

Experimental and numerical study on the R744 ejector with a suction nozzle bypass

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Abstract

State-of-the-art R744 cooling cycles include highly efficient ejectors as a crucial element for competitive energy consumption in warm climates. However, an improvement is still possible, as shown in this study. This study is the first to experimentally investigate a construction of the R744 ejector with a modulated opening of the suction nozzle bypass duct. Four bypass positions along the ejector axis were tested using three sets of motive nozzle conditions characteristic of a refrigeration unit operating in a warm climate. Two levels of evaporation temperature, as well as the superheat influence, were investigated to evaluate the application potential. The efficiency of the prototype was equal to or greater than that of the standard construction because the closed bypass duct did not result in deteriorated ejector performance. The best bypass positioning resulted in improved efficiency for the pressure lift up to 7 bar. The maximum efficiency improvement was 37% with application potential for systems with low-pressure lift modes. The simulation of the full 3-D bypass ejector allowed for insight into the efficiency improvement. Finally, guidelines were given for further improvement considering the connection between the standard suction chamber and the bypass chamber.

Keywords: R744 ejector, experimental tests, performance improvement, bypass, numerical analysis

Nomenclature

Abbreviations

\dot{W}	Expansion work rate	W
D	Diameter	m
d	Width	m
h	Specific enthalpy	J·kg ⁻¹
L	Length	m

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m	Mass flow rate	$\text{kg}\cdot\text{s}^{-1}$
p	Absolute pressure	Pa
s	Specific entropy	$\text{J}\cdot\text{kg}^{-1}\cdot\text{K}^{-1}$

Subscripts

β, γ	Angle	$^{\circ}$
χ	Mass entrainment ratio	-
δ_m	Relative error of the mass flow rate	%
η	Efficiency	-
Ψ	Dimensionless position of the bypass duct along the ejector axis	-

Superscripts

bps	Bypass	-
CFD	Computational Fluid Dynamics	-
DIFF	Diffuser	-
ej	Ejector	-
EXP	Experimental	-
g	Saturated gas	-
l	Saturated liquid	-
max	Maximum	-
mix	Mixer	-
MN	Motive nozzle	-
OUT	Outlet	-
rec	Recovered expansion work	-
sat	Saturation state	-
SIM	Simulation	-
SN	Suction nozzle	-

Roman Symbols

$\text{CO}_2, \text{R744}$	Carbon dioxide	-
HEM	Homogeneous equilibrium model	-
HPV	Expansion valve	-
HVAC&R	Heating, ventilation, air conditioning refrigeration	-
IHX	Internal heat exchanger	-

1. Introduction

1.1. State-of-the-art R744 HVAC&R systems

Intensive research on carbon dioxide (R744) cooling technology as a standard solution in mobile applications with wide potential for large-scale heating, ventilation, air conditioning and refrigeration (HVAC&R) was proposed over two decades ago [1]. Carbon dioxide installations present substantial benefits in the form of unit compactness, easily accessible cooling-heating integration and inexpensive and unlimited availability concerning current and future legal requirements [2, 3, 4, 5]. Innovative HVAC&R systems based on R744 became a standard solution for supermarkets [6], as well as a domestic solution for heat pumps [7]. The R744 development provided integrated solutions for hotels [8], office buildings [9] and even entire islands, such as Mauritius [10].

13 A crucial and common feature in state-of-the-art R744 cooling and heating tech-
14 nology is the implementation of ejectors to the system layout [11], which brings
15 substantial differences between particular CO₂-based systems [12]. The study pre-
16 sented by Gullo et al. [13] underlined the substantial influence of the R744 transcritical
17 ejector cycles in overcoming challenging applications in warm climates. Further-
18 more, according to Gullo et al. [14], the next generations of large-scale R744 systems
19 with the integration of heating and cooling purposes should be based on cutting-
20 edge ejector solutions. Moreover, the potential for R744 cycles in various operating
21 conditions is continuously explored in areas outside standard HVACR applications.
22 For example, cryogenic cooling cycles involving ejectors were investigated in [15],
23 while advanced systems of energy storage with condensing ejectors were presented
24 in [16].

25 *1.2. Regulation and control strategies of the ejector performance*

26 Considering the importance of ejector utilisation in R744 systems, several re-
27 search areas could be indicated. Namely, the design process, control strategies and
28 continuously growing number of cycle configurations, including those in commer-
29 cial applications, were underlined in the comprehensive review presented by Elbel
30 and Lawrence [17]. This paper presented a rapid development of the ejector research
31 field that could be captured in significant improvement of the design tools starting
32 from the first 1-D models to advanced approaches based on computational fluid dy-
33 namics (CFD) methods. Furthermore, strategies and mechanisms of ejector control
34 were underlined as a key aspect for further development because of numerous ap-
35 plications in which off-design conditions are inevitable. As a consequence of this
36 operation, the ejector performance deteriorates because of degraded mass entrain-
37 ment.

38 The first experimental analysis [18] of the control mechanism based on needle
39 implementation into the R744 ejector motive nozzle to regulate ejector performance
40 was characterised as an effective solution. On the other hand, the reduction of the
41 motive nozzle throat in the controllable ejector should be carefully adjusted because
42 of the possibility of choking phenomena along with increased friction related to the
43 increased surface area introduced to the high-speed motive flow area [19].

44 The multi-ejector concept was presented by Hafner et al. [20] and experimen-
45 tally investigated by Banasiak et al. [21], while this technology is now available on
46 the commercial market. This solution could be located opposite the fluent needle
47 positioning in the motive nozzle throat and could result in continuous regulation.
48 The multi-ejector approach is based on multiple ejectors operating in parallel and
49 depends on the requested or delivered motive stream. Hence, the regulation of the
50 motive ports is managed in a binary manner by turning on or off the proper ejec-
51 tor regarding its size and operation range. Compared to the controllable ejector, the
52 multi-ejector solution brings the benefits of reduced friction in the motive nozzle
53 and stable performance of the ejectors. However, manufacturing multiple ejectors
54 and control valves involves higher investment costs.

55 Lawrence and Elbel [22] presented an experimental analysis of control strate-
56 gies based on the controllable ejector and expansion valves in serial and parallel

57 to the motive port. The comparison revealed the advantage of the adjustable ejec-
58 tor in terms of the ejector efficiency regarding the off-design conditions of ejector
59 operation. On the other hand, the level of this advantage could be lowered regard-
60 ing the simplicity and low cost of the expansion valves. A novel solution based on
61 vortex introduction to the motive stream through the tangentially introduced duct
62 was proposed and experimentally investigated by Zhu and Elbel [23]. Based on the
63 R134a flow, the authors delivered an analysis of the effective stream control and vi-
64 sualised the resulting expansion jet [23]. Furthermore, the developed mechanism
65 was implemented on the R744 ejector with positive results concerning the ejector
66 performance and the aforementioned effective regulation [24]. The aforementioned
67 studies improved the state-of-the-art knowledge and application potential based on
68 effective motive nozzle regulation in the R744 ejector. On the other hand, each anal-
69 ysed solution of motive stream regulation involves a noticeable pressure loss and
70 lower performance factors compared to those of the on-design conditions.

71 *1.3. Application of the suction nozzle bypass*

72 An unfavourable pressure distribution in the ejector mixing chamber could be
73 present even despite the motive flow regulation in the off-design conditions. The
74 relationship between the motive nozzle, premixing chamber and mixer was investi-
75 gated in the study presented by Palacz et al. [25], in which comprehensive optimi-
76 sation of the R744 ejector geometry was provided. The inappropriate dimensions of
77 the mixing section for the given motive nozzle states resulted in high entropy gen-
78 eration in the shock-wave pattern, correlated choking phenomena and finally con-
79 strained entrainment. A similar design criterion was confirmed for other working
80 fluids as well [26, 27].

81 In the off-design conditions, avoidance of the flow with high entropy generation
82 can be realised based on the additional bypass duct where suction flow is introduced
83 at the end or after the mixing section of the ejector. On this basis, the second inlet
84 of the suction nozzle is implemented in the ejector, which is then called a two-stage
85 ejector. However, in this study, the nomenclature of the bypass ejector will be used.
86 In the study presented by Chen et al. [28], the air ejector with bypass was proposed
87 and numerically analysed. Further CFD-based investigations focused on the geo-
88 metrical and operational factors that influence the performance of the bypass ejec-
89 tor [29]. Depending on the pressure conditions at the ejector ports, the mass en-
90 trainment ratio was intensified by up to 32.8%. Moreover, the crucial effect of the
91 geometry and positioning of the bypass duct was emphasized.

92 The bypass implementation to the ejector with an adjustable motive nozzle throat
93 (spindle insertion) was numerically analysed with methane as a working fluid [30].
94 The simulation results of a baseline ejector were validated against the data obtained
95 from an industrial natural gas field located in northwestern China. The numerical
96 analysis of the combined ejector with the spindle and the bypass revealed a large
97 potential for improvement of 75.0% compared to the baseline ejector. The same
98 methane ejector was analysed in the study by Chen et al. [31], where the implemen-
99 tation of two separated bypass inlets was considered. The authors analysed various

100 geometrical configurations of the bypass inlets. Moreover, the suction pressure of
101 each bypass inlet was analysed to evaluate the maximum potential in a proper con-
102 figuration of the serviced natural gas wells. A computed improvement of 48.93% over
103 the aforementioned baseline ejector was obtained.

104 Tang et al. [32] proposed a steam ejector with the suction nozzle bypass. The
105 positioning and geometry of the bypass inlet were numerically analysed [33], pro-
106 viding the optimum configuration for the maximum improvement of the mass en-
107 trainment. However, the reported improvement was 3.8%, which could be consid-
108 ered substantially lower than those of studies in which R134a [29] and methane [30]
109 flows were analysed. On the other hand, the bypass duct in the steam ejector was
110 considered a potential tool for pressure regulation in off- and on-design conditions
111 [34].

112 A numerical analysis of the bypass ejector with R744 as a working fluid was pre-
113 sented by Bodys et al. [35]. The best bypass duct shape and positioning with respect
114 to the mixing section improvement of the mass entrainment ratio was 37.0% for the
115 lowest tested pressure lift of 4 bar. The aforementioned studies reported a large po-
116 tential for performance improvement in the case of properly designed bypass ejec-
117 tors. However, none of the proposed bypass ejectors was examined experimentally
118 because the validation procedures covered only the baseline ejectors. To the best of
119 the authors' knowledge and comprehensive review of the ejector research field [36],
120 an experimental analysis of the R744 bypass ejector is not available in the literature.
121 Moreover, a control strategy for the bypass duct opening has not been investigated.

122 In this study, experimental analysis of the R744 bypass ejector is presented and
123 supplemented by a full 3-D numerical simulation of selected cases. The research
124 objectives include evaluating the performance of the bypass ejector and developing
125 a control strategy for the bypass duct opening. The ejector geometry with the bypass
126 investigated in the preliminary study [35] was used to manufacture the research con-
127 struction with stepwise regulation of the bypass duct opening. The prototype was
128 implemented in a laboratory R744 refrigeration unit dedicated for ejector tests. The
129 exchangeable modules of the ejector prototype allowed for the performance map-
130 ping of the four bypass positions. The refrigeration conditions of the unit operating
131 in warm and hot climate zones were investigated. Up to 37% of the ejector efficiency
132 improvement in the off-design conditions with low-pressure lift was registered. Re-
133 garding supercritical motive conditions, the prototype was numerically studied us-
134 ing the homogeneous equilibrium model (HEM) [37] of the two-phase carbon diox-
135 ide flow through the ejector. Consequently, features of the flow with the closed and
136 open bypass ducts were analysed based on the absolute pressure and velocity mag-
137 nitude distribution. On this basis, the application potential of the bypass ejector was
138 experimentally confirmed along with a proposition for further improvement of this
139 device.

140 **2. R744 laboratory installation dedicated to the ejector performance evaluation**

141 A laboratory R744 refrigeration unit dedicated to ejector tests was used for the
142 experimental analysis of the bypass ejector. An enhanced description and stability

143 analysis of the installation was published by Haida et al. [38]. The test rig, presented
144 in Fig. 1, was designed to cover the motive nozzle capacity at $360 \text{ kg}\cdot\text{h}^{-1}$. For this rea-
145 son, the Dorin CD1400H compressor was selected. The high-side pressure was ad-
146 ditionally controlled using the Danfoss CCMT type expansion valve (HPV) operating
147 in parallel with the ejector lines. Hence, the R744 loop was a transcritical booster sys-
148 tem with medium-temperature evaporation supported by ejector lines and an inter-
149 nal heat exchanger (IHX). The heat sources were served by auxiliary glycol loops that
150 were connected with an additional heat exchanger. The heat was delivered by six
151 heaters in the glycol tank and by the aforementioned glycol-glycol heat exchanger.
152 All heat exchangers were manufactured by the SWEP company. Finally, the con-
153 trol system was based on the AK-PC-782A Danfoss unit, and all measurements were
154 recorded by Danfoss StoreView software and then by an in-house developed script.

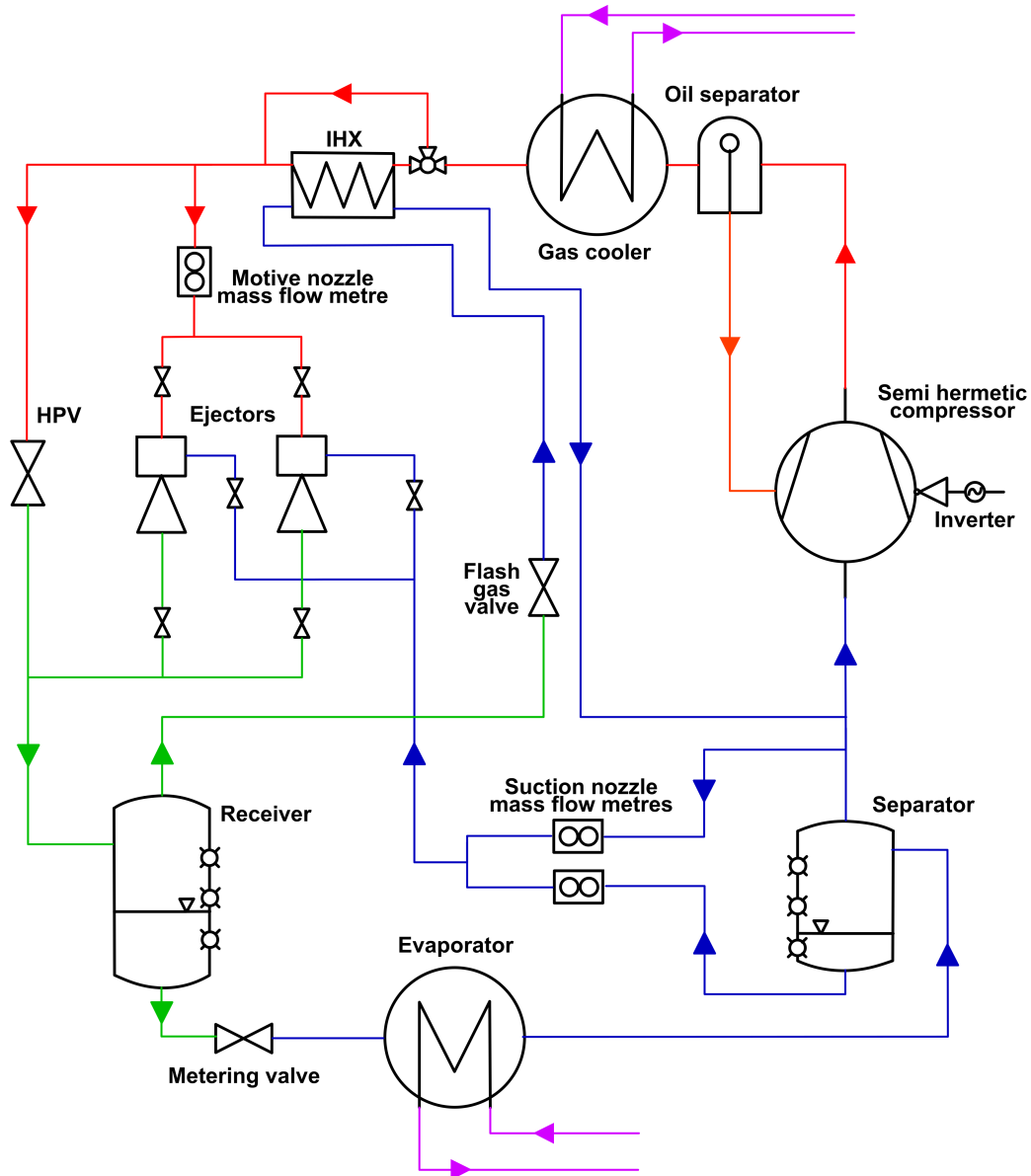


Figure 1: Overall scheme of the laboratory R744 test rig for ejector analysis

155 The dedicated temperature and pressure sensors were mounted approximately
 156 10 cm from the ejector ports. The location of the sensors is presented in Fig. 2,
 157 along with the investigated bypass ejector and the visualisation ejector for other
 158 research. Moreover, the motive and suction mass flow rates were measured using
 159 Coriolis-type mass flow metres manufactured by Endress+Hauser. The accuracy of
 160 the sensors and uncertainty of the output parameters were computed on the basis
 161 of NIST guidelines [39] and are listed in Table 1. The steady-state conditions of the
 162 considered operating point were assumed on the basis of 10-minute periods with a
 163 probing step of 5 seconds and the uncertainty values satisfying the levels listed in
 164 Table 1. Consequently, the period contained 120 probes per operating point, which
 165 allowed for the reliable evaluation of the ejector operation.

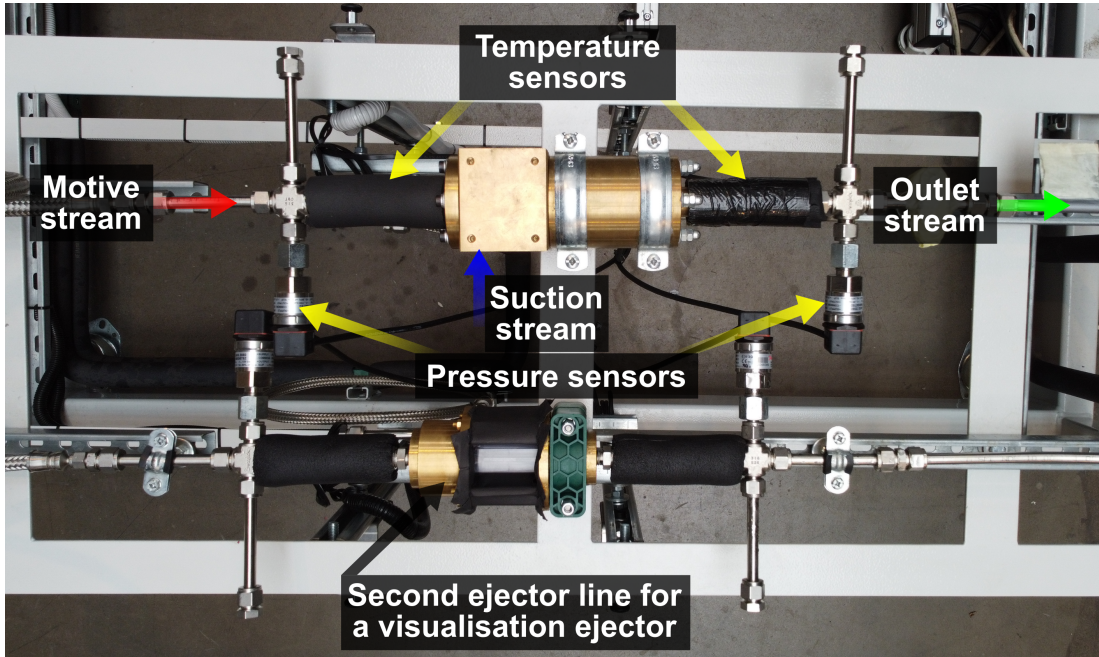


Figure 2: Localisation of the measurement sensors dedicated to the ejector inlet and outlet ports

Table 1: Accuracy and uncertainty of the measured parameters and the computed factors [39]

Parameter	Data source	Accuracy / Uncertainty Type A
Pressure	Danfoss AKS 32R Ratiometric pressure transmitter	$\pm 0.3\%$ / $\pm 0.35\text{bar}$ & $\pm 0.15\text{bar}$ (motive & suction)
Temperature	Danfoss AKS-21 PT1000	$\pm (0.3 + 0.005 \cdot \text{reading})$ / $\pm 0.05^\circ\text{C}$
Motive nozzle mass flow rate	Endress+Hauser Coriolis type flowmeter	$\pm 0.75\%$ / $\pm 3.0 \text{ kg}\cdot\text{h}^{-1}$
Suction nozzle mass flow rate	Endress+Hauser Coriolis type flowmeter	$\pm 0.75\%$ / $\pm 3.0 \text{ kg}\cdot\text{h}^{-1}$
Factor	Formulation	Uncertainty Type C [39]
Efficiency	formulation of Elbel and Hrnjak [18]	$\pm 1.0\%$
Mass Entrainment Ratio	ratio of the suction to the motive mass flow rate	± 0.01
Pressure lift	difference between the outlet and the suction port pressure	$\pm 0.15 \text{ bar}$

166 3. Design of the prototype ejector with a suction nozzle bypass

167 The ejector scheme is presented in Fig. 3, and the dimensions of the motive nozzle
 168 zle are presented in Table 2. The utilised ejector shape with an axial suction nozzle
 169 inlet was proposed by Banasiak et al. [40]. In this study, the tangential inlet of the
 170 suction nozzle was used. Namely, the ejector was dedicated for transcritical oper-
 171 ation of the system located in warm climate (approximately 36°C of ambient tem-
 172 perature) with evaporation temperatures at the level of -6°C and pressure-lift condi-
 173 tions at the level of 8 bar (i.e. corresponding temperature level demanded for air-
 174 conditioning). The geometry of the bypass duct and its positioning are described
 175 separately in subsections 3.1 and 3.2 for the sake of scheme clarity. Brass was used
 176 for the entire construction of the ejector ducts. The manufacturing tolerances were
 177 in the ranges of $\pm 0.01 \text{ mm}$, $\pm 0.05 \text{ mm}$ and $\pm 0^\circ 3'$ for the diameters, lengths and an-
 178 gles, respectively. The roughness of the internal surfaces was $R_a=0.5$. The entire con-
 179 struction was assembled using independent parts for the motive nozzle, the suction
 180 chamber, the mixing chamber, the movable diffuser and the outlet port. This means

181 that the prototype is based on several exchangeable pairs of movable diffuser and
 182 mixing chambers, which allows for the analysis of the various bypass positionings.

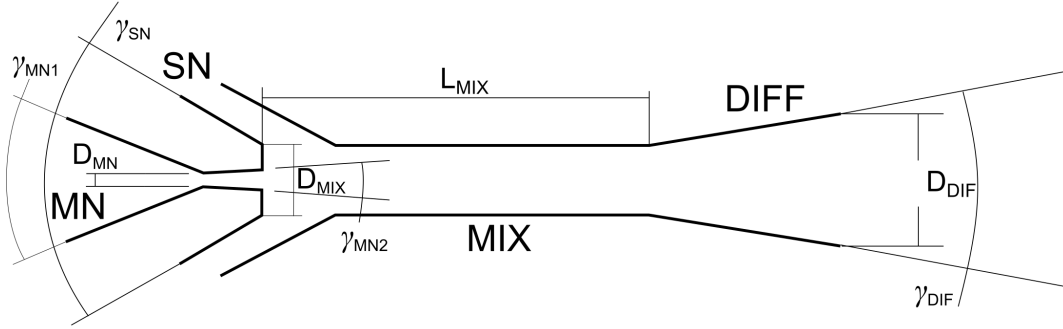


Figure 3: General scheme for a standard ejector geometry: motive nozzle (MN) section, suction nozzle (SN) section, mixing (MIX) section and diffuser (DIFF) section

Table 2: Geometrical parameters of the tested ejector motive nozzle

Parameter name (symbol)	Unit	Value
Throat diameter (D_{MN})	mm	1.4
Converging angle ($\gamma_{MN,1}$)	°	30.0
Diverging angle ($\gamma_{MN,2}$)	°	2.0
Suction angle (γ_{SN})	°	38.0
Mixing diameter (D_{MIX})	mm	6.0
Mixing length (L_{MIX})	mm	16.0
Diffuser diameter (D_{DIF})	mm	8.4
Diffuser angle (γ_{DIF})	°	5.0

183 3.1. Opening and closing mechanism of the bypass duct

184 According to the bypass regulation concept presented in [35], one tangential suc-
 185 tion port delivers R744 into two chambers: the standard one for the suction nozzle
 186 and the additional one for the bypass nozzle. The flow between these chambers
 187 is controlled by the bypass opening procedure. Hence, the bypass opening mecha-
 188 nism is located inside the ejector walls, as presented in Fig. 4 in the 2-D scheme (top)
 189 and 3-D view (bottom). Namely, the diffuser part (B2) is strictly correlated with the
 190 mixer part (B1) to provide the positioning and shape of the bypass duct. The prop-
 191 erly designed regulation screws hold the diffuser part (B2) in the closed position.
 192 The opening is based on the translation of the diffuser part (B2). To open the bypass
 193 duct, the screws need to be loosened, which results in translation of the diffuser part
 194 (B2). The translation is provided on the basis of the pressure difference between the
 195 suction chamber and ambient environment. A closing procedure is realised in the
 196 opposite way where the screws need to push the diffuser part. Both procedures have
 197 to be realised manually, which takes up to 5 seconds. Finally, the opening/closing of
 198 the bypass is conducted during the installation operation. Hence, the effect of the

199 bypass opening/closing is recorded *online* and in a continuous way for the same set
 200 of operating conditions.

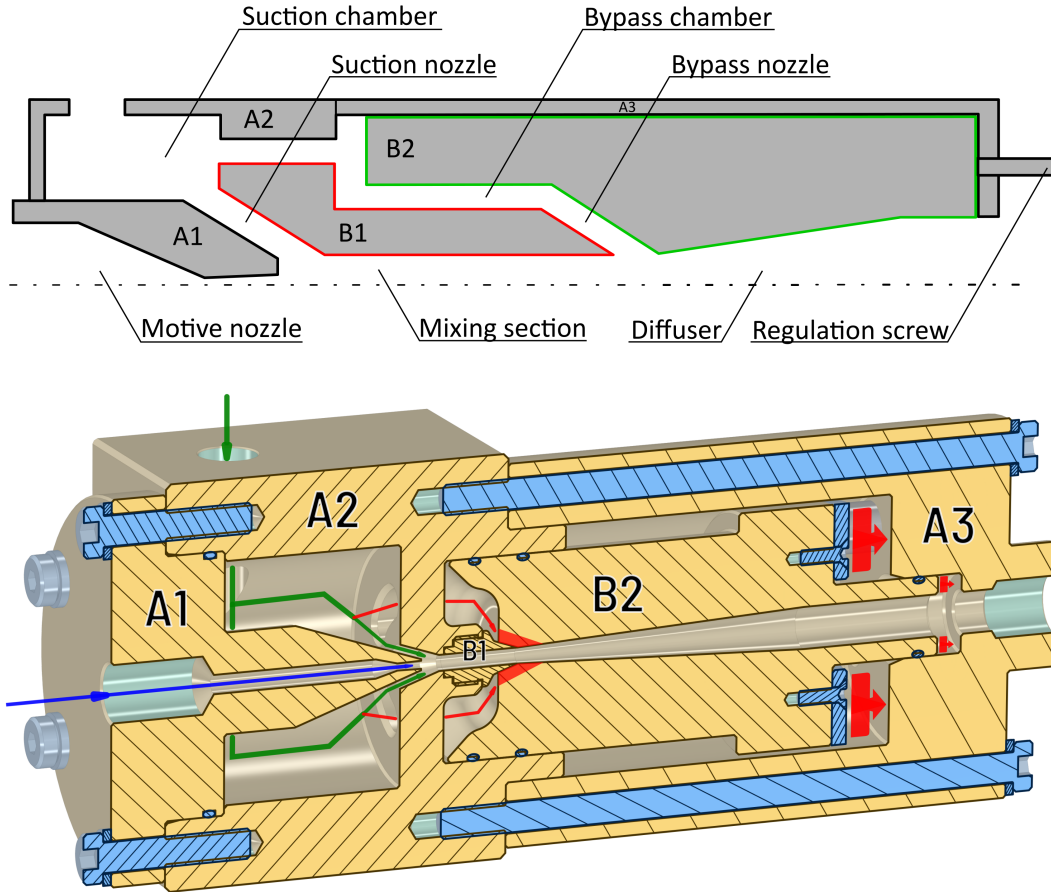


Figure 4: Cross section of the bypass ejector with stationary parts (A1, A2, A3 and B1) and a movable diffuser part (B2) using the 2-D scheme (top) and 3-D view (bottom)

201 **3.2. Positioning of the bypass duct**

202 A change in the bypass position along the ejector axis involves more time than
 203 the opening/closing procedure because it requires stoppage of the installation. Then,
 204 the ejector line is closed, and a pair of mixer and diffuser parts needs to be ex-
 205 changed. In the last step, the ejector line is vacuumed and refilled before further
 206 tests. The pairs of mixing and diffuser sections were designed on the basis of nu-
 207 merical assessment of the bypass ejector [35]. The connection point of the parts
 208 indicates the location of the bypass duct along the ejector axis. The scheme of the
 209 investigated bypass duct geometry is presented in Fig. 5 The angle β_{bps} was 19° ,
 210 while the bypass nozzle width d_{bps} was 1.6 mm. The shape of the bypass nozzle was
 211 adapted strictly from the preliminary 2D numerical analysis [35]. The conclusions
 212 from this preliminary analysis showed that the bypass nozzle shape was less impor-
 213 tant than the positioning of the bypass duct. Hence, the shape from the preliminary
 214 analysis was used in this study.

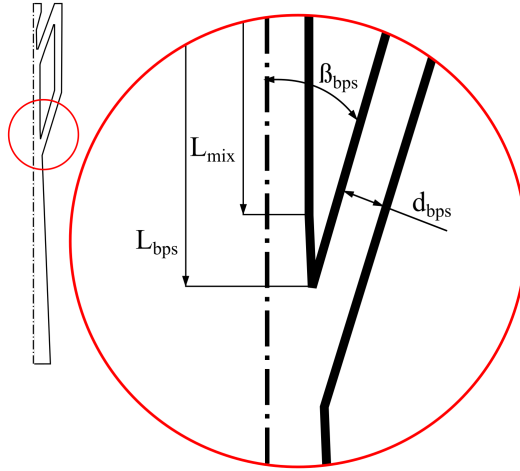


Figure 5: Scheme of the bypass duct with characteristic geometrical parameters

In general, the bypass duct takes the role of the second suction nozzle and has the same dimensions as the baseline suction nozzle based on the previous numerical assessment of the R744 bypass ejector performance [35]. The positioning of the bypass duct along the ejector axis was defined as follows:

$$\Psi = L_{bps}/L_{mix} \quad (1)$$

where Ψ represents the dimensionless position of the bypass duct along the ejector axis—for clarity, this nomenclature will be used in this study. L_{mix} is the length of the constant-area mixing section that was constant for the tested positions, and L_{bps} is the length along the ejector axis from the beginning of the constant-area mixing section to the location of the bypass nozzle introduction. Four pairs of mixing and diffuser parts were manufactured to investigate Ψ equal to 1.0, 1.1, 1.2 and 1.3. Hence, $\Psi=1.0$ indicates the bypass nozzle at the mixer and diffuser connection. A higher value of Ψ indicates that the bypass nozzle is introduced farther into the diffuser. The distance between each position is approximately 1.6 mm.

4. Methodology of the experimental evaluation of the bypass ejector performance

The investigated range of the motive nozzle conditions was based on the operating curve of the motive nozzle port defined by Gullo et al. [41] for the R744 refrigeration system equipped with a multi-ejector in a warm climate condition, such as a Mediterranean climate. Three operating conditions for the motive nozzle were used, as illustrated in Fig. 6, in groups A, B and C. The groups were represented by absolute pressure levels of 81 bar, 86 bar and 91 bar and corresponding temperatures of 33°C, 36°C and 39°C. Regarding the aforementioned operating curve dedicated to the refrigeration conditions, the pressure level of the suction nozzle port was chosen using typical refrigeration conditions, such as those of chillers. Hence, saturation pressures for evaporation temperatures of -10°C and -6°C were tested. The suction

235 temperature was controlled at 0°C and 4°C for the lower and higher evaporation tem-
 236 peratures, respectively. Each pair of motive and suction conditions was tested with
 237 5 levels of pressure lift from 4 bar to 11 bar. The range of the pressure lift was cor-
 238 related with the criterion of steady-state ejector operation, and the aim of this study
 239 focused on the bypass application potential. Namely, transcritical booster R744 sys-
 240 tems, such as those of local retail points or the food processing industry, should be
 241 applied.

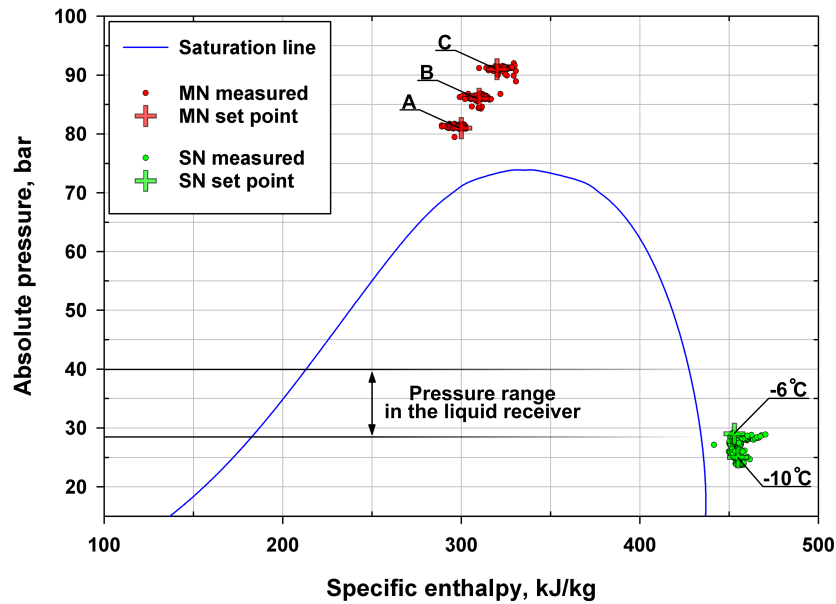


Figure 6: Investigated motive nozzle and suction nozzle conditions on the p-h diagram of R744

242 The aim of the study was to evaluate the performance of an ejector equipped
 243 with a bypass duct compared to that of a standard ejector. Nevertheless, some vari-
 244 ations in the operating conditions occurred, as presented in Fig. 6. The following
 245 measurement procedure was used to provide maximum similarity of the operating
 246 conditions during the comparison of ejector performances with open and closed
 247 bypass ducts. Namely, the tests for each point were started by an assembly of the
 248 proper bypass position in the ejector line. Next, a pair of motive conditions and
 249 evaporation temperatures was selected. Stabilisation of the test rig thermal condi-
 250 tions required approximately one hour. Finally, the measurement procedure for the
 251 given steady-state conditions was as follows:

- 252 1. Measurement of the ejector performance with a **closed bypass duct**
- 253 2. Bypass duct opening
- 254 3. Measurement of the ejector performance with an **open bypass duct**
- 255 4. Bypass duct closing

256 5. Increase in the pressure lift for the next measurement point

Consequently, each operating point with open bypass possessed an individual corresponding point with standard ejector operation for reliable comparison. Hence, the proposed procedure allowed for the evaluation of the ejector performance based on the efficiency formulation described by Elbel and Hrnjak [18]. The efficiency is defined using the absolute pressure, temperature and mass flow rates as follows:

$$\eta_{ej} = \frac{\dot{W}_{rec}}{\dot{W}_{rec,max}} = \chi \cdot \frac{h(p_{OUT}, s_{SN}) - h(p_{SN}, s_{SN})}{h(p_{MN}, s_{MN}) - h(p_{OUT}, s_{MN})} \quad (2)$$

257 where η_{ej} is the ejector efficiency, χ is the mass entrainment ratio, \dot{W} is the expansion
258 work rate, s is the specific entropy and the subscript OUT denotes the ejector
259 outlet. The formulation represents an expansion work rate recovered (subscript rec)
260 by the ejector with respect to the maximum possible expansion work rate recovery
261 potential (subscript rec, max).

262 5. Modelling approach for the numerical evaluation of the bypass ejector

263 The CFD techniques allowed for the analysis of the 3-D flow behaviour in the proto-
264 type bypass ejector. The numerical analysis was focused on the R744 flow evalua-
265 tion in the suction chambers what was not possible to realise on the basis of exper-
266 imental tests. Consequently, a discussion of potential further improvements of the
267 device was possible. The simulation of the two-phase carbon dioxide flow through
268 the bypass ejector was based on the data delivered from the dedicated laboratory
269 test rig discussed in Section 2. Hence, a direct validation process of the model output
270 was provided. Moreover, the numerical domain contained ducts up to the location
271 of the pressure sensors (see Fig. 2). This means that the pressure drop between the
272 measurement point and the ejector ports was included in the computational model.

273 5.1. Approach for the two-phase transonic simulation of the R744 flow

274 The two-phase flow simulation was based on the HEM presented in [37]. This
275 approach assumes thermodynamic and mechanical equilibrium between the two
276 phases flowing through the ejector ducts. Moreover, an instantaneous evaporation
277 process is assumed during the expansion process in the motive nozzle. Regarding
278 the range of operating conditions at the ejector motive port, this approach was ex-
279 tensively validated, resulting in a high motive nozzle mass flow prediction accuracy
280 of 10% [42]. The accuracy of the suction nozzle mass flow rate prediction was vali-
281 dated at 15% of the relative error [42]. In this study, the computed mass flow rates
282 at the motive and suction ports were validated against data from the laboratory test
283 rig.

284 5.2. Computational procedure

The 3-D domain of the prototype ejector was generated. The computational plat-
form *ejectorPL* described in the work of [42] generated numerical grids characterised

by negligible influence on the numerical solution. In this study, the same numerical grids were used for the main flow ducts of the ejector. The additional inlet ducts, suction and bypass chambers were generated separately and connected with the ejector. According to Section 2, the absolute pressure and temperature measured in closure of each ejector port were used as the boundary condition of the computational procedure. The simulation process was assumed to be finished when the relative residuals of each equation were below 10^{-4} , and the mass imbalance was lower than 0.1% of the suction nozzle mass flow rate. The total mass flow rate at each ejector port was compared to the measured values, and relative errors were computed as follows:

$$\delta_m = \frac{m_{SIM} - m_{EXP}}{m_{EXP}} \cdot 100\% \quad (3)$$

285 where δ_m is the relative error of the mass flow rate data obtained using the CFD
 286 model (subscript SIM) compared with the experimental (subscript EXP) data.

287 **6. Results and discussion**

288 *6.1. Bypass positioning and performance improvement*

289 In the experimental procedure, the tested ejector with the specified Ψ was ex-
 290 posed to all the operating conditions, while the tests of the closed and open bypass
 291 ducts were realised successively, as discussed in Section 4. Hence, the variation in
 292 the ejector performance with the closed and open bypass ducts could be described
 293 as a relative difference in the ejector efficiency, as presented in Fig. 7, for each of
 294 the four investigated Ψ values. The only variation between the operating conditions
 295 measured with closed and open bypass duct were due to natural instabilities of the
 296 control unit handling with the thermal inertia of the installation. However those in-
 297 stabilities were maintained below the level described in the Table 1 hence it could
 298 be evaluated as a negligible from the point of view of the ejector efficiency. The rel-
 299 ative difference of the efficiency presented in Fig. 7 is affected in vast majority by
 300 the change of the mass entrainment ratio, specifically by the change of the suction
 301 nozzle mass flow rate. However, despite that the authors decided to use efficiency
 302 values in order to include those minimal variations of the operating conditions re-
 303 garding full reliability of the results. In Fig. 7 series were marked according to the
 304 evaporation temperature, where -6°C and -10°C were described by circles and trian-
 305 gles, respectively. The conditions at the motive nozzle inlet are represented in Fig. 6
 306 by the letters A (85 bar and 33°C), B (86 bar and 36°C) and C (91 bar and 39°C) and
 307 additionally by the colours blue, green and red, respectively. Additionally, the pair
 308 of cases ($\Psi=1.1$) considered further in the numerical analysis in Section 7 is marked
 309 by a green dashed ring. These cases are selected on the basis of positioning analysis
 310 below and will be use to evaluate the R744 flow in the suction chambers which could
 311 not be done by experimental test.

312 A previous study showed that the implementation of the bypass duct into the
 313 R744 ejector achieved the most success in the case of the choked mixing section
 314 during unfavourable off-design operating conditions [35]. In this study, experimen-
 315 tal tests confirmed this statement. Namely, the higher the motive nozzle conditions

316 were, the more improvement of the ejector efficiency could be obtained after the
317 bypass opening. This is indicated by red-coloured markers positioned higher than
318 green-coloured markers. Simultaneously, the lower the suction pressure reflected
319 by the evaporation temperature was, the more improvement was available for the
320 tested bypass ejector compared to the standard design. Hence, the circular markers
321 (evaporation temperature set to -6°C) indicate lower values than the corresponding
322 triangular markers (evaporation temperature set to -10°C). Second, considering the
323 outlet port conditions, the pressure lift was an investigated parameter. Namely, the
324 improvement level of the ejector performance was reduced with the increasing pres-
325 sure lift value in all the tested Ψ . This relationship could be described as almost lin-
326 ear. Unfortunately the range of the pressure lift with potential benefits of the bypass
327 utilisation could be evaluated as a narrow in the case of the evaporation temperature
328 set to -6°C . Wider range (looking at the pressure lift value) was obtained in the case
329 of $T_0 = -10^{\circ}\text{C}$. The ejector efficiency improvement was observed for a maximum
330 pressure lift of almost 7.5 bar ($\Psi=1.1$), which should be considered a perspective for
331 further development of the bypass ejector. Namely, compared to the preliminary
332 numerical evaluation of the bypass idea, the efficiency of the ejector after bypass
333 opening could be improved only for a pressure lift of 4 bar [35].

334 The obtained efficiency improvement was correlated with Ψ in a more signifi-
335 cant way than that presented in the preliminary numerical analysis [35]. The high-
336 est efficiency increment was indicated with $\Psi=1.1$, when an improvement of 5% for
337 the pressure lift of 7 bar was obtained under the operating conditions of the mo-
338 tive conditions from group C and a lower evaporation temperature. Reducing the
339 pressure lift to 5 bar allowed for 34% of the efficiency improvement. Under the mo-
340 tive nozzle conditions from group A and an evaporation temperature of -10°C , it was
341 challenging to obtain steady operation with the lower pressure lift values for which
342 even higher improvements are expected. The reasons were related to the unstable
343 regulation of the metering valve and consequent fluctuation of the suction mass flow
344 rate. For $\Psi=1.2$, the maximum pressure lift correlated with the performance incre-
345 ment was reported at 6.5 bar. However, the lowest pressure lifts allowed for an 8
346 percentage points lower relative difference than in the case of $\Psi=1.1$. The maxi-
347 mum reported improvement was 37% in the case of $\Psi=1.3$ for a pressure lift of 5 bar.
348 Nevertheless, the latter position did not result in improved ejector operation with a
349 pressure lift higher than 6 bar. The bypass positioned directly at the connection of
350 the mixing section and diffuser ($\Psi=1.0$) provided the lowest improvement values of
351 approx. 10%, demanding additionally even lower pressure lifts.

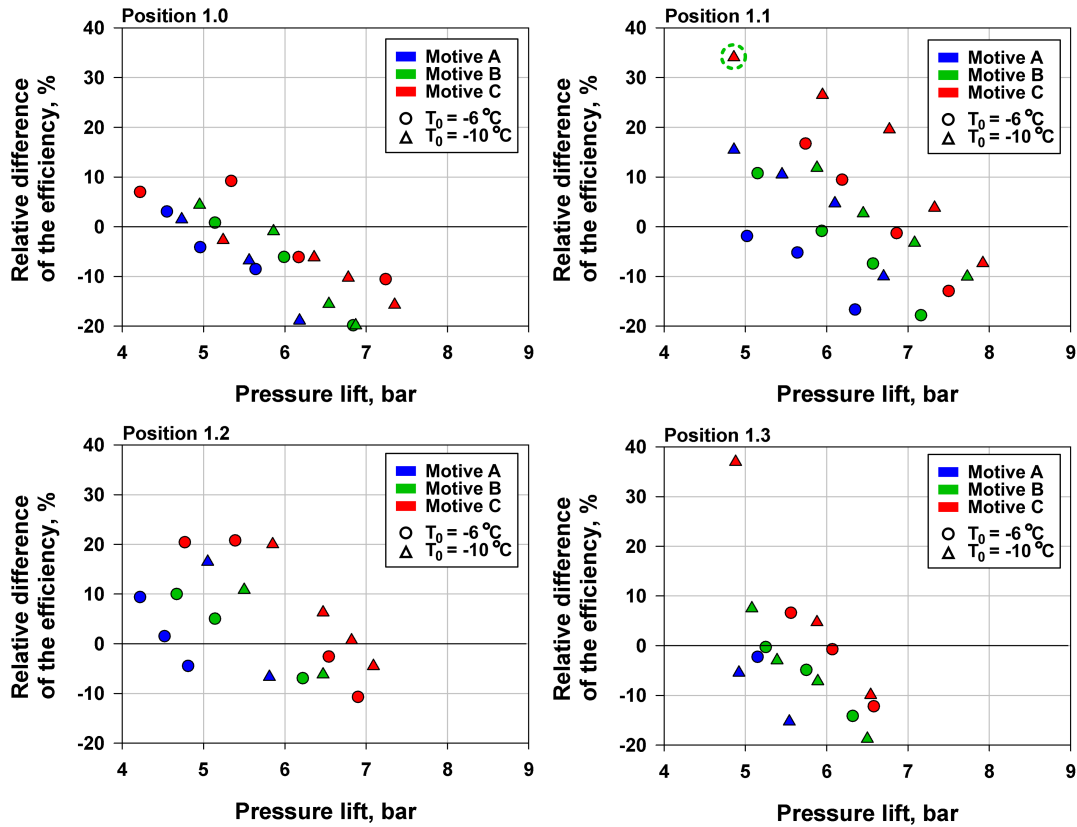


Figure 7: Relative difference in the ejector efficiency for four bypass opening positions (the case analysed numerically is indicated by a green ring)

352 6.2. Efficiency of the ejector with proper control of the bypass duct opening

353 This subsection presents the efficiency characteristics of the bypass ejector, including
 354 proper control of the bypass duct opening. The measurement results obtained for the
 355 higher suction and lower suction pressures are presented in Fig. 8 and Fig. 9, respectively.
 356 The cases in which the bypass duct should be closed for higher efficiency are marked by
 357 crosses. Situations in which the bypass duct should be open are marked by circles (for
 358 higher suction pressure) and triangles (for lower suction pressure). Similar to Fig. 7,
 359 the motive nozzle conditions were organised by blue (A), green (B) and red (C) colours
 360 according to the increasing pressure and temperature values. These results were divided
 361 into four graphs corresponding to Ψ .

362 The control strategy of the bypass duct opening should be aimed at the highest possible
 363 ejector efficiency. Consequently, the efficiency of the bypass ejector is higher than that
 364 of the standard design in the area of the lower pressure lifts. Considering the standard
 365 ejector efficiency, this factor takes the lowest values at the aforementioned low-pressure
 366 lift operation. Lifting the efficiency in this region provides a substantially flatter
 367 character to the efficiency distribution with varying pressure lift values. For an
 368 evaporation temperature of -6°C , the minimum registered efficiency was increased from
 369 18.5% to 19.8%. However, when the suction pressure

370 corresponded to a saturation pressure of -10°C , the minimum registered efficiency
 371 was improved more substantially from 14.4% to 19.3%. Moreover, in addition to the
 372 improved efficiency, more stability of the ejector performance was a benefit of the
 373 bypass introduction. Namely, considering all measurement points with the evapora-
 374 tion temperature set to -6°C , the average bypass ejector efficiency with a properly
 375 controlled opening was 28.4% with a standard deviation of 2.2 percentage point. The
 376 measurement results at the lower suction pressure provided an average bypass ejec-
 377 tor efficiency of 26.6% with a standard deviation of 1.8 percentage point. Finally,
 378 the introduced bypass raised the efficiencies in those regions which were out of the
 379 design conditions – from the point of view of the pressure lift. Consequently, the
 380 prototype ejector could cover a wider range of operating conditions with high and
 381 more uniform efficiency values than those of the standard construction.

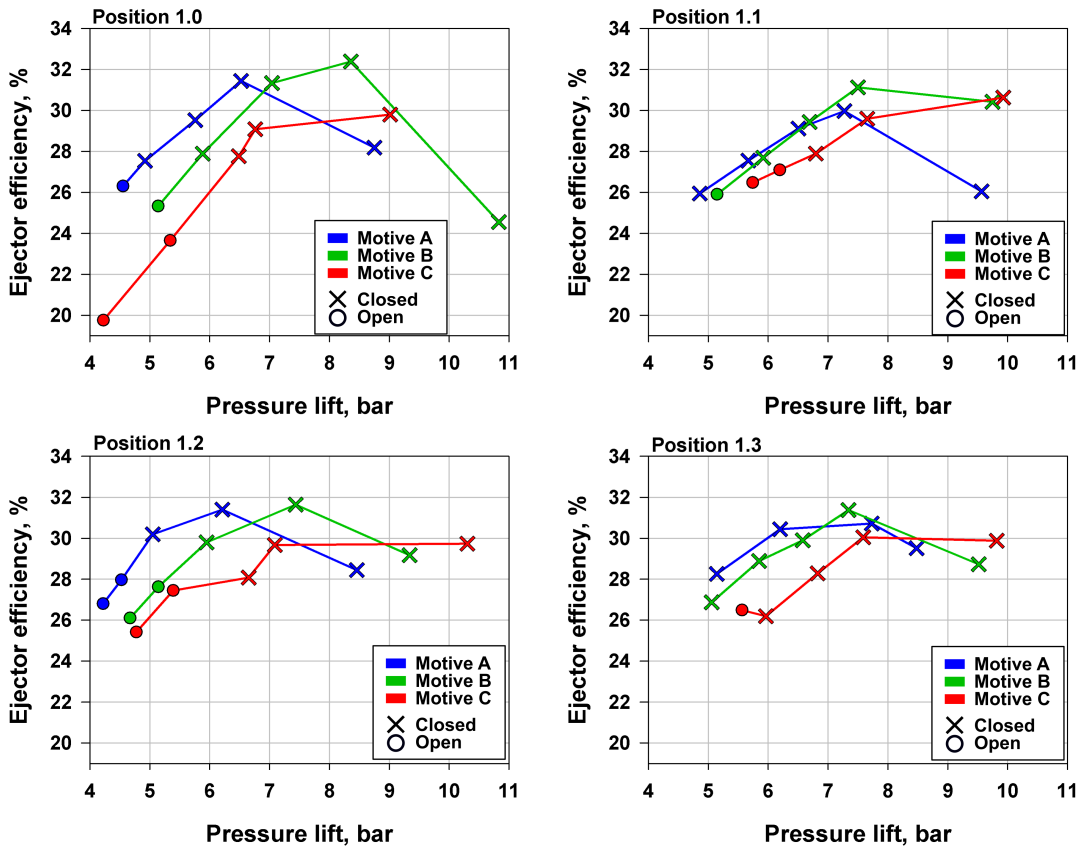


Figure 8: Compilation of the highest ejector efficiency with proper opening of the bypass at an evaporation temperature set to -6°C for four bypass opening positions

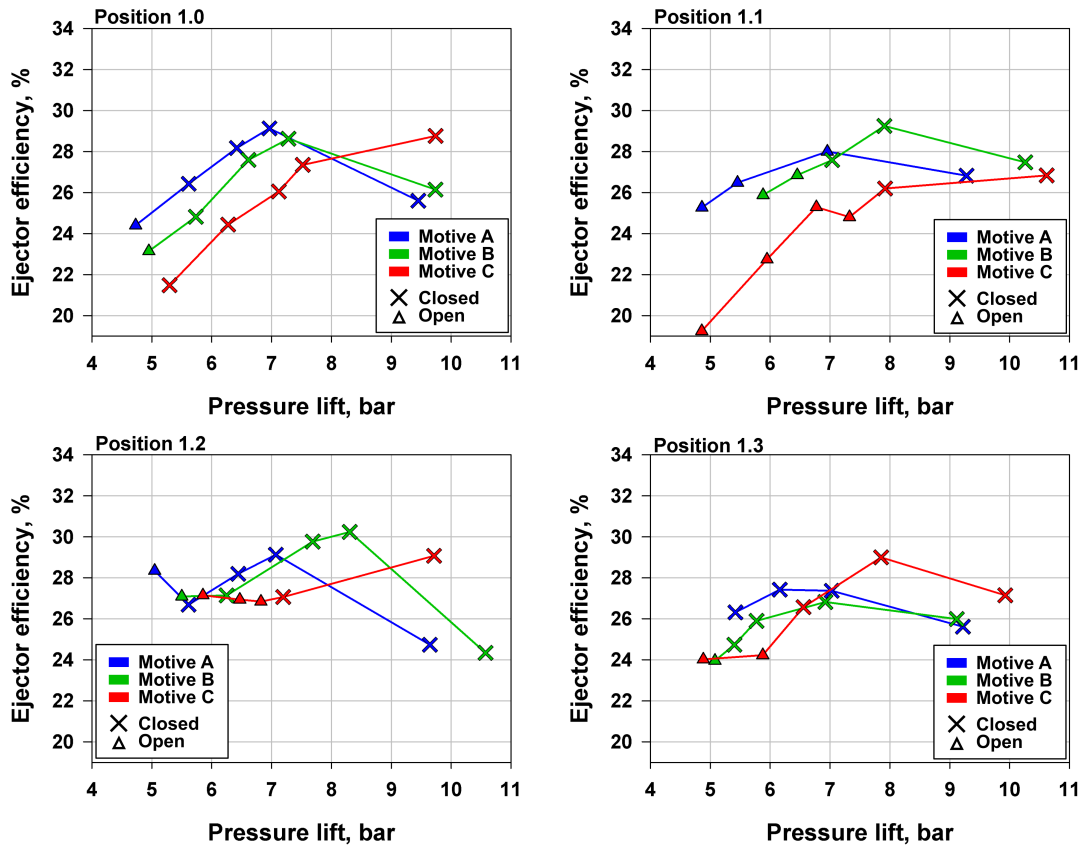


Figure 9: Compilation of the highest ejector efficiency with proper opening of the bypass at an evaporation temperature set to -10°C for four bypass opening positions

6.3. Influence of the motive and suction temperatures on the bypass performance

An analysis of the R744 temperature influence on the bypass potential was provided for the motive and suction ports separately. The aim was to check the sensitivity and then the potential instabilities in the ejector work in function of the motive and the suction temperatures which would be delivered by the various thermal states of the system, especially variable conditions of the heat rejection in the gas cooler and different heat load of the evaporator. For each port, three temperature levels were investigated, while other operating conditions were maintained at the same level, as presented in Table 3. Two positions of the bypass opening resulted in six operating points per motive and suction port analysis. The motive port analysis was performed at 29°C , 33°C and 37°C and a pressure lift of 4.5 bar. The influence of the temperature at the suction port on ejector performance was assumed to be less significant. Hence, a pressure lift of 6.0 bar (higher efficiency) was used to clearly present the influence of the suction temperature. Moreover, a higher step of 8 K between the points was used starting from the suction temperature of 9°C . The suction temperatures provide deterioration of the R744 refrigeration system coefficient of performance; however, they were used to evaluate its influence on the bypass ejector.

Table 3: Operating conditions for the ejector efficiency analysis at different motive and suction nozzle temperatures

No.	Bypass	Motive conditions		Suction conditions		Outlet conditions	
		bar	°C	bar	°C	bar	°C
M1	closed	90.6	29.6	28.1	18.5	32.4	-1.3
	open	90.9	28.8	27.9	19.0	32.6	-1.0
M2	closed	91.0	33.7	28.0	18.3	32.8	-0.7
	open	91.2	33.9	27.9	18.7	33.0	-0.4
M3	closed	91.3	37.5	28.1	20.1	32.8	-0.7
	open	91.1	37.6	27.9	20.5	33.1	-0.2
S1	closed	91.0	38.3	28.0	9.6	34.0	0.5
	open	91.2	38.6	28.0	8.8	34.2	0.8
S2	closed	91.2	38.7	28.2	16.6	34.1	0.7
	open	91.2	39.1	28.2	16.9	34.2	0.8
S3	closed	90.9	37.9	28.2	24.1	34.1	0.9
	open	91.1	37.9	28.3	25.3	34.1	0.8

400 The output of the R744 temperature analysis is presented in Fig. 10 for the motive
401 nozzle (left graph) and for the suction nozzle (right graph). The ejector efficiency val-
402 ues with closed (black bars) and open (red bars) were compared using relative differ-
403 ences (green bars). The motive nozzle analysis revealed similar efficiency improve-
404 ments for M1 and M2. Moreover, the bypass ejector efficiency increased from M2 to
405 M3, while the standard ejector design obtained efficiency (black bars) at 20.0%. The
406 influence of the motive nozzle was substantial, as presented in the aforementioned
407 analysis and in Fig. 7. However, the improvement potential of the bypass ejector
408 was at constant and high levels despite the changes in the motive nozzle tempera-
409 ture. Consequently, the aforementioned similarity between the efficiency at condi-
410 tions M2 and M3 provided information that starting from approximately 33°C the
411 ejector operates with the uniform efficiency at the level of 20% despite unfavourable
412 low pressure-lift value at the level of 4 bar. On the other hand, this region of the
413 operating conditions could be improved in the most significant manner what was
414 represented by the highest green bar indicating over 30% of the relative difference of
415 ejector efficiency comparing closed and opened bypass.

416 The analysis of the suction nozzle temperature presented in the rightward graph
417 of Fig. 10 showed the moderate importance of this parameter. The opening of the
418 bypass duct resulted in a similar improvement of approx. 8.5% at lower suction tem-
419 peratures S1 and S2. The highest suction temperature resulted in a lowered (-7.0%)
420 efficiency of the bypass ejector. Nevertheless, despite large temperature increments,
421 the ejector performance remained in the range of 26.0% to 28.0%. Consequently, the
422 influence of the suction temperature should be considered negligible because of the
423 similar prototype and standard design efficiencies in all examined cases.

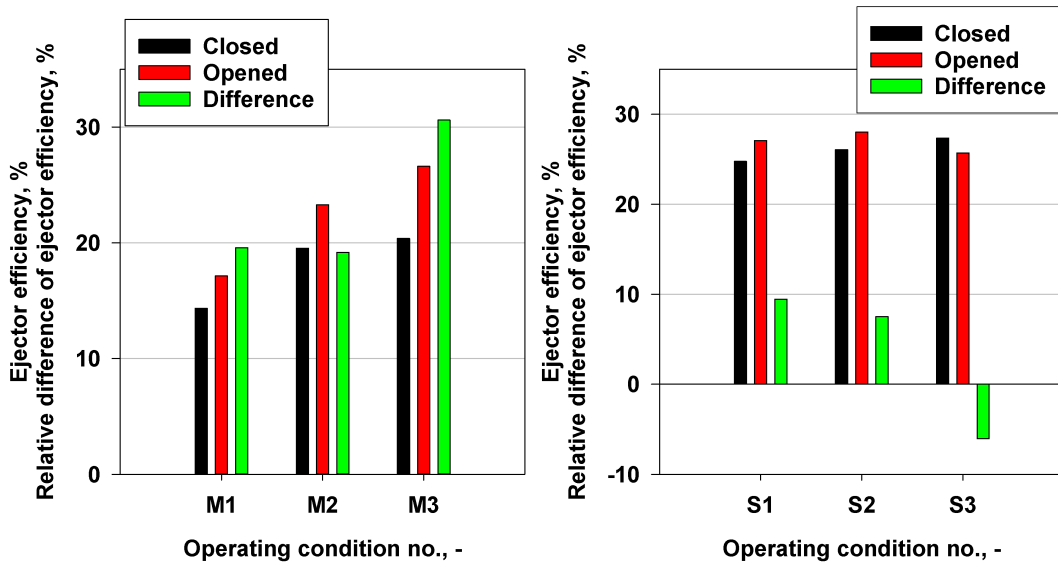


Figure 10: Influence of the motive nozzle temperature (left) and the suction nozzle temperature (right) on closed (black bars) and open (red bars) bypass ducts and the resulting relative difference (green bars)

424 6.4. Analysis of the opening degree of the bypass duct

425 The effect of the bypass opening degree on ejector performance was investigated
 426 for $\Psi=1.1$. The opening degree of the bypass duct in the prototype ejector can un-
 427 dergo stepless modulation from 0.0% to 100.0% and was formulated as the ratio of
 428 actual Part B translation (see Fig. 4) to the maximum translation of 5.0 mm. For this
 429 analysis, operating condition C-10 and a pressure lift of 5.0 bar were selected. Hence,
 430 it provided the maximum improvement for the selected bypass position.

431 The results obtained from the opening degree analysis are presented in Fig. 11.
 432 The results showed that full available improvement for each case was obtained with
 433 approximately 7.0% to 10.0% of the opening degree. In this opening, the width of the
 434 bypass nozzle d_{bps} was approximately 0.16 mm (see Fig. 5). It might be interpreted
 435 that the influence of the pressure introduced at the beginning of the diffuser could
 436 be a crucial factor behind the bypass improvement. Resulting mixing pressure was
 437 increased and provided reduction of the choked mixer phenomenon (further dis-
 438 cussed in Section 7.3 and Fig 12). It could be stated that the aforementioned width
 439 of the bypass duct at 10.0% of the opening was large enough to provide losses-free
 440 flow through this secondary bypass nozzle. On the other hand, this could be a con-
 441 sequence of the fact that this opening was sufficiently large to provide a full available
 442 suction stream through the bypass duct. Moreover, in this situation, the ratio of the
 443 suction nozzle and bypass duct mass flow rates should be considered a high value
 444 because of the low value of the latter component. Hence, the geometry of the suc-
 445 tion nozzle and the mixer chamber could still be described as a substantial design
 446 feature for the ejector performance even in the case of bypass duct utilisation. The
 447 measurement points of the higher opening degree were characterised by variations
 448 of $\pm 1.5\%$ of the ejector efficiency, which should be related rather to the variations

449 of the operating conditions. Finally, in the analysed ejector design, a large buffer is
 450 available for the additional suction stream flow through the bypass duct at evapora-
 451 tion temperatures lower than the investigated values.

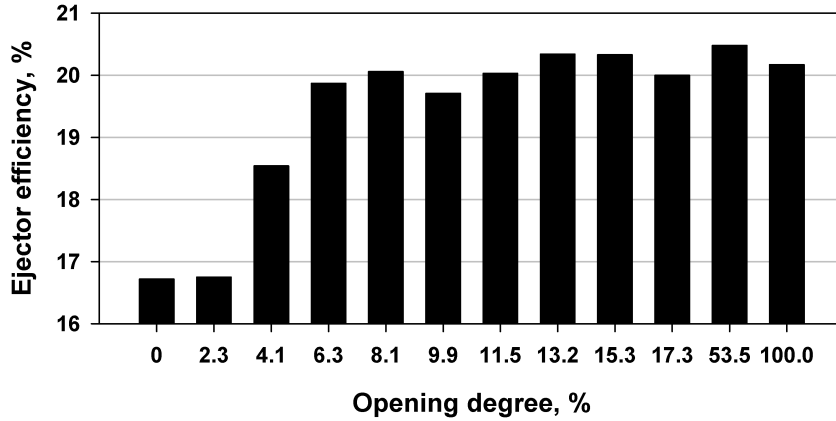


Figure 11: Ejector efficiency for different opening degrees of the bypass duct

452 7. Numerical analysis of the ejector operation with closed and open bypass ducts

453 7.1. Validation results

454 The numerical analysis in this study aimed for the evaluation of the R744 flow
 455 in the suction chambers of the prototype ejector. For this purpose only one operat-
 456 ing condition was required hence this was a range of necessary validation process.
 457 Namely, two cases for that operating condition were validated – one case with open
 458 and one case with closed bypass duct. The performance of the suction chambers
 459 during the ejector operation with open and closed bypasses was numerically anal-
 460 ysed for the best case from $\Psi=1.1$. The analysis would reveal a potential for further
 461 improvement of the suction phenomena in the case of already high increment of the
 462 ejector efficiency. Namely, the lowest pressure lift in operating condition C-10 was
 463 selected (marked by the green ring in Fig. 7) for the numerical analysis provided in
 464 this Section.

465 First, the main output data from the simulation were validated against the exper-
 466 imental data from the test rig, as presented in Table 4. Regarding the simulation of
 467 the closed bypass scenario, the accuracy of the mass flow rate predictions (as defined
 468 in Eq. 5.2) was -2.3% and 11.3% for the motive and suction ports, respectively. In the
 469 case of the open bypass duct, the aforementioned accuracy was slightly higher. In
 470 both cases, the stream at the suction port was overpredicted, and the level of these
 471 inaccuracies was similar. Hence, the validation results were considered sufficiently
 472 good for further analysis in which the ejector operations with the open and closed
 473 bypasses were compared.

Table 4: Validation data of the ejector simulation and results of the suction stream analysis

Validation parameter		Bypass closed	Bypass open
Motive port	simulation, kg/s	0.0695	0.0691
	experiment, kg/s	0.0712	0.0695
	$\delta_m, \%$	-2.3	-0.6
Suction port	simulation, kg/s	0.0318	0.0362
	experiment, kg/s	0.0286	0.0337
	$\delta_m, \%$	11.3	7.3

474 *7.2. Analysis of the suction stream distribution*

475 The results of the analysis based on the suction stream distribution between the
476 suction nozzle and the bypass nozzle are presented in Table 5. Such data could not
477 be obtained from the laboratory test rig due to the closed construction of the ejec-
478 tor. Hence, measurement of the mass flow rate is available only for the total suction
479 stream. A substantial amount of the total sucked R744 was directed to the suction
480 nozzle. The bypassed stream was on the level of approximately 15.0% of the total
481 suction stream computed at the suction port. The low ratio between the bypassed
482 and total suction streams could be found as a reflection of the data presented during
483 the opening degree analysis in which a small opening (see Fig. 11) was sufficient for a
484 full improvement of the ejector performance. Moreover, the simulation showed that
485 the change in the suction nozzle mass flow rate was equal to -3.6%, which is negli-
486 gible compared the cases with closed and open bypass duct operations. Hence, the
487 vast majority of the additional R744 sucked after the bypass opening was directed
488 to the bypass nozzle. Namely, the overall change in the suction stream was 13.9% in
489 the case of the simulation output, while a slightly higher (by 4.3% percentage points)
490 change was measured at the test rig.

Table 5: Results of the suction stream analysis

Parameter		Value
Simulation	suction nozzle, kg/s	0.0306
	bypass nozzle, kg/s	0.0056
	suction nozzle / total suction stream, %	84.6
	bypass nozzle / total suction stream, %	15.4
	suction nozzle stream change, %	-3.6
	total suction stream change, %	13.9
Experimental	total suction stream change, %	18.2

491 *7.3. Discussion of the possible further shape improvements of the bypass ducts*

492 The absolute pressure distribution in the ejector was obtained from the numeri-
493 cal simulation for both variants. These results are presented in Fig. 12 in the form of
494 the cross-sectional field (top) and the corresponding distribution along the ejector
495 axis (bottom). The maximum value at the field range and vertical axis of the graph
496 was set to 3.0 MPa to clearly illustrate the absolute pressure distribution in the mixer
497 region.

498 Cross-section A indicates the beginning of the mixing section where choked flow
 499 and Prandtl-Meyer shock-train flow are visible. The bypass opening resulted in in-
 500 creased absolute pressure in the aforementioned region. In the case of the field data,
 501 the near-wall region at the beginning of the mixer changed colour from azure to
 502 green, i.e., by approximately 0.55 MPa. The pressure profiles in this region differ by
 503 0.17 MPa. Considering the ejector axis, the pressure difference is higher at the end of
 504 the mixing section marked by cross-section B (0.44 MPa) and at the beginning of the
 505 bypass chamber indicated by cross-section C. Cross-section D indicates a uniform
 506 pressure distribution across the duct and the same pressure values in the ejector
 507 axis for both cases. The pressure distribution in the diffuser is similar; however, the
 508 red circles of the open bypass case are located slightly higher than the markers of
 509 the closed bypass. The absolute pressure is identical at the bypass nozzle and suc-
 510 sion nozzle inlets as indicated by the yellow colour in both areas. Consequently, the
 511 channel that connects these regions provides a negligible pressure drop. Finally, the
 512 mechanism of ejector unchoking could be related to the flattened pressure profile
 513 and higher pressure level at the end of the mixing section after bypass duct opening.

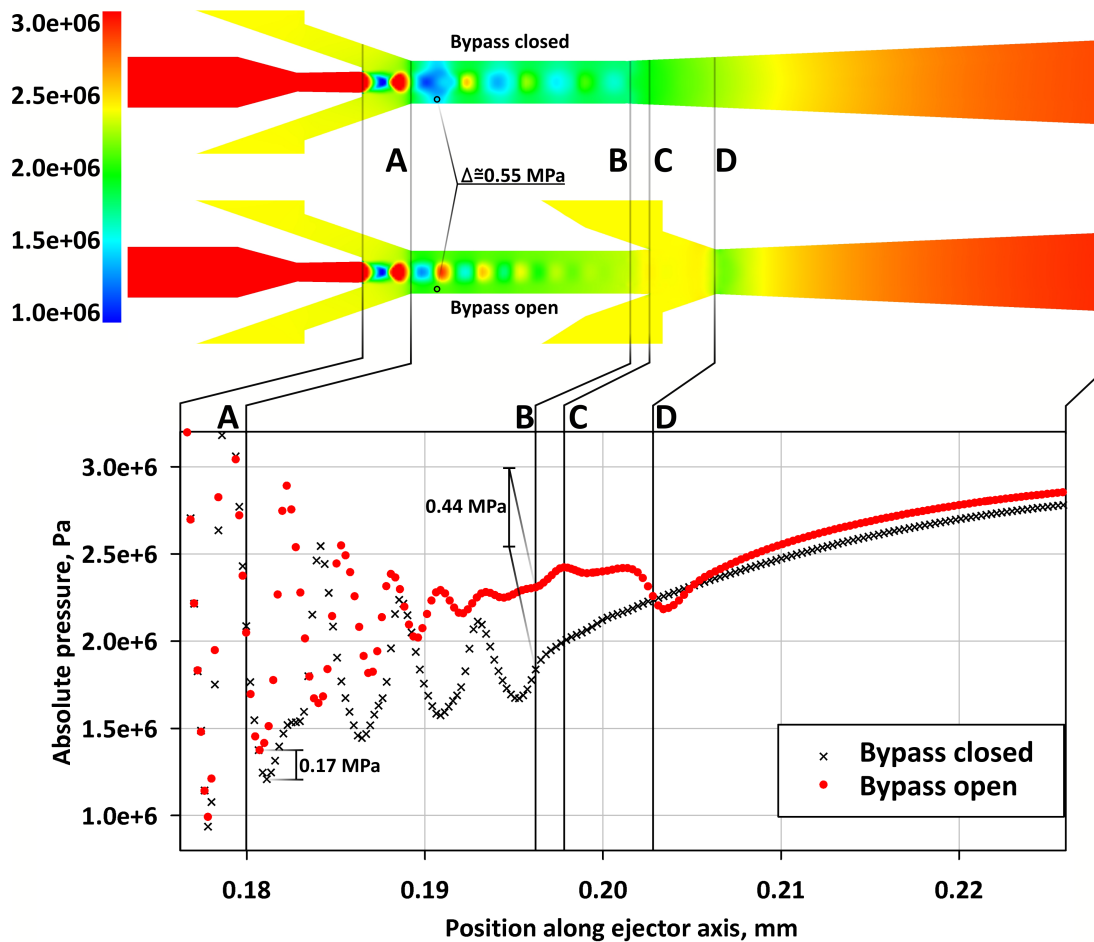


Figure 12: Comparison of the absolute pressure (Pa) fields (top) and profiles (bottom) of the ejector operation with the closed (black crosses) and open (red circles) bypass ducts

514 Additional analysis of the suction ducts is provided based on Fig. 13. Fig. 13
515 contains a composition of the three planes (schematically presented on the right-
516 hand side) with velocity magnitude contours and path lines. In this figure, the veloc-
517 ity range was reduced to 20 m/s to analyse the behaviour of the flow in the suction
518 chamber and the bypass chamber. Two of the planes (top) were placed across ducts
519 that connect the suction and the bypass chamber. The third plane was rotated by
520 45°(bottom). The velocity magnitude in the suction chamber is 15 m/s, while sig-
521 nificantly slower flow was observed in the bypass chamber. The path lines in the
522 suction chamber show uniform swirling flow around the ejector axis. The strong
523 vortex and long pathlines from the suction port to the suction nozzle were a result
524 of the bypass duct design. Namely, the radius of the suction chamber was enlarged
525 (comparing to standard construction) in order to contain the opening mechanism
526 (moving cylinder correlated with proper mixing part, please see Fig. 4) and the ducts
527 which connect the suction chamber and the bypass chamber. Scheme located on
528 the right-hand side of the Fig. 13 presents a cross-section view of the aforemen-
529 tioned connecting ducts (4 shapes similar to a cashew nut). These ducts needed to
530 be located around the part which provided mixing zone hence it was at the larger di-
531 ameter. This enlargement resulted in the final diameter of the suction chamber. On
532 the other hand, flow in the bypass chamber is more turbulent, and rotational move-
533 ment around the ejector axis is substantially less visible. Moreover, the distance for
534 effective acceleration of the fluid should be longer in the case of the bypass nozzle.
535 The resulting distribution (see Table 5) of the sucked R744 could be affected by the
536 described differences between the flow in the suction and bypass chambers. The
537 aforementioned differences in the flow pattern in the suction and bypass chambers
538 could be correlated with the cross-sectional shapes of the inlet ducts that connect
539 both chambers. Finally, the introduction of additional swirl motion in the bypass
540 chamber could bring further improvement.

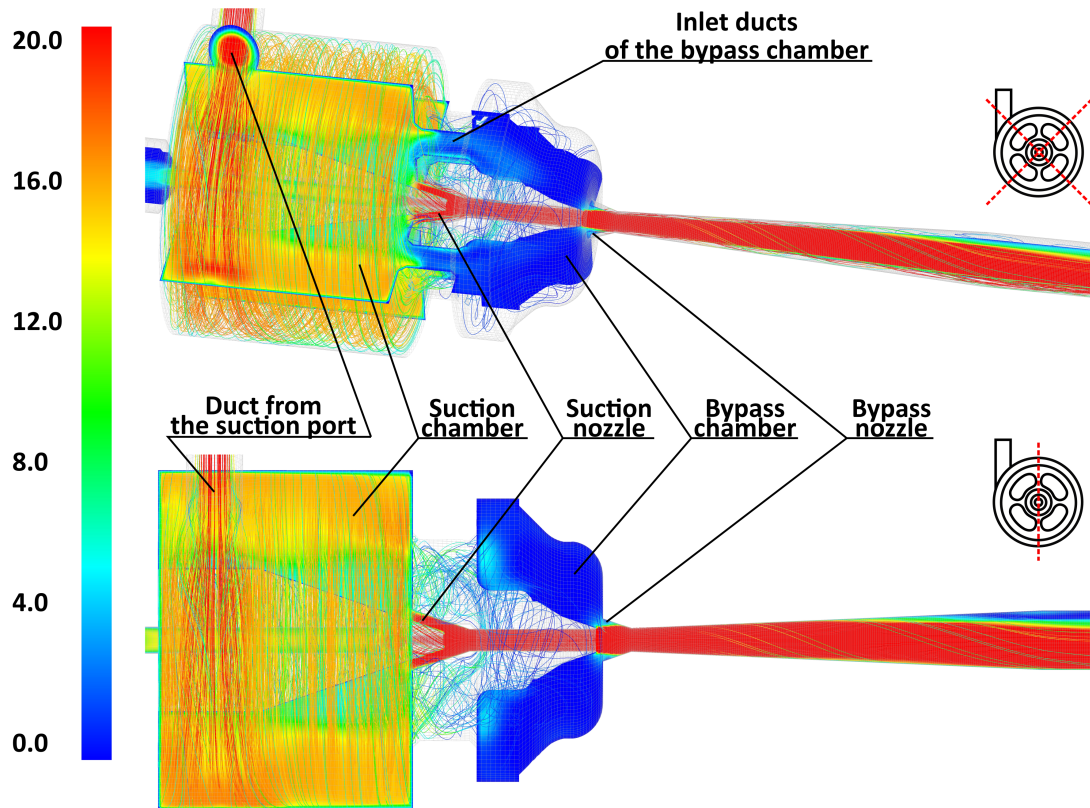


Figure 13: Distribution of the velocity magnitude (m/s) and path lines of the velocity magnitude (m/s) in the suction and bypass chambers in isometric view (top) and side view (bottom)

541 8. Conclusions and further work

542 The prototype R744 ejector with the bypass duct of the suction nozzle was de-
 543 signed and manufactured based on baseline ejector design prepared for high-pressure
 544 lift conditions. Experimental investigation at a dedicated R744 lab installation was
 545 conducted for four bypass nozzle positions along the ejector axis and variable open-
 546 ing degrees. Three sets of high-pressure side conditions characteristic of warm cli-
 547 mates were used along with two evaporation temperatures correlated with chilling
 548 and refrigeration conditions.

549 The improvement of the ejector efficiency after the bypass opening was strong
 550 and almost linear as a function of the pressure lift. The bypass position described by
 551 $\Psi=1.1$ was evaluated as the best case in the investigated range. The ejector with the
 552 bypass duct open in $\Psi=1.1$ obtained an efficiency improvement for the pressure lift
 553 of up to 7 bar. The maximum measured efficiency increment was 37%. The ejector
 554 performance was improved in the off-design conditions characterised by the low-
 555 pressure lift values and the lowest level of the baseline efficiency. Consequently, the
 556 efficiency curve presented higher and more uniform values in a full range of ejec-
 557 tor operations. Regarding mobile or integrated HVAC&R applications in which high
 558 variation of the cooling load is expected, the bypass ejector should reduce instabil-
 559 ities as a result of a more uniform performance in less favourable operating condi-

560 tions. Moreover, manufacturing one ejector for a wide range of pressure lift opera-
561 tions could bring economic advantages over two different ejectors designed for low-
562 and high-pressure lift operating ranges.

563 The pressure and temperature conditions at the motive port of the bypass ejec-
564 tor were more influential than the suction conditions. This feature should be consid-
565 ered an advantage of the examined idea, considering that R744 systems are standard
566 applications in warmer climates. Moreover, the results showed that the application
567 of the bypass ejector should not be affected by the suction temperature. From the
568 point of view of the ejector, this temperature is usually correlated with the tempera-
569 ture after the cooling load component.

570 The bypass duct in the prototype ejector allowed for the fluent regulation of the
571 bypass opening. The mechanism used the pressure difference between the suction
572 port and ambient environment. Approximately 10% of the opening degree allowed
573 for the full available improvement of the ejector efficiency at the considered operat-
574 ing point. The importance of the bypass suction chamber length was evaluated as
575 less significant because further opening did not influence the ejector performance.
576 Full available improvement after the bypass opening required a small range of part
577 B (see Fig. 4) translation of only 0.5 mm. The mechanism of bypass duct opening
578 should be characterised as less demanding in the design process. Namely, easy im-
579 plementation of the control system is possible using a simple on/off electromagnetic
580 valve. Hence, a high potential could be indicated for application in the current tran-
581 scritical booster R744 systems, such as in local retail points or the food processing
582 industry.

583 Numerical analysis of the ejector with closed and open bypass ducts delivered
584 additional data on the prototype device. First, the pressure distribution in the case
585 of the closed bypass duct resulted in unfavourable choking of the mixing section.
586 Reduction of the shock train generation after bypass duct opening provided higher
587 pressure values and unblocked the flow in the mixing cross section. Similar conclu-
588 sions were reached in a previous numerical study [35].

589 Simulations of the R744 flow in the suction and bypass chambers revealed a po-
590 tential for further improvement. Disordered and highly turbulent flow in the latter
591 chamber was correlated with the cross-sectional shape of the channels, which led to
592 flow from the suction chamber. This region could be the basis for an optimisation
593 study of the investigated device.

594 Further work could include an enhanced analysis of ejectors designed for sys-
595 tems characterised by higher cooling capacities and the development of control li-
596 braries based on performance mapping. The automatic mechanism of the bypass
597 duct could be connected with the unit's control system as a demonstrative version
598 of the bypass ejector characterised by a higher technology readiness level. Another
599 solution should be considered for the ejector designed for the low-pressure lift to
600 improve the performance under the high-pressure lift conditions.

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