

Thermoacoustic modes of intrinsic and acoustic origin and their interplay with exceptional points

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Abstract

We propose a general classification of all the modes of a given thermoacoustic system into two sets: one of acoustic origin and one of intrinsic thermoacoustic (ITA) origin. To do this, the definition of intrinsic modes, traditionally based on anechoic boundary conditions, is reformulated in terms of the gain n of the Flame Transfer Function (FTF). As a consequence of this classification, we show how theoretical results for the estimation of all thermoacoustic modes can be derived in the limit $n \rightarrow 0$, for both axial and annular combustors, independent of the acoustic boundary conditions. Starting from this limit and using standard continuation methods while increasing n , all the eigenvalues of interest in a given domain in the frequency space can be identified. We also discuss how thermoacoustic modes of acoustic and ITA origin can interact, and in some cases coalesce generating exceptional points (EPs).

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Although all EPs found have negative growth rates, in their vicinity thermoacoustic eigenmodes have very large sensitivities and exhibit strong mode veering. We demonstrate how, in some cases, mode veering is responsible for the occurrence of thermoacoustic instabilities, and propose a numerical method to identify EPs. All the theoretical results are numerically verified using two generic thermoacoustic configurations.

Keywords: Thermoacoustics, Intrinsic modes, Exceptional point, Combustion instabilities

1. Introduction

Intrinsic thermoacoustic (ITA) instabilities were first recognized in [1, 2], where it was shown that even a thermoacoustic system with anechoic boundary conditions can exhibit thermoacoustic instabilities. Analytical models of ITA instabilities were developed in [3, 4], which allowed for their connection to an intrinsic feedback loop that does not require acoustic reflections at boundaries. Using an $n-\tau$ model, it was shown that the ITA resonance frequency can be calculated from the time delay of the acoustic flame response only, and that ITA resonance frequencies are directly related to the peaks of the elements of the scattering matrix [5, 6].

Direct numerical simulations of anechoic systems that exhibited ITA instabilities [7, 8], together with some experimental evidence [2, 9], further corroborate that ITA instabilities are indeed physical, and not just the result of a mathematical artifact. The physical mechanism governing ITA instabilities relies on the feedback between acoustic waves generated by unsteady heat release rate and the latter being sensitive to acoustic velocity fluctua-

17 tions upstream of the flame [10, 11]. ITA instabilities do not require ideal
18 anechoic conditions to exist: in [12, 13] it has been demonstrated that this
19 kind of instability is relevant in thermoacoustic systems with partially re-
20 flecting boundary conditions. In [14] it was shown that, for a fully reflective
21 Rijke-tube like systems with an $n-\tau$ flame response model, the resonance
22 frequency characterizing the ITA feedback loop for $n \rightarrow 0$ is the same as the
23 one obtained in anechoic systems. In [15] the evolution of thermoacoustic
24 eigenfrequency trajectories from fully reflecting to anechoic conditions was
25 reported. Some trajectories are pushed to infinitely negative growth rates
26 in the anechoic limit, others retain a finite frequency and growth rate. The
27 former were called ITA eigenfrequencies (for nonzero reflecting conditions)
28 and the latter “pure ITA” eigenfrequencies (in the anechoic limit).

29 One objective of the present study is demonstrating that the conclusions
30 derived for ITA modes in simple configurations (straight Rijke tubes) and/or
31 with simple flame models ($n-\tau$) can be generalized to a much larger set of
32 thermoacoustic systems. We will show that a clear distinction between “ITA
33 driven” and “acoustic driven” thermoacoustic modes (as in [15]) is not al-
34 ways possible when using the definitions of ITA and acoustic modes given
35 in the literature; an alternative definition that allows for this distinction is
36 proposed. Furthermore, we will show that the interplay between modes of
37 acoustic and ITA origin leads to the existence of exceptional points (EPs).
38 EPs have been identified as the fundamental concept at the base of many
39 scientific phenomena in several fields, including non-Hermitian quantum me-
40 chanics, optics and acoustics [16–18]. In thermoacoustics, they have only
41 been recently discussed [19]. At EPs two or more eigenvalues coalesce and

42 so do their corresponding modeshapes. The resulting eigenvalue is, thus,
43 degenerate and defective. In the simplest case, a defective eigenvalue has
44 algebraic multiplicity two and geometric multiplicity one. This is accompa-
45 nied by special properties like infinite parameter sensitivity. The effects of
46 EPs on thermoacoustic eigenvalue trajectories is the second main objective
47 of the study. In particular, we will demonstrate how thermoacoustic eigen-
48 modes can become unstable because of strong mode veering caused by their
49 interaction with EPs, even if the latter have negative growth rates. The iden-
50 tification of EPs, thus, helps understanding the origin of some thermoacoustic
51 instabilities.

52 In §2, the definition of ITA modes is revised, and the findings of [14]
53 are generalized to arbitrary thermoacoustic systems. It is also shown that a
54 given thermoacoustic mode could be thought of as originating from either an
55 acoustic or a “pure ITA” mode, depending on whether a thermoacoustic con-
56 figuration is considered to be originating from (i) an acoustic cavity in which
57 the strength of the flame is gradually increased, or (ii) an anechoic com-
58 bustion chamber in which the reflection coefficients are gradually increased.
59 In §3, this ambiguity is tied to the existence of EPs. A general strategy to
60 numerically identify EPs is proposed, and their effect on the trajectories of
61 thermoacoustic modes is discussed. The theory is demonstrated in §4 using
62 two thermoacoustic configurations modeled with the 3D Helmholtz equa-
63 tion. In §4.1 an axial combustor is considered, and EPs stemming from the
64 interaction between acoustic and ITA modes are found; in §4.2 an annular
65 combustor is considered, and an EP is identified as the coalescence of two
66 thermoacoustic modes of acoustic origin.

67 **2. Acoustic, intrinsic, and thermoacoustic modes**

68 Thermoacoustic phenomena arise from the interaction between unsteady
 69 heat release rate and acoustic fluctuations. The linear stability of thermoacoustic systems in the low-Mach-number limit can be assessed by investigating the eigenvalues of the Helmholtz equation with a heat release source
 70 term [20, 21]. In the frequency domain it reads [22, 23]
 71
 72

$$\nabla \cdot (c^2 \nabla \hat{p}) - s^2 \hat{p} - \frac{(\gamma - 1) \bar{q}(\mathbf{x})}{\bar{\rho} \bar{u}} \mathcal{F}(s) \nabla \hat{p}_{\text{ref}} \cdot \hat{\mathbf{n}}_{\text{ref}} = 0, \quad (1)$$

73 where \hat{p} represents the pressure fluctuations in frequency domain, γ the heat
 74 capacity ratio, c the local speed of sound, $\bar{\rho}$ the mean density, and $s \equiv$
 75 $\sigma + i\omega$ the Laplace variable – where σ is the growth rate and $\omega \equiv 2\pi f$
 76 is the angular frequency – representing the eigenvalues of interest. $\mathcal{F}(s)$ is
 77 the Flame Transfer Function (FTF), which represents the linear response
 78 of heat release rate fluctuations resulting from perturbations in the acoustic
 79 velocity field at a reference location, indicated with the subscript $(\)_{\text{ref}}$. The
 80 heat release is located only in a (compact) sub-domain of the total volume,
 81 where the local mean heat release, $\bar{q}(\mathbf{x})$ in Eq. (1), is non-zero. Here we are
 82 implicitly assuming that the unsteady heat release rate is proportional to
 83 the mean one, which is not strictly necessary but inconsequential due to the
 84 acoustic compactness of the flame.

85 The FTF is often expressed in terms of the n - τ model

$$\mathcal{F}(s) \equiv \frac{\bar{u} \hat{q}}{\bar{Q} \hat{u}} \equiv n e^{-s\tau}, \quad (2)$$

86 where n and τ represent the heat release gain and time-lag with respect to
 87 velocity fluctuations, and \bar{u} and \bar{Q} are the mean characteristic velocity and

88 mean global heat release rate, respectively. A non-dimensional impedance Z
 89 specifies the boundary conditions:

$$\hat{p} + \frac{cZ}{s} \nabla \hat{p} \cdot \hat{\mathbf{n}} = 0. \quad (3)$$

90 Traditionally, thermoacoustic modes have been understood as perturba-
 91 tions of purely acoustic modes (as defined in §1). Consequently, their eigen-
 92 frequencies and modeshapes have been sought in the vicinity of those of the
 93 same system without any unsteady heat release rate ($\mathcal{F} = 0$ in eq. (1)). By
 94 this assumption, Galerkin mode expansions of thermoacoustic modes based
 95 on the acoustic modes have been proposed [24]. The recently discovered in-
 96 trinsic thermoacoustic modes, however, show that this is not always appropri-
 97 ate. Thermoacoustic oscillations can have frequencies which are not directly
 98 related to the purely acoustic eigenfrequencies of the combustor [1, 9]. This
 99 is because these modes arise from the feedback loop created by upstream
 100 traveling acoustic perturbations generated by the flame, which trigger veloc-
 101 ity fluctuations upstream of the flame [10, 11, 25]. These in turn lead to
 102 the generation of heat release rate fluctuations (see Fig. 2). Because this
 103 mechanism does not require any interaction with acoustic waves reflected at
 104 the boundaries, its associated modes have been labelled ITA modes.

105 *2.1. Origin of thermoacoustic modes*

106 From the literature, modes of ITA origin are defined to be those associated
 107 with the eigenfrequency loci that contain pure ITA eigenfrequencies in the
 108 anechoic limit [26]. Equivalently, modes of acoustic origin are defined to be
 109 those associated with the eigenfrequency loci that contain purely acoustic
 110 eigenfrequencies in the $n \rightarrow 0$ limit. It appears therefore meaningful that a

111 given thermoacoustic mode can be referred to as of acoustic origin or of ITA
 112 origin [15]. There is, however, an inconsistency between the definitions given
 113 above and the idea of classifying thermoacoustic modes depending on their
 114 origin, which we now demonstrate.

115 For this purpose, we consider a Rijke tube configuration: a straight duct
 116 of length 0.5 m with a temperature jump $T_2/T_1 = 4$ across a flame element
 117 located in the middle of the tube. Explicit expressions for the dispersion
 118 relation of the thermoacoustic eigenvalues can be found in this configuration
 119 using network approaches (see [27–29] and §1 in the supplementary material).

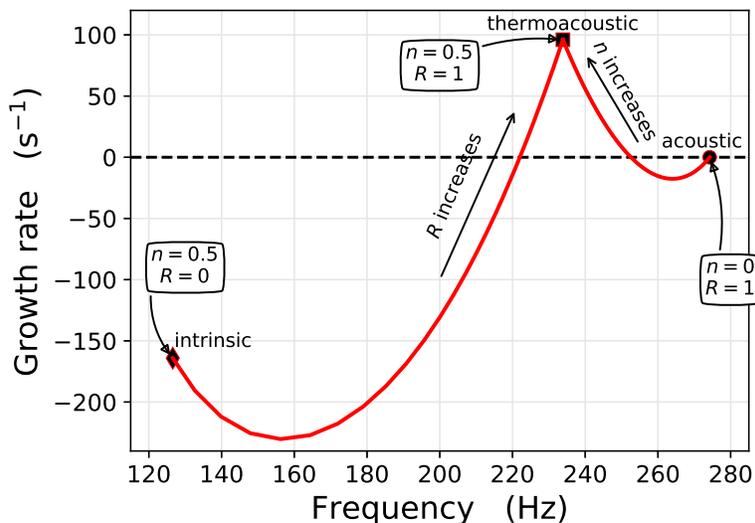


Figure 1: Thermoacoustic mode trajectories while varying the reflection coefficient R or the interaction index n . Depending on which parameter is varied towards zero, the thermoacoustic mode converges to either an acoustic or a pure ITA mode. This illustrates the need for an unambiguous definition of the origin of the thermoacoustic mode origin.

120 We model the Rijke tube using finite elements, solving the nonlinear
 121 eigenvalue problem (1). We use continuation methods – based on high-order

122 adjoint-based eigenvalue sensitivity [30] – to track the evolution of a specific
 123 eigenvalue while varying two parameters: the flame interaction index n , and
 124 the magnitude of the reflection coefficients $|R|$. The flame time delay is fixed
 125 to $\tau = 3.96$ ms. In the fully reflective case, $|R| = 1$, the tube is assumed to
 126 be acoustically closed ($R = 1$) in the cold region, and open ($R = -1$) in the
 127 hot one. The corresponding non-dimensional impedances (3) are calculated
 128 as

$$Z = \frac{1 + R}{1 - R}. \quad (4)$$

129 We start by setting $|R| = 1$ (so that the up- and downstream impedances
 130 are ∞ and 0, respectively), and $n = 0$. Then, only purely acoustic modes
 131 exist, featuring zero growth rate as no damping mechanisms are modeled.
 132 An acoustic mode with an angular frequency of 274 Hz is found (see Fig. 1).
 133 This value agrees well with that predicted from low-order network models
 134 (see supplementary material §1). We then vary the interaction index n and
 135 track the eigenvalue evolution. Note that, for small values of the interac-
 136 tion index the flame damps the mode, but for larger values this trend is
 137 reversed and the mode becomes unstable. Once we have reached $n = 0.5$, a
 138 reasonable value for a flame response, we maintain this value and vary the
 139 reflection coefficient magnitudes $|R|$ from 1 to 0. The up- and downstream
 140 impedances are calculated according to Eq. (4), with R positive (negative)
 141 in the upstream (downstream) region. When $|R| = 0$, the mode is a pure
 142 ITA mode. There are analytical expressions available from the literature for
 143 pure ITA modes in Rijke tubes (see [1, 3] and supplementary material §1).

144 These are given by

$$s_{\text{ITA}} \equiv \frac{1}{\tau} \ln \left[\left(\sqrt{\frac{T_2}{T_1}} - 1 \right) n \right] + \frac{(2k+1)\pi}{\tau} i, \quad k \in \mathbb{Z}. \quad (5)$$

145 For the chosen values of n and τ , the resulting intrinsic frequency and growth
 146 rate are $f_{\text{ITA}} = 1/(2\tau) \approx 126$ Hz and $\sigma_{\text{ITA}} = -175 \text{ s}^{-1}$, which agree well with
 147 the values obtained from the Helmholtz model (see Fig. 1).

148 2.2. An alternative definition of intrinsic modes

149 The previous example indicates that there exist thermoacoustic modes
 150 that can be arbitrarily associated to either acoustic or pure ITA eigenfre-
 151 quencies. This depends on whether the parameter responsible for the ther-
 152 moacoustic coupling is considered to be n or R . With the given definitions of
 153 acoustic modes ($n = 0$) and pure intrinsic modes ($R = 0$), it is therefore am-
 154 biguous to think of a thermoacoustic mode as of “acoustic” or “ITA” origin.
 155 In order to assign to thermoacoustic modes a specific source, the definition
 156 of acoustic and intrinsic modes must rely on *one* parameter.

157 A step in this direction has been done in [14], considering a 1D Rijke
 158 tube with an n - τ flame response. Damping, mean flow and entropy-wave
 159 effects have been neglected. For this configuration a dispersion relation can
 160 be derived, whose zeroes represent the thermoacoustic modes. Using it, it
 161 was shown that, when steering n towards zero, any thermoacoustic mode
 162 will either reduce to an acoustic mode or move towards an infinitely damped
 163 mode, with the same angular frequency as an intrinsic mode, regardless of
 164 the boundary conditions. Thus, using only the parameter n it is still possible
 165 to identify intrinsic eigenfrequencies, and uniquely associate an origin to each
 166 thermoacoustic mode.

167 These findings are in fact much more general. In the following, we will
 168 present a derivation of these results with a method different from that of [14].
 169 Our proof is independent of the dispersion relation and, thus, valid for con-
 170 figurations with mean flow, damping, arbitrary reflection coefficients, and
 171 arbitrary expression for the FTF.

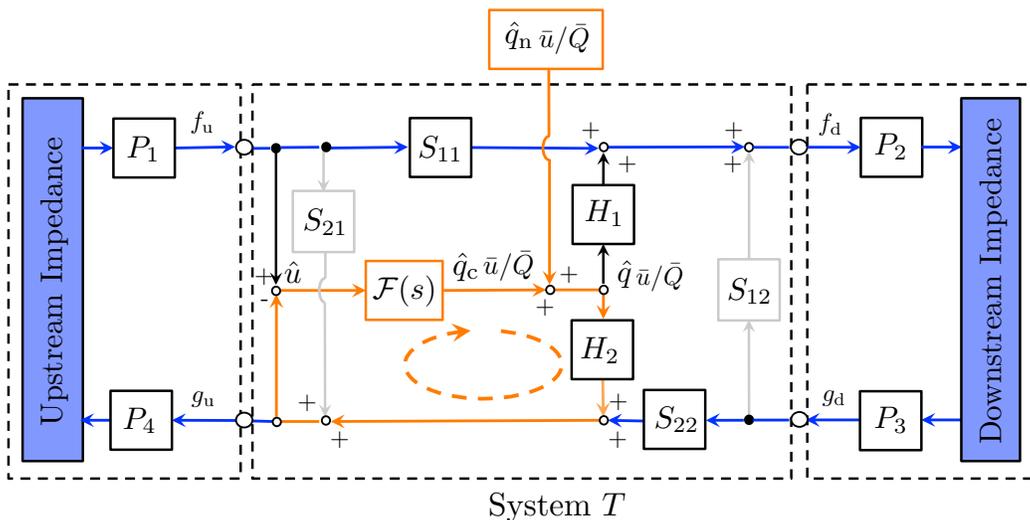


Figure 2: Block-diagram representation of a generic axial combustor. The upstream and downstream impedances can be complex-valued and frequency dependent, so that they can be used to model arbitrarily shaped volumes. The flame is considered compact, so that jump conditions across it can be expressed in terms of a scattering matrix \mathbf{S} . The ITA feedback loop is highlighted in orange, and does not involve any interaction with the acoustic boundary conditions.

172 We consider an arbitrary axial combustor, in which only plane waves
 173 propagate, containing an acoustically compact flame. No assumption is made
 174 on the actual shape of the setup, presence of a mean flow, or the acoustic
 175 boundary conditions. Such a generic configuration can still be represented in
 176 block form (Fig. 2), as commonly done in control theory and network model

177 approaches. The propagation blocks, $P_j = e^{-s\tau_j}$, transport the acoustic
 178 waves from a location x to $x + \Delta x$. τ_j is a characteristic acoustic propagation
 179 time delay [20], proportional to Δx , and generally a function of the mean
 180 flow. Damping models add an imaginary term in the exponential of the
 181 propagating terms, multiplying s . The jump conditions across the compact
 182 flame are contained in the coefficients of the scattering matrix, relating the
 183 incident and outgoing acoustic waves:

$$\begin{bmatrix} f_d \\ g_u \end{bmatrix} = \mathbf{S} \begin{bmatrix} f_u \\ g_d \end{bmatrix} + \mathbf{H}\hat{q} = \begin{bmatrix} S_{11} & S_{12} \\ S_{21} & S_{22} \end{bmatrix} \begin{bmatrix} f_u \\ g_d \end{bmatrix} + \begin{bmatrix} H_1 \\ H_2 \end{bmatrix} \hat{q}. \quad (6)$$

184 The factors H_j account for the scaling between the heat release rate fluc-
 185 tuations and the acoustic waves, and are not frequency dependent (see sup-
 186plementary material). The coefficients of the scattering matrix, S_{ij} , can in
 187 general be function of the frequency, when losses or inertial effects in terms
 188 of effective lengths are considered. However, it is standard in the analysis of
 189 intrinsic modes in thermoacoustics to assume that they are not, to allow for
 190 analytical treatment. The coefficients of \mathbf{S} are derived from jump conditions,
 191 which are conservative and show no frequency dependence. Their expression
 192 depends on the presence/absence of an area increase/decrease, temperature
 193 jump, entropy and/or vorticity waves. The theory presented here is gen-
 194eral in this respect, and the definition of the coefficients S_{ij} is kept implicit;
 195 an example of their expression is provided in the supplementary material.
 196 Lastly, the response of the acoustic configuration upstream and downstream
 197 of the flame is modeled by means of arbitrary impedances, which are con-
 198verted into reflection coefficients R via Eq. (4). These can be complex-valued
 199 and/or frequency dependent, and may contain the effects of area variations.

200 From Fig. 2, considering the balance of the waves at each node, it can be
 201 shown that the response of acoustic velocity fluctuations \hat{u} to heat release
 202 rate fluctuations \hat{q} is

$$\hat{u} = \frac{(1 - P_1 P_4 R_1)[H_2 + (P_2 P_3 R_2)(H_1 S_{22} - H_2 S_{12})]}{P_1 P_4 R_1 S_{21} + P_2 P_3 R_2 S_{12} + P_1 P_2 P_3 P_4 R_1 R_2 (S_{22} S_{11} - S_{21} S_{12}) - 1} \hat{q}. \quad (7)$$

203 Assuming that all the components of (7) are analytic functions of the eigen-
 204 frequency s , its numerator, $N(s)$, does not have poles. Thus, the *acoustic*
 205 eigenfrequencies are those for which the denominator of (7), $D(s)$, vanishes:

$$D(s) := P_1 P_4 R_1 S_{21} + P_2 P_3 R_2 S_{12} + P_1 P_2 P_3 P_4 R_1 R_2 (S_{22} S_{11} - S_{21} S_{12}) - 1 = 0 \quad (8)$$

207 The heat release rate response, $\hat{q}_c/\bar{Q} = \mathcal{F}(s)\hat{u}/\bar{u}$, is **assumed to have**
 208 **no finite poles, for simplicity, although this assumption could be relaxed.**
 209 This is true for many traditional flame response approximations, including
 210 $n - \tau$ models with constant or polynomial coefficients, and more generally for
 211 sum of time delay models, which can well fit Flame Transfer and Describing
 212 Functions [31, 32].

213 The ITA loop of the system is highlighted in Fig. 2. It is characterized
 214 by the transfer function

$$\hat{q} = \frac{1}{H_2 \mathcal{F}(s) + 1} \hat{q}_n, \quad (9)$$

215 where \hat{q}_n can be understood as a source of combustion noise [12]. This implies
 216 that ITA modes are found when

$$H_2 \mathcal{F}(s) + 1 = 0, \quad (10)$$

217 which is equivalent to definitions found in the literature [1, 3] for a Rijke
 218 tube with anechoic boundary conditions. Coupling the acoustic and flame
 219 responses yields the closed-loop transfer function

$$\hat{u} = \frac{N(s)}{D(s) - N(s)\mathcal{F}(s)}\hat{q}_n. \quad (11)$$

220 The *thermoacoustic* modes are found¹ when $D(s) - N(s)\mathcal{F}(s) = 0$.

221 One can then track these eigenvalues by slowly varying the gain of the
 222 FTF from a finite value towards zero. In the following, we shall assume
 223 that $\mathcal{F}(s) = ne^{-s\tau}$, as this permits a direct comparison of our results with
 224 those available in the literature, and eases the notation. Two scenarios are
 225 possible:

- 226 1. $\lim_{n \rightarrow 0} |ne^{-s\tau}| \rightarrow 0$. Then, for s to be a pole of Eq. (11), the condition
 227 $D(s) = 0$ must be satisfied. This coincides with the acoustic modes, as
 228 per Eq. (8);
- 229 2. $\lim_{n \rightarrow 0} |ne^{-s\tau}| \rightarrow \mathcal{O}(1)$. This is possible if and only if, asymptotically,
 230 $e^{-\sigma\tau} \sim \alpha/n$, with $\alpha \in \mathbb{R}$. This value is not arbitrary, but can be linked
 231 to the elements of the scattering matrix and the heat release scaling
 232 coefficients H_j (see [Appendix A](#)). This implies that

$$\sigma \sim \frac{1}{\tau} \log(n/\alpha) \quad \text{as } n \rightarrow 0, \quad (12)$$

233 and that $\lim_{n \rightarrow 0} ne^{-(\sigma+i\omega)\tau} \propto e^{-i\omega\tau}$. In other words, the infinite growth rate
 234 of the time-delayed terms is balanced by the vanishing flame strength
 235 when $n \rightarrow 0$. We now want to find those values of s that are poles

¹This assumes that no zero-pole cancellations, which are nonetheless extremely rare, occurs; otherwise, some extra modes are identified.

236 of (11) in this limit. It can be shown (see [Appendix A](#)) that the
237 angular frequencies of these infinitely damped modes are identical to
238 those of ITA modes (Eq. (5)).

239 This proves that, in the $n \rightarrow 0$ limit, thermoacoustic modes are split into
240 two distinct sets. One of them is equivalent to the set of acoustic modes, the
241 other is connected to the ITA modes (with infinite damping), regardless of
242 the boundary conditions.

243 We remark that the intrinsic loop highlighted in Fig. 2 and defined in
244 Eq. (10) exists in an isolated fashion only when no area variations are present
245 in the volumes upstream/downstream of the flame. Otherwise, even when
246 purely anechoic boundary conditions are imposed, reflection of acoustic waves
247 will occur due to the area changes. This modifies the anechoic intrinsic loop
248 and the consequent definition of pure intrinsic modes, as discussed in [7]. This
249 is, however, unimportant in the current analysis because only the limit $n \rightarrow 0$
250 is considered. The amplitude of the ITA generated waves that propagate
251 away from the flame vanish in this limit because they are infinitely damped,
252 regardless of the presence of area variations. As a consequence of this, the
253 definition of intrinsic modes for vanishing n depends only on the values of
254 the scattering matrix and flame scaling coefficients, as shown in [Appendix](#)
255 [A](#).

256 Using a single parameter to define both acoustic and ITA frequencies
257 allows for unambiguously associating each thermoacoustic mode with a spe-
258 cific origin. This is not possible when different parameters are used to define
259 ITA and acoustic modes. Here, the chosen parameter is the flame gain (as
260 it retains the natural definition of acoustic modes), but the reflection coeffi-

261 cient could be chosen equivalently (which would lead to a different, but still
262 unique, classification of the eigenvalues).

263 Furthermore, the results of this section have an important consequence
264 for practical applications and the numerical calculation of thermoacoustic
265 modes. Commonly, thermoacoustic modes are found by initializing Helmholtz
266 solvers with the purely acoustic frequencies (found when $n = 0$) and then
267 by gradually increasing the value of n and track the evolution of the eigen-
268 values [33]. However, this method fails in identifying all thermoacoustic
269 modes, as some modes are of ITA origin. Using the results of this proof,
270 we have a new set of guesses that can be used to identify the remaining
271 thermoacoustic modes: starting from a small but non-zero value of the in-
272 teraction index, thermoacoustic modes of intrinsic origin are approximately
273 given by Eq. (A.5). No other modes can be found because, when $n \rightarrow 0$,
274 all modes *must* belong to one of these two sets. Once located in this limit,
275 all thermoacoustic eigenvalues can be tracked to the desired value of n us-
276 ing continuation methods. Thus, with this strategy the space that needs to
277 be explored is limited to that in the vicinity of the theoretically estimated
278 solutions. This leads to a gain in both the numerical time needed to locate
279 the thermoacoustic modes, and in the confidence that all modes (in a given
280 frequency range) have been identified. This will be demonstrated in §4.

281 2.2.1. The annular combustor case

282 In this section we qualitatively discuss how the results of §2.2 can be
283 extended to annular and can-annular thermoacoustic configurations. In par-
284 ticular, we will consider systems featuring a discrete rotational symmetry.
285 Such systems are generally modeled with two acoustic volumes connected by

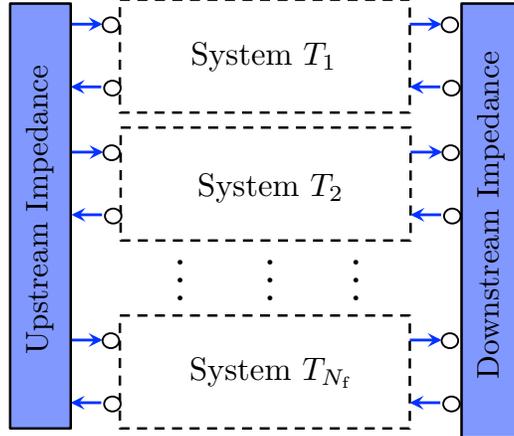


Figure 3: Block-diagram representation of an annular combustor with discrete rotational symmetry and N_f flames. The flames are located in ducts that connect upstream and downstream acoustic cavities (annular plenum and annular combustion chamber).

286 N_f ducts, in which identical flames are located (see Fig. 3). The acoustic
 287 boundary conditions couple the responses of the various ducts, in which the
 288 acoustic field is assumed to be one dimensional, and can be modeled with
 289 impedance matrices [34, 35]. These contain the acoustic response in all the
 290 ducts when the acoustic field in a given duct is excited. Given the rota-
 291 tional symmetry of the system, these matrices are circulant, which has direct
 292 connections with a possible Bloch representation of the dynamics [36–38].

293 In the limit $n \rightarrow 0$, the same results of the previous section must hold for
 294 each duct. In this limit, there is no physical mechanism that couples the N_f
 295 intrinsic loops. Since the flames in the various ducts are identical, so are all
 296 intrinsic loops, which are still governed by the dispersion relation (10). This
 297 results in N_f identical intrinsic eigenfrequencies, infinitely damped, and with
 298 the same frequency of pure ITA modes. This discussion is kept qualitative
 299 because, given the matrix formulation needed to represent annular systems

300 with discrete symmetries, a closed-form formulation of analytical results is
301 impractical.

302 These intrinsic loops can be thought of as a set of identical, decoupled
303 oscillators. A weak coupling between them is achieved by either consider-
304 ing small but non-zero up- and downstream reflection coefficients when the
305 flame interaction index is finite, or a small but non-zero flame interaction
306 index when the reflection coefficients are finite. In both cases, the dynam-
307 ics of ITA modes is governed by weakly coupled oscillators. When identical
308 oscillators are weakly coupled, the eigenvalues of the weakly coupled system
309 form clusters of closely spaced eigenvalues [39]. This is the case, for example,
310 in can-annular systems, in which the acoustic coupling between the various
311 cans is weak, and clustering of thermoacoustic modes can be observed [40].
312 The same holds true for modes of ITA origin in annular systems. This has
313 been first observed numerically in [41], where clusters of modes with eigen-
314 frequencies close to those of ITA modes were found.

315 **3. Interaction of acoustic and intrinsic modes with exceptional** 316 **points**

317 In the previous section, we have discussed how the origin of thermoacous-
318 tic modes can be assessed using only the interaction index n of the flame.
319 There remain however some points in the spectrum of thermoacoustic systems
320 that elude this classification. These points are known as exceptional points
321 (EPs). EPs are a particular type of degenerate, defective eigenvalues, with
322 the additional property of being singularities in the parameter space [42]. EPs
323 have recently been identified in the spectrum of a Rijke tube-like thermo-

324 coustic system using an explicitly known dispersion relation [19]. Here, we
 325 show how EPs of any thermoacoustic systems can be identified numerically
 326 without using the dispersion relation, but the self-orthogonality property of
 327 the eigenfunctions at EPs.

328 3.1. Self-orthogonality

329 The eigenvalue sensitivity of thermoacoustic modes w.r.t. a parameter ξ
 330 is given by [43, 44]:

$$\frac{\partial s_j}{\partial \xi} = - \frac{\langle \hat{\mathbf{p}}_j^\dagger | \frac{\partial \mathbf{L}}{\partial \xi} \hat{\mathbf{p}}_j \rangle}{\langle \hat{\mathbf{p}}_j^\dagger | \frac{\partial \mathbf{L}}{\partial s} \hat{\mathbf{p}}_j \rangle} \Big|_{s=s_j}, \quad (13)$$

331 where the adjoint eigenvectors have been denoted with the superscript \dagger ,
 332 and the matrix \mathbf{L} contains the discretization of the thermoacoustic eigen-
 333 value problem (1). Equation (13) is valid whenever the denominator is non-
 334 zero. This is always guaranteed to be the case for non-defective eigenvalues
 335 (even if they are degenerate), but it is zero for defective eigenvalues [45]. In
 336 fact, the derivation of equation (13) assumes that a bi-orthonormal set of
 337 direct/adjoint eigenfunctions can be chosen [46]:

$$\langle \hat{\mathbf{p}}_i^\dagger | \frac{\partial \mathbf{L}}{\partial s} \hat{\mathbf{p}}_j \rangle = \delta_{i,j}. \quad (14)$$

338 This breaks down at defective points, because the basis of the eigenvectors is
 339 incomplete. In particular, it is possible to show that, for defective eigenvalues,
 340 the direct and corresponding adjoint eigenvectors satisfy [47]

$$\langle \hat{\mathbf{p}}_{\text{def}}^\dagger | \frac{\partial \mathbf{L}}{\partial s} \hat{\mathbf{p}}_{\text{def}} \rangle = 0. \quad (15)$$

341 This property is known as self-orthogonality. At EPs, it manifests itself in
 342 infinite eigenvalue sensitivity. We reference to the supplementary material §2
 343 for remarks on numerical aspects of self-orthogonality.

344 *3.1.1. General method for the identification of exceptional points*

345 We exploit the infinite eigenvalue sensitivity at EPs to devise a general
 346 strategy for their identification in thermoacoustic systems. At EPs, we have

347

$$\lim_{\xi \rightarrow \xi_{\text{EP}}} \left| \frac{\partial s}{\partial \xi} \right|^{-1} = 0 \quad (16)$$

348 Thus, the identification of EPs is reduced to a root-finding problem, which
 349 can be straightforwardly solved numerically with iterative methods. Note
 350 that, every time the parameter ξ is updated in the iterative scheme, a new
 351 eigenvalue problem needs to be solved, and the sensitivity can then be calcu-
 352 lated using Eq. (13). Furthermore, since the eigenvalues of thermoacoustic
 353 problems are generally complex-valued, also the value of the parameter ξ at
 354 which the EP is found using this strategy can be complex-valued. These
 355 complex-valued parameters may or may not be physically realizable: an EP
 356 found in a Rijke tube having a complex-valued length would not be realizable,
 357 but one found for a complex-valued impedance would. In order to identify
 358 EPs for real-valued parameters, we need to extend the parameter space un-
 359 der consideration to two independent parameters [42]. The identification of
 360 EPs in the real-valued parameter space reads

$$\lim_{\substack{\xi_1 \rightarrow \xi_{1,\text{EP}} \\ \xi_2 \rightarrow \xi_{2,\text{EP}}} } \left| \frac{\partial s}{\partial \xi_i} \right|^{-1} = 0 \quad \text{for } i = 1 \text{ or } 2, \quad (17)$$

361 which can be solved using standard multi-parameter root finding algorithms.

362 Despite their peculiar nature, EPs are not rare, and have been observed
 363 in a large variety of physical systems [48]. In thermoacoustics, they have
 364 been first discussed only recently [19]. In the latter study, they have been
 365 identified making use of the dispersion relation for the eigenvalues, which is

366 available only for simple networks. The method outlined in this study is more
367 general because it does not rely on the explicit knowledge of the dispersion
368 relation, which is typically not available, for instance, when using Helmholtz
369 solvers. The method has been tested on several configurations. For all tested
370 cases, identifying real-valued EPs was possible, as discussed in §4.

371 Lastly, we highlight that there appears to be evidence in the literature
372 that the effects of EPs in the spectra of thermoacoustic systems, even if not
373 investigated directly, have already been observed. In [49], eigenvalues hav-
374 ing infinite sensitivity have been identified analytically, in a Rijke tube-like
375 system. They have however imprecisely been linked to arbitrary degenerate
376 states rather than to EPs. Large eigenvalue sensitivities were also observed
377 in [50], together with the phenomenon of mode veering. Mode veering is
378 a manifestation of avoided crossing of two eigenvalues. This always occurs
379 when the thermoacoustic system parameters are close to a degenerate point,
380 at which the eigenvalue trajectories intersect. The behavior of the eigen-
381 value trajectories tells whether the degeneracy is defective or not. For a
382 degenerate eigenvalue with multiplicity 2, if the trajectories approach each
383 other from nearly opposite direction and then veer by 90° (as is the case
384 in [50]), then there is an EP close in parameter space, at which the eigen-
385 value sensitivity is infinite [42]. On the other hand, if the trajectories veer at
386 different angles, then the veering is due to a degenerate, non-defective point,
387 and the eigenvalue sensitivity remains finite. The existence of EPs is evident
388 also in [26, 51], where regions of high sensitivity exhibiting mode veering
389 have been identified, and their universality in connection to non-dimensional
390 groups has been demonstrated. Regarding annular combustors, the presence

391 of EPs can be inferred from the eigenvalue trajectories shown in [52, Fig. 8]
392 and [53, Fig. 6], constructed while varying n and τ . This example will be
393 further discussed in §4.2.

394 3.2. The effect of exceptional points on thermoacoustic eigenvalue trajectories

395 We shall now return to the Rijke tube example of §2.1. Using the nu-
396 merical method outlined in the previous section, with $\xi_1 = n$ and $\xi_2 = \tau$,
397 an EP is identified in this configuration for $n_{\text{EP}} = 0.075$ and $\tau_{\text{EP}} = 4.66$ ms,
398 having a frequency close to that of the acoustic mode of Fig. 1 and a negative
399 growth rate. Starting from the EP, we vary the flame gain n in the range
400 $[0, 0.5]$ and the reflection coefficient $|R|$ in the range $[0, 1.3]$. The resulting
401 eigenvalue trajectories are shown in Fig. 4.

402 One of the two trajectories obtained while varying n starts (for $n = 0$) at
403 the acoustic eigenvalue (on the neutral line); the other comes instead from
404 a very negative growth rate. On the contrary, the two trajectories obtained
405 while varying R start (for $R = 0$) from pure ITA modes, whose values es-
406 timated from Eq. (5) are reported with red circular markers in Fig. 4. The
407 trajectories meet at the exceptional value, and turn by 90° across it. Further-
408 more, because no discontinuities in the parameters are present in the modeled
409 system, the eigenvalue trajectories must be continuous. In the vicinity of the
410 exceptional point, this means that the trajectories must strongly veer, to
411 avoid a crossing. This explains the behavior of the eigenvalue trajectory
412 shown in Fig. 1: increasing n , the eigenvalue trajectory is first attracted to-
413 wards the EP, which has a negative growth rate. However, since crossing is
414 prohibited, the trajectory must strongly veer, which leads to a sudden change
415 in the trend of the eigenvalue sensitivity and eventually to the existence of

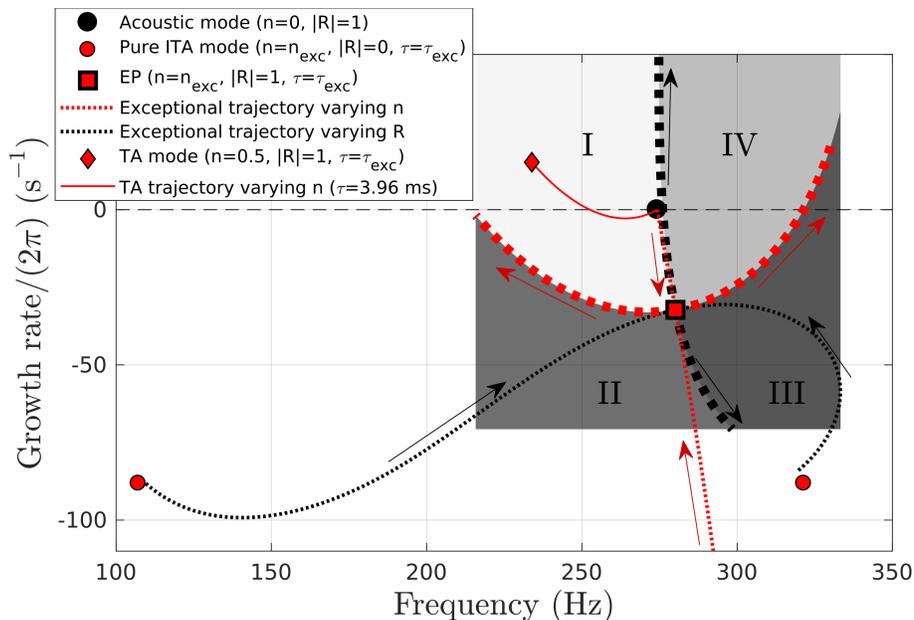


Figure 4: Behaviour of the eigenvalue trajectories obtained while varying the flame interaction index and reflection coefficients across an EP. The arrows indicate the direction that the trajectories follow when the parameters are increased. The acoustic solution is marked with a black circle, and the pure ITA modes for $\tau = \tau_{EP}$ with red circles. Because trajectories cannot intersect in the vicinity of an EP, four regions, labelled I to IV, corresponding to different behaviors in the limits $n \rightarrow 0$ and $R \rightarrow 0$ are identified. The thermoacoustic mode and the eigenvalue trajectory of Fig. 1 lie in region I.

416 an unstable mode. This is true even though the identified EP has a negative
 417 growth rate. Rather than the EP per se, it is its interaction with the eigen-
 418 value trajectories which is relevant: identifying the parameters at which EPs
 419 are found gives information about when strong changes in the eigenvalues
 420 sensitivities are expected. As demonstrated in Fig. 4, this sudden change in
 421 sensitivity can lead to thermoacoustic instabilities.

422 The fact that eigenvalue trajectories cannot cross in the vicinity of an

423 EP leads to a classification of the eigenvalue space in its vicinity. Consider
 424 the exceptional trajectories for n highlighted with thick red lines in Fig. 4:
 425 the acoustic mode is contained in the portion of the plane above this line.
 426 Because eigenvalue trajectories cannot intersect, thermoacoustic modes that
 427 start above this line must converge to the acoustic mode when $n \rightarrow 0$. On the
 428 other hand, modes that start below this line cannot converge to an acoustic
 429 mode when $n \rightarrow 0$, and will be pushed towards $\sigma \rightarrow -\infty$. Similarly, the
 430 exceptional trajectories for R highlighted with thick black lines in Fig. 4
 431 delimit the region of convergence towards two separate pure ITA modes when
 432 $R \rightarrow 0$: on the left, eigenvalues must be attracted towards the ITA mode
 433 with frequency $1/(2\tau)$; on the right, towards the ITA mode with frequency
 434 $3/(2\tau)$. Thus, four regions exist (I-IV in Fig. 4) in which the behavior of the
 435 eigenvalues in the limits $n \rightarrow 0$ and $R \rightarrow 0$ differs. For example in region I
 436 (top-left), eigenvalues must be attracted towards an acoustic solution when
 437 $n \rightarrow 0$ and to the ITA solution with frequency $1/(2\tau)$ when $R \rightarrow 0$. Similar
 438 arguments hold for the remaining regions. The thermoacoustic mode shown
 439 in Fig. 1 lies in region I in Fig. 4, which is consistent with the seemingly
 440 ambiguous $n \rightarrow 0$ and $R \rightarrow 0$ limits.

441 In summary, the “basins of attraction” of acoustic and pure ITA modes
 442 are determined by the parameter chosen to describe the thermoacoustic cou-
 443 pling. Because the exceptional trajectories for varying n and R are different,
 444 the resulting basins differ too. This explains why a classification of the ther-
 445 moacoustic modes using two separate parameters can be ambiguous. These
 446 findings on the identification of EPs and the behavior of eigenvalue trajecto-
 447 ries in their vicinity are general, and will be demonstrated on more complex

448 geometries in the last part of the study.

449 4. Numerical examples

450 In this section we demonstrate with two examples (an axial and an
451 annular configurations) the theoretical findings of this study. Both cases
452 are solved using the freely available 3D FEM code PyHoltz², dedicated to
453 (thermo)acoustic eigenvalue problems.

454 4.1. BRS combustor

455 As an axial configuration, we focus on the so-called BRS combustor [54].
456 Thermoacoustic oscillations with a frequency which is not close to any acous-
457 tic mode have been experimentally observed in this combustor [9], and have
458 been related to the effect of ITA modes in the literature [13]. The mod-
459 eled combustor is shown in Fig. 7 (see the aforementioned references for an
460 exhaustive geometrical description of the combustor).

461 We assume that the inlet/outlet are respectively acoustically closed and
462 open, and that a sudden temperature jump, $T_2/T_1 = 5.46$, occurs across
463 the flame. Starting from the purely acoustic scenario ($n = 0$, fully reflective
464 boundaries), we identify several acoustic modes having zero growth rate. The
465 lowest two frequencies correspond to a Helmholtz mode with $f_H = 54$ Hz and
466 a quarter-wave like mode, with a frequency of $f_{1/4} = 589$ Hz, and are shown
467 in Fig. 5.

468 We then set the interaction index to the small value $n = 0.001$ and the
469 time delay (arbitrarily) to $\tau = 6.88$ ms. The eigenvalues of this thermoacous-

²Freely available at <https://bitbucket.org/pyholtzdevelopers/public>

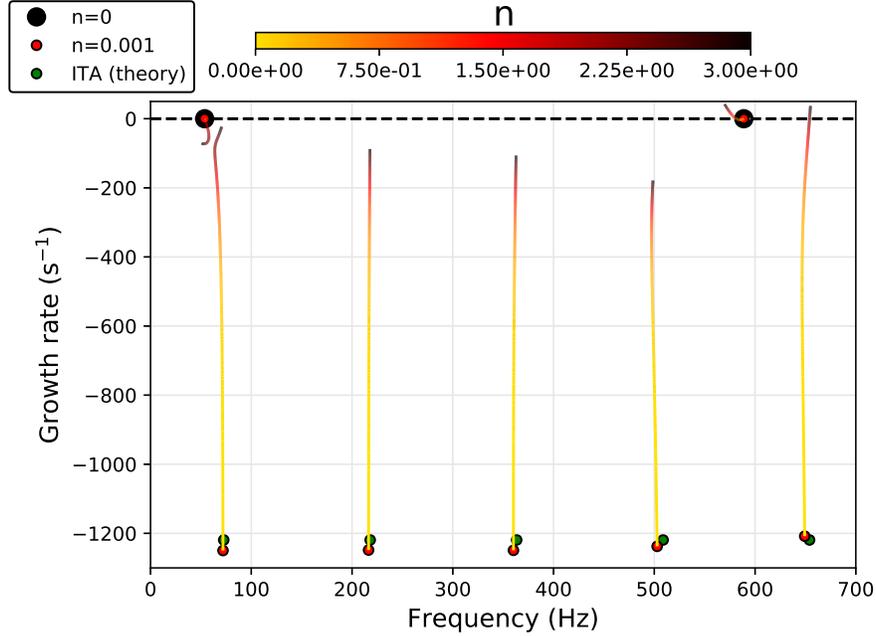


Figure 5: Location of the acoustic (black circles) and thermoacoustic (red circles) eigenvalues when $n = 0$ and $n = 0.001$, respectively. The theoretical guesses for the locations of the thermoacoustic modes of ITA origin are indicated with green markers. The lines track the eigenvalue trajectories for $n \in [0.001, 3]$.

470 tic system are shown in Fig. 5 with red markers; because of the weak effect of
 471 the flame, the eigenvalues of acoustic origin are almost unaffected. However,
 472 a new set of modes, having ITA origin, is found. These modes have been iden-
 473 tified using as guesses the expression (A.5), with $\beta^{-1} = (\theta^2 - 1)/(A_r\theta + 1)$,
 474 obtained from (A.3) when a temperature jump ($\theta = \sqrt{T_2/T_1}$) and an area
 475 jump ($A_r = A_2/A_1 = 7.95$) are found across the flame, and mean flow is
 476 neglected. The theoretical guesses are marked in Fig. 5 with green circles,
 477 and are used to initialize the search of thermoacoustic eigenvalues via New-
 478 ton's method. The converged thermoacoustic eigenvalues for $n = 0.001$ are

479 marked with red circles, and agree well with the theoretical predictions. The
480 configuration at hand has an area jump upstream of the flame, which affects
481 the definition of pure ITA modes in the anechoic limit [7]. However, this has
482 no effect on the definition of the ITA modes originating in the limit $n \rightarrow 0$.

483 Using these initial guesses, we can then track the evolution of *all* ther-
484 moacoustic eigenvalues in the region of interest while increasing the value
485 of the interaction index to any desired value. These trajectories are shown
486 with lines in Fig. 5: the growth rates of thermoacoustic modes of ITA origin
487 are far more sensitive to changes in n than the growth rate of the modes of
488 acoustic origin. For large values of n , the modes of ITA origin can become
489 unstable and feature the largest growth rates. Also, mode veering between
490 an eigenvalue of acoustic origin and one of ITA origin is visible at a frequency
491 of about 60 Hz.

492 This mode veering is relevant for the experimental observations of [9],
493 where oscillations with a frequency of 100 Hz were observed, and have been
494 associated to an ITA mode instability [4]. Therefore, we shall focus the
495 attention around the low-frequency Helmholtz mode only. To understand the
496 influence of the flame response on the spectrum, we vary $n \in [0, 3]$ and $\tau \in$
497 $[0, 0.016]$. The maximum τ is chosen to be $< 1/f_H$, to avoid that eigenvalue
498 trajectories intersect in the vicinity of the acoustic solution. The resulting
499 stability map is shown in Fig. 6; as commonly observed, depending on the
500 particular choice of both n and τ , the resulting thermoacoustic mode can
501 be stable or unstable. In the vicinity of the acoustic solution the eigenvalue
502 sensitivity with respect to changes in the interaction index is nonlinear: a
503 small increase in n from zero first stabilizes the pole, but the trajectory

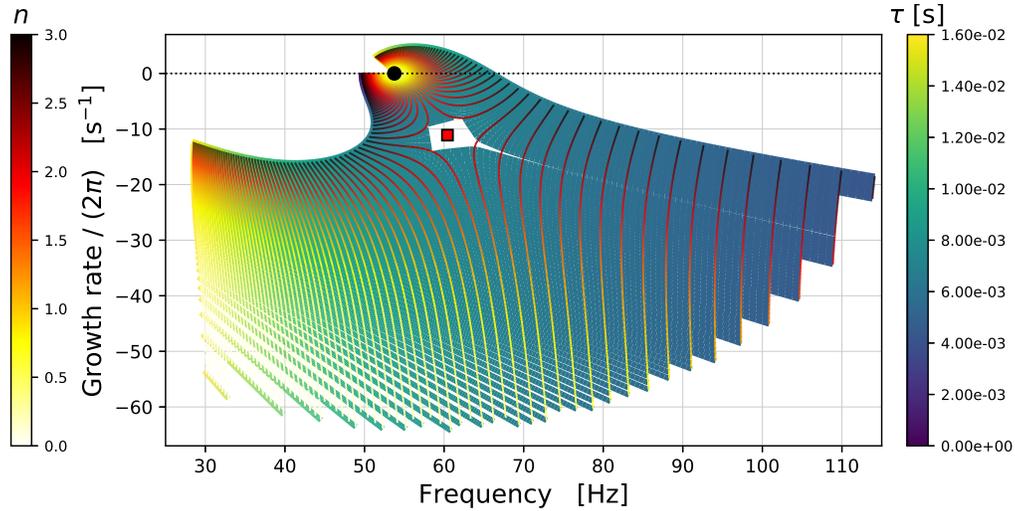


Figure 6: Trajectories of the lowest frequency eigenvalue of the BRS combustor when varying the parameters of the flame model. The trajectories with constant n and constant τ are highlighted with different colormaps. The acoustic solution ($f = 53.75$ Hz) is highlighted with a black circle. In the region which is avoided by the eigenvalues, there exists an EP ($n_{\text{EP}} = 2.181$, $\tau_{\text{EP}} = 6.96$ ms), indicated in red. The local behavior of the trajectories in this region is shown in Fig. 8a.

504 strongly veers and the mode can become unstable for larger values of n , as
 505 was observed in Fig. 1. Because of their veering, which can be observed for
 506 both the n - and τ -isolines, the eigenvalue trajectories avoid a region in the
 507 complex-frequency space, as discussed in [26]. The presence of this avoided
 508 region is one of the characteristics of EPs. Its existence can be confirmed
 509 using the procedure outlined in §3.1: starting from an educated guess based
 510 on Fig. 6, a root of Eq. (17) is found while varying n and τ . This root
 511 identifies an EP (see Fig. 8). Its modeshape is reported in Fig. 7, together
 512 with that of the purely acoustic Helmholtz mode (plenum dominant) and of
 513 the mode of ITA origin in the $n \rightarrow 0$ limit (flame region dominant). These

514 are the two modes that coalesce to create the EP, whose modeshape has
 515 clearly inherited features from both of them.

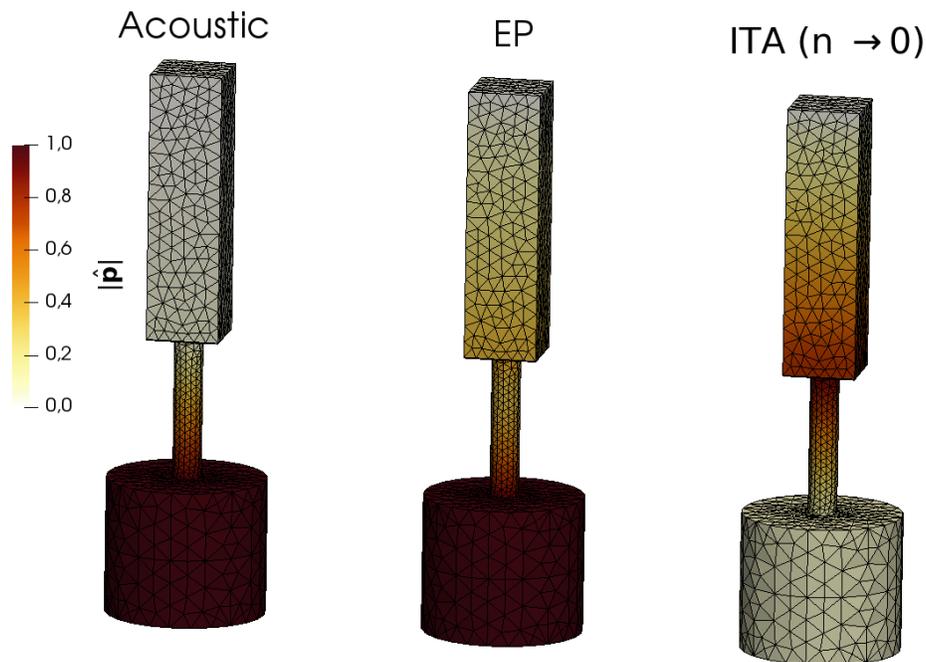


Figure 7: Geometry of the BRS combustor and pressure modeshapes of the low-frequency acoustic (left), ITA origin (right) and exceptional (middle) modes.

516 The trajectories of the eigenvalues around the EP are shown in Fig. 8. At
 517 the EP, two eigenvalues, one of acoustic and one of intrinsic origin, coalesce.
 518 Due to the high sensitivity of the eigenvalues in the vicinity of the EP, a small
 519 parameter range is considered. When one parameter (n or τ) is fixed at the
 520 exceptional value and the other is varied, the trajectories collide at the EP,
 521 and branch off at angles of 90° (Fig. 8a). The further the parameter values
 522 are from those of the EP, the less intense is the veering. Figures 8b and 8c
 523 show the frequency and growth rate surfaces as functions of n and τ close
 524 to the EP. Because real and imaginary part of the eigenvalue surfaces are

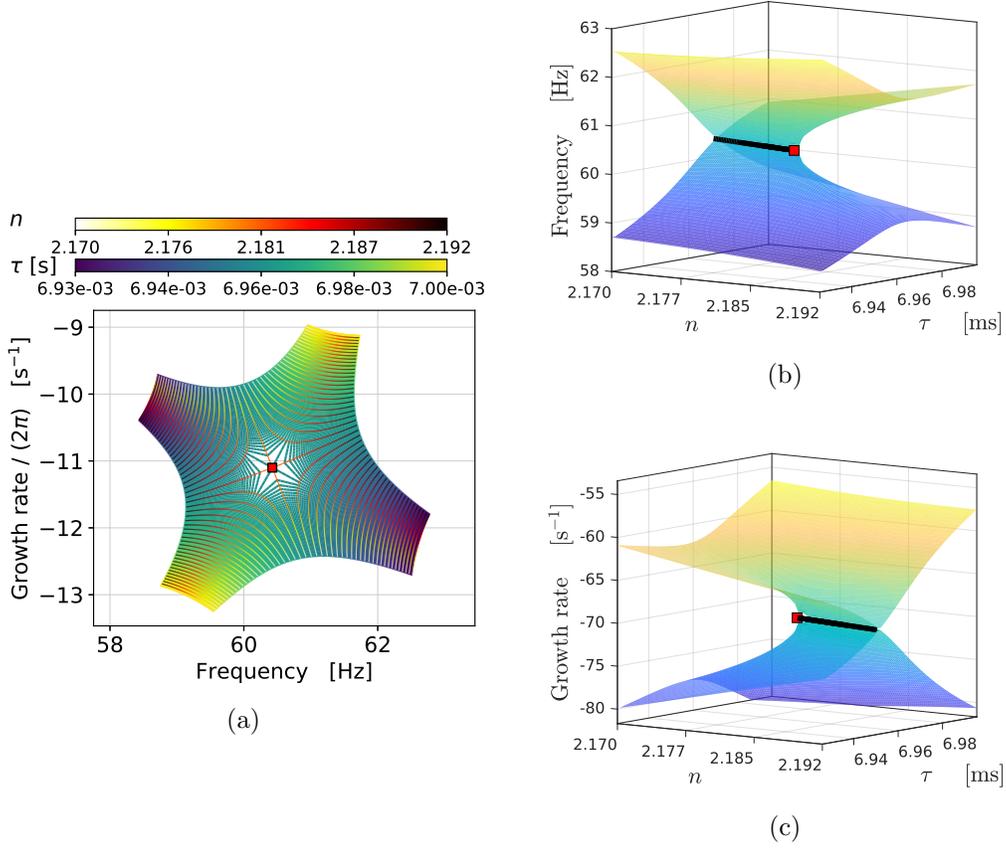


Figure 8: (a): Trajectories of the eigenvalues in the vicinity of the EP. At the EP the trajectories cross forming right angles. (b)-(c): Frequency and growth rate surfaces as functions of n and τ close to the EP. The Riemann cuts of the surfaces are highlighted with black lines, the EP in red.

525 plotted separately, each surface self-intersects, forming Riemann cuts [18].
 526 The number of eigenvalues found for an arbitrary pair of values (n, τ)
 527 to the EP is equal to the number of intersections of vertical lines passing
 528 through (n, τ) with the complex-valued surface. This is always equal to 2
 529 in Figs. 8b and 8c, except at the EP at which it is equal to 1, indicating

530 eigenvalue crossing.

531 Our analysis of the BRS system is consistent with the theoretical discus-
532 sion that thermoacoustic modes can uniquely be classified as of acoustic or
533 ITA origin in the limit $n \rightarrow 0$, regardless of the boundary conditions and
534 presence of area variations upstream of the flame. An EP exists in the spec-
535 tra of this combustor for specific values of n and τ , at which two eigenvalues
536 (one of acoustic and one of intrinsic origin) coalesce. Even if the system is not
537 operated at EP conditions, being sufficiently close to it in parameter space
538 results in strong mode veering. This can explain why the unstable frequency
539 observed experimentally in [9] significantly differs from all the acoustic eigen-
540 frequencies of the system, although the unstable mode may still be of acoustic
541 origin.

542 *4.2. Annular configuration*

543 As for an annular configuration, we consider the Helmholtz model of
544 a generic geometry formed by plenum and chamber volumes connected by
545 a given number of ducts (see Fig. 10). These configurations can also be
546 analyzed using network models, as discussed in [35, 55–57]. We investigate
547 the $N_f = 4$ burners setup presented in [52] because (i) the configuration has
548 closely spaced acoustic eigenvalues, and (ii) different regimes (uncoupled,
549 weakly coupled, and strongly coupled) have been identified in [52], which are
550 revisited here. This will show that the occurrence of these different regimes
551 can be explained with the existence of EPs.

552 The geometrical and thermodynamical parameters are taken from [56],
553 with the difference that a smaller temperature jump has been considered,
554 $T_2/T_1 = 1.5$, in order to have closely spaced acoustic eigenvalues when $n = 0$,

555 as in [52]. The two acoustic eigenvalues with the lowest frequencies are re-
 556 ported with black circles in Fig. 9. The two modes are plenum- and chamber-
 557 dominant, respectively, with azimuthal order $m = 1$, therefore degenerate.
 558 Their modeshapes³ are reported in Fig. 10, and the frequencies are close to
 559 the frequencies of the plenum/chamber, $f_i = c_i/(\pi D_i)$, where D_i are the
 560 diameters of the volumes. The deviation from these values is due to the
 561 cross-talking of the plenum-chamber volumes via the connecting ducts.

562 To investigate the effect of the flame response on the eigenvalues, we vary
 563 $n \in [0, 2]$ and $\tau \in [0, 0.015]$ s. The maximum time delay value is chosen
 564 to be $1/f_{\text{chamber}}$ so that eigenvalue trajectories looping around the acoustic
 565 eigenvalues will not cross each other [26, 58]. We track the eigenvalues using
 566 continuation methods and show the eigenvalue trajectories in Fig. 9. In the
 567 range of parameters studied, we identify three topologically different groups
 568 of eigenvalue trajectories:

- 569 I. when $n < 1.17$, varying τ results in looping the thermoacoustic eigenval-
 570 ues that originate from the plenum and chamber acoustic modes around
 571 these solutions. This corresponds to the “weakly coupled” regime dis-
 572 cussed in [56], in which it is appropriate to associate each thermoacous-
 573 tic mode to an acoustic mode.
- 574 II. when $1.17 < n < 1.44$, the trajectories do not follow closed loops any-
 575 more. The thermoacoustic eigenvalue that starts close to the plenum
 576 acoustic mode is shifted towards a value close to that of the chamber

³Because these solutions are degenerate with geometric multiplicity 2, they each have
 2 linearly independent modeshapes. Only one of them is shown in Fig. 10.

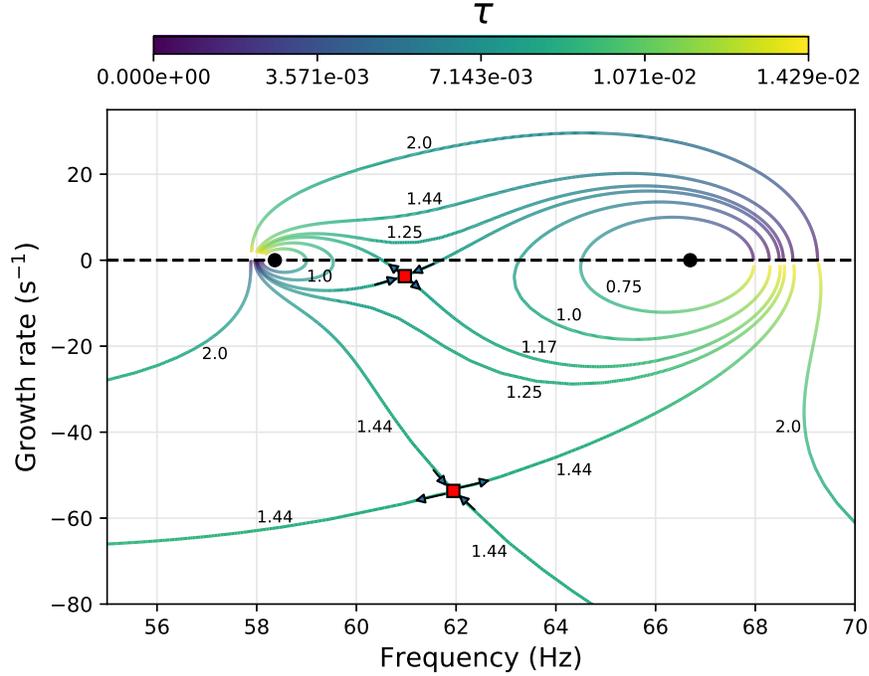


Figure 9: Eigenvalue trajectories of the annular configuration for constant n (isoline values) when τ is varied. The system has two closely spaced purely acoustic eigenvalues (black dots). Two exceptional points (red squares) are identified: one due to the interaction between two modes of acoustic origin, close to the real axis, and one due to the interaction between acoustic and intrinsic modes. The values of n at which EPs are located determine topological changes in the eigenvalue trajectories.

577 acoustic mode, and vice versa. In other words, the nature of these ther-
 578 moacoustic modes strongly varies depending on the value of τ consid-
 579 ered. For intermediate values of τ , the modes have frequencies which lie
 580 between those of the two acoustic modes, and their modeshapes are not
 581 anymore dominant in only the plenum or chamber, but instead in both
 582 cavities (see Fig. 10). For this reason, in this regime, which corresponds
 583 to the “strongly coupled” regime of [56], it would be inappropriate to

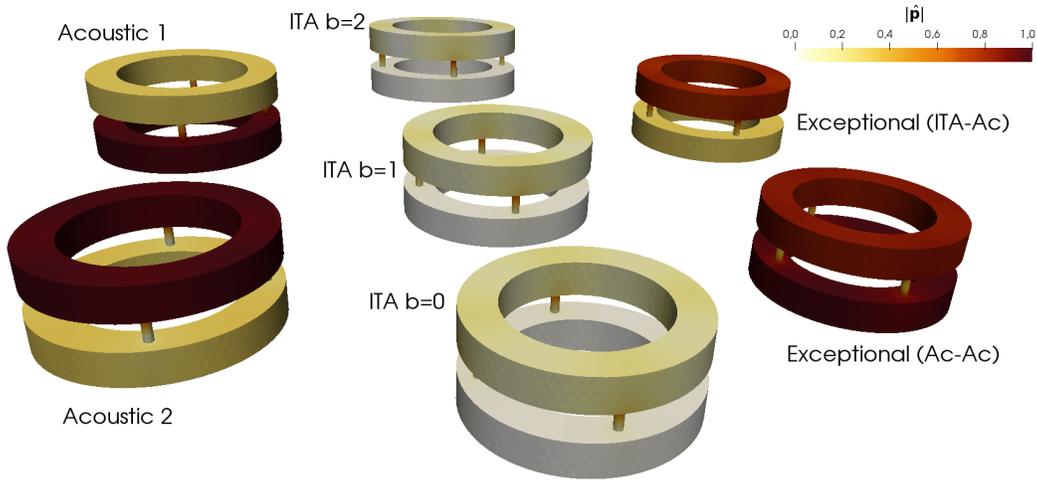


Figure 10: Absolute value of the pressure of various modes found in the annular combustor configuration. Left column: plenum and chamber dominant acoustic modes (see Fig. 9). Middle column: cluster of ITA modes found in the limit $n \rightarrow 0$ (see Fig. 11); the modeshapes are identical and dominant in the flame region, as expected for ITA modes. Right column: modeshapes of the two identified exceptional points (see Fig. 9). One is due to the interaction between two modes of acoustic origin, the other between a mode of acoustic origin and one of ITA origin.

584 perform a single-mode (degenerate) Galerkin expansion. A two-mode
 585 Galerkin expansion that accounts for both the plenum- and chamber
 586 dominant modes should yield a good approximation, as hinted by the
 587 modeshapes shown in both Figs. 7 and 10.

588 III. Another topological change in the eigenvalue trajectories is observed
 589 when $n > 1.44$. The eigenvalues that start close to the chamber acoustic
 590 mode are shifted towards plenum-dominant solutions when τ increases,
 591 as in the previous regime. The same is, however, not anymore true
 592 for the other eigenvalue. The mode that starts close to the plenum-
 593 dominant solution is pushed away from any known acoustic solution,

594 and a trajectory that starts from a solution that is not related to any
595 known acoustic mode (on the right of Fig. 9) ends with a frequency close
596 to the chamber-dominant acoustic mode. In this regime, which has not
597 been discussed before, also a two-mode Galerkin expansion based on
598 the acoustic modes cannot yield a good approximation of the original
599 system, because the thermoacoustic modes can be significantly different
600 from the acoustic ones.

601 From the topological structure of the various eigenvalue trajectories, it is
602 possible to infer that EPs must exist for specific pairs of n and τ . These EPs
603 in fact discriminate between the three regimes just discussed. Starting from
604 educated guesses based on shape of the eigenvalue trajectories, and using the
605 numerical procedure outlined in §3.1, we identify two EPs in the parameter
606 region investigated. One is found for $n_{\text{EP},aa} = 1.17$ and $\tau_{\text{EP},aa} = 8.19$ ms,
607 and discriminates between the aforementioned regimes I and II. The other
608 is found for $n_{\text{EP},ai} = 1.44$ and $\tau_{\text{EP},ai} = 8.91$ ms, and discriminates between
609 regimes II and III.

610 The first EP is labelled with the subscript $_{aa}$ because it results from
611 the collision of two modes of acoustic origin. In [19] only EPs that arise
612 from the interaction of a mode of acoustic origin and one of ITA origin
613 were discussed. EPs arising from two acoustic eigenvalues are already known
614 from the discussions of EPs in acoustic systems, and have a relevance, e.g.,
615 in optimizing the performance of Helmholtz dampers [18, 59]. However, in
616 the current example the coalescence of two eigenvalues of acoustic origin
617 at the EP is driven by the flame response parameters. Such EPs were not
618 discussed in the literature before. The acoustic–acoustic nature of this EP

619 is also visible in its modeshape, shown in Fig. 10: in contrast to the acoustic
 620 modes – purely plenum- or chamber dominant – this modeshape has the
 621 same magnitude in both cavities, suggesting that a single-mode Galerkin
 622 expansion (that preserves only the plenum’s or chamber’s structure) would
 623 not be a suitable approximation in its vicinity.

624 The other EP is labelled with the subscript $_{ai}$ because it originates from
 625 the interaction between modes of acoustic and ITA origin. Its behavior is
 626 analogous to that discussed in §4.1 and [19], as is more evident from Fig. 11:
 627 starting from $n_{EP,ai}$ and decreasing the value of n towards zero, two eigen-
 628 values stem from this EP: one converges to an acoustic solution (that of the
 629 chamber), the other tends towards an ITA mode. Also, because of the dis-
 630 crete rotational symmetry of the annular configuration under investigation,
 631 each eigenvalue shown in Fig. 9 is degenerate with algebraic and geometric
 632 multiplicity 2. As both EPs identified for this configuration are found when
 633 2 eigenvalues (each having algebraic multiplicity 2) coalesce, the EPs have
 634 algebraic multiplicity 4, but are defective in that only two linearly indepen-
 635 dent modeshapes exist, i.e. they have geometric multiplicity 2. Only one
 636 of these modeshapes is shown in Fig. 10, the other is phase-inverted, as for
 637 degenerate (thermo)acoustic modes.

638 Lastly, we verify that, for annular configurations, in the limit $n \rightarrow 0$ the
 639 N_f modes of ITA origin are almost decoupled, as discussed in §2.2.1. Thus,
 640 a cluster of $N_f = 4$ eigenvalues of ITA origin is expected to be found in
 641 the vicinity of the value predicted by Eq. (A.5). Starting from this theoret-
 642 ical guess, a Newton method is employed to identify the close eigenvalues.
 643 Given the fact that the modes cluster, however, it is difficult to identify all

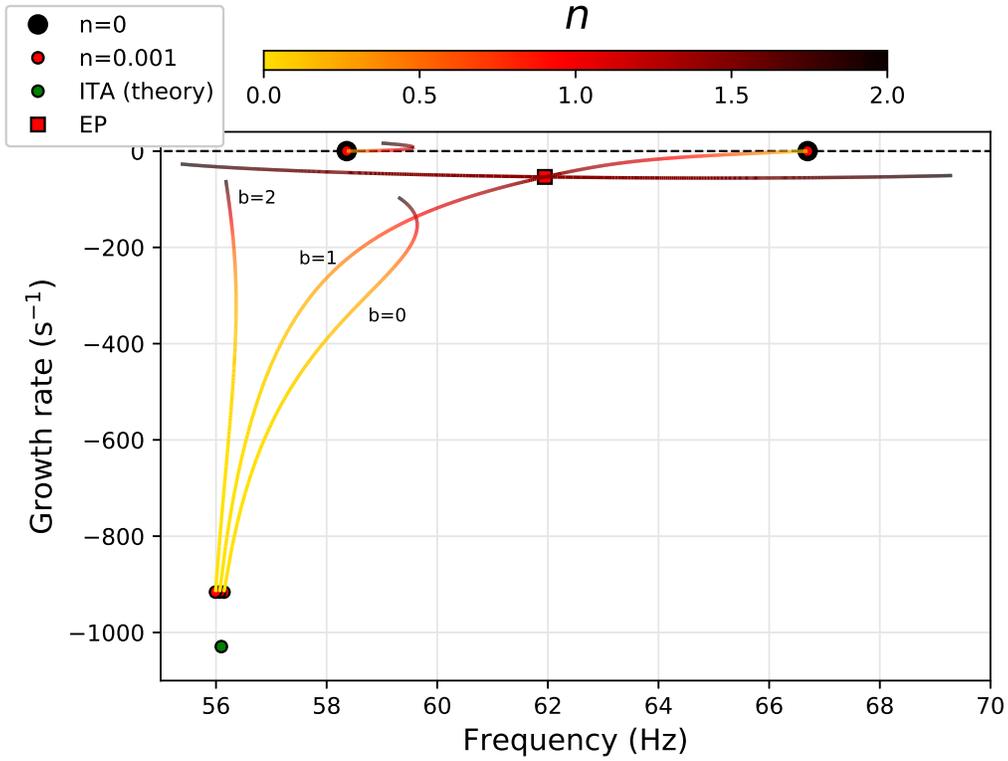


Figure 11: Modes of ITA origin found in the annular setup when $n \rightarrow 0$. The cluster of ITA modes estimated using equation (A.5) for $n = 0.001$ is shown in green. Four thermoacoustic eigenvalues (one for each burner, and two are degenerate due to the symmetries of the annular configuration) are found in its vicinity. The lines track all the eigenvalues for $n \in [0.001, 2]$: two modes of ITA and acoustic origin coalesce at $n_{\text{EP},ai}$.

644 the modes, as the iterative algorithm tends to converge to the same solu-
 645 tion. Bloch-wave theory comes to aid [37]. By using this formalism, the
 646 clustered modes are naturally split across the various Bloch-wave numbers
 647 $b = 0, 1, \dots, N_f - 1$. In the $n \rightarrow 0$ limit, a mode of ITA origin is found for
 648 each Bloch-wave number for the configuration at hand when $n = 0.001$, as
 649 shown in Fig. 11. Because of the mirror symmetry of the system, the modes

650 found for $b = 1$ and $b = 3$ are degenerate for any value of n . In general,
 651 for a system with an even number N_f of burners, the cluster of eigenvalues
 652 found in the $n \rightarrow 0$ limit will be formed by $N_f/2 + 1$ distinct eigenvalues, of
 653 which $N_f/2 - 1$ are degenerate with multiplicity 2. We can then track these
 654 solutions of ITA origin by increasing n towards any desired finite value. The
 655 eigenvalue trajectories are shown in Fig. 11. Because we have fixed the time
 656 delay to $\tau_{\text{EP},ai}$, we identify again the EP of acoustic–intrinsic nature. No-
 657 tably, only the mode of ITA origin with $b = 1$ (or equivalently 3, given the
 658 degeneracy) interacts with the mode of acoustic origin and generates an EP.
 659 This is because modes associated with different Bloch-wave numbers are or-
 660 thogonal [30]. Because both modes of acoustic origin are azimuthal modes
 661 of order $m = 1$, thus associated with Bloch-wave numbers $b = 1$, only the
 662 modes of ITA origin associated with the latter Bloch-wave numbers can in-
 663 teract with them and lead to the formation of EPs, or more generally to
 664 mode veering. This also explains why the ITA modes found in the clustered
 665 region do not interact with each other and neither exhibit veering nor form
 666 EPs despite their eigenvalues being so closed.

667 5. Conclusions

668 In this study, a different perspective to the notion of intrinsic modes
 669 has been presented, with the aim of associating each thermoacoustic mode
 670 with a unique origin without altering the acoustic state (in particular the
 671 reflection coefficients) of the system. We have demonstrated that this is
 672 not possible with the traditional definitions of acoustic and intrinsic modes,
 673 because these definitions are based on two separate parameters: the former

674 are defined in the absence of heat release dynamics, when $n = 0$, whereas
 675 the latter are defined in anechoic conditions, when $R = 0$. Instead, we
 676 propose to use only one parameter, chosen to be n , to define the origin
 677 of all the modes independent of the acoustic boundary conditions of the
 678 system. We have shown in a rather general way that any thermoacoustic
 679 mode can be uniquely associated with one of these two classes of modes when
 680 $n \rightarrow 0$. Furthermore, an explicit expression has been found for the modes of
 681 ITA origin when an $n - \tau$ model is adopted. These are independent of the
 682 acoustic properties even for systems which, despite having anechoic boundary
 683 conditions, still have an acoustic response due to, e.g., area expansions. Their
 684 expressions are functions only of the coefficients of the scattering matrix
 685 \mathbf{S} and the heat release scaling coefficients. In some special cases (absence
 686 of mean flow and cross-sectional area variations across the flame, anechoic
 687 boundary conditions), the definition of ITA modes proposed in this study
 688 coincides with that found in the literature for anechoic conditions. We have
 689 also discussed how, in the case of rotationally symmetric annular combustors,
 690 modes of ITA origin tend to form clusters of eigenvalues in the limit $n \rightarrow 0$,
 691 and generally behave as (weakly) coupled oscillators for finite values of n .

692 The presented theory enables us to theoretically estimate the location
 693 of *all* thermoacoustic eigenvalues in the limit $n \rightarrow 0$, in a given range of
 694 eigenfrequencies. The estimate is based on the numerical identification of
 695 eigenvalues of acoustic origin, found in the vicinity of acoustic modes and
 696 easily obtainable with standard Helmholtz solvers, and eigenvalues of ITA
 697 origin, in the vicinity of theoretically estimated values having large negative
 698 growth rates. Having at hand the solutions in the limit $n \rightarrow 0$, continuation

699 methods can be used to track the trajectories of these eigenvalues to any
700 desired value of n . This reduces the numerical effort needed for identifying
701 large sets of thermoacoustic eigenvalues, and increases the confidence that
702 all modes in a given frequency range have been identified. For finite val-
703 ues of n , the modes of acoustic and ITA origin may interact, giving rise to
704 strong veering of the eigenvalue trajectories. This effect has been related to
705 the existence of exceptional points (EPs) in the spectra of thermoacoustic
706 systems, at which eigenvalues and their corresponding eigenfunctions coa-
707 lesce. Even though the identified EPs always have negative growth rates,
708 we have demonstrated how mode veering in their vicinity is responsible for
709 strong changes in the eigenvalues sensitivities: in some cases, this can cause
710 eigenvalues that are predicted to stabilize by linear stability analysis (for
711 weak flames) to become unstable. In this respect, EPs can be considered as
712 one of the causes of thermoacoustic instabilities, and their identification is
713 practically relevant. A numerical method for the identification of real-valued
714 EPs has been presented, which uses the self-orthogonality property of the
715 defective eigenvalues found at EPs.

716 All the theoretical results presented have also been demonstrated numeri-
717 cally on 3D axial and annular thermoacoustic configurations. The theoretical
718 predictions on the locations of the modes of ITA origin agree well with numer-
719 ical results in all tested cases. Clustering of modes is predicted and observed
720 in annular configurations. EPs have been identified in all configurations and
721 can result from the interaction of (i) modes of acoustic and of intrinsic origin,
722 or (ii) modes of only acoustic origin. No EPs resulting from the interaction
723 between two intrinsic modes have been identified so far. The modeshapes of

724 the EPs contain a strong signature of which modes are responsible for their
725 formation. We have linked the existence of EPs to the topological behavior of
726 the eigenvalue trajectories in parameter space, and related those to regimes
727 that have been previously indicated as “weakly” or “strongly” coupled, as
728 well as identified a new regime which is triggered at moderately high values
729 of the interaction index n . Generally, knowledge on the EPs’ locations leads
730 to the identification of several regimes within which the topological behavior
731 of eigenvalue trajectories is preserved, and to a good qualitative prediction
732 and understanding of the eigenvalue trajectories of thermoacoustic systems
733 in parameter space.

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737 **Appendix A. Poles of the closed thermoacoustic feedback loop** 738 **when $|ne^{-s\tau}| \rightarrow \mathcal{O}(1)$ for $n \rightarrow 0$**

739 Given the expression for the growth rate (12), all the propagation terms
740 $P_j = e^{-s\tau_j}$ diverge to infinity in the limit $n \rightarrow 0$. To find the poles of (11)
741 when the growth rate σ becomes infinitely negative, it is convenient to rewrite
742 it as

$$\hat{u} = \frac{1}{\frac{D(s)}{N(s)} - \mathcal{F}(s)} \hat{q}_n. \quad (\text{A.1})$$

743 The fraction at the denominator of the above equation is an indeterminate
744 form in the $\sigma \rightarrow -\infty$ limit. To solve it, we divide numerator and denominator

745 of Eq. (7) by $P_1 P_2 P_3 P_4$, obtaining

$$\frac{D(s)}{N(s)} = \frac{P_2^{-1} P_3^{-1} R_1 S_{21} + P_1^{-1} P_4^{-1} R_2 S_{12} + R_1 R_2 (S_{22} S_{11} - S_{21} S_{12}) - (P_1 P_2 P_3 P_4)^{-1}}{(P_1^{-1} P_4^{-1} - R_1) [P_2^{-1} P_3^{-1} H_2 + R_2 (H_1 S_{22} - H_2 S_{12})]}. \quad (\text{A.2})$$

746 Considering now the limit $\sigma \rightarrow -\infty$, all the terms containing P_j^{-1} vanish
747 because of the growth rate expression (12). Thus, Eq. (A.2) reduces to

$$\lim_{\sigma \rightarrow 0} \frac{D(s)}{N(s)} = -\frac{R_1 R_2 (S_{22} S_{11} - S_{21} S_{12})}{R_1 R_2 (H_1 S_{22} - H_2 S_{12})} \equiv -\beta, \quad (\text{A.3})$$

748 where we have defined the factor β as a function of the scattering matrix
749 elements S_{ij} and the scaling factors between flame and acoustic responses
750 H_i . Note that the reflection coefficients, which generally include also possible
751 area variations in the regions upstream/downstream of the flame, simplify in
752 the above expressions.

753 The poles of (A.1) in the limit of infinitely negative growth rate are
754 therefore given by

$$\frac{1}{\beta} \mathcal{F}(s) + 1 = 0. \quad (\text{A.4})$$

755 The latter equation can always be solved numerically, for arbitrary expres-
756 sions of the FTF. In the special case in which the flame response can be
757 modelled with an $n - \tau$ model, $\mathcal{F}(s) = n e^{-s\tau}$, analytical solutions can be
758 found:

$$s = \frac{1}{\tau} \log \left(\frac{n}{\beta} \right) + \frac{(2k+1)\pi}{\tau} \mathbf{i}, \quad k \in \mathbb{Z}. \quad (\text{A.5})$$

759 The angular frequencies of these solutions are identical to those of the in-
760 trinsic ones, as per Eq. (5). The expression for the growth rate is consistent
761 with that obtained in the asymptotic limit (see Eq. (12)), with $\alpha = \beta$, which
762 makes the solutions valid. In the special case in which no mean flow effects

763 are considered, it can be proven that $\beta = \frac{1}{H_2}$, so that the dispersion rela-
764 tion (A.4) is formally equivalent to that of the pure ITA modes (10), and so
765 are the eigensolutions. This was implicitly shown in [14], but can be derived
766 from first principles given the explicit expressions of the scattering matrix
767 elements.

768 In summary, we have proven that there always exists a set of modes
769 in the $n \rightarrow 0$ limit which are infinitely damped, and whose frequencies are
770 identical to those of the pure ITA modes, which are found when the reflection
771 coefficients are set to zero.

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