

Effect of electrochemical charging on the hydrogen embrittlement susceptibility of Alloy 718

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Abstract

The susceptibility of age-hardened nickel-based 718 superalloy to hydrogen embrittlement was studied by the controlled electrochemical charging combined with slow strain-rate tensile tests (SSRT) and advanced characterization techniques. We proposed some novel ideas of explaining hydrogen embrittlement mechanisms of the studied material in regard to two cracking morphologies: transgranular and intergranular cracking. It is for the first time to report that electrochemical charging alone could cause slip lines, surface and subsurface cracks on nickel-based superalloys. The formation of pre-damages was discussed by calculating the hydrogen concentration gradient generated during cathodic charging. Pre-damages were proved to result in transgranular cracks and lead to the evident reduction of mechanical properties. In addition, the STRONG (Slip Transfer Resistance of

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5 Neighbouring Grains) model was used to analyze the dependence of hydrogen-assisted
6 intergranular cracking on the microscopic incompatibility of the grain boundaries. The
7 results show that in the presence of hydrogen, grain boundaries with a lower dislocation
8 slip transmission are more prone to cracking during loading and vice versa.
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15 **Keywords:** Hydrogen embrittlement; nickel-based superalloy; hydrogen-induced cracking;
16 electrochemical hydrogen charging; slip transmission.
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21 **1 Introduction**

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23 Precipitation-hardened nickel-based superalloys, with their combination of excellent
24 mechanical properties, good corrosion resistance, and wide working temperature range, are
25 some of the best candidates used for subsea oil and gas industries. Despite their good
26 mechanical properties, these high-strength alloys are susceptible to hydrogen embrittlement
27 (HE) when they are used in a hydrogen-containing environment [1-9]. The mechanism of
28 HE has been extensively debated for decades [10]. Both experimental techniques and
29 computational models were applied to unravel the hydrogen-assisted failure process [11-13].
30 Several acceptable mechanisms proposed thus far include hydrogen-enhanced localized
31 plasticity (HELP) [12, 14, 15], hydrogen-enhanced decohesion (HEDE) [7, 16-18],
32 hydrogen adsorption-induced dislocation emission (AIDE) [11, 19-22], hydrogen-enhanced
33 strain-induced vacancy (HESIV) [23] and crack induction by hydride formation [6, 24-26].
34 Through years of understanding the role of hydrogen in nickel-based superalloys,
35 particularly precipitation-hardened alloys, several prevalent mechanisms were proposed.
36 When this alloy is deformed in hydrogen containing environment, two types of cracking
37 prevail: transgranular and intergranular cracking, which are explained by HELP [3, 27] and
38 HEDE [28, 29] the most. On one hand, recent studies [8, 28, 30] concluded that
39 transgranular cracking originated from the intersections of slip bands. The local stress
40 concentrations and strain discontinuities at the intersection points attract hydrogen, and the
41 local plastic instability in combination with non-uniform hydrogen concentration lead to the
42 primary cracking. On the other hand, hydrogen-assisted intergranular cracking was
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observed at delta (δ) phase decorated grain boundaries (GBs) [30], GB triple junctions [28], and high angle GBs [31]. The δ phase was proved to promote GB cracking through forming micro-cracks along its interfaces with the matrix in the framework of HEDE mechanism [7, 28, 29]. In addition, high elastic mismatch and strain localization at GBs and triple junctions facilitate decohesion and crack formation [28]. Kimura and Birnbaum [6] reported that a critical amount of hydrogen is a prerequisite for generating intergranular cracks in nickel alloys depending on the nature of the GBs and the corresponding stress state. Additionally, it has been documented that the cracking behavior of GBs is highly dependent on the misorientation angle [32, 33]. In other words, cracks are likely to initiate at high-angle GBs, while low-angle or low- Σ boundaries are more resistant to fracture [34]. However, to determine whether a GB has a high propensity to cracking or not, the measurement of misorientation alone, which is based on the 2D surface information, is far from being adequate due to the complex structure of GBs and the synergetic effect of adjacent grains upon deformation. During deformation of a polycrystalline metal, some GBs act as barriers for dislocation slip that causes dislocation pile-ups and local stress concentrations [35, 36]. Other GBs allows partial or full transmission of dislocations depending on the deformation behavior of adjacent grains and the character of GBs. By considering this, the concept of slip transferability and its impact on crack initiation along GBs was brought up to analyze the fatigue crack initiation and growth [37, 38]. And 3D characterization of the GB makes it possible to analyze the incompatibility stresses and reveal the transmission of the plasticity across the GB. However, this idea has not been involved in uncovering the underlying mechanisms of materials that failed in the hydrogenating environment. Analyzing stress compatibility of GBs is of great help to understand hydrogen-assisted intergranular cracking behavior.

Moreover, in most of the previous studies, hydrogen was introduced into the material through electrochemical charging. And it is known that harsh charging conditions, especially electrochemical charging, can cause surface damage in the form of blisters on steel and pure iron [39-42]. However, in face-centered-cubic (FCC) metals, where hydrogen diffusion is sluggish, the impact of hydrogen absorption and its influence on the cracking behavior during deformation is rarely documented. The most important reason for

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5 this is the difficulty of preserving a clean surface after long electrochemical hydrogen
6 charging. Thus far, most of the solutions used for electrochemical charging have been
7 water-based, which could cause surface corrosion after charging. In ex-situ tests, further
8 surface polishing before loading would introduce new stresses and thus conceal the useful
9 information of charging. In the case of in-situ tests, what we observe is a combined effect
10 of cathodic charging and mechanical loading, from which it is more difficult to disentangle
11 the HE mechanisms responsible for each type of cracking behavior.
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18 In the present study, a systematic analysis was conducted on the effect of cathodic charging
19 and its subsequent impact on the mechanical properties of age-hardened Alloy 718
20 superalloys. We used a glycerol-based electrolyte to preserve the surface integrity of the
21 alloy. Thus, direct observations and studies of the influence of electrochemical charging
22 could be realized. The material was cathodically charged and then subjected to direct-
23 surface and subsurface-damage characterization. A linkage between hydrogen-induced pre-
24 cracking and mechanical properties was set up. Later, the 3D information of GBs was
25 collected using Focused ion beam (FIB) cross section milling technique on the failed
26 hydrogenated samples after mechanical loading. We adopted the STRONG (Slip Transfer
27 Resistance of Neighbouring Grains) model proposed by Knorr [43] to analyze the
28 relationship between the slip system compatibility and hydrogen-assisted intergranular
29 cracking behavior. In this manner, we conducted a thorough analysis of the failure behavior
30 of Ni-based superalloys and tried to figure out the factors controlling transgranular and
31 intergranular cracking in the electrochemical charging conditions.
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46 **2 Materials and experimental procedure**

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48 In this work, the standard UNS N07718 (Alloy 718) alloy was used; its main chemical
49 composition is listed in Table 1. As received from the supplier, the alloy went through
50 standard API heat treatment with an aging temperature of 782 °C for 6.5 h before air
51 cooling. Two-step dog-bone tensile test specimens for the slow strain-rate tensile test
52 (SSRT) were cut by electrical discharge machining (EDM) as shown in Fig. 1(a).
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5 **Table 1** Chemical composition of Alloy 718 alloy
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10 **Fig. 1** (a) Specimen geometry for slow strain rate tensile test (all dimensions are shown in
11 mm) and (b) schematic of subsurface characterization direction.
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14 Before charging, the specimens were ground sequentially from 220 grit to 4000 grit SiC
15 papers and then polished with 3 μm and 1 μm diamond pastes. Electropolishing was
16 conducted in a methanolic H_2SO_4 solution at 26 V 30 s as the final step to remove the
17 deformed layer by previous mechanical polishing. Ex-situ hydrogen charging was
18 performed in a three-electrode electrochemical charging cell. A platinum net was used as
19 the counter electrode and Ag/AgCl was used as the reference electrode due to its wide
20 working temperature range. The charging process was performed in a 2:1 mixture of
21 glycerol and H_3PO_4 [44, 45] at 75 $^\circ\text{C}$ at a constant potential of -1050 mV (equals to a
22 cathodic current density of 15 mA/cm²). This solution offers the possibility of preserving a
23 corrosion-free surface after electrochemical charging. Later, the specimens were washed
24 with distilled water and ethanol and dried in circulating air flow. The dwell time between
25 finishing hydrogen charging and the damage characterization was controlled within 10 min.
26 In order to study the effect of electrochemical charging, different charging times of 20, 40,
27 60, 80, 100, 120 and 168 h were chosen.
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41 SSRT was performed at a constant nominal strain rate of 2×10^{-5} s⁻¹ on a Kammrath &
42 Weiss tensile/compression module inside the high-resolution Quanta FEG 650 scanning
43 electron microscope (SEM, Thermo Fisher Scientific Inc., USA) for a real-time recording.
44 The specimens were loaded until fracture to study the deformation process and crack
45 evolution in vacuum condition (4.7×10^{-3} Pa) and hydrogen-charged condition. The dwell
46 time before testing was controlled within 30 min. As the diffusion rate of hydrogen in the
47 FCC material is low (about 10^{-15} m²/s) [46], hydrogen loss within this period was
48 negligible. In order to study the possible damage below the fracture surface, the fractured
49 parts were electrodeposited in a Ni electroplating solution before subsurface analysis to
50 provide appropriate edge retention and mechanical stability during sample preparation. A
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5 dense layer of Ni with about 40 μm thickness was deposited. Afterwards, the coated
6 samples were mounted, grinded and polished. By using a caliper, the sample depth that was
7 removed from the fracture surface was precisely controlled to be 50, 350, and 650 μm
8 perpendicular to the tensile axis, as shown schematically in Fig 1(b). This sectioning
9 process was proved to cause no subsurface damages on the analyzed cross-sections. To
10 study the effect of hydrogen alone, the same subsurface characterization procedure was
11 conducted on the hydrogenated specimens without further mechanical loading. A thorough
12 microstructural characterization on both fracture surfaces and subsurfaces was conducted
13 using SEM embedded with backscattered electron (BSE) and electron dispersive
14 spectroscopy (EDS) detectors. Electron channeling contrast imaging (ECCI) was used to
15 reveal deformation structures after fracture. Electron backscattered diffraction (EBSD) was
16 conducted at an accelerating voltage of 20 kV with a step size of 200 nm-1 μm . Before
17 EBSD characterization, the samples were mechanically polished and finished with a 0.04
18 μm colloidal silica suspension (OPS) to obtain deformation-free surfaces. FIB (Helios
19 Nanolab DualBeam FIB, Thermo Fisher Inc., USA.) milling was performed to cut a trench
20 across each boundary to achieve a complete grain boundary trace information on
21 orthogonal planes to the sample surface. In addition, thermal desorption spectroscopy (TDS)
22 test was carried out to determine the total hydrogen content in the sample after hydrogen
23 charging. The TDS test was performed by a Bruker G4 PHONIX DH analyzer with a mass
24 spectroscopy detector at a heating rate of 25 $^{\circ}\text{C}/\text{min}$ from 50 $^{\circ}\text{C}$ to 750 $^{\circ}\text{C}$.
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44 **3 Results**

45 **3.1 Microstructural analysis**

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47 The initial microstructure of Alloy 718 after aging is shown in Fig. 2(a) with equiaxed
48 grains and an average grain size of 73.5 μm . Annealing twins can be observed inside the
49 grains, as highlighted by the yellow lines. Block-shaped nitrides (Nb, Ti)N and carbides
50 (Nb, Ti)C are present in the matrix as illustrated in the EDS results, respectively. The
51 coherent strengthening phases, spherical γ' and disc γ'' , are homogeneously distributed in
52 the matrix (Fig. 2(b)). A small fraction of the GBs is covered with needle-shaped
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5 intermetallic precipitate delta (δ) phases, which extend into the matrix (Fig. 2(c)) and left
6 with a precipitation-free zone (PFZ) due to the local depletion of Nb.
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11 **Fig. 2** Microstructure of Alloy 718 showing (a) carbides, (b) fine γ' and γ'' precipitates,
12 and (c) the δ phase. (d) EDS information on incoherent (Nb, Ti)N nitrides and (Nb, Ti)C
13 carbides.
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18 **3.2 Surface and subsurface response after electrochemical charging**

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20 The surface micrographs after charging are presented in Fig. 3, where we can see slip lines
21 on the surfaces after charging. Besides, it was also possible to detect cracks on the surface
22 after charging for more than 60 h (Fig. 3(c)–(g)). The entire formed cracks exhibit
23 transgranular modes, as illustrated in Fig. 3(c1). Cracks larger than $15 \mu\text{m}/\text{mm}^2$ started to
24 form after charging the samples for more than 80 h. In severe cases, e.g., when the sample
25 was hydrogenated for 168 h, its entire surface was filled with cracks with a tendency of
26 peeling (Fig. 3(g)). As it is difficult to determine the exact number of cracks in this
27 condition, the surface damage statistics excluding this case were characterized and
28 summarized in Fig. 3(h); in this diagram, it can be observed that both the number density
29 and the length density of surface cracks increase with charging time (≥ 40 h). And there is a
30 crack length jump at 80 h charging. In the current study, a time period of 40 h is set as the
31 threshold for hydrogen-induced surface cracking.
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44 **Fig. 3** BSE images ((a)-(c)), and SE images ((d)-(g)) showing slip lines and surface cracks
45 on Alloy 718 samples after electrochemical charging. The magnified view of the area
46 highlighted by the dashed circle in (c) is shown in (c1). (h) Statistical summary of the crack
47 number and crack length densities on the surfaces after charging from 20 h to 120 h.
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51 Apart from the surface damage induced by electrochemical charging, subsurface damages
52 were also studied in hydrogen-charged samples alone without the application of any
53 mechanical loading. Subsurface cracks can be observed after 100 h of charging. Take 120 h
54 charged condition as an example, we found that the crack cut through carbides in a
55 transgranular mode inside the grain (Fig. 4(a)). The crack path follows $\{111\}$ plane trace
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5 (red dash lines in the slip traces in Fig. 4(b)) and exhibits a symmetric morphology across
6 $\Sigma 3$ annealing twin boundaries (TBs), but it is not deflected by carbides. The displacement
7 and rotation of cracking attributes to their transmission through TBs. The formation of
8 crack caused a large amount of plasticity, which results in a bulge on the surface (Fig. 4(a))
9 as shown by the kernel average misorientation (KAM) in Fig. 4(c). A misorientation of 2°
10 to 8° is detected between the crack and the surface. When the crack reaches GBs, it is
11 transmitted into another grain (Fig. 4(e)). However, the crack has a tendency to propagate
12 along the GB which is decorated with precipitates (Fig. 4(d, d1)), and causes the debonding
13 of δ phase-matrix interface.
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24 **Fig. 4** SEM and EBSD analysis of subsurface cracks in the 120 h charged sample. (a), (d),
25 and (e) Micrographs illustrating crack paths. (b) ND-IPF (inverse pole figure) map and (c)
26 KAM map of the crack. (d1) is the ND-IPF map of the area highlighted by the yellow box
27 in (d). (TB: twin boundary, GB: grain boundary. The color scale of the IPF maps is shown
28 in the IPF triangle. The color scale of the KAM map is shown in the color bar.)
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33 **3.3 Slow strain rate tensile testing**

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35 To further investigate the effect of pre-formed cracks on the mechanical behavior of Alloy
36 718, SSRT was conducted after surface characterization. The engineering stress-strain
37 curves generated under different charging conditions are presented in Fig. 5(a). The 168 h-
38 charged sample was not considered as the sample was severely damaged during charging.
39 Fig. 5(b) shows that hydrogen decreases the ultimate tensile strength (UTS) due to the
40 reduction of the total elongation (from 27.6% without hydrogen to 2.06% after 120 h of
41 charging). The yield strength (YS), defined as 0.2% offset stress, increases with hydrogen
42 charging up to 60 h, after which it reduces gradually to a value lower than the YS in air at
43 120 h. The embrittlement factor (EF) is defined as eq. (1) [31, 47] and its evolution with
44 respect to the charging time is presented in Fig. 5(c). It describes the relative elongation
45 loss of the hydrogenated samples in comparison to the non-charged samples.
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$$55 \quad EF = \left(1 - \frac{\varepsilon_H}{\varepsilon_{no H}} \right) \times 100\% \quad (1)$$

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5 where ε_H is the engineering strain of a precharged sample after fracture and $\varepsilon_{no H}$
6 represents the engineering strain of a hydrogen-free sample. The EF increases with
7 charging time and becomes almost constant after 80 h of hydrogenation.
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14 **Fig. 5** (a) Engineering stress-strain curves of the tested specimens under different charging
15 conditions. Effect of hydrogen on (b) 0.2 % YS and UTS (b) and (c) EF.
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18 **3.4 Fractography**

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20 Post mortem fractography characterization with and without hydrogen exposure was done
21 after tensile tests. Fig. 6(a1)–(a2), (a4) show the fractography of hydrogen-free samples; the
22 fracture surface exhibited microvoid coalescence (MVC)-assisted failure in a ductile
23 fracture mode. Necking is a distinct feature of the ductile failure process (Fig. 6(a1)).
24 Higher magnification images of carbonitride cracks and dimples are presented in Fig. 6(a2)
25 and (a4). Micrographs in the vicinity of necking region show severely deformed grains with
26 secondary cracks across carbonitrides (Fig. 6(a3)). Two activated slip systems revealed by
27 ECCI are shown in Fig. 6(a5).
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35 Conversely, the hydrogenated samples displayed a multi-mode failure process. A
36 representative fracture surface of 20 h electrochemically charged sample is shown in Fig.
37 6(b1): it consists of a brittle area near the sample surface, a ductile region in the sample
38 interior enclosed by the red dotted line, and a transition from brittle to ductile regions in
39 between. In the brittle region, both intergranular and transgranular cracks are observed.
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41 Ridges, steps, and river patterns in the brittle region display transgranular features, and
42 nanovoids along the GBs exhibit a tendency for intergranular cracking (Fig. 6(b2)). The
43 steps and cleavage facets in Fig. 6(b3) show that dislocation slip preferred along certain slip
44 planes. Nanovoids are observed at slip line intersections (Fig. 6(b4)). In addition, secondary
45 transgranular cracks on the surface follow the direction of slip bands and exhibit zig-zag
46 morphologies (Fig. 6(b5)). The same case as in the non-charged sample, most carbides on
47 the surface are cracked.
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8 **Fig. 6** (a1)-(a5): Micrograph analysis of no-hydrogen sample after failure showing (a1, a4)
9 ductile fracture features, (a2, a3) carbonitride cracks, and (a5) surface slip lines; (b1)-(b5):
10 micrograph analysis of a fractured sample with 20 h-charging showing (b1) transition from
11 brittle to ductile fracture, (b2) GB microvoids, (b3) transgranular steps, (b4) microvoids at
12 slip lines intersections, and (b5) surface secondary cracks.
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15 Percentage of brittle area (POB) is introduced to visualize the effect of hydrogenation time
16 on the fracture mode and assess HE sensitivity:
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$$19 \text{POB} = \frac{A_T - A_D}{A_T} \times 100\% \quad (2)$$

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22 where A_T is the total area of the fracture surface and A_D is the area of the ductile region.
23 The results in Fig. 7(a) indicate an increase in POB with charging time. A drastic increase
24 in POB happened between 60 and 80 h. After 120 h of charging, the fracture surface
25 exhibits a fully brittle fracture mode. The corresponding brittle depths of the areas showing
26 brittle fracture in each charging condition are summarized in Fig. 7(b), which are consistent
27 with the changes in POB.
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36 **Fig. 7** Effect of hydrogen on (a) POB and (b) the corresponding brittle depth.
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40 **3.5 Subsurface crack analysis after loading**

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42 Subsurface cracks on three cross section layers were analyzed after loading the samples to
43 fracture. In the non-charged sample, the subsurface shown in Fig. 8(a) exhibits cracks in the
44 secondary-phase particles without intruding into the matrix, which is fully consistent with
45 the fractography in Fig. 6(a1). In comparison, both intergranular and transgranular cracks
46 are observed on the hydrogenated samples. Representative subsurface cracks with their
47 ND-IPF maps are shown in Fig. 8(b)–(e). In accordance with previous studies [37, 48, 49],
48 TBs show great resistance to crack propagation. An intergranular crack is deflected and a
49 transgranular crack becomes blunted when they reach TBs as shown in Fig. 8(c).
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6 **Fig. 8** Micrographs of the cross sections of (a) a hydrogen-free sample showing carbides
7 cracks. (b)-(c) BSE images and ND-IPF maps, respectively, of a hydrogen-charged sample
8 showing both intergranular and transgranular cracks. (d)-(e) BSE images and ND-IPF maps,
9 respectively, of a hydrogen-charged sample showing only intergranular cracks, step size:
10 0.2 μm .
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13 For a statistic analysis of these cracks, we divide them into three categories: transgranular
14 cracks, low- Σ ($\Sigma = 1-29$) [49], and general GB (all other GBs) cracks, as shown in Fig. 9(a).
15 Only one subsurface crack is observed at 40 h of hydrogen charging and three at 60 h of the
16 examined cross-sections. The total number of intergranular cracks remains almost constant
17 after charging the samples for more than 80 h, while that of transgranular cracks increases
18 drastically with respect to the charging time. Here we assumed that intergranular cracks
19 originate from the nearest triple junctions to the surface, and transgranular cracks initiation
20 points are measured by the distance from the surface to the nearest cracking point.
21 Examples of the above measurements are highlighted by the yellow arrows in Fig. 8(b)-(d).
22 Fig. 9(b) summarizes the results and indicates that most transgranular cracks are formed
23 near free surfaces. For samples charged for 100 h and 120 h, half of the cracks connect to
24 the surfaces. In comparison, intergranular cracks tend to nucleate deeper in the material
25 matrix. For samples charged between 80 to 120 h, the outermost intergranular cracks occur
26 at an average distance of (21.50 ± 4.99) μm to the free surface, which is in agreement with
27 the average outermost triple junction sites distance of (26.83 ± 4.56) μm . A detailed
28 discussion of both transgranular and intergranular cracking will be presented in sections 4.3
29 and 4.4.
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46 **Fig. 9** Statistical analysis of subsurface cracks: (a) the number of different types of cracks
47 on the cross-sections; and (b) distance between surface and cracks.
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50 51 52 **3.6 Slip transferability at GBs**

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54 Take a close look at the intergranular cracks and to understand the cracking mechanism in a
55 new perspective, 3D information of GBs was collected using FIB cross section milling
56 technique. Basically, in the STRONG model, a GB is described by two pairs of Euler
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5 angles (ϕ_1 Φ ϕ_2) in adjacent grains and two angles (η , ε) determining the direction of GB,
6 as illustrated in Fig. 10. The resistance of GB to slip transmission can be expressed as eq.
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9 (3) [37]:

$$\omega_{ij} = 1 - \cos \alpha_{ij} \cdot \cos \beta_{ij} \quad (3)$$

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11 where α is the angle between the intersection lines of the active slip plane in each grain
12 with the GB and β is the angle between the corresponding slip directions in adjacent grains.
13 A schematic of the residual Burges vector \vec{b}_r is defined by the difference of Burgers vector
14 of both slip systems. The matrix ω_{ij} contains in total 144 single slip systems couplings,
15 which yield a comprehensive description of the GB resistance to slip transmission. Fig. 11
16 shows two examples of GB cracks and the necessary parameters needed for ω calculation.
17 The depth tilt angle ε can be obtained from the cross sections of the GBs (Fig. 11(a3) and
18 (b4)). To measure the surface trace angle η , the direction parallel to the gauge length was
19 used as the reference line. The $\{111\}$ slip traces are marked in Fig. 11(a2) and (b2)–(b3)
20 with different colors indicating Schmid factor (SF) variation from the highest to the lowest
21 value (red-blue-light blue-dark). The active slip systems can be determined by comparing
22 the SFs with $\{111\}$ slip traces that were unambiguously measured by ECCI (Fig. 11(a2)
23 and (b2)–(b3)), and they are marked by dot lines.
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40 **Fig. 10** Schematic of slip system coupling at a GB, together with GB traces, represented by
41 angles (η , ε).
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45 **Fig. 11** ND-IPF maps (a) (b), and corresponding SE images (a1) and (b1) showing GB
46 cracks. The ECC micrographs (a2) (b2)-(b3) indicate active slip lines that represent $\{111\}$
47 traces (the magnitude of Schmid factors varies from the highest to the lowest value: red,
48 blue, light blue, black; the active slip planes are marked in dotted lines), (b2)-(b3) are
49 magnified ECC images of areas in (b1) that are marked in yellow. (a3) (b4) The FIB cross
50 sections show GB depth tilt angles.
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54 A summary of the measured GB parameters (including GB misorientation axes and angles),
55 calculated SFs, GB resistance on several cracked GBs (C-GBs) of a sample that fractured
56 after hydrogen charging is presented in Table 2. Non-cracked GBs (NC-GBs) next to the C-
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5 GBs, which are assumed to have similar stress level as those C-GBs, are presented as well.
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7 Since low- Σ GBs are relatively resistance to cracking compared to general GBs, we focused
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9 only on general GBs for the current study. The highest SF for each slip plane was
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11 calculated and presented in Table 2. The analyzed C-GBs and NC-GBs are in the range of
12
13 high misorientation angles. For each of the grains adjacent to the GB, at least two slip
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15 planes were activated. When the divergence between the intersection lines of active slip
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17 planes with the GBs (α angle) is lower than 15° , there is a higher probability of slip
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19 transferability [50]. All the NC-GBs shown in Table 2 have at least one coupling slip plane
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21 with relatively low α ($\leq 15^\circ$). In comparison, α angles of the C-GBs are much higher, with
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23 only one exception (gA-3/gB-3). Also, the parameter representing the resistance of GB to
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25 slip transmission, ω , for C-GBs are higher than those for NC-GBs. This indicates that, in
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27 the presence of hydrogen, GBs that possess low slip transmission have high stress
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29 incompatibility and can initialize crack easily.
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32 **Table 2** Measured GB parameters, active slip planes with corresponding SFs, and GB
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34 resistance calculated from eq. (3). (U V W)/ θ represents the misorientation axis and angle
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36 between adjacent grains.
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38 **4 Discussion**

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40 The aim of this study is to perform a systematic hydrogen charging process to analyze the
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42 impact of electrochemical charging alone on the microstructure change of Alloy 718, and
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44 the relation between hydrogen-induced pre-damages and the cracking behaviors in the
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46 process of mechanical loading. In the discussion part, firstly hydrogen concentration profile
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48 along the sample was calculated to help explain the hydrogen-induced slip lines and cracks
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50 during hydrogen adsorption. Subsequently, the possible reasons for transgranular cracking
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52 and intergranular cracking were discussed in detail.
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54 **4.1 Pre-damages induced by cathodic charging**

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57 The results presented above show that slip lines were formed on the surface of Alloy 718
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59 followed by surface and subsurface cracking in the process of electrochemical charging.
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5 These phenomena were rarely detected and documented. The most probable reason for not
6 being able to detect them can be attributed to the surface condition after charging, which
7 exhibits a relatively high surface roughness or corrosion. In the present study, the sample
8 surface was electropolished before hydrogen charging to guarantee a surface with no
9 residual stress and the usage of a glycerol-based electrolyte ensured surface integrity.
10 Therefore, the microstructure change on the sample in the process of cathodic charging
11 could be recorded. The formation of surface and subsurface cracks shows severe damage
12 caused by hydrogen ingress alone. Due to a low hydrogen diffusivity in Alloy 718, it is
13 reasonable to infer that a gradient in hydrogen concentration from the edge to the inner
14 region is developed with the highest concentration at the surface. And the internal stress
15 due to hydrogen distribution inhomogeneity plays a crucial role. To visualize hydrogen
16 concentration distribution during different charging time, the plane sheet diffusion model
17 [51] with a fixed diffusion coefficient was used. In this context, we need to solve the one-
18 dimensional form of Fick's second law, which is represented by eq. (4):
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$$\frac{\partial C}{\partial t} = D \frac{\partial^2 C}{\partial^2 x} \quad (4)$$

31
32 where C is the hydrogen concentration at a specific time t and distance x from the center of
33 the specimen along the thickness direction. D is the hydrogen diffusion constant, which is
34 independent of hydrogen concentration and local stress state, and can be calculated as eq. (5)
35 [52]:
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$$D = 4.06 \times 10^{-7} \exp\left(-\frac{48.63 \text{ kJ/mol}}{RT}\right) \text{ m}^2/\text{s} \quad (5)$$

42
43 where R is the gas constant, and T is the charging temperature (348.15 K in the present
44 study). This yields the value of D $2.05 \times 10^{-14} \text{ m}^2/\text{s}$. Considering that the specimen is
45 symmetric at $x = 0$ in the center, the boundary conditions in the region of $-L \leq x \leq L$ (L
46 equals half length of the thickness) for this equation can therefore be written as eqs. (6)-(8):
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$$C = C_0, x = L, t \geq 0 \quad (6)$$

$$C = C_0, x = -L, t \geq 0 \quad (7)$$

$$\frac{\partial C}{\partial x} = 0, x = 0, t \geq 0 \quad (8)$$

where C_0 is the concentration of hydrogen adsorbed at the surface. Assuming that hydrogen concentration in the specimen before charging is 0 and use Laplace transformation, the solution of eq. (4) can be expressed as eq. (9):

$$\frac{C}{C_0} = \sum_{n=0}^{\infty} (-1)^n \operatorname{erfc} \frac{(2n+1)L-x}{2\sqrt{Dt}} + \sum_{n=0}^{\infty} (-1)^n \operatorname{erfc} \frac{(2n+1)L+x}{2\sqrt{Dt}} \quad (9)$$

The solution of eq. (9) is schematically shown in Fig. 12(a) in section 4.2. The calculated hydrogen concentration ratio terminates at different points, corresponding to a variation in samples' thickness. Obviously, a steep concentration gradient is generated along the sample, which is a common feature in FCC metals during cathodic charging [30, 53, 54]. In this alloy, the aging process annealed out most of the internal stress induced by water quenching, and electropolishing before hydrogen charging guaranteed the surfaces free from residual stresses. By hydrogen charging, the lattice expands, and stress evolves in the crystal. As soon as the stress reaches the level required for activation of dislocations, dislocation loops are emitted to accommodate the strain gradient in the sample and those dislocations can be assumed to be geometrically necessary dislocations (GND). Besides, the introduction of hydrogen can reduce the dislocation line energy, which further contributes to the activation of Frank-Read source [55-57]. And a high supersaturation of hydrogen accompanied by the high stresses near the surface [12] act as a driving force for dislocations to slip and leave slip traces at free surfaces (Fig. 3). This phenomenon was also reported in an austenitic phase of a super duplex stainless steel during in-situ electrochemical hydrogen charging [55]. It is also reasonable to infer that, at longer hydrogen charging time, cracks on the surface can generate at slip band intersections where stress concentration is high (Fig. 3(c1)). A similar process occurred during the pulsed plasma nitriding (PPN) of austenitic stainless steel [58]; in that case, the nitrogen content across the PPN layer was measured using an electron probe micro-analyzer, which is responsible for the observed dislocation density gradient. However, in the present study, a direct measurement of hydrogen content along the sample was difficult due to the limitations of achieving resolution within the available techniques.

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6 It was previously proposed that subsurface cracks are caused by the accumulation of
7 hydrogen atoms at microscopic defects, such as voids, second phase particles, and grain
8 boundaries [59]. However, a recent study by Griesche et al. [60] stated that the presence of
9 a second phase is not a prerequisite for cracks formation. And no hydrogen segregation was
10 detected at the interfaces of carbonitrides by using Nanoscale Secondary Iron Mass
11 Spectrometry [29]. In this case, hydrogen-induced subsurface cracks could be related to
12 hydrogen segregation at pre-existing voids or grain boundaries. Meanwhile, hydrogen-
13 induced cracking with a noticeable opening displacement beneath the surface were reported
14 to be caused by surface blisters [39-41, 61]. However, most of the studies were on body-
15 centered cubic (BCC) metals. Only one paper mentioned the hydrogen-induced blister on
16 aluminum alloys [62], but subsurface characterization was not performed. Moreover, none
17 of these studies reported the formation of slip traces on the surface before blister formation.
18 To determine whether such internal crack (Fig. 4) is a hydrogen-induced blister or not,
19 hydrogen fugacity inside the crack and on the surface was discussed. On one hand,
20 assuming that the internal gas pressure P_{H_2} at the internal cavity that was caused by the
21 recombination of hydrogen atoms into hydrogen molecules around the defects can be
22 roughly estimated from eq. (10) [63]:
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$$37 \quad P_{H_2} = \frac{K_{IC}\sqrt{\pi}}{2\sqrt{a}} \quad (10)$$

38 where K_{IC} is the fracture toughness of Alloy 718 alloy ($95.8 \text{ MPa}\sqrt{\text{m}}$) [64] and a is the
39 radius of the crack, which is assumed to be half size of the crack length. Here we chose the
40 120 h-charged sample for an estimation of the internal hydrogen pressure. This yields P_{H_2}
41 the value of 8.5 GPa. And the hydrogen fugacity in the cavity can be calculated by using
42 the Abel-Noble equation (eq. (11)) [65]:
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$$50 \quad f = P_{H_2} \cdot \exp\left(P_{H_2} \frac{b}{RT}\right) \quad (11)$$

51 where $b = 1.584 \cdot 10^{-5} \text{ m}^3 \text{ mol}^{-1}$, and this results in the extremely high values of f (order
52 of 10^{25} atm). On the other hand, the hydrogen fugacity on the sample surface f_s is
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5 proportional to the hydrogen surface concentration C_0 during electrochemical charging, as
6 described by the Sivert's law [66]:
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$$C_0 = S_H \sqrt{f_s} \quad (12)$$

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11 where S_H is the solubility limit of hydrogen during charging, which can be measured from
12 eq. (13) [67]:
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$$S_H = 8.8 \cdot \exp\left(\frac{-8.1 \text{ kJ/mol}}{RT}\right) \text{ wt ppm H}/\sqrt{\text{atm}} \quad (13)$$

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17 And the hydrogen surface concentration can be estimated from the total hydrogen
18 concentration in the sample C_M by using eq. (14) [68]:
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$$C_0 = \frac{\omega C_M}{4} \sqrt{\frac{\pi}{Dt}} \quad (14)$$

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24 where ω equals to the sample thickness. As shown in Fig. 3, 20 h is the least hydrogen
25 charging time that can generate slip lines on the surface without causing any surface and
26 subsurface damages. Therefore, TDS test has been performed after the sample was
27 hydrogenated for 20 h, which yields C_M as 36.5 wt. ppm. The corresponding hydrogen
28 surface concentration C_0 and the hydrogen fugacity f_s yield the values of 189.4 wt. ppm
29 and 1.2×10^5 bar, respectively. The huge difference between hydrogen fugacity in the cavity
30 and on the surface indicates that the internal crack generated during cathodic charging is
31 not a high internal pressure-induced blister. If the blister formation were assumed, the
32 hydrogen fugacity in the cavity and on the surface should be in a comparable level due to
33 the continuous equilibrium of the hydrogen activities near to the surface. Thus, it is
34 reasonable to infer that these cracks were initiated at stress concentrated areas such as GBs
35 and TBs due to the formation of internal stress during charging.
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49 To substantiate that the surface slip lines and cracks are caused by the internal stress due to
50 hydrogen distribution inhomogeneity along the sample, the internal stress σ is calculated
51 and compared with the critical resolved shear stress (CRSS) for dislocation slip. The
52 internal stress induced by abundant dissolved hydrogen can be estimated by eq. (15):
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$$\sigma = B \cdot \frac{\Delta V}{V_0} \quad (15)$$

where B is the bulk modulus of Alloy 718, and $\Delta V/V_0$ is the lattice volume change by hydrogen. The value of B cannot be obtained directly from the literature, however, one can estimate it from the bulk modulus of γ'' phase (211 GPa) and γ' phase (229 GPa) in Inconel 718 [69]. In addition, the lattice expansion $\Delta V/V_0$ has a linear relationship with the dissolved hydrogen [70]:

$$\frac{\Delta V}{V_0} = \frac{H_a}{M_a} \frac{\Delta v}{\Omega} \quad (16)$$

where H_a/M_a represents the atomic ratio between hydrogen and metal near to the surface during charging, Δv is the volume change per hydrogen atom, and Ω is the mean volume of metal atom. For Ni, this ratio was reported as 0.28 [71]. The atomic ratio H_a/M_a can be calculated by eq. (17):

$$H_a/M_a = \frac{C_0 \times N_A/M_H}{(10^6 - C_0) \times N_A/M_A} \quad (17)$$

Where N_A is the Avogadro constant, M_H and M_A are the molar mass of hydrogen (1 g/mol) and the studied Alloy 718 (59.5 g/mol), respectively. As a result, the internal stress σ near to the surface in 20 h charging condition yields 665.7 MPa. This value is higher than the CRSS (218.9 MPa) for slip in a single crystal that is deduced from the measured polycrystalline yield stress σ_Y (669.8 MPa for 20 h charging) using the Taylor factor (3.06) [72]. The above calculations confirm that the internal stress generated during electrochemical charging is high enough to be the driving force for the hydrogen-induced surface damages. Worthy of note, the estimated value of the internal stress, which is only an approximation based on the Hook's law, though not accurate, is able to qualitatively testify the mechanism for the plastic deformation during hydrogen charging. This deduction also applies to the longer time charging conditions.

4.2 Failure mode of Alloy 718 with the presence of hydrogen

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5 The deformation mechanism in hydrogen-charged samples is different from that of
6 hydrogen-free samples. The failure mode of Alloy 718 transfers from ductile to brittle
7 fracture. Both transgranular and intergranular fracture modes were found on samples failed
8 in a hydrogenation environment. By introducing hydrogen, YS increased at first due to the
9 solid solution strengthening effect of hydrogen, which was also reported for iron [73] and
10 FCC high-entropy alloys [74, 75]. During the increment of charging time, YS after 80 h
11 charging reduced, which might be caused by the internal and external pre-damages
12 generated during hydrogen charging (Fig. 3). This resulted in a rapid fracture after reaching
13 the yield point. At the same time, there is a drastic change of EF and POB results between
14 60 and 80 h (Fig. 7(a)), which fits well with the sudden increment in surface crack length
15 density after charging (Fig. 3(h)), indicating again that pre-damages induced by cathodic
16 charging assist both the formation and growth of cracks in the process of mechanical
17 loading. Once a pre-crack is large enough, it propagates readily at a lower stress level,
18 which results in the early brittle fracture of the sample.
19

20 It is worth mentioning that the brittle depth at each charging condition measured from
21 fractography corresponds well to the hydrogen concentration at approximately $0.12 C_0$ (Fig.
22 12 (a)) that obtained from the simulation, as illustrated in Fig. 12(b). This result means that
23 the sample region with a hydrogen concentration reaches to this level (22.7 wt. ppm) during
24 electrochemical charging can cause a brittle fracture upon mechanical loading. Although
25 hydrogen charging solution in Ref. [67] is different from the current study, the usage of
26 both acidic electrolytes can, to some extent, give a hint for the amount of hydrogen that can
27 initiate brittleness.
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33 **Fig. 12** (a) Hydrogen concentration profile for different charging time and (b) comparison
34 of hydrogen effect on brittle depth measured from fracture surface and the simulated 0.12
35 C_0 results.
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38 **4.3 Hydrogen-assisted transgranular cracking during deformation**

39 Concentration gradient during hydrogen charging results in compressive stresses in the
40 outer regions of the sample [12], which in turn causes slip lines and surface/subsurface
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5 cracks even without mechanical loading. Once the sample is loaded, those pre-damages can
6 easily propagate with the assistance of hydrogen accumulated at these stress concentration
7 regions. Since part of the transgranular cracks were surface cracks, and there is a drastic
8 increment of transgranular cracks between 80 h and 120 h (Fig. 9(a)) even though they
9 fractured at similar deformation level, it is reasonable to infer that surface cracks induced
10 by hydrogen charging promoted the formation and propagation of transgranular cracks.
11 Concomitantly, intersections of slip lines act as strain concentration areas that attract
12 hydrogen, and the formation of embryos of nanovoids becomes easier (Fig. 6(b4)) [8, 28,
13 30]. Quasi-cleavage is a commonly reported fracture feature in the hydrogenated
14 precipitation-hardened Alloy 718 [9, 29, 76, 77]. Hydrogen-induced cracking along
15 dislocation slip bands (DSBs), particularly at nonparallel DSBs intersections near the
16 surface [8, 9, 76], has been claimed to be the reason for the observed quasi-cleavage feature.
17 Hydrogen atmospheres reduce the tendency for dislocation cross-slip by stabilizing the
18 edge component [78]. Furthermore, the reduction of SFE due to the presence of hydrogen
19 [68, 79, 80] promotes slip planarity and enhances the stress concentration at the
20 intersections of {111} slip planes. In the current study, another possible reason for the
21 occurrence of surface transgranular cracking upon loading is the difference in the stress
22 conditions between surfaces and subsurfaces. At regions close to the surface where plane
23 stresses dominate, grains experience relatively lower incompatibility stresses and
24 constraints from neighbouring grains. Thus, surface GBs are tougher and show less
25 cracking compared to the grain interior. This in return promotes the possibility of
26 transgranular cracking on the surface. To the authors' knowledge, this is for the first time to
27 report that hydrogen charging alone causes slip lines and cracks in nickel-based superalloys.
28 It is worth mentioning that the charging condition is relatively moderate and without any
29 hydrogen poison. Compared to the room temperature (25 °C), the current charging
30 temperature (75 °C) increases the hydrogen diffusion rate and solubility by 16.7 and 1.6
31 times, respectively, according to eqs. (5) and (13). Thus, one would expect a more severe
32 hydrogen charging condition with a steeper hydrogen concentration gradient at room
33 temperature. Compared to pure Ni and other nickel-based alloys showing intergranular
34 failure, the pronounced transgranular fracture in the tested Alloy 718 could be attributed to
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5 several reasons. It was reported that the failure mode of Alloy 718 is highly dependent on
6 the virgin microstructure [7, 8, 30]. When the GBs are heavily decorated with δ phase,
7 intergranular failure is more frequently observed, which is not the case in the current
8 material with merely small fraction of δ phase. Additionally, hydrogen-induced internal
9 stress initiates dislocation slips and possible transgranular cracks with prolonged charging
10 time. Those pre-damages act as pre-existing stress concentrators that can substantially
11 promote transgranular cracking upon loading. Moreover, the pronounced localized planar
12 slip due to the high volume of nano-sized precipitates facilitates the slip planes as the
13 sources of cracks rather than the GBs. With mechanical loading, transgranular crack
14 evolves more, albeit both types of cracks are evolving. The results showed that hydrogen-
15 charging induced pre-damages had a detrimental effect on the mechanical behavior of
16 nickel-based superalloys. And it is necessary to consider this effect on HE in future
17 experimental studies.
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32 **4.4 Hydrogen-assisted intergranular cracking during deformation**

33 Hydrogen-assisted intergranular cracking was observed in the precipitation hardened
34 nickel-based superalloys. Different from the debonding mechanism proposed for
35 intergranular cracking in the presence of δ phase [28, 30, 53], in this study, we introduced a
36 new parameter, i.e. GB resistance to slip transmission, to understand the stress
37 incompatibility between the adjacent grains and its impact on the propensity for GB
38 cracking. In general, GBs will block dislocations from intragranular movement and
39 enhance local stress incompatibility, which in turn increases the risk of crack initiation [81].
40 The transferability of slip systems through GBs has been proved to play a crucial role in the
41 evaluation of local GB flow stress in pure Ni [82] and GB crack initiation in mild steels
42 [43]. To predict which slip system is favorable for slip transfer, Lee, Robertson and
43 Birnbaum [83] proposed three criteria: (a) the angle between the intersection lines of the
44 slipping planes with the GB plane should be minimal; (b) the component of the residual
45 Burgers vector that protrudes from the grain boundary should be minimal; (c) the resolved
46 shear stress acting on the active slipping system should be maximum. The introduction of
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5 GB resistance to slip transmission, ω_{ij} , provides the possibility of describing dislocations
6 transferability at GBs by considering conditions (a) and (b). Condition (c) is not so easy to
7 determine due to the superimposed local stress caused by the anisotropic elastic
8 incompatibility of the two neighbouring grains. The normalized effective normal stress
9 acting on a GB can be described by the Schmid-Modified Grain Boundary Stress (SMGBS)
10 model [84]. However, it does not provide quantitative results, but rather an indication of the
11 stress field variation with a discrepancy in grain orientations. Meanwhile, one should
12 consider the synergetic effect of all these factors, since the slip system experiencing the
13 maximum resolved shear stress need not be activated if conditions (a) and (b) are not
14 satisfied [85]. In the current study, we focused on conditions (a) and (b), which are more
15 important when more than one active incoming or outgoing slip system operates. Assuming
16 that the critical value for ω_{ij} is $\omega = 1 - \cos 15^\circ \cdot \cos 15^\circ = 0.067$, which is in line with the
17 suggestions made by Davis [50] and Werner [86], it is quite obvious that in the hydrogen-
18 charged samples, ω_{ij} of C-GBs is much higher than the critical value (Table 2). This means
19 that the transferability of slip at those GBs is much more difficult, and a high stress
20 incompatibility will arise at those regions. An intuitive example of the non-transferrable
21 slip is the high flow stress and local buckling at GBs due to dislocation pile-ups, which was
22 reported on the deformed high-angle GB bi-crystal micropillars [82]. Furthermore,
23 molecular dynamic simulations of iron in Ref. [87] claimed that hydrogen atoms
24 segregating along the GB distort the lattice structure, thus consistently increase the energy
25 barrier to slip transmission. Meanwhile, slip planarity is enhanced in FCC metals in the
26 presence of hydrogen owing to the reduction of SFE [79, 80]. Hence, local flow stress
27 increases at the intersections of GBs with $\{111\}$ sliding planes. The accumulation of
28 hydrogen atoms at those stress concentration areas further reduces the defect-formation
29 energy in the framework of “the Defactant” model proposed by Kirchheim [57, 88]. As a
30 result, cracks occur readily. On the other hand, the GB resistance of NC-GBs is very close
31 to the critical value, which indicates that a slip can pass through more easily in those GBs
32 with higher microscopic compatibility. It is worth mentioning that all the analyzed GBs in
33 Table 3 are free of δ phase, which eliminates the detrimental effect of δ phase when
34 assessing slip transferability at GBs. The δ phase was proposed to exert a nontrivial effect
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5 on the intergranular GB cracking by initializing micro-cracks [28, 29], since a soft PFZ
6 around δ enhances stress localization. In addition to avoiding the formation of the δ phase
7 through optimizing the heat treatment processes, attention should also be paid on how to
8 achieve GBs that hold high slip transferability in the concept of GB engineering.
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15 **5. Conclusions**

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17 Hydrogen embrittlement in electrochemically charged Alloy 718 was investigated using
18 slow strain rate tensile test (SSRT) in combination with advanced characterization
19 techniques, i.e. secondary electron microscopy (SEM), electron channeling contrast
20 imaging (ECCI), electron backscattered diffraction (EBSD), and electron dispersive
21 spectroscopy (EDS). Focused ion beam (FIB) was used to collect 3D information of GBs.
22 A systematic study was conducted on the impact of hydrogen charging on the alloy's
23 microstructure change and its mechanical properties during mechanical loading. The
24 transgranular cracking mechanism was discussed in detail by calculating hydrogen
25 concentration profiles. In this study, GB compatibility, i.e., dislocation slip transferability
26 was considered for the first time to explain the intergranular fracture in nickel-based
27 superalloys by using STRONG model. The major conclusions can be summarized as:
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- 37 1) The presence of hydrogen has a strong detrimental effect on the mechanical
38 properties of Alloy 718; the mechanical property degradation is accompanied by a
39 transition from microvoid coalescence (MVC)-induced ductile failure to hydrogen-
40 assisted brittle fracture. Both transgranular and intergranular cracks could be
41 observed. In the current material, sample regions with a hydrogen concentration
42 reaches to 22.7 wt. ppm during electrochemical charging can initiate brittleness
43 upon mechanical loading.
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- 50 2) Electrochemical charging has a prominent effect on the cracking behavior of Alloy
51 718. Cathodic charging alone caused slip lines, surface and subsurface cracks. The
52 formation of pre-damages results in a reduction of yield strength, an increment of
53 embrittlement factor and percent of brittle area, and finally assists an early failure of
54 the alloy. Hydrogen concentration gradient accompanied by the high stresses near
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5 the surface is responsible for the initiation of the pre-damages. These damages are
6 cracking sources in the process of mechanical loading.
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9 3) The formation of transgranular cracks resides in two reasons: (a) pre-damages
10 caused by electrochemical charging before loading promoted the initiation and
11 propagation of transgranular cracks; (b) intersections of dislocation slip bands on
12 {111} planes act as stress concentrators that attract hydrogen and lead to void
13 nucleation, which finally results in cracking.
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15 4) The GB transferability of dislocations is an important factor determining the
16 propensity for intergranular cracking. In the presence of hydrogen, GBs with high
17 resistance to slip transmission increases the probability for crack initiation upon
18 loading due to the synergetic effect of hydrogen and local stress incompatibility,
19 and GBs with lower resistance to slip transmission are less susceptible to cracking.
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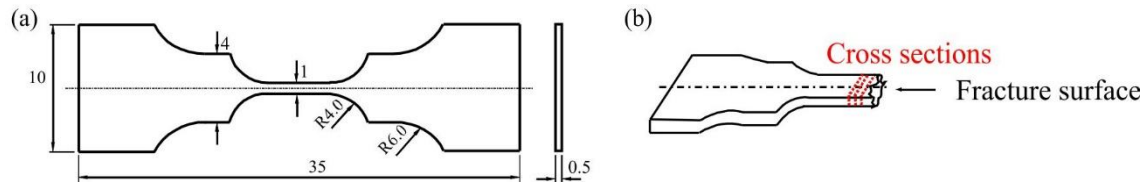


Fig. 1 (a) Specimen geometry for slow strain rate tensile test (all dimensions are shown in mm) and (b) schematic of subsurface characterization direction.

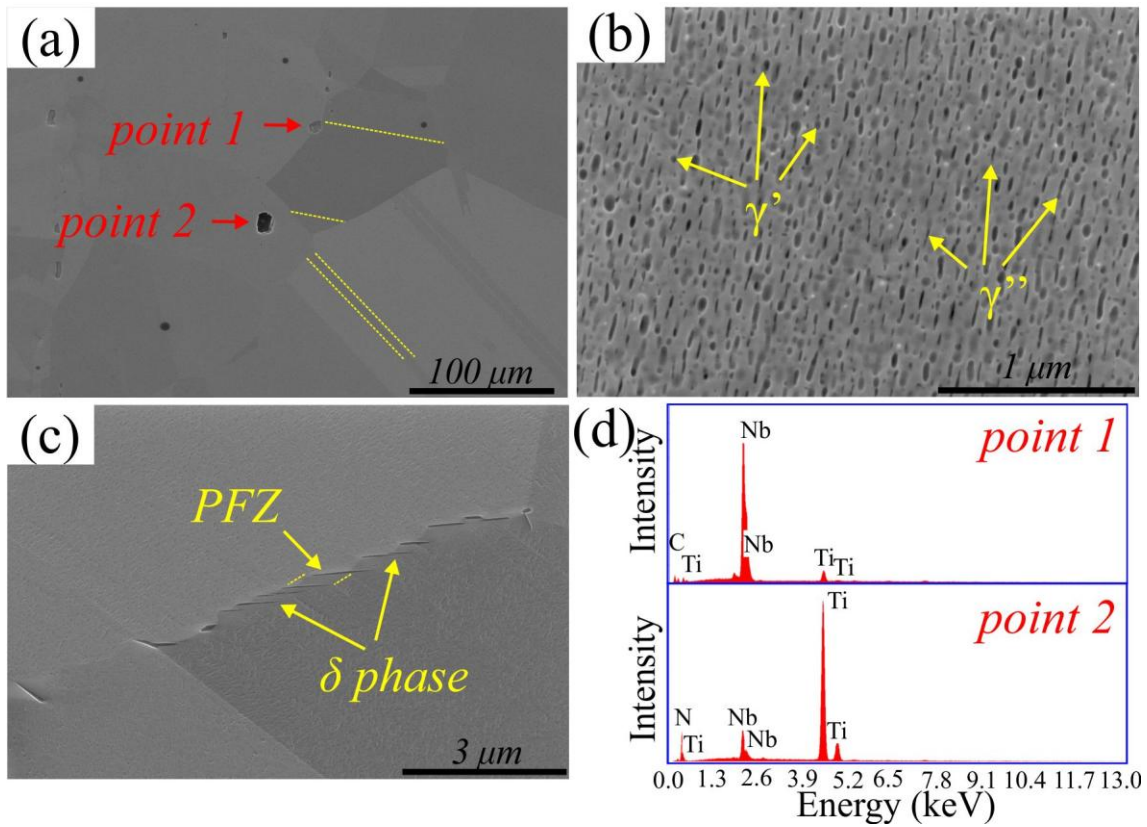


Fig. 2 Microstructure of Alloy 718 showing (a) carbides, (b) fine γ' and γ'' precipitates, and (c) the δ phase. (d) EDS information on incoherent (Nb, Ti)N nitrides and (Nb, Ti)C carbides.

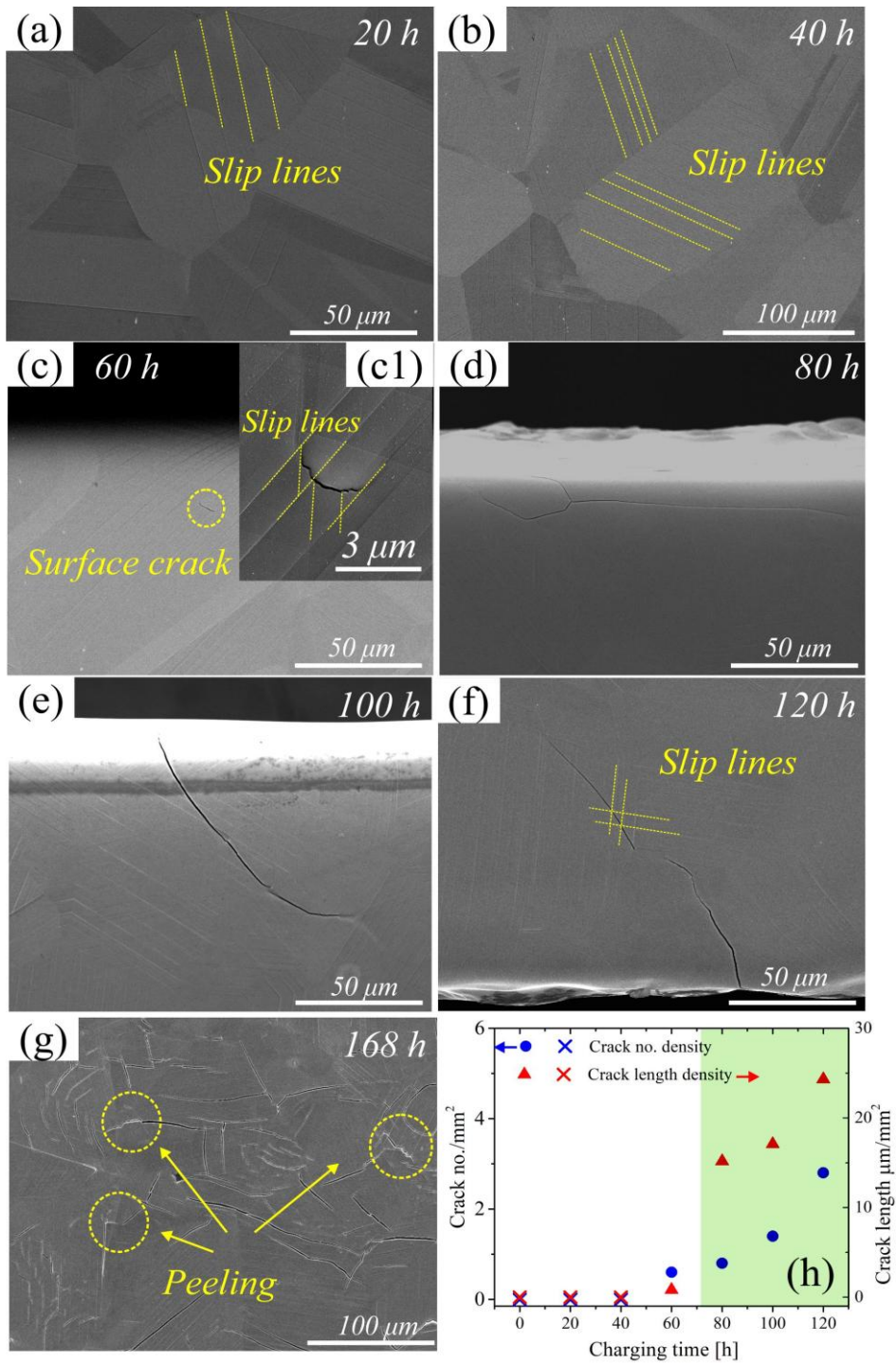


Fig. 3 BSE images ((a)-(c)), and SE images ((d)-(g)) showing slip lines and surface cracks on Alloy 718 samples after electrochemical charging. The magnified view of the area

highlighted by dashed circle in (c) is shown in (d1). (h) Statistical summary of the crack number and crack length densities on the surfaces after charging from 20 h to 120 h.

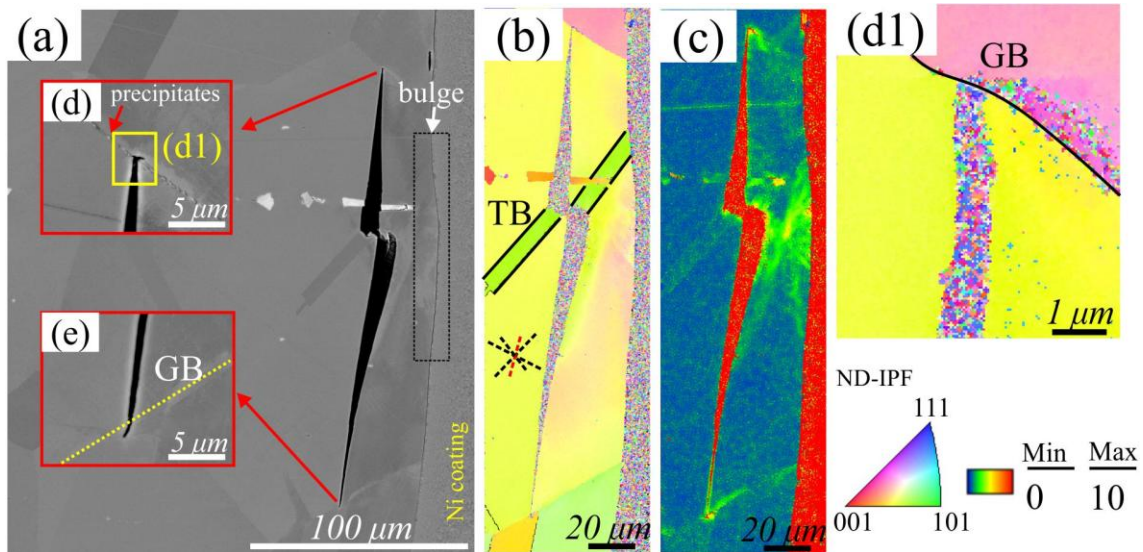


Fig. 4 SEM and EBSD analysis of subsurface cracks in the 120 h charged sample. (a), (d), and (e) Micrographs illustrating crack paths. (b) ND-IPF (inverse pole figure) map and (c) KAM map of the crack. (d1) is the ND-IPF map of the area highlighted by the yellow box in (d). (TB: twin boundary, GB: grain boundary. The color scale of the IPF maps is shown in the IPF triangle. The color scale of the KAM map is shown in the color bar.)

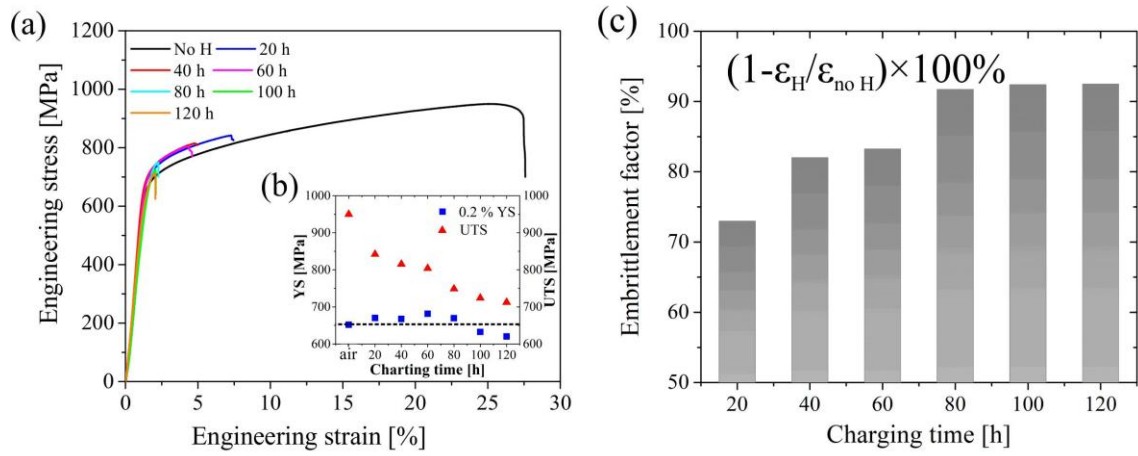
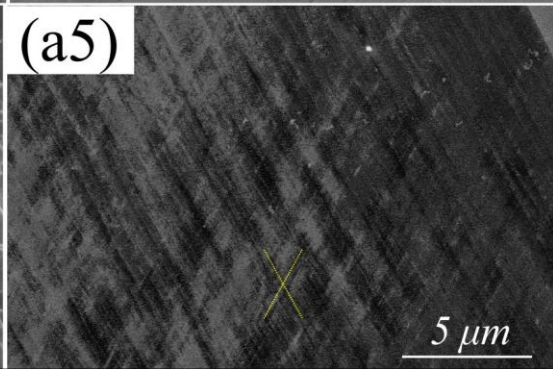
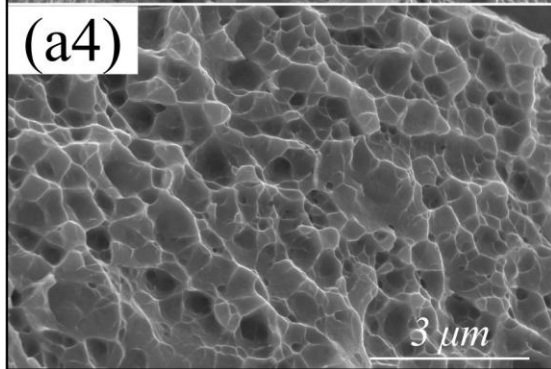
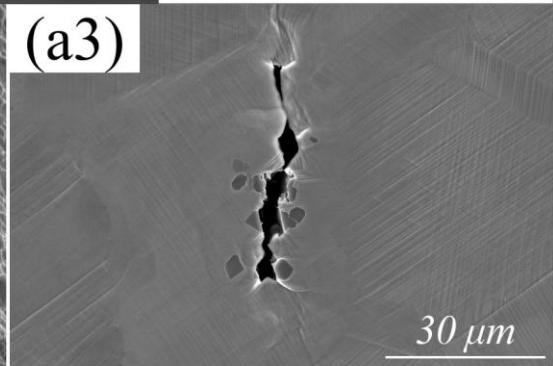
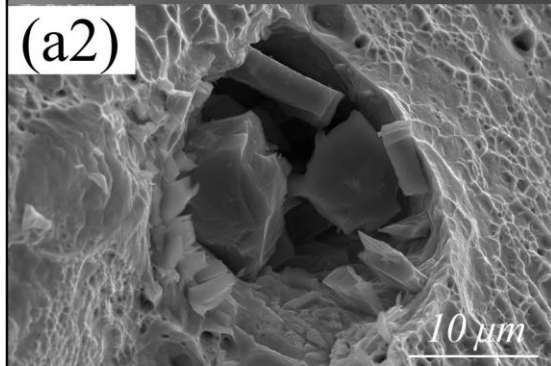
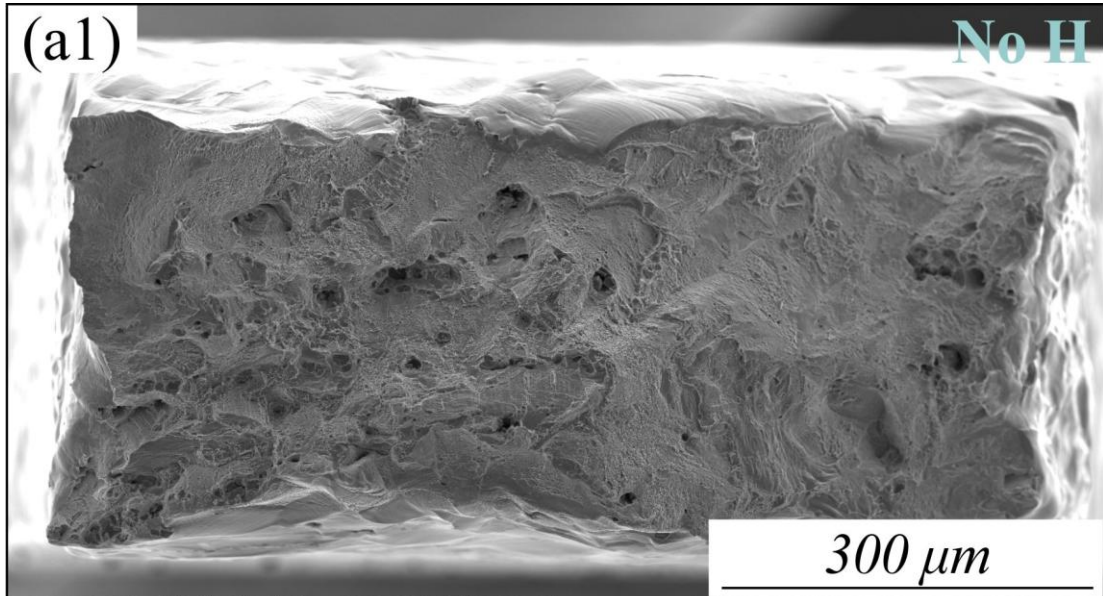


Fig. 5 (a) Engineering stress-strain curves of the tested specimens under different charging conditions. Effect of hydrogen on (b) 0.2 % YS and UTS (b) and (c) EF.



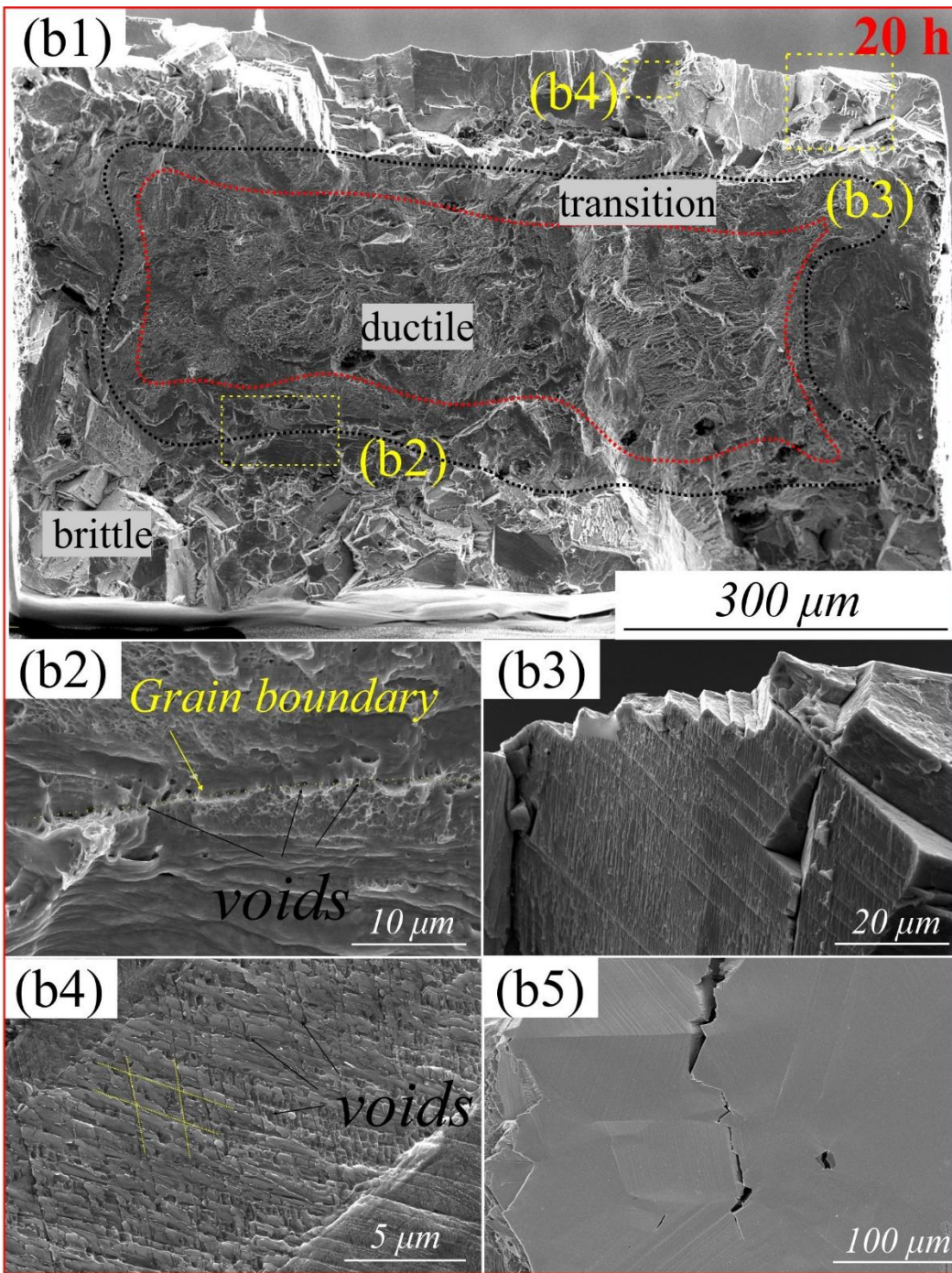


Fig. 6 (a1)-(a5): Micrograph analysis of no-hydrogen sample after failure showing (a1, a4) ductile fracture features, (a2, a3) carbonitride cracks, and (a5) surface slip lines; (b1)-(b5): micrograph analysis of a fractured sample with 20 h-charging showing (b1) transition from brittle to ductile fracture, (b2) GB microvoids, (b3) transgranular steps, (b4) microvoids at slip lines intersections, and (b5) surface secondary cracks.

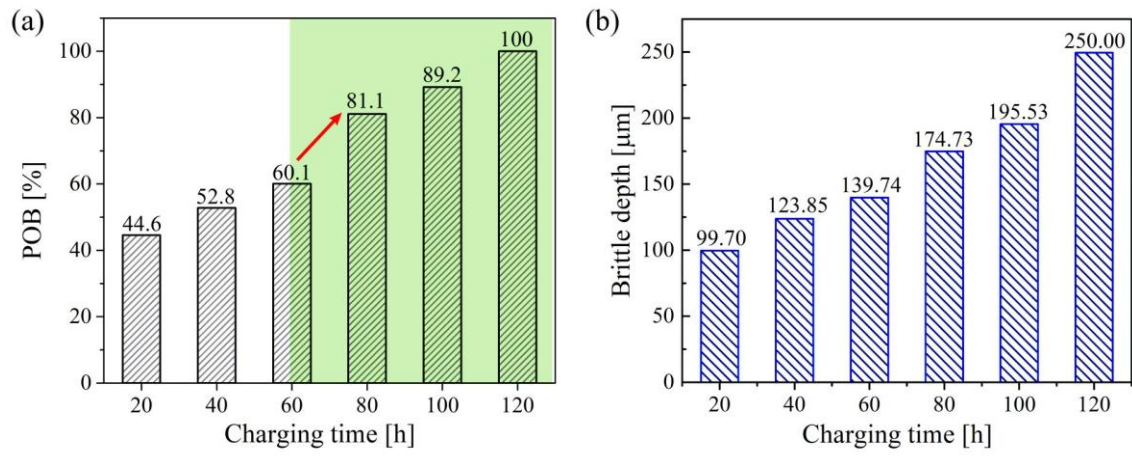


Fig. 7 Effect of hydrogen on (a) POB and (b) the corresponding brittle depth.

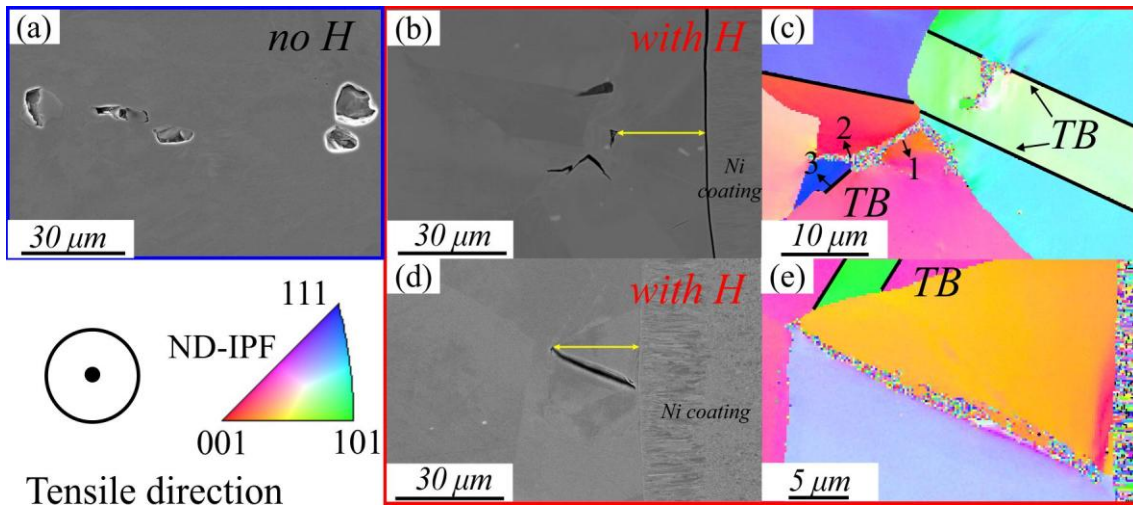


Fig. 8 Micrographs of the cross sections of (a) a hydrogen-free sample showing carbides cracks. (b)-(c) BSE images and ND-IPF maps, respectively, of a hydrogen-charged sample showing both intergranular and transgranular cracks. (d)-(e) BSE images and ND-IPF maps, respectively, of a hydrogen-charged sample showing only intergranular cracks, step size: 0.2 μm.

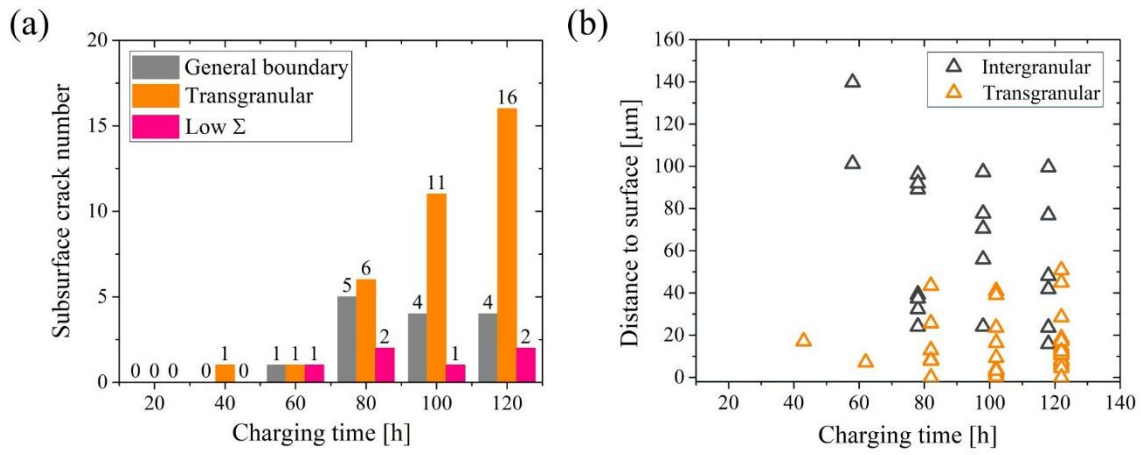


Fig. 9 Statistical analysis of subsurface cracks: (a) the number of different types of cracks on the gauge area (5mm^2) and (b) distance between surface and cracks.

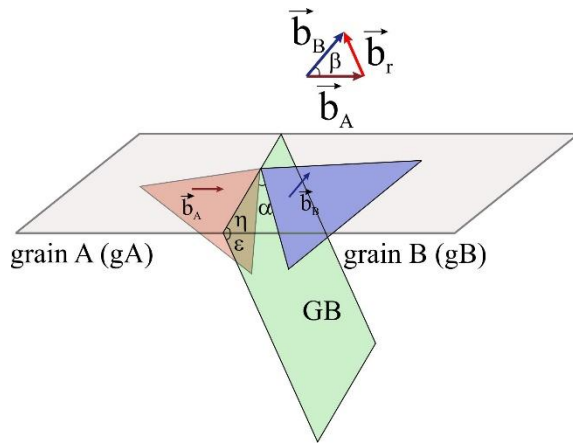


Fig. 10 Schematic of slip system coupling at a GB, together with GB traces, represented by angles (η , ϵ).

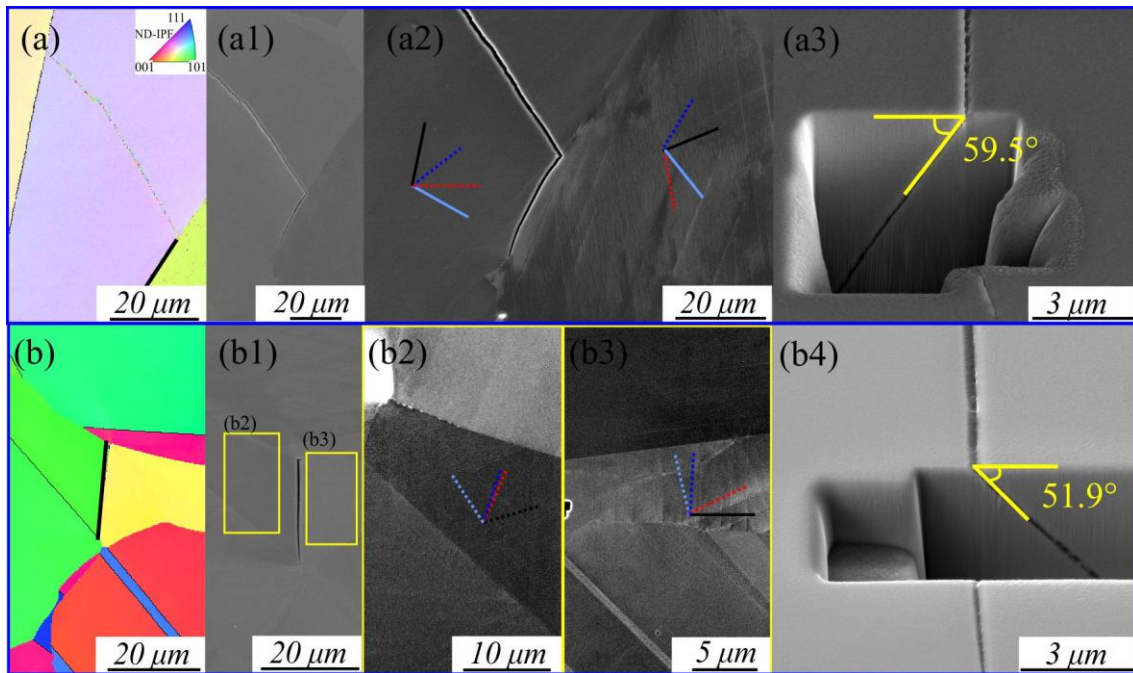


Fig. 11 ND-IPF maps (a) (b), and corresponding SE images (a1) and (b1) showing GB cracks. The ECC micrographs (a2) (b2)-(b3) indicate active slip lines that represent $\{111\}$ traces (the magnitude of Schmid factors varies from the highest to the lowest value: red, blue, light blue, black; the active slip planes are marked in dotted lines), (b2)-(b3) are magnified ECC images of areas in (b1) that are marked in yellow. (a3) (b4) The FIB cross sections show GB depth tilt angles.

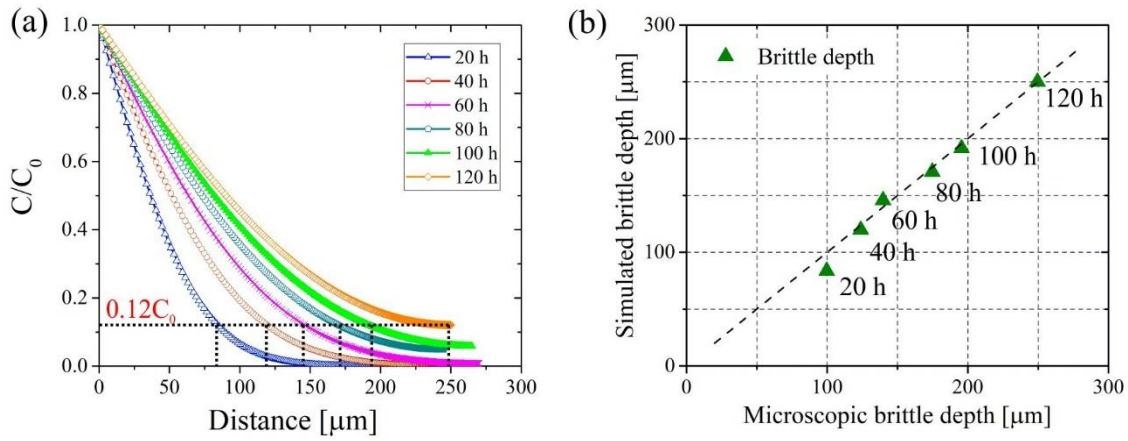


Fig. 12 (a) Hydrogen concentration profile for different charging time and (b) comparison of hydrogen effect on brittle depth measured from fracture surface and the simulated $0.12 C_0$ results.

Table 1 Chemical composition of the tested Alloy 718.

Element	C	Fe	Cr	Nb	Mo	Ti	Al	Co	Ni
wt. %	0.018	19.14	17.62	4.98	2.87	0.95	0.52	0.05	Balance

Table 2 Measured GB parameters, active slip planes with corresponding SFs, and GB resistance calculated from eq. (3). $(U\ V\ W)/\theta$ represents the misorientation axis and angle between adjacent grains.

C-GB	Euler angle ($\phi_1\ \Phi\ \phi_2$)	GB parameter ($U\ V\ W$)/ θ	η, ε	Active slip plane (hkl)	SF	$\alpha (\leq 15^\circ)$ ($h_a k_a l_a$)– ($h_b k_b l_b$)	Min. ω ($h_a k_a l_a$)– ($h_b k_b l_b$)
gA1	(12.1 60.7 287.2)	(-5 0 -1) /55.4	63.3, 59.5	(-1-11)	0.48	no	0.83
				(-111)	0.39		
gB1	(250.4 151.6 353.6)			(111)	0.48		(-1-11)–
				(-111)	0.47		(-111)
gA2	(194.8 27.9 33.1)	(1 -1 -1) /49.3	26.6, 90.0	(-111)	0.47	no	0.23
				(1-11)	0.40		
				(111)	0.39		
gB2	(98.4 46.5 340)			(1-11)	0.46		
				(-1-11)	0.40		
gA3	(308.9 24.5 98.6)	(-1 1 -1) /29.9	51.9 51.9	(111)	0.49	14.4	0.14
				(1-11)	0.48	(111)–	(1-11)–
				(-1-11)	0.31	(1-11)	(1-11)
gB3	(249.9 90.1 317.7)			(1-11)	0.43	5.5	
				(-111)	0.42	(1-11)–	
				(111)	0.05	(1-11)	
				(-1-11)	0.02		
NC-GB	Euler angle ($\phi_1\ \Phi\ \phi_2$)	GB parameter ($U\ V\ W$)/ θ	η, ε	Active slip plane	SF	$\alpha (\leq 15^\circ)$ ($h_a k_a l_a$)– ($h_b k_b l_b$)	Min. ω ($h_a k_a l_a$)– ($h_b k_b l_b$)
gA4	(12.1 60.7 287.2)	(-7 -1 0) /46.8	64.0, 88.8	(-1-11)	0.48	11.5	0.09
				(-111)	0.39	(-1-11)–	(-1-11)–
gB4	(59.4 158.5 266.0)			(111)	0.49	(111)	(111)
				(1-11)	0.49		
gA5	(225.6 87.2 302.0)	(7 1 0) /16.4	13.2, 89.0	(1-11)	0.48	1.1	0.01
				(-111)	0.47	(-111)–	(-111)–
gB5	(239.0 77.8 297.8)			(1-11)	0.49	(-111)	(-111)
				(-111)	0.42	12.2	
				(-1-11)	0.33	(1-11)– (-1-11)	
gA6	(12.1 60.7 287.2)	(-1 1 -1) /51.9	26.3, 76.4	(-1-11)	0.48	1.3	0.04
				(-111)	0.39	(-111)–	(-111)–
gB6	(239.0 77.8 297.8)			(1-11)	0.49	(1-11)	(1-11)
				(-111)	0.42	12.8	
				(-1-11)	0.33	(-111)– (-1-11)	

gA7	(308.9 24.5 98.6)	(0 -1 -1) /34.8	82.6, 44.2	(111)	0.49	1.8	0.1
				(1-11)	0.48	(111)-	(-1-11)-
				(-1-11)	0.31	(-1-11)	(-111)
gB7	(76.8 82.1 3.8)			(-111)	0.46	3.3	
				(111)	0.45	(-1-11)-	
				(-1-11)	0.42	(-111)	

