

# Thermoacoustic modes of intrinsic and acoustic origin and their interplay with exceptional points

Alessandro Orchini<sup>a,\*</sup>, Camilo F. Silva<sup>b</sup>, Georg A. Mensah<sup>c</sup>, Jonas P.  
Moeck<sup>d</sup>

<sup>a</sup>*Institut für Strömungsmechanik und Technische Akustik, Technische Universität Berlin,  
Berlin, Germany*

<sup>b</sup>*Fakultät für Maschinenwesen, Technische Universität München, Garching, Germany*

<sup>c</sup>*CAPS Laboratory, Department of Mechanical and Process Engineering, ETH Zürich,  
Zurich Switzerland*

<sup>d</sup>*Department of Energy and Process Engineering, Norwegian University of Science and  
Technology, Trondheim, Norway*

---

## Abstract

We propose a general classification of all the modes of a given thermoacoustic system into two sets: one of acoustic origin and one of intrinsic thermoacoustic (ITA) origin. To do this, the definition of intrinsic modes, traditionally based on anechoic boundary conditions, is reformulated in terms of the gain  $n$  of the Flame Transfer Function (FTF). As a consequence of this classification, we show how theoretical results for the estimation of all thermoacoustic modes can be derived in the limit  $n \rightarrow 0$ , for both axial and annular combustors, independent of the acoustic boundary conditions. Starting from this limit and using standard continuation methods while increasing  $n$ , all the eigenvalues of interest in a given domain in the frequency space can be identified. We also discuss how thermoacoustic modes of acoustic and ITA origin can interact, and in some cases coalesce generating exceptional points (EPs).

---

\*Corresponding Author

*Email address:* [a.orchini@campus.tu-berlin.de](mailto:a.orchini@campus.tu-berlin.de) (Alessandro Orchini)

Although all EPs found have negative growth rates, in their vicinity thermoacoustic eigenmodes have very large sensitivities and exhibit strong mode veering. We demonstrate how, in some cases, mode veering is responsible for the occurrence of thermoacoustic instabilities, and propose a numerical method to identify EPs. All the theoretical results are numerically verified using two generic thermoacoustic configurations.

*Keywords:* Thermoacoustics, Intrinsic modes, Exceptional point, Combustion instabilities

---

## 1. Introduction

Intrinsic thermoacoustic (ITA) instabilities were first recognized in [1, 2], where it was shown that even a thermoacoustic system with anechoic boundary conditions can exhibit thermoacoustic instabilities. Analytical models of ITA instabilities were developed in [3, 4], which allowed for their connection to an intrinsic feedback loop that does not require acoustic reflections at boundaries. Using an  $n-\tau$  model, it was shown that the ITA resonance frequency can be calculated from the time delay of the acoustic flame response only, and that ITA resonance frequencies are directly related to the peaks of the elements of the scattering matrix [5, 6].

Direct numerical simulations of anechoic systems that exhibited ITA instabilities [7, 8], together with some experimental evidence [2, 9], further corroborate that ITA instabilities are indeed physical, and not just the result of a mathematical artifact. The physical mechanism governing ITA instabilities relies on the feedback between acoustic waves generated by unsteady heat release rate and the latter being sensitive to acoustic velocity fluctua-

17 tions upstream of the flame [10, 11]. ITA instabilities do not require ideal  
18 anechoic conditions to exist: in [12, 13] it has been demonstrated that this  
19 kind of instability is relevant in thermoacoustic systems with partially re-  
20 flecting boundary conditions. In [14] it was shown that, for a fully reflective  
21 Rijke-tube like systems with an  $n-\tau$  flame response model, the resonance  
22 frequency characterizing the ITA feedback loop for  $n \rightarrow 0$  is the same as the  
23 one obtained in anechoic systems. In [15] the evolution of thermoacoustic  
24 eigenfrequency trajectories from fully reflecting to anechoic conditions was  
25 reported. Some trajectories are pushed to infinitely negative growth rates  
26 in the anechoic limit, others retain a finite frequency and growth rate. The  
27 former were called ITA eigenfrequencies (for nonzero reflecting conditions)  
28 and the latter “pure ITA” eigenfrequencies (in the anechoic limit).

29 One objective of the present study is demonstrating that the conclusions  
30 derived for ITA modes in simple configurations (straight Rijke tubes) and/or  
31 with simple flame models ( $n-\tau$ ) can be generalized to a much larger set of  
32 thermoacoustic systems. We will show that a clear distinction between “ITA  
33 driven” and “acoustic driven” thermoacoustic modes (as in [15]) is not al-  
34 ways possible when using the definitions of ITA and acoustic modes given  
35 in the literature; an alternative definition that allows for this distinction is  
36 proposed. Furthermore, we will show that the interplay between modes of  
37 acoustic and ITA origin leads to the existence of exceptional points (EPs).  
38 EPs have been identified as the fundamental concept at the base of many  
39 scientific phenomena in several fields, including non-Hermitian quantum me-  
40 chanics, optics and acoustics [16–18]. In thermoacoustics, they have only  
41 been recently discussed [19]. At EPs two or more eigenvalues coalesce and

42 so do their corresponding modeshapes. The resulting eigenvalue is, thus,  
43 degenerate and defective. In the simplest case, a defective eigenvalue has  
44 algebraic multiplicity two and geometric multiplicity one. This is accompa-  
45 nied by special properties like infinite parameter sensitivity. The effects of  
46 EPs on thermoacoustic eigenvalue trajectories is the second main objective  
47 of the study. In particular, we will demonstrate how thermoacoustic eigen-  
48 modes can become unstable because of strong mode veering caused by their  
49 interaction with EPs, even if the latter have negative growth rates. The iden-  
50 tification of EPs, thus, helps understanding the origin of some thermoacoustic  
51 instabilities.

52 In §2, the definition of ITA modes is revised, and the findings of [14]  
53 are generalized to arbitrary thermoacoustic systems. It is also shown that a  
54 given thermoacoustic mode could be thought of as originating from either an  
55 acoustic or a “pure ITA” mode, depending on whether a thermoacoustic con-  
56 figuration is considered to be originating from (i) an acoustic cavity in which  
57 the strength of the flame is gradually increased, or (ii) an anechoic com-  
58 bustion chamber in which the reflection coefficients are gradually increased.  
59 In §3, this ambiguity is tied to the existence of EPs. A general strategy to  
60 numerically identify EPs is proposed, and their effect on the trajectories of  
61 thermoacoustic modes is discussed. The theory is demonstrated in §4 using  
62 two thermoacoustic configurations modeled with the 3D Helmholtz equa-  
63 tion. In §4.1 an axial combustor is considered, and EPs stemming from the  
64 interaction between acoustic and ITA modes are found; in §4.2 an annular  
65 combustor is considered, and an EP is identified as the coalescence of two  
66 thermoacoustic modes of acoustic origin.

67 **2. Acoustic, intrinsic, and thermoacoustic modes**

68 Thermoacoustic phenomena arise from the interaction between unsteady  
 69 heat release rate and acoustic fluctuations. The linear stability of thermoacoustic systems in the low-Mach-number limit can be assessed by investigating the eigenvalues of the Helmholtz equation with a heat release source  
 70 term [20, 21]. In the frequency domain it reads [22, 23]  
 71  
 72

$$\nabla \cdot (c^2 \nabla \hat{p}) - s^2 \hat{p} - \frac{(\gamma - 1) \bar{q}(\mathbf{x})}{\bar{\rho} \bar{u}} \mathcal{F}(s) \nabla \hat{p}_{\text{ref}} \cdot \hat{\mathbf{n}}_{\text{ref}} = 0, \quad (1)$$

73 where  $\hat{p}$  represents the pressure fluctuations in frequency domain,  $\gamma$  the heat  
 74 capacity ratio,  $c$  the local speed of sound,  $\bar{\rho}$  the mean density, and  $s \equiv$   
 75  $\sigma + i\omega$  the Laplace variable – where  $\sigma$  is the growth rate and  $\omega \equiv 2\pi f$   
 76 is the angular frequency – representing the eigenvalues of interest.  $\mathcal{F}(s)$  is  
 77 the Flame Transfer Function (FTF), which represents the linear response  
 78 of heat release rate fluctuations resulting from perturbations in the acoustic  
 79 velocity field at a reference location, indicated with the subscript  $(\ )_{\text{ref}}$ . The  
 80 heat release is located only in a (compact) sub-domain of the total volume,  
 81 where the local mean heat release,  $\bar{q}(\mathbf{x})$  in Eq. (1), is non-zero. Here we are  
 82 implicitly assuming that the unsteady heat release rate is proportional to  
 83 the mean one, which is not strictly necessary but inconsequential due to the  
 84 acoustic compactness of the flame.

85 The FTF is often expressed in terms of the  $n$ - $\tau$  model

$$\mathcal{F}(s) \equiv \frac{\bar{u} \hat{q}}{\bar{Q} \hat{u}} \equiv n e^{-s\tau}, \quad (2)$$

86 where  $n$  and  $\tau$  represent the heat release gain and time-lag with respect to  
 87 velocity fluctuations, and  $\bar{u}$  and  $\bar{Q}$  are the mean characteristic velocity and

88 mean global heat release rate, respectively. A non-dimensional impedance  $Z$   
 89 specifies the boundary conditions:

$$\hat{p} + \frac{cZ}{s} \nabla \hat{p} \cdot \hat{\mathbf{n}} = 0. \quad (3)$$

90 Traditionally, thermoacoustic modes have been understood as perturba-  
 91 tions of purely acoustic modes (as defined in §1). Consequently, their eigen-  
 92 frequencies and modeshapes have been sought in the vicinity of those of the  
 93 same system without any unsteady heat release rate ( $\mathcal{F} = 0$  in eq. (1)). By  
 94 this assumption, Galerkin mode expansions of thermoacoustic modes based  
 95 on the acoustic modes have been proposed [24]. The recently discovered in-  
 96 trinsic thermoacoustic modes, however, show that this is not always appropri-  
 97 ate. Thermoacoustic oscillations can have frequencies which are not directly  
 98 related to the purely acoustic eigenfrequencies of the combustor [1, 9]. This  
 99 is because these modes arise from the feedback loop created by upstream  
 100 traveling acoustic perturbations generated by the flame, which trigger veloc-  
 101 ity fluctuations upstream of the flame [10, 11, 25]. These in turn lead to  
 102 the generation of heat release rate fluctuations (see Fig. 2). Because this  
 103 mechanism does not require any interaction with acoustic waves reflected at  
 104 the boundaries, its associated modes have been labelled ITA modes.

### 105 *2.1. Origin of thermoacoustic modes*

106 From the literature, modes of ITA origin are defined to be those associated  
 107 with the eigenfrequency loci that contain pure ITA eigenfrequencies in the  
 108 anechoic limit [26]. Equivalently, modes of acoustic origin are defined to be  
 109 those associated with the eigenfrequency loci that contain purely acoustic  
 110 eigenfrequencies in the  $n \rightarrow 0$  limit. It appears therefore meaningful that a

111 given thermoacoustic mode can be referred to as of acoustic origin or of ITA  
 112 origin [15]. There is, however, an inconsistency between the definitions given  
 113 above and the idea of classifying thermoacoustic modes depending on their  
 114 origin, which we now demonstrate.

115 For this purpose, we consider a Rijke tube configuration: a straight duct  
 116 of length 0.5 m with a temperature jump  $T_2/T_1 = 4$  across a flame element  
 117 located in the middle of the tube. Explicit expressions for the dispersion  
 118 relation of the thermoacoustic eigenvalues can be found in this configuration  
 119 using network approaches (see [27–29] and §1 in the supplementary material).

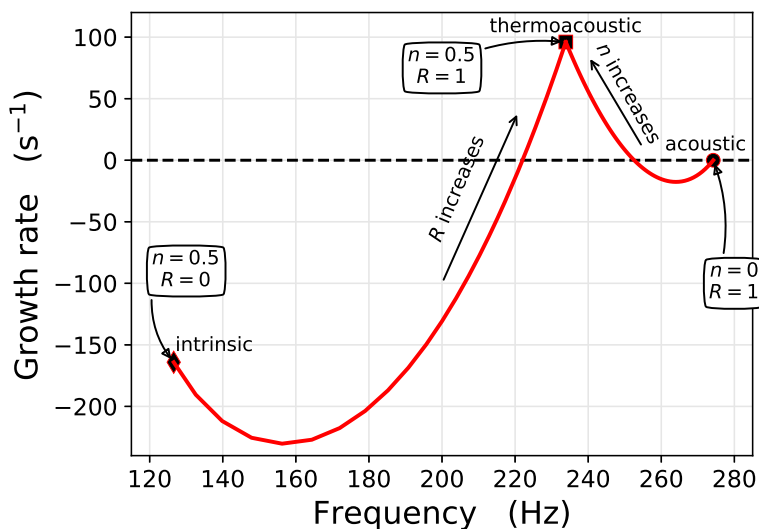


Figure 1: Thermoacoustic mode trajectories while varying the reflection coefficient  $R$  or the interaction index  $n$ . Depending on which parameter is varied towards zero, the thermoacoustic mode converges to either an acoustic or a pure ITA mode. This illustrates the need for an unambiguous definition of the origin of the thermoacoustic mode origin.

120 We model the Rijke tube using finite elements, solving the nonlinear  
 121 eigenvalue problem (1). We use continuation methods – based on high-order

122 adjoint-based eigenvalue sensitivity [30] – to track the evolution of a specific  
 123 eigenvalue while varying two parameters: the flame interaction index  $n$ , and  
 124 the magnitude of the reflection coefficients  $|R|$ . The flame time delay is fixed  
 125 to  $\tau = 3.96$  ms. In the fully reflective case,  $|R| = 1$ , the tube is assumed to  
 126 be acoustically closed ( $R = 1$ ) in the cold region, and open ( $R = -1$ ) in the  
 127 hot one. The corresponding non-dimensional impedances (3) are calculated  
 128 as

$$Z = \frac{1 + R}{1 - R}. \quad (4)$$

129 We start by setting  $|R| = 1$  (so that the up- and downstream impedances  
 130 are  $\infty$  and 0, respectively), and  $n = 0$ . Then, only purely acoustic modes  
 131 exist, featuring zero growth rate as no damping mechanisms are modeled.  
 132 An acoustic mode with an angular frequency of 274 Hz is found (see Fig. 1).  
 133 This value agrees well with that predicted from low-order network models  
 134 (see supplementary material §1). We then vary the interaction index  $n$  and  
 135 track the eigenvalue evolution. Note that, for small values of the interac-  
 136 tion index the flame damps the mode, but for larger values this trend is  
 137 reversed and the mode becomes unstable. Once we have reached  $n = 0.5$ , a  
 138 reasonable value for a flame response, we maintain this value and vary the  
 139 reflection coefficient magnitudes  $|R|$  from 1 to 0. The up- and downstream  
 140 impedances are calculated according to Eq. (4), with  $R$  positive (negative)  
 141 in the upstream (downstream) region. When  $|R| = 0$ , the mode is a pure  
 142 ITA mode. There are analytical expressions available from the literature for  
 143 pure ITA modes in Rijke tubes (see [1, 3] and supplementary material §1).



144 These are given by

$$s_{\text{ITA}} \equiv \frac{1}{\tau} \ln \left[ \left( \sqrt{\frac{T_2}{T_1}} - 1 \right) n \right] + \frac{(2k+1)\pi}{\tau} i, \quad k \in \mathbb{Z}. \quad (5)$$

145 For the chosen values of  $n$  and  $\tau$ , the resulting intrinsic frequency and growth  
 146 rate are  $f_{\text{ITA}} = 1/(2\tau) \approx 126$  Hz and  $\sigma_{\text{ITA}} = -175 \text{ s}^{-1}$ , which agree well with  
 147 the values obtained from the Helmholtz model (see Fig. 1).

## 148 2.2. An alternative definition of intrinsic modes

149 The previous example indicates that there exist thermoacoustic modes  
 150 that can be arbitrarily associated to either acoustic or pure ITA eigenfre-  
 151 quencies. This depends on whether the parameter responsible for the ther-  
 152 moacoustic coupling is considered to be  $n$  or  $R$ . With the given definitions of  
 153 acoustic modes ( $n = 0$ ) and pure intrinsic modes ( $R = 0$ ), it is therefore am-  
 154 biguous to think of a thermoacoustic mode as of “acoustic” or “ITA” origin.  
 155 In order to assign to thermoacoustic modes a specific source, the definition  
 156 of acoustic and intrinsic modes must rely on *one* parameter.

157 A step in this direction has been done in [14], considering a 1D Rijke  
 158 tube with an  $n$ - $\tau$  flame response. Damping, mean flow and entropy-wave  
 159 effects have been neglected. For this configuration a dispersion relation can  
 160 be derived, whose zeroes represent the thermoacoustic modes. Using it, it  
 161 was shown that, when steering  $n$  towards zero, any thermoacoustic mode  
 162 will either reduce to an acoustic mode or move towards an infinitely damped  
 163 mode, with the same angular frequency as an intrinsic mode, regardless of  
 164 the boundary conditions. Thus, using only the parameter  $n$  it is still possible  
 165 to identify intrinsic eigenfrequencies, and uniquely associate an origin to each  
 166 thermoacoustic mode.

167 These findings are in fact much more general. In the following, we will  
 168 present a derivation of these results with a method different from that of [14].  
 169 Our proof is independent of the dispersion relation and, thus, valid for con-  
 170 figurations with mean flow, damping, arbitrary reflection coefficients, and  
 171 arbitrary expression for the FTF.

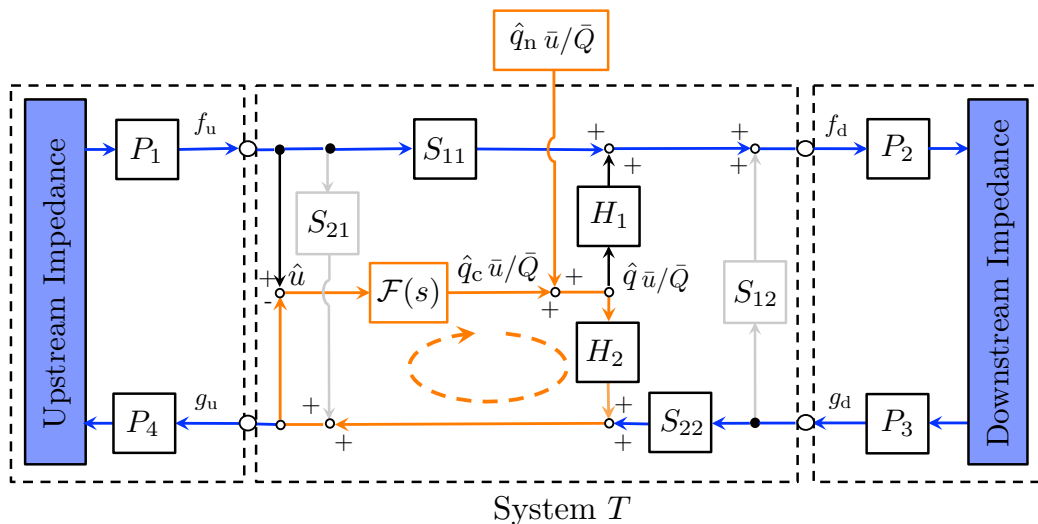


Figure 2: Block-diagram representation of a generic axial combustor. The upstream and downstream impedances can be complex-valued and frequency dependent, so that they can be used to model arbitrarily shaped volumes. The flame is considered compact, so that jump conditions across it can be expressed in terms of a scattering matrix  $\mathbf{S}$ . The ITA feedback loop is highlighted in orange, and does not involve any interaction with the acoustic boundary conditions.

172 We consider an arbitrary axial combustor, in which only plane waves  
 173 propagate, containing an acoustically compact flame. No assumption is made  
 174 on the actual shape of the setup, presence of a mean flow, or the acoustic  
 175 boundary conditions. Such a generic configuration can still be represented in  
 176 block form (Fig. 2), as commonly done in control theory and network model

177 approaches. The propagation blocks,  $P_j = e^{-s\tau_j}$ , transport the acoustic  
 178 waves from a location  $x$  to  $x + \Delta x$ .  $\tau_j$  is a characteristic acoustic propagation  
 179 time delay [20], proportional to  $\Delta x$ , and generally a function of the mean  
 180 flow. Damping models add an imaginary term in the exponential of the  
 181 propagating terms, multiplying  $s$ . The jump conditions across the compact  
 182 flame are contained in the coefficients of the scattering matrix, relating the  
 183 incident and outgoing acoustic waves:

$$\begin{bmatrix} f_d \\ g_u \end{bmatrix} = \mathbf{S} \begin{bmatrix} f_u \\ g_d \end{bmatrix} + \mathbf{H}\hat{q} = \begin{bmatrix} S_{11} & S_{12} \\ S_{21} & S_{22} \end{bmatrix} \begin{bmatrix} f_u \\ g_d \end{bmatrix} + \begin{bmatrix} H_1 \\ H_2 \end{bmatrix} \hat{q}. \quad (6)$$

184 The factors  $H_j$  account for the scaling between the heat release rate fluc-  
 185 tuations and the acoustic waves, and are not frequency dependent (see sup-  
 186plementary material). The coefficients of the scattering matrix,  $S_{ij}$ , can in  
 187 general be function of the frequency, when losses or inertial effects in terms  
 188 of effective lengths are considered. However, it is standard in the analysis of  
 189 intrinsic modes in thermoacoustics to assume that they are not, to allow for  
 190 analytical treatment. The coefficients of  $\mathbf{S}$  are derived from jump conditions,  
 191 which are conservative and show no frequency dependence. Their expression  
 192 depends on the presence/absence of an area increase/decrease, temperature  
 193 jump, entropy and/or vorticity waves. The theory presented here is gen-  
 194eral in this respect, and the definition of the coefficients  $S_{ij}$  is kept implicit;  
 195 an example of their expression is provided in the supplementary material.  
 196 Lastly, the response of the acoustic configuration upstream and downstream  
 197 of the flame is modeled by means of arbitrary impedances, which are con-  
 198verted into reflection coefficients  $R$  via Eq. (4). These can be complex-valued  
 199 and/or frequency dependent, and may contain the effects of area variations.

200 From Fig. 2, considering the balance of the waves at each node, it can be  
 201 shown that the response of acoustic velocity fluctuations  $\hat{u}$  to heat release  
 202 rate fluctuations  $\hat{q}$  is

$$\hat{u} = \frac{(1 - P_1 P_4 R_1)[H_2 + (P_2 P_3 R_2)(H_1 S_{22} - H_2 S_{12})]}{P_1 P_4 R_1 S_{21} + P_2 P_3 R_2 S_{12} + P_1 P_2 P_3 P_4 R_1 R_2 (S_{22} S_{11} - S_{21} S_{12}) - 1} \hat{q}. \quad (7)$$

203 Assuming that all the components of (7) are analytic functions of the eigen-  
 204 frequency  $s$ , its numerator,  $N(s)$ , does not have poles. Thus, the *acoustic*  
 205 eigenfrequencies are those for which the denominator of (7),  $D(s)$ , vanishes:

$$D(s) := P_1 P_4 R_1 S_{21} + P_2 P_3 R_2 S_{12} + P_1 P_2 P_3 P_4 R_1 R_2 (S_{22} S_{11} - S_{21} S_{12}) - 1 = 0 \quad (8)$$

207 The heat release rate response,  $\hat{q}_c/\bar{Q} = \mathcal{F}(s)\hat{u}/\bar{u}$ , is **assumed to have**  
 208 **no finite poles, for simplicity, although this assumption could be relaxed.**  
 209 This is true for many traditional flame response approximations, including  
 210  $n - \tau$  models with constant or polynomial coefficients, and more generally for  
 211 sum of time delay models, which can well fit Flame Transfer and Describing  
 212 Functions [31, 32].

213 The ITA loop of the system is highlighted in Fig. 2. It is characterized  
 214 by the transfer function

$$\hat{q} = \frac{1}{H_2 \mathcal{F}(s) + 1} \hat{q}_n, \quad (9)$$

215 where  $\hat{q}_n$  can be understood as a source of combustion noise [12]. This implies  
 216 that ITA modes are found when

$$H_2 \mathcal{F}(s) + 1 = 0, \quad (10)$$

217 which is equivalent to definitions found in the literature [1, 3] for a Rijke  
 218 tube with anechoic boundary conditions. Coupling the acoustic and flame  
 219 responses yields the closed-loop transfer function

$$\hat{u} = \frac{N(s)}{D(s) - N(s)\mathcal{F}(s)}\hat{q}_n. \quad (11)$$

220 The *thermoacoustic* modes are found<sup>1</sup> when  $D(s) - N(s)\mathcal{F}(s) = 0$ .

221 One can then track these eigenvalues by slowly varying the gain of the  
 222 FTF from a finite value towards zero. In the following, we shall assume  
 223 that  $\mathcal{F}(s) = ne^{-s\tau}$ , as this permits a direct comparison of our results with  
 224 those available in the literature, and eases the notation. Two scenarios are  
 225 possible:

- 226 1.  $\lim_{n \rightarrow 0} |ne^{-s\tau}| \rightarrow 0$ . Then, for  $s$  to be a pole of Eq. (11), the condition  
 227  $D(s) = 0$  must be satisfied. This coincides with the acoustic modes, as  
 228 per Eq. (8);
- 229 2.  $\lim_{n \rightarrow 0} |ne^{-s\tau}| \rightarrow \mathcal{O}(1)$ . This is possible if and only if, asymptotically,  
 230  $e^{-\sigma\tau} \sim \alpha/n$ , with  $\alpha \in \mathbb{R}$ . This value is not arbitrary, but can be linked  
 231 to the elements of the scattering matrix and the heat release scaling  
 232 coefficients  $H_j$  (see [Appendix A](#)). This implies that

$$\sigma \sim \frac{1}{\tau} \log(n/\alpha) \quad \text{as } n \rightarrow 0, \quad (12)$$

233 and that  $\lim_{n \rightarrow 0} ne^{-(\sigma+i\omega)\tau} \propto e^{-i\omega\tau}$ . In other words, the infinite growth rate  
 234 of the time-delayed terms is balanced by the vanishing flame strength  
 235 when  $n \rightarrow 0$ . We now want to find those values of  $s$  that are poles

---

<sup>1</sup>This assumes that no zero-pole cancellations, which are nonetheless extremely rare, occurs; otherwise, some extra modes are identified.

236 of (11) in this limit. It can be shown (see [Appendix A](#)) that the  
237 angular frequencies of these infinitely damped modes are identical to  
238 those of ITA modes (Eq. (5)).

239 This proves that, in the  $n \rightarrow 0$  limit, thermoacoustic modes are split into  
240 two distinct sets. One of them is equivalent to the set of acoustic modes, the  
241 other is connected to the ITA modes (with infinite damping), regardless of  
242 the boundary conditions.

243 We remark that the intrinsic loop highlighted in Fig. 2 and defined in  
244 Eq. (10) exists in an isolated fashion only when no area variations are present  
245 in the volumes upstream/downstream of the flame. Otherwise, even when  
246 purely anechoic boundary conditions are imposed, reflection of acoustic waves  
247 will occur due to the area changes. This modifies the anechoic intrinsic loop  
248 and the consequent definition of pure intrinsic modes, as discussed in [7]. This  
249 is, however, unimportant in the current analysis because only the limit  $n \rightarrow 0$   
250 is considered. The amplitude of the ITA generated waves that propagate  
251 away from the flame vanish in this limit because they are infinitely damped,  
252 regardless of the presence of area variations. As a consequence of this, the  
253 definition of intrinsic modes for vanishing  $n$  depends only on the values of  
254 the scattering matrix and flame scaling coefficients, as shown in [Appendix](#)  
255 [A](#).

256 Using a single parameter to define both acoustic and ITA frequencies  
257 allows for unambiguously associating each thermoacoustic mode with a spe-  
258 cific origin. This is not possible when different parameters are used to define  
259 ITA and acoustic modes. Here, the chosen parameter is the flame gain (as  
260 it retains the natural definition of acoustic modes), but the reflection coeffi-

261 cient could be chosen equivalently (which would lead to a different, but still  
262 unique, classification of the eigenvalues).

263 Furthermore, the results of this section have an important consequence  
264 for practical applications and the numerical calculation of thermoacoustic  
265 modes. Commonly, thermoacoustic modes are found by initializing Helmholtz  
266 solvers with the purely acoustic frequencies (found when  $n = 0$ ) and then  
267 by gradually increasing the value of  $n$  and track the evolution of the eigen-  
268 values [33]. However, this method fails in identifying all thermoacoustic  
269 modes, as some modes are of ITA origin. Using the results of this proof,  
270 we have a new set of guesses that can be used to identify the remaining  
271 thermoacoustic modes: starting from a small but non-zero value of the in-  
272 teraction index, thermoacoustic modes of intrinsic origin are approximately  
273 given by Eq. (A.5). No other modes can be found because, when  $n \rightarrow 0$ ,  
274 all modes *must* belong to one of these two sets. Once located in this limit,  
275 all thermoacoustic eigenvalues can be tracked to the desired value of  $n$  us-  
276 ing continuation methods. Thus, with this strategy the space that needs to  
277 be explored is limited to that in the vicinity of the theoretically estimated  
278 solutions. This leads to a gain in both the numerical time needed to locate  
279 the thermoacoustic modes, and in the confidence that all modes (in a given  
280 frequency range) have been identified. This will be demonstrated in §4.

### 281 *2.2.1. The annular combustor case*

282 In this section we qualitatively discuss how the results of §2.2 can be  
283 extended to annular and can-annular thermoacoustic configurations. In par-  
284 ticular, we will consider systems featuring a discrete rotational symmetry.  
285 Such systems are generally modeled with two acoustic volumes connected by

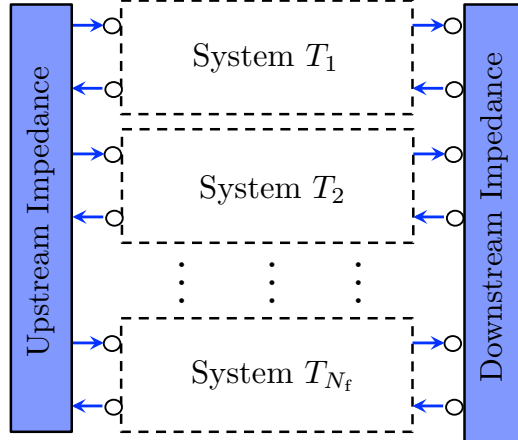


Figure 3: Block-diagram representation of an annular combustor with discrete rotational symmetry and  $N_f$  flames. The flames are located in ducts that connect upstream and downstream acoustic cavities (annular plenum and annular combustion chamber).

286  $N_f$  ducts, in which identical flames are located (see Fig. 3). The acoustic  
 287 boundary conditions couple the responses of the various ducts, in which the  
 288 acoustic field is assumed to be one dimensional, and can be modeled with  
 289 impedance matrices [34, 35]. These contain the acoustic response in all the  
 290 ducts when the acoustic field in a given duct is excited. Given the rota-  
 291 tional symmetry of the system, these matrices are circulant, which has direct  
 292 connections with a possible Bloch representation of the dynamics [36–38].

293 In the limit  $n \rightarrow 0$ , the same results of the previous section must hold for  
 294 each duct. In this limit, there is no physical mechanism that couples the  $N_f$   
 295 intrinsic loops. Since the flames in the various ducts are identical, so are all  
 296 intrinsic loops, which are still governed by the dispersion relation (10). This  
 297 results in  $N_f$  identical intrinsic eigenfrequencies, infinitely damped, and with  
 298 the same frequency of pure ITA modes. This discussion is kept qualitative  
 299 because, given the matrix formulation needed to represent annular systems



300 with discrete symmetries, a closed-form formulation of analytical results is  
301 impractical.

302 These intrinsic loops can be thought of as a set of identical, decoupled  
303 oscillators. A weak coupling between them is achieved by either consider-  
304 ing small but non-zero up- and downstream reflection coefficients when the  
305 flame interaction index is finite, or a small but non-zero flame interaction  
306 index when the reflection coefficients are finite. In both cases, the dynam-  
307 ics of ITA modes is governed by weakly coupled oscillators. When identical  
308 oscillators are weakly coupled, the eigenvalues of the weakly coupled system  
309 form clusters of closely spaced eigenvalues [39]. This is the case, for example,  
310 in can-annular systems, in which the acoustic coupling between the various  
311 cans is weak, and clustering of thermoacoustic modes can be observed [40].  
312 The same holds true for modes of ITA origin in annular systems. This has  
313 been first observed numerically in [41], where clusters of modes with eigen-  
314 frequencies close to those of ITA modes were found.

### 315 **3. Interaction of acoustic and intrinsic modes with exceptional** 316 **points**

317 In the previous section, we have discussed how the origin of thermoacous-  
318 tic modes can be assessed using only the interaction index  $n$  of the flame.  
319 There remain however some points in the spectrum of thermoacoustic systems  
320 that elude this classification. These points are known as exceptional points  
321 (EPs). EPs are a particular type of degenerate, defective eigenvalues, with  
322 the additional property of being singularities in the parameter space [42]. EPs  
323 have recently been identified in the spectrum of a Rijke tube-like thermo-

324 coustic system using an explicitly known dispersion relation [19]. Here, we  
 325 show how EPs of any thermoacoustic systems can be identified numerically  
 326 without using the dispersion relation, but the self-orthogonality property of  
 327 the eigenfunctions at EPs.

### 328 3.1. Self-orthogonality

329 The eigenvalue sensitivity of thermoacoustic modes w.r.t. a parameter  $\xi$   
 330 is given by [43, 44]:

$$\frac{\partial s_j}{\partial \xi} = - \frac{\langle \hat{\mathbf{p}}_j^\dagger | \frac{\partial \mathbf{L}}{\partial \xi} \hat{\mathbf{p}}_j \rangle}{\langle \hat{\mathbf{p}}_j^\dagger | \frac{\partial \mathbf{L}}{\partial s} \hat{\mathbf{p}}_j \rangle} \Big|_{s=s_j}, \quad (13)$$

331 where the adjoint eigenvectors have been denoted with the superscript  $\dagger$ ,  
 332 and the matrix  $\mathbf{L}$  contains the discretization of the thermoacoustic eigen-  
 333 value problem (1). Equation (13) is valid whenever the denominator is non-  
 334 zero. This is always guaranteed to be the case for non-defective eigenvalues  
 335 (even if they are degenerate), but it is zero for defective eigenvalues [45]. In  
 336 fact, the derivation of equation (13) assumes that a bi-orthonormal set of  
 337 direct/adjoint eigenfunctions can be chosen [46]:

$$\langle \hat{\mathbf{p}}_i^\dagger | \frac{\partial \mathbf{L}}{\partial s} \hat{\mathbf{p}}_j \rangle = \delta_{i,j}. \quad (14)$$

338 This breaks down at defective points, because the basis of the eigenvectors is  
 339 incomplete. In particular, it is possible to show that, for defective eigenvalues,  
 340 the direct and corresponding adjoint eigenvectors satisfy [47]

$$\langle \hat{\mathbf{p}}_{\text{def}}^\dagger | \frac{\partial \mathbf{L}}{\partial s} \hat{\mathbf{p}}_{\text{def}} \rangle = 0. \quad (15)$$

341 This property is known as self-orthogonality. At EPs, it manifests itself in  
 342 infinite eigenvalue sensitivity. We reference to the supplementary material §2  
 343 for remarks on numerical aspects of self-orthogonality.

344 *3.1.1. General method for the identification of exceptional points*

345 We exploit the infinite eigenvalue sensitivity at EPs to devise a general  
 346 strategy for their identification in thermoacoustic systems. At EPs, we have

347

$$\lim_{\xi \rightarrow \xi_{\text{EP}}} \left| \frac{\partial s}{\partial \xi} \right|^{-1} = 0 \quad (16)$$

348 Thus, the identification of EPs is reduced to a root-finding problem, which  
 349 can be straightforwardly solved numerically with iterative methods. Note  
 350 that, every time the parameter  $\xi$  is updated in the iterative scheme, a new  
 351 eigenvalue problem needs to be solved, and the sensitivity can then be calcu-  
 352 lated using Eq. (13). Furthermore, since the eigenvalues of thermoacoustic  
 353 problems are generally complex-valued, also the value of the parameter  $\xi$  at  
 354 which the EP is found using this strategy can be complex-valued. These  
 355 complex-valued parameters may or may not be physically realizable: an EP  
 356 found in a Rijke tube having a complex-valued length would not be realizable,  
 357 but one found for a complex-valued impedance would. In order to identify  
 358 EPs for real-valued parameters, we need to extend the parameter space un-  
 359 der consideration to two independent parameters [42]. The identification of  
 360 EPs in the real-valued parameter space reads

$$\lim_{\substack{\xi_1 \rightarrow \xi_{1,\text{EP}} \\ \xi_2 \rightarrow \xi_{2,\text{EP}}} } \left| \frac{\partial s}{\partial \xi_i} \right|^{-1} = 0 \quad \text{for } i = 1 \text{ or } 2, \quad (17)$$

361 which can be solved using standard multi-parameter root finding algorithms.

362 Despite their peculiar nature, EPs are not rare, and have been observed  
 363 in a large variety of physical systems [48]. In thermoacoustics, they have  
 364 been first discussed only recently [19]. In the latter study, they have been  
 365 identified making use of the dispersion relation for the eigenvalues, which is

366 available only for simple networks. The method outlined in this study is more  
367 general because it does not rely on the explicit knowledge of the dispersion  
368 relation, which is typically not available, for instance, when using Helmholtz  
369 solvers. The method has been tested on several configurations. For all tested  
370 cases, identifying real-valued EPs was possible, as discussed in §4.

371 Lastly, we highlight that there appears to be evidence in the literature  
372 that the effects of EPs in the spectra of thermoacoustic systems, even if not  
373 investigated directly, have already been observed. In [49], eigenvalues hav-  
374 ing infinite sensitivity have been identified analytically, in a Rijke tube-like  
375 system. They have however imprecisely been linked to arbitrary degenerate  
376 states rather than to EPs. Large eigenvalue sensitivities were also observed  
377 in [50], together with the phenomenon of mode veering. Mode veering is  
378 a manifestation of avoided crossing of two eigenvalues. This always occurs  
379 when the thermoacoustic system parameters are close to a degenerate point,  
380 at which the eigenvalue trajectories intersect. The behavior of the eigen-  
381 value trajectories tells whether the degeneracy is defective or not. For a  
382 degenerate eigenvalue with multiplicity 2, if the trajectories approach each  
383 other from nearly opposite direction and then veer by  $90^\circ$  (as is the case  
384 in [50]), then there is an EP close in parameter space, at which the eigen-  
385 value sensitivity is infinite [42]. On the other hand, if the trajectories veer at  
386 different angles, then the veering is due to a degenerate, non-defective point,  
387 and the eigenvalue sensitivity remains finite. The existence of EPs is evident  
388 also in [26, 51], where regions of high sensitivity exhibiting mode veering  
389 have been identified, and their universality in connection to non-dimensional  
390 groups has been demonstrated. Regarding annular combustors, the presence

391 of EPs can be inferred from the eigenvalue trajectories shown in [52, Fig. 8]  
392 and [53, Fig. 6], constructed while varying  $n$  and  $\tau$ . This example will be  
393 further discussed in §4.2.

### 394 3.2. The effect of exceptional points on thermoacoustic eigenvalue trajectories

395 We shall now return to the Rijke tube example of §2.1. Using the nu-  
396 merical method outlined in the previous section, with  $\xi_1 = n$  and  $\xi_2 = \tau$ ,  
397 an EP is identified in this configuration for  $n_{\text{EP}} = 0.075$  and  $\tau_{\text{EP}} = 4.66$  ms,  
398 having a frequency close to that of the acoustic mode of Fig. 1 and a negative  
399 growth rate. Starting from the EP, we vary the flame gain  $n$  in the range  
400  $[0, 0.5]$  and the reflection coefficient  $|R|$  in the range  $[0, 1.3]$ . The resulting  
401 eigenvalue trajectories are shown in Fig. 4.

402 One of the two trajectories obtained while varying  $n$  starts (for  $n = 0$ ) at  
403 the acoustic eigenvalue (on the neutral line); the other comes instead from  
404 a very negative growth rate. On the contrary, the two trajectories obtained  
405 while varying  $R$  start (for  $R = 0$ ) from pure ITA modes, whose values es-  
406 timated from Eq. (5) are reported with red circular markers in Fig. 4. The  
407 trajectories meet at the exceptional value, and turn by  $90^\circ$  across it. Further-  
408 more, because no discontinuities in the parameters are present in the modeled  
409 system, the eigenvalue trajectories must be continuous. In the vicinity of the  
410 exceptional point, this means that the trajectories must strongly veer, to  
411 avoid a crossing. This explains the behavior of the eigenvalue trajectory  
412 shown in Fig. 1: increasing  $n$ , the eigenvalue trajectory is first attracted to-  
413 wards the EP, which has a negative growth rate. However, since crossing is  
414 prohibited, the trajectory must strongly veer, which leads to a sudden change  
415 in the trend of the eigenvalue sensitivity and eventually to the existence of

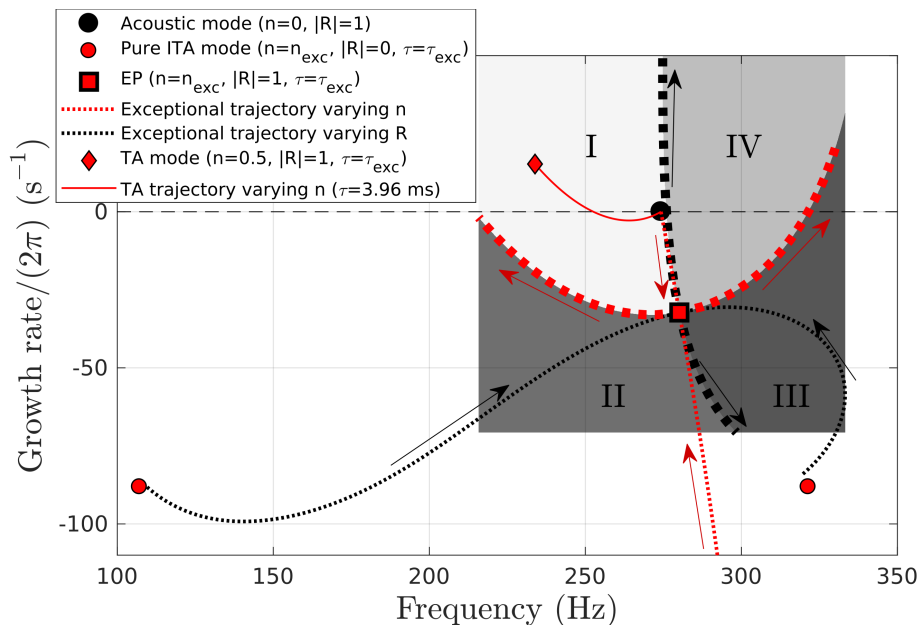


Figure 4: Behaviour of the eigenvalue trajectories obtained while varying the flame interaction index and reflection coefficients across an EP. The arrows indicate the direction that the trajectories follow when the parameters are increased. The acoustic solution is marked with a black circle, and the pure ITA modes for  $\tau = \tau_{\text{EP}}$  with red circles. Because trajectories cannot intersect in the vicinity of an EP, four regions, labelled I to IV, corresponding to different behaviors in the limits  $n \rightarrow 0$  and  $R \rightarrow 0$  are identified. The thermoacoustic mode and the eigenvalue trajectory of Fig. 1 lie in region I.

416 an unstable mode. This is true even though the identified EP has a negative  
 417 growth rate. Rather than the EP per se, it is its interaction with the eigen-  
 418 value trajectories which is relevant: identifying the parameters at which EPs  
 419 are found gives information about when strong changes in the eigenvalues  
 420 sensitivities are expected. As demonstrated in Fig. 4, this sudden change in  
 421 sensitivity can lead to thermoacoustic instabilities.

422 The fact that eigenvalue trajectories cannot cross in the vicinity of an

423 EP leads to a classification of the eigenvalue space in its vicinity. Consider  
 424 the exceptional trajectories for  $n$  highlighted with thick red lines in Fig. 4:  
 425 the acoustic mode is contained in the portion of the plane above this line.  
 426 Because eigenvalue trajectories cannot intersect, thermoacoustic modes that  
 427 start above this line must converge to the acoustic mode when  $n \rightarrow 0$ . On the  
 428 other hand, modes that start below this line cannot converge to an acoustic  
 429 mode when  $n \rightarrow 0$ , and will be pushed towards  $\sigma \rightarrow -\infty$ . Similarly, the  
 430 exceptional trajectories for  $R$  highlighted with thick black lines in Fig. 4  
 431 delimit the region of convergence towards two separate pure ITA modes when  
 432  $R \rightarrow 0$ : on the left, eigenvalues must be attracted towards the ITA mode  
 433 with frequency  $1/(2\tau)$ ; on the right, towards the ITA mode with frequency  
 434  $3/(2\tau)$ . Thus, four regions exist (I-IV in Fig. 4) in which the behavior of the  
 435 eigenvalues in the limits  $n \rightarrow 0$  and  $R \rightarrow 0$  differs. For example in region I  
 436 (top-left), eigenvalues must be attracted towards an acoustic solution when  
 437  $n \rightarrow 0$  and to the ITA solution with frequency  $1/(2\tau)$  when  $R \rightarrow 0$ . Similar  
 438 arguments hold for the remaining regions. The thermoacoustic mode shown  
 439 in Fig. 1 lies in region I in Fig. 4, which is consistent with the seemingly  
 440 ambiguous  $n \rightarrow 0$  and  $R \rightarrow 0$  limits.

441 In summary, the “basins of attraction” of acoustic and pure ITA modes  
 442 are determined by the parameter chosen to describe the thermoacoustic cou-  
 443 pling. Because the exceptional trajectories for varying  $n$  and  $R$  are different,  
 444 the resulting basins differ too. This explains why a classification of the ther-  
 445 moacoustic modes using two separate parameters can be ambiguous. These  
 446 findings on the identification of EPs and the behavior of eigenvalue trajecto-  
 447 ries in their vicinity are general, and will be demonstrated on more complex

448 geometries in the last part of the study.

## 449 4. Numerical examples

450 In this section we demonstrate with two examples (an axial and an  
451 annular configurations) the theoretical findings of this study. Both cases  
452 are solved using the freely available 3D FEM code PyHoltz<sup>2</sup>, dedicated to  
453 (thermo)acoustic eigenvalue problems.

### 454 4.1. BRS combustor

455 As an axial configuration, we focus on the so-called BRS combustor [54].  
456 Thermoacoustic oscillations with a frequency which is not close to any acous-  
457 tic mode have been experimentally observed in this combustor [9], and have  
458 been related to the effect of ITA modes in the literature [13]. The mod-  
459 eled combustor is shown in Fig. 7 (see the aforementioned references for an  
460 exhaustive geometrical description of the combustor).

461 We assume that the inlet/outlet are respectively acoustically closed and  
462 open, and that a sudden temperature jump,  $T_2/T_1 = 5.46$ , occurs across  
463 the flame. Starting from the purely acoustic scenario ( $n = 0$ , fully reflective  
464 boundaries), we identify several acoustic modes having zero growth rate. The  
465 lowest two frequencies correspond to a Helmholtz mode with  $f_H = 54$  Hz and  
466 a quarter-wave like mode, with a frequency of  $f_{1/4} = 589$  Hz, and are shown  
467 in Fig. 5.

468 We then set the interaction index to the small value  $n = 0.001$  and the  
469 time delay (arbitrarily) to  $\tau = 6.88$  ms. The eigenvalues of this thermoacous-

---

<sup>2</sup>Freely available at <https://bitbucket.org/pyholtzdevelopers/public>



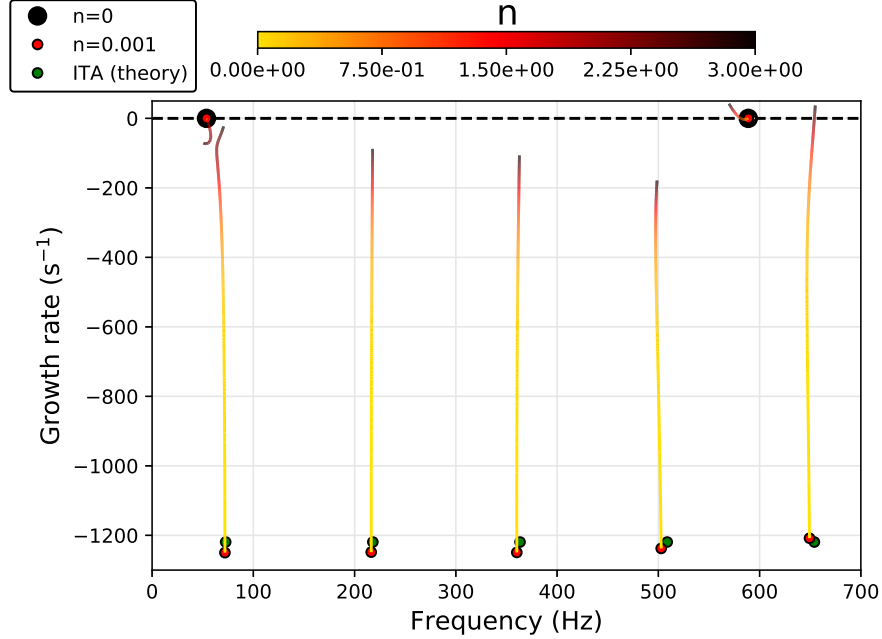


Figure 5: Location of the acoustic (black circles) and thermoacoustic (red circles) eigenvalues when  $n = 0$  and  $n = 0.001$ , respectively. The theoretical guesses for the locations of the thermoacoustic modes of ITA origin are indicated with green markers. The lines track the eigenvalue trajectories for  $n \in [0.001, 3]$ .

470 tic system are shown in Fig. 5 with red markers; because of the weak effect of  
 471 the flame, the eigenvalues of acoustic origin are almost unaffected. However,  
 472 a new set of modes, having ITA origin, is found. These modes have been identi-  
 473 fied using as guesses the expression (A.5), with  $\beta^{-1} = (\theta^2 - 1)/(A_r\theta + 1)$ ,  
 474 obtained from (A.3) when a temperature jump ( $\theta = \sqrt{T_2/T_1}$ ) and an area  
 475 jump ( $A_r = A_2/A_1 = 7.95$ ) are found across the flame, and mean flow is  
 476 neglected. The theoretical guesses are marked in Fig. 5 with green circles,  
 477 and are used to initialize the search of thermoacoustic eigenvalues via New-  
 478 ton's method. The converged thermoacoustic eigenvalues for  $n = 0.001$  are

479 marked with red circles, and agree well with the theoretical predictions. The  
480 configuration at hand has an area jump upstream of the flame, which affects  
481 the definition of pure ITA modes in the anechoic limit [7]. However, this has  
482 no effect on the definition of the ITA modes originating in the limit  $n \rightarrow 0$ .

483 Using these initial guesses, we can then track the evolution of *all* ther-  
484 moacoustic eigenvalues in the region of interest while increasing the value  
485 of the interaction index to any desired value. These trajectories are shown  
486 with lines in Fig. 5: the growth rates of thermoacoustic modes of ITA origin  
487 are far more sensitive to changes in  $n$  than the growth rate of the modes of  
488 acoustic origin. For large values of  $n$ , the modes of ITA origin can become  
489 unstable and feature the largest growth rates. Also, mode veering between  
490 an eigenvalue of acoustic origin and one of ITA origin is visible at a frequency  
491 of about 60 Hz.

492 This mode veering is relevant for the experimental observations of [9],  
493 where oscillations with a frequency of 100 Hz were observed, and have been  
494 associated to an ITA mode instability [4]. Therefore, we shall focus the  
495 attention around the low-frequency Helmholtz mode only. To understand the  
496 influence of the flame response on the spectrum, we vary  $n \in [0, 3]$  and  $\tau \in$   
497  $[0, 0.016]$ . The maximum  $\tau$  is chosen to be  $< 1/f_H$ , to avoid that eigenvalue  
498 trajectories intersect in the vicinity of the acoustic solution. The resulting  
499 stability map is shown in Fig. 6; as commonly observed, depending on the  
500 particular choice of both  $n$  and  $\tau$ , the resulting thermoacoustic mode can  
501 be stable or unstable. In the vicinity of the acoustic solution the eigenvalue  
502 sensitivity with respect to changes in the interaction index is nonlinear: a  
503 small increase in  $n$  from zero first stabilizes the pole, but the trajectory

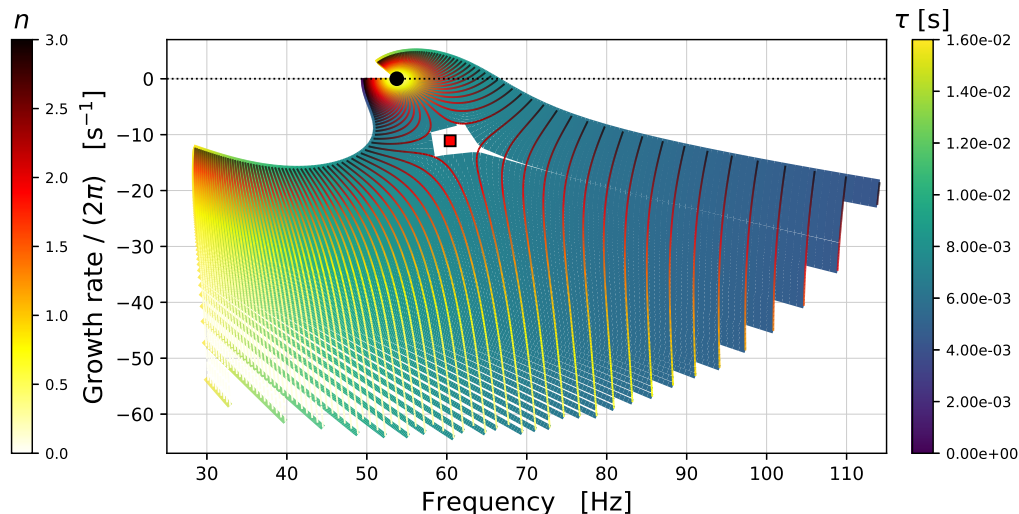


Figure 6: Trajectories of the lowest frequency eigenvalue of the BRS combustor when varying the parameters of the flame model. The trajectories with constant  $n$  and constant  $\tau$  are highlighted with different colormaps. The acoustic solution ( $f = 53.75$  Hz) is highlighted with a black circle. In the region which is avoided by the eigenvalues, there exists an EP ( $n_{\text{EP}} = 2.181$ ,  $\tau_{\text{EP}} = 6.96$  ms), indicated in red. The local behavior of the trajectories in this region is shown in Fig. 8a.

504 strongly veers and the mode can become unstable for larger values of  $n$ , as  
 505 was observed in Fig. 1. Because of their veering, which can be observed for  
 506 both the  $n$ - and  $\tau$ -isolines, the eigenvalue trajectories avoid a region in the  
 507 complex-frequency space, as discussed in [26]. The presence of this avoided  
 508 region is one of the characteristics of EPs. Its existence can be confirmed  
 509 using the procedure outlined in §3.1: starting from an educated guess based  
 510 on Fig. 6, a root of Eq. (17) is found while varying  $n$  and  $\tau$ . This root  
 511 identifies an EP (see Fig. 8). Its modeshape is reported in Fig. 7, together  
 512 with that of the purely acoustic Helmholtz mode (plenum dominant) and of  
 513 the mode of ITA origin in the  $n \rightarrow 0$  limit (flame region dominant). These

514 are the two modes that coalesce to create the EP, whose modeshape has  
 515 clearly inherited features from both of them.

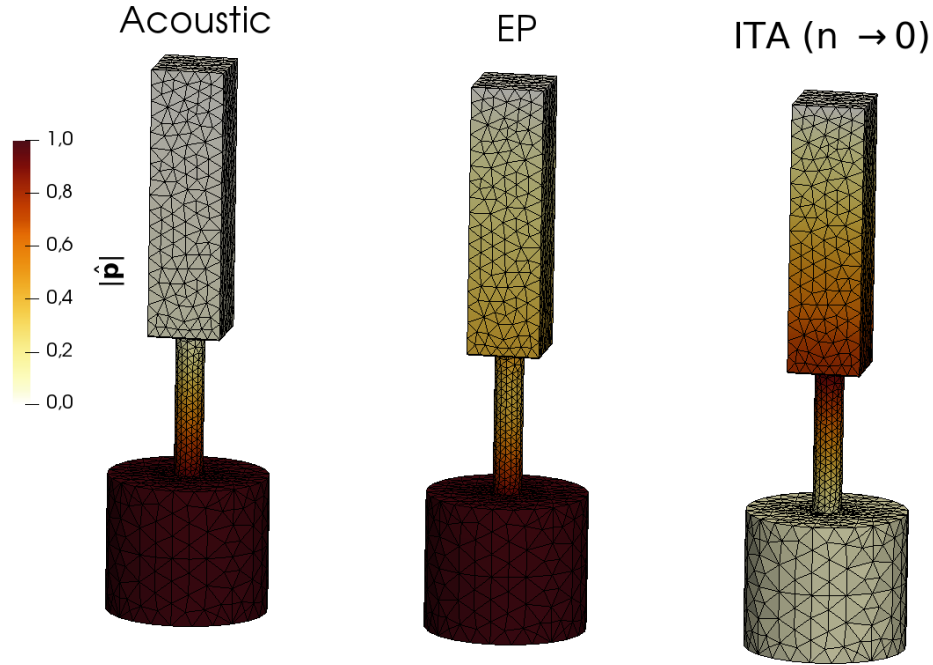


Figure 7: Geometry of the BRS combustor and pressure modeshapes of the low-frequency acoustic (left), ITA origin (right) and exceptional (middle) modes.

516 The trajectories of the eigenvalues around the EP are shown in Fig. 8. At  
 517 the EP, two eigenvalues, one of acoustic and one of intrinsic origin, coalesce.  
 518 Due to the high sensitivity of the eigenvalues in the vicinity of the EP, a small  
 519 parameter range is considered. When one parameter ( $n$  or  $\tau$ ) is fixed at the  
 520 exceptional value and the other is varied, the trajectories collide at the EP,  
 521 and branch off at angles of  $90^\circ$  (Fig. 8a). The further the parameter values  
 522 are from those of the EP, the less intense is the veering. Figures 8b and 8c  
 523 show the frequency and growth rate surfaces as functions of  $n$  and  $\tau$  close  
 524 to the EP. Because real and imaginary part of the eigenvalue surfaces are

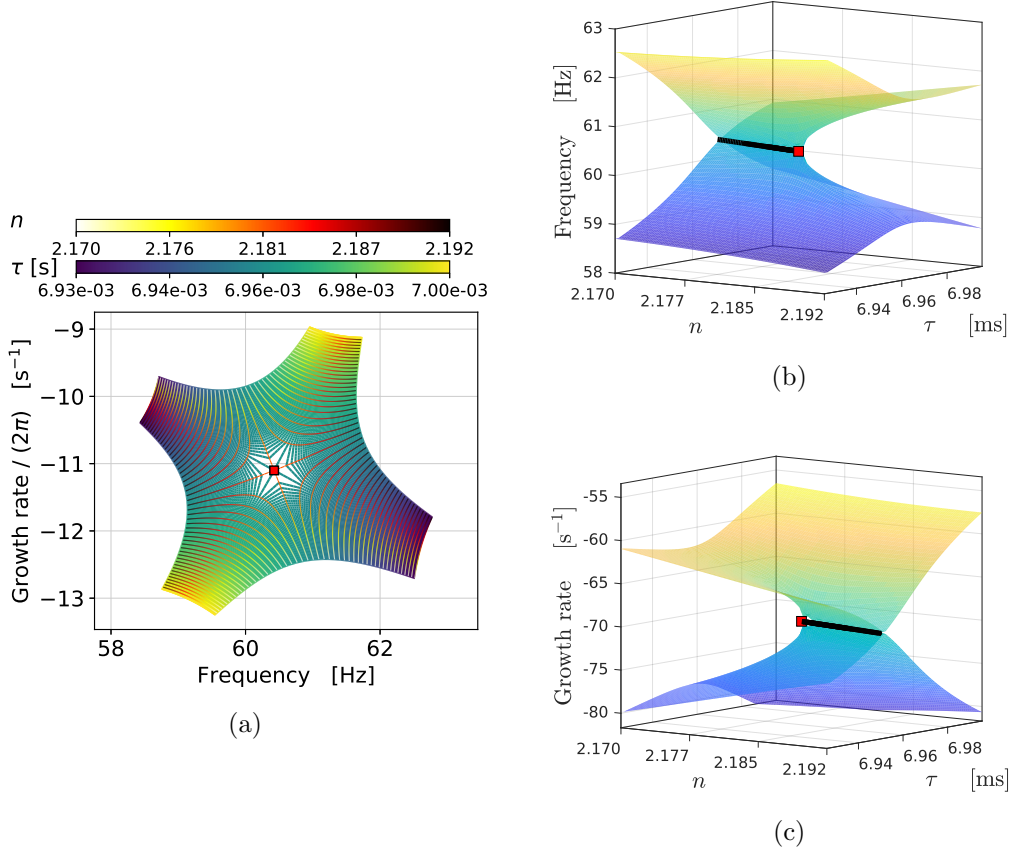


Figure 8: (a): Trajectories of the eigenvalues in the vicinity of the EP. At the EP the trajectories cross forming right angles. (b)-(c): Frequency and growth rate surfaces as functions of  $n$  and  $\tau$  close to the EP. The Riemann cuts of the surfaces are highlighted with black lines, the EP in red.

525 plotted separately, each surface self-intersects, forming Riemann cuts [18].  
 526 The number of eigenvalues found for an arbitrary pair of values ( $n, \tau$ )  
 527 to the EP is equal to the number of intersections of vertical lines passing  
 528 through ( $n, \tau$ ) with the complex-valued surface. This is always equal to 2  
 529 in Figs. 8b and 8c, except at the EP at which it is equal to 1, indicating

530 eigenvalue crossing.

531 Our analysis of the BRS system is consistent with the theoretical discus-  
532 sion that thermoacoustic modes can uniquely be classified as of acoustic or  
533 ITA origin in the limit  $n \rightarrow 0$ , regardless of the boundary conditions and  
534 presence of area variations upstream of the flame. An EP exists in the spec-  
535 tra of this combustor for specific values of  $n$  and  $\tau$ , at which two eigenvalues  
536 (one of acoustic and one of intrinsic origin) coalesce. Even if the system is not  
537 operated at EP conditions, being sufficiently close to it in parameter space  
538 results in strong mode veering. This can explain why the unstable frequency  
539 observed experimentally in [9] significantly differs from all the acoustic eigen-  
540 frequencies of the system, although the unstable mode may still be of acoustic  
541 origin.

#### 542 *4.2. Annular configuration*

543 As for an annular configuration, we consider the Helmholtz model of  
544 a generic geometry formed by plenum and chamber volumes connected by  
545 a given number of ducts (see Fig. 10). These configurations can also be  
546 analyzed using network models, as discussed in [35, 55–57]. We investigate  
547 the  $N_f = 4$  burners setup presented in [52] because (i) the configuration has  
548 closely spaced acoustic eigenvalues, and (ii) different regimes (uncoupled,  
549 weakly coupled, and strongly coupled) have been identified in [52], which are  
550 revisited here. This will show that the occurrence of these different regimes  
551 can be explained with the existence of EPs.

552 The geometrical and thermodynamical parameters are taken from [56],  
553 with the difference that a smaller temperature jump has been considered,  
554  $T_2/T_1 = 1.5$ , in order to have closely spaced acoustic eigenvalues when  $n = 0$ ,

555 as in [52]. The two acoustic eigenvalues with the lowest frequencies are re-  
 556 ported with black circles in Fig. 9. The two modes are plenum- and chamber-  
 557 dominant, respectively, with azimuthal order  $m = 1$ , therefore degenerate.  
 558 Their modeshapes<sup>3</sup> are reported in Fig. 10, and the frequencies are close to  
 559 the frequencies of the plenum/chamber,  $f_i = c_i/(\pi D_i)$ , where  $D_i$  are the  
 560 diameters of the volumes. The deviation from these values is due to the  
 561 cross-talking of the plenum-chamber volumes via the connecting ducts.

562 To investigate the effect of the flame response on the eigenvalues, we vary  
 563  $n \in [0, 2]$  and  $\tau \in [0, 0.015]$  s. The maximum time delay value is chosen  
 564 to be  $1/f_{\text{chamber}}$  so that eigenvalue trajectories looping around the acoustic  
 565 eigenvalues will not cross each other [26, 58]. We track the eigenvalues using  
 566 continuation methods and show the eigenvalue trajectories in Fig. 9. In the  
 567 range of parameters studied, we identify three topologically different groups  
 568 of eigenvalue trajectories:

- 569 I. when  $n < 1.17$ , varying  $\tau$  results in looping the thermoacoustic eigenval-  
 570 ues that originate from the plenum and chamber acoustic modes around  
 571 these solutions. This corresponds to the “weakly coupled” regime dis-  
 572 cussed in [56], in which it is appropriate to associate each thermoacous-  
 573 tic mode to an acoustic mode.
- 574 II. when  $1.17 < n < 1.44$ , the trajectories do not follow closed loops any-  
 575 more. The thermoacoustic eigenvalue that starts close to the plenum  
 576 acoustic mode is shifted towards a value close to that of the chamber

---

<sup>3</sup>Because these solutions are degenerate with geometric multiplicity 2, they each have  
 2 linearly independent modeshapes. Only one of them is shown in Fig. 10.

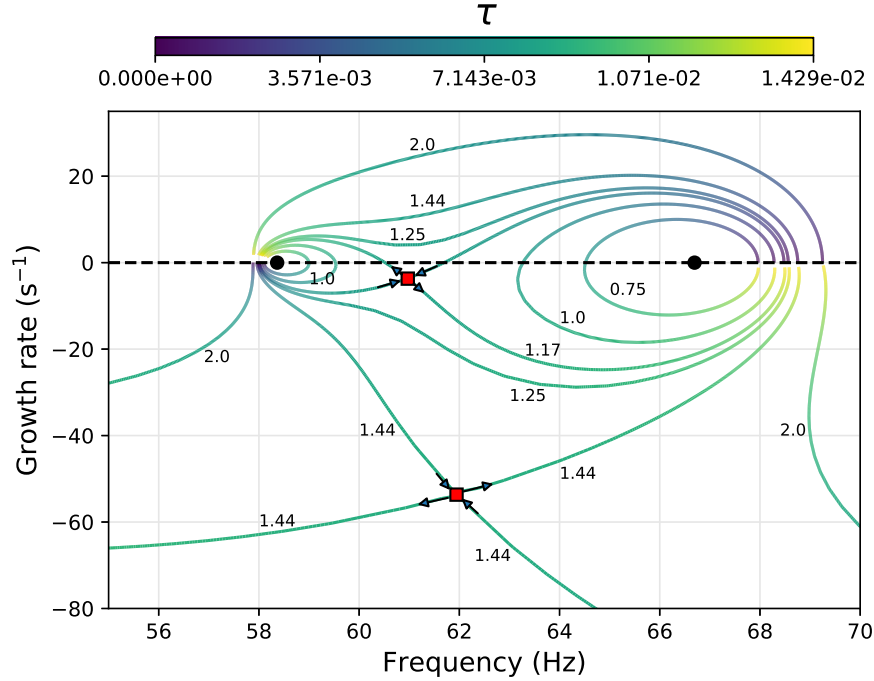


Figure 9: Eigenvalue trajectories of the annular configuration for constant  $n$  (isoline values) when  $\tau$  is varied. The system has two closely spaced purely acoustic eigenvalues (black dots). Two exceptional points (red squares) are identified: one due to the interaction between two modes of acoustic origin, close to the real axis, and one due to the interaction between acoustic and intrinsic modes. The values of  $n$  at which EPs are located determine topological changes in the eigenvalue trajectories.

577 acoustic mode, and vice versa. In other words, the nature of these ther-  
 578 moacoustic modes strongly varies depending on the value of  $\tau$  consid-  
 579 ered. For intermediate values of  $\tau$ , the modes have frequencies which lie  
 580 between those of the two acoustic modes, and their modeshapes are not  
 581 anymore dominant in only the plenum or chamber, but instead in both  
 582 cavities (see Fig. 10). For this reason, in this regime, which corresponds  
 583 to the “strongly coupled” regime of [56], it would be inappropriate to



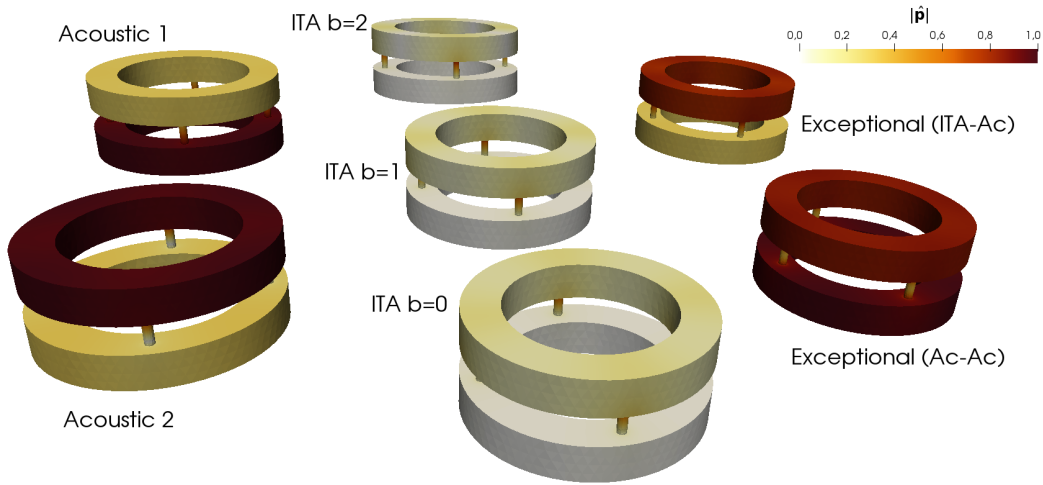


Figure 10: Absolute value of the pressure of various modes found in the annular combustor configuration. Left column: plenum and chamber dominant acoustic modes (see Fig. 9). Middle column: cluster of ITA modes found in the limit  $n \rightarrow 0$  (see Fig. 11); the modeshapes are identical and dominant in the flame region, as expected for ITA modes. Right column: modeshapes of the two identified exceptional points (see Fig. 9). One is due to the interaction between two modes of acoustic origin, the other between a mode of acoustic origin and one of ITA origin.

584 perform a single-mode (degenerate) Galerkin expansion. A two-mode  
 585 Galerkin expansion that accounts for both the plenum- and chamber  
 586 dominant modes should yield a good approximation, as hinted by the  
 587 modeshapes shown in both Figs. 7 and 10.

588 III. Another topological change in the eigenvalue trajectories is observed  
 589 when  $n > 1.44$ . The eigenvalues that start close to the chamber acoustic  
 590 mode are shifted towards plenum-dominant solutions when  $\tau$  increases,  
 591 as in the previous regime. The same is, however, not anymore true  
 592 for the other eigenvalue. The mode that starts close to the plenum-  
 593 dominant solution is pushed away from any known acoustic solution,

594 and a trajectory that starts from a solution that is not related to any  
595 known acoustic mode (on the right of Fig. 9) ends with a frequency close  
596 to the chamber-dominant acoustic mode. In this regime, which has not  
597 been discussed before, also a two-mode Galerkin expansion based on  
598 the acoustic modes cannot yield a good approximation of the original  
599 system, because the thermoacoustic modes can be significantly different  
600 from the acoustic ones.

601 From the topological structure of the various eigenvalue trajectories, it is  
602 possible to infer that EPs must exist for specific pairs of  $n$  and  $\tau$ . These EPs  
603 in fact discriminate between the three regimes just discussed. Starting from  
604 educated guesses based on shape of the eigenvalue trajectories, and using the  
605 numerical procedure outlined in §3.1, we identify two EPs in the parameter  
606 region investigated. One is found for  $n_{\text{EP},aa} = 1.17$  and  $\tau_{\text{EP},aa} = 8.19$  ms,  
607 and discriminates between the aforementioned regimes I and II. The other  
608 is found for  $n_{\text{EP},ai} = 1.44$  and  $\tau_{\text{EP},ai} = 8.91$  ms, and discriminates between  
609 regimes II and III.

610 The first EP is labelled with the subscript <sub>aa</sub> because it results from  
611 the collision of two modes of acoustic origin. In [19] only EPs that arise  
612 from the interaction of a mode of acoustic origin and one of ITA origin  
613 were discussed. EPs arising from two acoustic eigenvalues are already known  
614 from the discussions of EPs in acoustic systems, and have a relevance, e.g.,  
615 in optimizing the performance of Helmholtz dampers [18, 59]. However, in  
616 the current example the coalescence of two eigenvalues of acoustic origin  
617 at the EP is driven by the flame response parameters. Such EPs were not  
618 discussed in the literature before. The acoustic–acoustic nature of this EP

619 is also visible in its modeshape, shown in Fig. 10: in contrast to the acoustic  
 620 modes – purely plenum- or chamber dominant – this modeshape has the  
 621 same magnitude in both cavities, suggesting that a single-mode Galerkin  
 622 expansion (that preserves only the plenum’s or chamber’s structure) would  
 623 not be a suitable approximation in its vicinity.

624 The other EP is labelled with the subscript  $_{ai}$  because it originates from  
 625 the interaction between modes of acoustic and ITA origin. Its behavior is  
 626 analogous to that discussed in §4.1 and [19], as is more evident from Fig. 11:  
 627 starting from  $n_{EP,ai}$  and decreasing the value of  $n$  towards zero, two eigen-  
 628 values stem from this EP: one converges to an acoustic solution (that of the  
 629 chamber), the other tends towards an ITA mode. Also, because of the dis-  
 630 crete rotational symmetry of the annular configuration under investigation,  
 631 each eigenvalue shown in Fig. 9 is degenerate with algebraic and geometric  
 632 multiplicity 2. As both EPs identified for this configuration are found when  
 633 2 eigenvalues (each having algebraic multiplicity 2) coalesce, the EPs have  
 634 algebraic multiplicity 4, but are defective in that only two linearly indepen-  
 635 dent modeshapes exist, i.e. they have geometric multiplicity 2. Only one  
 636 of these modeshapes is shown in Fig. 10, the other is phase-inverted, as for  
 637 degenerate (thermo)acoustic modes.

638 Lastly, we verify that, for annular configurations, in the limit  $n \rightarrow 0$  the  
 639  $N_f$  modes of ITA origin are almost decoupled, as discussed in §2.2.1. Thus,  
 640 a cluster of  $N_f = 4$  eigenvalues of ITA origin is expected to be found in  
 641 the vicinity of the value predicted by Eq. (A.5). Starting from this theoret-  
 642 ical guess, a Newton method is employed to identify the close eigenvalues.  
 643 Given the fact that the modes cluster, however, it is difficult to identify all

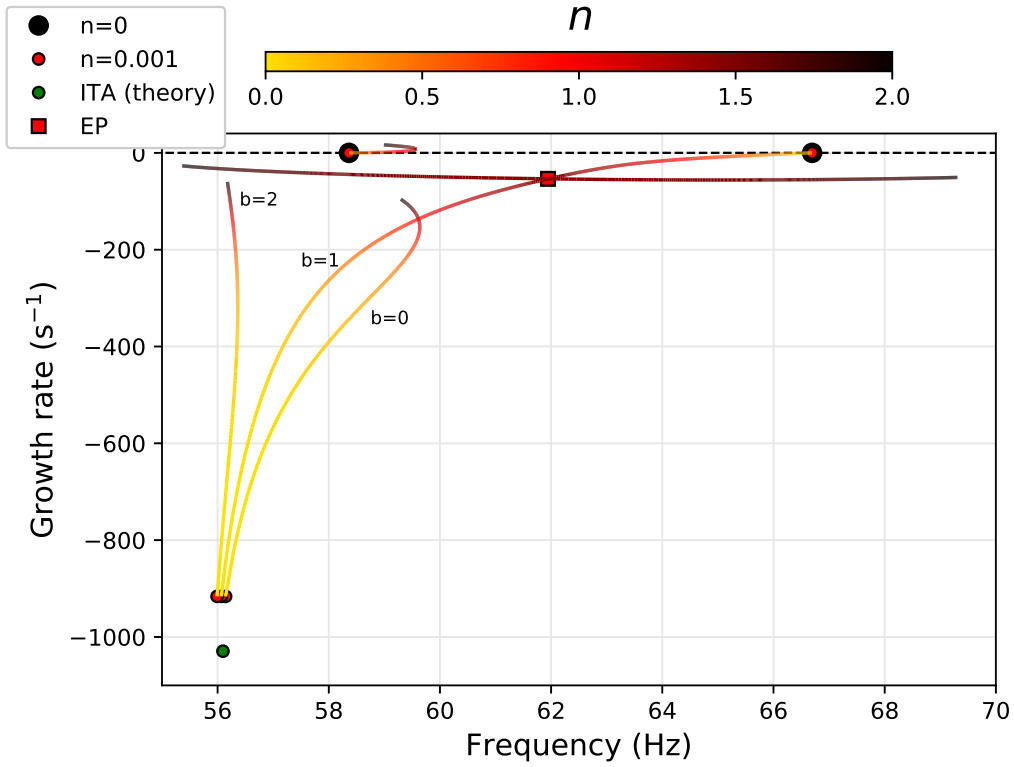


Figure 11: Modes of ITA origin found in the annular setup when  $n \rightarrow 0$ . The cluster of ITA modes estimated using equation (A.5) for  $n = 0.001$  is shown in green. Four thermoacoustic eigenvalues (one for each burner, and two are degenerate due to the symmetries of the annular configuration) are found in its vicinity. The lines track all the eigenvalues for  $n \in [0.001, 2]$ : two modes of ITA and acoustic origin coalesce at  $n_{\text{EP},ai}$ .

644 the modes, as the iterative algorithm tends to converge to the same solu-  
 645 tion. Bloch-wave theory comes to aid [37]. By using this formalism, the  
 646 clustered modes are naturally split across the various Bloch-wave numbers  
 647  $b = 0, 1, \dots, N_f - 1$ . In the  $n \rightarrow 0$  limit, a mode of ITA origin is found for  
 648 each Bloch-wave number for the configuration at hand when  $n = 0.001$ , as  
 649 shown in Fig. 11. Because of the mirror symmetry of the system, the modes

650 found for  $b = 1$  and  $b = 3$  are degenerate for any value of  $n$ . In general,  
 651 for a system with an even number  $N_f$  of burners, the cluster of eigenvalues  
 652 found in the  $n \rightarrow 0$  limit will be formed by  $N_f/2 + 1$  distinct eigenvalues, of  
 653 which  $N_f/2 - 1$  are degenerate with multiplicity 2. We can then track these  
 654 solutions of ITA origin by increasing  $n$  towards any desired finite value. The  
 655 eigenvalue trajectories are shown in Fig. 11. Because we have fixed the time  
 656 delay to  $\tau_{\text{EP},ai}$ , we identify again the EP of acoustic–intrinsic nature. No-  
 657 tably, only the mode of ITA origin with  $b = 1$  (or equivalently 3, given the  
 658 degeneracy) interacts with the mode of acoustic origin and generates an EP.  
 659 This is because modes associated with different Bloch-wave numbers are or-  
 660 thogonal [30]. Because both modes of acoustic origin are azimuthal modes  
 661 of order  $m = 1$ , thus associated with Bloch-wave numbers  $b = 1$ , only the  
 662 modes of ITA origin associated with the latter Bloch-wave numbers can in-  
 663 teract with them and lead to the formation of EPs, or more generally to  
 664 mode veering. This also explains why the ITA modes found in the clustered  
 665 region do not interact with each other and neither exhibit veering nor form  
 666 EPs despite their eigenvalues being so closed.

## 667 5. Conclusions

668 In this study, a different perspective to the notion of intrinsic modes  
 669 has been presented, with the aim of associating each thermoacoustic mode  
 670 with a unique origin without altering the acoustic state (in particular the  
 671 reflection coefficients) of the system. We have demonstrated that this is  
 672 not possible with the traditional definitions of acoustic and intrinsic modes,  
 673 because these definitions are based on two separate parameters: the former

674 are defined in the absence of heat release dynamics, when  $n = 0$ , whereas  
 675 the latter are defined in anechoic conditions, when  $R = 0$ . Instead, we  
 676 propose to use only one parameter, chosen to be  $n$ , to define the origin  
 677 of all the modes independent of the acoustic boundary conditions of the  
 678 system. We have shown in a rather general way that any thermoacoustic  
 679 mode can be uniquely associated with one of these two classes of modes when  
 680  $n \rightarrow 0$ . Furthermore, an explicit expression has been found for the modes of  
 681 ITA origin when an  $n - \tau$  model is adopted. These are independent of the  
 682 acoustic properties even for systems which, despite having anechoic boundary  
 683 conditions, still have an acoustic response due to, e.g., area expansions. Their  
 684 expressions are functions only of the coefficients of the scattering matrix  
 685 **S** and the heat release scaling coefficients. In some special cases (absence  
 686 of mean flow and cross-sectional area variations across the flame, anechoic  
 687 boundary conditions), the definition of ITA modes proposed in this study  
 688 coincides with that found in the literature for anechoic conditions. We have  
 689 also discussed how, in the case of rotationally symmetric annular combustors,  
 690 modes of ITA origin tend to form clusters of eigenvalues in the limit  $n \rightarrow 0$ ,  
 691 and generally behave as (weakly) coupled oscillators for finite values of  $n$ .

692 The presented theory enables us to theoretically estimate the location  
 693 of *all* thermoacoustic eigenvalues in the limit  $n \rightarrow 0$ , in a given range of  
 694 eigenfrequencies. The estimate is based on the numerical identification of  
 695 eigenvalues of acoustic origin, found in the vicinity of acoustic modes and  
 696 easily obtainable with standard Helmholtz solvers, and eigenvalues of ITA  
 697 origin, in the vicinity of theoretically estimated values having large negative  
 698 growth rates. Having at hand the solutions in the limit  $n \rightarrow 0$ , continuation

699 methods can be used to track the trajectories of these eigenvalues to any  
700 desired value of  $n$ . This reduces the numerical effort needed for identifying  
701 large sets of thermoacoustic eigenvalues, and increases the confidence that  
702 all modes in a given frequency range have been identified. For finite val-  
703 ues of  $n$ , the modes of acoustic and ITA origin may interact, giving rise to  
704 strong veering of the eigenvalue trajectories. This effect has been related to  
705 the existence of exceptional points (EPs) in the spectra of thermoacoustic  
706 systems, at which eigenvalues and their corresponding eigenfunctions coa-  
707 lesce. Even though the identified EPs always have negative growth rates,  
708 we have demonstrated how mode veering in their vicinity is responsible for  
709 strong changes in the eigenvalues sensitivities: in some cases, this can cause  
710 eigenvalues that are predicted to stabilize by linear stability analysis (for  
711 weak flames) to become unstable. In this respect, EPs can be considered as  
712 one of the causes of thermoacoustic instabilities, and their identification is  
713 practically relevant. A numerical method for the identification of real-valued  
714 EPs has been presented, which uses the self-orthogonality property of the  
715 defective eigenvalues found at EPs.

716 All the theoretical results presented have also been demonstrated numeri-  
717 cally on 3D axial and annular thermoacoustic configurations. The theoretical  
718 predictions on the locations of the modes of ITA origin agree well with numer-  
719 ical results in all tested cases. Clustering of modes is predicted and observed  
720 in annular configurations. EPs have been identified in all configurations and  
721 can result from the interaction of (i) modes of acoustic and of intrinsic origin,  
722 or (ii) modes of only acoustic origin. No EPs resulting from the interaction  
723 between two intrinsic modes have been identified so far. The modeshapes of

724 the EPs contain a strong signature of which modes are responsible for their  
725 formation. We have linked the existence of EPs to the topological behavior of  
726 the eigenvalue trajectories in parameter space, and related those to regimes  
727 that have been previously indicated as “weakly” or “strongly” coupled, as  
728 well as identified a new regime which is triggered at moderately high values  
729 of the interaction index  $n$ . Generally, knowledge on the EPs’ locations leads  
730 to the identification of several regimes within which the topological behavior  
731 of eigenvalue trajectories is preserved, and to a good qualitative prediction  
732 and understanding of the eigenvalue trajectories of thermoacoustic systems  
733 in parameter space.

### 734 **Acknowledgments**

735 A. Orchini is grateful to the Alexander von Humboldt Foundation for  
736 financial support via their PostDoctoral Research Fellowship.

### 737 **Appendix A. Poles of the closed thermoacoustic feedback loop** 738 **when $|ne^{-s\tau}| \rightarrow \mathcal{O}(1)$ for $n \rightarrow 0$**

739 Given the expression for the growth rate (12), all the propagation terms  
740  $P_j = e^{-s\tau_j}$  diverge to infinity in the limit  $n \rightarrow 0$ . To find the poles of (11)  
741 when the growth rate  $\sigma$  becomes infinitely negative, it is convenient to rewrite  
742 it as

$$\hat{u} = \frac{1}{\frac{D(s)}{N(s)} - \mathcal{F}(s)} \hat{q}_n. \quad (\text{A.1})$$

743 The fraction at the denominator of the above equation is an indeterminate  
744 form in the  $\sigma \rightarrow -\infty$  limit. To solve it, we divide numerator and denominator



745 of Eq. (7) by  $P_1P_2P_3P_4$ , obtaining

$$\frac{D(s)}{N(s)} = \frac{P_2^{-1}P_3^{-1}R_1S_{21}+P_1^{-1}P_4^{-1}R_2S_{12}+R_1R_2(S_{22}S_{11}-S_{21}S_{12})-(P_1P_2P_3P_4)^{-1}}{(P_1^{-1}P_4^{-1}-R_1)[P_2^{-1}P_3^{-1}H_2+R_2(H_1S_{22}-H_2S_{12})]}. \quad (\text{A.2})$$

746 Considering now the limit  $\sigma \rightarrow -\infty$ , all the terms containing  $P_j^{-1}$  vanish  
747 because of the growth rate expression (12). Thus, Eq. (A.2) reduces to

$$\lim_{\sigma \rightarrow 0} \frac{D(s)}{N(s)} = -\frac{R_1R_2(S_{22}S_{11} - S_{21}S_{12})}{R_1R_2(H_1S_{22} - H_2S_{12})} \equiv -\beta, \quad (\text{A.3})$$

748 where we have defined the factor  $\beta$  as a function of the scattering matrix  
749 elements  $S_{ij}$  and the scaling factors between flame and acoustic responses  
750  $H_i$ . Note that the reflection coefficients, which generally include also possible  
751 area variations in the regions upstream/downstream of the flame, simplify in  
752 the above expressions.

753 The poles of (A.1) in the limit of infinitely negative growth rate are  
754 therefore given by

$$\frac{1}{\beta}\mathcal{F}(s) + 1 = 0. \quad (\text{A.4})$$

755 The latter equation can always be solved numerically, for arbitrary expres-  
756 sions of the FTF. In the special case in which the flame response can be  
757 modelled with an  $n - \tau$  model,  $\mathcal{F}(s) = ne^{-s\tau}$ , analytical solutions can be  
758 found:

$$s = \frac{1}{\tau} \log\left(\frac{n}{\beta}\right) + \frac{(2k+1)\pi}{\tau}i, \quad k \in \mathbb{Z}. \quad (\text{A.5})$$

759 The angular frequencies of these solutions are identical to those of the in-  
760 trinsic ones, as per Eq. (5). The expression for the growth rate is consistent  
761 with that obtained in the asymptotic limit (see Eq. (12)), with  $\alpha = \beta$ , which  
762 makes the solutions valid. In the special case in which no mean flow effects

763 are considered, it can be proven that  $\beta = \frac{1}{H_2}$ , so that the dispersion rela-  
764 tion (A.4) is formally equivalent to that of the pure ITA modes (10), and so  
765 are the eigensolutions. This was implicitly shown in [14], but can be derived  
766 from first principles given the explicit expressions of the scattering matrix  
767 elements.

768 In summary, we have proven that there always exists a set of modes  
769 in the  $n \rightarrow 0$  limit which are infinitely damped, and whose frequencies are  
770 identical to those of the pure ITA modes, which are found when the reflection  
771 coefficients are set to zero.

## 772 References

- 773 [1] M. Hoeijmakers, V. Kornilov, I. Lopez Arteaga, P. de Goey, H. Nijmeijer,  
774 Intrinsic instability of flame-acoustic coupling, *Combustion and Flame*  
775 161 (2014) 2860–2867.
- 776 [2] M. Hoeijmakers, V. Kornilov, I. Lopez Arteaga, P. de Goey, H. Nijmei-  
777 jer, Flame dominated thermoacoustic instabilities in a system with high  
778 acoustic losses, *Combustion and Flame* 169 (2016) 209–215.
- 779 [3] T. Emmert, S. Bomberg, W. Polifke, Intrinsic thermoacoustic instability  
780 of premixed flames, *Combustion and Flame* 162 (2015) 75–85.
- 781 [4] S. Bomberg, T. Emmert, W. Polifke, Thermal versus acoustic response  
782 of velocity sensitive premixed flames, *Proceedings of the Combustion*  
783 *Institute* 35 (2015) 3185–3192.
- 784 [5] Y. Aurégan, R. Starobinski, Determination of acoustical energy dissipa-

- 785 tion/production potentiality from the acoustical transfer functions of a  
786 multiport, *Acta Acustica* 85 (1999) 788–792.
- 787 [6] A. Gentemann, W. Polifke, Scattering and generation of acoustic energy  
788 by a premix swirl burner, in: *ASME Turbo Expo*, 2007, pp. 125–133.
- 789 [7] C. F. Silva, T. Emmert, S. Jaensch, W. Polifke, Numerical study on  
790 intrinsic thermoacoustic instability of a laminar premixed flame, *Combustion and Flame* journal 162 (2015) 3370–3378.
- 792 [8] E. Courtine, L. Selle, T. Poinso, DNS of intrinsic thermoacoustic modes  
793 in laminar premixed flames, *Combustion and Flame* 162 (2015) 4331–  
794 4341.
- 795 [9] L. Tay-Wo-Chong, S. Bomberg, A. Ulhaq, T. Komarek, W. Polifke,  
796 Comparative validation study on identification of premixed Flame  
797 Transfer Function, *Journal of Engineering for Gas Turbines and Power*  
798 134 (2012) 021502 (8 pages).
- 799 [10] L. Crocco, Theoretical studies on liquid-propellant rocket instability,  
800 in: *The Combustion Institute: Tenth Symposium (International) on*  
801 *Combustion*, 1965, pp. 1101–1128.
- 802 [11] T. Poinso, Simulation methodologies and open questions for acoustic  
803 combustion instability studies, in: *Center for Turbulence Research Annual*  
804 *Research Briefs*, 2013, pp. 179–188.
- 805 [12] C. F. Silva, M. Merk, T. Komarek, W. Polifke, The contribution of  
806 intrinsic thermoacoustic feedback to combustion noise and resonances

- 807 of a confined turbulent premixed flame, *Combustion and Flame* 182  
808 (2017) 269–278.
- 809 [13] T. Emmert, S. Bomberg, S. Jaensch, W. Polifke, Acoustic and intrinsic  
810 thermoacoustic modes of a premixed combustor, in: *Proceedings of the*  
811 *Combustion Institute*, Vol. 36, 2017, pp. 3835–3842.
- 812 [14] N. K. Mukherjee, V. Shrira, Intrinsic flame instabilities in combustors:  
813 Analytic description of a 1-D resonator model, *Combustion and Flame*  
814 185 (2017) 188–209.
- 815 [15] N. Hosseini, V. N. Kornilov, I. Lopez Arteaga, W. Polifke, O. J. Teerling,  
816 L. P. H. de Goey, Intrinsic thermoacoustic modes and their interplay  
817 with acoustic modes in a Rijke burner, *International Journal of Spray*  
818 *and Combustion Dynamics* 10 (2018) 315–325.
- 819 [16] W. D. Heiss, The physics of exceptional points, *Journal of Physics A:*  
820 *Mathematical and Theoretical* 45 (2012) 444016 (11pp).
- 821 [17] M. Miri, A. Alù, Exceptional points in optics and photonics, *Science* 363  
822 (2019) eaar7709 (11 pages).
- 823 [18] C. Bourquard, N. Noiray, Stabilization of acoustic modes using  
824 Helmholtz and Quarter-Wave resonators tuned at exceptional points,  
825 *Journal of Sound and Vibration* 445 (2019) 288–307.
- 826 [19] G. A. Mensah, L. Magri, C. F. Silva, P. E. Buschmann, J. P. Moeck,  
827 Exceptional points in the thermoacoustic spectrum, *Journal of Sound*  
828 *and Vibration* 433 (2018) 124–128.

- 829 [20] A. P. Dowling, S. R. Stow, Acoustic analysis of gas turbine combustors,  
830 Journal of Propulsion and Power 19 (2003) 751–764.
- 831 [21] F. E. C. Culick, Unsteady motions in combustion chambers for propul-  
832 sion systems, AGARDograph, RTO AG-AVT-039, 2006.
- 833 [22] T. Poinsot, D. Veynante, Theoretical and numerical combustion, 2nd  
834 Edition, R T Edwards, 2005.
- 835 [23] F. Nicoud, L. Benoit, C. Sensiau, T. Poinsot, Acoustic modes in com-  
836 bustors with complex impedances and multidimensional active flames,  
837 AIAA Journal 45 (2007) 426–441.
- 838 [24] F. E. C. Culick, Nonlinear behavior of acoustic waves in combustion  
839 chambers - I, Acta Astronautica 3 (1976) 715–734.
- 840 [25] B. E. Courtine, L. Selle, F. Nicoud, W. Polifke, C. Silva, M. Bauerheim,  
841 Causality and intrinsic thermoacoustic instability modes, in: Center for  
842 Turbulence Research, 2014, pp. 169–178.
- 843 [26] C. F. Silva, K. J. Yong, L. Magri, Thermoacoustic modes of quasi- one-  
844 dimensional combustors in the region of marginal stability, Journal of  
845 Engineering for Gas Turbines and Power 141 (2019) 021022 (8 pages).
- 846 [27] A. P. Dowling, Nonlinear self-excited oscillations of a ducted flame, Jour-  
847 nal of Fluid Mechanics 346 (1997) 271–290.
- 848 [28] W. Polifke, Combustion instabilities, in: J. Anthoine, A. Hirschberg  
849 (Eds.), Advances in Aeroacoustics and Applications, Von Karman Insti-  
850 tute, 2004, pp. 1–44.

- 851 [29] A. Orchini, S. J. Illingworth, M. P. Juniper, Frequency domain and  
852 time domain analysis of thermoacoustic oscillations with wave-based  
853 acoustics, *Journal of Fluid Mechanics* 775 (2015) 387–414.
- 854 [30] G. Mensah, L. Magri, A. Orchini, J. Moeck, Effects of asymmetry on  
855 thermoacoustic modes in annular combustors: a higher-order perturba-  
856 tion study, *J. Eng. Gas Turbines Power* 141 (2018) 041030 (8 pages).
- 857 [31] S. R. Stow, A. P. Dowling, Low-Order modelling of thermoacoustic limit  
858 cycles, in: *Proceedings of ASME Turbo Expo, 2004*, pp. GT2004–54245.
- 859 [32] B. Semlitsch, A. Orchini, A. P. Dowling, M. P. Juniper, G-equation  
860 modelling of thermoacoustic oscillations of partially premixed flames,  
861 *International Journal of Spray and Combustion Dynamics* 9 (2017) 260–  
862 276.
- 863 [33] M. Juniper, Sensitivity analysis of thermoacoustic instability with ad-  
864 joint helmholtz solvers, *Physical Review Fluids* 3 (2018) 110509.
- 865 [34] B. Schuermans, V. Bellucci, C. O. Paschereit, Thermoacoustic modeling  
866 and control of multi burner combustion systems, in: *ASME Turbo Expo,*  
867 *2003*, pp. GT–38688 (11 pages).
- 868 [35] A. Orchini, G. A. Mensah, J. P. Moeck, Effects of nonlinear modal inter-  
869 actions on the thermoacoustic stability of annular combustors, *Journal*  
870 *of Engineering for Gas Turbines and Power* 141 (2018) 021002 (10 pages).
- 871 [36] G. J. Tee, Eigenvectors of block circulant and alternating circulant ma-  
872 trices, *New Zealand Journal of Mathematics* 36 (2007) 195–211.

- 873 [37] G. A. Mensah, G. Campa, J. P. Moeck, Efficient Computation of Ther-  
874 moacoustic Modes in Industrial Annular Combustion Chambers Based  
875 on Bloch-Wave Theory, *Journal of Engineering for Gas Turbines and*  
876 *Power* 138 (2016) 081502 (7 pages).
- 877 [38] P. J. Schmid, M. F. de Pando, N. Peake, Stability analysis for  $n$ -periodic  
878 arrays of fluid systems, *Physical Review Fluids* 2 (2017) 113902.
- 879 [39] C. Pierre, E. H. Dowell, Localization of vibrations by structural irregu-  
880 larity, *Journal of Sound and Vibration* 114 (1987) 549–564.
- 881 [40] G. Ghirardo, C. D. Giovine, J. P. Moeck, M. R. Bothien, Thermo-  
882 acoustics of Can-Annular Combustors, *Journal of Engineering for Gas*  
883 *Turbines and Power* 141 (2019) 011007 (10 pages).
- 884 [41] P. E. Buschmann, G. A. Mensah, F. Nicoud, J. P. Moeck, Solution of  
885 thermoacoustic eigenvalue problems with a non-iterative method, in:  
886 *ASME Turbo Expo, 2019*, pp. GT2019–90834 (15 pages).
- 887 [42] A. P. Seyranian, O. N. Kirillov, A. A. Mailybaev, Coupling of eigenval-  
888 ues of complex matrices at diabolic and exceptional points, *Journal of*  
889 *Physics A: Mathematical and General* 38 (2005) 1723–1740.
- 890 [43] L. Magri, M. P. Juniper, Sensitivity analysis of a time-delayed thermo-  
891 acoustic system via an adjoint-based approach, *Journal of Fluid Me-*  
892 *chanics* 719 (2013) 183–202.
- 893 [44] A. Orchini, M. P. Juniper, Linear stability and adjoint sensitivity anal-  
894 ysis of thermoacoustic networks with premixed flames, *Combustion and*  
895 *Flame* 165 (2016) 97–108.

- 896 [45] R. Mennicken, M. Möller, Non-self-adjoint boundary eigenvalue prob-  
897 lems, Elsevier, 2003.
- 898 [46] P. Luchini, A. Bottaro, Adjoint equations in stability analysis, Annual  
899 Review of Fluid Mechanics 46 (2014) 493–517.
- 900 [47] N. Moiseyev, Non-Hermitian quantum mechanics, Cambridge University  
901 Press, 2011.
- 902 [48] E. Narevicius, P. Serra, N. Moiseyev, Critical phenomena associated  
903 with self-orthogonality in non-Hermitian quantum mechanics, Euro-  
904 physics Letters 62 (2003) 789–794.
- 905 [49] M. P. Juniper, R. I. Sujith, Sensitivity and nonlinearity of thermoacous-  
906 tic oscillations, Annual Review of Fluid Mechanics 50 (2018) 661–689.
- 907 [50] F. Sogaro, P. Schmid, A. Morgans, Sensitivity analysis of thermoacoustic  
908 instabilities, in: International Congress on Sound and Vibration, 2018,  
909 pp. 2063–2070.
- 910 [51] C. F. Silva, W. Polifke, Non-dimensional groups for similarity analysis  
911 of thermoacoustic instabilities, Proceedings of the Combustion Institute  
912 37 (2019) 5289–5297.
- 913 [52] M. Bauerheim, F. Nicoud, T. Poinsot, Progress in analytical methods to  
914 predict and control azimuthal combustion instability modes in annular  
915 chambers, Physics of Fluids 28 (2016) 021303.
- 916 [53] M. Bauerheim, A. Ndiaye, P. Constantine, S. Moreau, F. Nicoud, Sym-



- 917       metry breaking of azimuthal thermoacoustic modes: the UQ perspective,  
918       Journal of Fluid Mechanics 789 (2016) 534–566.
- 919 [54] T. Komarek, W. Polifke, Impact of swirl fluctuations on the flame re-  
920       sponse of a perfectly premixed swirl burner, Journal of Engineering for  
921       Gas Turbines and Power 132 (2010) 061503 (7 pages).
- 922 [55] M. A. Macquisten, M. Whiteman, S. R. Stow, A. J. Moran, Exploitation  
923       of measured Flame Transfer Functions for a two- phase lean fuel injector  
924       to predict thermoacoustic modes in full annular combustors, in: ASME  
925       Turbo Expo, 2014, pp. GT2014–25036.
- 926 [56] M. Bauerheim, J. F. Parmentier, P. Salas, F. Nicoud, T. Poinsot, An an-  
927       alytical model for azimuthal thermoacoustic modes in an annular cham-  
928       ber fed by an annular plenum, Combustion and Flame 161 (2014) 1374–  
929       1389.
- 930 [57] M. R. Bothien, N. Noiray, B. Schuermans, Analysis of azimuthal thermo-  
931       acoustic modes in annular gas turbine combustion chambers, Journal of  
932       Engineering for Gas Turbines and Power 137 (2015) 061505.
- 933 [58] G. Ghirardo, M. P. Juniper, M. R. Bothien, The effect of the flame phase  
934       on thermoacoustic instabilities, Combustion and Flame 187 (2018) 165–  
935       184.
- 936 [59] K. Ding, G. Ma, M. Xiao, Z. Q. Zhang, C. T. Chan, Emergence, coa-  
937       lescence, and topological properties of multiple exceptional points and  
938       their experimental realization, Physical Review X 6 (2016) 021007.