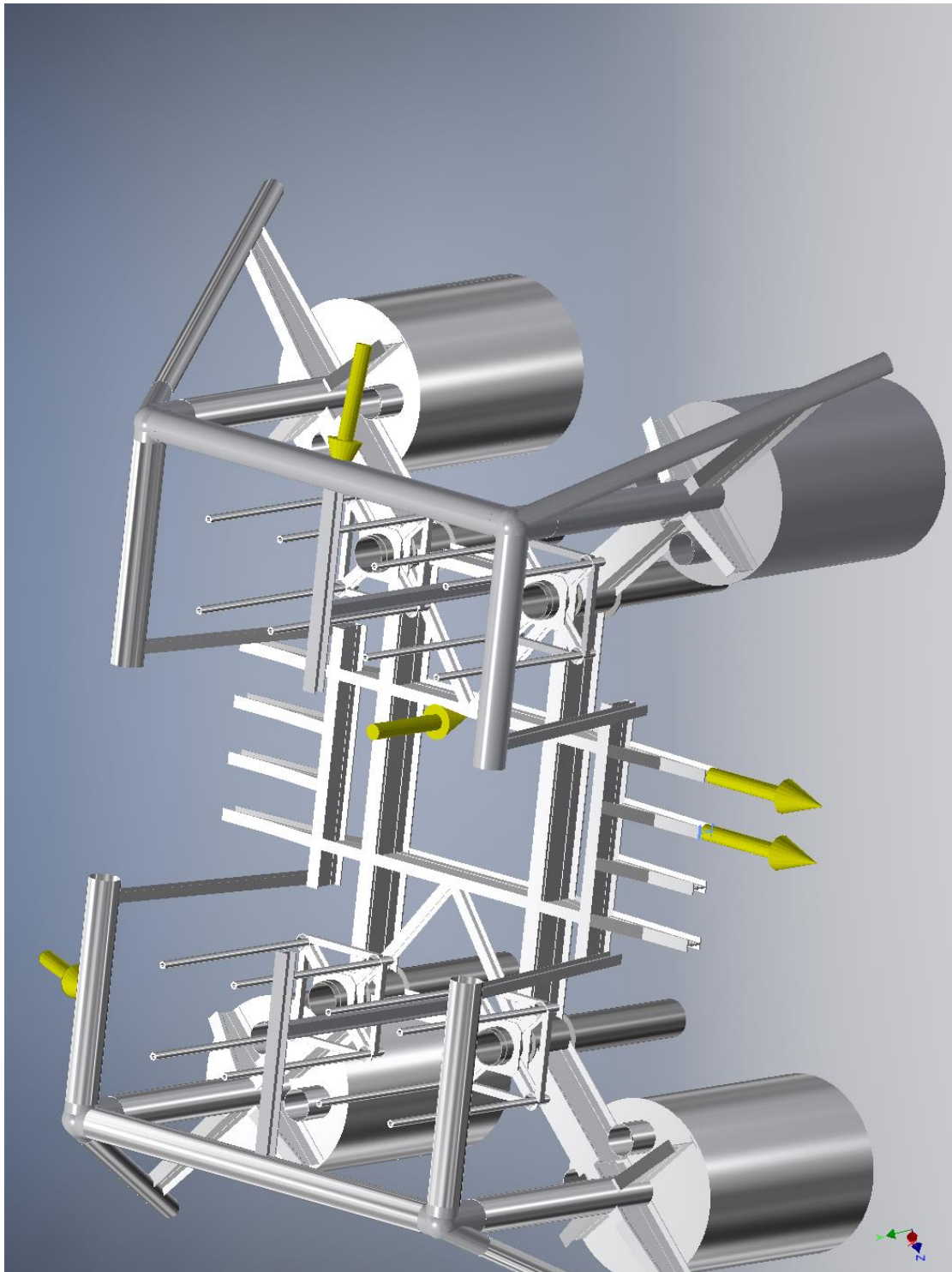
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	300000.000 N
Vector X	198108.677 N
Vector Y	-221589.070 N
Vector Z	40635.406 N

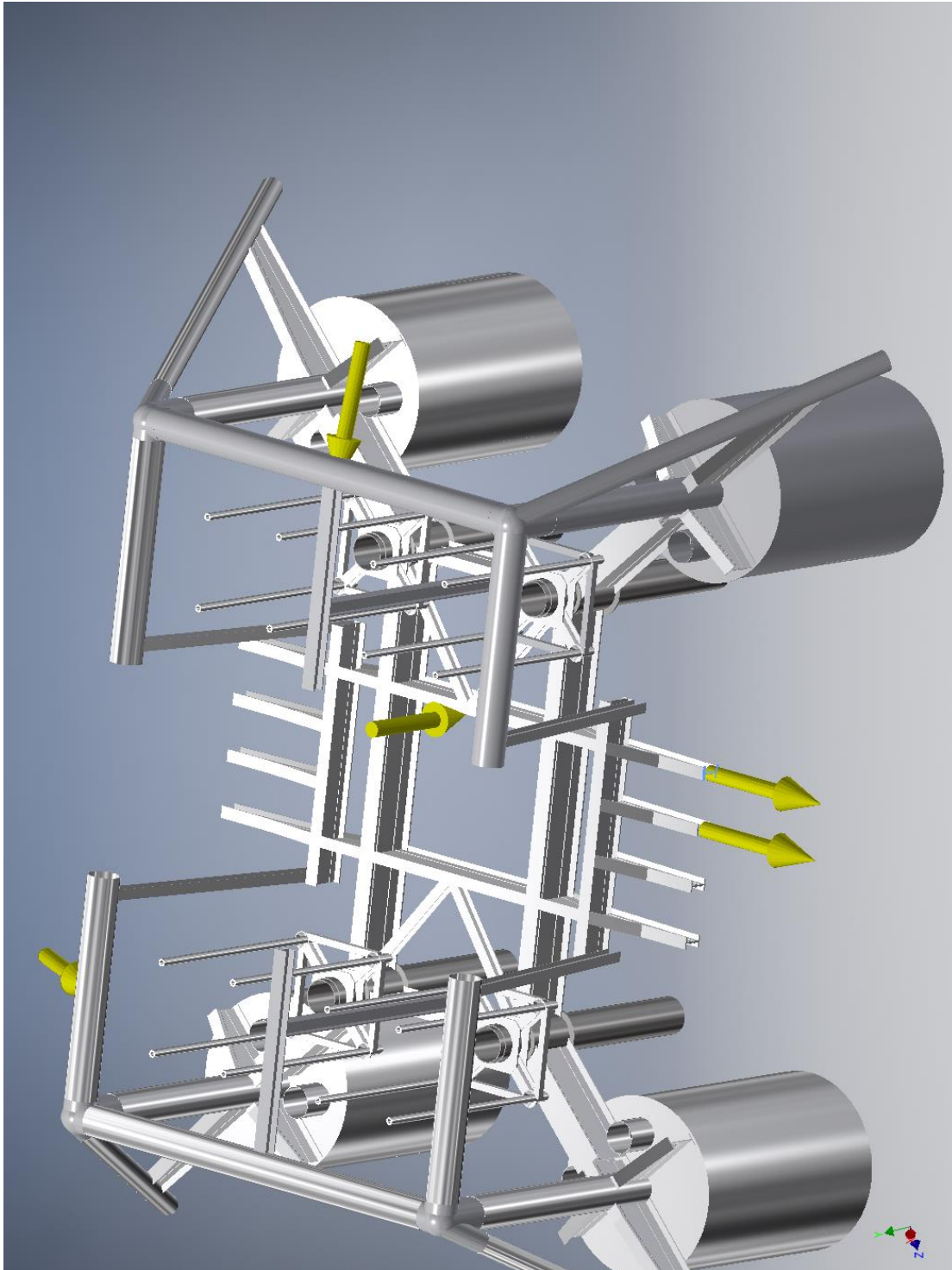
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	300000.000 N
Vector X	198108.677 N
Vector Y	-221589.070 N
Vector Z	40635.406 N

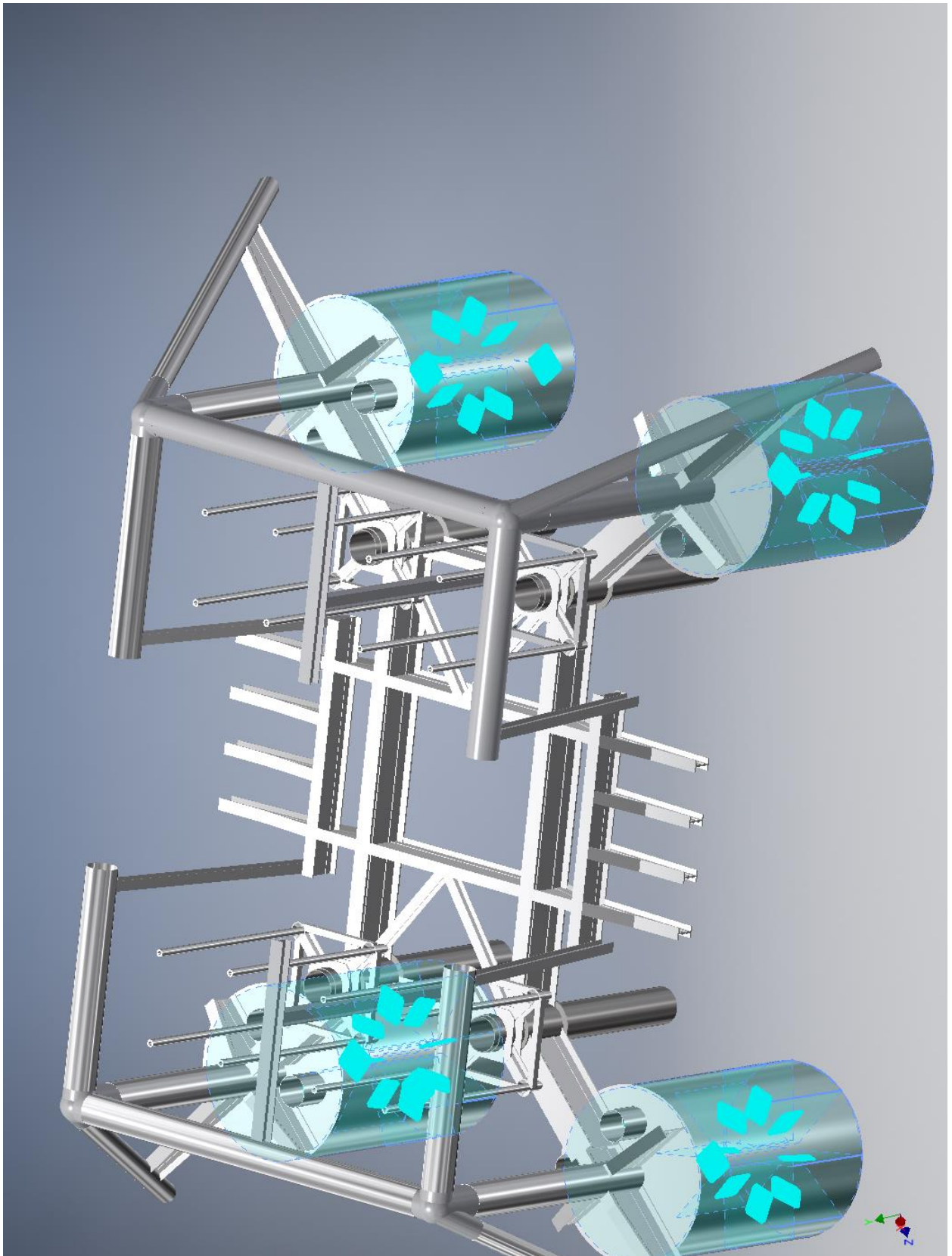
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

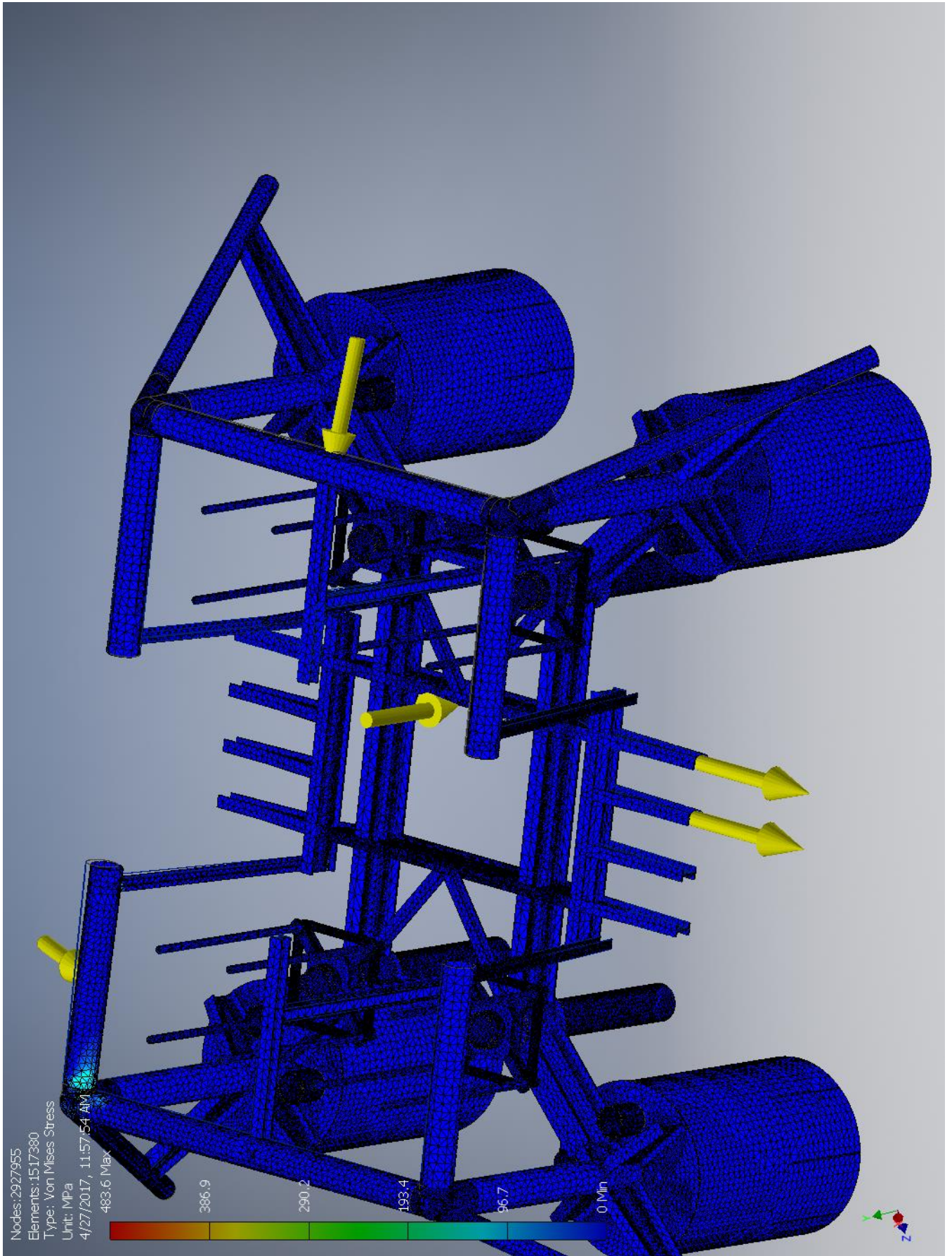
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1855620 N	-1594070 N	20792300 N m	-6465650 N m
		390515 N		11709100 N m
		-865888 N		15918900 N m

☐ Result Summary

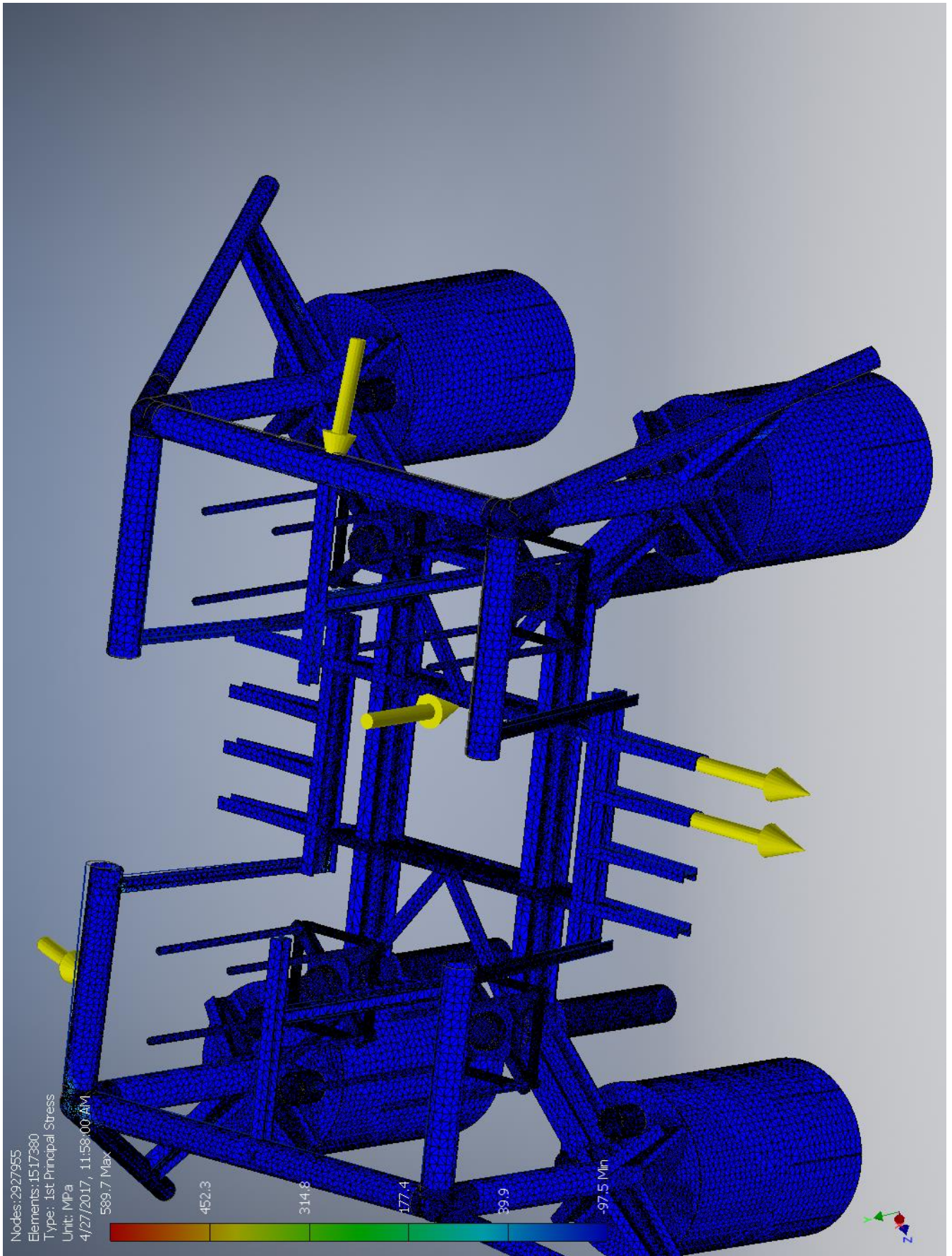
Name	Minimum	Maximum
Volume	6.09185E+10 mm ³	
Mass	182983 kg	
Von Mises Stress	0 MPa	483.624 MPa
1st Principal Stress	-97.5239 MPa	589.698 MPa
3rd Principal Stress	-575.036 MPa	133.04 MPa
Displacement	0 mm	245.483 mm
Safety Factor	0.444559 ul	15 ul

☐ Figures

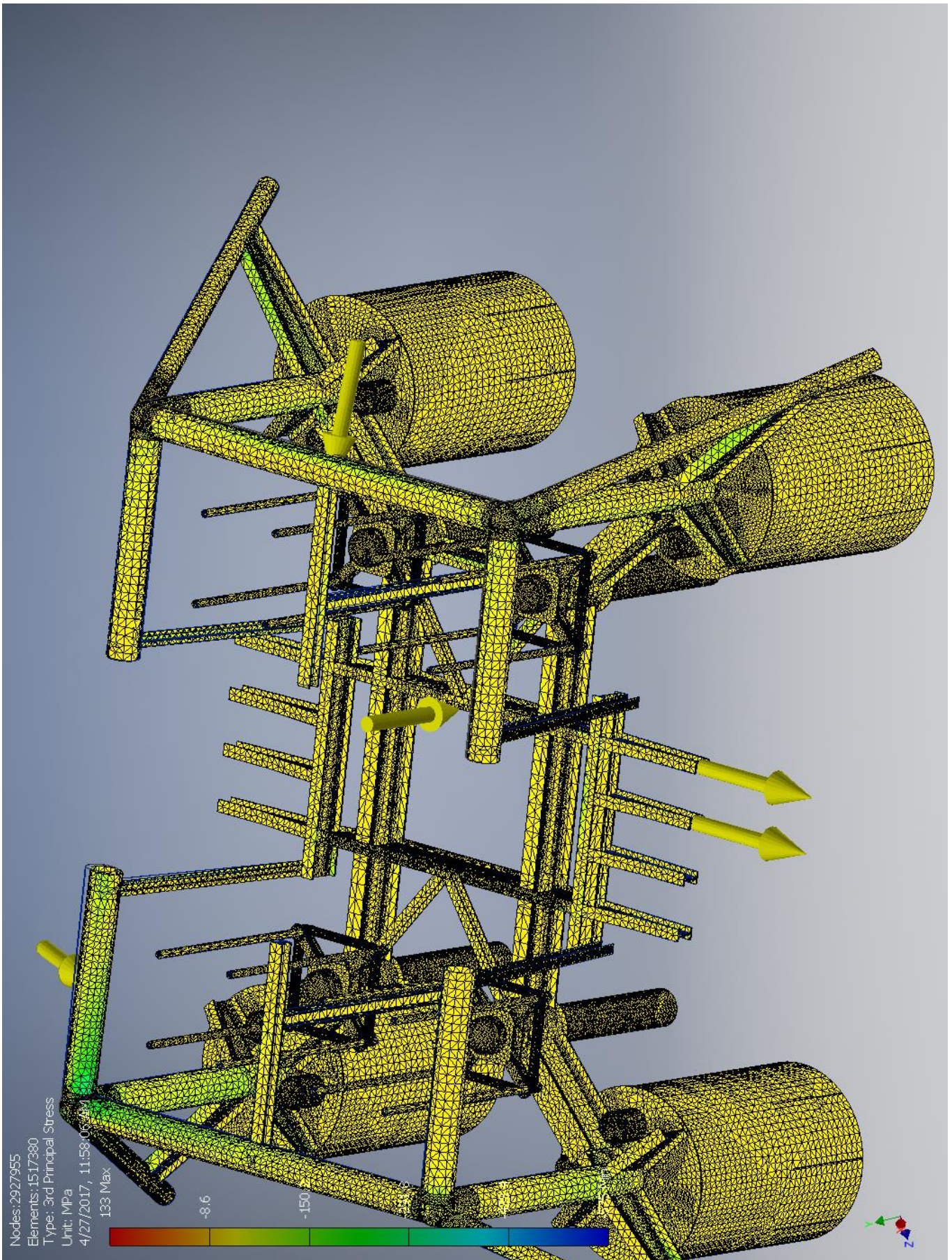
☐ Von Mises Stress



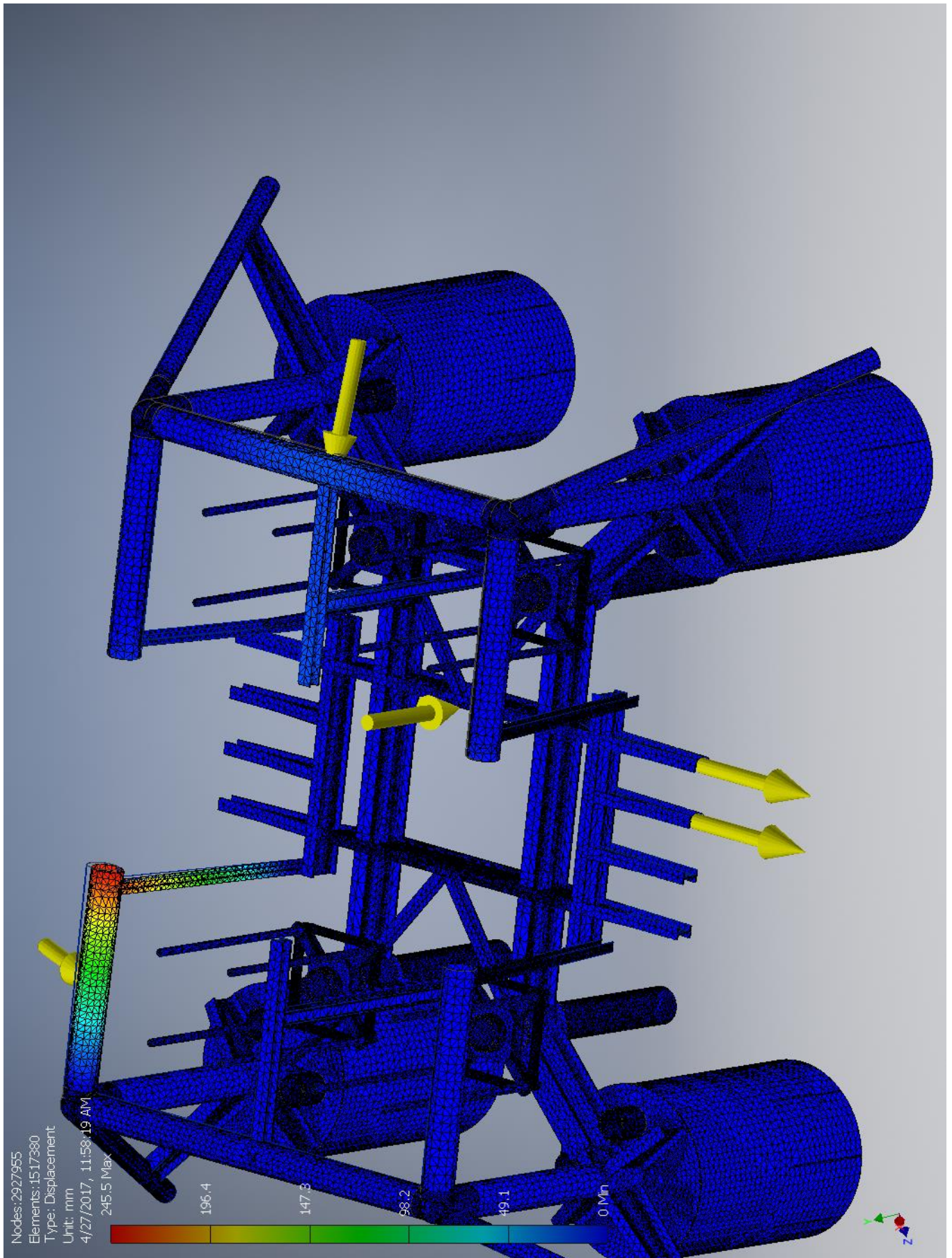
☐ 1st Principal Stress



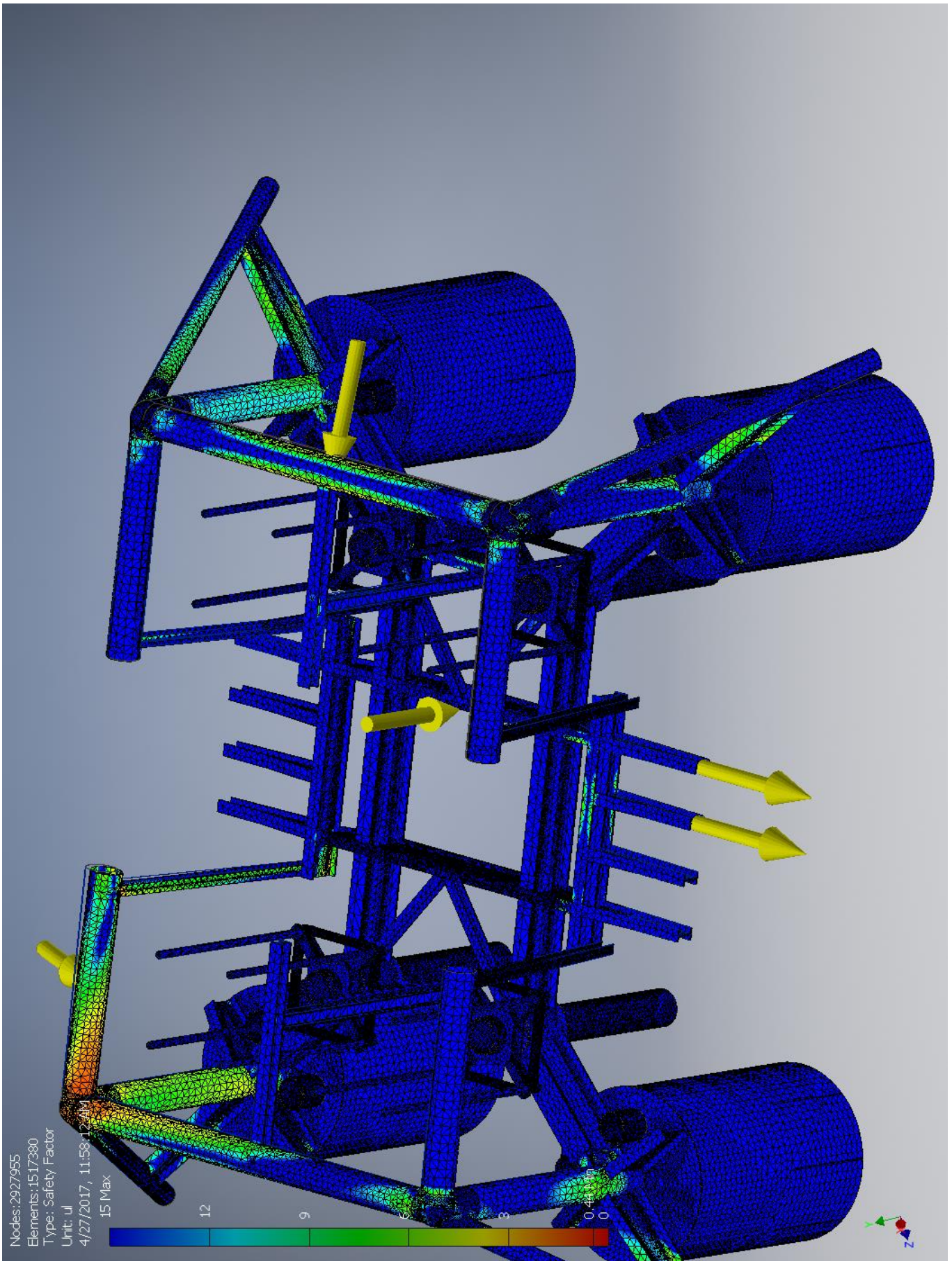
☐ 3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.3.4 Overtrawlable/snagfree design load, 400kN instead of 1000kN.

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	5/1/2017, 10:16 AM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	181171 kg
Area	3.52908E+09 mm ²
Volume	6.02475E+10 mm ³
Center of Gravity	x=-8288.42 mm y=6548.24 mm z=9857.84 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-C

General objective and settings:

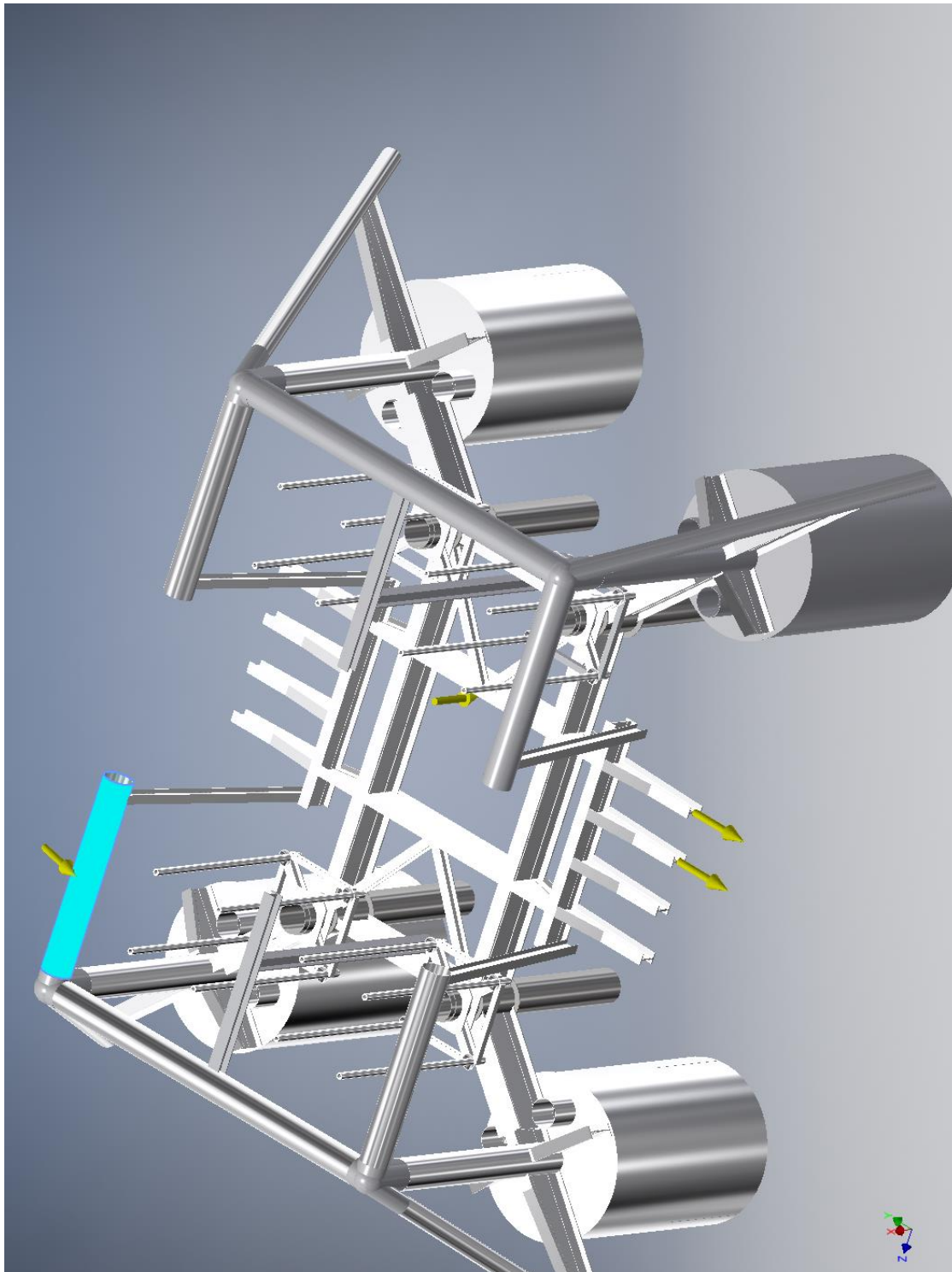
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	5/1/2017, 10:11 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	400000.000 N
Vector X	130246.406 N
Vector Y	-378200.838 N
Vector Z	0.000 N

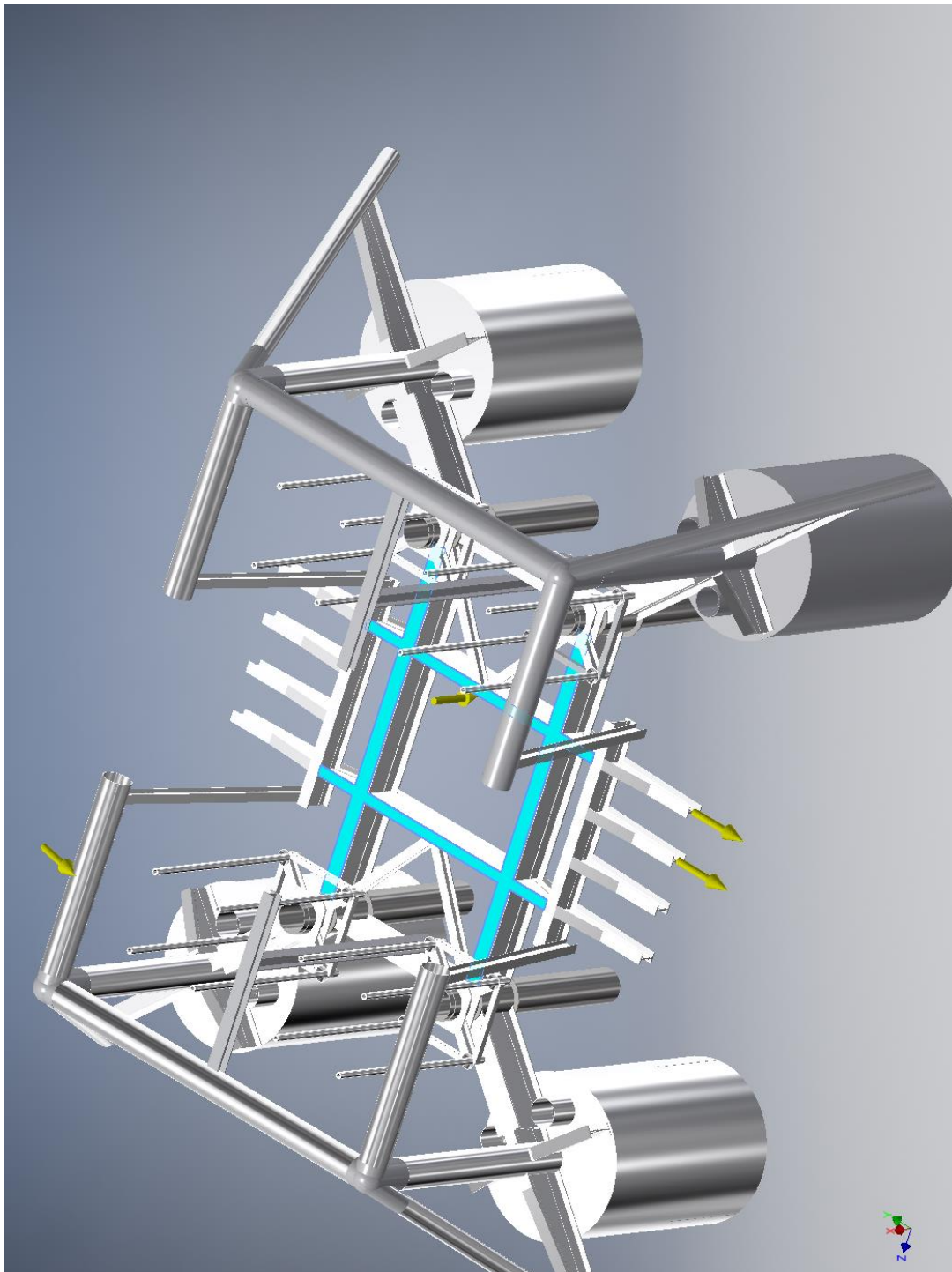
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	800000.000 N
Vector X	-800000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

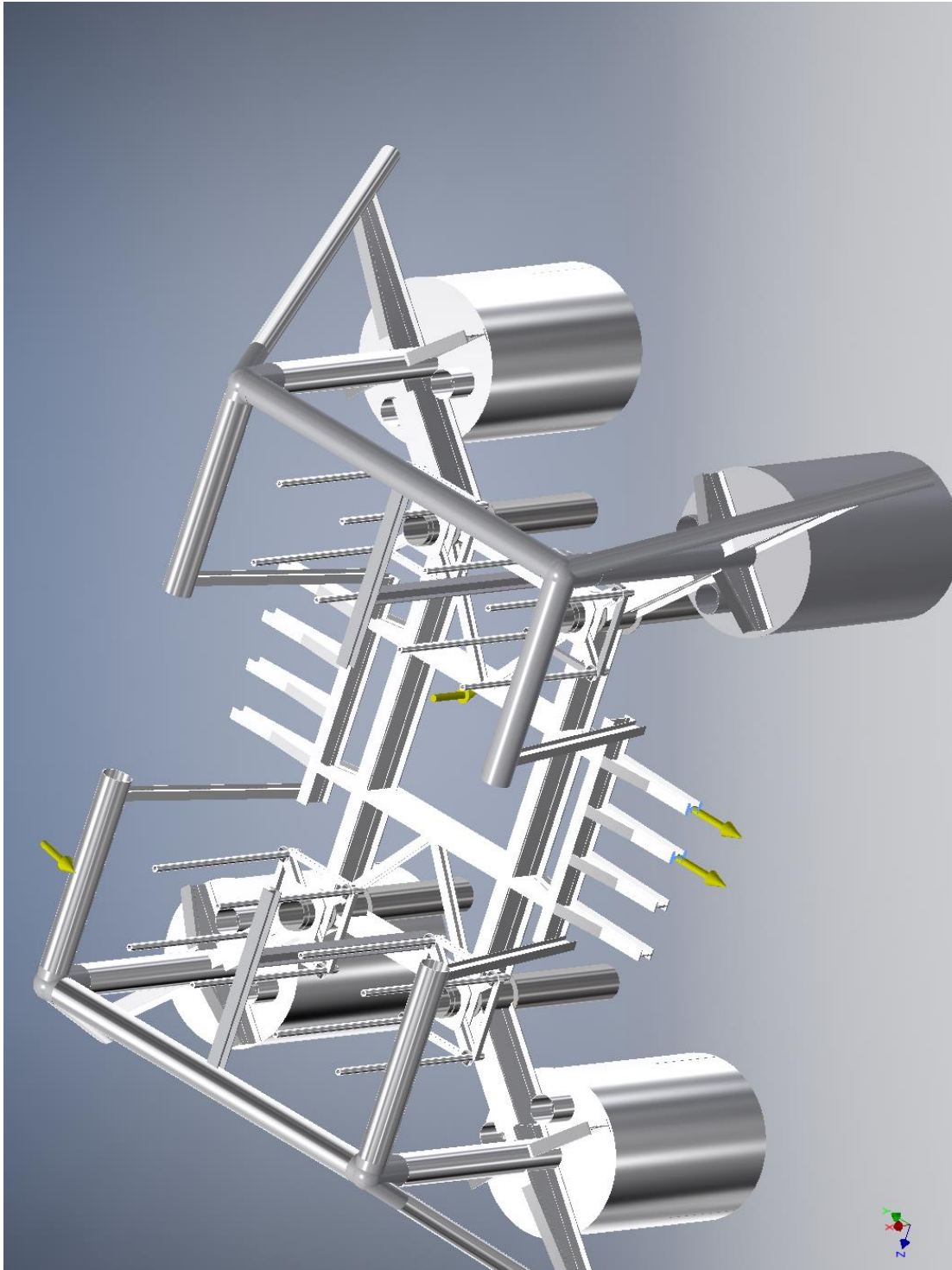
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	-300000.000 N
Vector Z	0.000 N

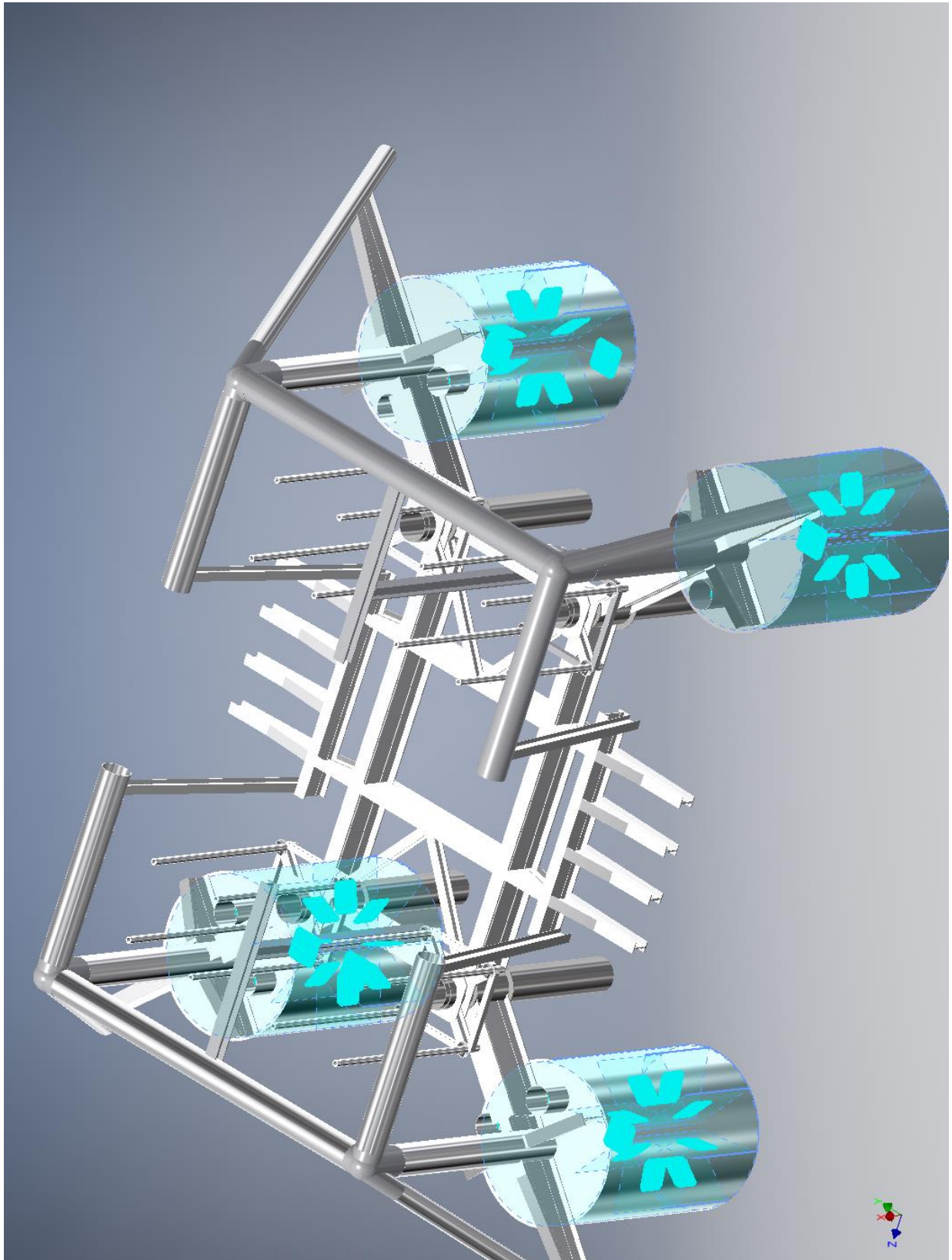
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

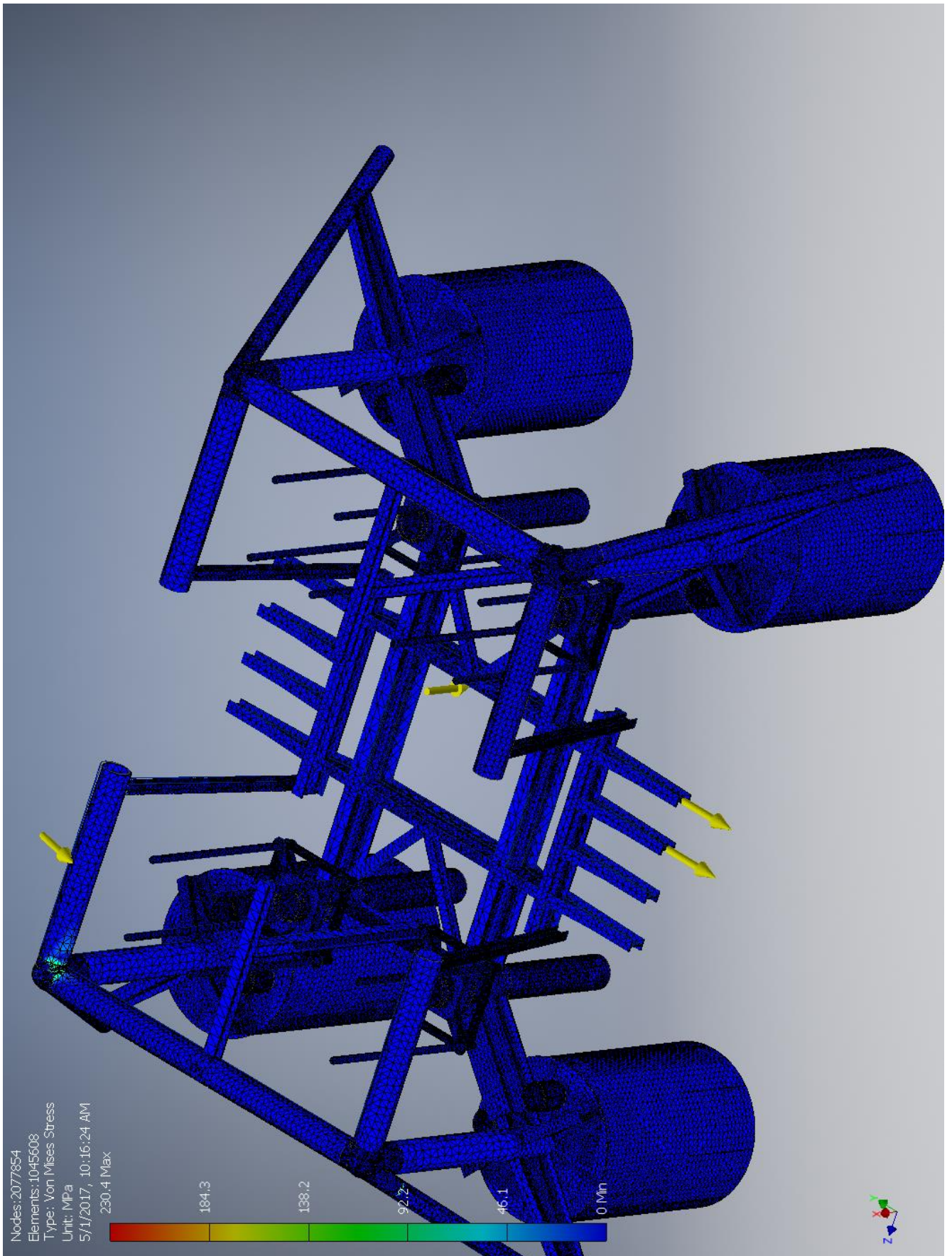
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	953166 N	669754 N	7388730 N m	-1858890 N m
		678201 N		-890576 N m
		0 N		7095400 N m

☐ Result Summary

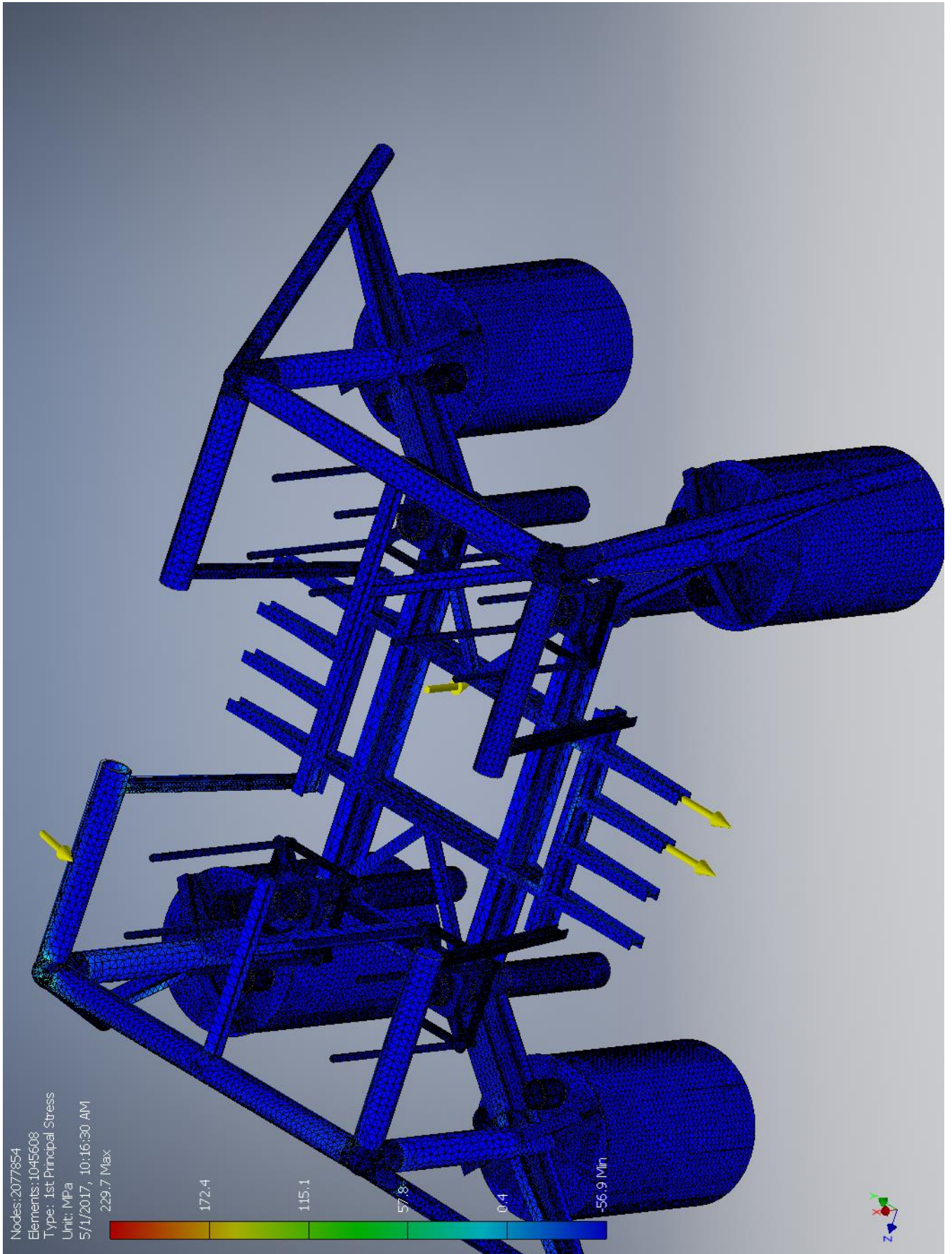
Name	Minimum	Maximum
Volume	6.02475E+10 mm ³	
Mass	181171 kg	
Von Mises Stress	0 MPa	230.383 MPa
1st Principal Stress	-56.8608 MPa	229.68 MPa
3rd Principal Stress	-222.992 MPa	70.3542 MPa
Displacement	0 mm	101.754 mm
Safety Factor	0.95315 ul	15 ul

☐ Figures

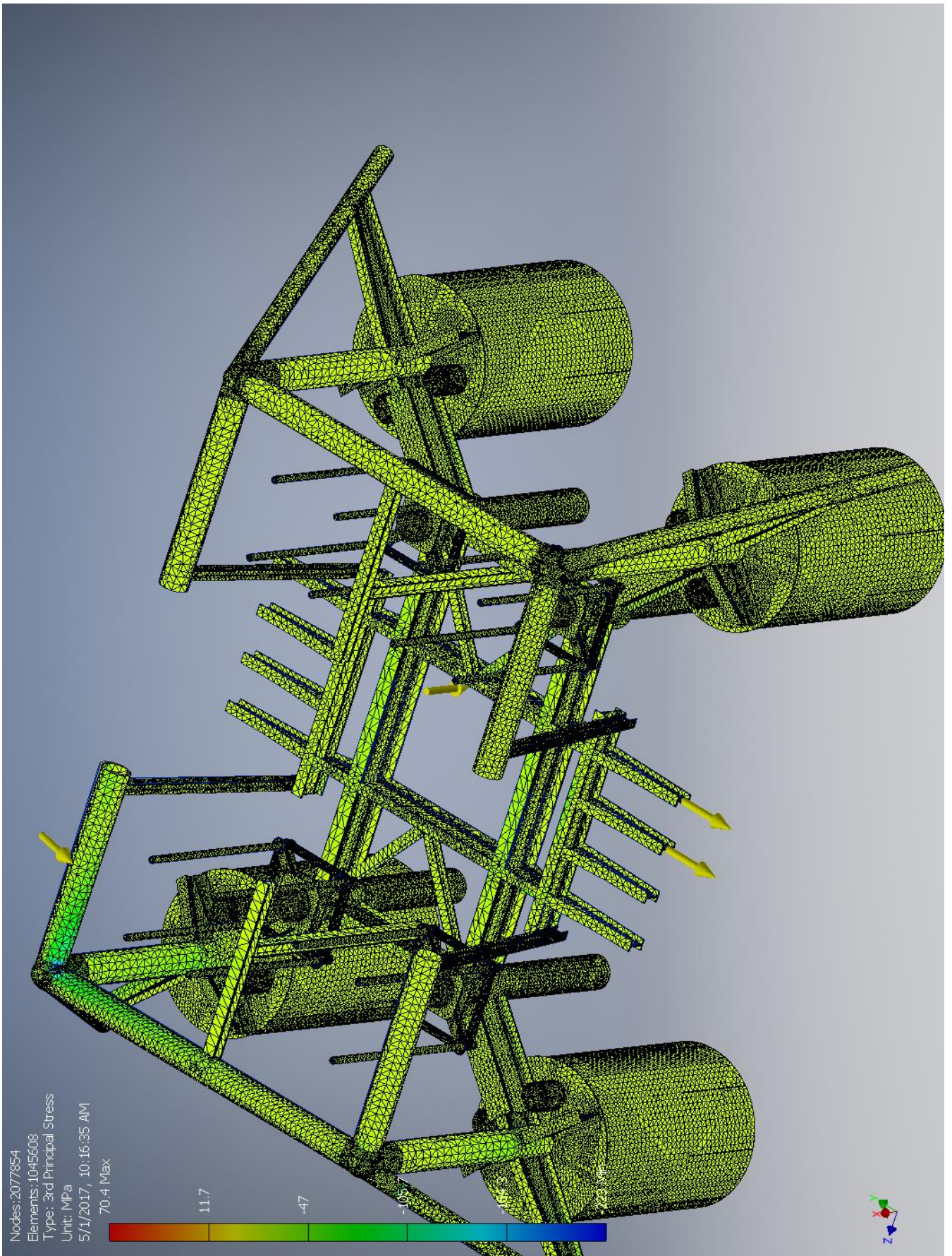
☐ Von Mises Stress



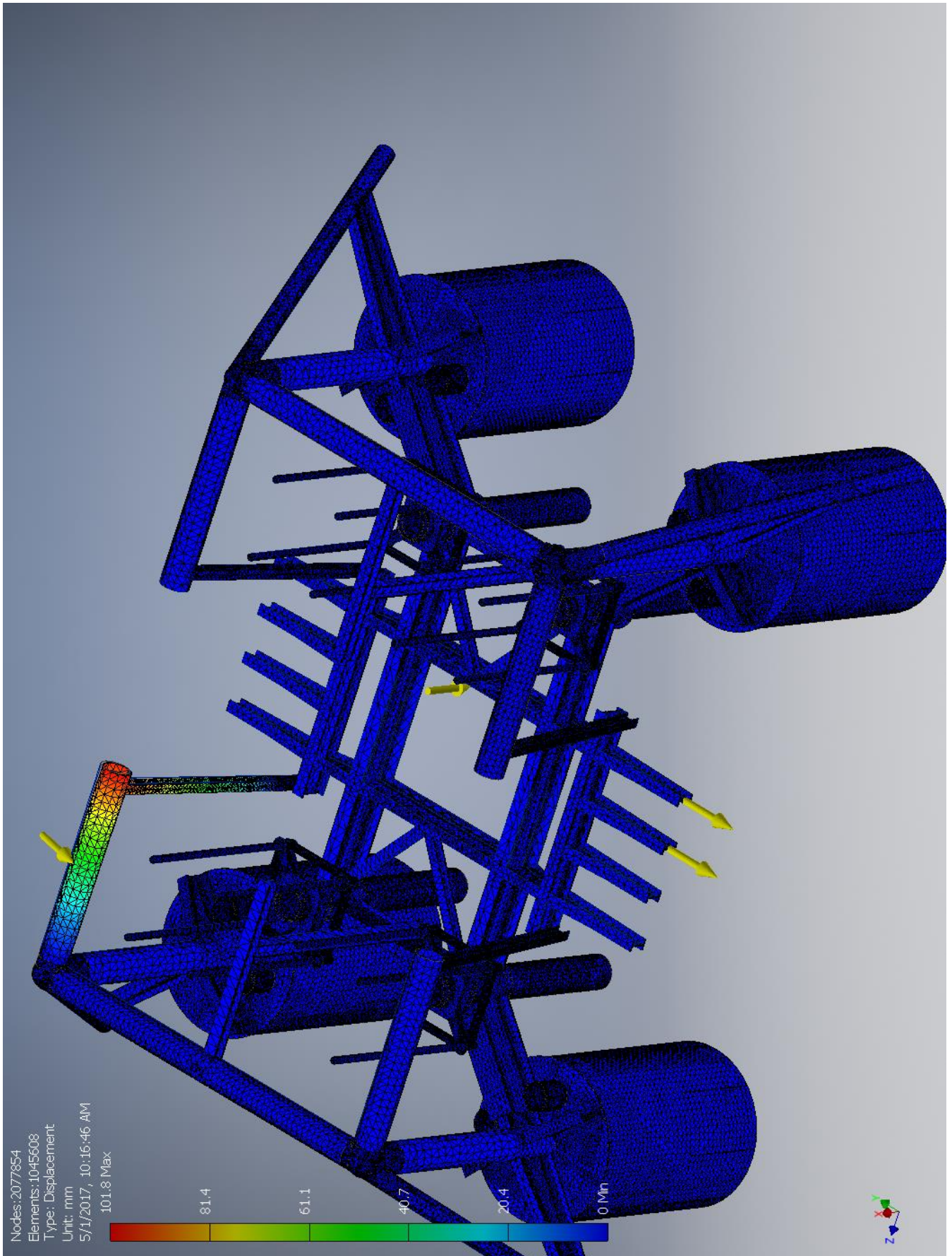
1st Principal Stress



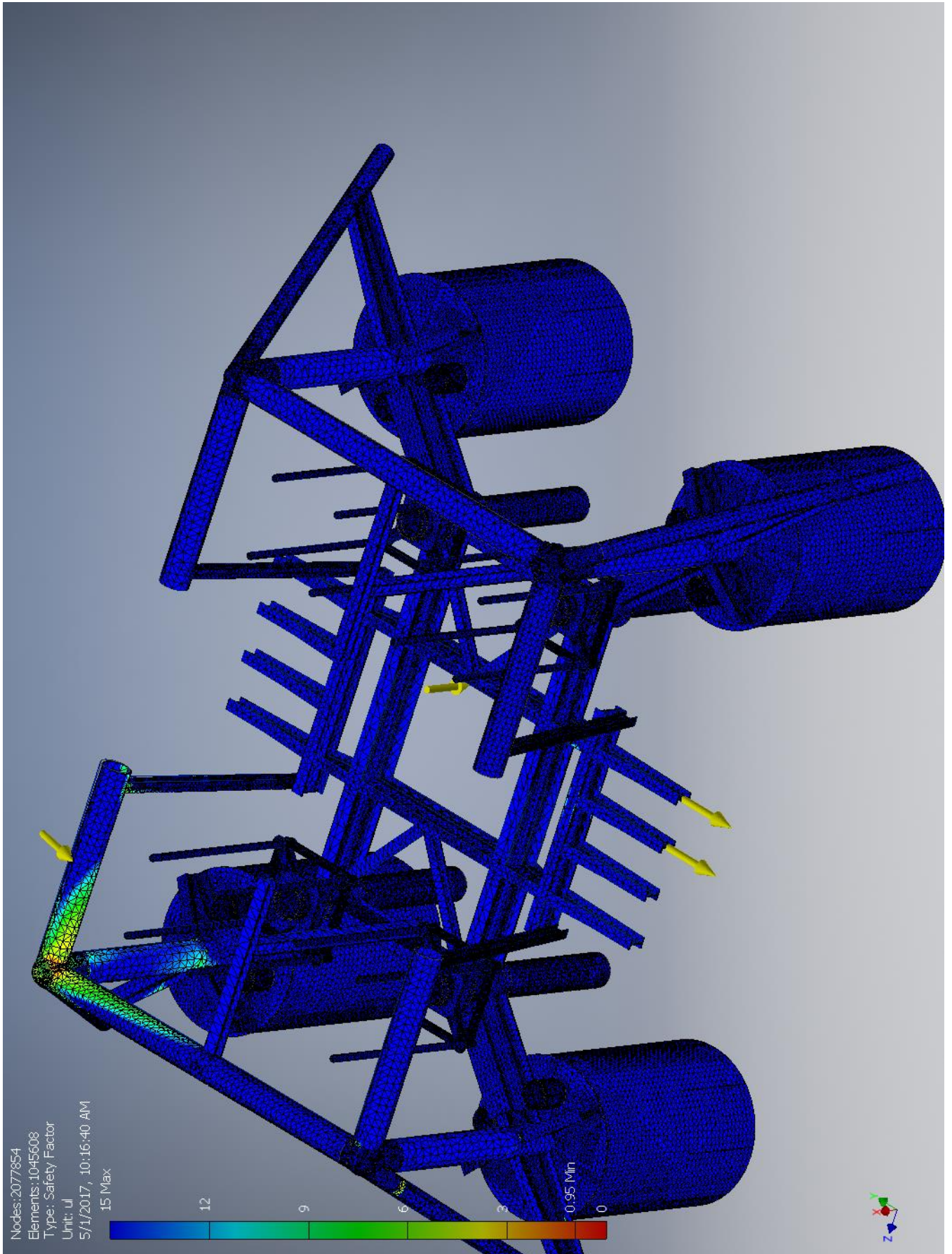
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.4 Case-D

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/28/2017, 5:10 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	181171 kg
Area	3.52908E+09 mm ²
Volume	6.02475E+10 mm ³
Center of Gravity	x=-8288.42 mm y=6548.24 mm z=9857.84 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-D

General objective and settings:

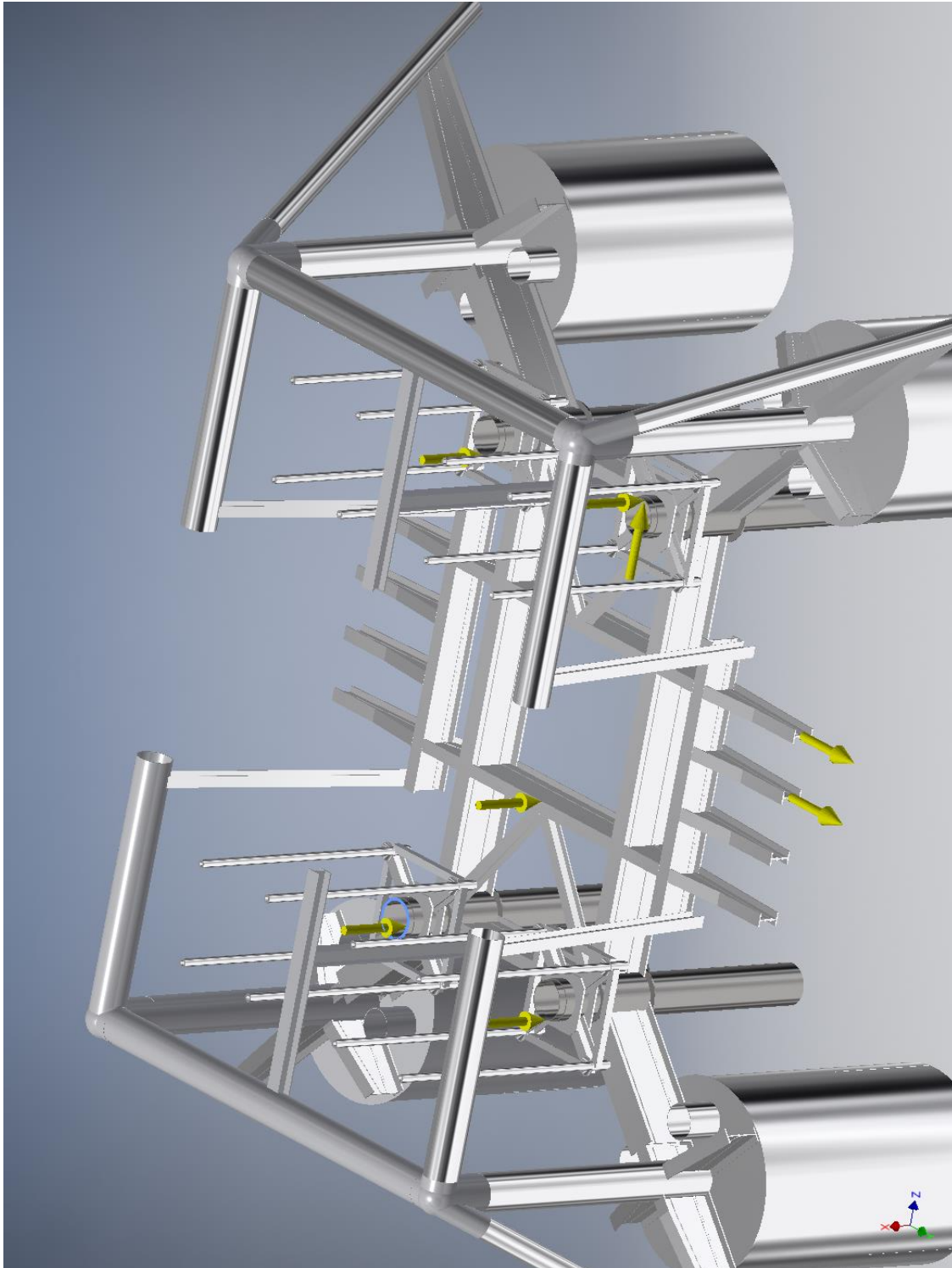
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/28/2017, 5:09 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

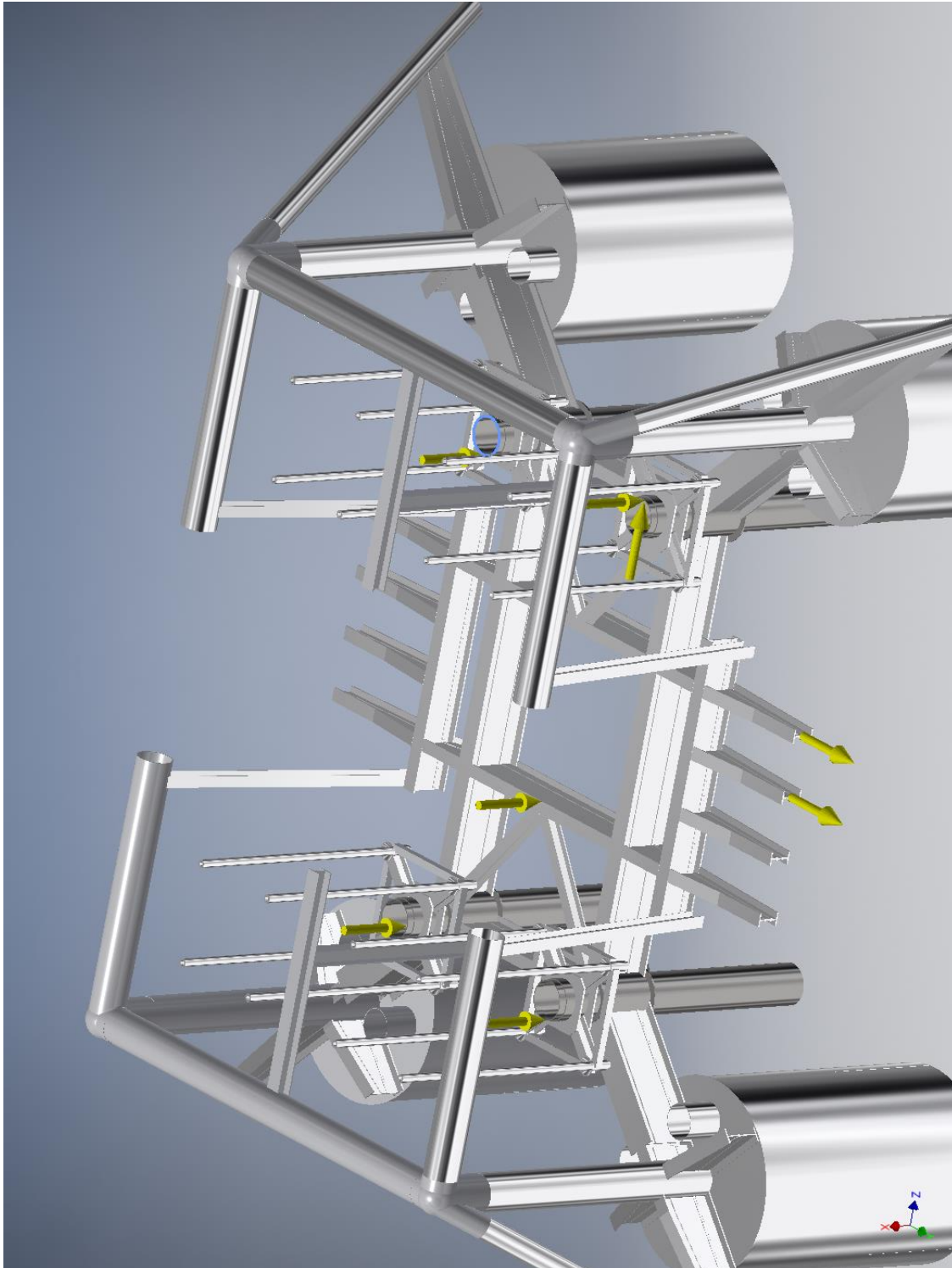
☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

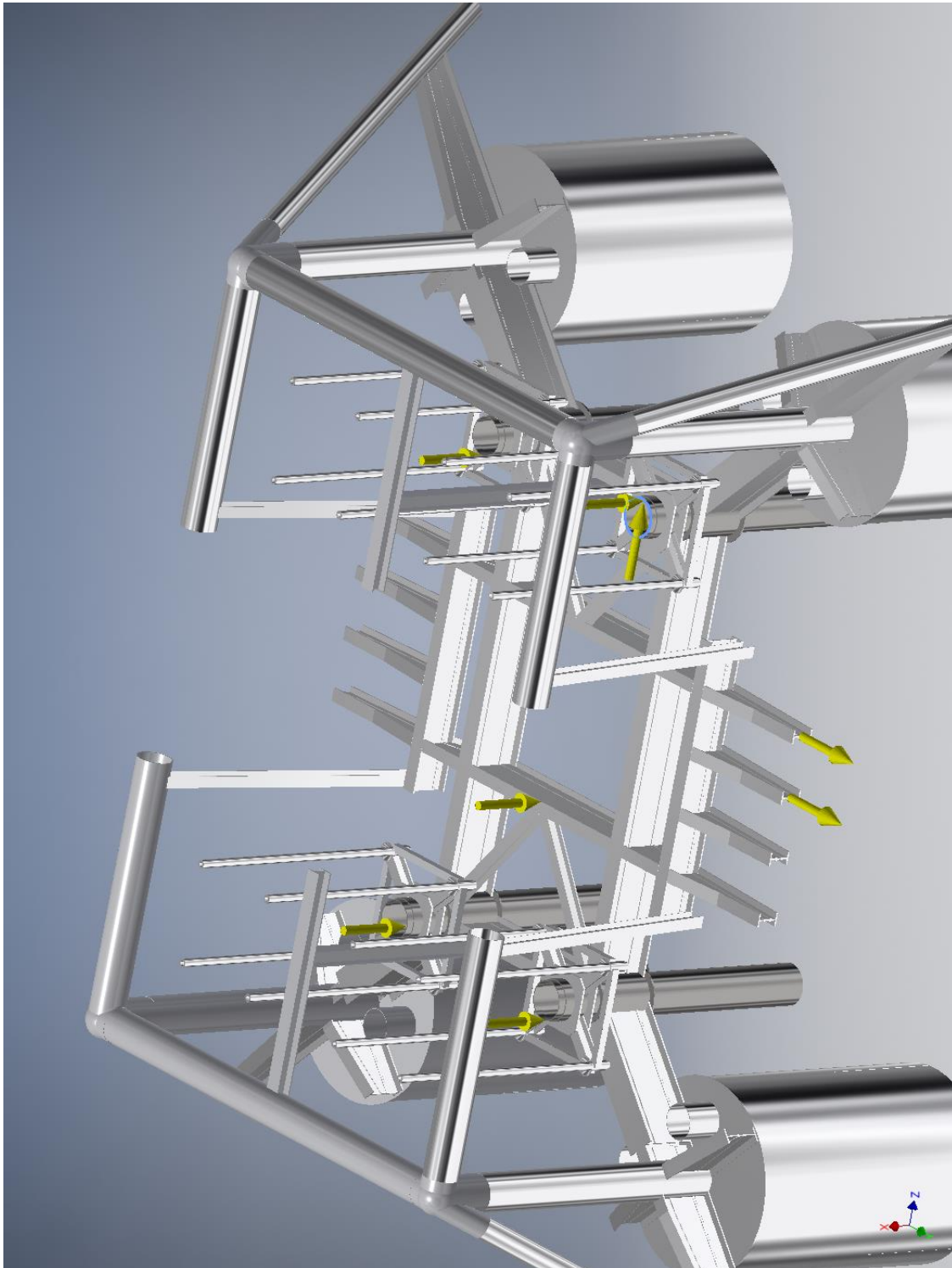
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

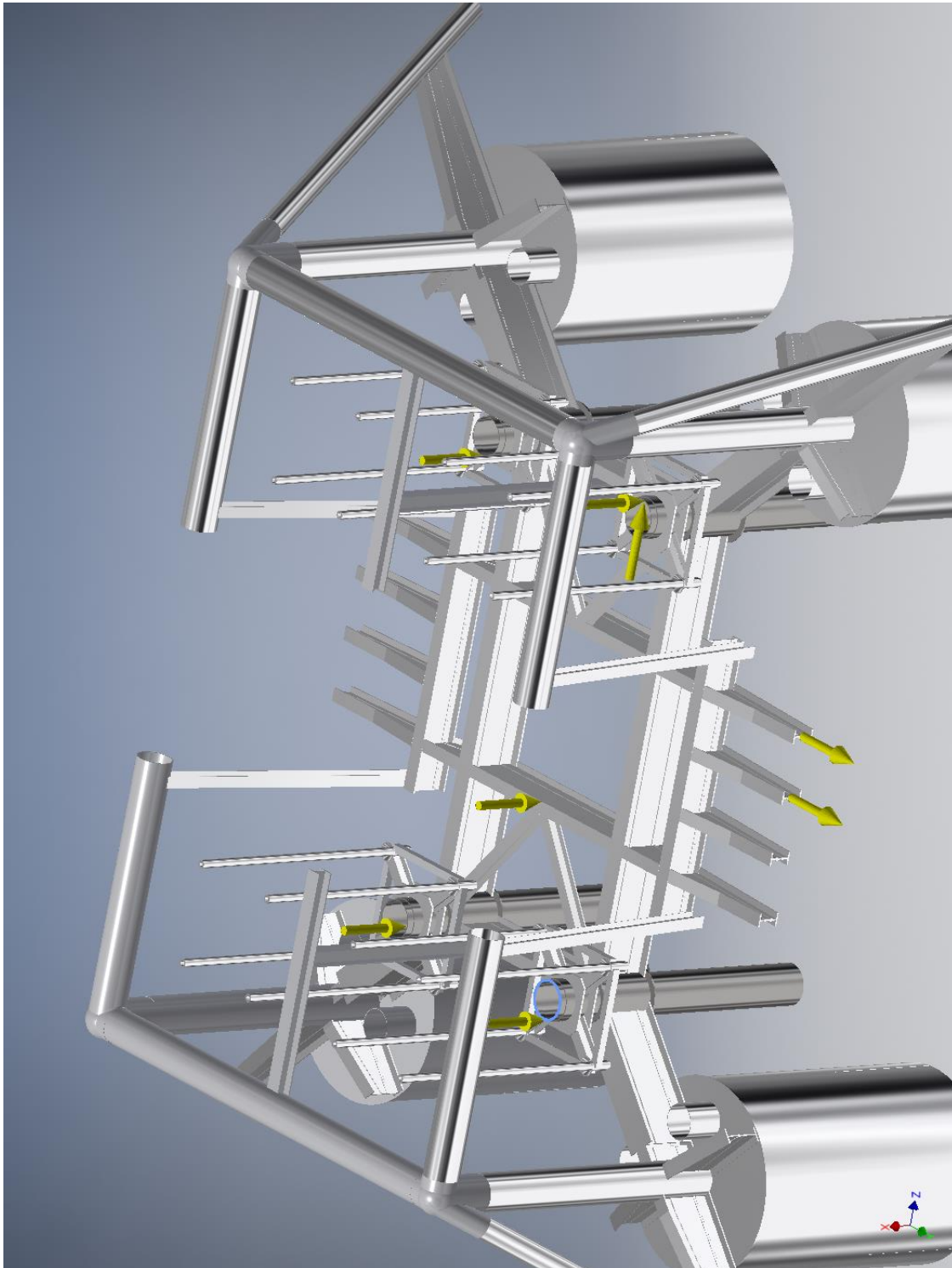
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

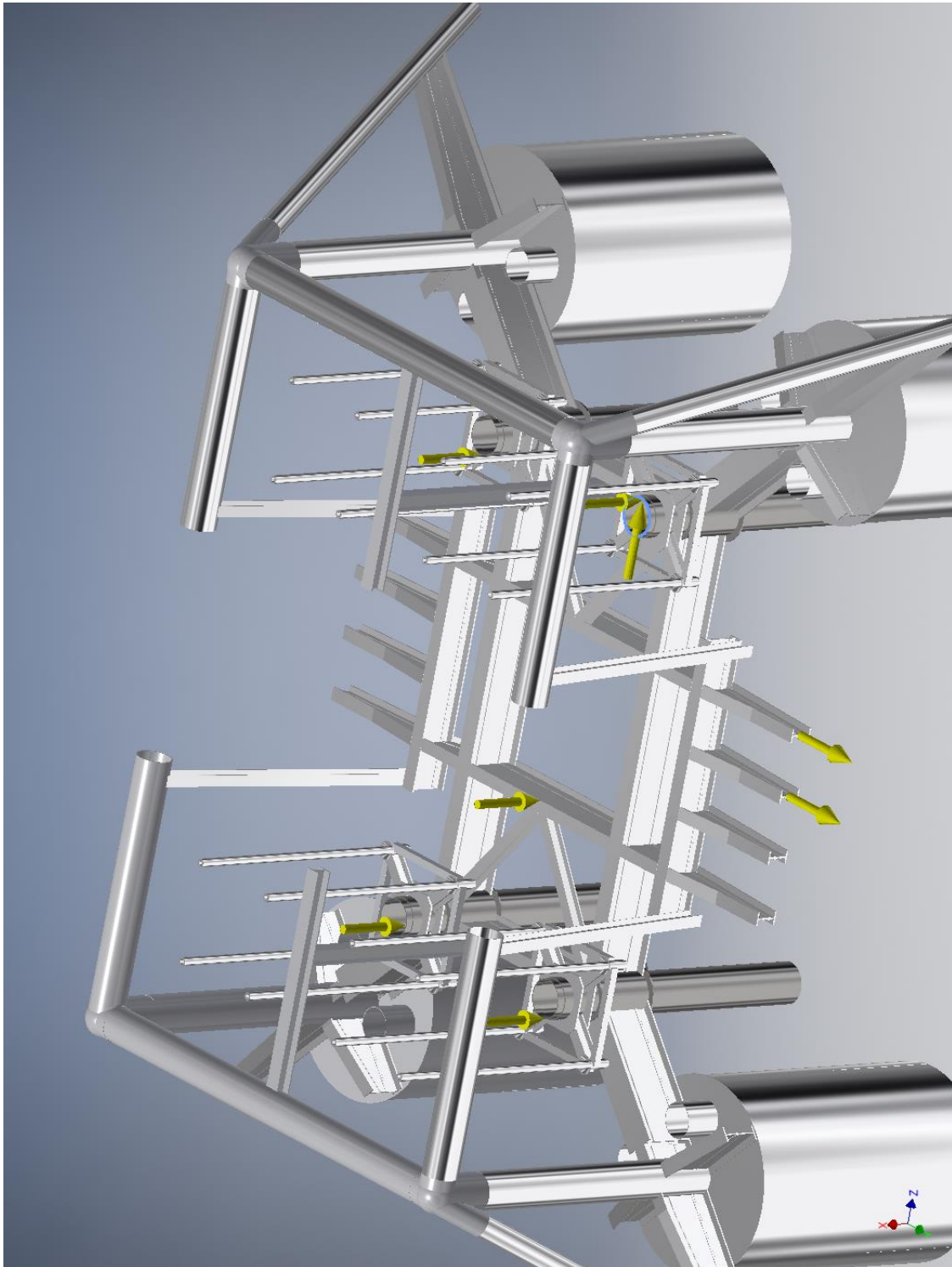
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	450000.000 N
Vector X	-450000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

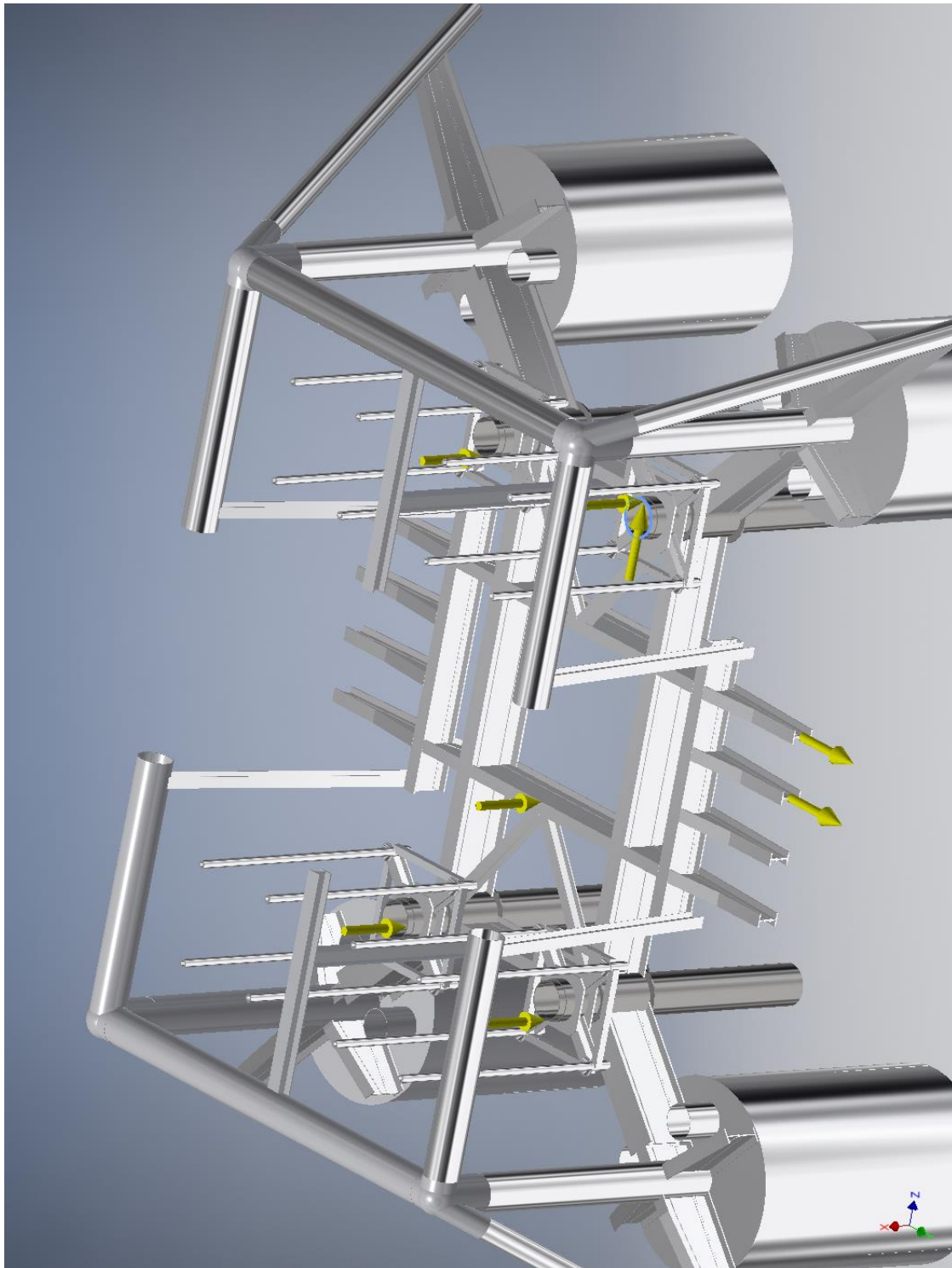
☐ **Selected Face(s)**



☐ **Force:6**

Load Type	Force
Magnitude	200000.000 N
Vector X	0.000 N
Vector Y	-0.000 N
Vector Z	200000.000 N

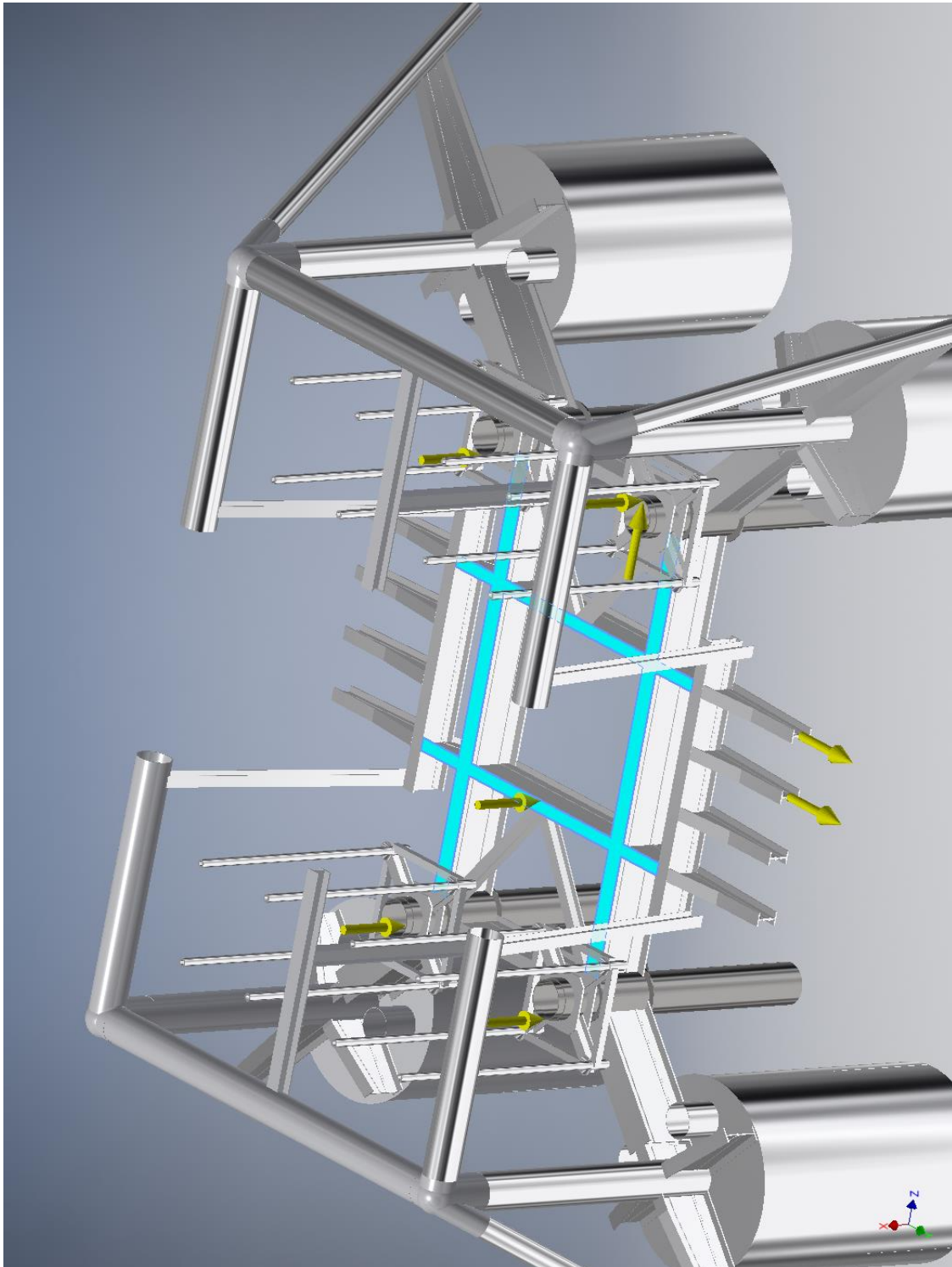
☐ **Selected Face(s)**



☐ **Force:7**

Load Type	Force
Magnitude	800000.000 N
Vector X	-800000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

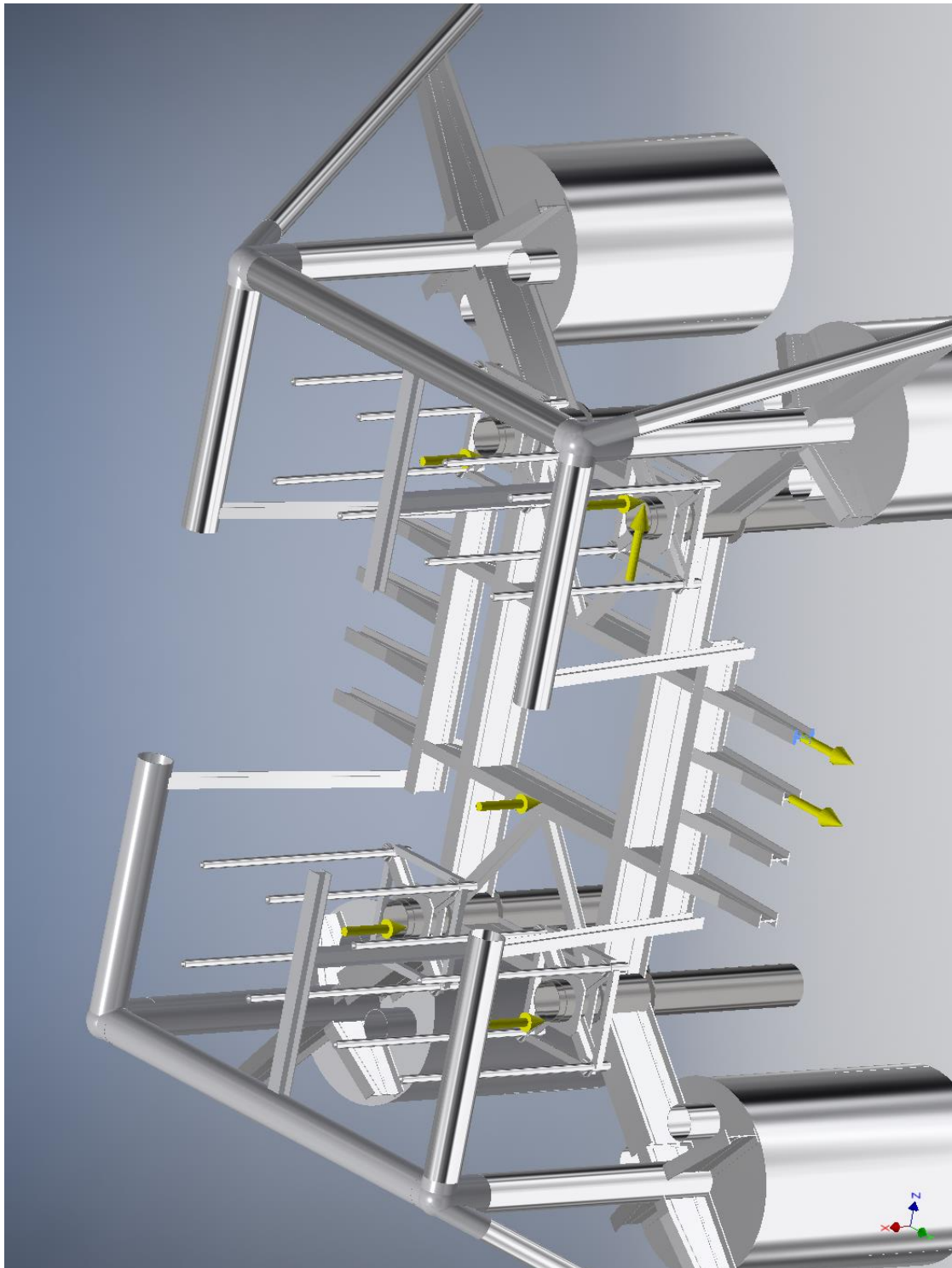
☐ **Selected Face(s)**



☐ **Force:8**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	300000.000 N
Vector Z	-0.000 N

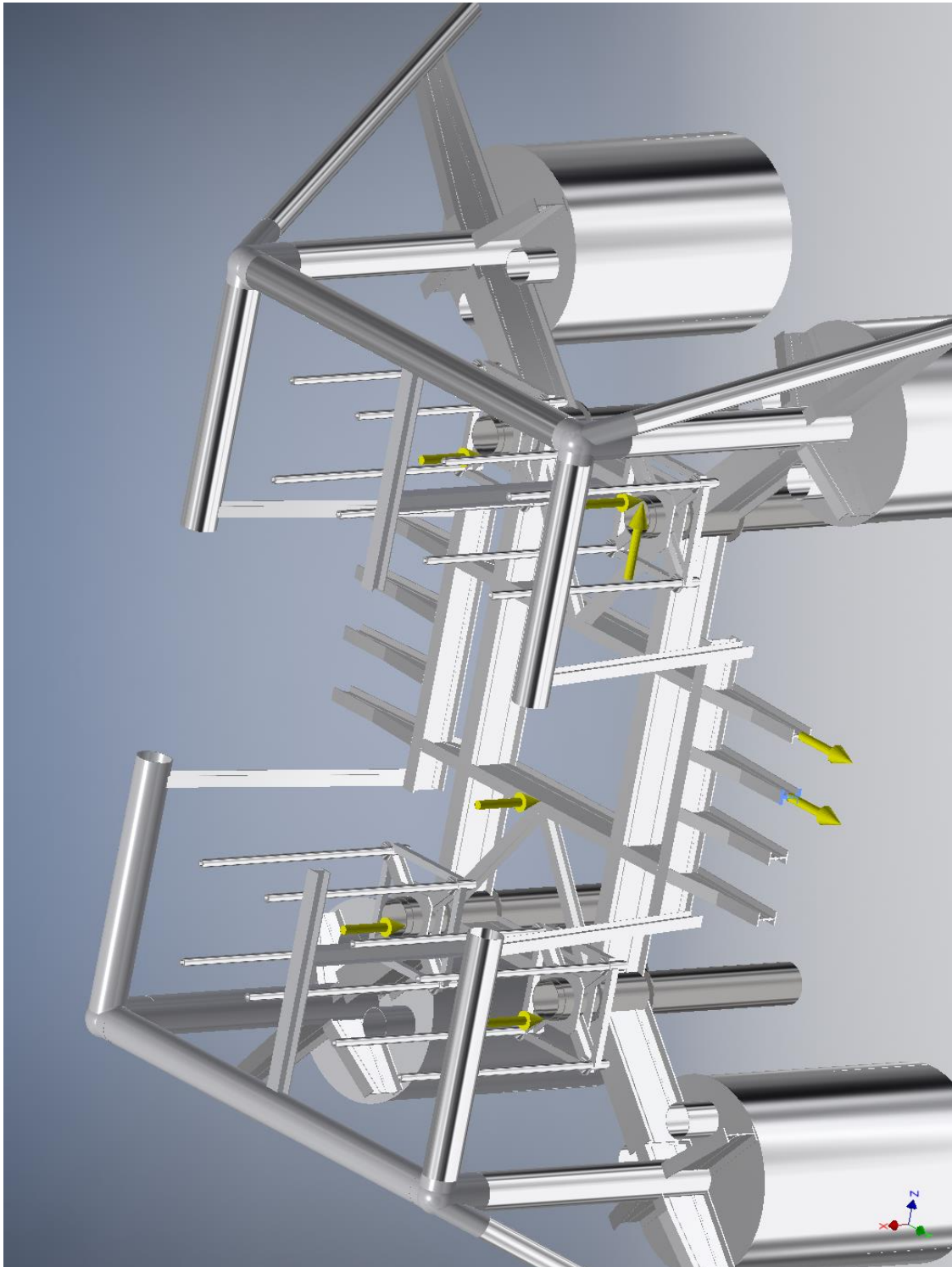
☐ **Selected Face(s)**



☐ **Force:9**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	300000.000 N
Vector Z	-0.000 N

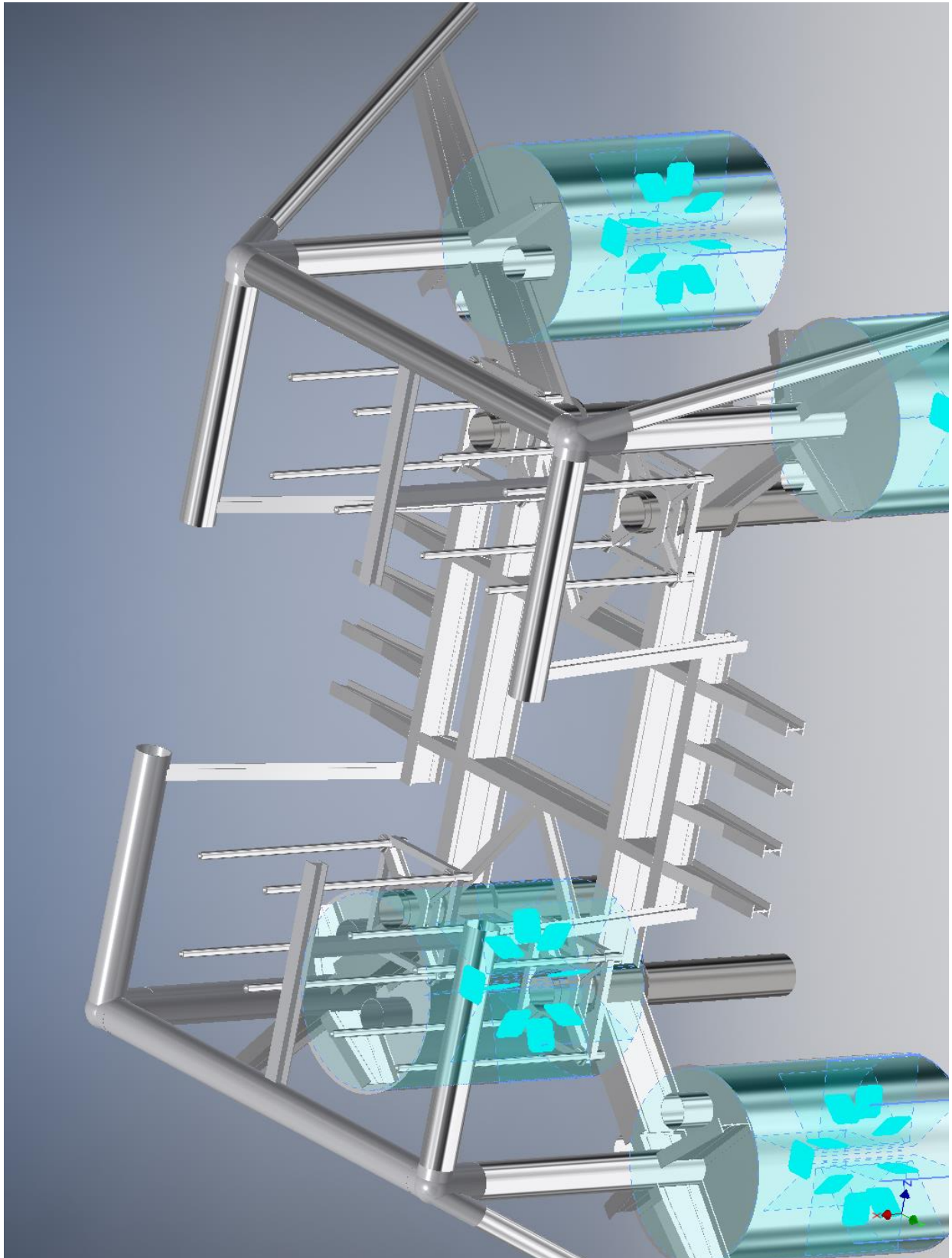
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
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☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

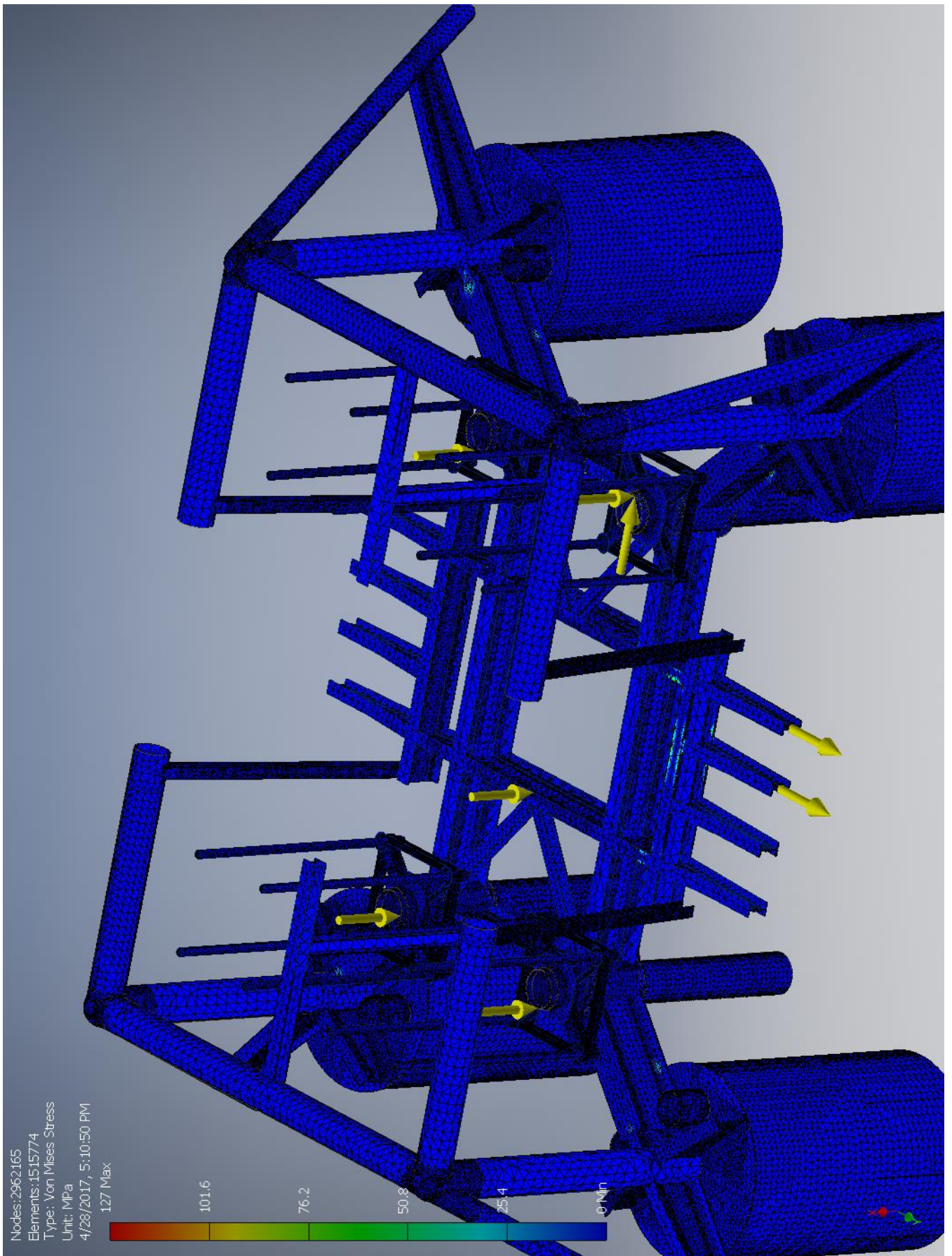
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	3704390 N	3650000 N	6204470 N m	365904 N m
		-600000 N		4253910 N m
		-200000 N		-4501760 N m

☐ Result Summary

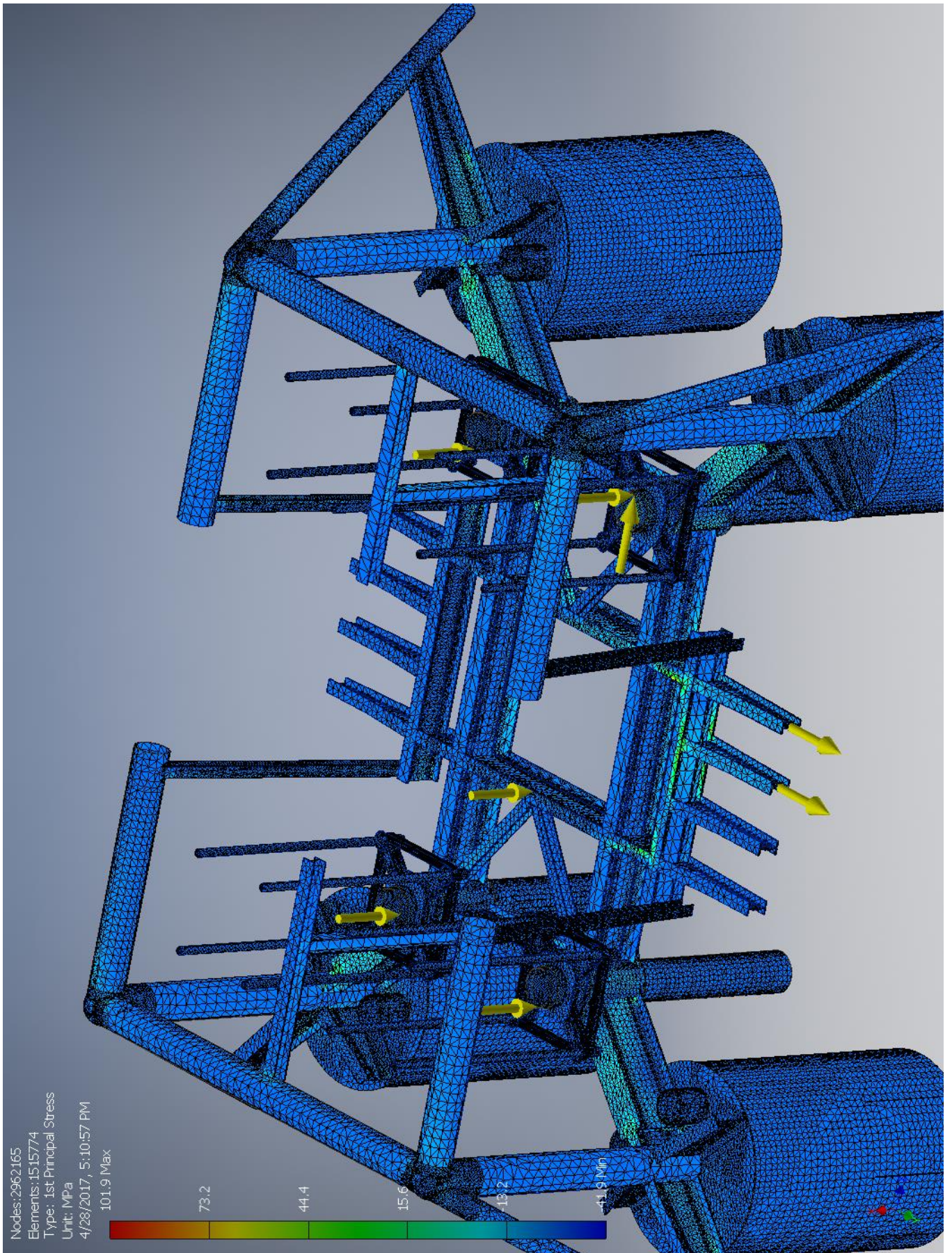
Name	Minimum	Maximum
Volume	6.02475E+10 mm ³	
Mass	181171 kg	
Von Mises Stress	0 MPa	127.049 MPa
1st Principal Stress	-41.9479 MPa	101.938 MPa
3rd Principal Stress	-146.946 MPa	39.752 MPa
Displacement	0 mm	17.6941 mm
Safety Factor	2.71356 ul	15 ul

☐ Figures

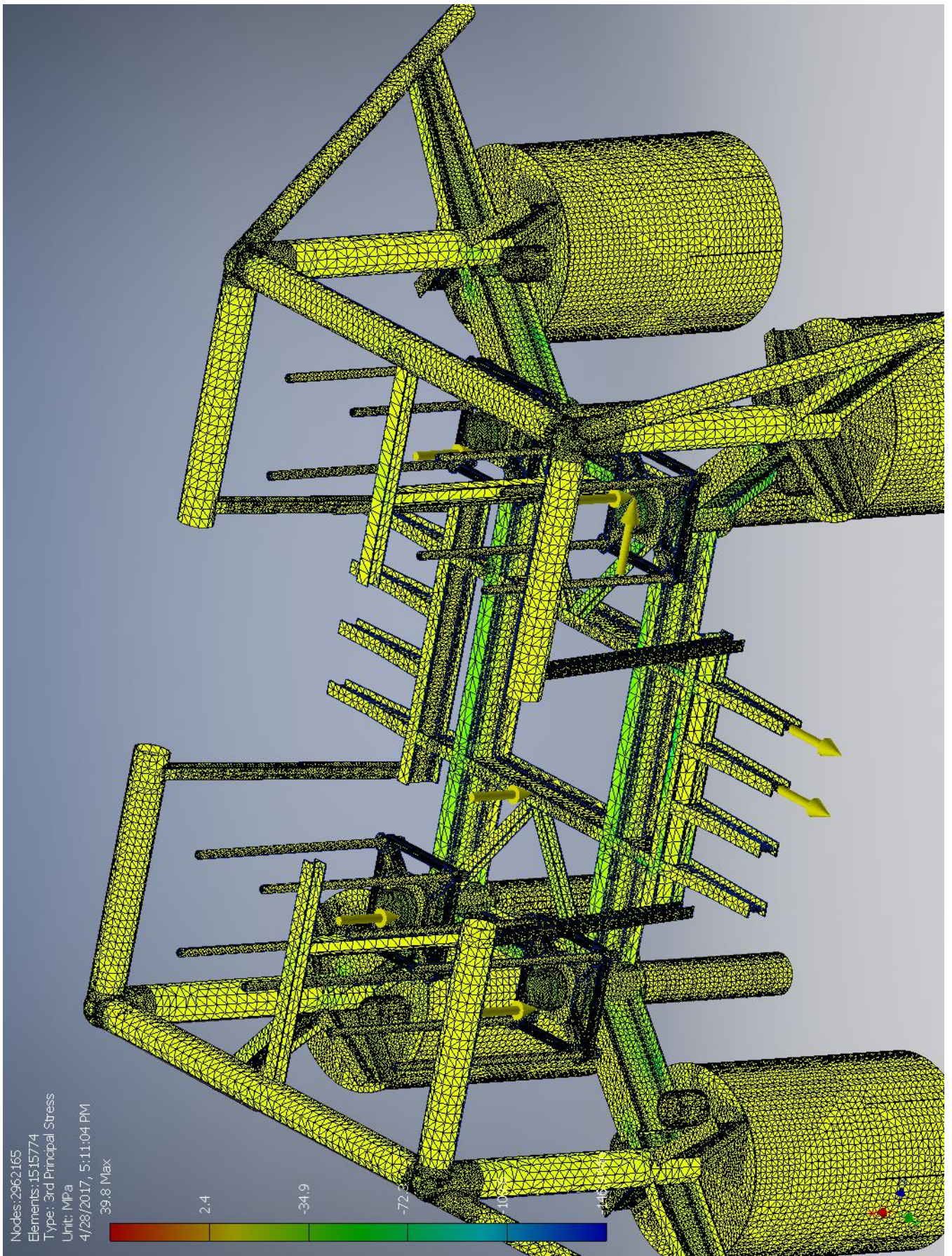
☐ Von Mises Stress



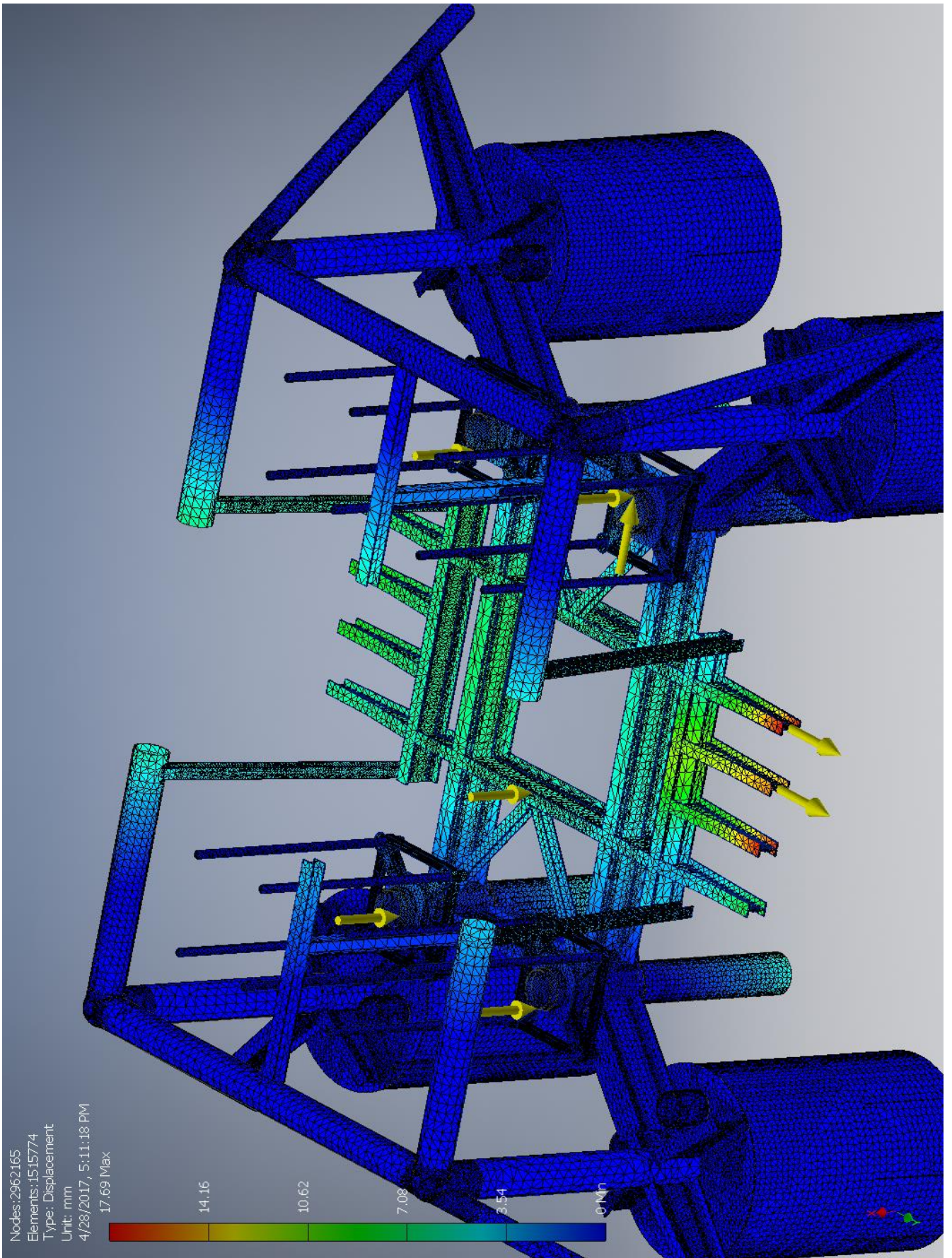
☐ 1st Principal Stress



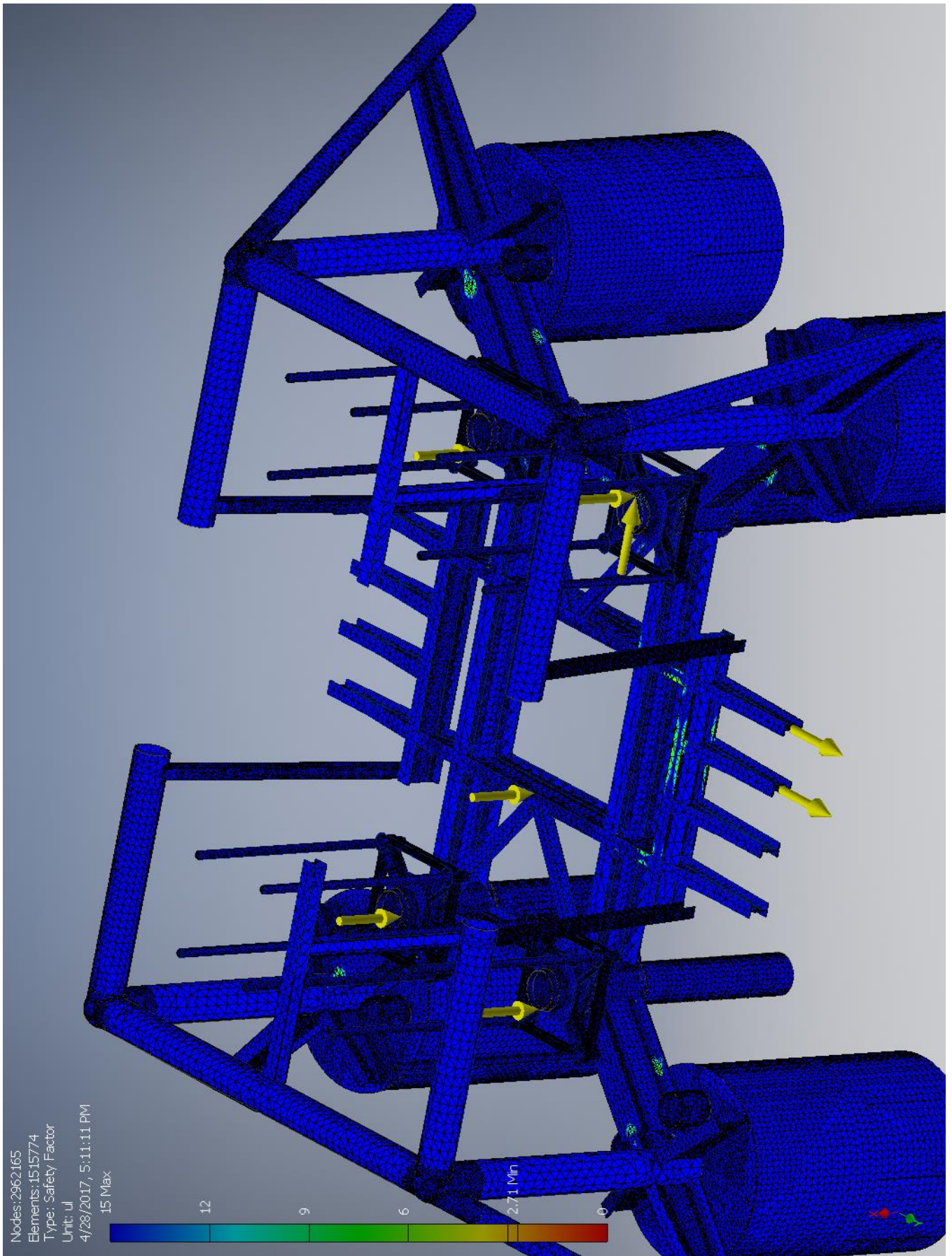
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.5 Case-E

I.5.1 Without convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/27/2017, 5:19 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	181171 kg
Area	3.52908E+09 mm ²
Volume	6.02475E+10 mm ³
Center of Gravity	x=-8288.42 mm y=6548.24 mm z=9857.84 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case_E

General objective and settings:

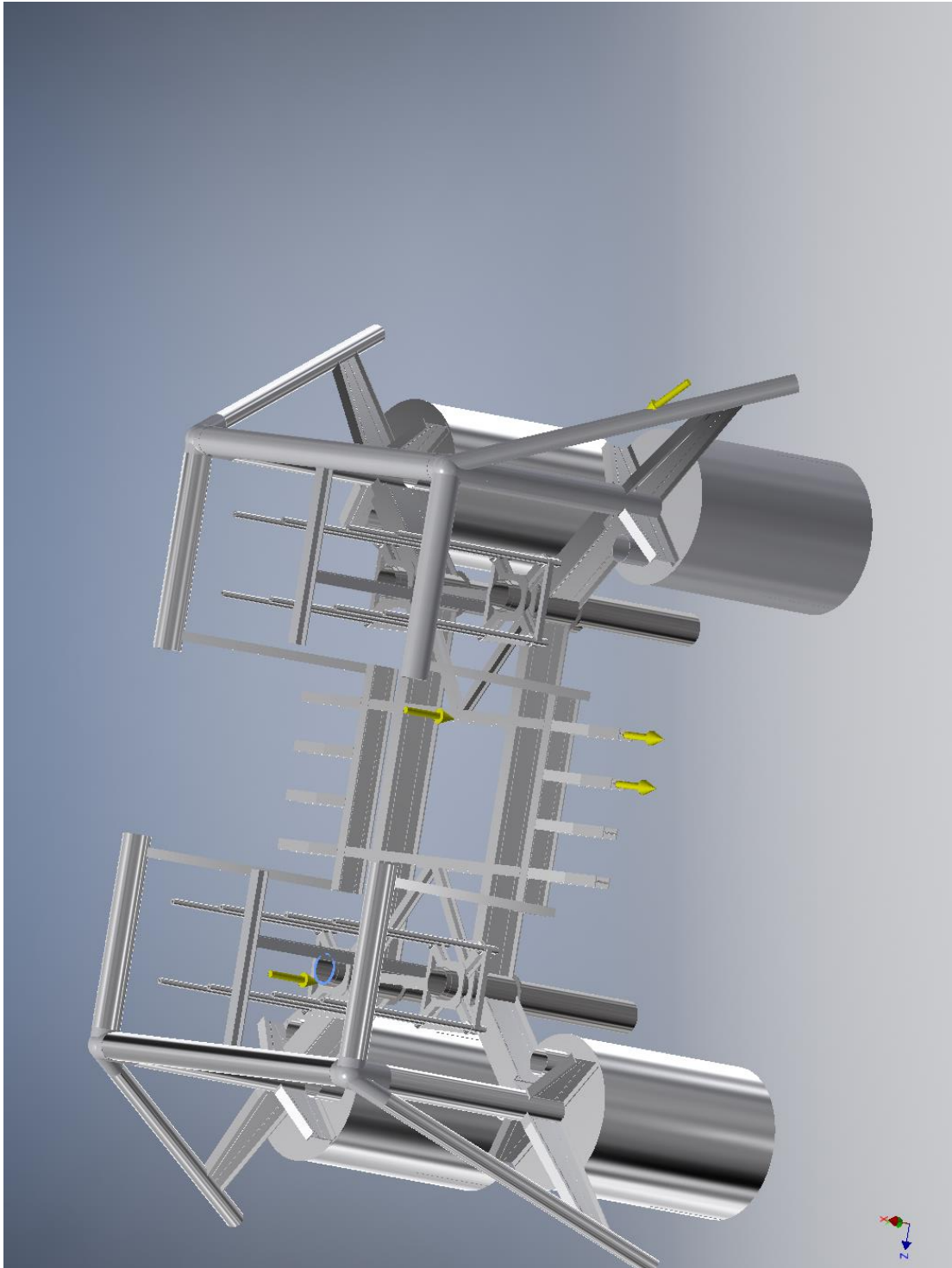
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/27/2017, 5:10 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

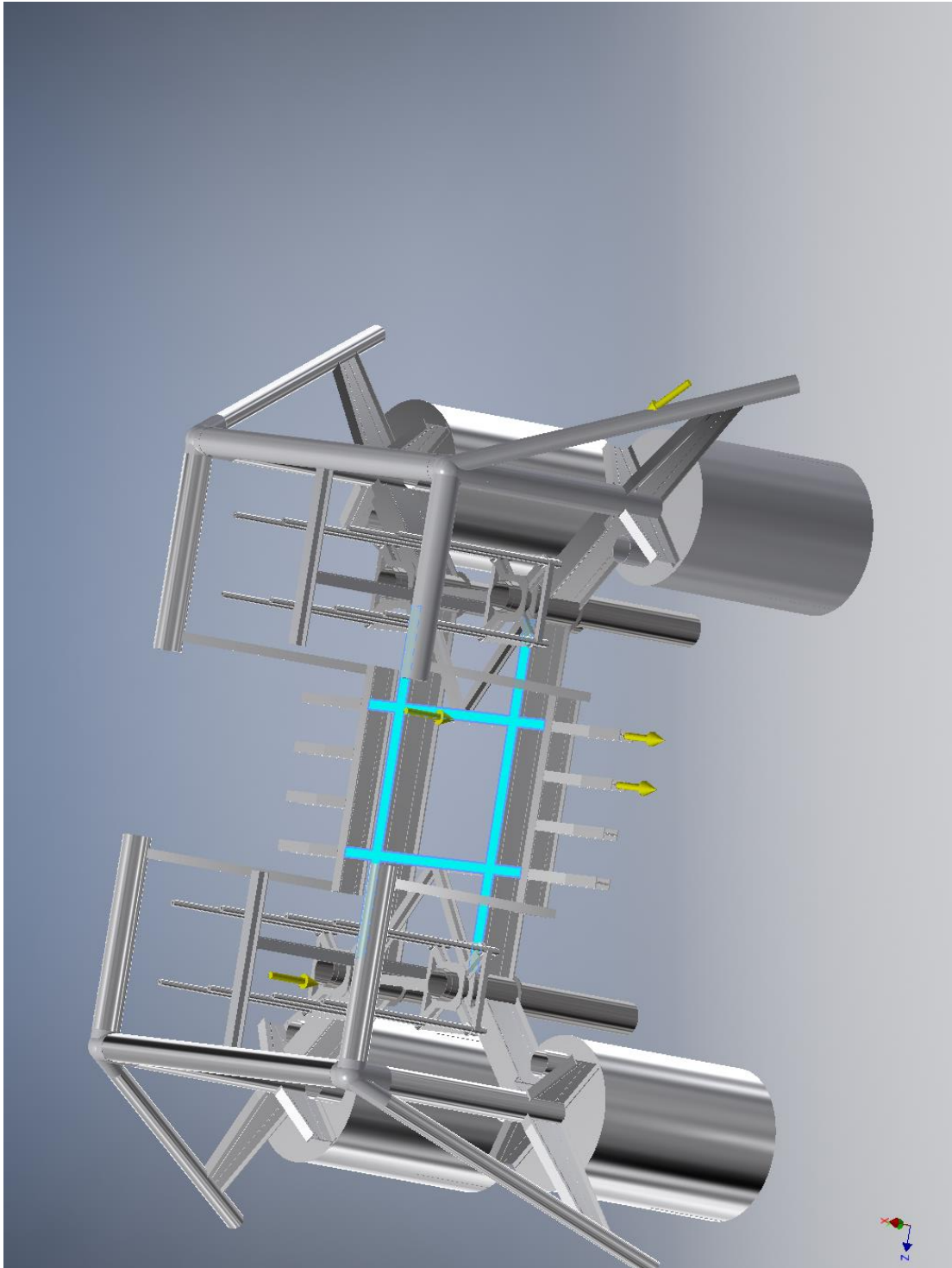
☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	800000.000 N
Vector X	-800000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

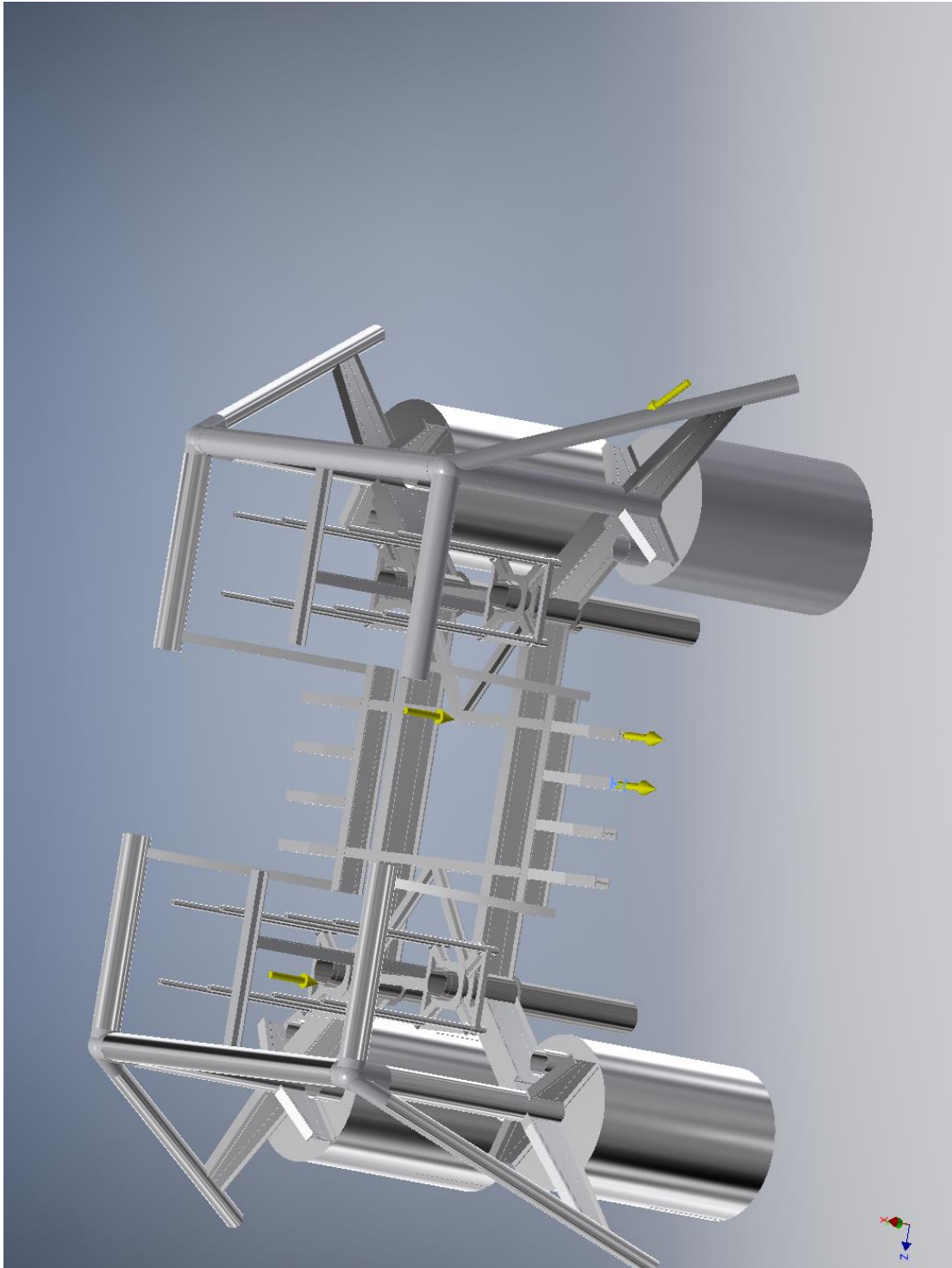
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	-300000.000 N
Vector Z	0.000 N

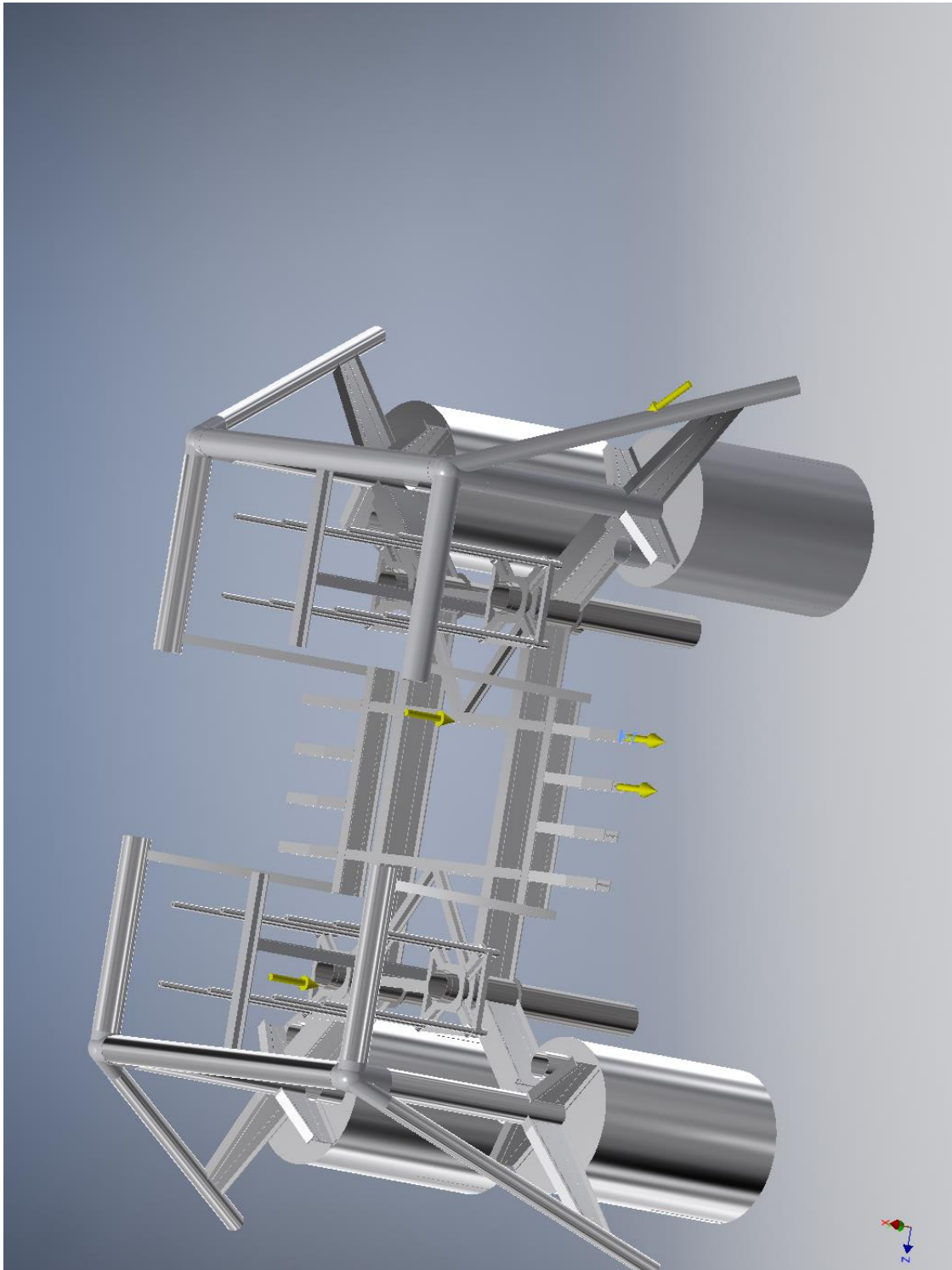
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	300000.000 N
Vector X	-0.000 N
Vector Y	-300000.000 N
Vector Z	0.000 N

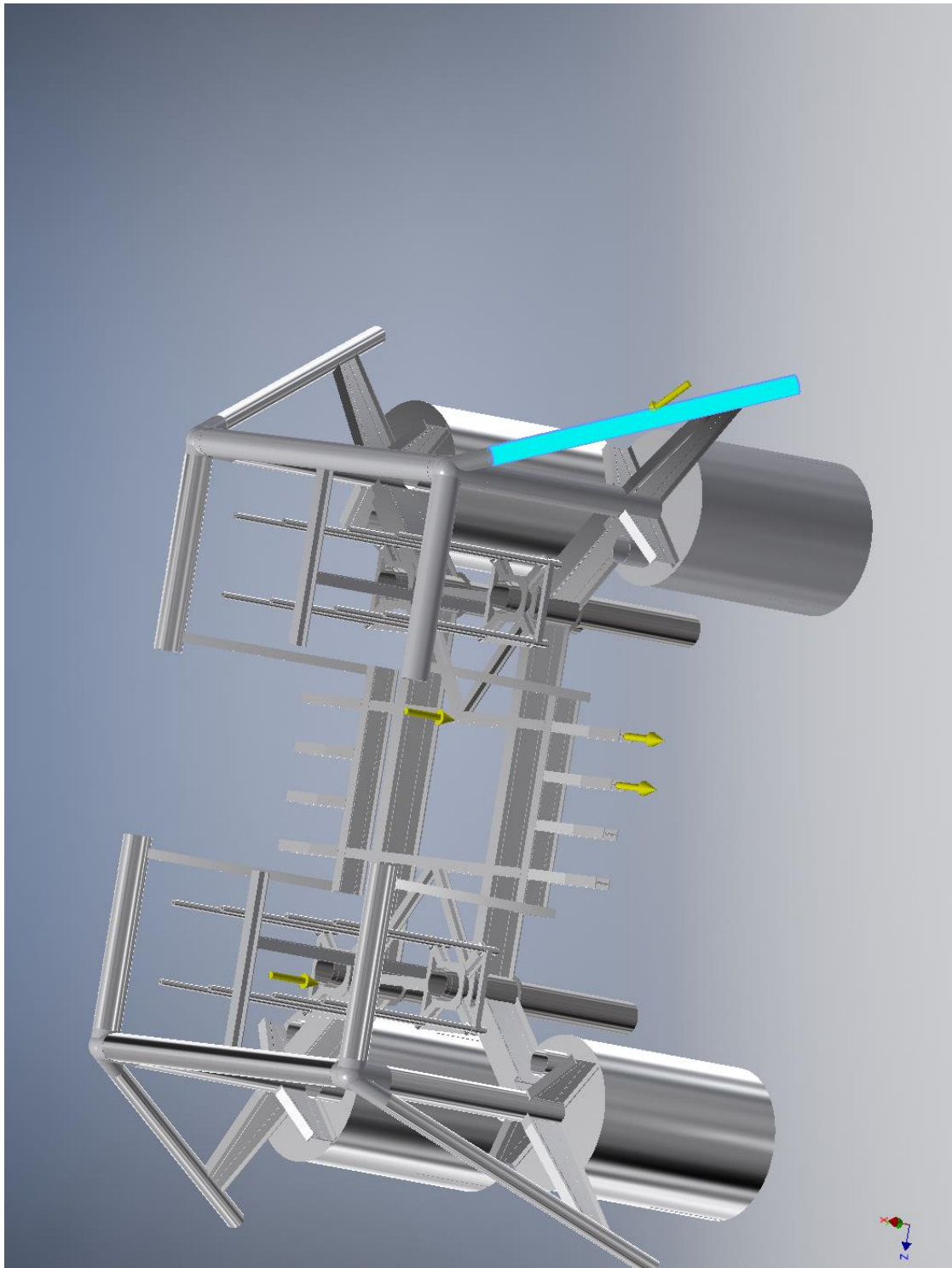
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	1047367.023 N
Vector X	342020.000 N
Vector Y	700000.000 N
Vector Z	700000.000 N

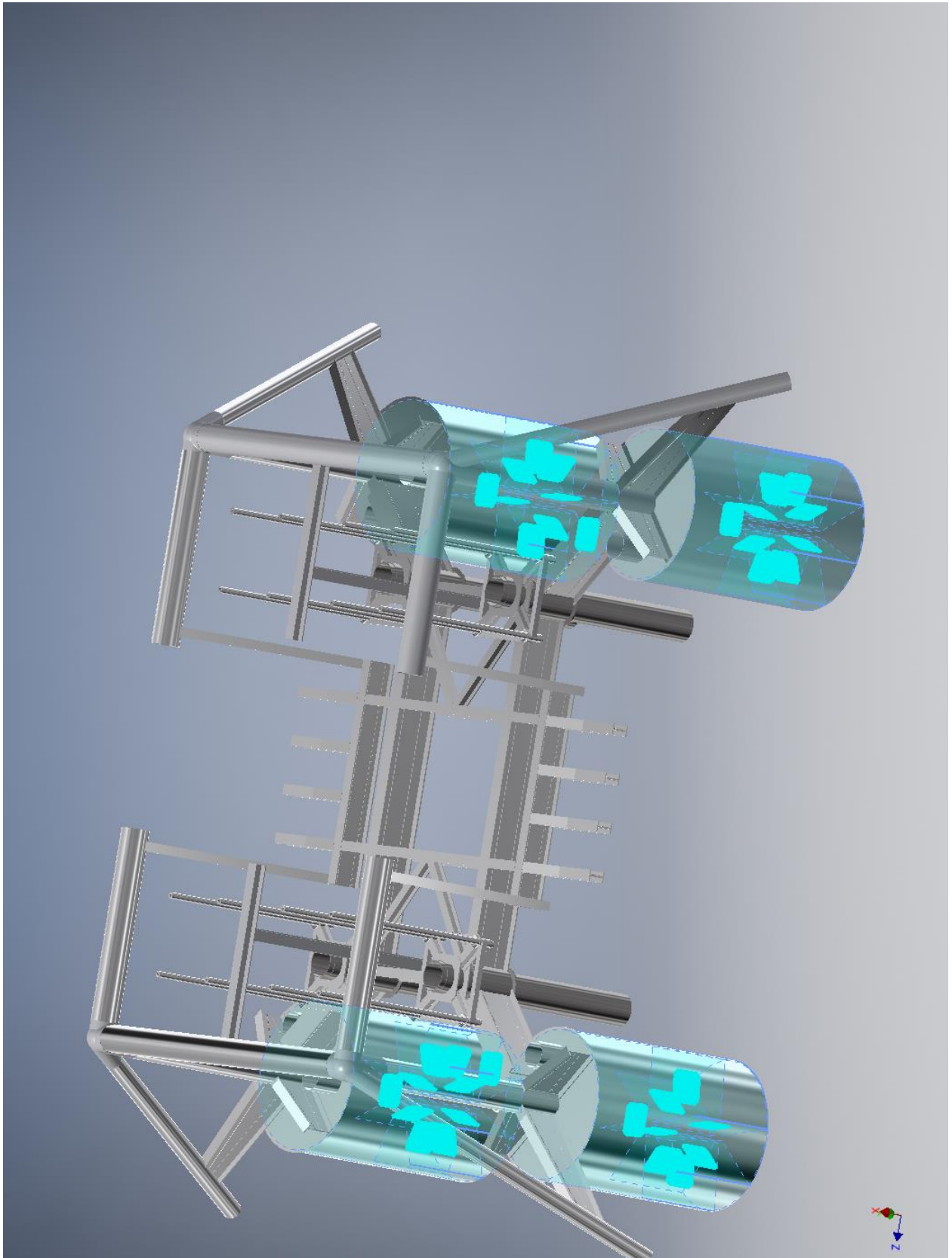
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
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☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

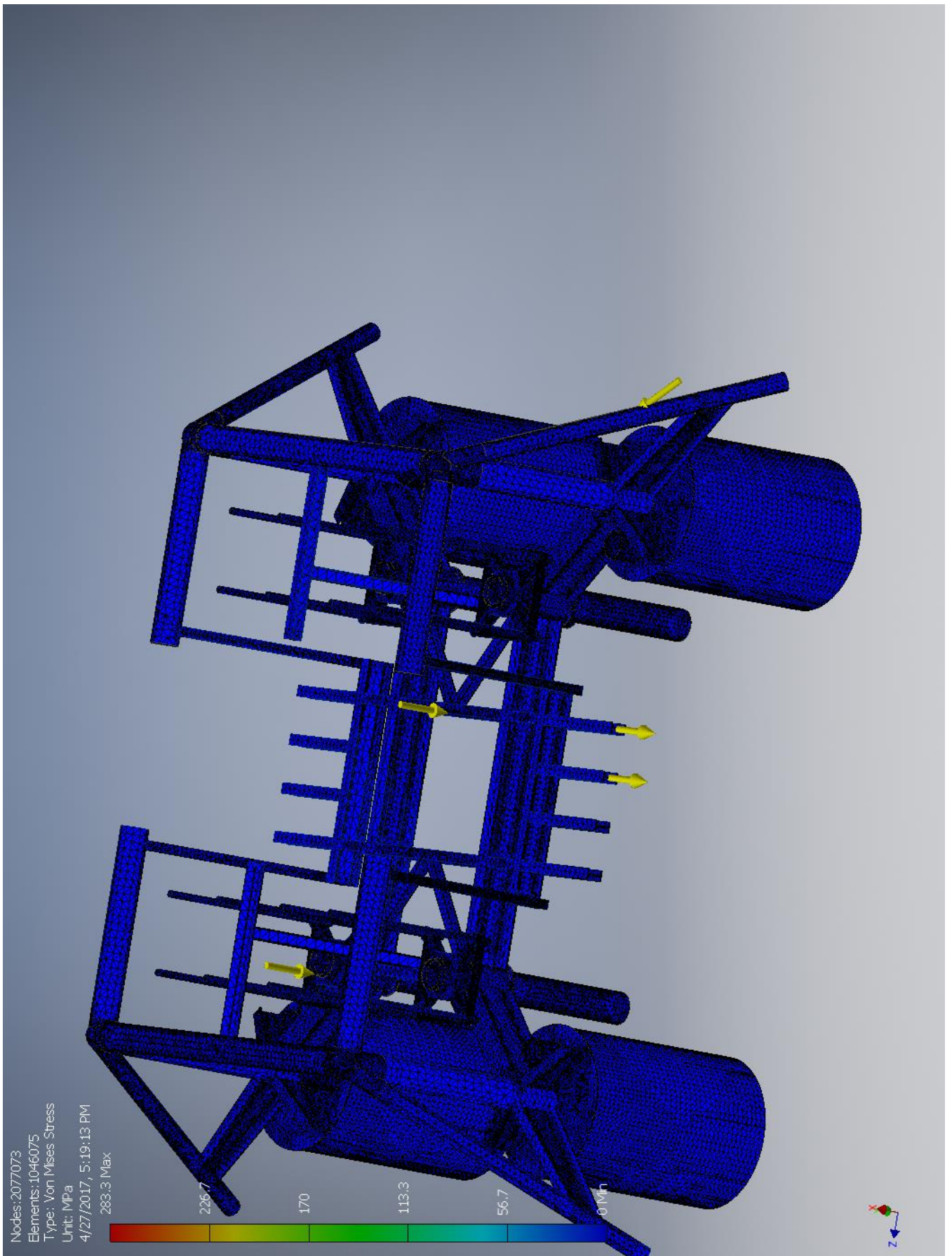
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1272530 N	1057980 N	14744600 N m	-1340550 N m
		-100000 N		13024400 N m
		-700000 N		-6780180 N m

☐ Result Summary

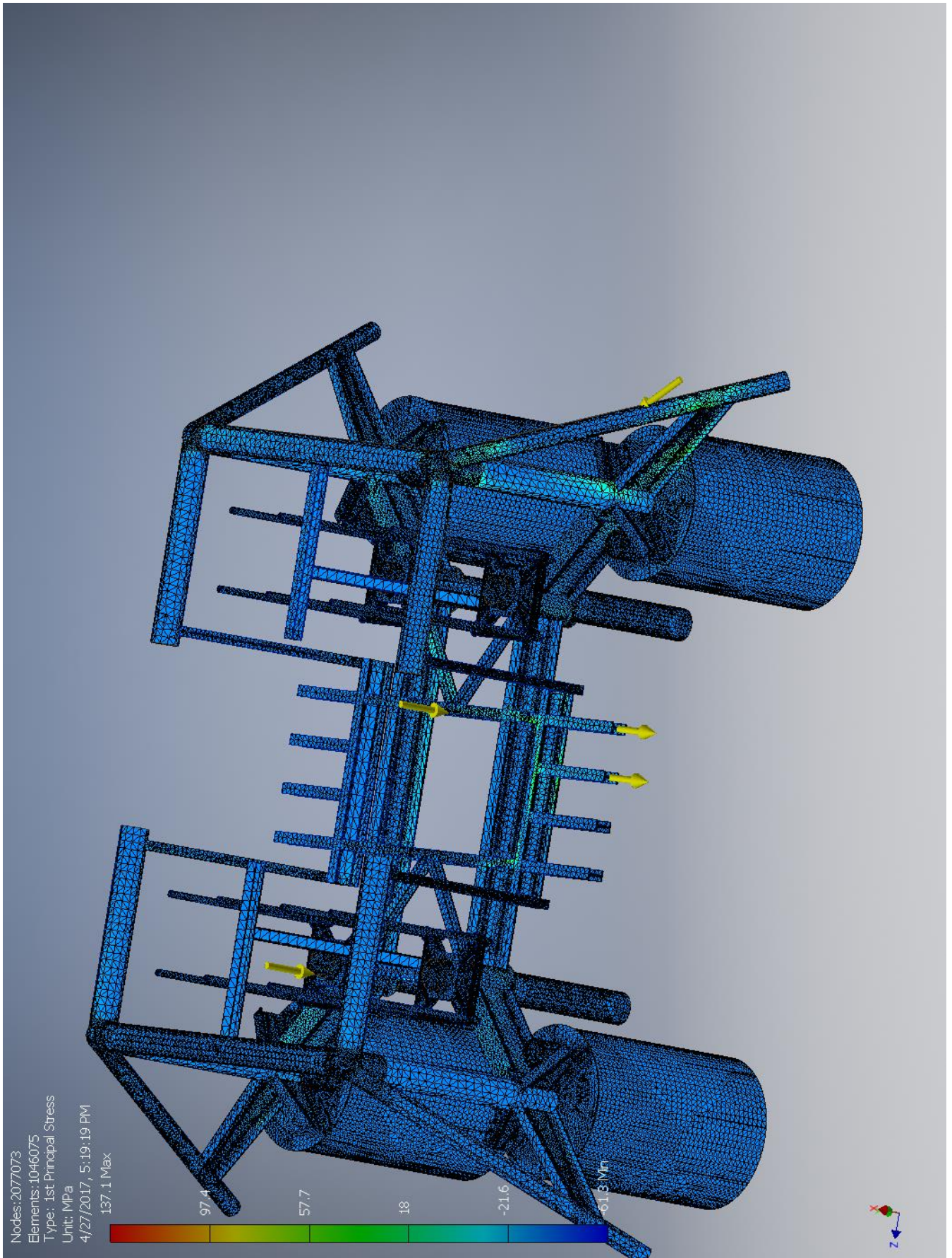
Name	Minimum	Maximum
Volume	6.02475E+10 mm ³	
Mass	181171 kg	
Von Mises Stress	0 MPa	283.315 MPa
1st Principal Stress	-61.317 MPa	137.09 MPa
3rd Principal Stress	-227.754 MPa	39.9197 MPa
Displacement	0 mm	19.5821 mm
Safety Factor	0.97065 ul	15 ul

☐ Figures

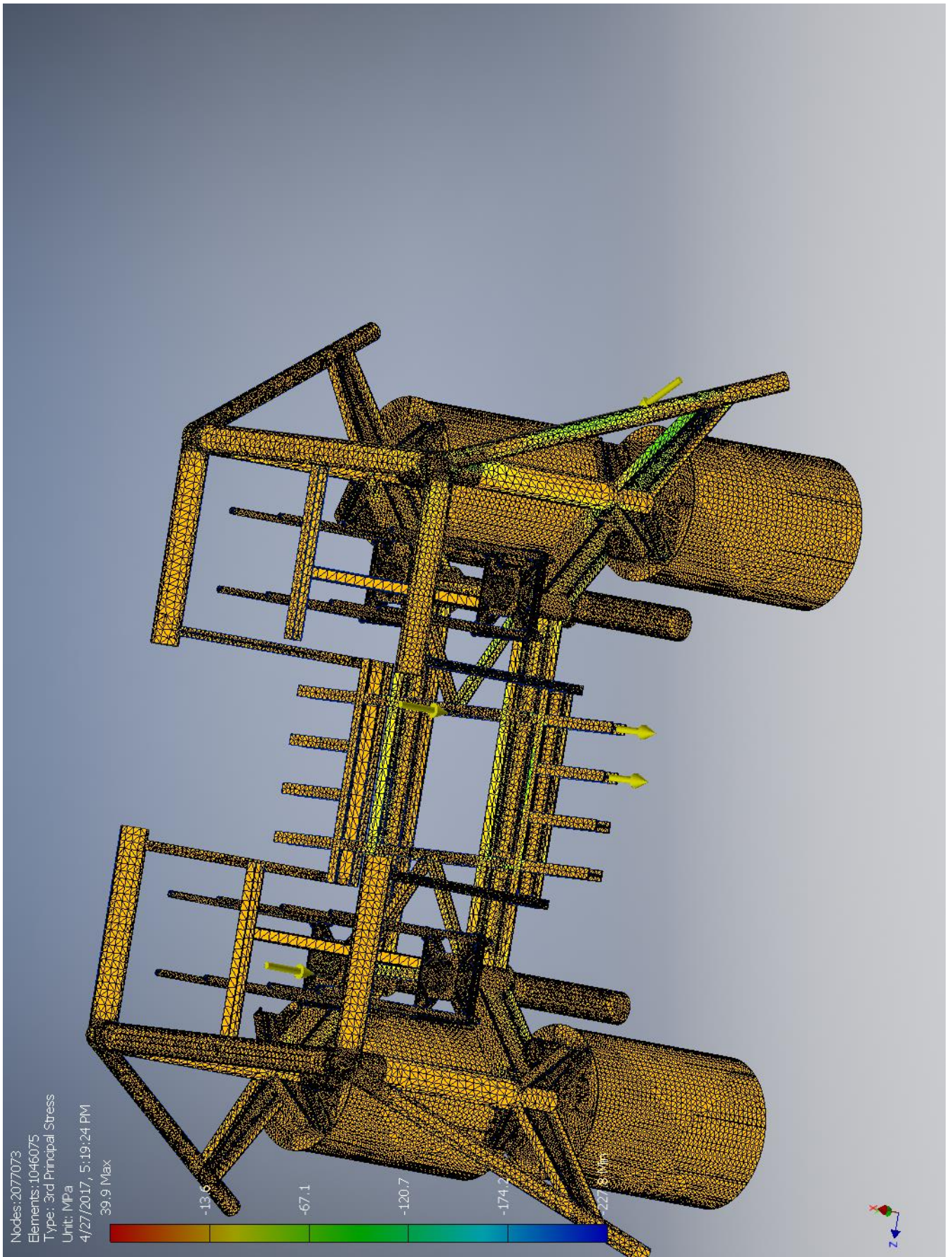
☐ Von Mises Stress



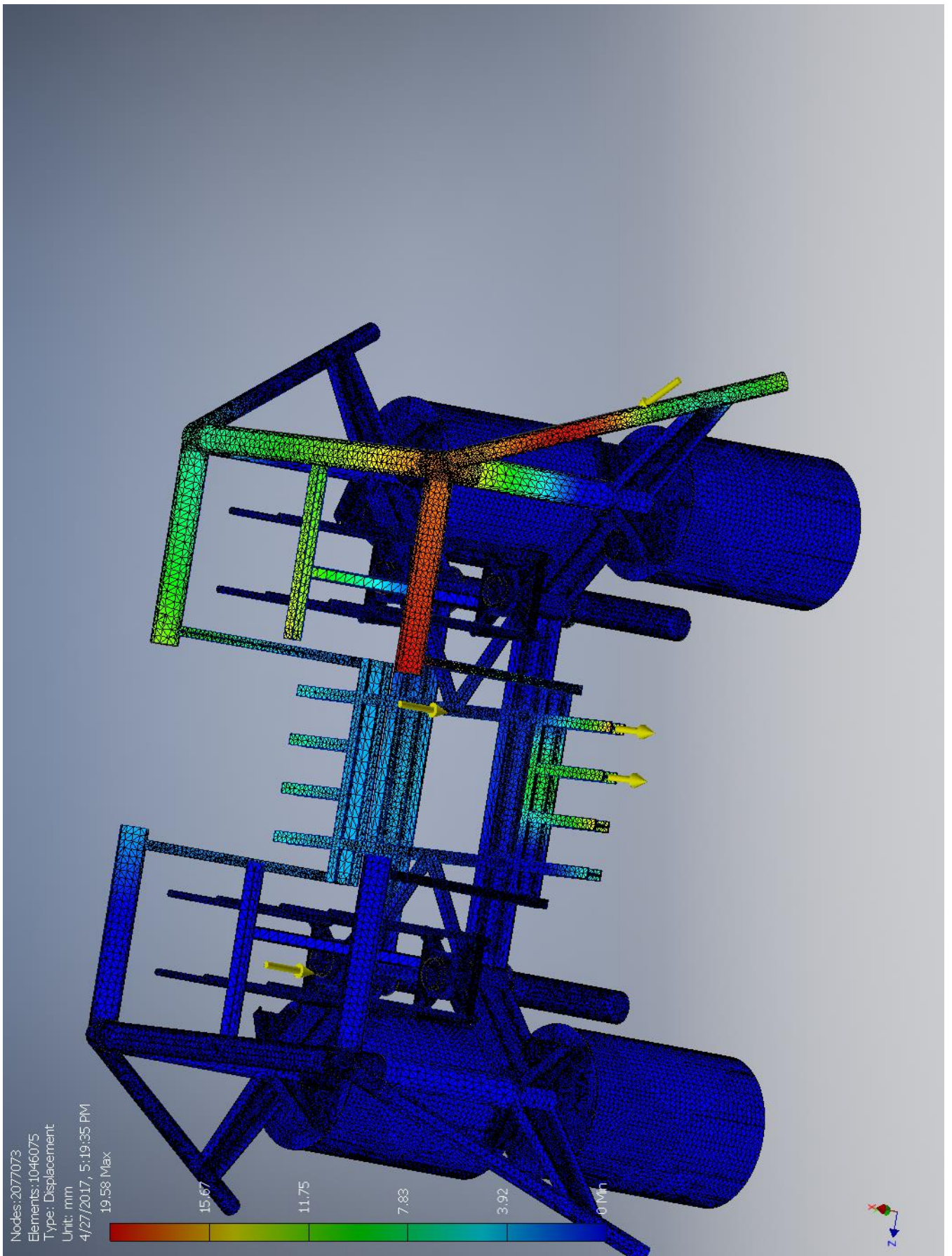
1st Principal Stress



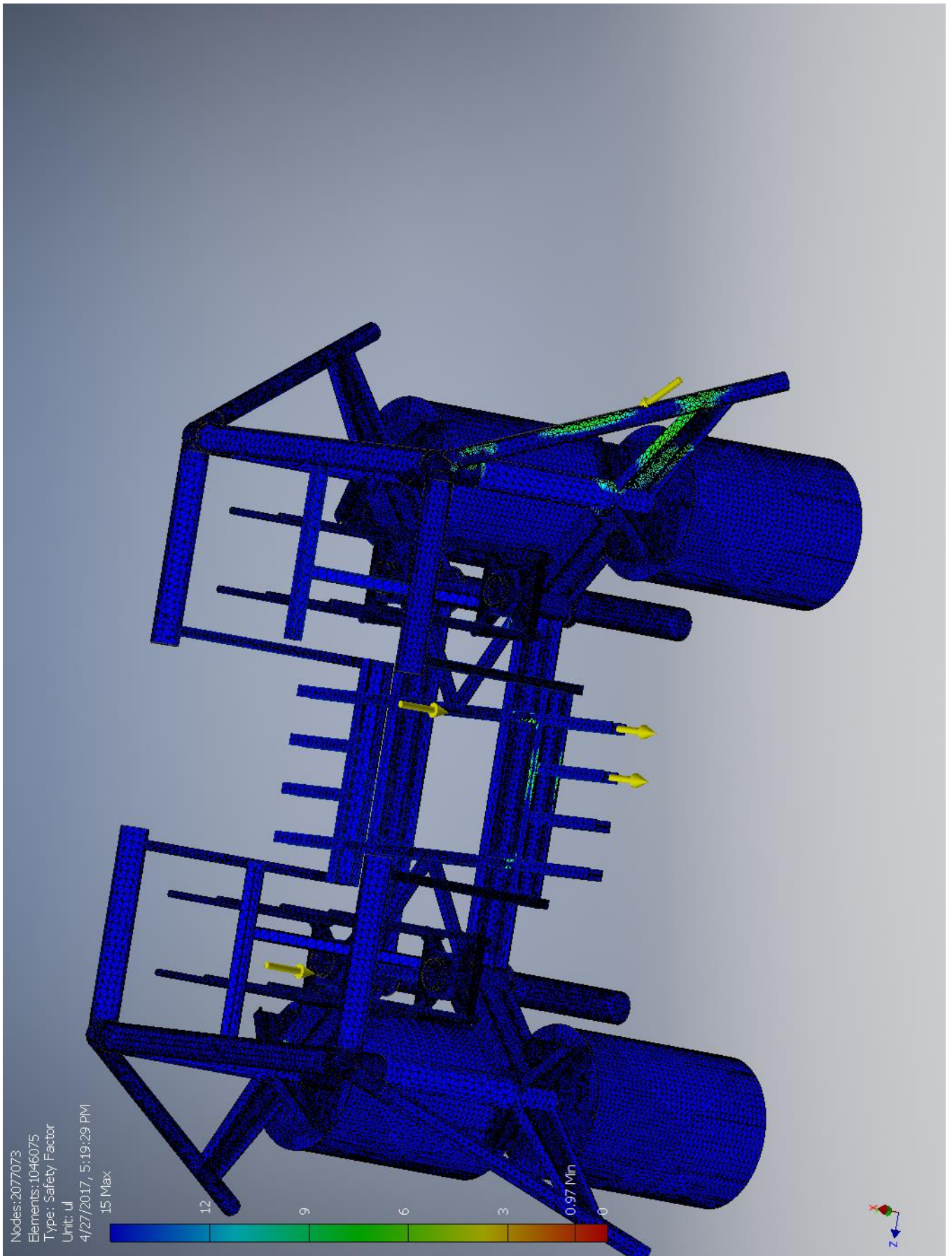
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.5.2 With convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/28/2017, 9:16 AM
Study Author:	henrikwn
Summary:	

Project Info (iProperties)

Summary

Title	
Author	henrikwn

Project

Designer	henrikwn
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Physical

Mass	181171 kg
Area	3.52908E+09 mm ²
Volume	6.02475E+10 mm ³
Center of Gravity	x=-8288.42 mm y=6548.24 mm z=9857.84 mm

Note: Physical values could be different from Physical values used by FEA reported below.

Case_E

General objective and settings:

Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/27/2017, 6:26 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

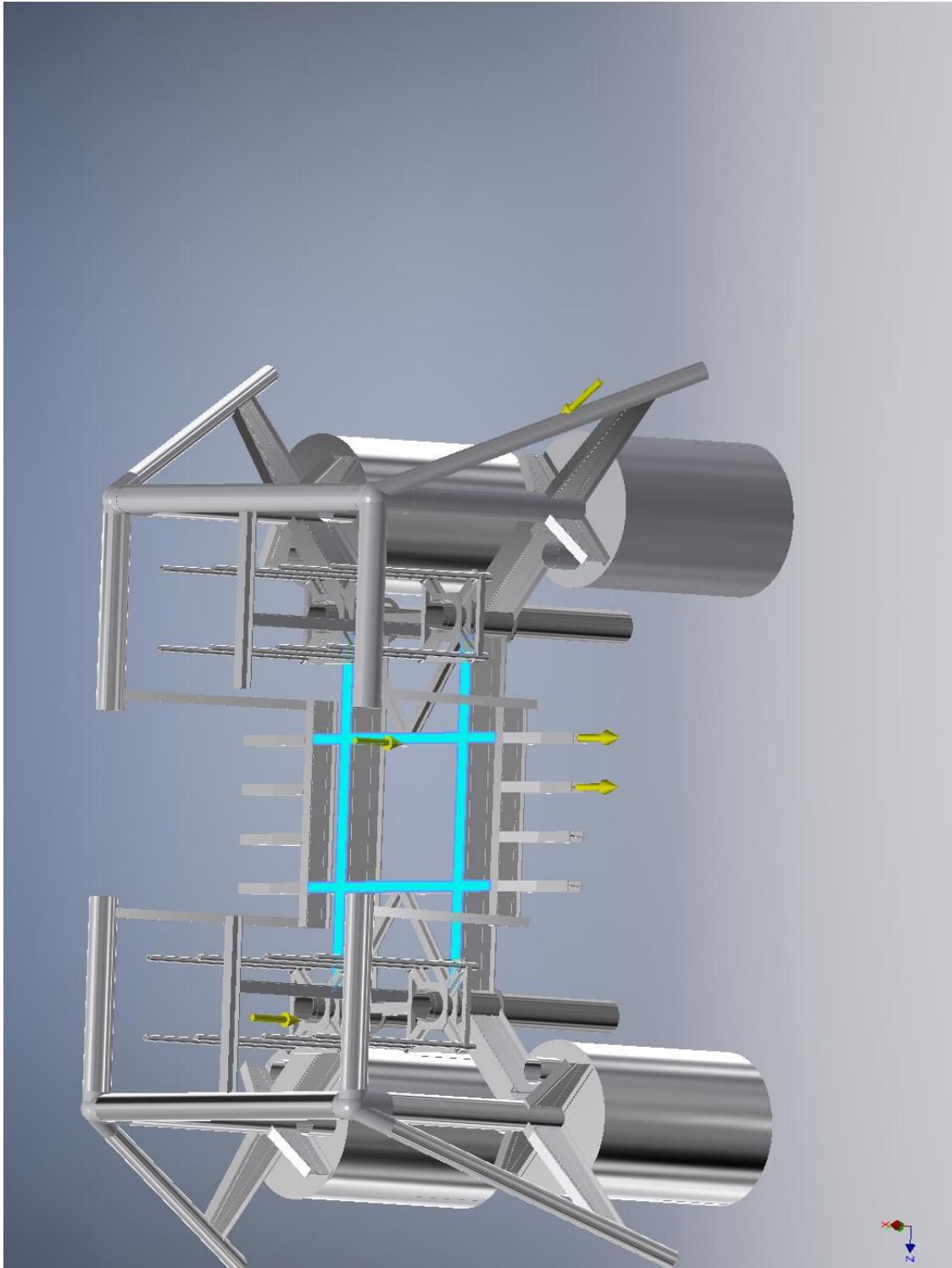
☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	800000.000 N
Vector X	-800000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	-300000.000 N
Vector Z	0.000 N

☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	300000.000 N
Vector X	-0.000 N
Vector Y	-300000.000 N
Vector Z	0.000 N

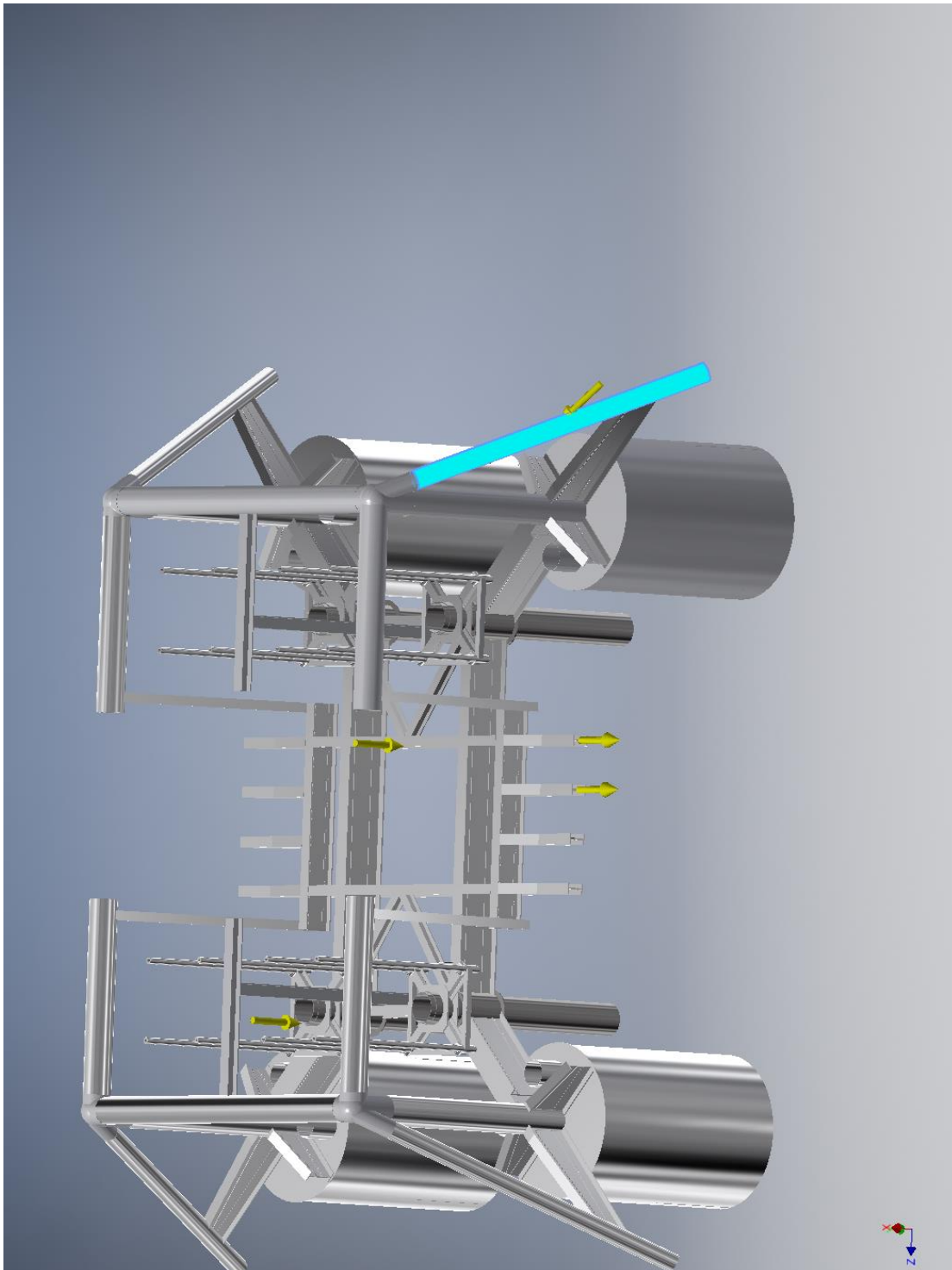
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	1047367.023 N
Vector X	342020.000 N
Vector Y	700000.000 N
Vector Z	700000.000 N

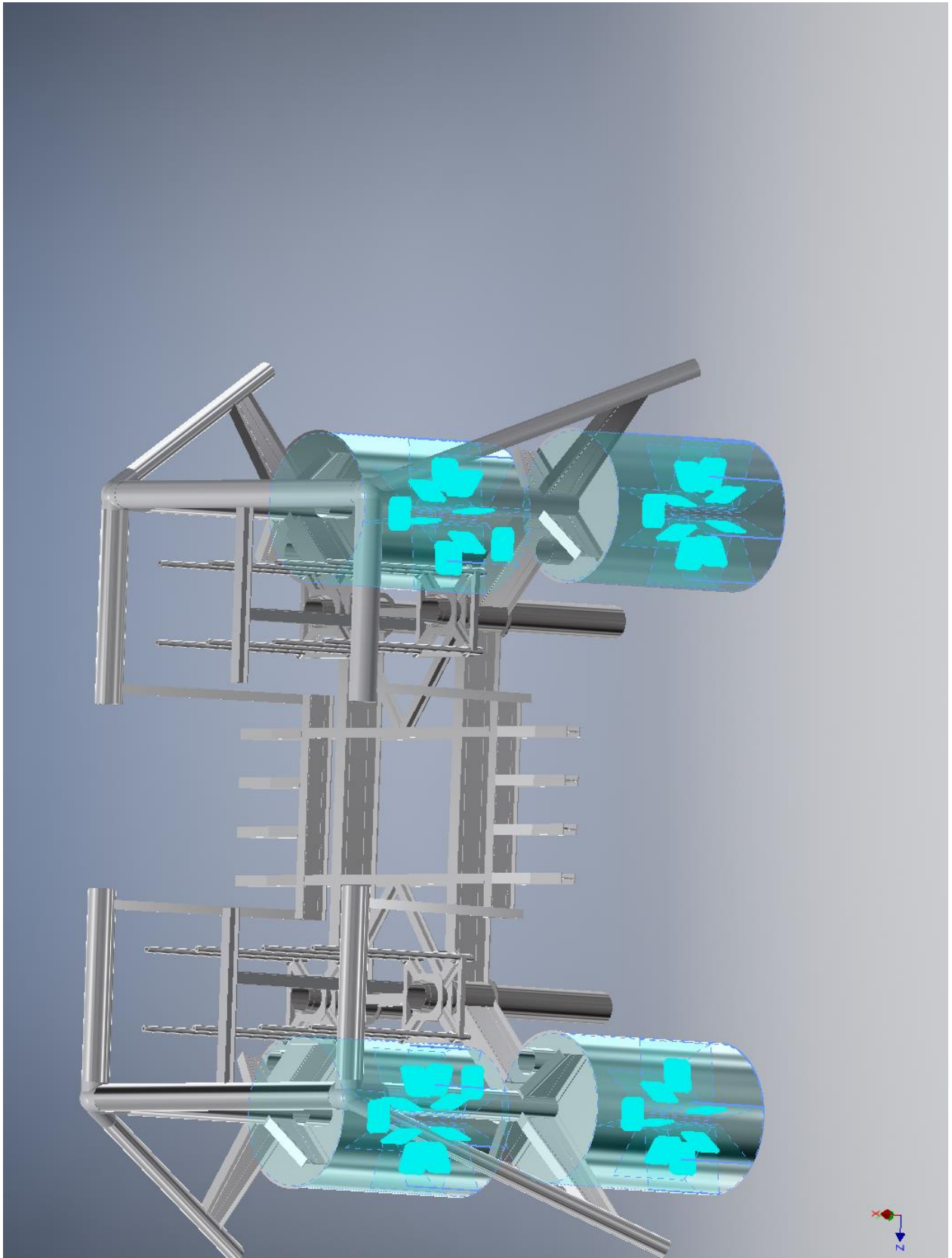
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
-----------------	------------------

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

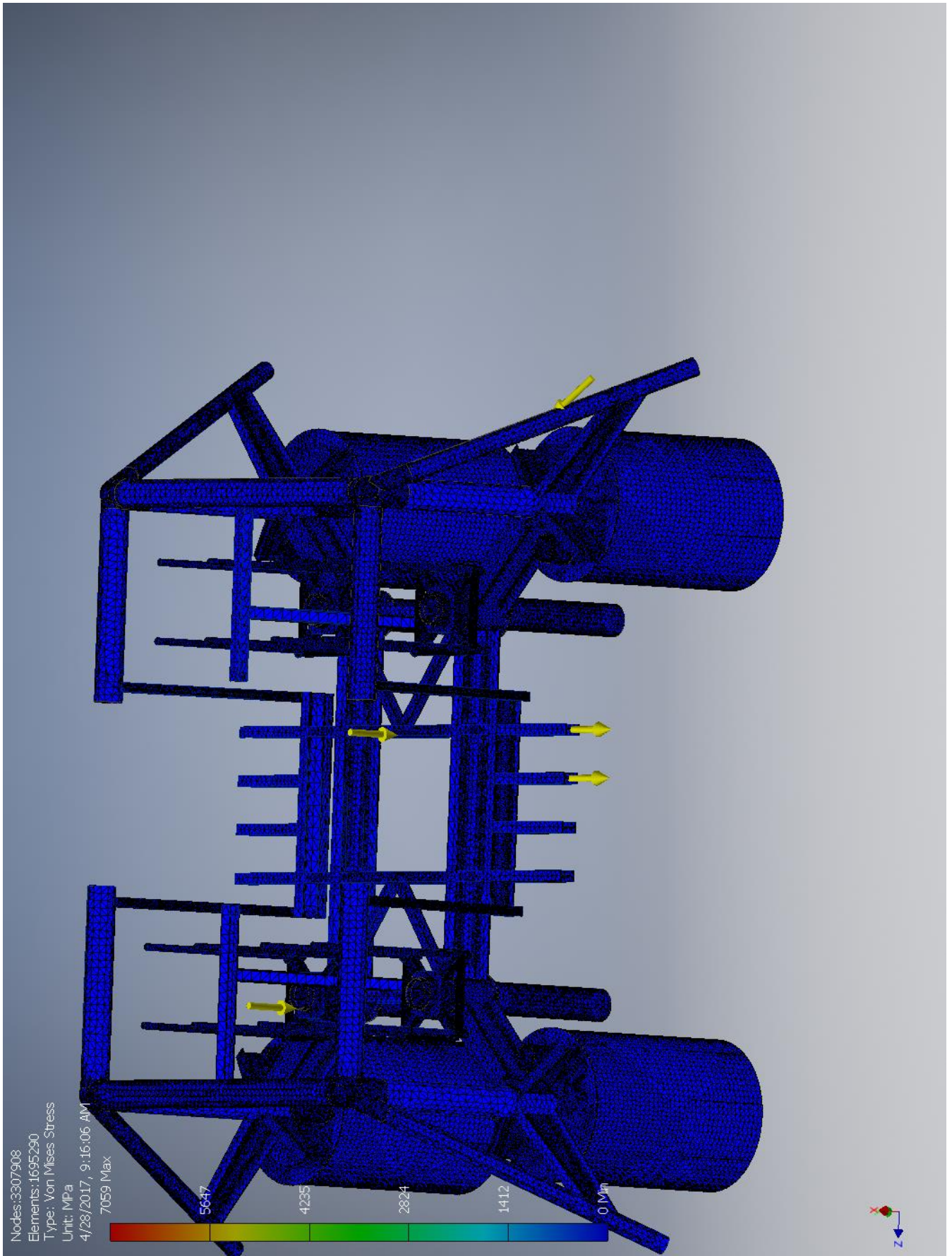
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	1272530 N	1057980 N	14740800 N m	-1341780 N m
		-100000 N		13020700 N m
		-700000 N		-6778770 N m

☐ Result Summary

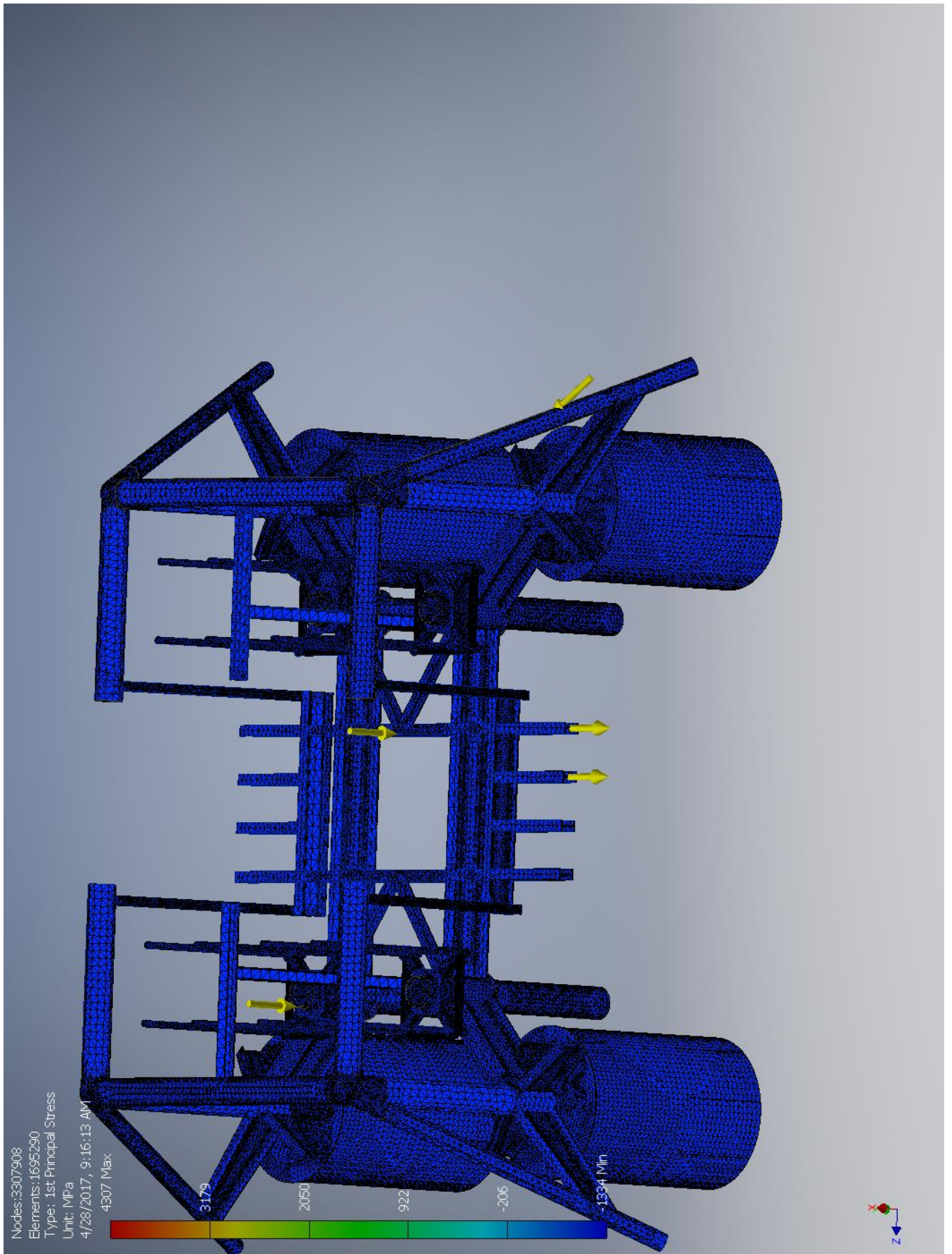
Name	Minimum	Maximum
Volume	6.02475E+10 mm ³	
Mass	181171 kg	
Von Mises Stress	0 MPa	7059.03 MPa
1st Principal Stress	-1334.08 MPa	4306.8 MPa
3rd Principal Stress	-8525.02 MPa	64.9763 MPa
Displacement	0 mm	20.6414 mm
Safety Factor	0.0389572 ul	15 ul

☐ Figures

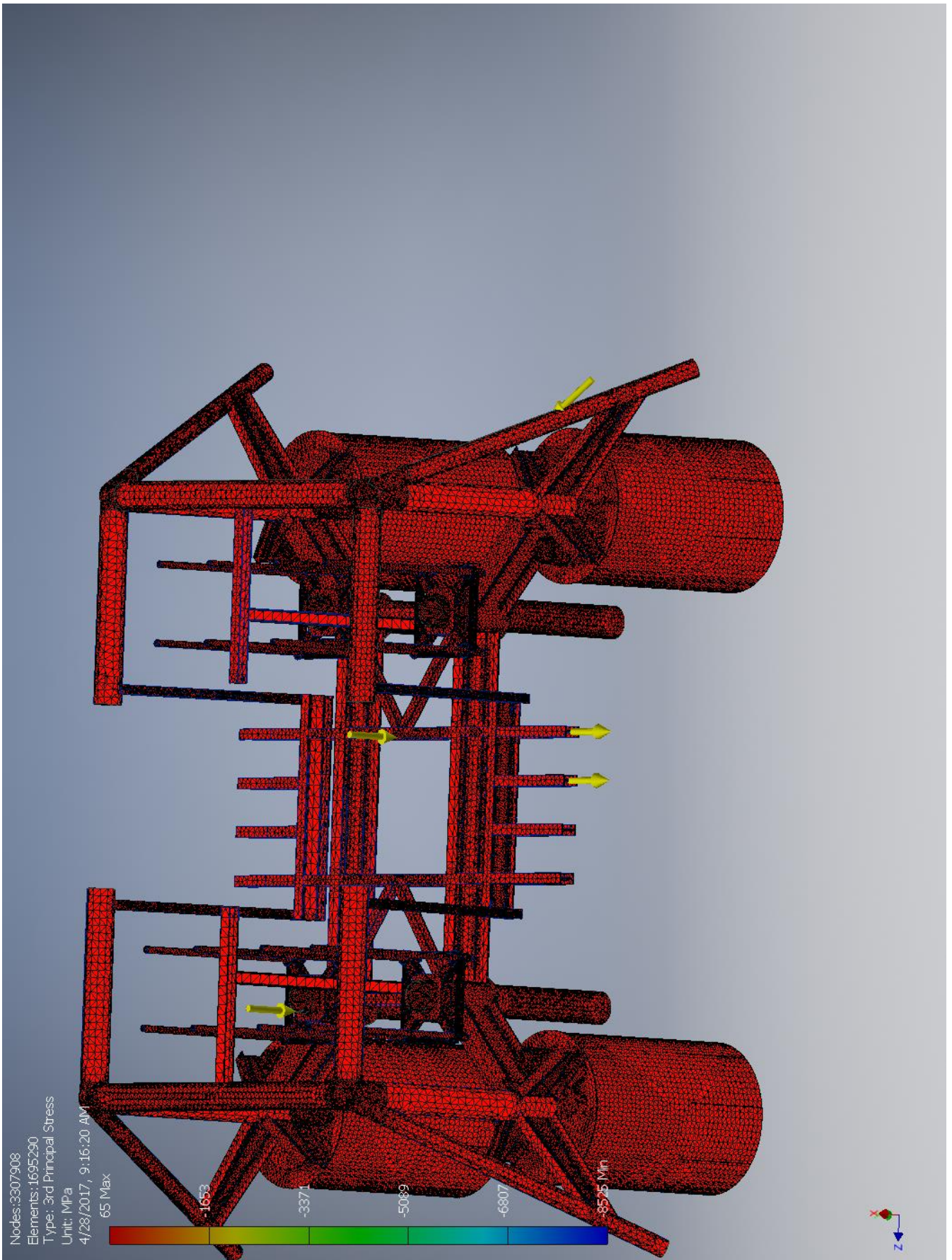
☐ Von Mises Stress



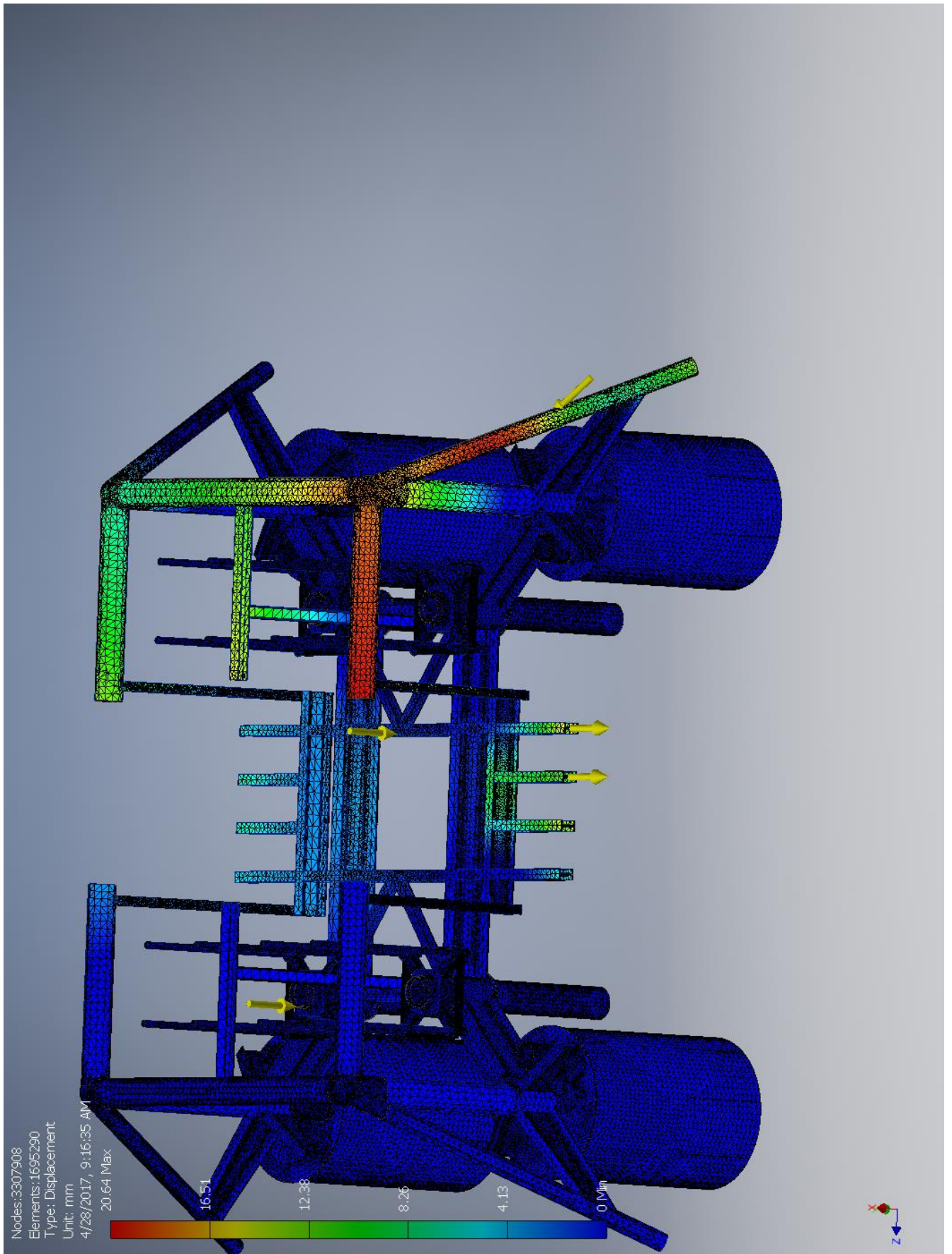
1st Principal Stress



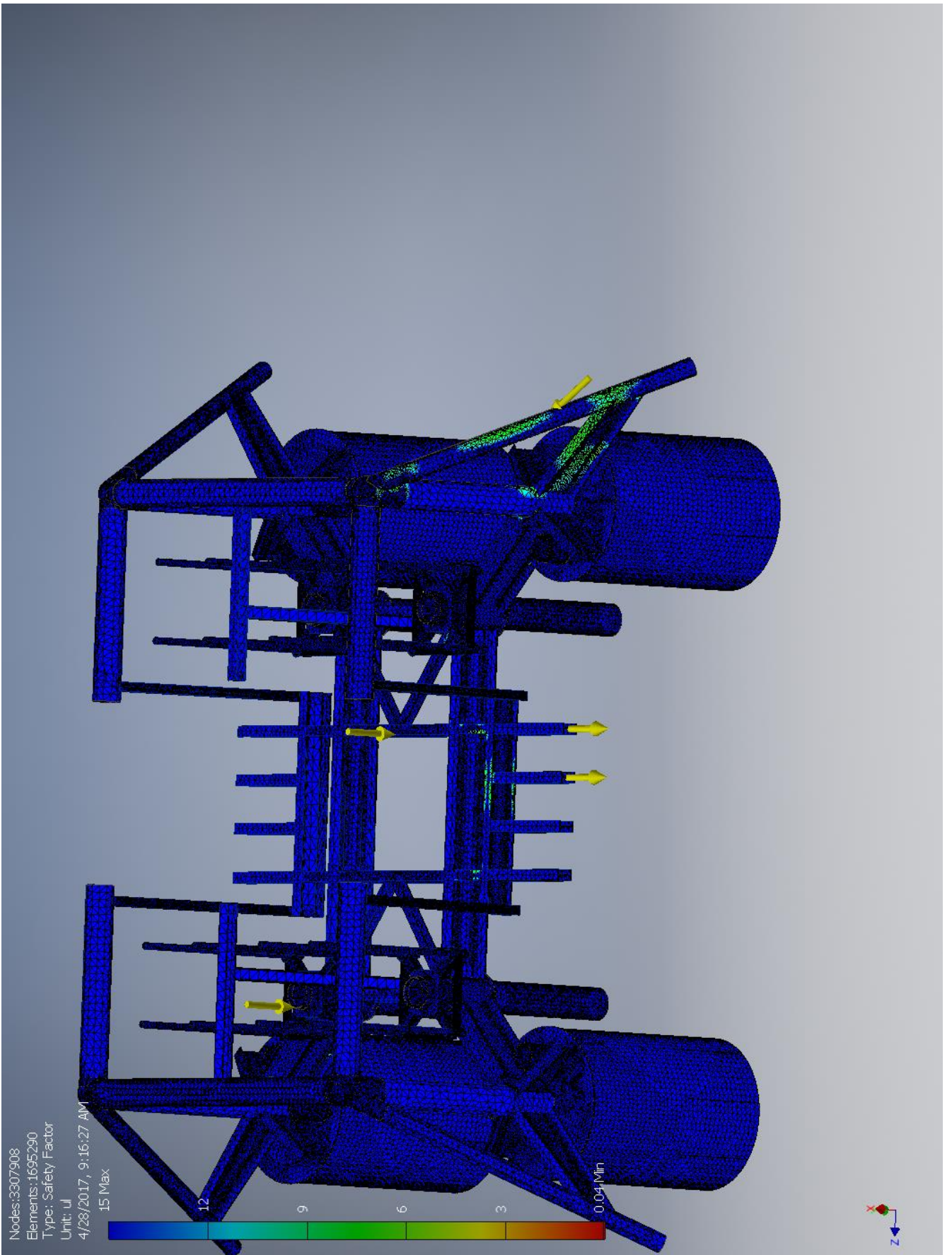
3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.6 Case-F

I.6.1 Without convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/28/2017, 12:02 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	181171 kg
Area	3.52908E+09 mm ²
Volume	6.02475E+10 mm ³
Center of Gravity	x=-8288.42 mm y=6548.24 mm z=9857.84 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-F

General objective and settings:

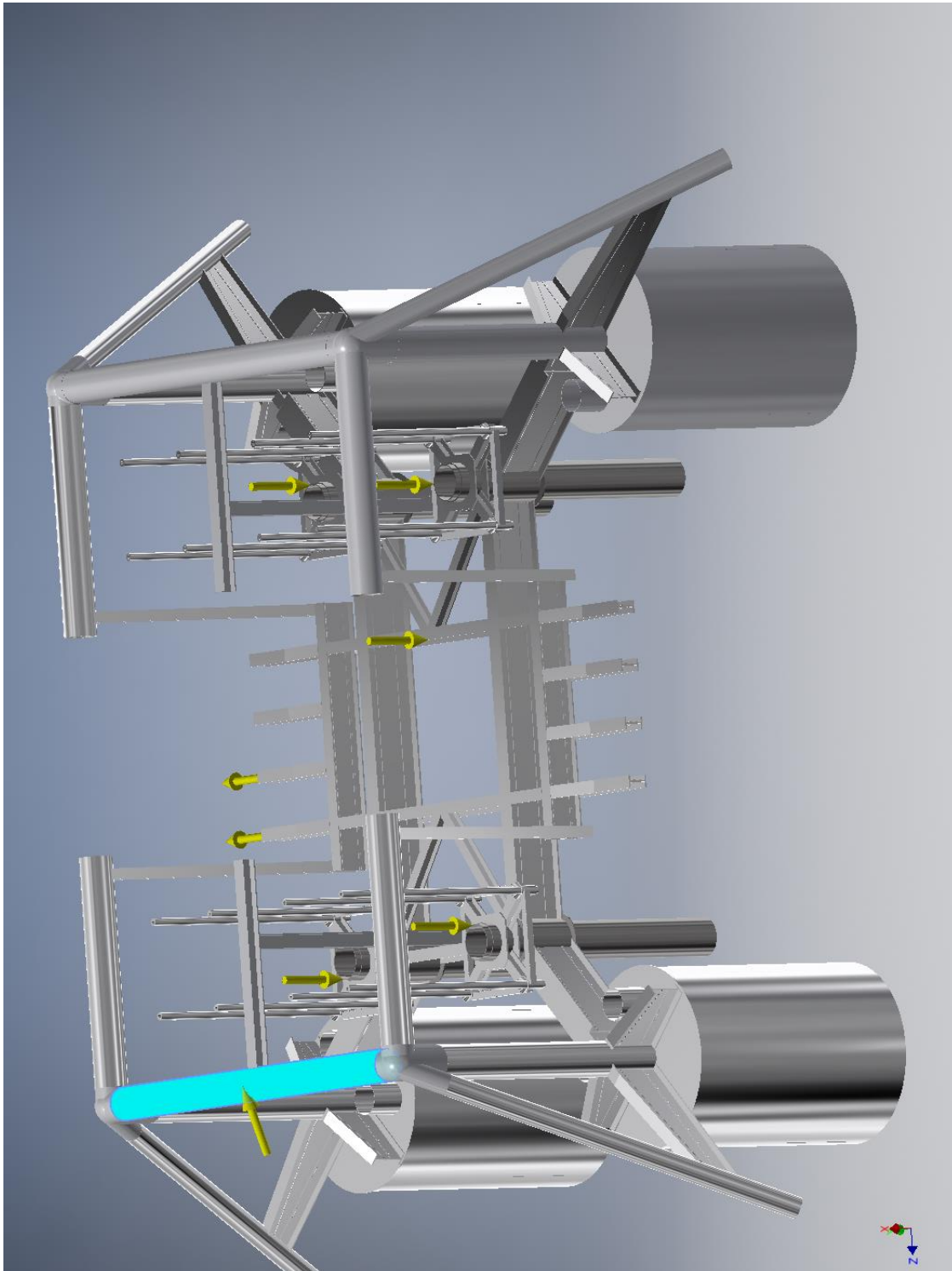
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/28/2017, 11:46 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.293 N
Vector X	342021.000 N
Vector Y	0.000 N
Vector Z	-939692.621 N

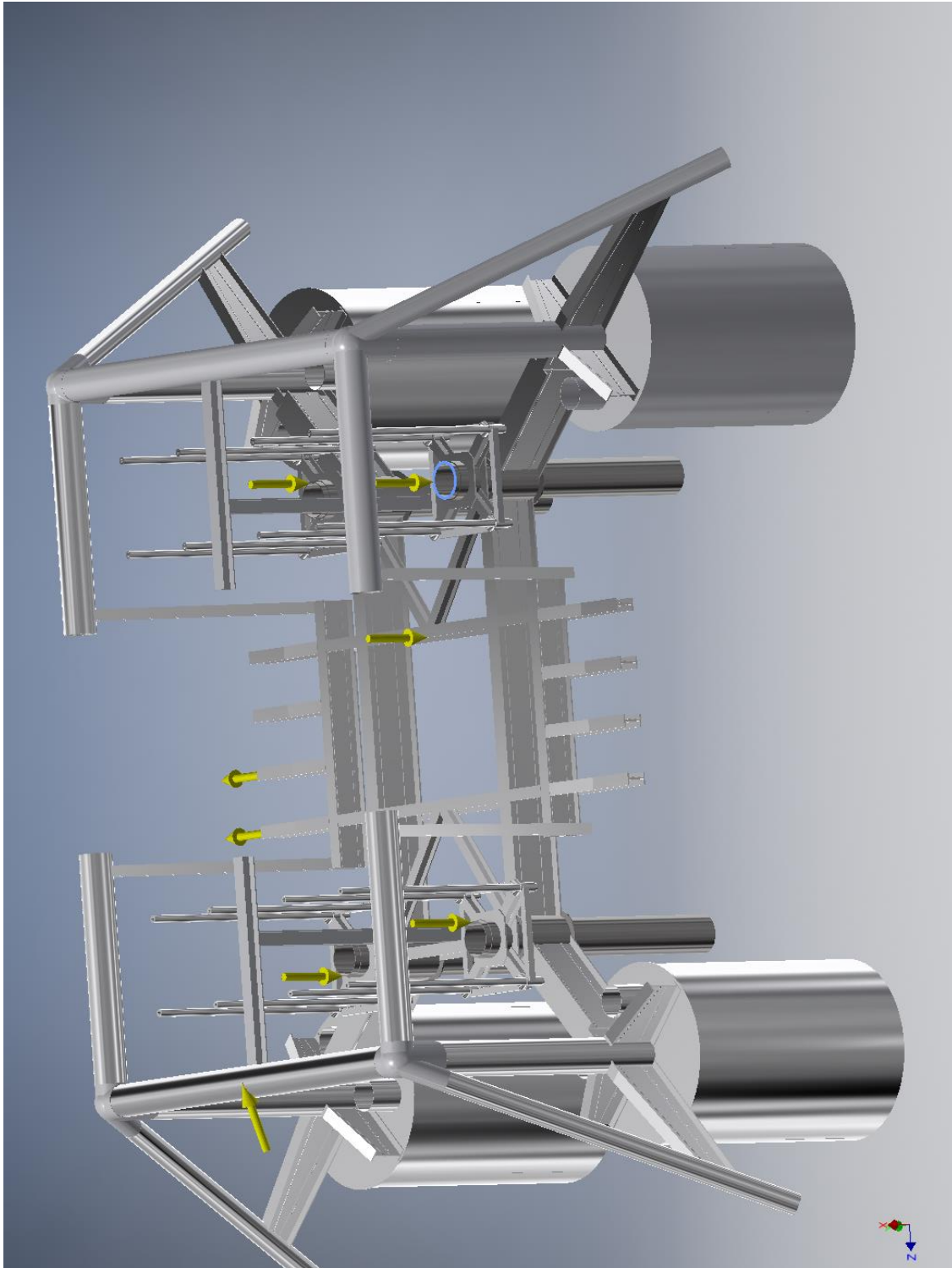
☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

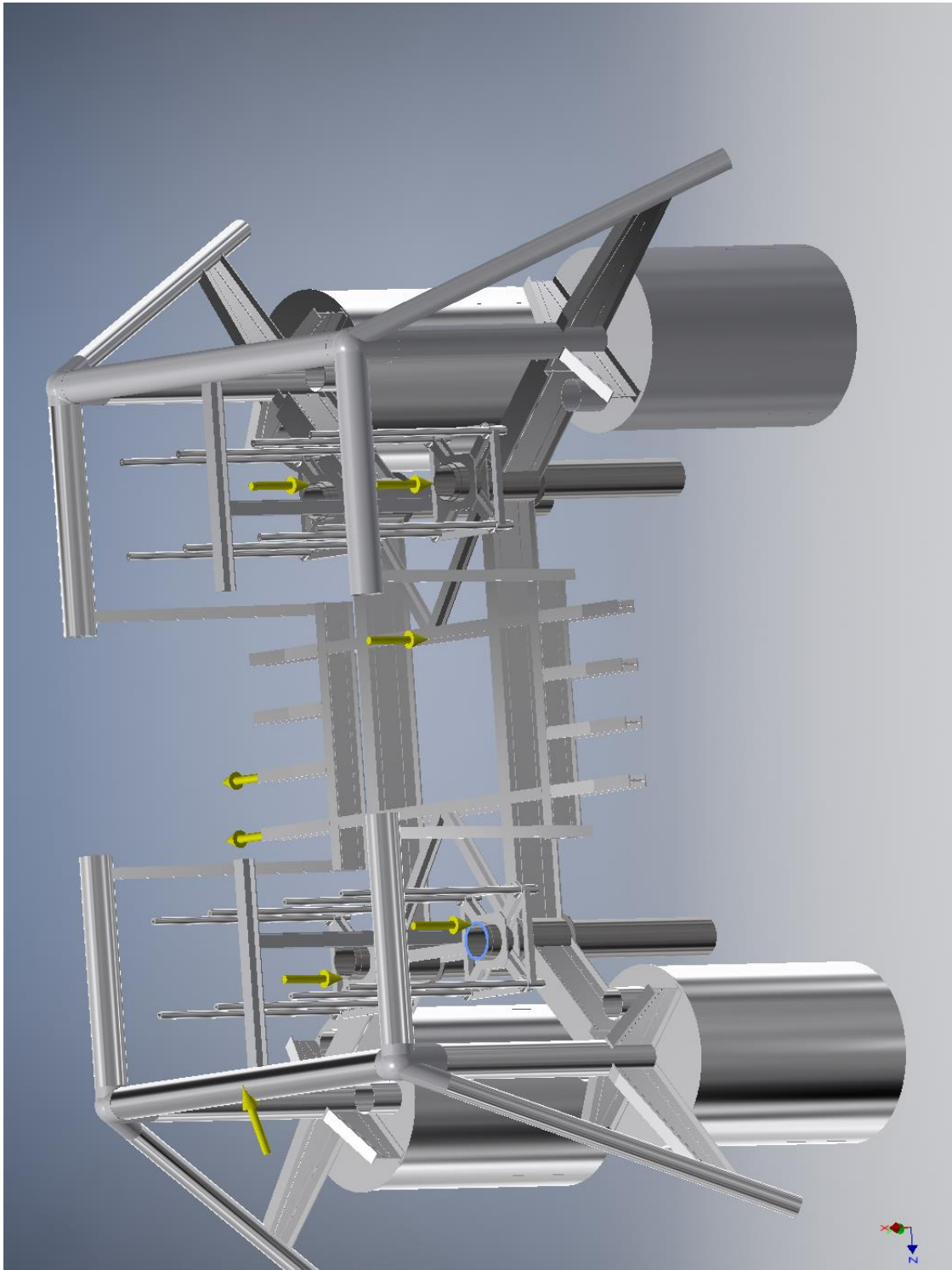
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

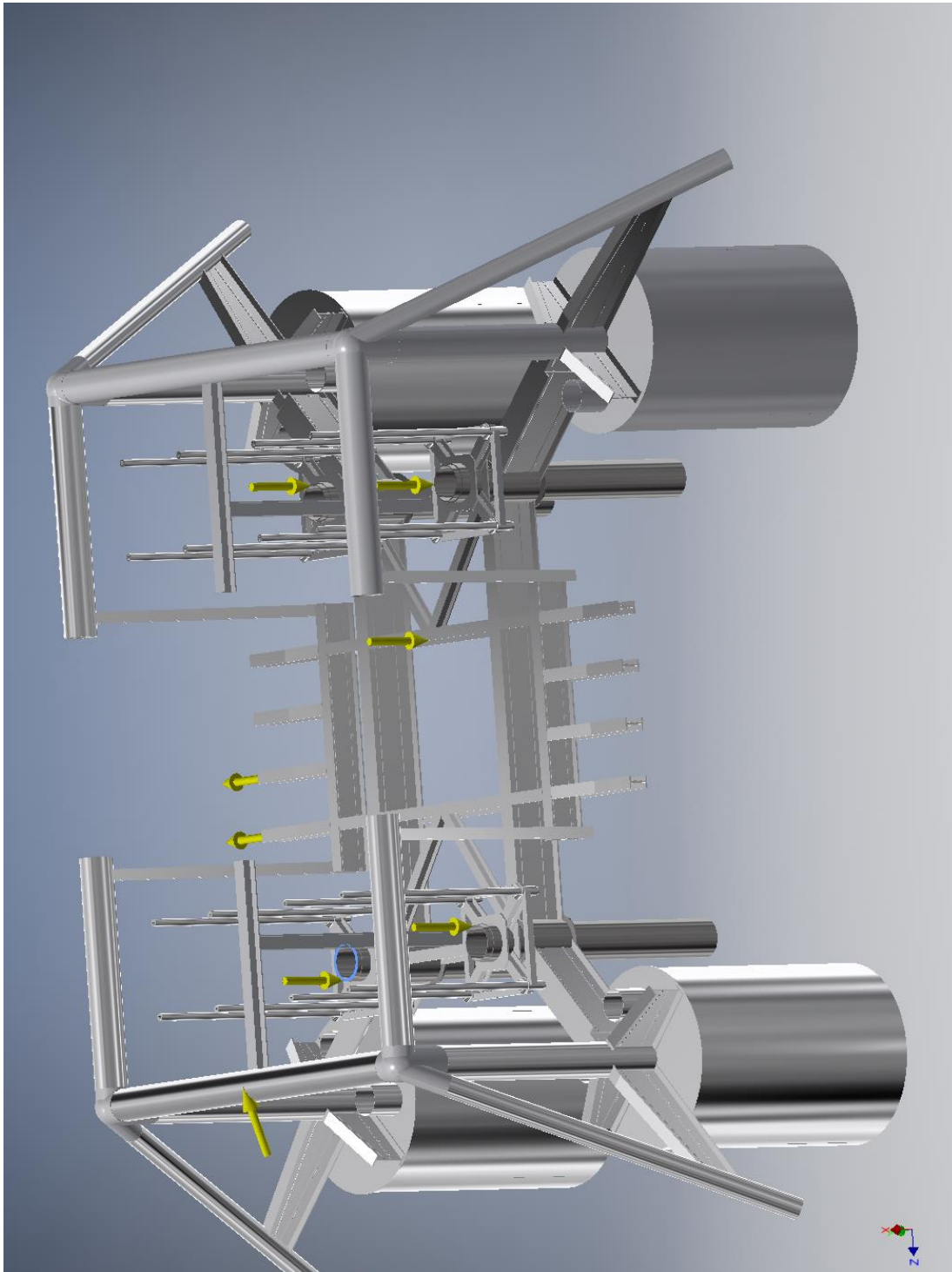
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

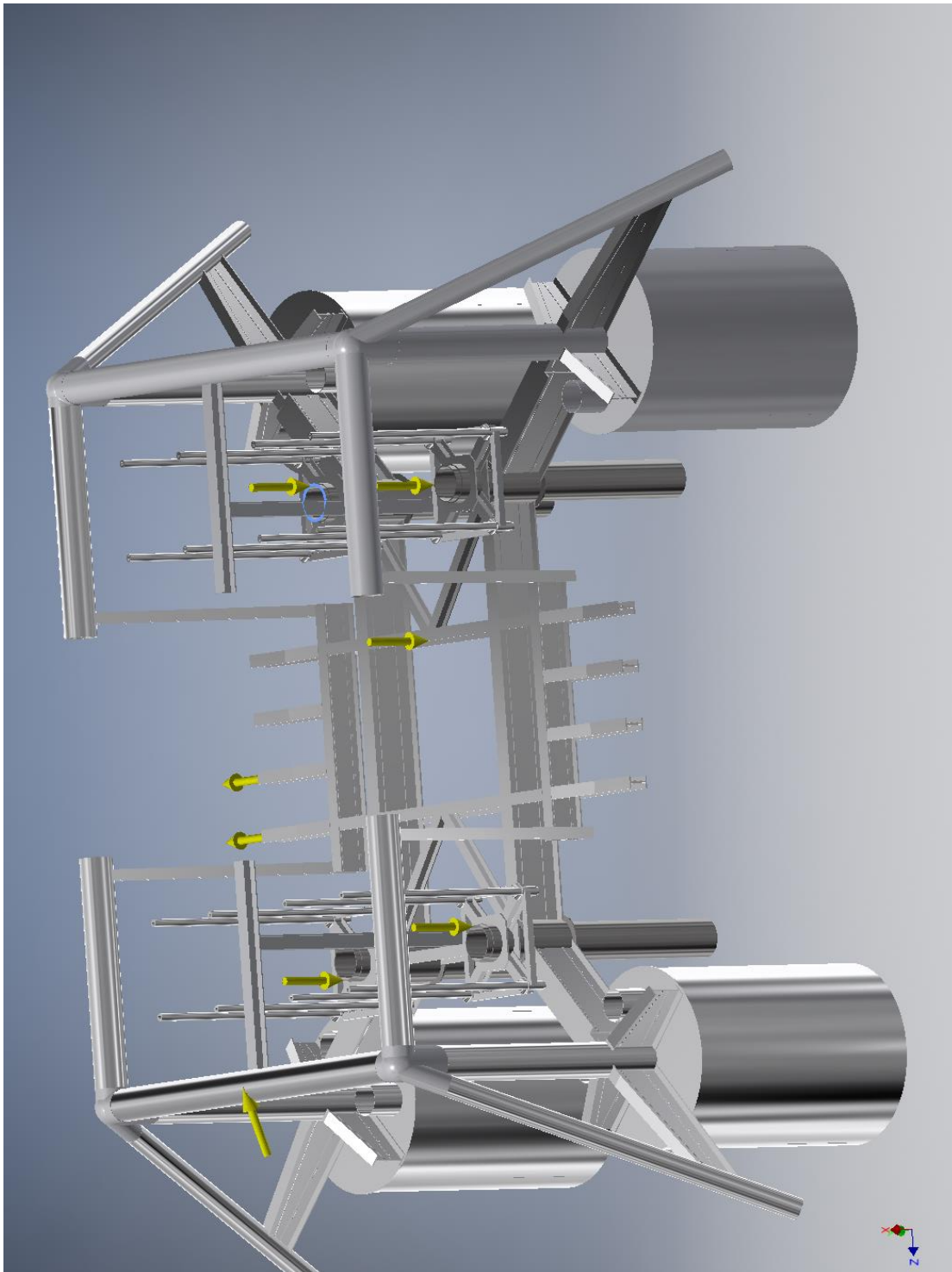
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

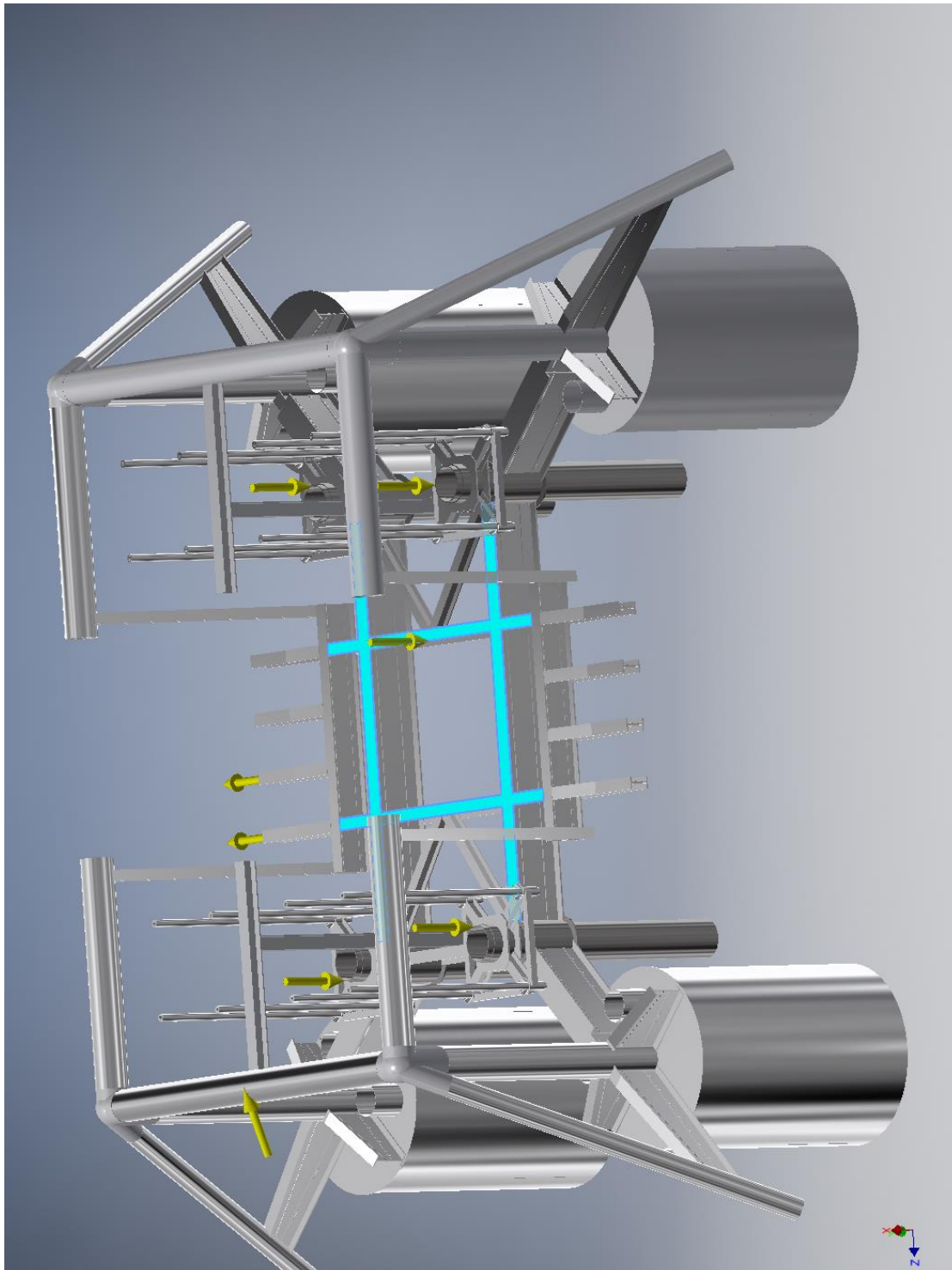
☐ **Selected Face(s)**



☐ **Force:6**

Load Type	Force
Magnitude	800000.000 N
Vector X	-800000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

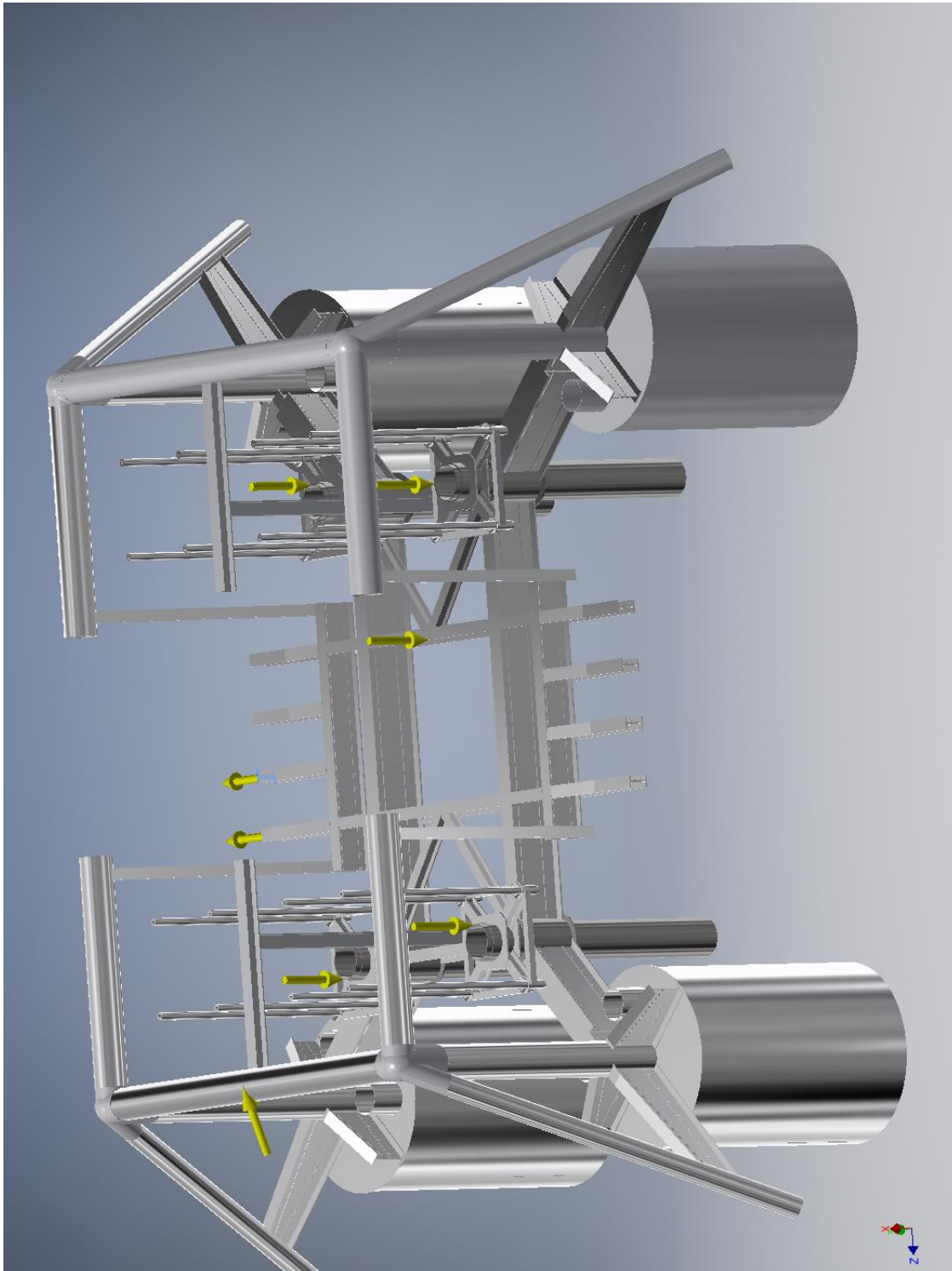
☐ **Selected Face(s)**



☐ **Force:7**

Load Type	Force
Magnitude	300000.000 N
Vector X	-0.000 N
Vector Y	300000.000 N
Vector Z	-0.000 N

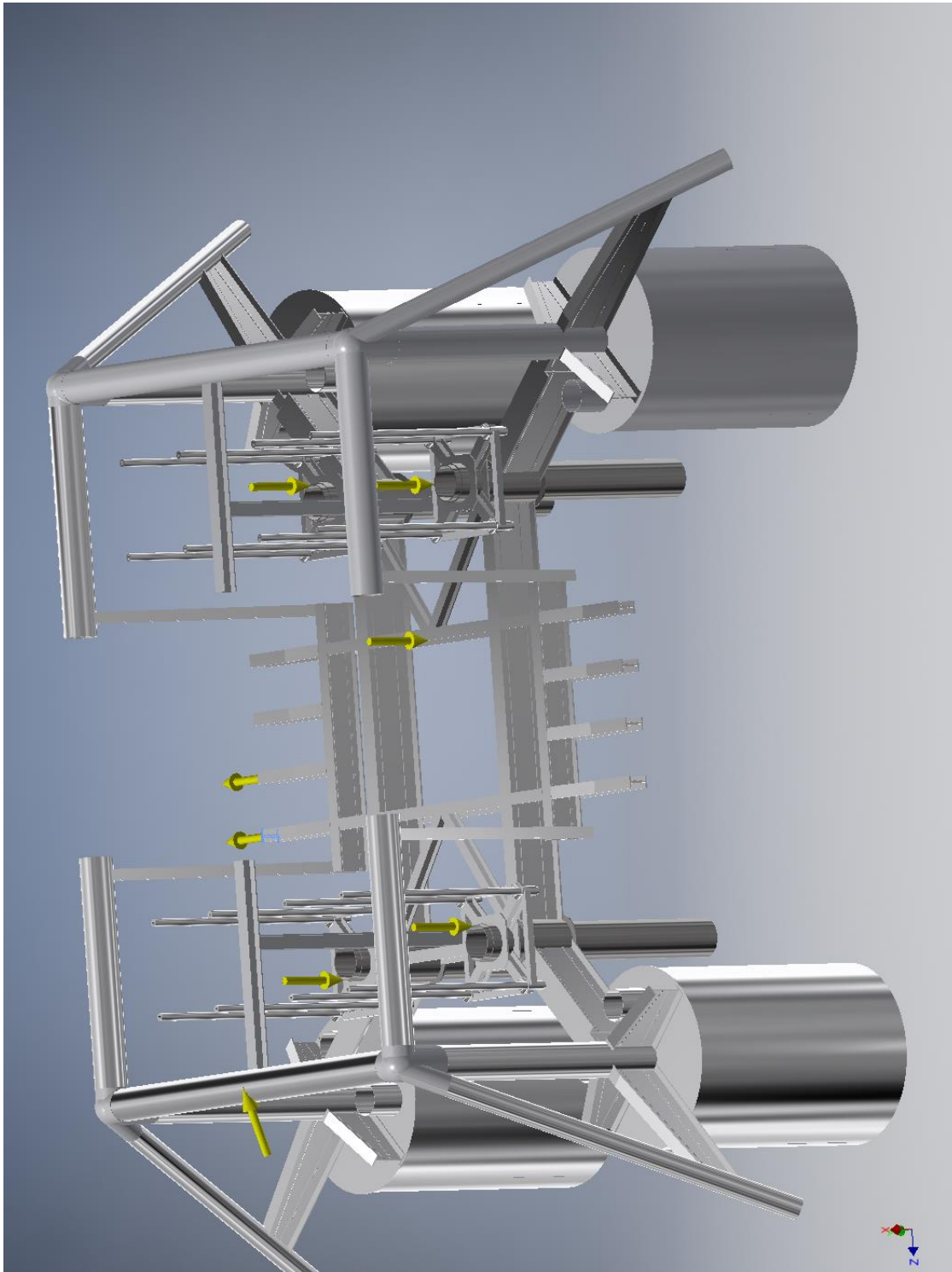
☐ **Selected Face(s)**



☐ **Force:8**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	300000.000 N
Vector Z	-0.000 N

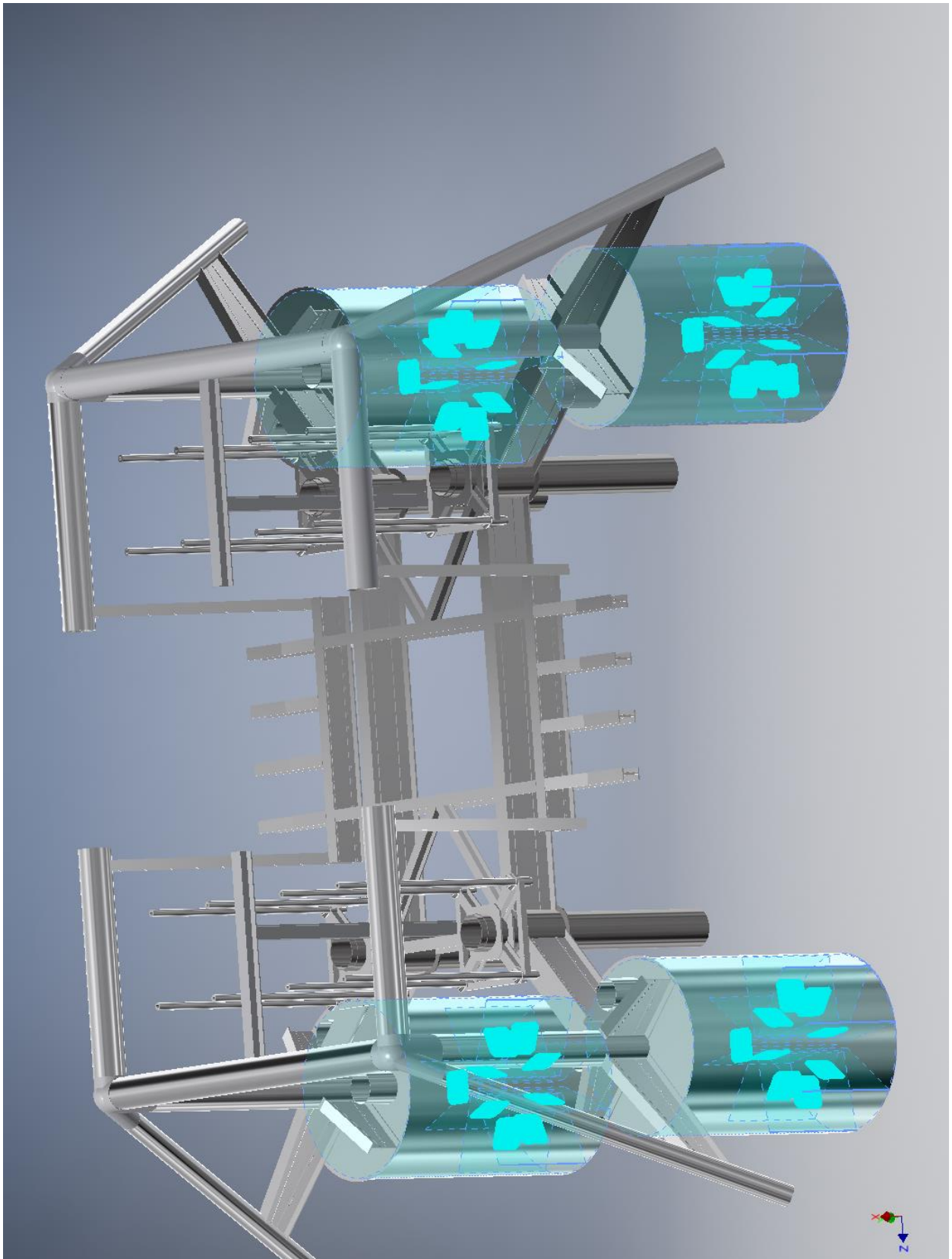
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type	Fixed Constraint
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☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

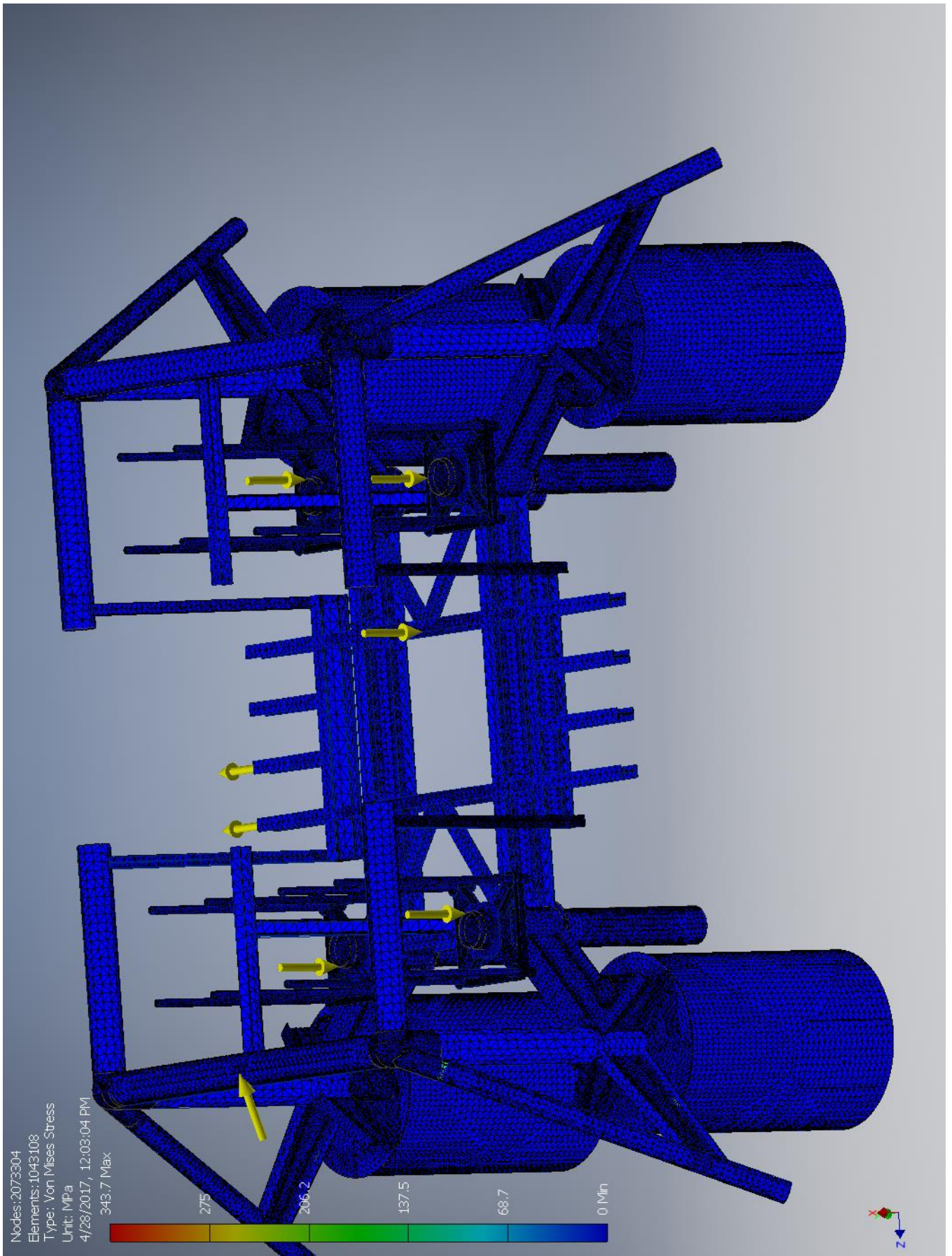
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	3067750 N	2857980 N	15469400 N m	1016440 N m
		-600000 N		-15111800 N m
		939693 N		-3146730 N m

☐ Result Summary

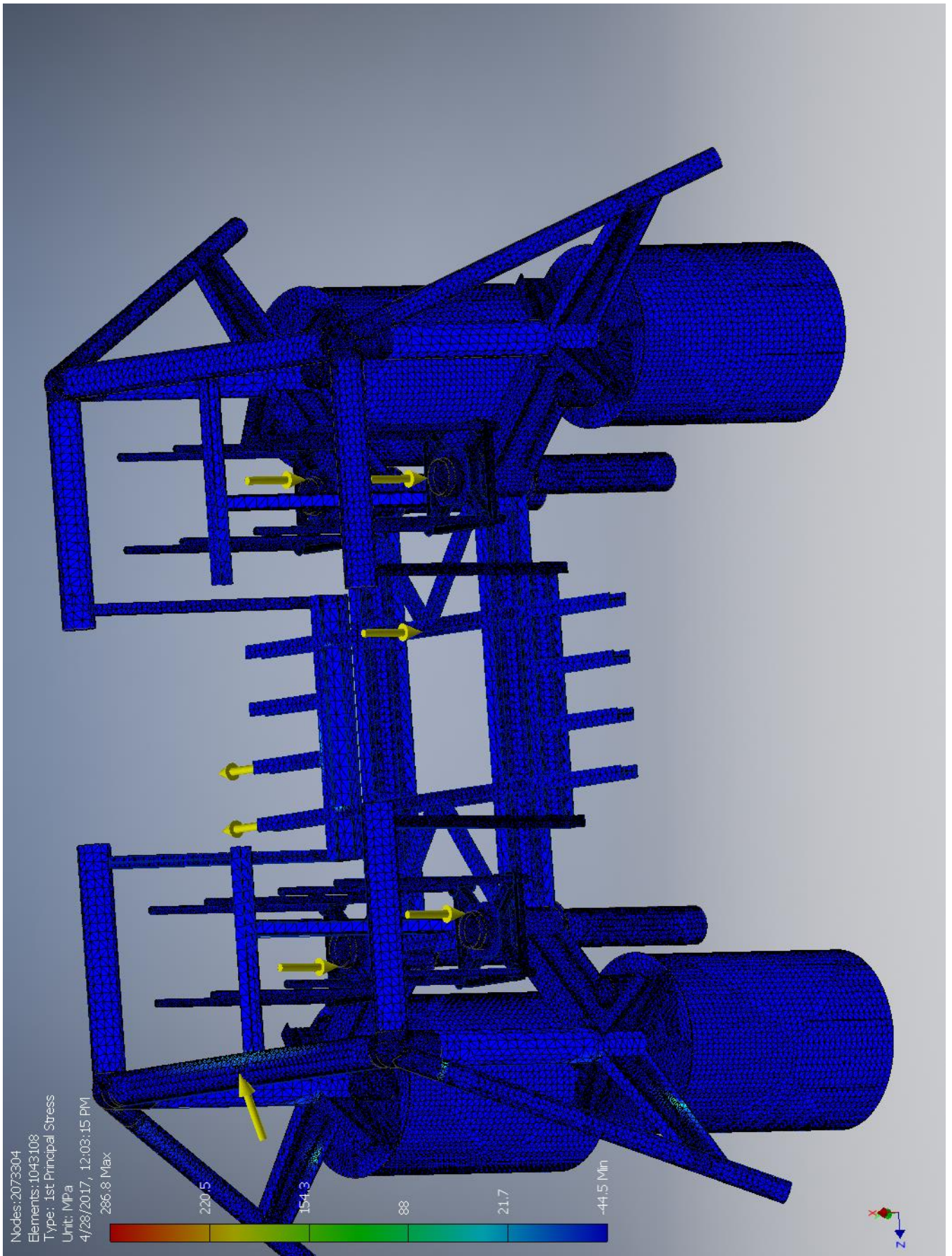
Name	Minimum	Maximum
Volume	6.02475E+10 mm ³	
Mass	181171 kg	
Von Mises Stress	0 MPa	343.71 MPa
1st Principal Stress	-44.5052 MPa	286.756 MPa
3rd Principal Stress	-268.213 MPa	44.1763 MPa
Displacement	0 mm	65.0658 mm
Safety Factor	0.687916 ul	15 ul

☐ Figures

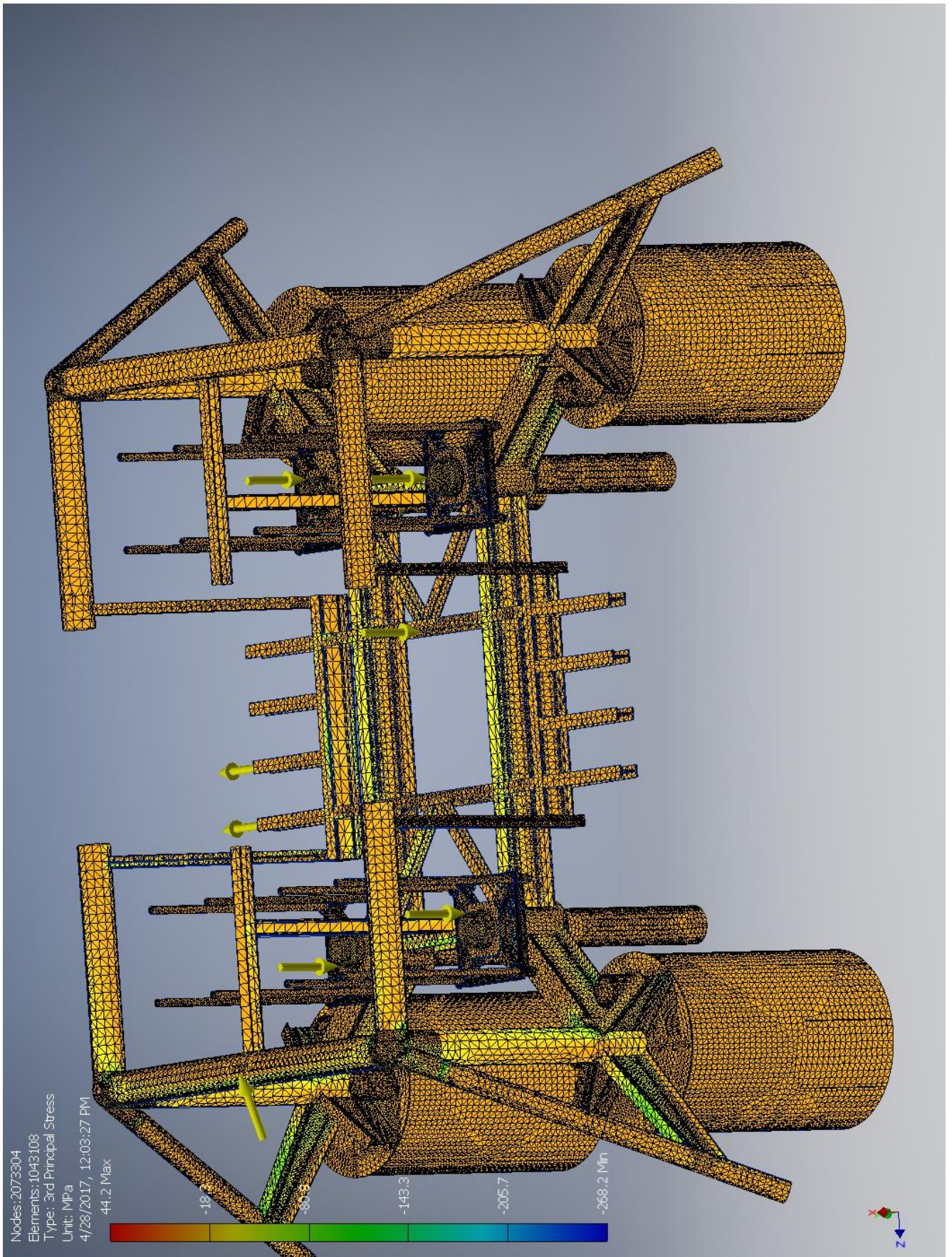
☐ Von Mises Stress



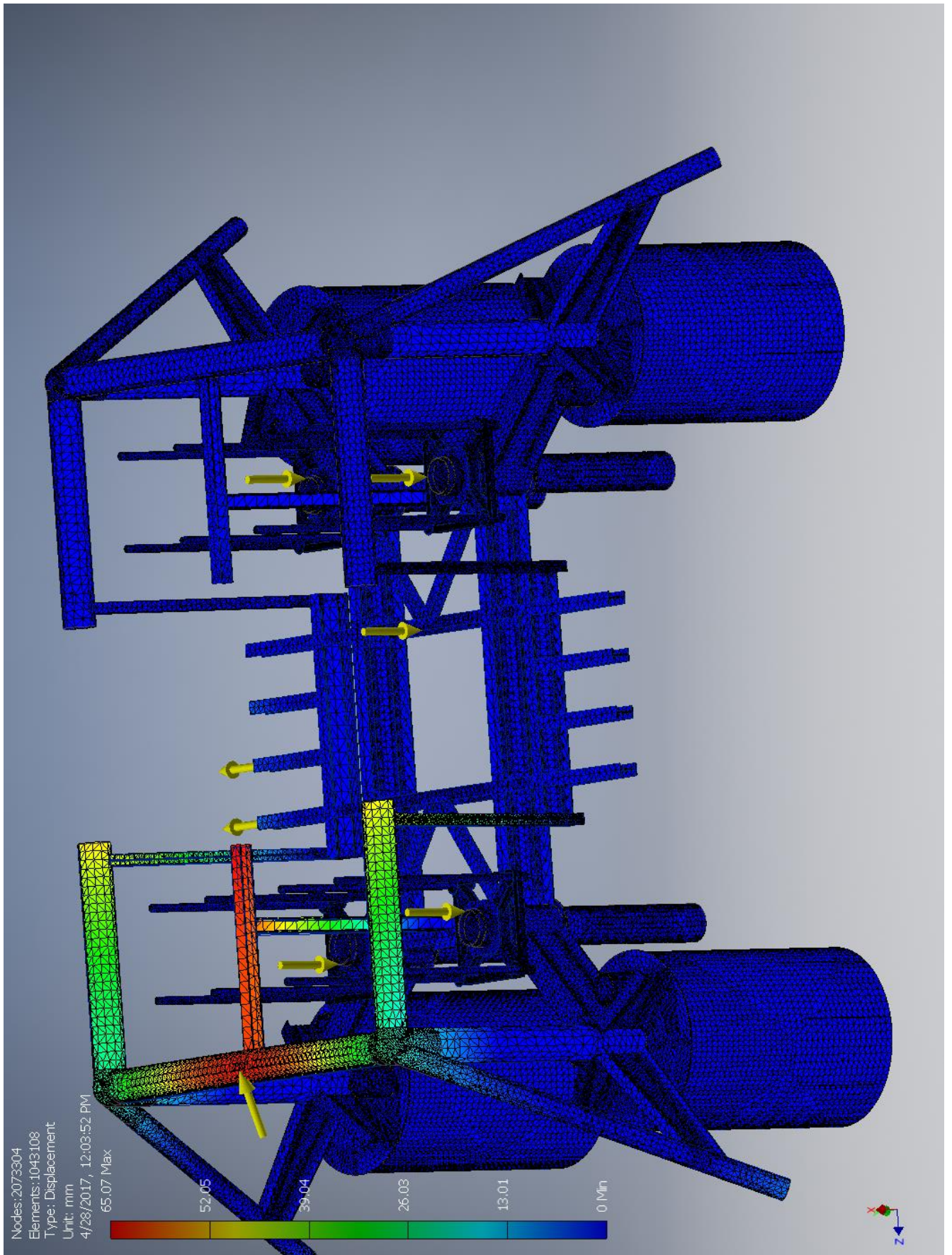
☐ 1st Principal Stress



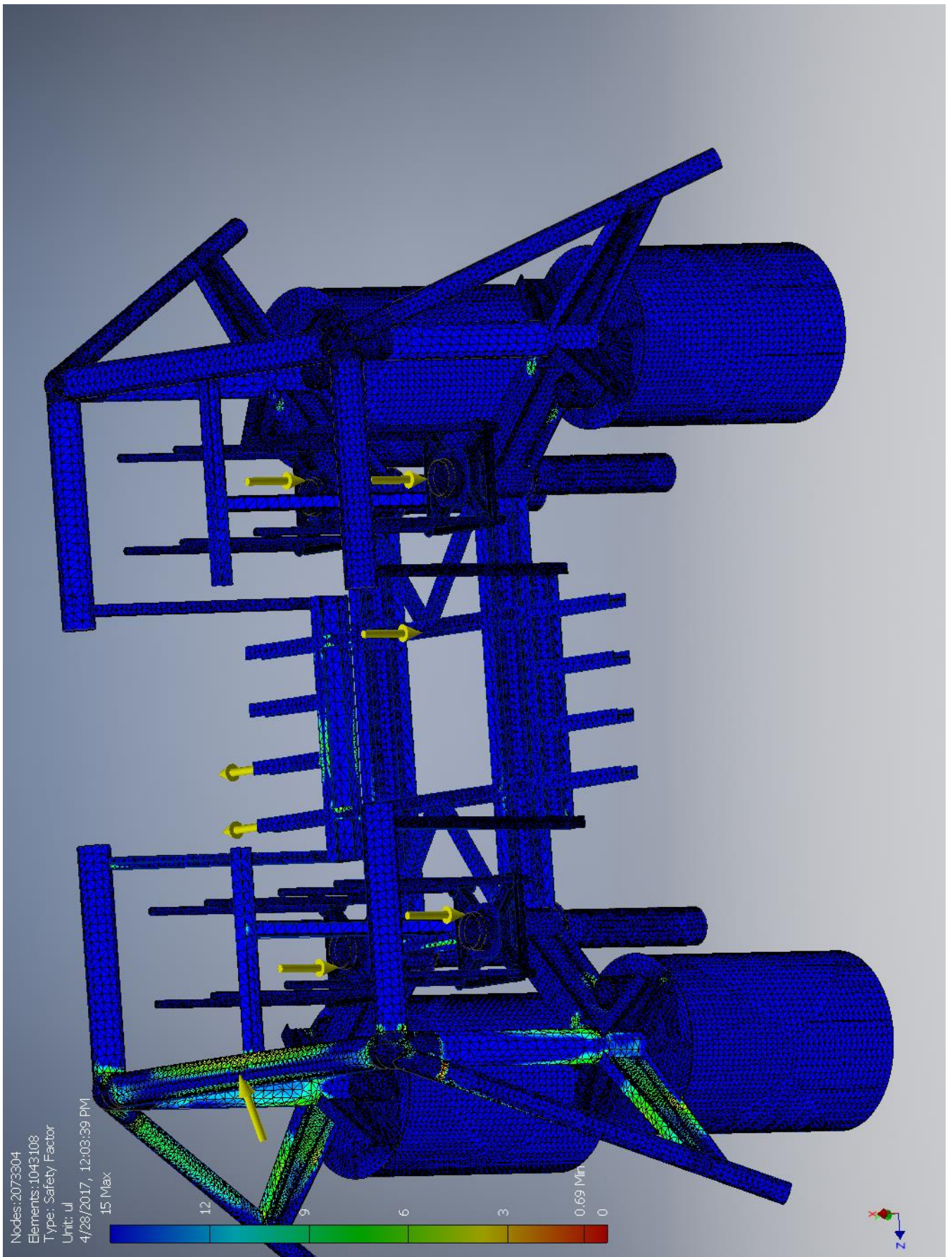
☐ 3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.6.2 With convergence

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/28/2017, 11:30 AM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	181171 kg
Area	3.52908E+09 mm ²
Volume	6.02475E+10 mm ³
Center of Gravity	x=-8288.42 mm y=6548.24 mm z=9857.84 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-F

General objective and settings:

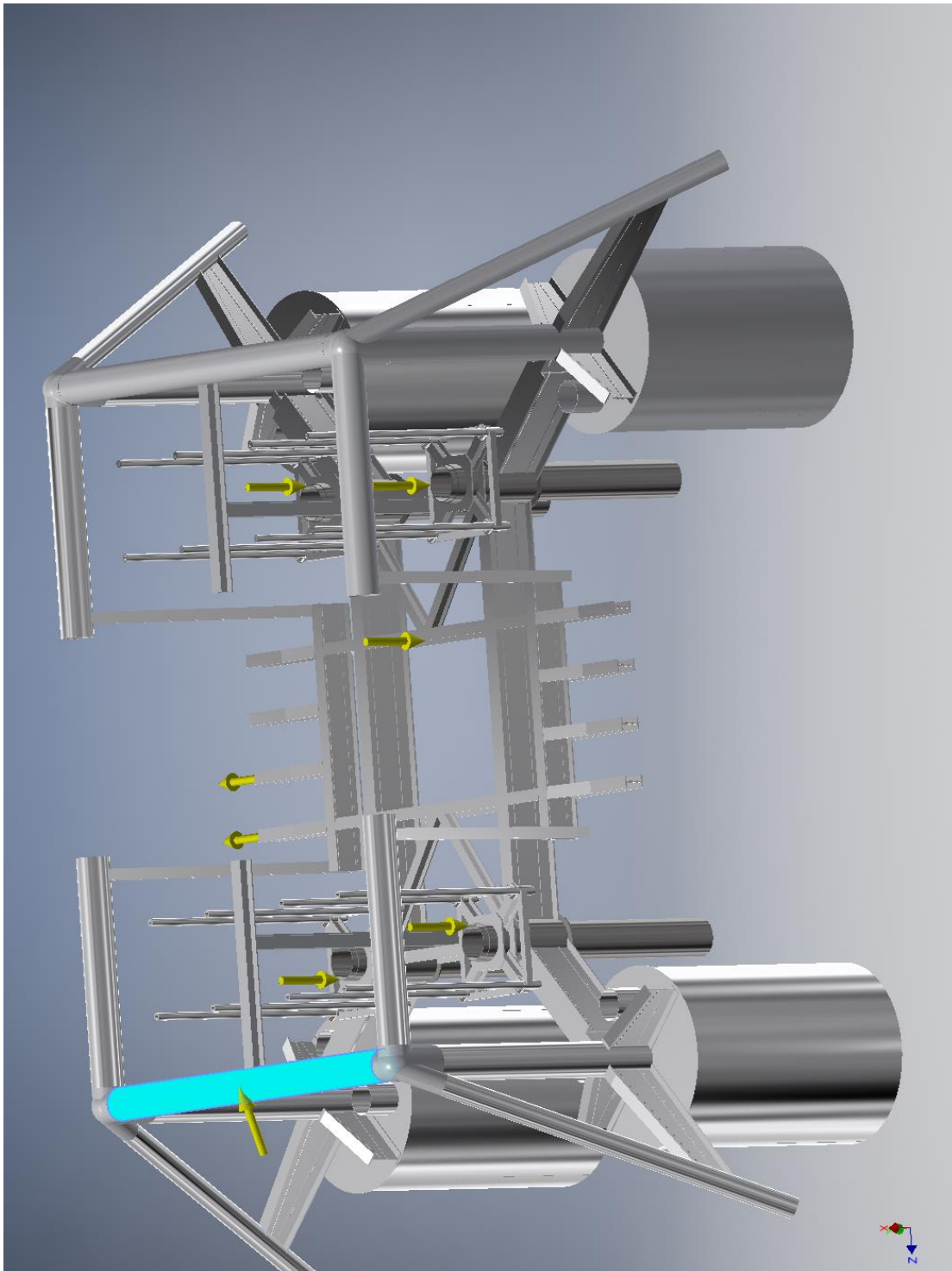
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/28/2017, 10:56 AM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

☐ **Force:1**

Load Type	Force
Magnitude	1000000.293 N
Vector X	342021.000 N
Vector Y	0.000 N
Vector Z	-939692.621 N

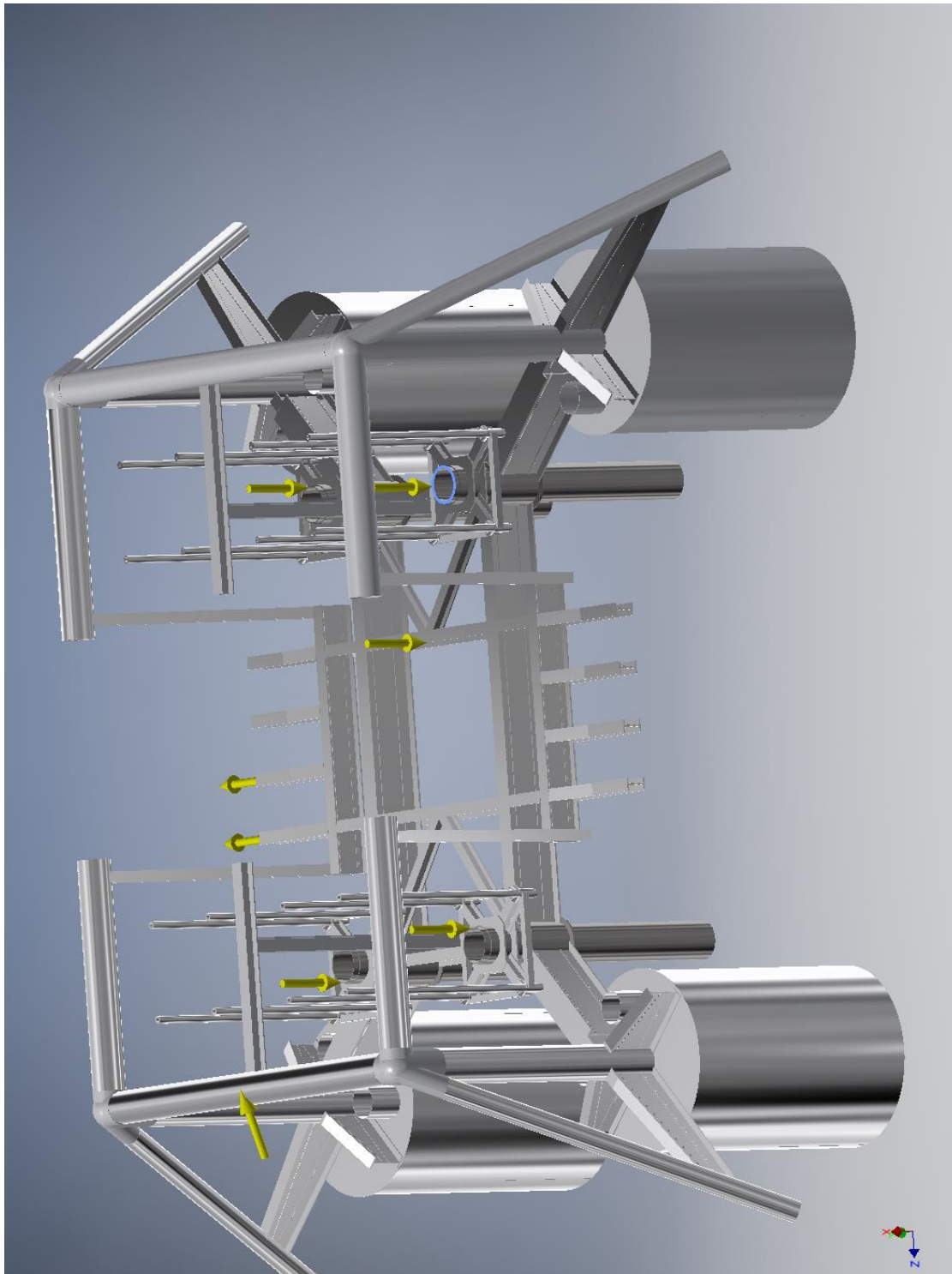
☐ **Selected Face(s)**



☐ **Force:2**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

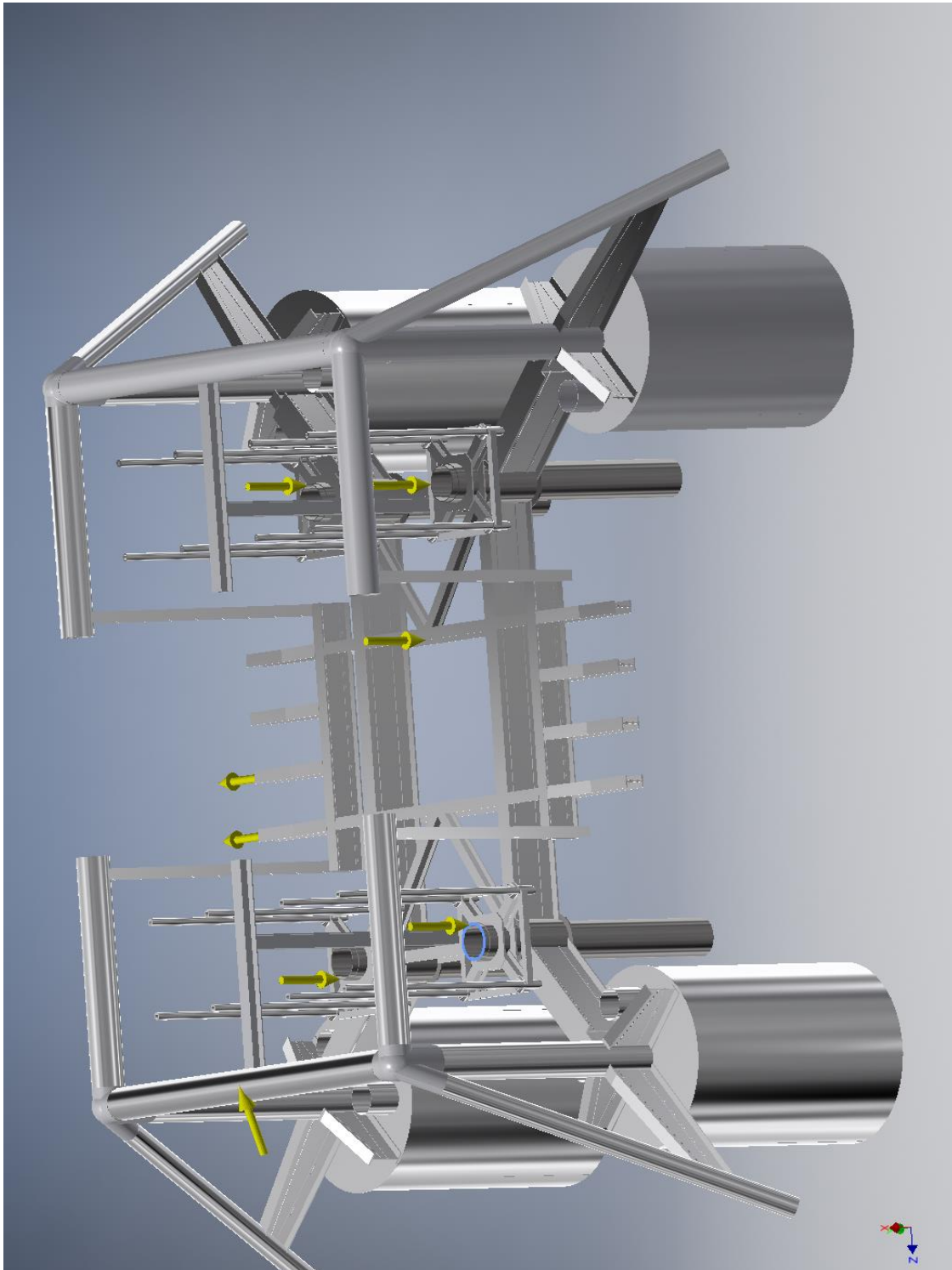
☐ **Selected Face(s)**



☐ **Force:3**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

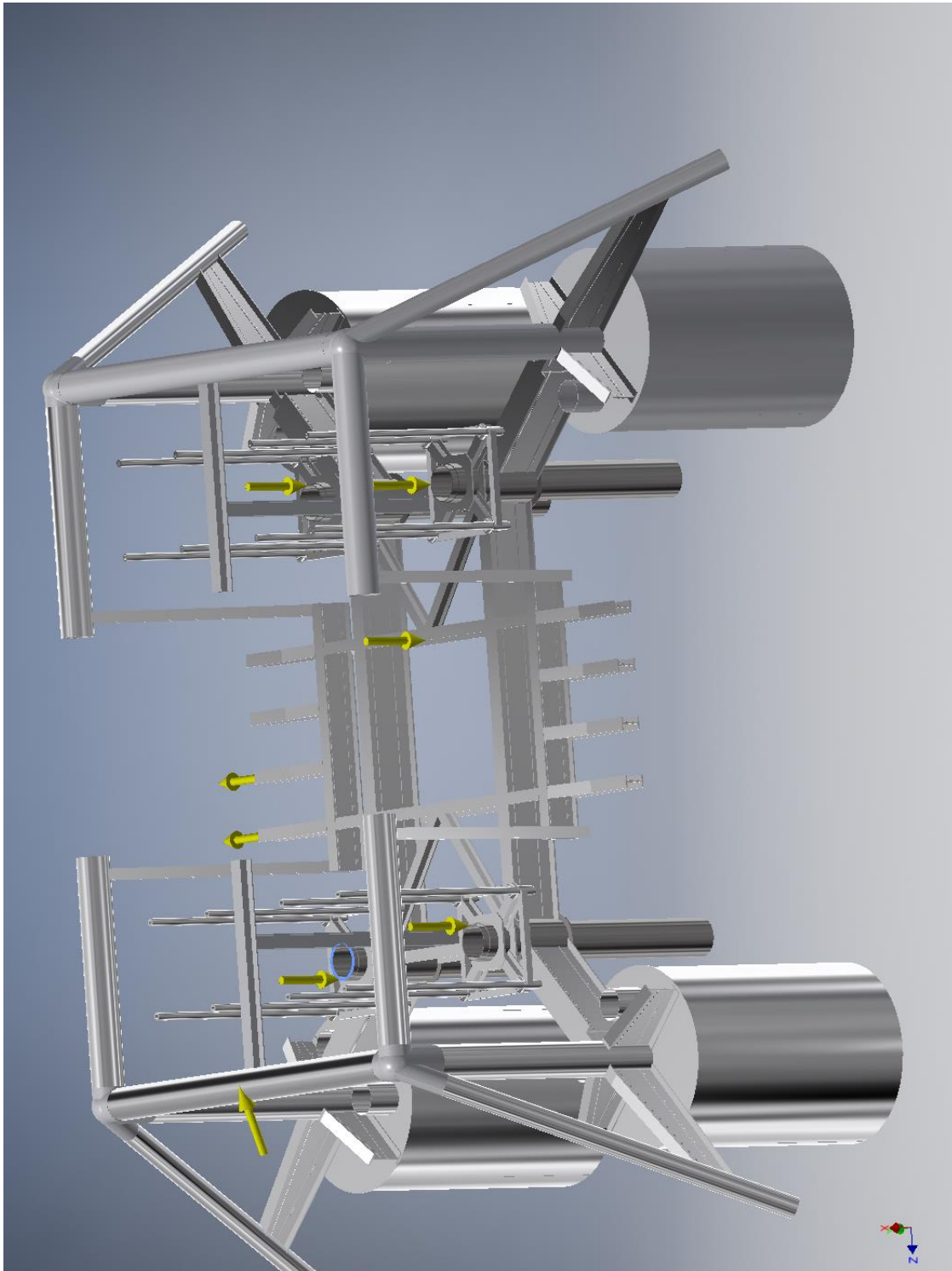
☐ **Selected Face(s)**



☐ **Force:4**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

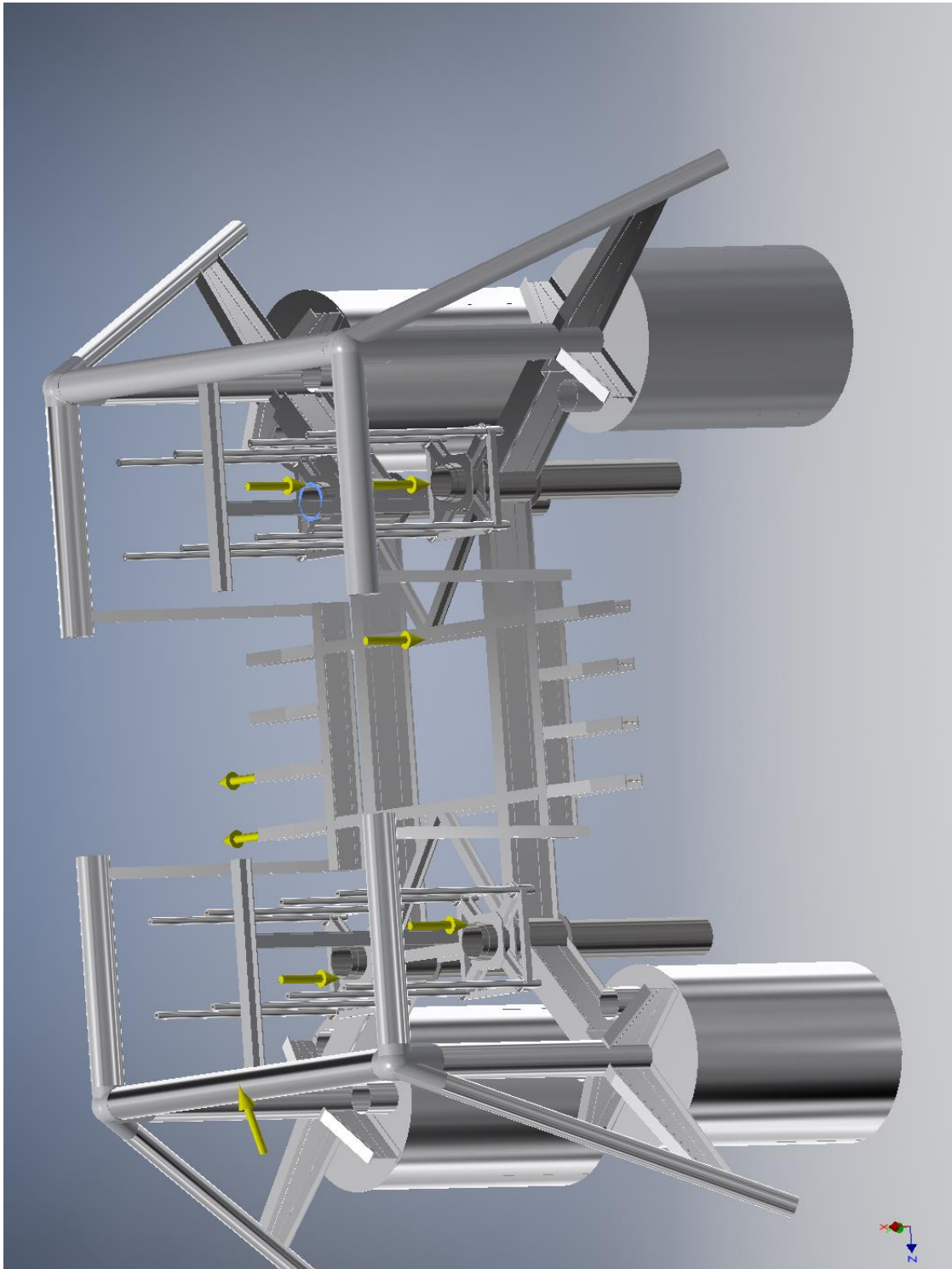
☐ **Selected Face(s)**



☐ **Force:5**

Load Type	Force
Magnitude	600000.000 N
Vector X	-600000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

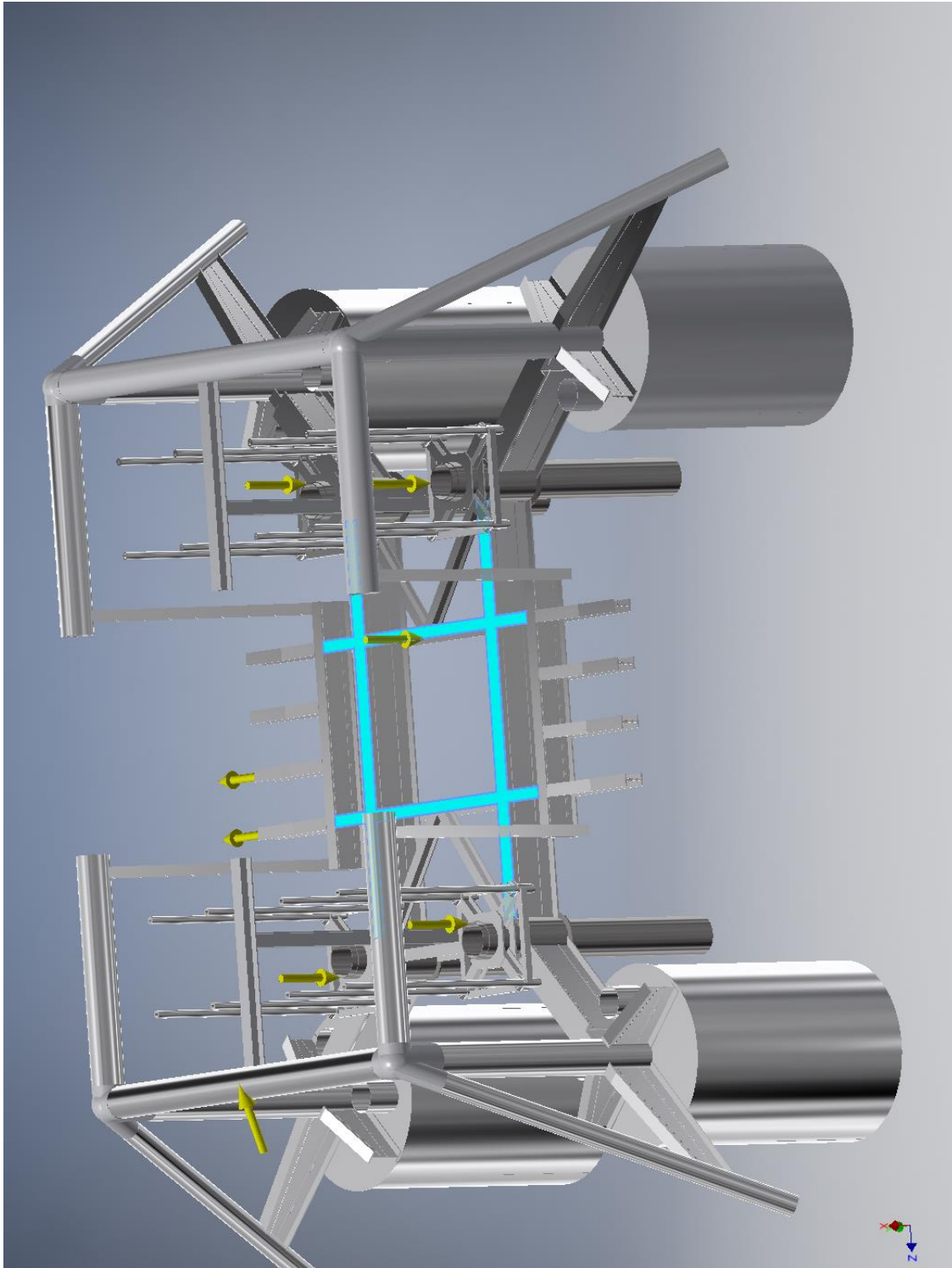
☐ **Selected Face(s)**



☐ **Force:6**

Load Type	Force
Magnitude	800000.000 N
Vector X	-800000.000 N
Vector Y	-0.000 N
Vector Z	-0.000 N

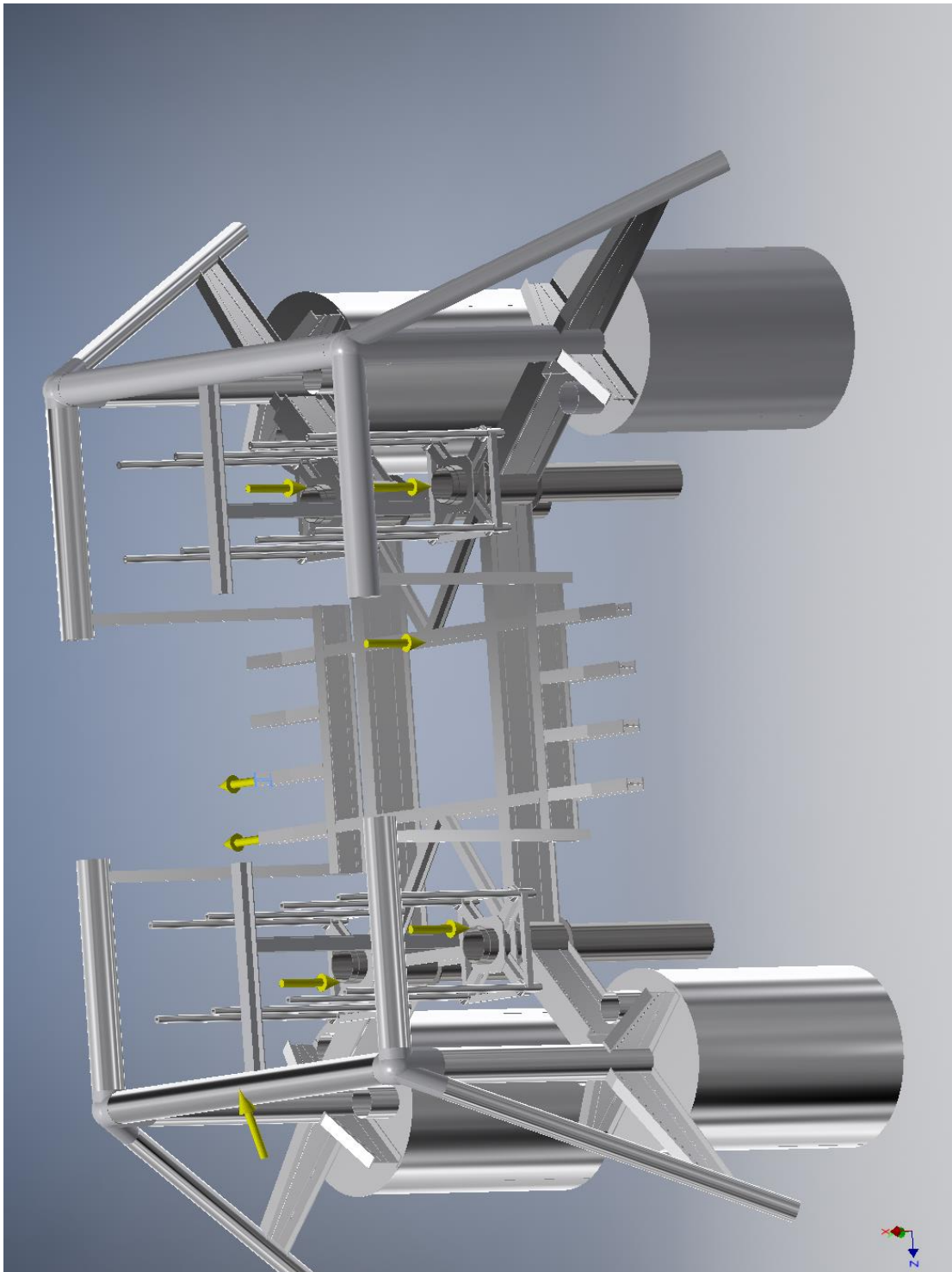
☐ **Selected Face(s)**



☐ **Force:7**

Load Type	Force
Magnitude	300000.000 N
Vector X	-0.000 N
Vector Y	300000.000 N
Vector Z	-0.000 N

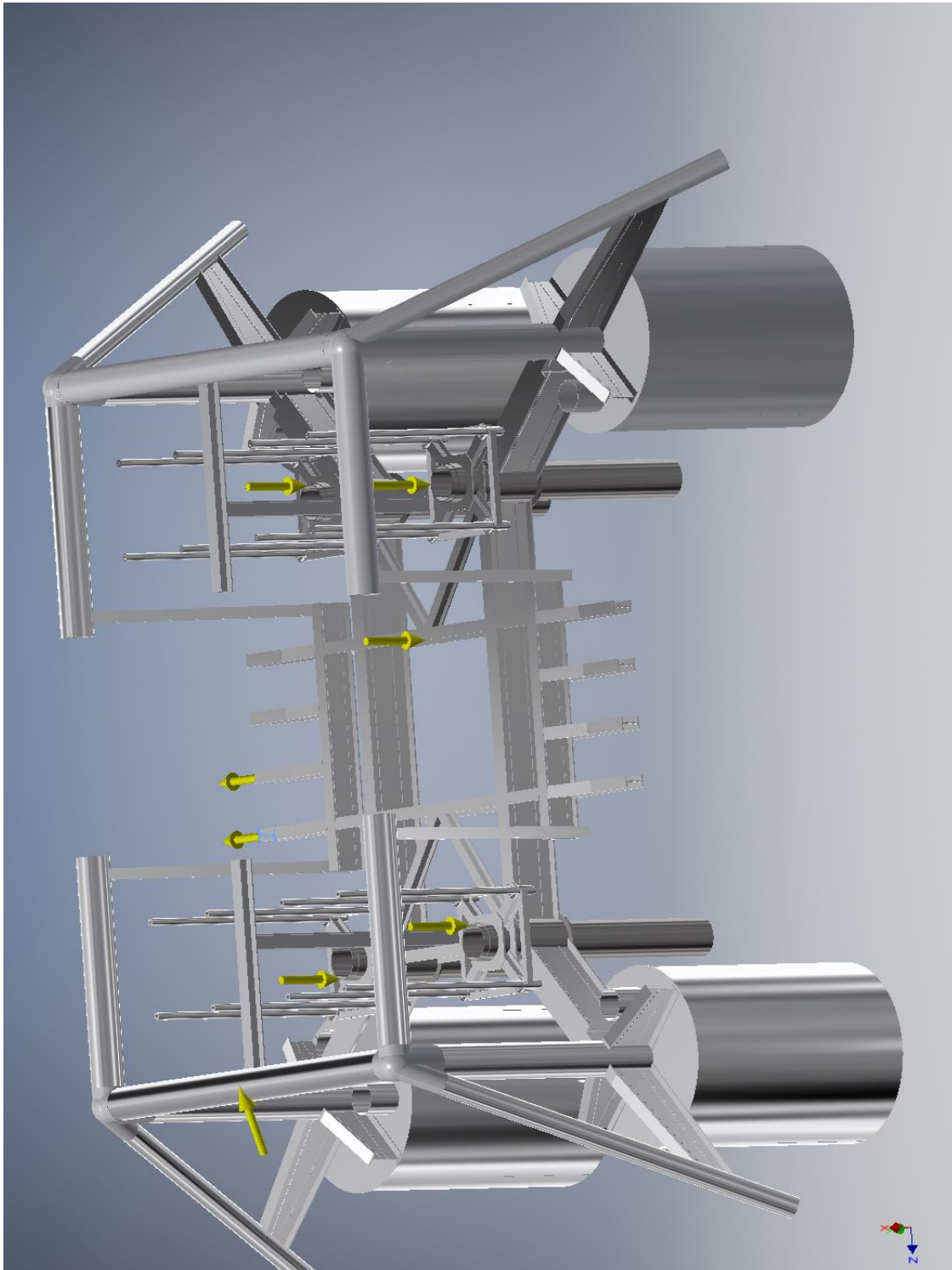
☐ **Selected Face(s)**



☐ **Force:8**

Load Type	Force
Magnitude	300000.000 N
Vector X	0.000 N
Vector Y	300000.000 N
Vector Z	-0.000 N

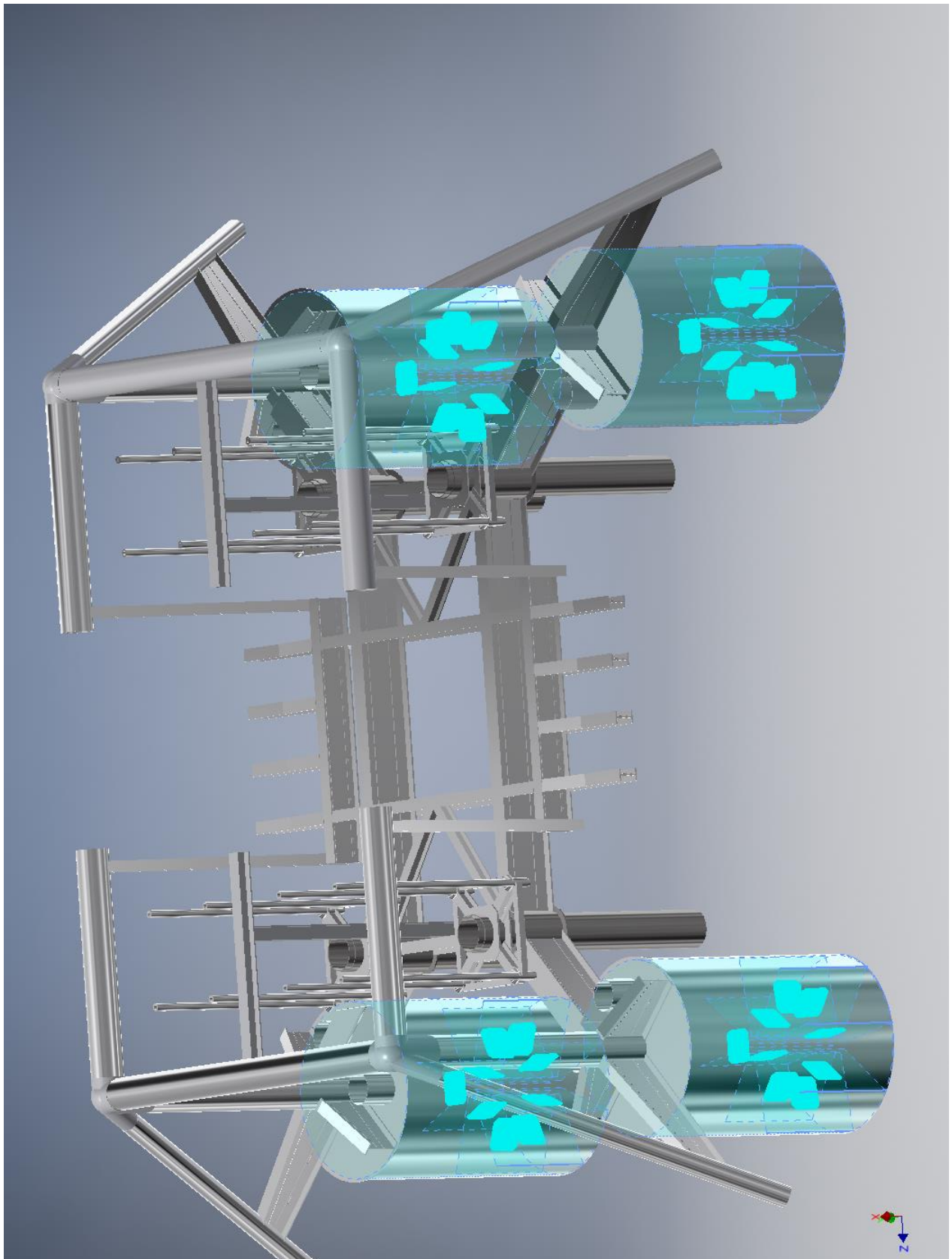
☐ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type Fixed Constraint

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

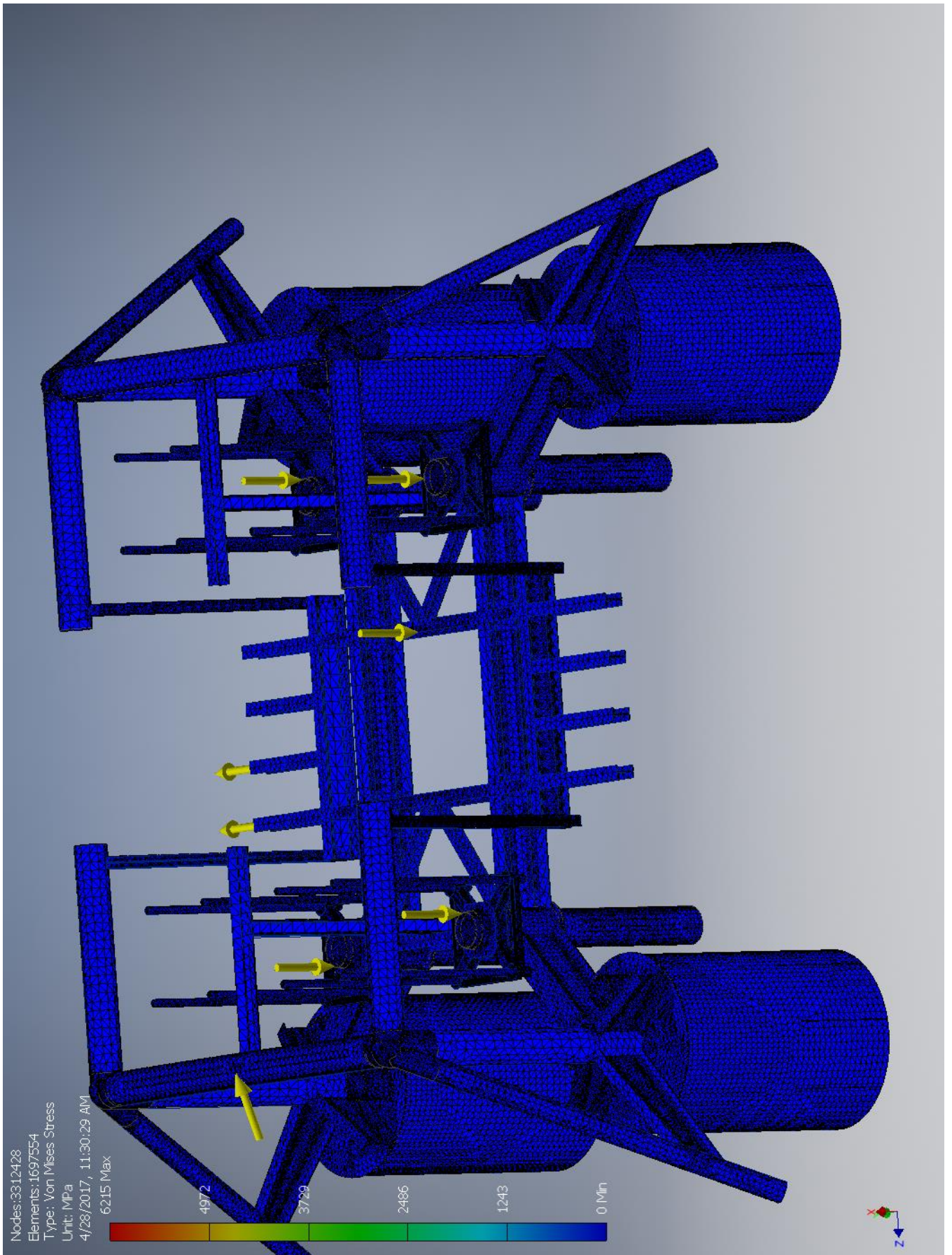
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	3067750 N	2857980 N	15457700 N m	1025160 N m
		-600000 N		-15100700 N m
		939693 N		-3140050 N m

☐ Result Summary

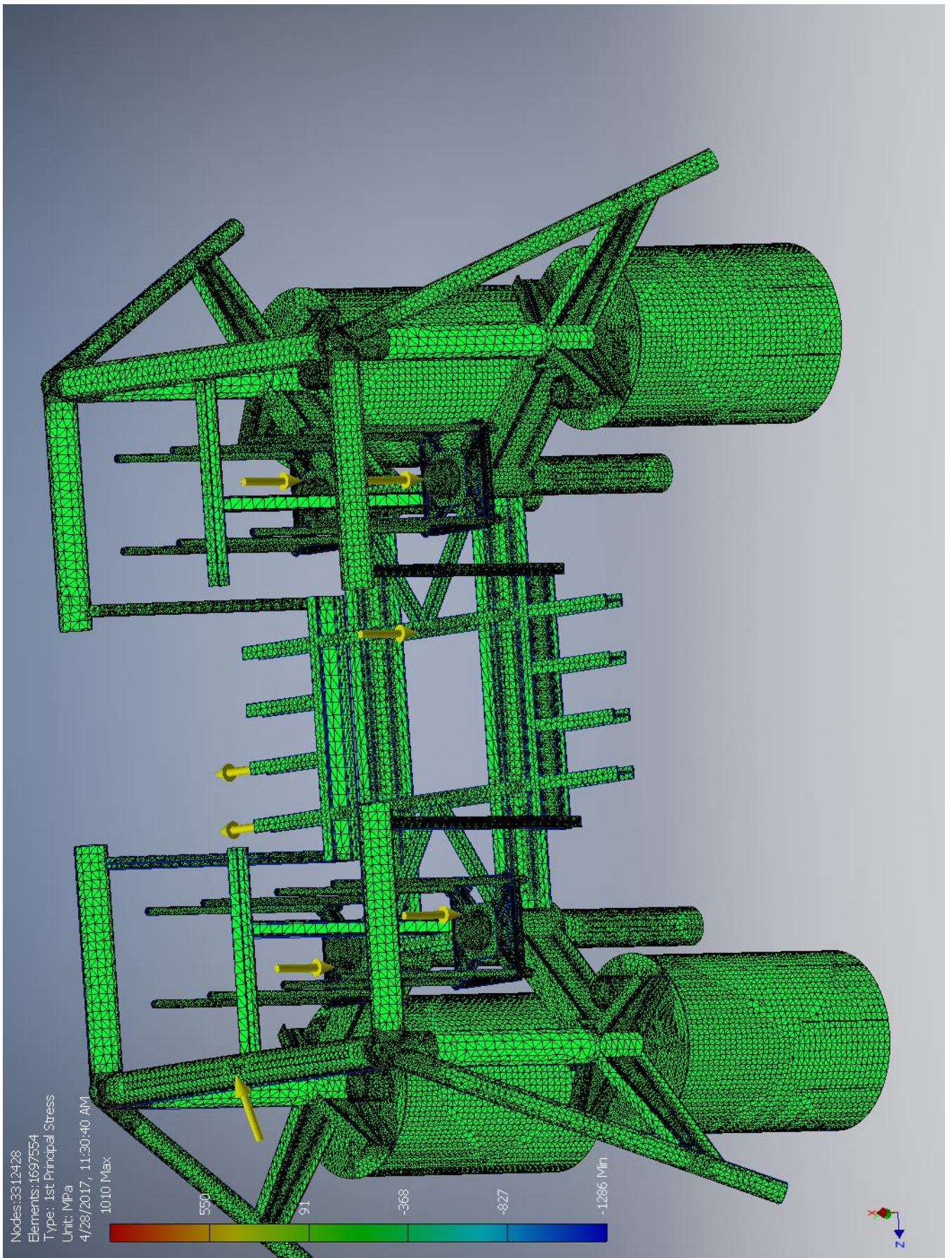
Name	Minimum	Maximum
Volume	6.02475E+10 mm ³	
Mass	181171 kg	
Von Mises Stress	0 MPa	6214.6 MPa
1st Principal Stress	-1286.42 MPa	1009.56 MPa
3rd Principal Stress	-7647.63 MPa	145.504 MPa
Displacement	0 mm	65.9224 mm
Safety Factor	0.0442506 ul	15 ul

☐ Figures

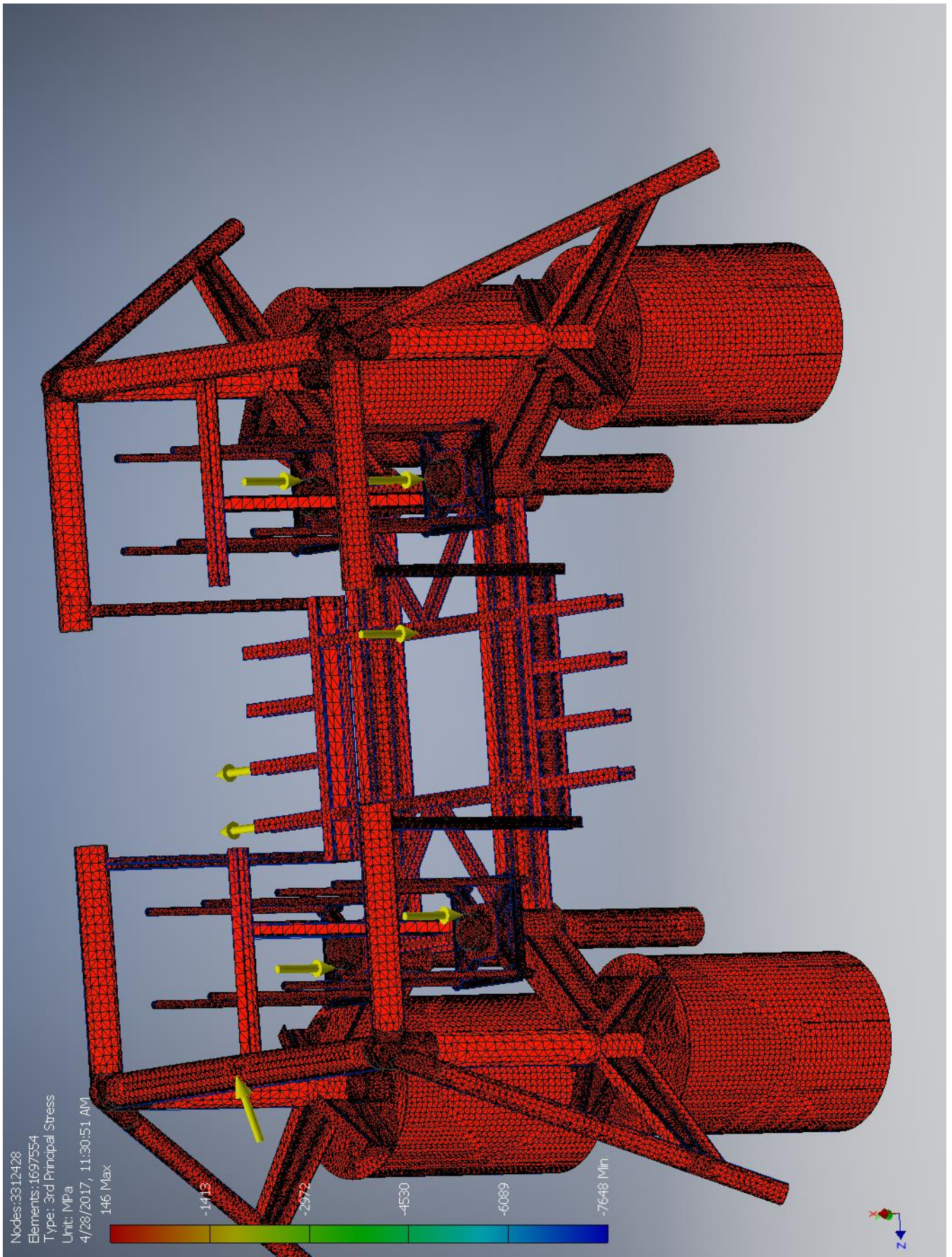
☐ Von Mises Stress



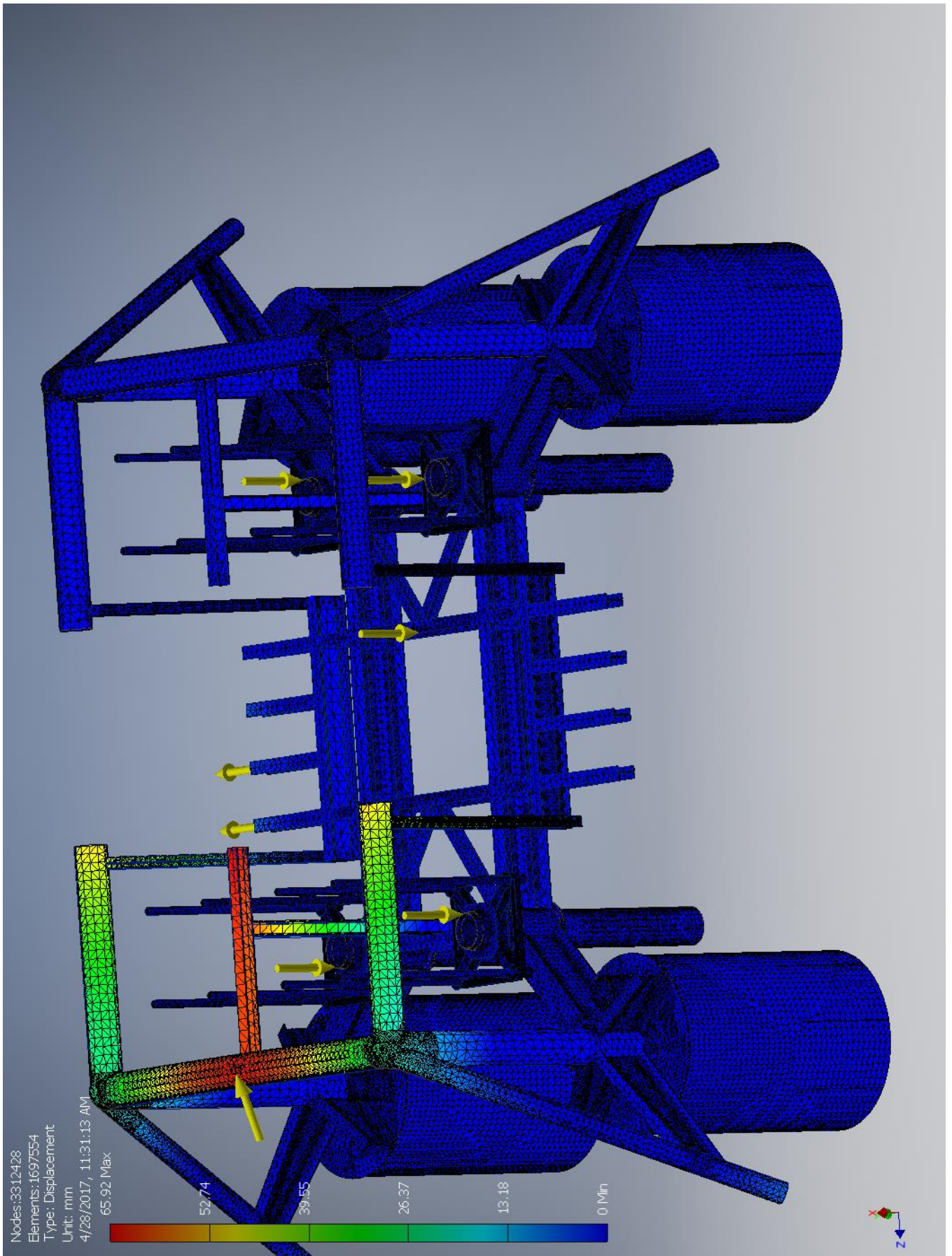
1st Principal Stress



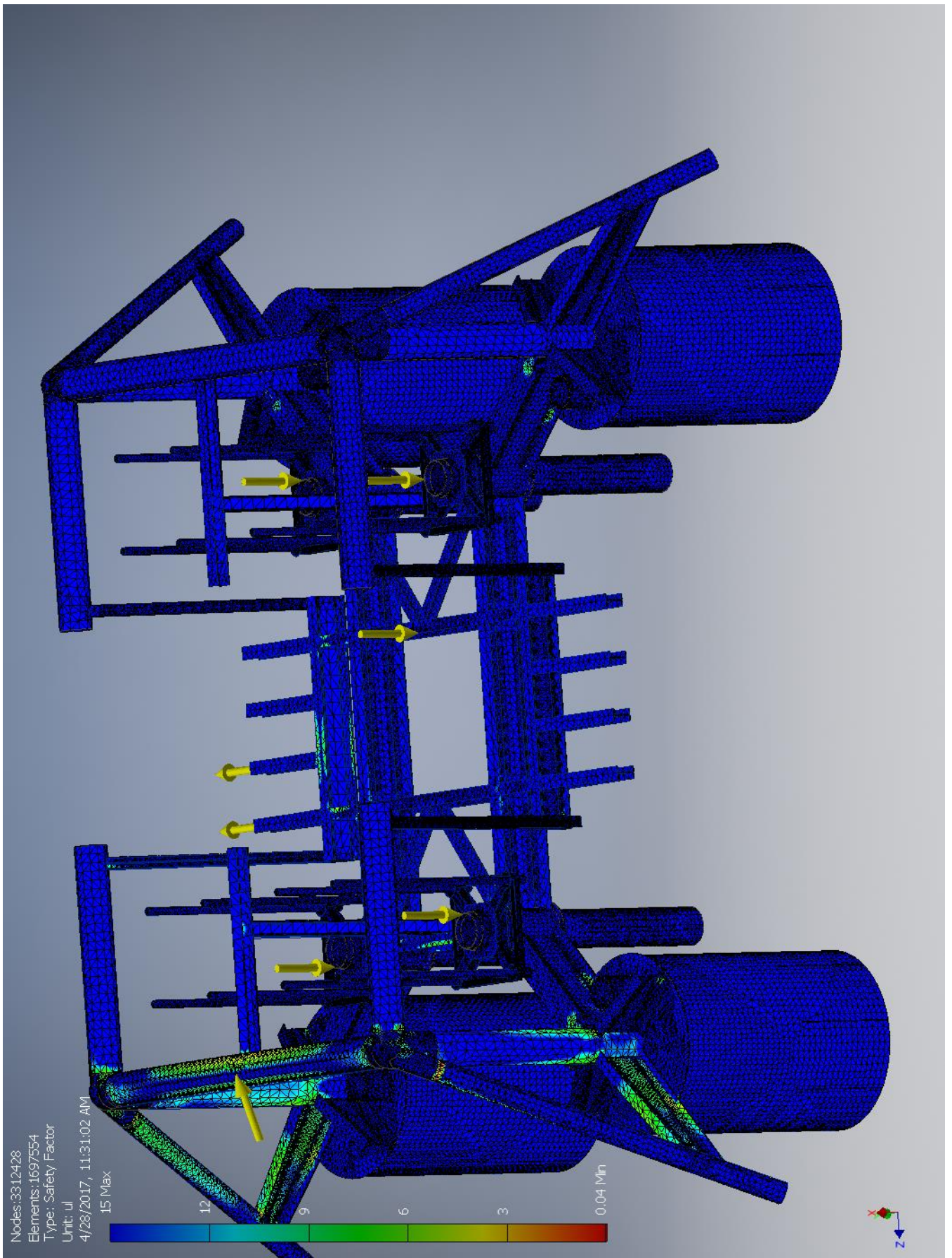
☐ 3rd Principal Stress



☐ Displacement



☐ Safety Factor



I.7 Case-G

Stress Analysis Report



Analyzed File:	Assembly.iam
Autodesk Inventor Version:	2017 (Build 210142000, 142)
Creation Date:	4/28/2017, 4:19 PM
Study Author:	henrikwn
Summary:	

☐ Project Info (iProperties)

☐ Summary

Title	
Author	henrikwn

☐ Project

Designer	henrikwn
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☐ Physical

Mass	182182 kg
Area	3.5408E+09 mm ²
Volume	6.06219E+10 mm ³
Center of Gravity	x=-8278.87 mm y=6548.2 mm z=9857.83 mm

Note: Physical values could be different from Physical values used by FEA reported below.

☐ Case-G

General objective and settings:

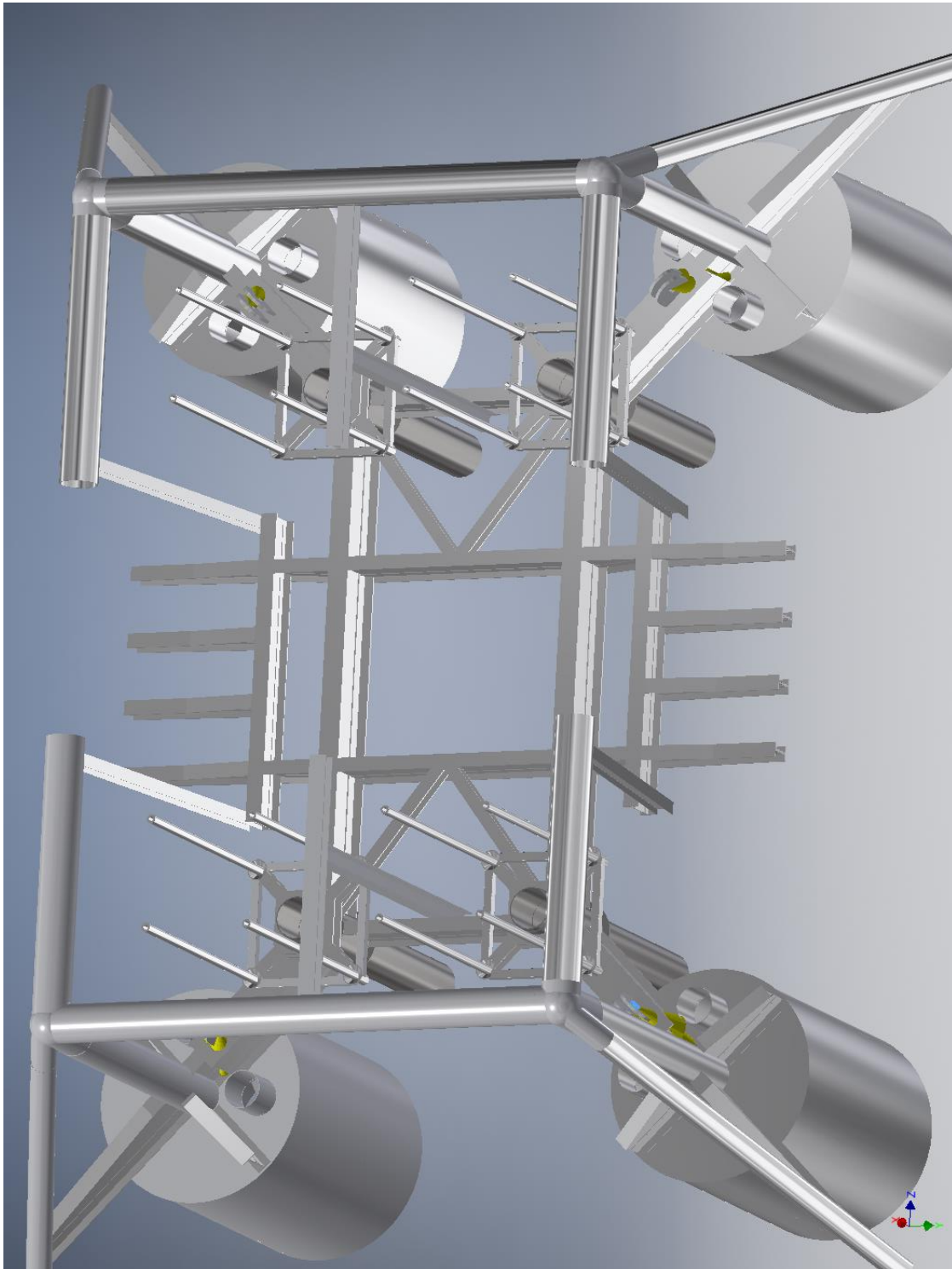
Design Objective	Single Point
Study Type	Static Analysis
Last Modification Date	4/28/2017, 3:30 PM
Detect and Eliminate Rigid Body Modes	No
Separate Stresses Across Contact Surfaces	No
Motion Loads Analysis	No

☐ Operating conditions

▣ **Bearing Load:1**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	-117346.000 N
Vector Z	117346.000 N

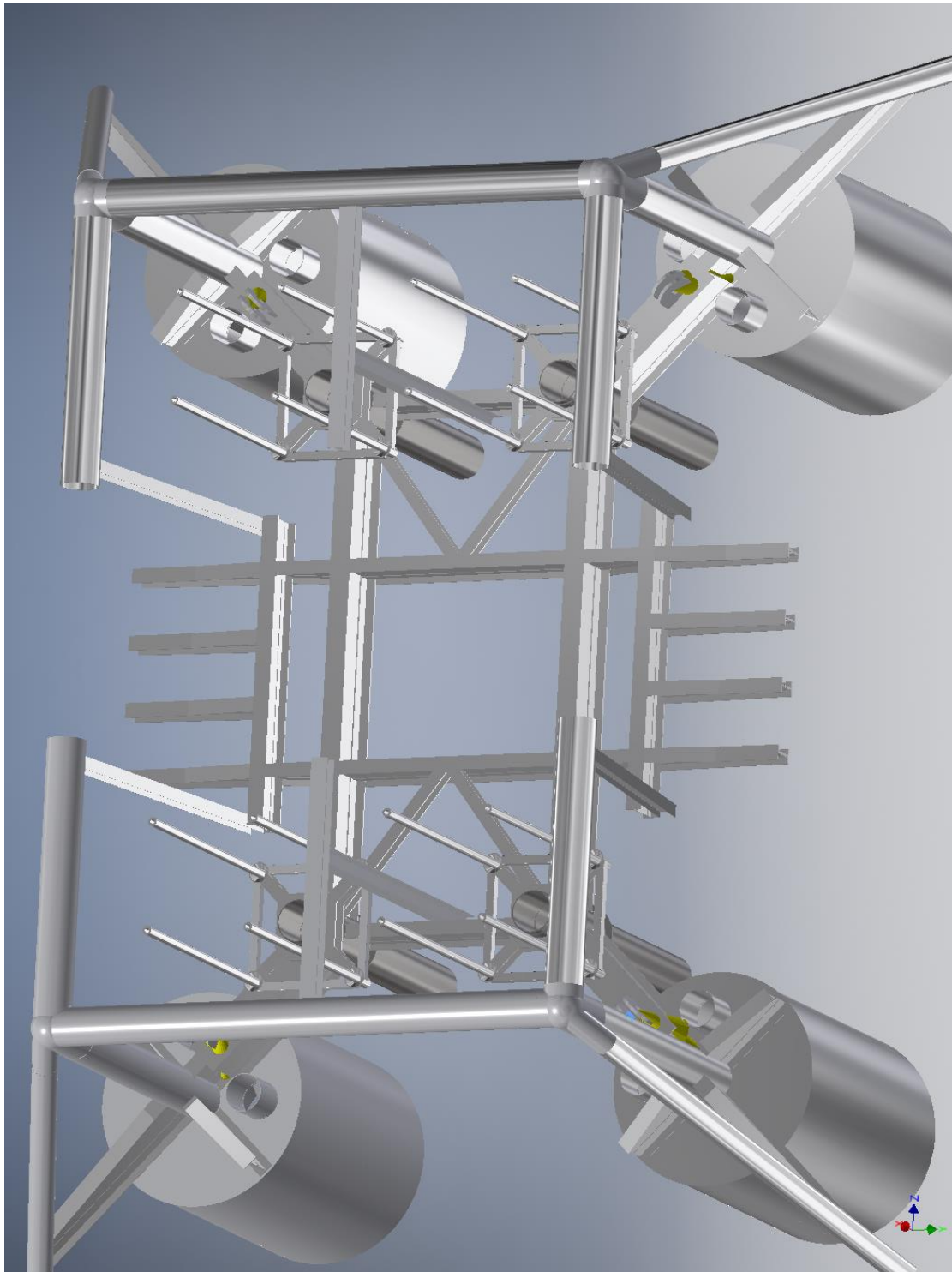
▣ **Selected Face(s)**



▣ **Bearing Load:2**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	-117346.000 N
Vector Z	117346.000 N

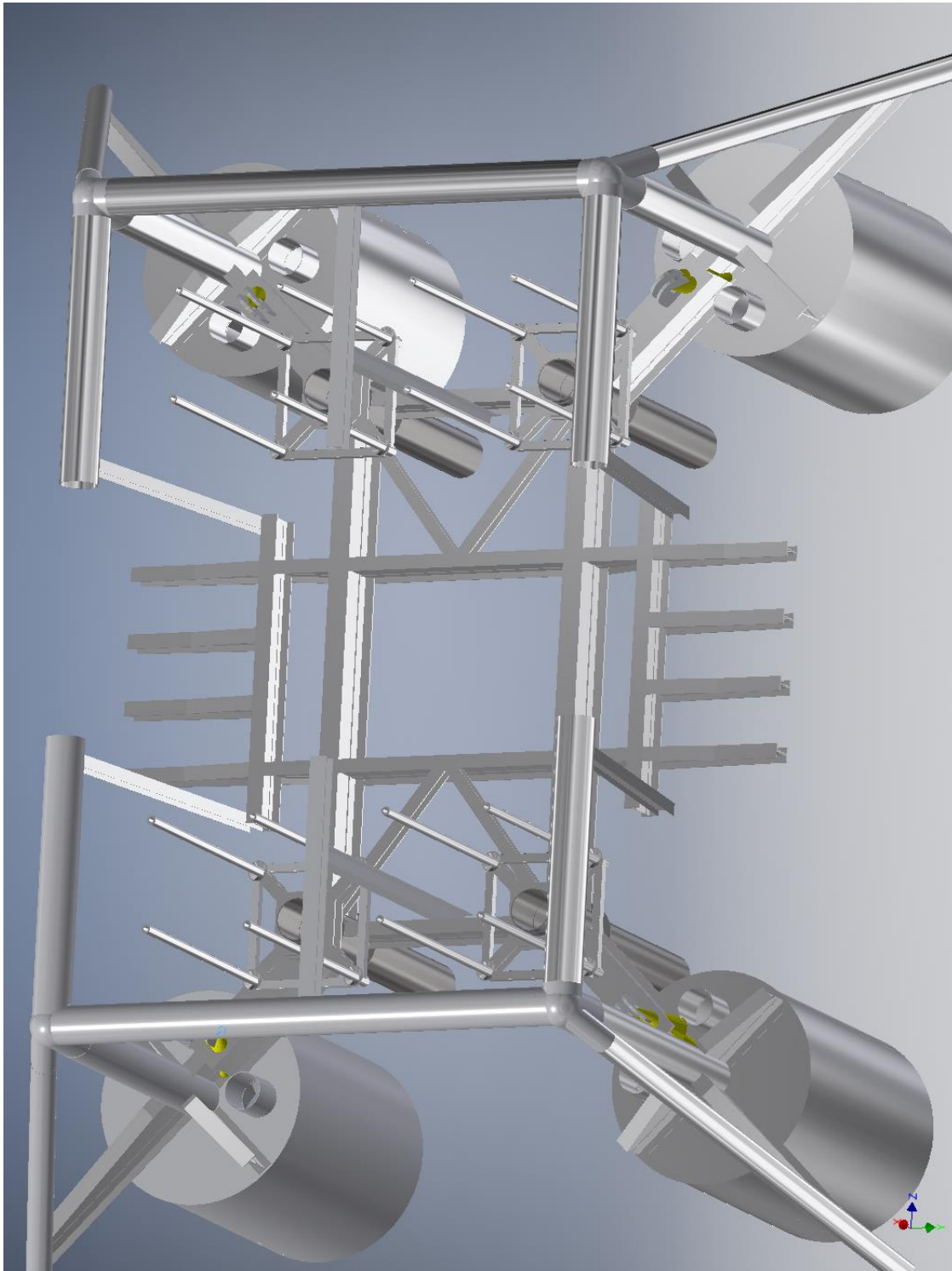
▣ **Selected Face(s)**



▣ **Bearing Load:3**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	117346.000 N
Vector Z	117346.000 N

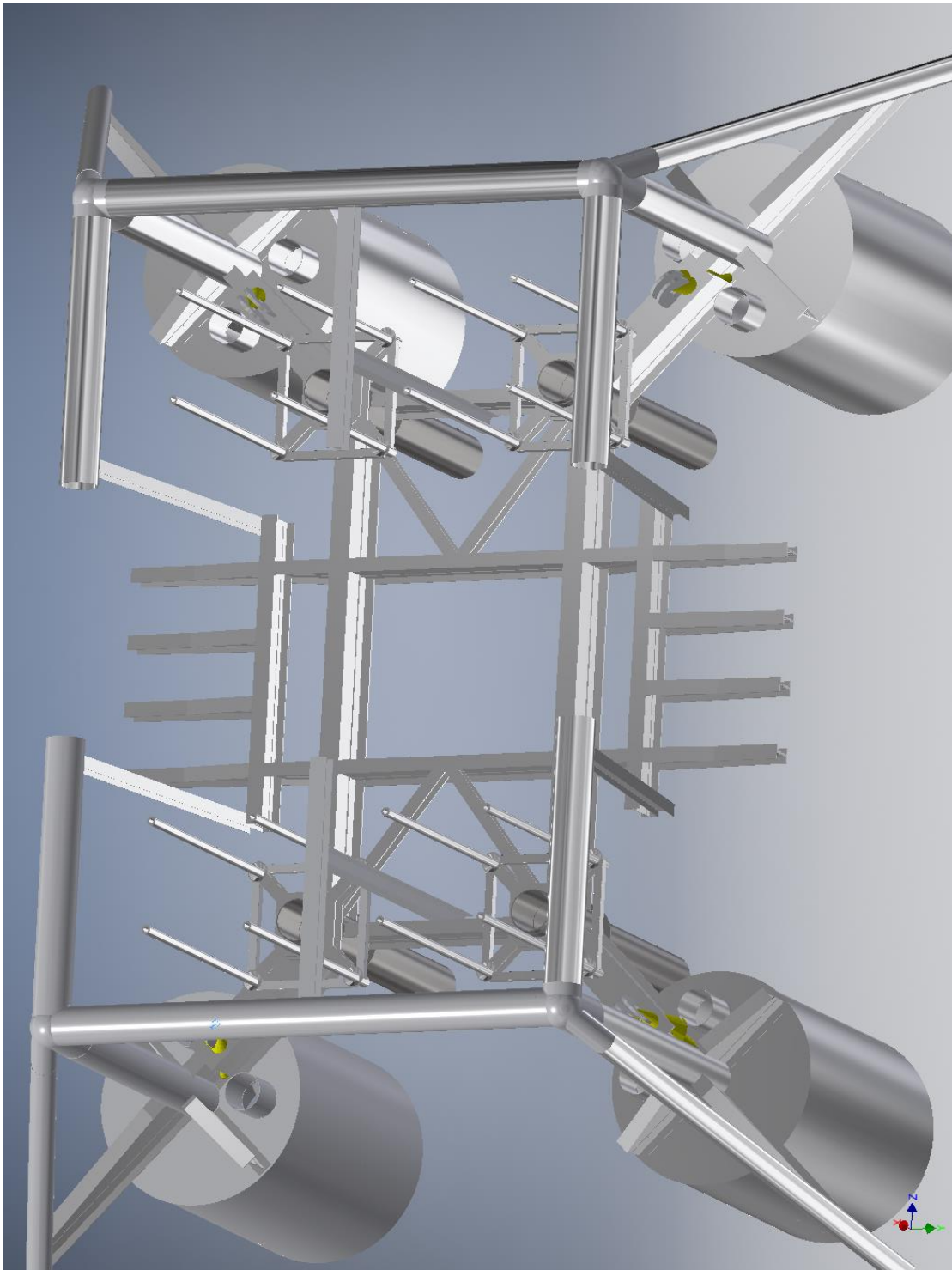
▣ **Selected Face(s)**



▣ **Bearing Load:4**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	117346.000 N
Vector Z	117346.000 N

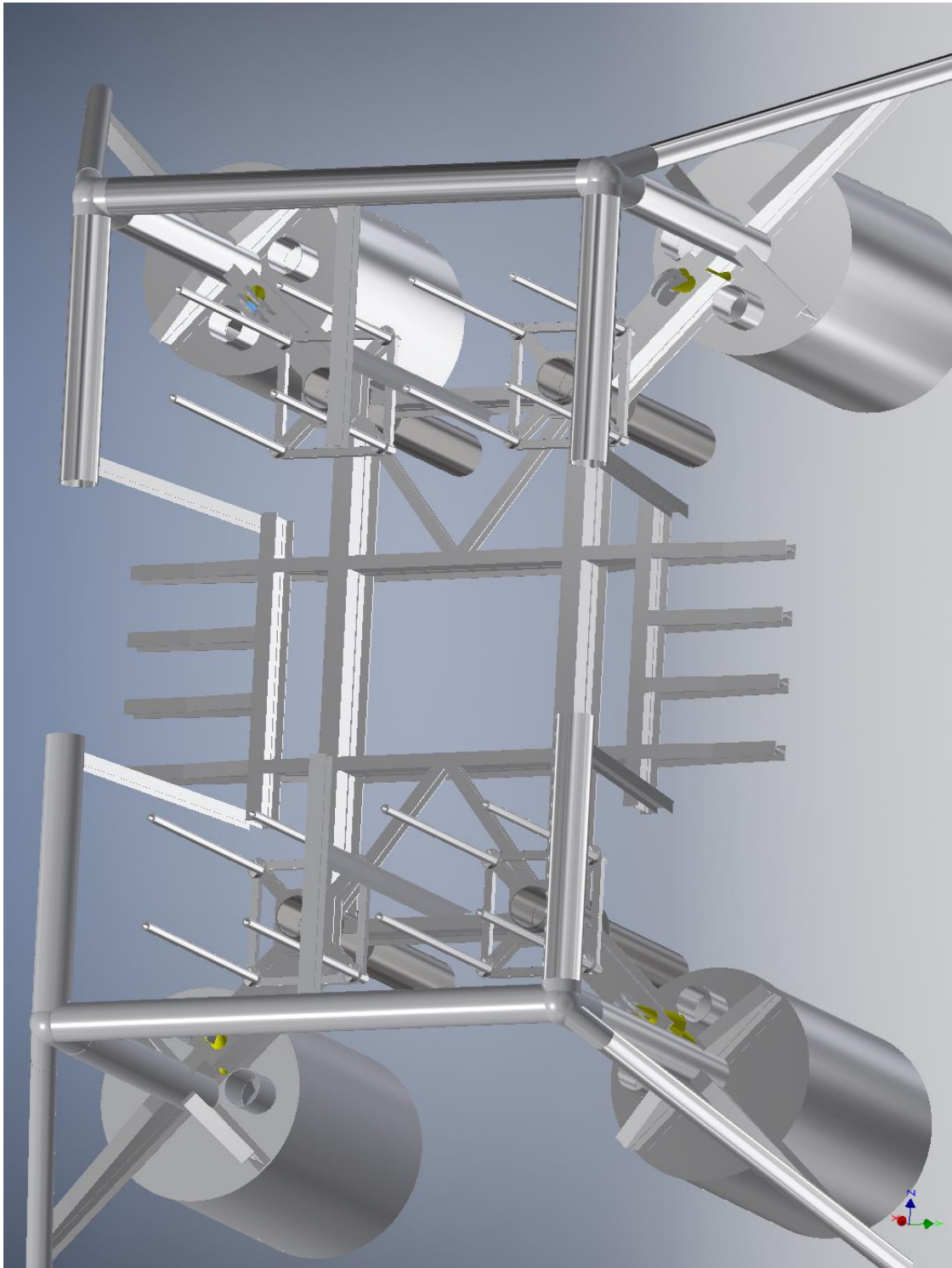
▣ **Selected Face(s)**



▣ **Bearing Load:5**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	117346.000 N
Vector Z	-117346.000 N

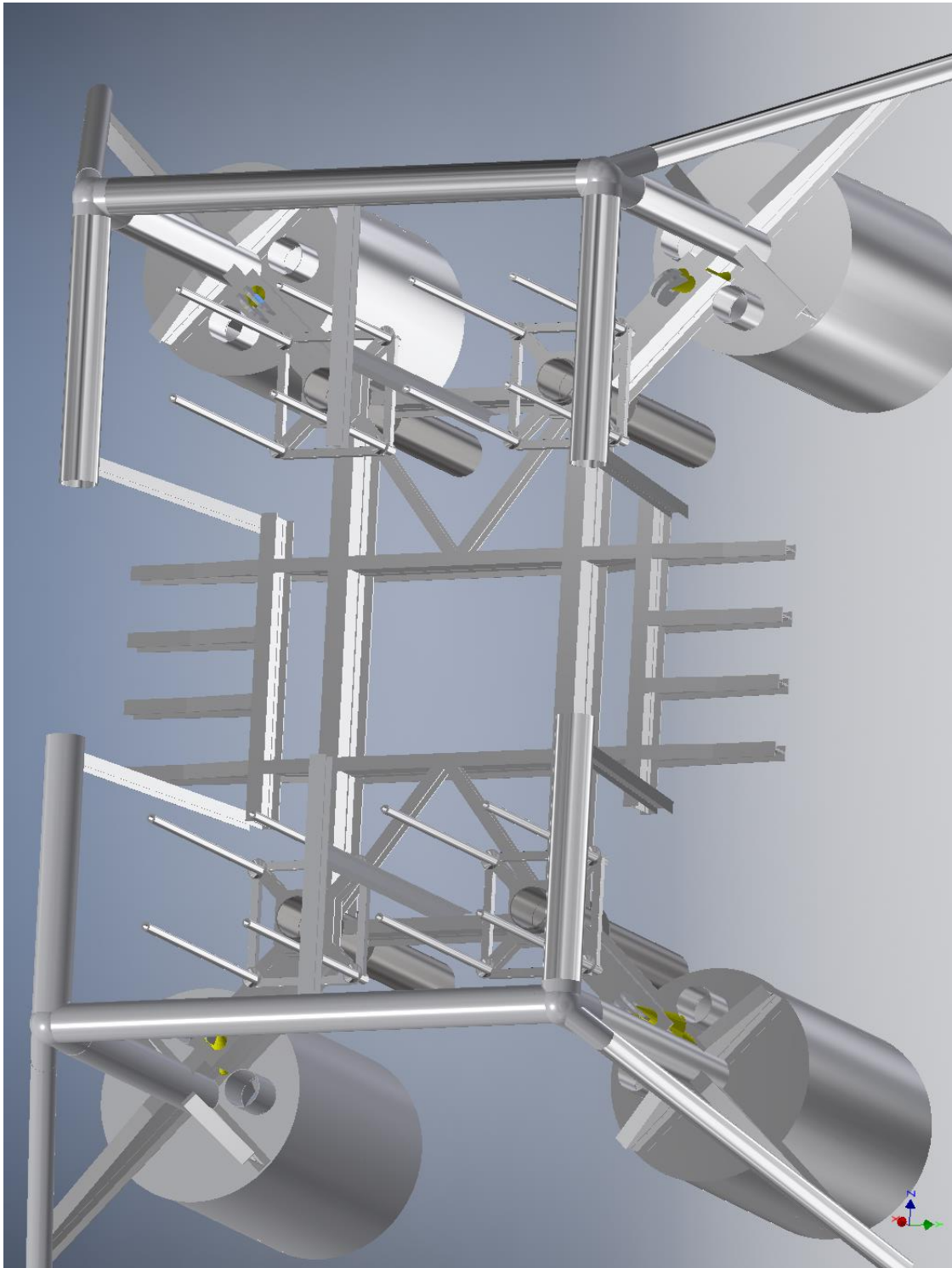
▣ **Selected Face(s)**



▣ **Bearing Load:6**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	117346.000 N
Vector Z	-117346.000 N

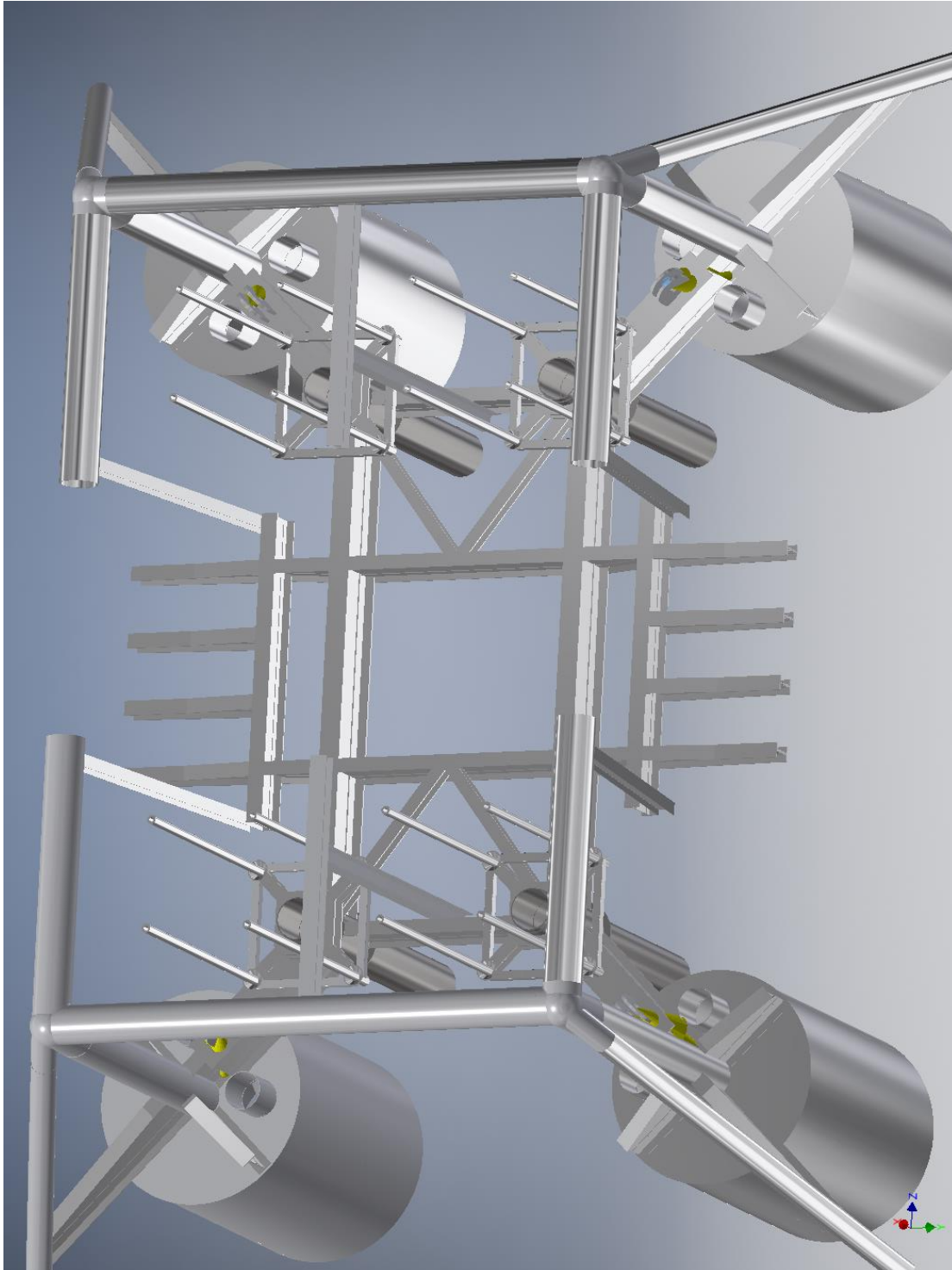
▣ **Selected Face(s)**



▣ **Bearing Load:7**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	-117346.000 N
Vector Z	-117346.000 N

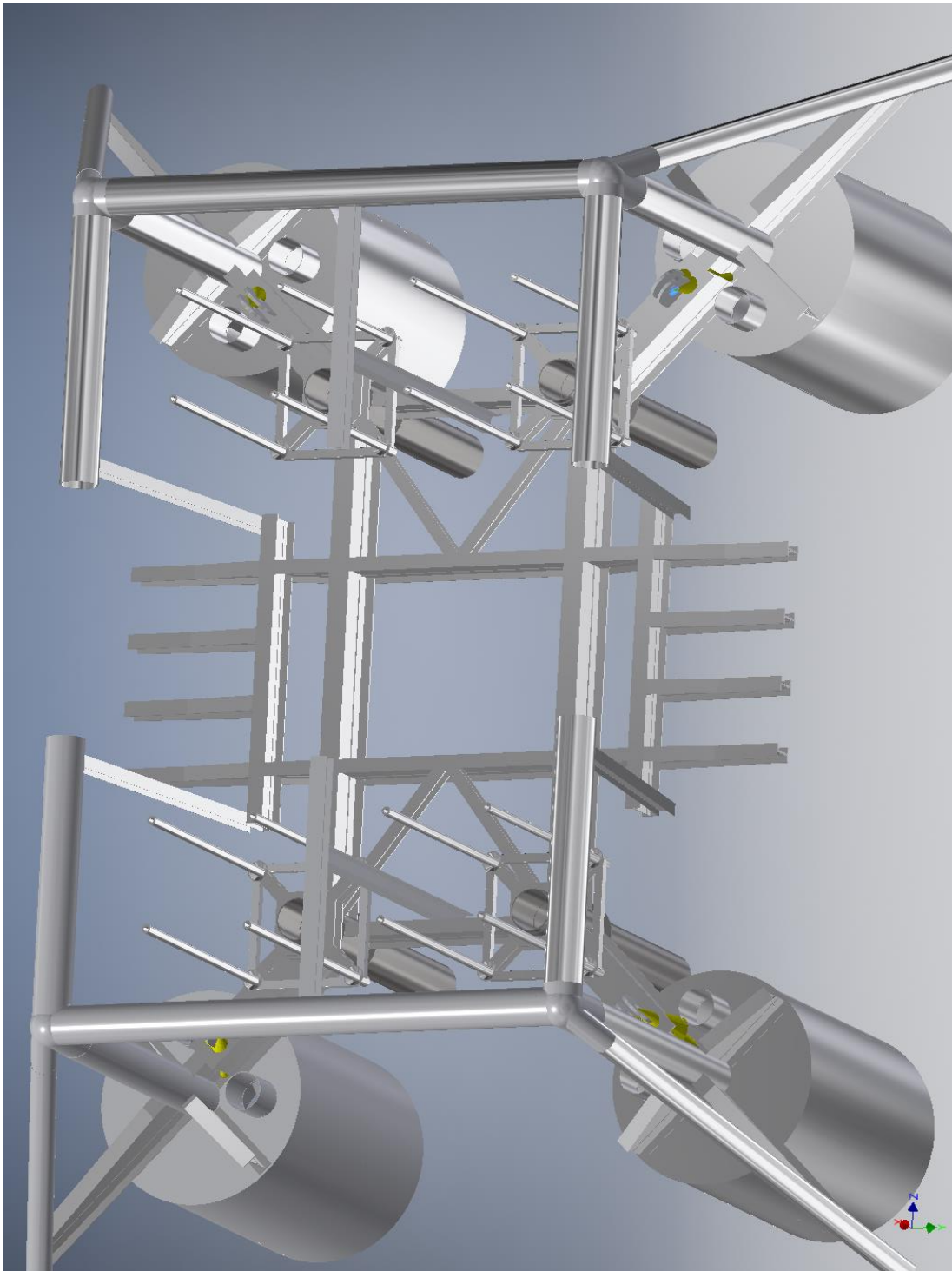
▣ **Selected Face(s)**



▣ **Bearing Load:8**

Load Type	Bearing Load
Magnitude	371080.984 N
Vector X	331905.000 N
Vector Y	-117346.000 N
Vector Z	-117346.000 N

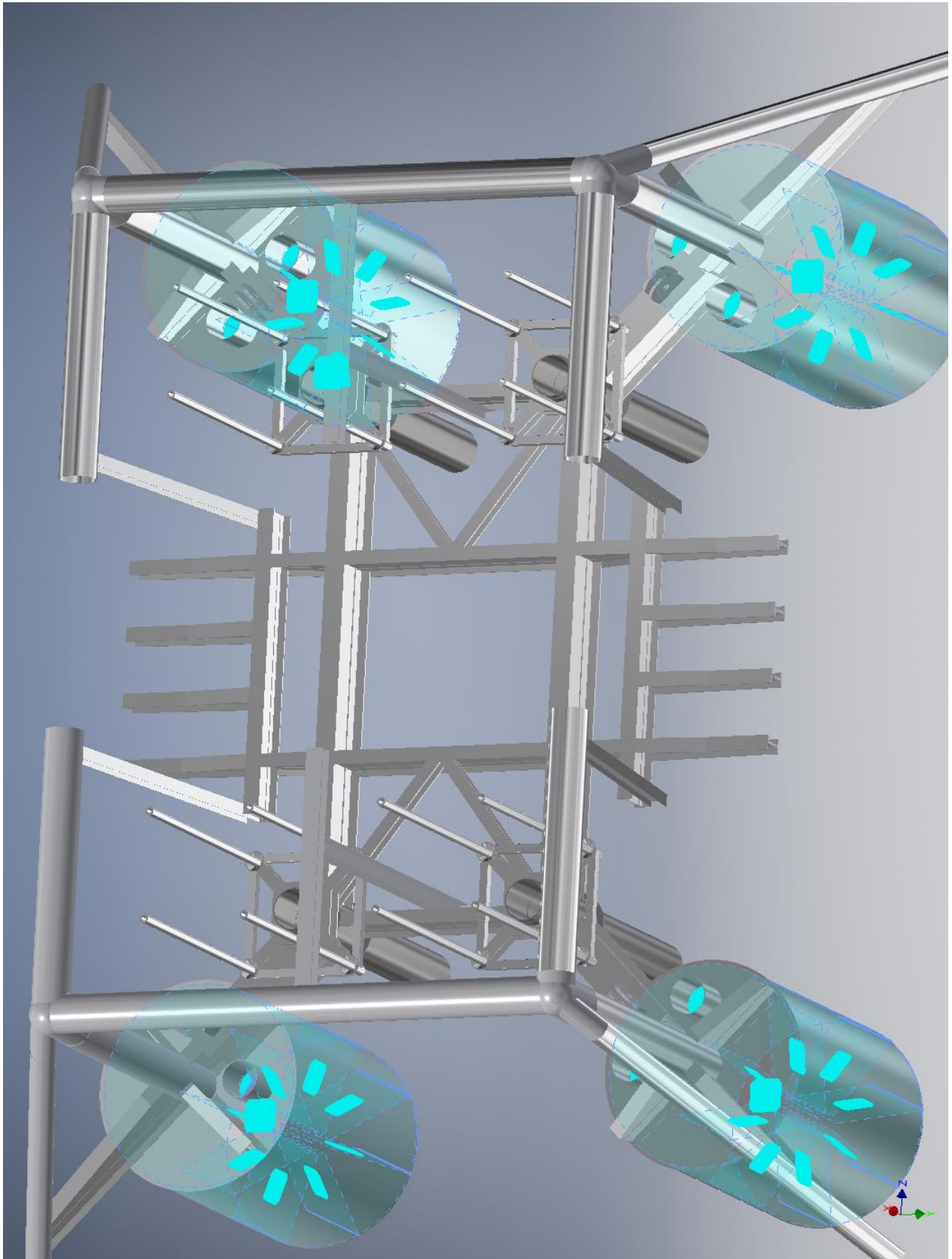
▣ **Selected Face(s)**



☐ **Fixed Constraint:1**

Constraint Type Fixed Constraint

☐ **Selected Face(s)**



☐ Results

☐ Reaction Force and Moment on Constraints

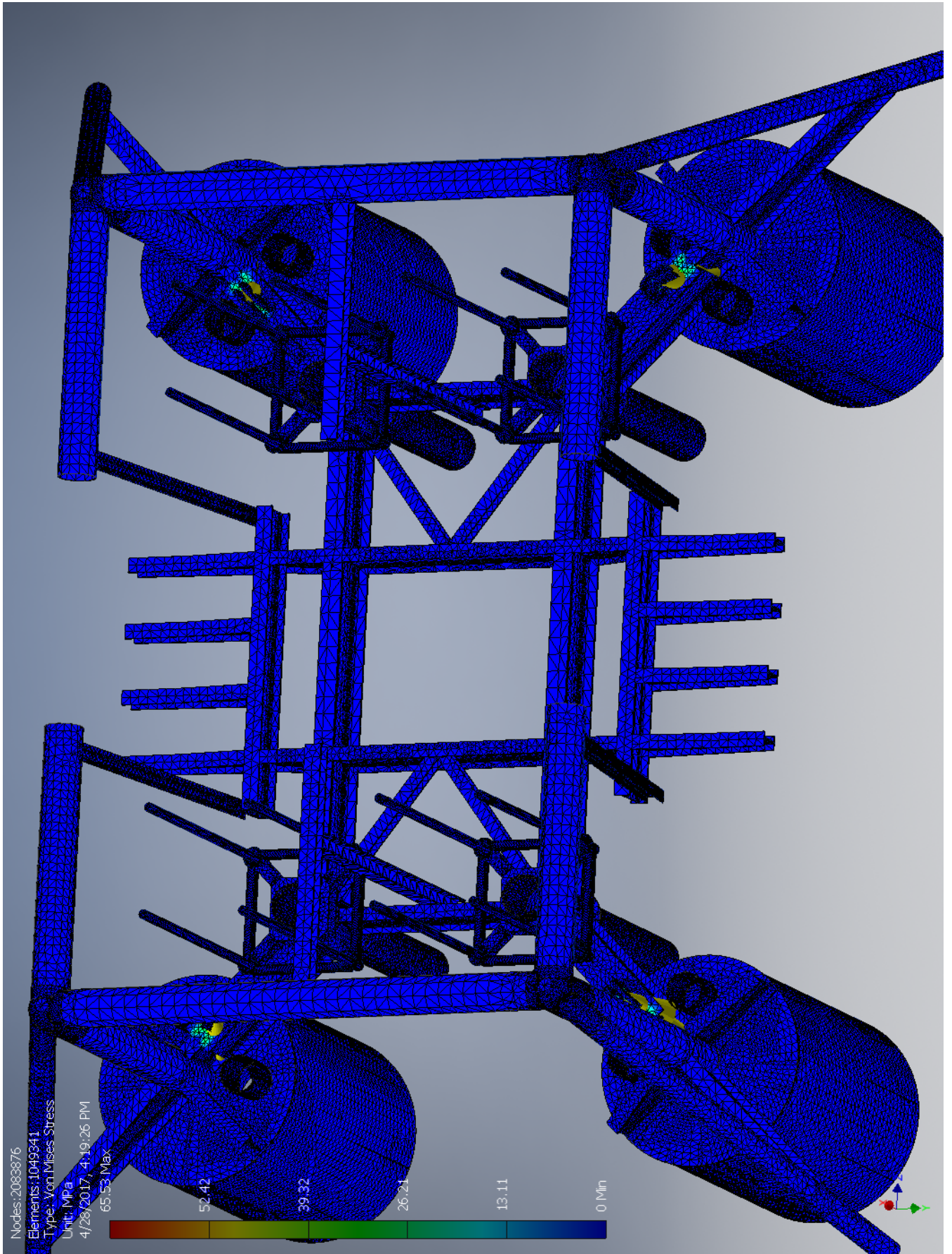
Constraint Name	Reaction Force		Reaction Moment	
	Magnitude	Component (X,Y,Z)	Magnitude	Component (X,Y,Z)
Fixed Constraint:1	2655240 N	-2655240 N	0 N m	0 N m
		0 N		0 N m
		0 N		0 N m

☐ Result Summary

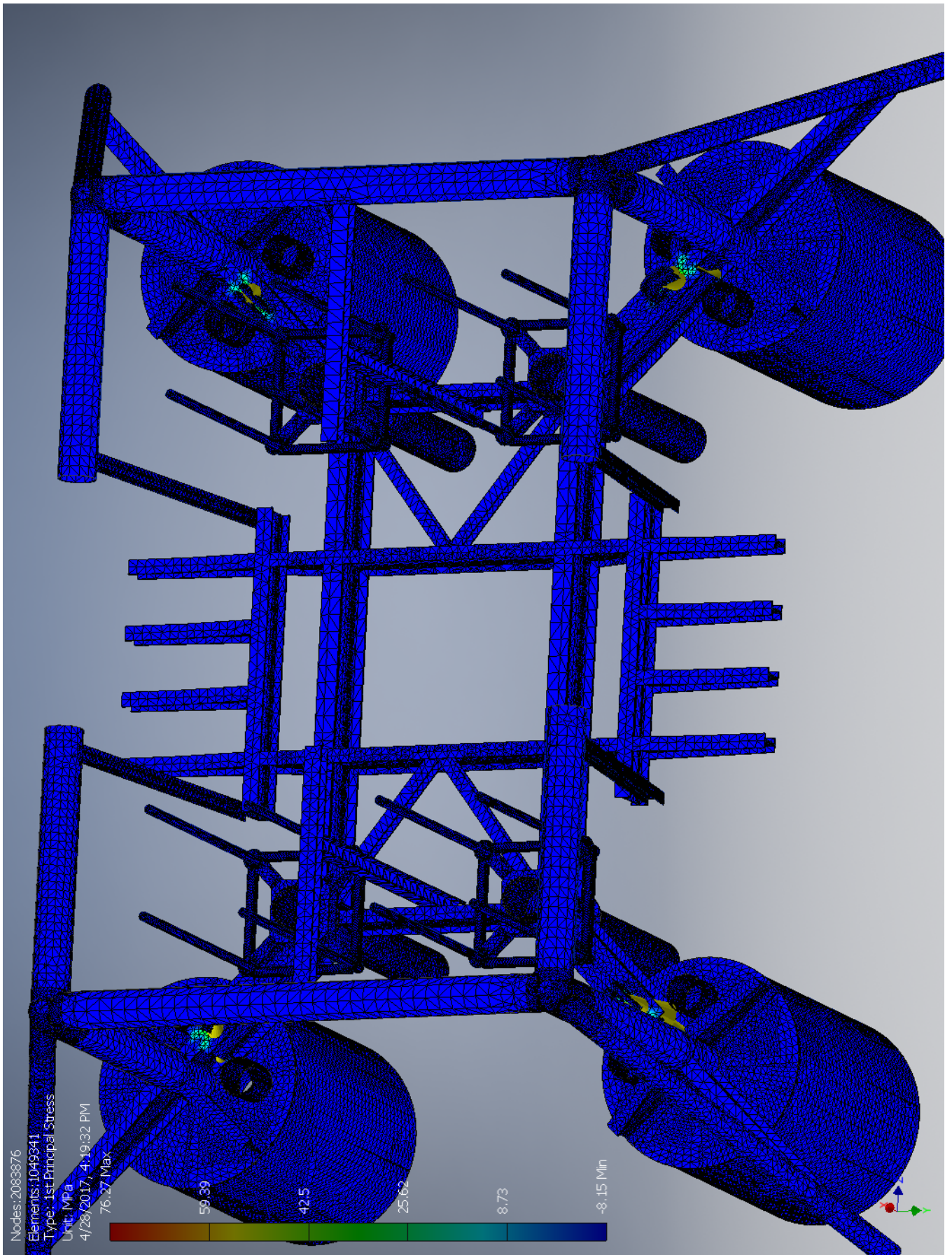
Name	Minimum	Maximum
Volume	6.06219E+10 mm ³	
Mass	182182 kg	
Von Mises Stress	0 MPa	65.5253 MPa
1st Principal Stress	-8.15124 MPa	76.2729 MPa
3rd Principal Stress	-57.1178 MPa	22.3187 MPa
Displacement	0 mm	2.51266 mm
Safety Factor	4.19686 ul	15 ul

☐ Figures

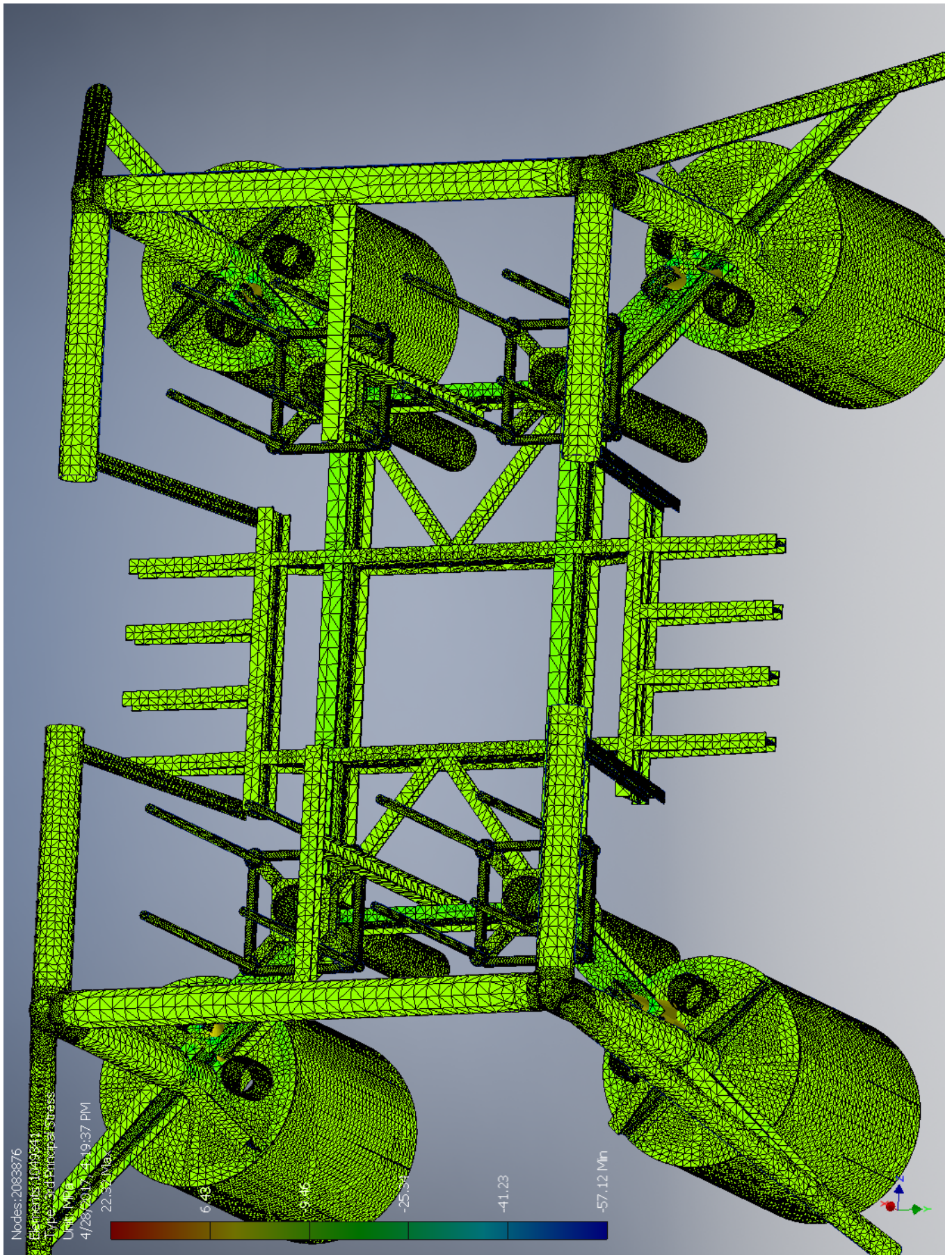
☐ Von Mises Stress



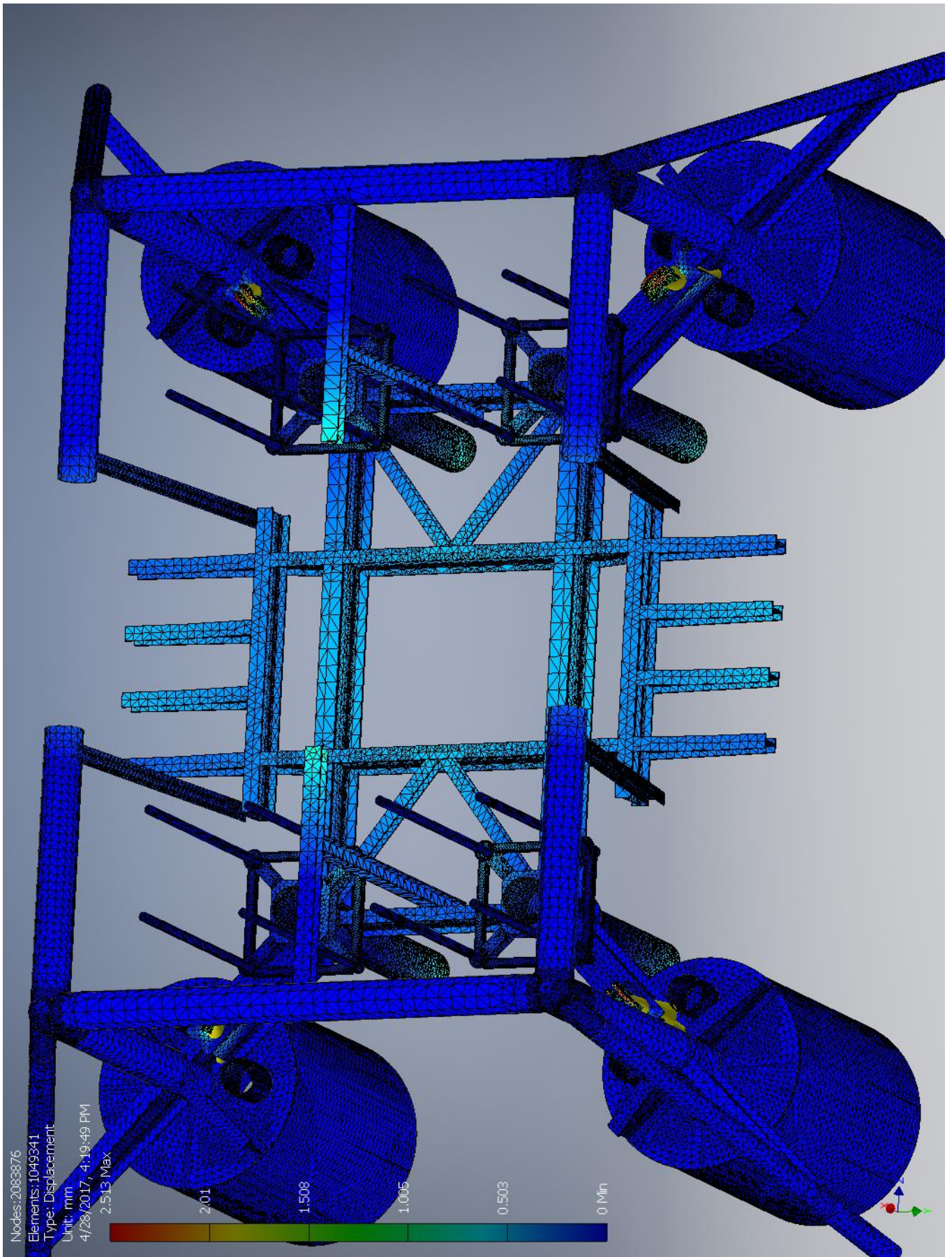
☐ 1st Principal Stress



☐ 3rd Principal Stress



☐ Displacement



☐ Safety Factor

