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1 **Flow over a single dimple recessed in a flat plate**

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6 Direct numerical simulations have been conducted to investigate a zero-pressure-gradient
 7 boundary layer flow over a single shallow dimple. Here the dimple depth to dimple diam-
 8 eter ratio (d/D) as well as the Reynolds number (based on D and free-stream velocity) are
 9 fixed at 0.05 and 20000, respectively. The effect of inlet boundary layer thickness δ on a
 10 given dimple is investigated by considering $\delta/D \in [0.023, 0.1]$. The flow within the dimple
 11 exhibits either a horseshoe vortex (a continuous core line through the two spirals within
 12 the dimple) or a tornado-like vortex pair (discontinuous core line). For the given param-
 13 eter range, four different flow patterns have been identified within the single dimple; *i*) a
 14 steady symmetric horseshoe vortex pattern for $\delta/D \in [0.053, 0.1]$; *ii*) a steady asymmet-
 15 ric horseshoe vortex pattern for $\delta/D = 0.04$; *iii*) a quasi-periodic asymmetric horseshoe
 16 vortex pattern for $\delta/D = 0.033$; *iv*) a mixed horseshoe and tornado-like vortex pattern
 17 for $\delta/D = 0.023$. The growth of the streamwise vorticity, mainly caused by the tilting of
 18 the vertical vorticity, plays a key role in the transition between the different flow patterns.
 19 Dimple-induced velocity streaks above the single dimple have been investigated in detail
 20 for the first time, showing four different streaks; *i*) a High-speed streak above the dimple;
 21 *ii*) two Side-low-speed streaks located outside the dimple span; *iii*) two Side-high-speed
 22 streaks and *iv*) a Mid-low-speed streak in between them. These are mainly caused by a
 23 flow acceleration effect and a flow diffuser effect over the dimple, as well as a 'lift-up'
 24 mechanism within the downstream part of the dimple, tilting the boundary layer upwards.

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25 I. INTRODUCTION

26 Dimple geometry as a passive flow control strategy has been widely investigated both numeri-
 27 cally and experimentally due to its practical applications such as, e.g., enhancing the heat-transfer
 28 rate^{1,2}, improving the aerodynamic performance of a wind turbine blade³⁻⁵ and airfoils⁶ as a vor-
 29 tex generator, trailing-edge noise reduction on the airfoil⁷, as well as its potential application in
 30 drag reduction⁸⁻¹⁰. Despite these applications and the previous research associated with them, the
 31 basic physical mechanisms of flow over dimples are yet to be fully understood. The purpose of
 32 the present paper is to fill in some of this knowledge gap by investigating the detailed laminar flow
 33 over one single dimple using direct numerical simulations, which allow us to provide a building
 34 stone for understanding the flow over multiple dimples.

35 Previous experimental and numerical studies of flow over a dimpled plate have been conducted
 36 in conjunction with two major flow configurations; *i*) channel flow and *ii*) zero-pressure-gradient
 37 boundary layer flow over a flat plate. Kovalenko *et al.*¹¹ collected and analyzed a substantial
 38 amount of experimental results for channel flow with a single dimple for $Re_c \in [500, 100000]$ and
 39 for the dimple depth d to dimple diameter D ratio $d/D \in [0.1, 0.5]$, where Re_c denotes the Reynolds
 40 number based on D and the centerline velocity U_c . Figure 1 shows the sketch of the dimple geom-
 41 etry. These experimental data were provided from previous works published by (among others)
 42 a range of scientists in the former Soviet Union, and later Russia and Ukraine (see Kovalenko *et*
 43 *al.*¹¹ and the references therein for the detailed publications). More recently detailed flow mea-
 44 surements for laminar incoming boundary layer flow over a single dimple recessed in a flat plate
 45 were conducted by Tay *et al.*¹² using dye flow visualization. Here Re_D ranges from 1000 to 28000
 46 while d/D ranges from 0.05 to 0.5 where Re_D denotes the Reynolds number based on D and the
 47 freestream velocity U . Two side-by-side dimples, one with a round edge and the other with a sharp
 48 edge, were used in their experiments with the distance between them being large enough so that
 49 the flow over each dimple was independent of the flow over the other dimple. Tay *et al.*¹² reported
 50 six qualitatively different flow patterns, of which four were previously observed by Kovalenko *et*
 51 *al.*¹¹.

52 Figure 2 shows a sketch of these six flow patterns. Here flow pattern *I* (figure 2*a*) is charac-
 53 terized by the streamlines first bending towards the streamwise centerline of the dimple (indicated
 54 by the red dotted dash line) and then bending away from it; flow pattern *II* (figure 2*b*) is similar
 55 to *I* except for the existence of a small flow recirculation region within the upstream part of the

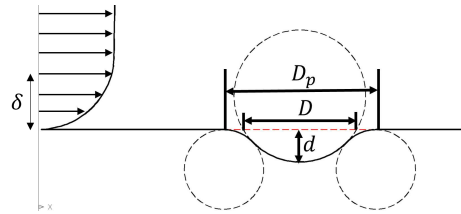
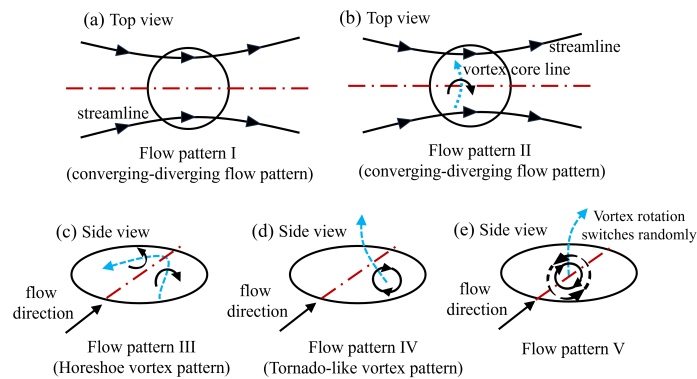


FIG. 1. Sketch of the dimple geometry and the inlet boundary layer flow


 FIG. 2. Sketch of the flow patterns found in the measurements of Tay *et al.*¹². The red dotted dash line denotes the streamwise centerline of the dimple; the blue dash line represents the vortex core line while the black arrow line around the vortex core line indicates the vortex rotation direction.

56 dimple as visualized by the vortex core line (indicated by the blue dashed arrow line where the
 57 surrounding black arrow line indicates the vortex rotation direction). These two flow patterns are
 58 similar to the 'diffuser-confuser' flow pattern classified by Kovalenko *et al.*¹¹. More recently, this
 59 flow pattern has also been referred to as the 'converging-diverging' flow pattern¹⁰.

60 Flow pattern III (figure 2c) consists of a counter-rotating vortex pair (with rotation indicated
 61 by the black arrows) with a continuous vortex core line (depicted by the blue dashed arrow line),
 62 which is symmetric about the streamwise centerline of the dimple. This flow pattern was pre-
 63 viously classified as the horseshoe vortex pattern by Kovalenko *et al.*¹¹. This flow pattern was
 64 also observed in Large eddy simulations (LES) conducted by Lan, Xie, and Zhang¹³ for turbulent
 65 boundary layer flow over a single dimple under an adverse pressure gradient, as well as in the

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66 Reynolds-averaged Navier-Stokes (RANS) simulations of Isaev *et al.*¹⁴ who investigated steady-
 67 state turbulent channel flow over a single dimple. Flow pattern *IV* is characterized by that one
 68 of the vortices in the counter-rotating vortex pair increases while the other shrinks, thus forming
 69 an asymmetric flow pattern with one dominating vortex as illustrated in figure 2(*d*). The core
 70 line of this dominating vortex, which is displaced from the streamwise centerline of the dimple
 71 (see figure 2*d*), is directed upwards from the dimple bottom with a slight inclination relative to
 72 the vertical axis, which remains stable once it is established. Here the location of the dominating
 73 vortex can appear on either side of the streamwise centerline through the dimple with opposite
 74 rotation direction, for repeated flow realizations. This flow pattern is qualitatively similar to the
 75 tornado-like vortex pattern identified by Kovalenko *et al.*¹¹. Qualitatively similar tornado-like
 76 vortex structures were also observed in unsteady RANS and LES simulations of Turnow *et al.*¹⁵
 77 for turbulent channel flow with a single dimple. They found that the unsteady RANS simulations
 78 can only capture the mean tornado-like vortex structure with an orientation of approximately 45°
 79 inclined to the streamwise direction while LES results showed that the vortex core switched ori-
 80 entation approximately $\pm 45^\circ$ inclined to the streamwise direction. They also reported that the
 81 mean turbulent flow within the dimple, obtained by averaging the streamlines over a long time
 82 interval, was qualitatively similar to the symmetric horseshoe vortex pattern. It should be noted
 83 here that none of these works have visualized the core line of the shrinking vortex. Flow pattern *V*
 84 (figure 2*e*) was only found in the deepest dimple for $d/D = 0.5$. Here only one vortex is observed
 85 within the dimple. The vortex core line near the dimple surface is vertical while higher up from
 86 the dimple surface, the core line is bent downstream (see figure 2*e*). Moreover, the vortex may
 87 switch its rotational direction randomly as illustrated in figure 2(*e*) by the full and dashed black
 88 arrow lines surrounding the vortex core line. In flow pattern *VI*, the flow exhibits a transition to
 89 turbulent/chaotic states (not shown here).

90 Until now, a substantial number of measurements and numerical simulations have been con-
 91 ducted for flow over a single dimple, resulting in the qualitative description of different flow pat-
 92 terns, described above. However, neither existing measurements nor existing numerical simu-
 93 lations have so far provided information on the detailed physical mechanisms underpinning the
 94 occurrence and transition between the different observed flow patterns. To fill in a part of this
 95 knowledge gap, direct numerical simulations (DNS) are applied to *i*) provide a more detailed de-
 96 scription of these flow patterns; *ii*) investigate the physical mechanisms underpinning the transition
 97 between these flow patterns; *iii*) investigate the effect of the flat-plate boundary layer thickness on

98 the flow within a given dimple and *iv*) explain the dimple-induced streaks that occur above the
 99 recessed dimple. Specifically, direct numerical simulations are conducted for a laminar zero-
 100 pressure-gradient boundary layer flow over a shallow dimple with $d/D = 0.05$ recessed in a flat
 101 plate for a fixed $Re_D = 20000$; the thickness δ/D of the boundary layer at the inlet ranges from
 102 0.023 to 0.1.

103 The paper is organized as follows. The problem definition, numerical method, computational
 104 domain, and grid convergence study are given in section *II* and the Appendix. The formation
 105 of and transition between the different flow patterns within the dimple as δ/D decreases, are
 106 discussed in detail in section *III*. The dimple-induced velocity streaks above the dimple and their
 107 formation mechanisms are discussed in section *IV*. Conclusions are given in section *V*.

108 II. PROBLEM DEFINITION AND GOVERNING EQUATIONS

109 The current paper addresses a zero-pressure-gradient boundary layer flow over a plate with one
 110 single dimple as shown in figure 1 for $Re_D = 20000$, corresponding to a value within a range where
 111 either the horseshoe vortex or the tornado-like vortex was observed for different dimple depths in
 112 the previous experiments. This allows us to investigate the evolution of these two major vortices
 113 within the dimple. The dimple depth-to-diameter ratio d/D and the round edge ratio D_p/D are
 114 0.05 and 1.06, respectively. Most of the previous work on this flow configuration has focused
 115 on $d/D > 0.1$. The inlet boundary layer flow is defined by the Blasius-like velocity profile (with
 116 zero vertical velocity w) corresponding to the boundary layer thickness δ , defined by the distance
 117 between the wall and the height where the streamwise velocity is 99% of the free-stream velocity.
 118 The values of δ/D range from 0.023 to 0.1, corresponding to the Reynolds number Re_θ based
 119 on the momentum thickness ranging from 280 to 64 as well as the Reynolds number Re_x based
 120 on the streamwise location ranging from 1.8×10^5 to 2.5×10^4 . The choice of δ/D ensures that
 121 the flat-plate boundary layer flow remains laminar since the transition Reynolds number Re_x from
 122 laminar to turbulent boundary layer flow in a flat plate is about 5×10^5 . Here, the incompressible
 123 flow with a constant density ρ and kinematic viscosity ν is governed by the three-dimensional
 124 Navier–Stokes equations given as

$$125 \quad \frac{\partial u_i}{\partial x_i} = 0 \quad (1)$$

$$\frac{\partial u_i}{\partial t} + \frac{\partial u_i u_j}{\partial x_j} = -\frac{1}{\rho} \frac{\partial p}{\partial x_i} + \nu \frac{\partial^2 u_i}{\partial x_j \partial x_j} \quad (2)$$

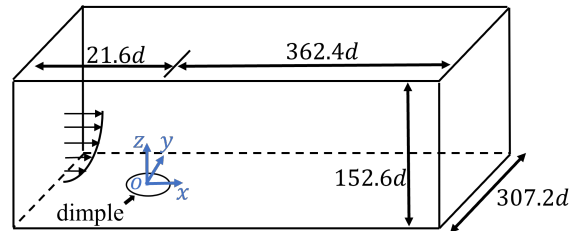


FIG. 3. Sketch of the computational domain

126 where the Einstein notation using repeated indices is applied. Here $u_i = (u, v, w)$ and $x_i = (x, y, z)$ for
 127 $i = 1, 2,$ and $3,$ indicate the velocity and Cartesian coordinates, respectively, while t and p denote
 128 the time and pressure, respectively.

129 A. Numerical methods

130 The DNS/LES solver MGLET^{16,17} utilizing a second-order finite volume method with a stag-
 131 gered grid is used for solving the Navier–Stokes equations. An explicit low-storage third-order
 132 Runge-Kutta scheme is used for time integration. The midpoint rule is used to approximate the
 133 surface integral of flow variables over the faces of the discrete volumes, and the Poisson equation
 134 for pressure correction is solved using Stone’s strongly implicit procedure (SIP). The dimple ge-
 135 ometry is taken into account by a direct-forcing immersed boundary method, which is described
 136 in detail in Peller *et al.*¹⁸. The code has been thoroughly validated and applied to investigate many
 137 complex flows, such as turbulent boundary layer flow over a flat plate¹⁹, turbulent flow in a rod-
 138 roughed channel²⁰, steady and oscillatory flow through a hexagonal sphere pack²¹, the spheroid
 139 wake²², the single-step cylinder wake²³ and the curved cylinder wake²⁴.

140 B. Computational domain and grid resolution

141 In the present work, numerical simulations of the zero-pressure-gradient boundary layer flow
 142 over the dimpled plate have been conducted for $\delta/D = 0.023, 0.04, 0.053, 0.069,$ and 0.1 with
 143 $Re_D = 20000$. It should be noted that for a given Re_D , the decrease of δ/D is equivalent to moving
 144 the dimple towards the front edge of the plate. Figure 3 shows the computational domain. The
 145 inlet and outlet boundaries are located $21.6d$ and $362.4d$ away from the dimple center. The top

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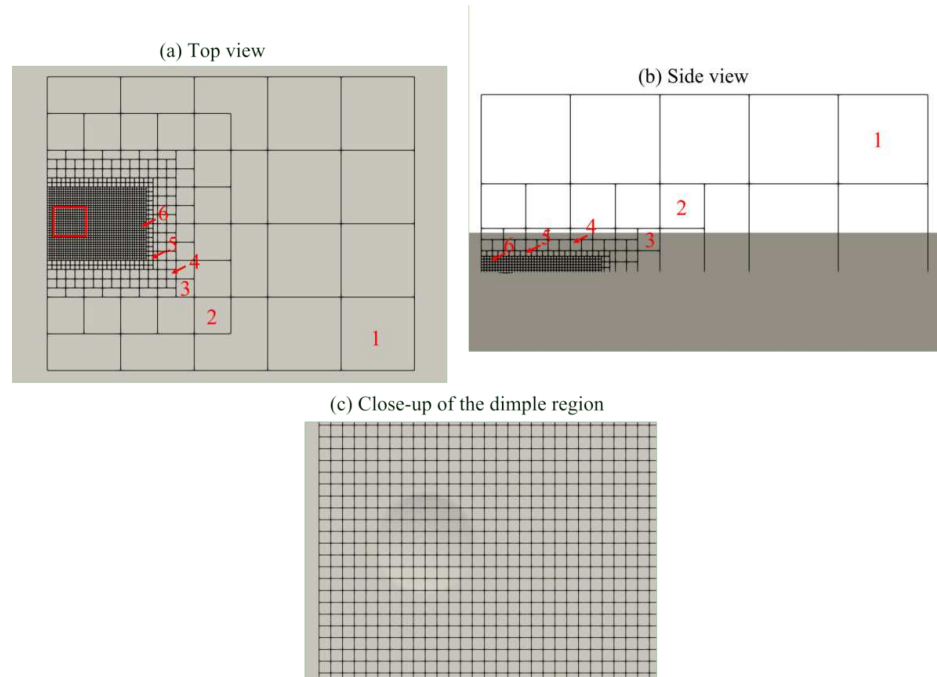


FIG. 4. An illustration of the multi-level grids from the (a) top view and (b) side view; (c) close-up of the grids near the dimple region. The back solid lines denote the edges of a grid box, which contains $40 \times 40 \times 40$ cubic grids.

146 boundary is located $152.6d$ away from the dimpled plate at the bottom while the boundaries in the
 147 spanwise direction are located $307.2d$ from the dimple center.

148 A Blasius-like velocity profile with $w = 0$ is applied at the inlet, while a Neumann condition
 149 for the velocity ($\partial u_i / \partial x = 0$) and a Dirichlet condition for the pressure ($p = 0$) are imposed at
 150 the outlet. At the top boundary, $p = 0$ and $\partial u_i / \partial z = 0$ are imposed in order to ensure a zero
 151 pressure gradient condition in the streamwise direction¹⁹. Slip conditions are used at the side
 152 boundaries. Figures 4(a) and 4(b) show the top view and side view of multi-level grids used for
 153 the simulations, respectively. The edge length (solid lines) of the cubic grid box is half of that of
 154 the grid box at a higher level (marked by red numbers), such as the edge length of the grid box
 155 at level 1 being twice as large as at level 2. Moreover, one grid box is composed of $40 \times 40 \times 40$
 156 cubic grids. The finest grid, i.e., the level-6 grid box with a grid size of $0.003D$ is applied over the

157 dimple region as shown in figure 4(c). The present results are all obtained from this level-6 grid
 158 region. The total grid number is approximately 0.7 billion. This zonal grid algorithm is described
 159 in detail by Manhart¹⁹.

160 III. FLOW PATTERNS WITHIN THE DIMPLE

161 In this section, the flow structure within the dimple of $d/D = 0.05$ is presented and discussed
 162 for δ/D ranging from 0.023 to 0.1 at $Re_D = 20000$. As δ/D decreases, four different flow patterns
 163 are found. The symmetric horseshoe vortex pattern^{11,12} depicted in figure 2(c) is observed for
 164 $\delta/D \in [0.053, 0.1]$. Two other sub-patterns have been identified within this pattern. These sub-
 165 patterns are distinguished by that one pattern consists of a steady asymmetric horseshoe vortex
 166 while the other consists of an asymmetric horseshoe vortex exhibiting quasi-periodic movement.
 167 Furthermore, a transient flow pattern has been observed between the asymmetric horseshoe vortex
 168 pattern and the tornado-like vortex pattern where there is no continuous core line passing through
 169 both vortices. The physical mechanisms underpinning the formation of the vortex structures, as
 170 well as the transition between the different flow patterns, will be discussed in detail below in
 171 subsections A to D.

172 A. Horseshoe vortex pattern

173 Figure 5(a) shows the three-dimensional streamlines (black lines with arrows) and the vortex
 174 core line (thick red line) for $\delta/D = 0.1$. Here the vortex core line is calculated from the vorticity
 175 vector²⁵. The flow is steady and symmetric about the streamwise centerline of the dimple and
 176 is spiraling within the dimple, forming a horseshoe vortex as visualized by the vortex core line.
 177 This flow pattern is the same as the horseshoe vortex pattern identified by Kovalenko *et al.*¹¹
 178 from measurements for channel flow over a recessed dimple, and by measurements conducted
 179 by Tay *et al.*¹² for laminar flat-plate boundary layer flow over a recessed dimple. The physical
 180 mechanisms underpinning the formation of the horseshoe vortex within the dimple will now be
 181 discussed thoroughly. In the following discussion, the flow spiral is decomposed into a spanwise
 182 flow spiral (i.e., a flow spiral in the xz -plane), a vertical flow spiral (i.e., a flow spiral in the xy -
 183 plane), and a streamwise flow spiral (i.e., a flow spiral in the yz -plane), named in accordance with
 184 the direction of the relevant vorticity vector component.

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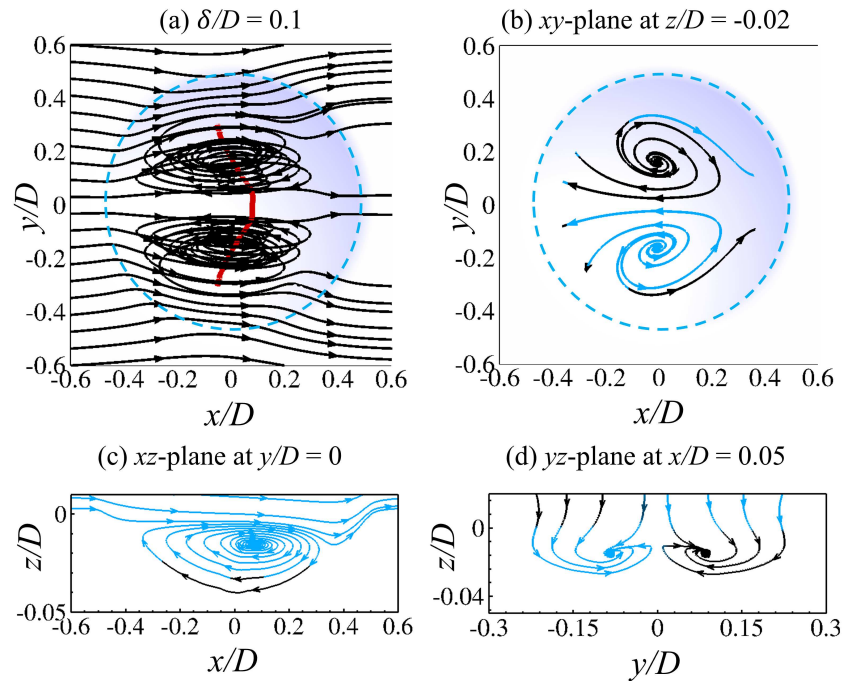


FIG. 5. (a) Three-dimensional (3D) streamlines (black lines with arrows) and the vortex core lines (red thick lines); Two-dimensional (2D) streamlines in the (b) xy -plane at $z/D = -0.02$, (c) xz -plane at $y/D = 0$ (symmetric plane) and (d) yz -plane at $x/D = 0.05$ for $\delta/D = 0.1$. The blue and black lines denote positive and negative values, respectively, of the relevant vorticity component. The dashed blue line denotes the dimple edge.

185 Thus, the vertical flow spiral is depicted in figure 5(b), showing the two-dimensional stream-
 186 lines colored by the vertical vorticity ω_z in the xy -plane at $z/D = -0.02$. The blue and black lines
 187 denote positive and negative values, respectively, of the relevant vorticity components. The span-
 188 wise flow spiral is illustrated by two-dimensional streamlines colored by the spanwise vorticity ω_y
 189 in the xz -plane at $y/D = 0$ shown in figure 5(c), while the streamwise flow spiral is visualized by
 190 two-dimensional streamlines colored by the streamwise vorticity ω_x in the yz -plane at $x/D = 0.05$
 191 shown in figure 5(d). Here the locations of these three planes are selected in order to best visualize

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192 the flow spiral center corresponding to the vortex core line.

193 As shown in figure 5(b), the vertical flow spiral induces negative ω_z for $y > 0$ and positive ω_z
 194 for $y < 0$. This leads to the vertical component of $\vec{\omega}_c$ along the vortex core line being directed
 195 downwards for $y > 0$ and upwards for $y < 0$, thus resulting in a vertically upwards tilting of the
 196 vortex core line. This tilting is visualized in figure 6, showing the vorticity vector $\vec{\omega}_c$ along (i.e.,
 197 tangential to) the vortex core line, where the core line is colored by the vertical location relative
 198 to the flat plate where $z/D = 0$. It should be noted that positive ω_z for $y > 0$ and negative ω_z for
 199 $y < 0$ (figure 5b) are observed near the dimple surface because of the shear layer formed as the
 200 horseshoe vortex approaches the wall.

201 The spanwise flow spiral (figure 5c) mainly induces negative ω_y near the dimple bottom (since
 202 $\partial u/\partial z < 0$ yields the dominating contribution to ω_y at the bottom) and positive ω_y in the central
 203 region of the flow spiral, which is consistent with the spanwise component of $\vec{\omega}_c$ being directed
 204 towards the positive y -direction in this region (figures 5a and 6). It should be noted that $\vec{\omega}_c$ is
 205 nearly parallel to the y -axis near the central part of the dimple.

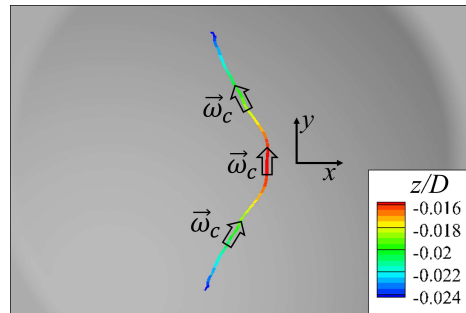


FIG. 6. The top view of the vortex core line.

206 The streamwise flow spiral in the yz -plane (figure 5d) induces negative ω_x for $y > 0$ and positive
 207 ω_x for $y < 0$ near the central region of the flow spiral, causing the downstream tilting of the vortex
 208 core line (figure 6); the streamwise component of $\vec{\omega}_c$ is directed towards the negative x -direction
 209 for $y > 0$ and towards the positive x -direction for $y < 0$.

210 Overall, the vortex core line undergoes deformation into a horseshoe shape as a result of the
 211 streamwise flow spiral, quantified by ω_x . Therefore, the physical mechanism underpinning the de-
 212 formation of the vortex core line into the horseshoe shape can be further illustrated by considering
 213 the transport equation of ω_x as follows

$$\frac{D\omega_x}{Dt} = \omega_x \frac{\partial u}{\partial x} + \omega_y \frac{\partial u}{\partial y} + \omega_z \frac{\partial u}{\partial z} + Re_D^{-1} \nabla^2 \omega_x \quad (3)$$

214 where the first term on the right-hand side denotes the production of ω_x due to the streamwise
 215 stretching or compression of ω_x , while the second and third terms indicate the production of ω_x
 216 due to the streamwise tilting of ω_y and ω_z , respectively^{26,27}. Here $|\omega_x|$ can only increase if the
 217 sign of the production term is the same as the sign of ω_x . Therefore, the increase of ω_x is due to

$$P_{s,x} = \frac{\omega_x}{|\omega_x|} \left(\omega_x \frac{\partial u}{\partial x} \right) \quad (4)$$

$$P_{t,y} = \frac{\omega_x}{|\omega_x|} \left(\omega_y \frac{\partial u}{\partial y} \right) \quad (5)$$

$$P_{t,z} = \frac{\omega_x}{|\omega_x|} \left(\omega_z \frac{\partial u}{\partial z} \right) \quad (6)$$

220 where $P_{s,x}$, $P_{t,y}$, and $P_{t,z}$ are the net production terms of ω_x caused by the streamwise stretching of
 221 ω_x , and the streamwise tilting of ω_y and ω_z , respectively.

222 Figure 7(a) shows the three specific production terms along the vortex core line shown in figure
 223 6 for $\delta/D = 0.1$. The corresponding velocity gradients of u are shown in figure 7(b); the vorticity
 224 components along the vortex core line are shown in figure 7(c). The magnitude of the production
 225 term due to streamwise stretching of ω_x , i.e., $P_{s,x}$, (figure 7a) is much smaller than the magnitude
 226 of the production terms due to the streamwise tilting of ω_y and ω_z , i.e., $P_{t,y}$ and $P_{t,z}$, implying that
 227 the streamwise stretching makes a negligible contribution to the production of ω_x . This is mainly
 228 due to that $|\frac{\partial u}{\partial x}|$ (figure 7b) is very small along the vortex core line.

229 The production term $P_{t,y}$ (figure 7a, showing $-P_{t,y}$) remains negative along the entire vortex core
 230 line, implying that the production of ω_x , and thus the deformation of the vortex core line into the
 231 horseshoe shape, is counteracted by the streamwise tilting of ω_y . The negative value of $P_{t,y}$ is due
 232 to that ω_y (figure 7c) remains positive along the vortex core line; ω_x (figure 7c) is negative for
 233 $y > 0$ and positive for $y < 0$, while $\frac{\partial u}{\partial y}$ (figure 7b) is positive for $y > 0$ and negative for $y < 0$ (see
 234 Eq.5). It appears that $|P_{t,y}|$ (figure 7a) exhibits two maxima; one for $y > 0$ and one for $y < 0$. Here
 235 the increase of $|P_{t,y}|$ from $|y| = 0.12$ towards the maxima is caused by the corresponding increase
 236 of ω_y (figure 7c) due to the strengthening of the spanwise flow spiral towards $y = 0$. Moreover, the
 237 decrease of $|P_{t,y}|$ from the maxima towards $y = 0$ is induced by the corresponding decrease of $|\frac{\partial u}{\partial y}|$
 238 (figure 7b) due to the weakening of the vertical flow spiral towards $y = 0$.

239 The production term $P_{t,z}$ (figure 7a) remains positive along the vortex core line, implying that
 240 the production of ω_x is enhanced by the streamwise tilting of ω_z . Thus the deformation of the

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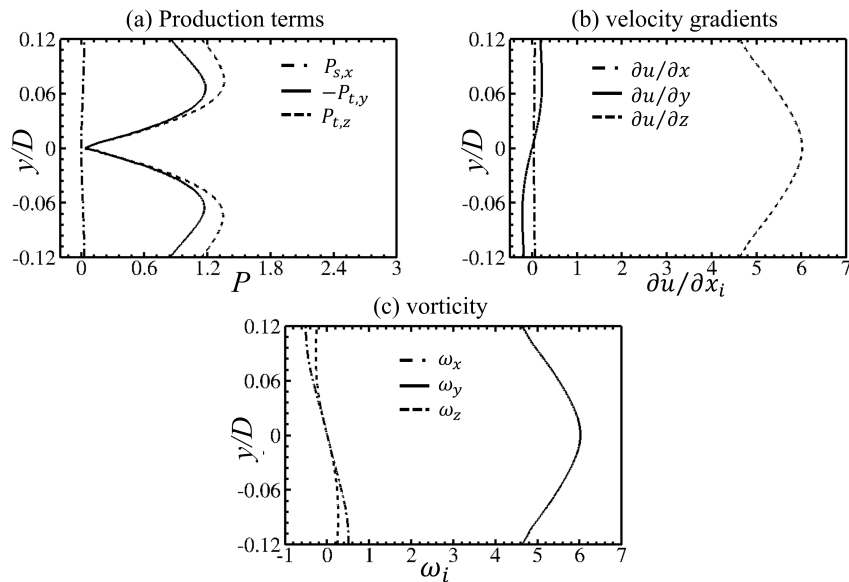


FIG. 7. (a) Production terms of ω_x ($P_{s,x}$, $P_{t,y}$, and $P_{t,z}$), (b) the velocity gradients of u , and (c) the vorticity (ω_x , ω_y , and ω_z) along the vortex core line for $\delta/D = 0.1$.

241 vortex core line into the horseshoe shape is mainly induced by the streamwise tilting of ω_z , which
 242 is consistent with the downstream tilting of the flow spiral. This tilting is mainly induced by the
 243 spanwise flow spiral, where $\frac{\partial u}{\partial z} > 0$ along the whole vortex core line as shown in figure 7(b).
 244 Furthermore, $P_{t,z}$ first increases and then decreases as $|y|$ decreases, qualitatively similar to that
 245 observed for $P_{t,y}$. The underpinning mechanism is similar to that for $P_{t,y}$; as $|y|$ decreases, the
 246 increase of the local dimple depth strengthens the spanwise flow spiral, leading to an increase of
 247 $\frac{\partial u}{\partial z}$ (figure 7b) but weakens the vertical flow spiral, thus resulting in a decrease of $|\omega_z|$ (figure 7c).

248 The symmetric horseshoe vortex pattern also occurs for $\delta/D = 0.069$ and 0.053 as illustrated in
 249 figure 8(a) and 8(b), respectively, showing three-dimensional streamlines and the corresponding
 250 vortex core lines (red thick line). Figures 8(c)-8(d) show the vortex core lines for $\delta/D = 0.1$, 0.069
 251 and 0.053 viewed from the top (xy -plane) and side (xz -plane), respectively. As δ/D decreases, the
 252 vertical tilting of the vortex core line (figure 8d) remains qualitatively similar while the vortex core
 253 line within the central part of the dimple is tilted farther downstream (figure 8c), i.e., the head of

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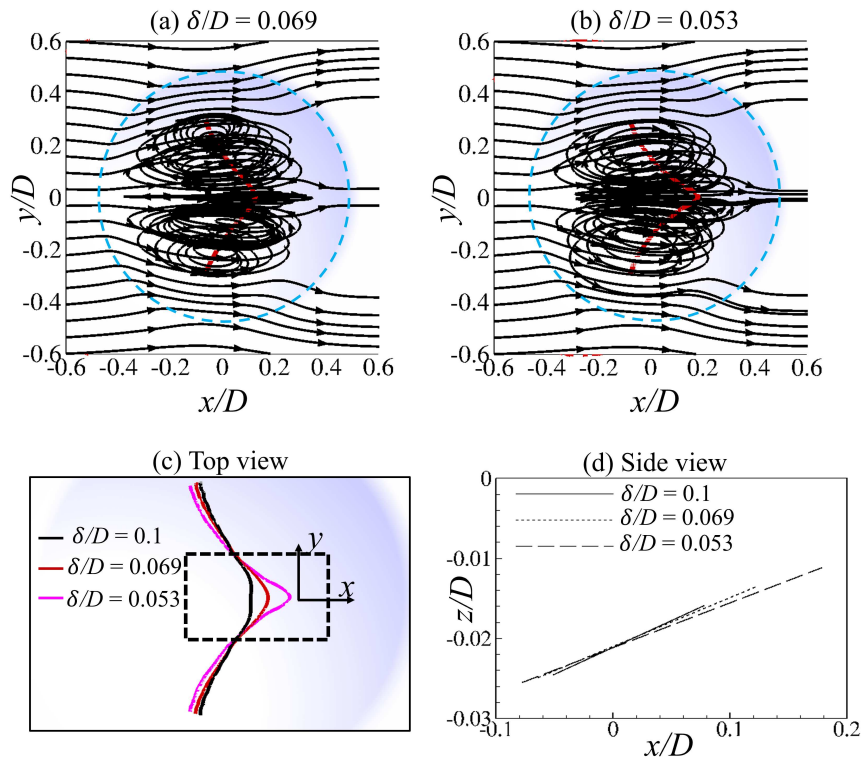


FIG. 8. (a)-(b) Three-dimensional (3D) streamline topologies (black lines with arrows) and the vortex core lines (red thick lines) for $\delta/D = 0.069$, and 0.053 ; (c)-(d) The vortex core lines for $\delta/D = 0.1$, 0.069 and 0.053 from the top and side views. The dashed blue line denotes the dimple edge.

254 the horseshoe vortex moves farther downstream, forming a sharper shape of the horseshoe vortex.
 255 This can be explained by that the streamwise flow spiral within the dimple (and thus ω_x) becomes
 256 stronger as δ/D decreases, and this effect is largest near the center of the dimple. The relative
 257 importance of the tilting and stretching on the growth of ω_x is assessed by evaluating the ratio
 258 P_t/P_{t+s} between the integral values of the tilting terms and the total production term given below

$$P_t = \int_V (P_{t,y} + P_{t,z}) dV \quad (7)$$

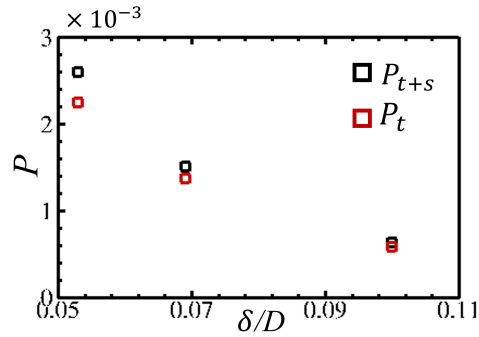


FIG. 9. Integral values of the tilting terms P_t and the total production term P_{t+s} over a control volume V for $\delta/D = 0.053, 0.069$, and 0.1 .

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$$P_{t+s} = \int_V (P_{t,x} + P_{t,y} + P_{t,z}) dV \quad (8)$$

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where V denotes a control volume ($x/D \in [-0.1, 0.4], y/D \in [-0.2, 0.2], z/D \in [-0.03, 0.0]$) covering the vortex core line with the largest deformation as shown by the box with dashed lines in figure 8(c). The values of P_t and P_{t+s} for $\delta/D = 0.1, 0.069$, and 0.053 are shown in figure 9, illustrating that as δ/D decreases from 0.1 to 0.053 , P_t and P_{t+s} increase, while the ratio P_t/P_{t+s} decreases from approximately 93% to 86%. This implies a small increase in the small contribution from the stretching to the production of ω_x .

Overall, the deformation of the vortex core line into the horseshoe form is caused by the stream-wise flow spiral, which is quantified by the streamwise vorticity component ω_x . The enhanced production of ω_x is mainly due to the tilting of ω_z , while the tilting of ω_y counteracts this production. Decreasing δ/D from 0.1 to 0.053 leads to a stronger flow spiral, and thus a larger production of ω_x , within the dimple. It appears that this effect is largest at the center of the dimple, which explains the sharper head of the horseshoe vortex as δ/D decreases.

B. Steady asymmetric horseshoe vortex pattern

As δ/D decreases to 0.04 , the flow exhibits an unsteady behavior initially, where an asymmetric flow pattern develops and results in a steady flow with an asymmetric horseshoe pattern. In the

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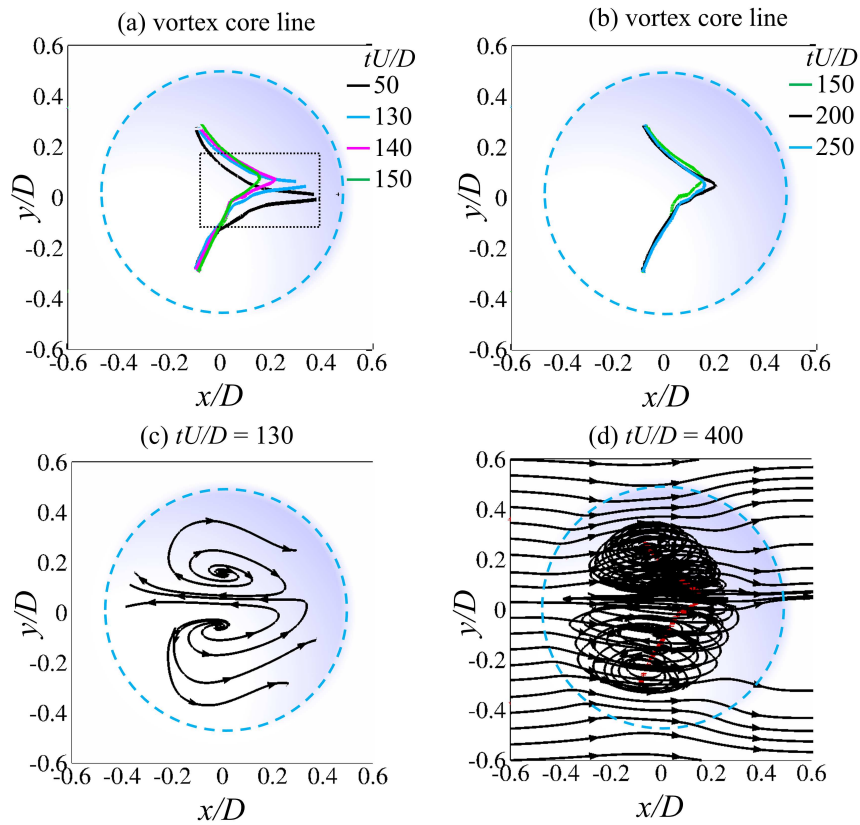


FIG. 10. Instantaneous vortex core lines at (a) $tU/D \in [50, 150]$, (b) $tU/D \in [150, 250]$; (c) Instantaneous two-dimensional streamlines in the xy -plane at $z/D = -0.02$ for $tU/D = 130$; (d) Instantaneous three-dimensional streamlines at $tU/D = 400$ for $\delta/D = 0.04$. The dashed blue line denotes the dimple edge.

275 forthcoming, this transient flow development towards the steady asymmetric vortex pattern will be
 276 described.

277 Figures 10(a) and 10(b) show the instantaneous vortex core lines at $tU/D \in [50, 250]$. For
 278 $tU/D = 50$ (figure 10a), a tornado-like vortex pair is present within the downstream part of the
 279 dimple, which can be further visualized by the corresponding ω_x -contours and by two-dimensional
 280 streamlines in the yz -plane at $x/D = 0.3$ shown in figure 11(a). The interaction between these

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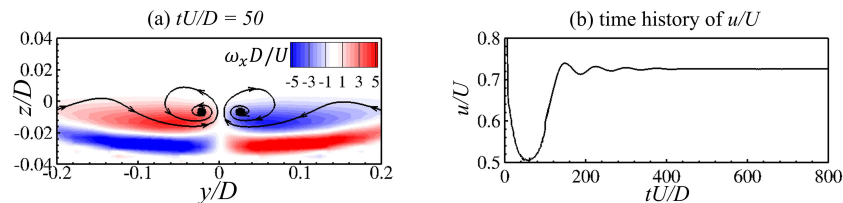


FIG. 11. (a) Two-dimensional streamlines and ω_x -contours in the yz -plane at $x/D = 0.3$; (b) Time history of u probed at $(x/D, y/D, z/D) = (0.5, 0.0, 0.02)$.

281 two tornado-like vortices appears to induce a flow instability, resulting in a flow asymmetry as
 282 visualized by the two-dimensional streamlines in the xy -plane at $z/D = -0.02$ for $tU/D = 130$ in
 283 figure 10(c). Here the flow spiral region for $y < 0$ (denoted the lower flow spiral region) grows
 284 while the flow spiral region for $y > 0$ (denoted the upper flow spiral region) shrinks, leading the
 285 vortex pair to move towards the positive y -direction as well as upstream from $tU/D = 50$ to 130
 286 as shown in figure 10(a). This flow pattern appears to be qualitatively similar to the tornado-like
 287 vortex pattern reported by Kovalenko *et al.*¹¹ and Tay *et al.*¹², although, in their work, only the
 288 dominating vortex core line was depicted. We will refer to this as the tornado-like vortex pair in the
 289 forthcoming. At $tU/D = 140$, the vortices within this tornado-like vortex pair connect, forming
 290 a horseshoe vortex, of which the head is located at $y > 0$ (pink line in figure 10a). Thereafter,
 291 at $tU/D = 150$, the head of the horseshoe vortex moves further upstream. As shown in figure
 292 10(b), the head of the horseshoe vortex moves slightly downstream for $tU/D = 200$, and then, at
 293 $tU/D = 250$, it moves upstream. This small oscillation of the head's position decays gradually
 294 and only lasts for a period of about 400 time units as illustrated by the time history of u probed
 295 at the downstream edge of the dimple, i.e., at $(x/D, y/D, z/D) = (0.5, 0, 0.02)$ as shown in figure
 296 11(b).

297 After $tU/D \approx 400$, the flow becomes steady and asymmetric about the streamwise centerline
 298 of the dimple as shown in figure 10(d), which shows the three-dimensional streamlines and the
 299 vortex core line for $tU/D = 400$. It should be noted that the head also moves vertically upwards
 300 and downwards as the head moves downstream and upstream, but this vertical displacement is less
 301 than 2% of the corresponding streamwise displacement, and will not be further discussed here.

302 The major mechanism underpinning the deformation of the vortex within the dimple can again

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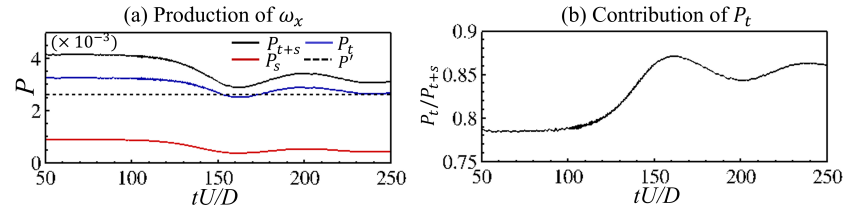


FIG. 12. (a) Time history of the production terms P_{t+s} , P_t , P_s and P' ; (b) Time history of the ratio P_t/P_{t+s} .

303 be explained by the production of ω_x owing to the stretching of ω_x and the tilting of ω_y and ω_z
 304 within the dimple as discussed for the symmetric horseshoe vortex pattern in Section III-A. Figure
 305 12(a) show the time history of $P_s = \int_V P_{s,x} dV$, P_t (Eq.7) and P_{t+s} (Eq.8) for $tU/D \in [50, 250]$. Here
 306 P' denotes the value of P_{t+s} for $\delta/D = 0.053$, which is taken as a reference. The area of the control
 307 volume V is shown by the box with dashed lines in figure 10(a). From $tU/D = 50$ to 130, P_{t+s} for
 308 $\delta/D = 0.04$ is larger than that for $\delta/D = 0.053$ (represented by P'), showing a larger production
 309 of the streamwise vorticity ω_x for $\delta/D = 0.04$ than for $\delta/D = 0.053$, leading to the formation of
 310 the tornado-like vortex pair. Then, from $tU/D = 130$ to 160, P_{t+s} decreases significantly, implying
 311 a decrease in the production of ω_x , resulting in the connection of vortices within the tornado-like
 312 vortex pair, thus forming a horseshoe vortex (figure 10a, pink line). Subsequently, P_{t+s} starts to
 313 increase again but reaches a smaller maximum value than before. The flow pattern remains a
 314 horseshoe vortex where the head moves slightly downstream from $tU/D = 150$ to 200 as shown
 315 in figure 10(b). Thereafter, P_{t+s} decreases as the head moves upstream from $tU/D = 200$ to
 316 250 (figure 10b). This slight upstream and downstream movement of the horseshoe vortex head
 317 continues for an intermediate period and finally reaches a steady state as shown by the time-history
 318 of streamwise velocity u/U in figure 11(b). Moreover, as shown in figure 12(a), the trend of P_{t+s}
 319 is almost the same as for P_t , which is larger than P_s , again implying that the tilting of ω_y and ω_z
 320 yields the dominating contribution to the production of ω_x , which leads to the deformation of the
 321 vortex within the dimple.

322 The relative importance of the tilting and stretching can be further quantified by the time history
 323 of the ratio P_t/P_{t+s} as shown in figure 12(b) for $tU/D \in [50, 250]$. For $tU/D \in [50, 100]$, the
 324 contribution from the tilting (P_t/P_{t+s}) is approximately 78% when the tornado-like vortex pair is
 325 located at the central part of the dimple (figure 10a). Then, P_t/P_{t+s} increases to the maximum

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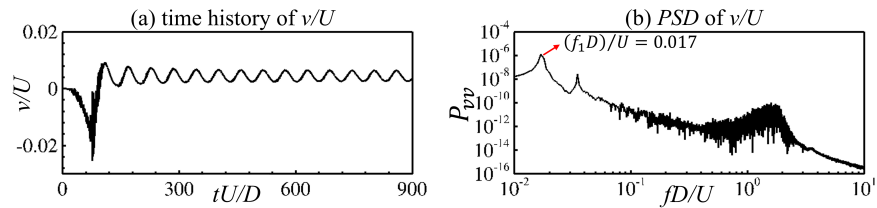


FIG. 13. (a) Time history of v/U probed at $(x/D, y/D, z/D) = (0.5, 0.0, 0.02)$ for $\delta/D = 0.033$; (b) Power spectral density (PSD) of v/U for $\delta/D = 0.033$.

326 value of approximately 87% from $tU/D = 100$ to 150 when the vortex pair starts to move towards
 327 the positive y direction and connect, forming the horseshoe vortex (figure 10a). A decrease of
 328 P_t/P_{t+s} from $tU/D = 150$ to 200 and an increase of P_t/P_{t+s} from $tU/D = 200$ to 250 are observed,
 329 coinciding with the downstream and upstream movements of the horseshoe vortex (figure 10b),
 330 respectively. Finally, P_t/P_{t+s} remains approximately 86% as the flow becomes steady.

331 C. Quasi-periodic asymmetric horseshoe vortex pattern

332 As δ/D decreases to 0.033, the flow exhibits qualitatively similar behavior as the steady asym-
 333 metric pattern discussed above; the flow structure within the dimple initially develops into a
 334 tornado-like vortex pair (see, e.g., figure 10a for $tU/D = 50$), and then these two tornado-like
 335 vortices connect, forming a horseshoe vortex (see, e.g., figure 10a for $tU/D = 140$). It should
 336 be noted that here the head of the horseshoe vortex keeps moving quasi-periodically within the
 337 dimple, i.e., there is no steady state. The flow periodicity is clearly illustrated by the oscillation of
 338 v/U probed at $(x/D, y/D, z/D) = (0.5, 0, 0.02)$ shown in figure 13(a). A mean value of v/U larger
 339 than zero implies that the flow is asymmetric about the streamwise centerline of the dimple where
 340 the flow spiral region grows for $y < 0$ and shrinks for $y > 0$, i.e., the head of the horseshoe vor-
 341 tex moves upstream and towards the positive y -direction. Furthermore, the oscillation frequency
 342 $(f_1 D/U)$ is 0.017 as shown by the power spectral density (PSD) of v/U in figure 13(b).

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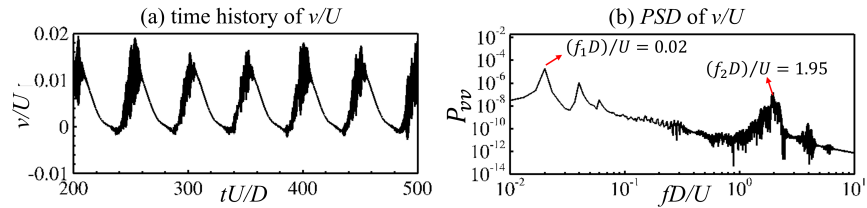


FIG. 14. (a) Time history of v/U probed at $(x/D, y/D, z/D) = (0.5, 0.0, 0.02)$ and (b) Power spectral density (PSD) of v/U for $\delta/D = 0.023$.

343 D. Mixed horseshoe and tornado-like vortex pattern

344 As δ/D decreases to 0.023, the flow becomes more unsteady than for $\delta/D = 0.033$. This can
 345 be clearly visualized by the time-history of v/U probed at $(x/D, y/D, z/D) = (0.5, 0, 0.02)$ for
 346 $\delta/D = 0.023$ in figure 14(a), where the fluctuation of v/U is clearly visible when v/U increases,
 347 e.g., from $tU/D = 340$ to 360. The corresponding vortex core lines are shown in figure 15(a). A
 348 tornado-like vortex pair is present for $tU/D = 340$ and then moves upstream as well as towards the
 349 positive y -direction from $tU/D = 350$ to 360. This behavior is qualitatively similar to that for the
 350 initial flow development in the steady asymmetric pattern (figure 10a) as well as the quasi-periodic
 351 asymmetric pattern, i.e., the formation of the tornado-like vortex pair. As the vortex pair connect
 352 with each other for $tU/D = 370$ (figure 15b), forming a horseshoe vortex, the fluctuation of v/U
 353 vanishes (figure 14a). This indicates that the fluctuation within the oscillation of v/U is mainly
 354 induced by the interaction between vortices within the tornado-like vortex pair (figure 15a). Then,
 355 for $tU/D = 380$, the head of the horseshoe vortex moves downstream and towards the negative
 356 y -direction, and the horseshoe vortex evolves into a tornado-like vortex pair again, coinciding with
 357 the corresponding fluctuation of v/U observed in figure 14(a). This behavior repeats itself quasi-
 358 periodically with alternating formations of tornado-like and horseshoe vortices; this represents
 359 a mixed horseshoe and tornado-like vortex pattern. Moreover, figure 14(b) depicts the power
 360 spectrum density of v/U , showing that the tornado-like vortex pair or horseshoe vortex within the
 361 dimple oscillates with a frequency $(f_1 D)/U$ of 0.02 which is higher than for $\delta/D = 0.033$ (where
 362 $f_1 D/U = 0.017$), while the interaction between the vortices within the tornado-like vortex pair
 363 induces a disturbance with a band of high frequencies with a peak frequency $(f_2 D)/U$ of 1.95.

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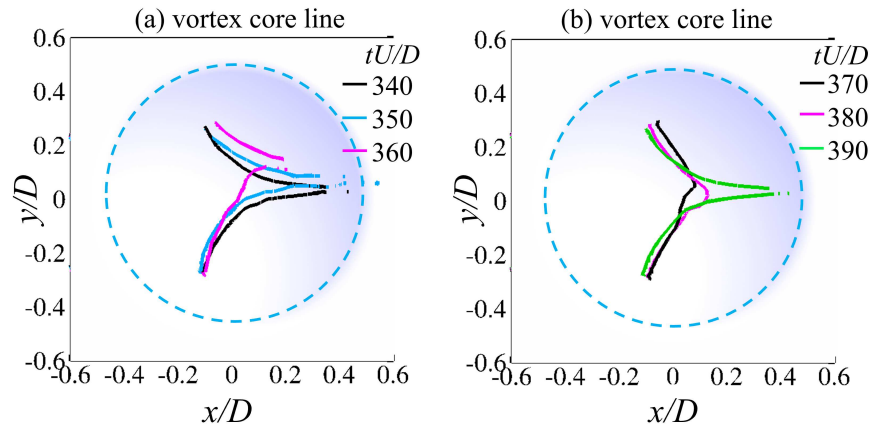


FIG. 15. The instantaneous vortex core lines for $\delta/D = 0.023$ (a) from $tU/D = 340$ to 360 and (b) from $tU/D = 370$ to 390 . The dashed blue line denotes the dimple edge.

364 Overall, a decrease of δ/D from 0.1 to 0.023 leads to increased net production of ω_x , which
 365 leads to a transition between four different flow patterns; *i*) the symmetric horseshoe vortex pat-
 366 tern observed for $\delta/D \in [0.1, 0.053]$ where the flow is steady and symmetric about the stream-
 367 wise centerline of the dimple; *ii*) the steady asymmetric horseshoe vortex pattern observed for
 368 $\delta/D = 0.04$ where the flow asymmetry appears to be induced by a flow instability mechanism
 369 within a tornado-like vortex pair; *iii*) the quasi-periodic asymmetric horseshoe vortex pattern ob-
 370 served for $\delta/D = 0.033$ where the head of the horseshoe vortex within the dimple moves slightly
 371 upstream and downstream quasi-periodically; *iv*) the mixed horseshoe and tornado-like vortex pat-
 372 tern observed for $\delta/D = 0.023$ where the flow within the dimple alternates between the tornado-
 373 like vortex pair and the horseshoe vortex in a quasi-periodic manner. Moreover, the first frequency
 374 ($f_1 D/U$) increases while the secondary frequency ($f_2 D/U$) becomes visible as δ/D decreases
 375 from 0.04 to 0.023 . Hence it is reasonable for us to assume that a decrease of δ/D for a given
 376 dimple depth is qualitatively equivalent to an increase of the dimple depth for a given δ/D , which
 377 might enhance these two frequencies. However, further work is required to confirm this assump-
 378 tion, which is outside the scope of the present work.

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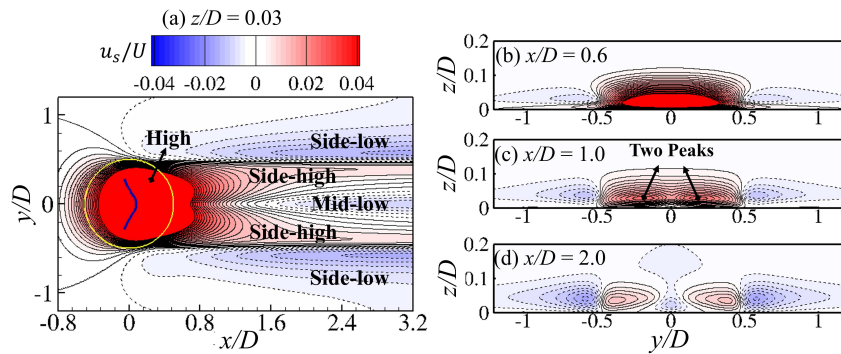


FIG. 16. The streak velocity (u_s/U) contours for $\delta/D = 0.1$ in the (a) xy -plane at $z/D = 0.03$, in the yz -plane at (b) $x/D = 0.6$, (c) 1.0 and (d) 2.0 . The black solid and dashed lines represent the positive and negative values of u_s/U , respectively. The yellow line denotes the dimple edge, and the dark blue solid line indicates the vortex core line within the dimple region.

379 IV. DIMPLE-INDUCED STREAKS ABOVE A DIMPLE

380 In this section, the dimple-induced velocity streaks for different flow patterns are presented
 381 and discussed for δ/D ranging from 0.023 to 0.1 at $Re_D = 20000$. Here the velocity streaks
 382 are depicted by the contours of $u_s/U = (u - u_0)/U$, i.e., by the normalized deviation from the
 383 streamwise velocity u_0 of a zero-pressure-gradient boundary layer flow over a flat plate²⁷, which
 384 is obtained from the present numerical simulations using the same grid resolution as applied for
 385 the dimpled plate. Here the streaks represent a deformation of the streamwise velocity induced by
 386 the dimple, affecting the laminar flat-plate boundary layer flow.

389 A. Symmetric pattern

390 The velocity streaks induced by the dimple mainly consist of four different streaks as depicted
 391 in figure 16(a), showing the contours of u_s/U in the xy -plane at $z/D = 0.03$ for the symmetric
 392 pattern at $\delta/D = 0.1$. These are *i*) a High-speed streak region ($u_s > 0$, marked as High) above the
 393 dimple and in the near-wake region; *ii*) two Side-low-speed streak ($u_s < 0$, marked as Side-low)
 394 located outside the span of the dimple ($y/D < -0.5$ and $y/D > 0.5$); *iii*) two Side-high-speed

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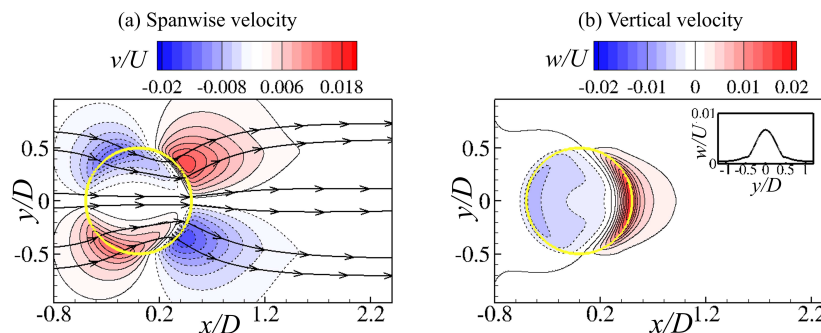


FIG. 17. (a) The spanwise velocity (v/U) contours and the vertical velocity (w/U) contours for $\delta/D = 0.1$ in the xy -plane at $z/D = 0.03$, where the black solid and dashed lines represent the positive and negative values, respectively. The yellow line denotes the dimple edge.

395 streaks ($u_s > 0$, marked as Side-high) and *iv*) a Mid-low-speed streak ($u_s < 0$, marked as Mid-low)
 396 formed farther downstream at $x/D \approx 1.6$. The mechanisms underpinning the formation of these
 397 streaks are now further discussed.

398 The High-speed streak region (figure 16a) is caused by the flow velocity at a given vertical
 399 position of $z > 0$ being located farther away from the dimple surface (located at $z < 0$) than from
 400 the corresponding flat plate (located at $z = 0$), thus leading to a larger flow velocity above the
 401 dimple (for $z > 0$) than above the flat plate, due to less friction. This represents an acceleration of
 402 the streamwise velocity u caused by the dimple, which is here denoted the flow acceleration effect.
 403 This High-speed streak ('High') is located within the span of the dimple at $y/D \in [-0.5, 0.5]$.

404 Two Side-low-speed streaks ($u_s < 0$, marked as Side-low) are formed outside the span of the
 405 dimple ($y/D < -0.5$ and $y/D > 0.5$); this is further illustrated by the u_s/U contours in the yz -plane
 406 at $x/D = 0.6$ in figure 16(b). The formation of the Side-low-speed streaks can be explained by
 407 the v/U contours and the two-dimensional streamlines in the xy -plane at $z/D = 0.03$ which are
 408 both shown in figure 17(a). In order to better visualize the curvature of the streamlines, the values
 409 of v/U are here amplified by a factor of 20 while the contours of v/U are not scaled. It appears
 410 that this flow exhibits the behavior of a confuser-diffuser flow¹¹ as the streamlines (from the left
 411 towards the right) first bend towards the streamwise centerline of the dimple and then bend away
 412 from it. The presence of the dimple-induced spanwise velocity v leads to a decrease of u in the

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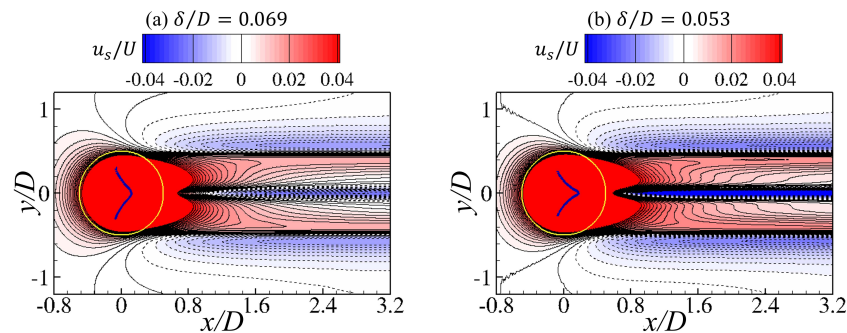


FIG. 18. The streak velocity (u_s/U) contours in the xy -plane at $z/D = 0.01$ for $\delta/D =$ (a) 0.069 and (b) 0.053, where the black solid and dashed lines represent the positive and negative values of u_s/U , respectively. The yellow line denotes the dimple edge, and the dark blue solid line indicates the vortex core line within the dimple region.

413 'Side-low' region marked in figure 16(a), relative to the corresponding streamwise velocity u_0
 414 over the flat plate. It should be noted that this decrease (caused by the diffuser effect) counteracts
 415 the increase of u induced by the flow acceleration effect in the 'High' region. Thus, the gradual
 416 growth of the Side-low-speed streak regions farther downstream is mainly caused by a dynamic
 417 balance between the diffuser effect and the flow acceleration effect. It appears that the Side-low-
 418 speed streaks are less affected by the dimple-induced vertical velocity w than the dimple-induced
 419 spanwise velocity v as illustrated in figure 17(b), showing the w/U -contours in the xy -plane at
 420 $z/D = 0.03$; the induced vertical velocity w is contained within the span of the dimple.

421 The High-speed streak becomes weaker farther downstream (figure 16a) due to the absence of
 422 the flow acceleration effect since the region downstream of the dimple consists of a flat plate. At
 423 $x/D \approx 1.6$, the High-speed streak splits into two Side-high-speed streaks (marked as Side-high)
 424 as the Mid-low-speed streak (marked as Mid-low) is formed in-between as further illustrated by
 425 the u_s/U contours in yz -plane at $x/D = 1.0$ and 2.0 in figure 16(c)-16(d). This is mainly caused
 426 by the local depth variation of the dimple and the horseshoe vortex (the vortex core line is marked
 427 as a solid blue line in figure 16a) within the dimple. As the local depth of the dimple increases
 428 from the spanwise sides of the dimple towards the centerline $y = 0$, the flow acceleration effect
 429 becomes stronger due to less friction, leading to an increase of u_s/U as $|y|$ approaches 0. However,

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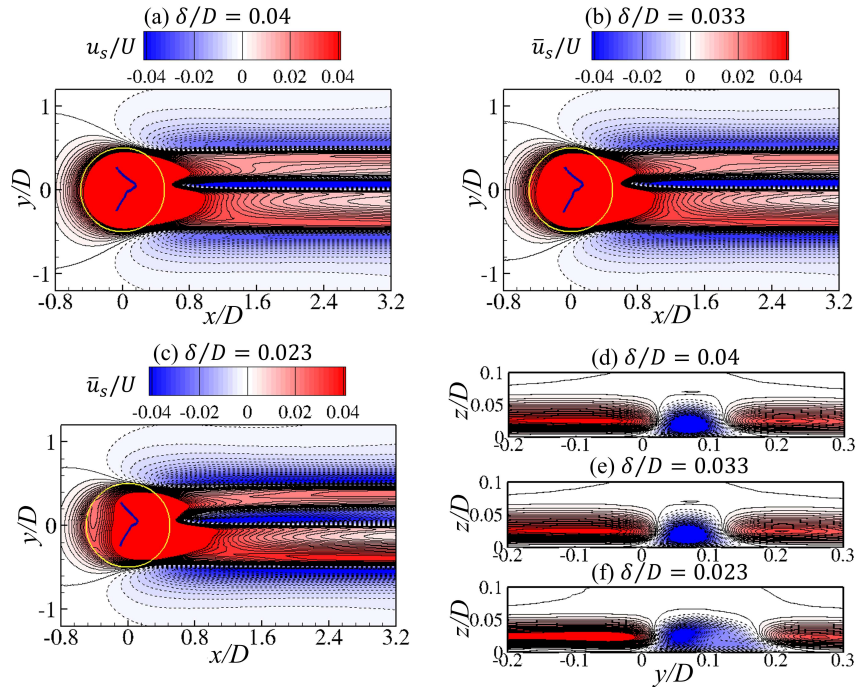


FIG. 19. The streak velocity (u_s/U) contours for $\delta/D = (a)$ 0.04 and the time-averaged \bar{u}_s/U contours for $\delta/D = (b)$ 0.033, and (c) 0.023 in the xy -plane at $z/D = 0.01$, where the black solid and dashed lines represent the positive and negative values of u_s/U , respectively. The yellow line denotes the dimple edge, and the dark blue solid line within the dimple region indicates the vortex core line. $(d)-(f)$ show the corresponding u_s/U or \bar{u}_s/U contours in the yz -plane at $x/D = 1.0$ for $\delta/D = 0.04, 0.033,$ and $0.023,$ respectively.

430 the strengthening of the spanwise flow spiral towards the head of the horseshoe vortex results in
 431 an increase of w/U towards $y = 0$ as illustrated in the top-right frame of figure 17(b), showing
 432 w/U along the spanwise direction at $(x/D, z/D) = (0.6, 0.03)$. Here the vertical velocity induces
 433 a 'lift-up' mechanism²⁸, which displaces low-momentum fluid away from the wall, thus tilting
 434 the boundary layer upwards, leading to a decrease of u for a given z/D . The value of u_s/U first
 435 increases (due to stronger flow acceleration effect) and then decreases (due to stronger 'lift-up'

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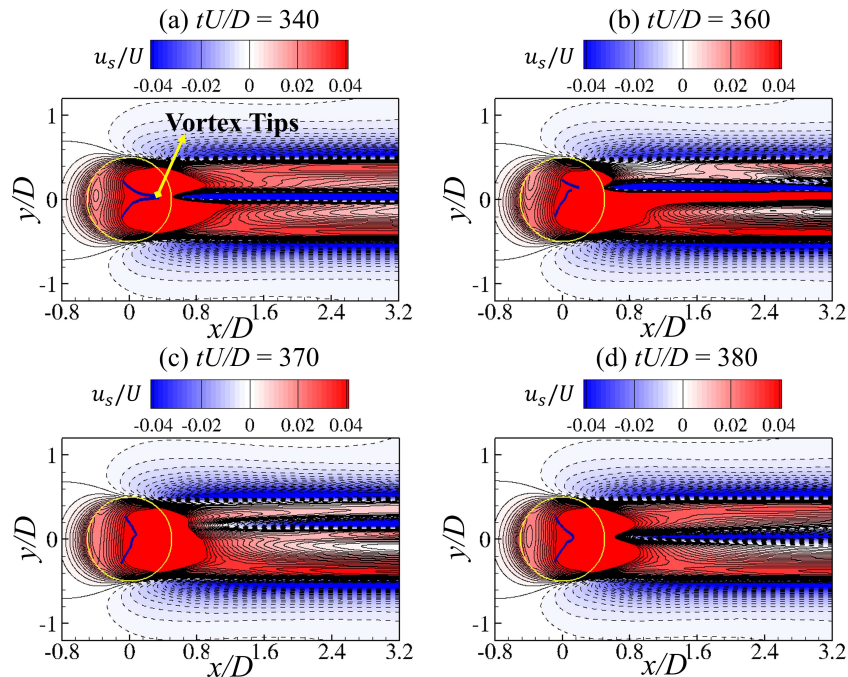


FIG. 20. The instantaneous streak velocity (u_s/U) contours for $\delta/D = 0.023$ at $tU/D =$ (a) 340, (b) 360, (c) 370, and (d) 380 in the xy -plane at $z/D = 0.03$, where the black solid and dashed lines represent the positive and negative values of u_s/U , respectively. The yellow line denotes the dimple edge, and the dark blue solid line indicates the vortex core line within the dimple region.

436 mechanism) from the dimple sides towards $y = 0$, leading to the formation of the Side-high-speed
 437 streaks and the Mid-low-speed streak in-between.

438 Now the effect of δ/D on the velocity streaks is further investigated for the symmetric pattern.
 439 Figures 18(a) and 18(b) show the contours of u_s/U in the xy -plane at $z/D = 0.03$ for $\delta/D =$
 440 0.069 and 0.053, respectively. As δ/D decreases, all velocity streaks become stronger because
 441 a decrease of δ/D implies an increase of the flow velocity at a given vertical position over the
 442 dimple, resulting in a stronger flow spiral, as discussed previously (figure 8), thus strengthening
 443 both the diffuser effect and the 'lift-up' mechanism as well as the flow acceleration effect. It should
 444 be noted that the Mid-low-speed streak is located closer to the dimple as δ/D decreases due to a

445 shift in the balance between the 'lift-up' mechanism and the flow acceleration effect. Moreover,
 446 the spanwise location of the Mid-low-speed streak always coincides with the head of the horseshoe
 447 vortex where the dominating spanwise flow spiral leads to a stronger 'lift-up' mechanism towards
 448 the horizontal line through the horseshoe vortex head. This behavior is also found in the steady
 449 and quasi-periodic asymmetric patterns discussed below.

450 B. Steady and quasi-periodic asymmetric patterns

451 Figure 19 shows the u_s/U contours for the steady asymmetric pattern at $\delta/D = 0.04$, as well
 452 as the time-averaged streak velocity \bar{u}_s/U contours for the quasi-periodic asymmetric pattern, at
 453 $\delta/D = 0.04, 0.033$ and 0.023 in the xy -plane at $z/D = 0.03$. The high- and low-speed streaks
 454 are here asymmetric about the streamwise centerline of the dimple; the High-speed streak located
 455 at $y < 0$ covers a larger region in the dimple wake than that located at $y > 0$. The spanwise
 456 location of the Mid-low-speed streak remains the same as that of the horseshoe vortex head for all
 457 these cases. Moreover, the velocity streak contours are quite similar for $\delta/D = 0.04$ and 0.033
 458 (figure 19a and 19b). However, the Mid-low-speed streak for $\delta/D = 0.023$ (figure 19c) diffuses
 459 in the spanwise direction, becoming wider and weaker than those for $\delta/D = 0.04$ and 0.033 as
 460 shown by the corresponding streak velocity contours in the yz -plane at $x/D = 1.0$ in figures 19(d)-
 461 19(f). This behavior will now be further explained by the instantaneous streak velocity contours
 462 for $\delta/D = 0.023$ shown in figure 20 for $tU/D \in [340, 380]$. The corresponding evolution of the
 463 vortex core line within the dimple was previously discussed in section D (figure 15)

464 For $tU/D = 340$ (figure 20a), the tornado-like vortex pair (the two dark blue solid vortex core
 465 lines) within the dimple, which is also clearly shown in figure 15(a), is slightly asymmetric about
 466 the streamwise centerline of the dimple. The spanwise location of the Mid-low-speed streak coin-
 467 cides with the horizontal line between the two tips of the vortex pair. Thus, the velocity streaks are
 468 asymmetric about the horizontal dimple centerline where the High-speed streak located at $y < 0$ is
 469 slightly larger than that located at $y > 0$. It should be noted that here the 'lift-up' mechanism can
 470 also be induced by the tornado-like vortex pair as discussed in Brandt and Henningson²⁹, but, in
 471 the present case, this effect is much weaker than that induced by the spanwise flow spiral. Then,
 472 for $tU/D = 360$ (figure 20b), the tornado-like vortex pair moves towards the positive y direction,
 473 leading to the corresponding movement of the Mid-low-speed streak, thus resulting in a more
 474 asymmetric streak pattern than that for $tU/D = 340$; the Side-low-speed streak located at $y > 0$

475 is much weaker and smaller than that located at $y < 0$. This is mainly because the flow accelera-
 476 tion effect for $y > 0$ is counteracted by the 'lift-up' mechanism (which is strongest between the
 477 two tips of the vortex pair) and strongly weakened by the friction due to the decrease of the local
 478 depth within the dimple towards the dimple side. Thereafter, for $tU/D = 370$ (figure 20c), the
 479 connection of vortices within the tornado-like vortex pair forms a horseshoe vortex, which moves
 480 upstream; the spanwise flow spiral (figure 15) becomes weaker, causing a weaker 'lift-up' mech-
 481 anism, thus leading to the decay and downstream movement of the Mid-low-speed streak. As the
 482 head of the horseshoe vortex moves towards the centerline $y = 0$ and downstream for $tU/D = 380$
 483 (figure 20d), the velocity streaks become nearly symmetric again while the Mid-low-speed streak
 484 grows and moves closer to the dimple. Overall, the Mid-low-speed streak is widened by the span-
 485 wise movement of the vortex pair (or by the horseshoe vortex) and weakened by the upstream
 486 movement of the horseshoe vortex.

487 Overall, there are three major mechanisms underpinning the formation and evolution of the
 488 dimple-induced velocity streaks; *i*) the flow acceleration effect, which induces a larger u above the
 489 dimple (for $z > 0$) than above a flat plate; *ii*) the flow diffuser effect, which induces a spanwise
 490 velocity in the dimple wake, resulting in a decrease of u ; *iii*) the 'lift-up' mechanism, which
 491 induces a vertical velocity w , thus leading to a decrease of u at a given vertical position.

492 V. SUMMARY AND CONCLUSION

493 In the present work, numerical investigations of a laminar zero-pressure-gradient boundary
 494 layer flow over a single shallow dimple recessed in a flat plate have been conducted for $Re_D =$
 495 20000 and $d/D = 0.05$ with $\delta/D \in [0.023, 0.1]$. The flow patterns consist of a transient connec-
 496 tion between the horseshoe vortex pattern and the tornado-like vortex pattern as shown in figure
 497 2(c) and 2(d), respectively. Here the horseshoe vortex is characterized by a continuous vortex
 498 core line through the two flow spirals within the dimple, while this core line is not continuous
 499 for the tornado-like vortex. This is consistent with the previous qualitative description of these
 500 vortex patterns (based on measurements) by Kovalenko *et al.*¹¹ and Tay *et al.*¹², although they
 501 did not present the vortex core line. Moreover, the physical mechanisms underpinning the vortex
 502 formation and deformation as well as the transition between the different flow patterns have been
 503 visualized and discussed. The dimple-induced streak patterns and the mechanisms underpinning
 504 their formation and evolution have also been analyzed in detail.

TABLE I. Four flow patterns and the corresponding characteristics for laminar boundary layer flow over a shallow dimple with $d/D = 0.05$ for δ/D ranging from 0.023 to 0.1 at $Re_D = 20000$.

| δ/D | Pattern | Characteristic of each pattern |
|------------|--|--|
| 0.1-0.053 | Symmetric horseshoe vortex pattern | A steady and symmetric horseshoe vortex, of which core line is tilted farther downstream with decreased δ/D . |
| 0.040 | Steady asymmetric horseshoe vortex pattern | A steady and asymmetric horseshoe vortex. |
| 0.033 | Quasi-periodic asymmetric horseshoe vortex pattern | An asymmetric horseshoe vortex, where the head moves quasi-periodically within the dimple. |
| 0.023 | Mixed horseshoe and tornado-like vortex pattern | An alternating formation of a horseshoe vortex and a tornado-like vortex pair. |

505 The results show that as δ/D decreases from 0.1 to 0.023 for $Re_D = 20000$, the flow with the
 506 single dimple exhibits a transition sequence between four different flow patterns as summarized
 507 in Table I; *i*) a symmetric horseshoe vortex pattern for $\delta/D \in [0.053, 0.1]$, where the horseshoe
 508 vortex is formed within the dimple with the head located farther downstream as δ/D decreases;
 509 *ii*) a steady asymmetric horseshoe vortex pattern for $\delta/D = 0.04$, where an asymmetric horseshoe
 510 vortex is formed within the dimple; *iii*) a quasi-periodic asymmetric horseshoe vortex pattern for
 511 $\delta/D = 0.033$, where the head of the asymmetric horseshoe vortex exhibits small oscillation within
 512 the dimple; *iv*) a mixed horseshoe and tornado-like vortex pattern for $\delta/D = 0.023$, characterized
 513 by the alternating formation of a horseshoe vortex and a tornado-like vortex pair (tornado-like
 514 vortex pair).

515 The deformation of the vortex core line into the horseshoe shape is mainly caused by the
 516 streamwise flow spiral, quantified by streamwise vorticity component ω_x . The enhancement of
 517 ω_x is mainly due to the tilting of the vertical vorticity component ω_z . A decrease of δ/D from 0.1
 518 to 0.053 leads to a stronger flow spiral, and thus an enhancement of ω_x within the dimple. This
 519 effect appears to be largest at the center of the dimple, thus explaining the sharper head of the
 520 horseshoe vortex as δ/D decreases. As δ/D decreases to 0.04, the interaction between the two
 521 tornado-like vortices appears to induce a flow instability, which results in an asymmetric steady

522 horseshoe vortex relative to the streamwise centerline of the dimple. An even further growth of
 523 ω_x for $\delta/D = 0.023$ leads to a quasi-periodic alternation between the horseshoe vortex and the
 524 tornado-like vortex pair.

525 There are four different dimple-induced velocity streaks above the dimple; *i*) a High-speed
 526 streak above the dimple caused by a flow acceleration effect due to less friction over the dimple
 527 than over the flat plate for a given $z/D > 0$; *ii*) two Side-low-speed streaks located outside the
 528 span of the dimple induced by the flow diffuser effect in the downstream part of the dimple, where
 529 the streamline is bent away from the streamwise centerline of the dimple; *iii*) two Side-high-speed
 530 streaks and *iv*) one Mid-low-speed streak in-between the two Side-high-speed streaks. These
 531 high- and low-speed streaks are mainly caused by a dynamic balance between a stronger flow
 532 acceleration over the dimple and a stronger 'lift-up' mechanism (which tilts the boundary layer
 533 upwards) from the dimple sides towards the streamwise line through the vortex head.

534 The velocity streaks above the dimple are symmetric about the streamwise centerline of the
 535 dimple for $\delta/D \in [0.053, 0.1]$, while for $\delta/D \in [0.023, 0.04]$, the velocity streaks become asym-
 536 metric where the Side-high-speed streak located at $y > 0$ is narrower and weaker than its counter-
 537 part located at $y < 0$. It should be noted that the spanwise location of the Mid-low-speed streak
 538 always coincides with the streamwise line through the vortex head or between the tips of the
 539 tornado-like vortex pair. This Mid-low-speed streak moves upstream as δ/D decreases due to the
 540 stronger 'lift-up' mechanism. Moreover, the Mid-low-speed streak is widened by the spanwise
 541 movement of the tornado-like vortex pair (or the horseshoe vortex) and weakened by the upstream
 542 movement of the horseshoe vortex.

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548 **DATA AVAILABILITY STATEMENT**

549 The data that support the findings of this study are available from the corresponding author
550 upon reasonable request.

TABLE II. Four flow patterns and the corresponding characteristics for laminar boundary layer flow over a shallow dimple with $d/D = 0.05$ for $\delta/D = 0.023$ at $Re_D = 20000$.

| Case | L_x/D | L_y/D | L_z/D | Δ_c/D | $f_1 D/U (\pm 0.0025)$ | $f_2 D/U (\pm 0.0025)$ |
|------------|---------|---------|---------|--------------|------------------------|------------------------|
| M1, Coarse | 19.2 | 15.26 | 7.63 | 0.004 | 0.02 | 1.95 |
| M1, Medium | 19.2 | 15.26 | 7.63 | 0.003 | 0.02 | 1.95 |
| M1, Fine | 19.2 | 15.26 | 7.63 | 0.0025 | 0.02 | 1.95 |
| M2, Medium | 26.88 | 23.04 | 11.78 | 0.003 | 0.02 | 1.95 |

551 Appendix A: Computational domain and grid convergence study

552 In this section, computational domain and grid convergence studies have been conducted for
 553 the zero-pressure-gradient boundary layer flow over the dimpled plate for $Re_D = 20000$ with
 554 $\delta/D = 0.023$. A cubic grid is applied over the dimple as shown in figure 4. As shown in ta-
 555 ble II, three different grid resolutions, i.e., the coarse ($\Delta_c/D = 0.004$), medium ($\Delta_c/D = 0.003$),
 556 and fine ($\Delta_c/D = 0.0025$) grid resolutions have been used to investigate the grid convergence
 557 within the computational domain (M1) of $(L_x/D, L_y/D, L_z/D) = (19.2, 15.26, 7.63)$, where Δ_c is
 558 the grid size for the level-6 grid region, while L_x , L_y , and L_z denote the streamwise, spanwise, and
 559 vertical lengths of the computational domain, respectively. Another computational domain (M2)
 560 which is 1.5 times larger than domain M1 in all directions with the medium grid resolution has
 561 been utilized for the domain convergence study.

562 Table II also shows the two dominating frequencies f_1 and f_2 corresponding to the vortex dy-
 563 namics within the dimple and the interaction between the two tornado-like vortices, respectively.
 564 These two frequencies remain the same for all cases given in table II. Figure 21 shows the com-
 565 parison of the time-averaged streamwise velocity \bar{u}/U along z/D at $y = 0$ within the dimple at
 566 $x/D = -0.3$ (black), 0.0 (red) and 0.3 (light blue) and within the dimple wake at $x/D = 1.0$ (dark
 567 blue), 2.0 (green) and 3.0 (pink) obtained by the three different grid resolutions (Coarse, Medium
 568 and Fine) and the two different computational domains (M1, M2). These time-averaged values are
 569 obtained from the data with 400 time units (D/U) after the flow is fully developed. The results
 570 obtained by the fine and medium grid resolutions (using the domain M1) show a good agreement
 571 while a small deviation can be observed for the coarse grid resolution (using the domain M1) at,

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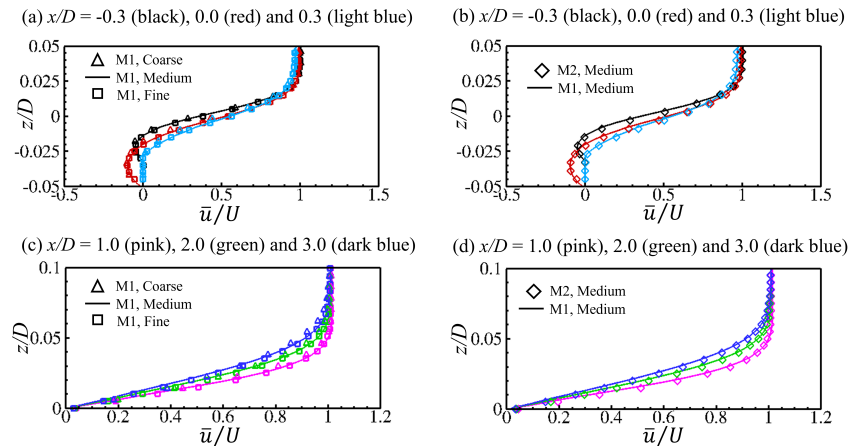


FIG. 21. The time-averaged streamwise velocity \bar{u}/U profile along z at $y = 0.0$ for three different grid resolutions (Coarse, Medium, and Fine) and two different computational domains (M1, M2) sampled (a)-(b) within the dimple at $x/D = -0.3$ (black), 0.0 (red), and 0.3 (light blue), as well as (c)-(d) within the downstream region at $x/D = 1.0$ (pink), 2.0 (green), and 3.0 (dark blue).

572 e.g., $x/D = 0.3$ and $x/D = 1.0$ as shown in figure 21(a) and 21(c), respectively. A good agree-
 573 ment is also obtained for the results obtained by the two different computational domains using
 574 the medium grid resolution as shown in figure 21(b) and 21(d). Furthermore, we have observed
 575 an initially transient flow, which contains small-scale fluctuations corresponding to the secondary
 576 frequency ($f_2 D/U$). Hence, the Kolmogorov scale η for the (M1, Medium)-case as given in Table
 577 II is calculated by $(\nu^3/\varepsilon)^{0.25}$, where ε denotes the local energy dissipation. The maximum ratio
 578 between the grid size Δ_c and the Kolmogorov scale, Δ_c/η , within the level-6 grid region is ap-
 579 proximately 3.5, implying that the smallest flow structures are resolved³⁰. Moreover, it should be
 580 noted that all the flow structures discussed in the present work are located within the level-6 grid
 581 region. Overall, the medium grid resolution with the computational domain 1 (M1, Medium) is
 582 applied in the present work to obtain grid- and domain-independent results.

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