

A theoretical approach to predict the performance of chevron-type plate heat exchangers

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Abstract

Manufacturers of plate and frame heat exchangers nowadays mainly offer plates with chevron (or herringbone) corrugation patterns. The inclination angle φ of the crests and furrows of that sinusoidal pattern relative to the main flow direction has been shown to be the most important design parameter with respect to fluid friction and heat transfer. Two kinds of flow may exist in the gap between two plates (pressed together with the chevron pattern of the second plate turned into the opposite direction): the crossing flow of small substreams following the furrows of the first and the second plate, respectively, over the whole width of the corrugation pattern, dominating at lower inclination angles (lower pressure drop); and the wavy longitudinal flow between two vertical rows of contact points, prevailing at high φ angles (high pressure drop). The combined effects of the longer flow paths along the furrows, the crossing of the substreams, flow reversal at the edges of the chevron pattern, and the competition between crossing and longitudinal flow are taken into account to derive a relatively simple but physically reasonable equation for the friction factor ξ as a function of the angle φ and the Reynolds number Re . Heat-transfer coefficients are then obtained from a theoretical equation for developing thermal boundary layers in fully developed laminar or turbulent channel flow — the generalized L  v  que equation — predicting heat-transfer coefficients as being proportional to $(\xi \cdot Re^2)^{1/3}$. It is shown, by comparison, that this prediction is in good agreement with experimental observations quoted in the literature.

Keywords: Theoretical approach; Chevron-type plate heat exchangers; Performance prediction; Heat-transfer coefficients; Developing thermal boundary layers

1. Introduction

Plate and frame heat exchangers are offered by a large number of manufacturers as standard series production equipment over a wide range of sizes. They consist of a number of gasketed metal plates clamped between a stationary head and a follower plate by tie bolts. The principle, application and design characteristics of this very successful type of heat exchanger is explained in detail in relevant texts and handbooks [1–3]. Due to the great variety of possible corrugation patterns, and to the proprietary nature of the details of each particular design, the aim to provide users with reliable design equations for thermal and hydraulic performance of such equipment seemed to be hopeless. The fact, however, that during the last decades the chevron wave pattern had proved to be the most successful design offered in rather similar shapes by the majority of manufacturers has partly changed this situ-

ation. In the meantime, a number of detailed experimental studies, some of a more fundamental nature [4–7], using model corrugation patterns and systematically varying parameters like amplitude, wavelength, inclination angle and flow rate, and others with real industrial series heat exchangers [8–10], have produced a relatively large amount of interesting facts about heat transfer and pressure drop in plate heat exchangers. This wealth of detailed knowledge, however, has not yet been properly exploited to build up a generalized thermal and hydraulic design method for plate heat exchangers, comparable to the well-established methods for shell-and-tube heat exchangers. The following presents an entirely new approach to this problem: based on the very detailed experimental observations of Focke et al. [6] and Gaiser [7], the flow behaviour as known qualitatively from these sources is used to develop a relatively simple model to describe the effect of the inclination angle and the Reynolds number of fluid

$$\xi_0 = \frac{B_0}{\text{Re}} \quad \text{Re} < 2000 \text{ (laminar, Poiseuille)} \quad (6)$$

($\sin \varphi = 0$, $\varphi = 0^\circ$, straight longitudinal flow)

$$\xi_0 = (1.8 \lg \text{Re} - 1.5)^{-2} \quad \text{Re} \geq 2000 \text{ (turbulent, Konakov)} \quad (7)$$

The constant $B_0 = \xi_0 \cdot \text{Re}$ depends on the shape of the cross-section; thus $B_0 = 64$ for a circular tube. Focke et al. [6] have undertaken a numerical calculation to obtain B_0 for the ducts of sine cross-section (Fig. 2, top) and obtained (Eq. (A3) in [6]) $B_{0,\pi} = 53.39$ (which is only valid for $\Lambda/\hat{a} = 4$). Numerical values for other values of Λ/\hat{a} , as well as for the other limit $B_{0,0}$, are not known to date, although one might expect a value of magnitude in the order of 90 (the long rectangular cross-section would give $B_0 = 96$). For practical applications, a mean value between these two is recommended for use in Eq. (6). That for a circular tube may be chosen since no better calculations are available, hence:

$$B_0 = 64 \quad (8)$$

For turbulent flow Prandtl's well-known semi-empirical law, i.e. $1/\sqrt{\xi} = 2 \lg(\text{Re}\sqrt{\xi}) - 0.8$, implicit in ξ , may be very well approximated by an explicit expression of the type $1/\sqrt{\xi} = (2-n) \lg \text{Re} + \text{const.}$, where the 'constant' is $\lg(\xi \cdot \text{Re}^n) - 0.8$. Konakov's equation [Eq. (7)] is one of the simplest, and best, of these approximations. It is to be recommended for turbulent pipe flow in place of the similar Filonenko equation, i.e. $1/\sqrt{\xi} = 1.82 \lg \text{Re} - 1.64$, in the next edition of *VDI-Wärmeatlas*.

The other limiting case, i.e. $\varphi = \pi/2 (= 90^\circ)$, is the longitudinal wavy flow in a duct of rectangular cross-section. In this case, the shapes shown as cross-sections in Fig. 2 are longitudinal sections. If the corrugations were phase shifted by π so that the plates would have line contacts along the crests, the flow would be blocked [$\xi_{1,\pi} \rightarrow \infty$]. If the corrugations are in phase, the plates have no contact (Fig. 2, bottom) and the wavy duct has a friction factor $\xi_{1,0}$, i.e. much larger than for a straight duct. It is known that when flow separation occurs at $\text{Re} > 20$, vortices rotate in the vicinity of the outer extrema of the duct walls and the main flow follows a sinusoidal path with the same wavelength, but with a much smaller amplitude. Focke et al. [6] correlated their experimental data for this case as

$$\xi_{1,0} = \frac{B_1}{\text{Re}} + C_1 \quad \text{Re} < 2000 \text{ (laminar, with vortices)} \quad (9)$$

($\sin \varphi = 1$, $\varphi = 90^\circ$, wavy longitudinal flow)

$$\xi_{1,0} = \frac{K_1}{\text{Re}^n} \quad \text{Re} \geq 2000 \text{ (fully turbulent)} \quad (10)$$

The constants in Ref. [6] were given for ξ_e and Re_e with d_e as the characteristic length and hence have to be recalculated here (with $\Phi = 1.464$ as given in Ref. [6] for

the plates used with $\Lambda = 10$ mm and $\hat{a} = 2.5$ mm). Thus with $B_{1e} = 1280$, $C_{1e} = 5.63$, $K_{1e} = 63.8$, $n = 0.289$, from Eqs. (14,15) in Ref. [6], one obtains: $B_1 = B_{1e}/\Phi^2$, $C_1 = C_{1e}/\Phi$, $K_1 = K_{1e}/\Phi^{1+n}$, i.e. $B_1 = 597$, $C_1 = 3.85$, $K_1 = 39$, and $n = 0.289$.

The critical Reynolds number (with d_e) was given as $\text{Re}_e = 3000$ in Ref. [6], comparable to $\text{Re} = 2049$ rounded off in Eqs. (9,10) to 2000. These empirical equations are only valid for the geometrical parameters used in Ref. [6], i.e. $\Lambda/\hat{a} = 4$. They will certainly depend on \hat{a}/Λ , as may readily be seen from the fact that for $\hat{a}/\Lambda \rightarrow 0$, the straight rectangular duct would have $B_1 = 96$, $C_1 = 0$ and Eq. (7) for turbulent flow. The friction factor ξ_1 will certainly depend very sensitively on slight changes in phase between the two plates, and on slight changes in the plate distance caused by pressure differences. With the present state of knowledge, the calculation of $\xi_{1,0}$ from Eqs. (9,10) can only be regarded as a rough estimate and will always be rather uncertain. In practice, the constants B_1 , C_1 , K_1 and n might be used as fitting parameters if experimental data are available.

The range of inclination angles between these two limits, i.e. $0^\circ < \varphi < 90^\circ$, may be modelled in the following way. The flow path along the furrows relative to the vertical increases proportional to $1/\cos \varphi$, and hence the Reynolds-dependent friction factor ξ_0 has to be replaced by $\xi_0/\cos \varphi$. Additional friction losses occur due to flow reversal at the edges of the corrugation pattern and to crossing of the substreams. These two effects may be taken approximately into account by constant friction coefficients multiplied by the number of flow reversals, or the number of crossing points, respectively.

The number of flow reversals (or Back turns) is:

$$n_b = \frac{L_p}{d_h} \cdot \frac{d_h}{B} \cdot \tan \varphi \quad (11)$$

where B is the width of the corrugation pattern (see Fig. 1). The number n_b is thus proportional to L_p/d_h , so that the additional friction due to back turning of the flow may be simply added to $\xi_0/\cos \varphi$ as

$$\xi_b = b \tan \varphi \quad (12)$$

where

$$b = \xi_b \frac{d_h}{B} \quad (13)$$

The number of crossing points in a vertical line is:

$$n_c = \frac{L_p}{d_h} \cdot \frac{2d_h}{\Lambda} \cdot \sin \varphi \quad (14)$$

where Λ is the wavelength of the corrugation pattern (see Fig. 1). The number n_c is also proportional to L_p/d_h , so that the additional friction due to crossing may be added to $\xi_0/\cos \varphi$ (and ξ_b) too, as

$$\xi_c = c \sin \varphi \quad (15)$$

where

$$c = \xi_c \frac{2d_h}{\Lambda} \quad (16)$$

The total friction factor for crossing flow is therefore given by

$$\xi_{\text{crossing}} = b \tan \varphi + c \sin \varphi + \xi_0(\text{Re})/\cos \varphi \quad (17)$$

The corresponding friction factor for longitudinal wavy flow, $\xi_1(\text{Re})$, has to be somewhere between $\xi_{1,0}(\text{Re})$ from Eqs. (9,10) and $\xi_{1,\pi} (= \infty)$ and may be taken as $\xi_1 = a \cdot \xi_{1,0}(\text{Re})$ with the factor $a \geq 1$. The flow rates of the two kinds of flow, driven by a common pressure gradient, are proportional to their respective cross-sectional fractions: $\cos \varphi$ for the crossing flow and $(1 - \cos \varphi)$ for the longitudinal wavy flow. To a first approximation, they are inversely proportional to the square roots of their respective friction factors, which leads to the relatively simply model equation for $\xi = f(\varphi, \xi_0(\text{Re}), \xi_1(\text{Re}), b, c)$ with $\xi_1(\text{Re}) \approx a \cdot \xi_{1,0}(\text{Re})$ as:

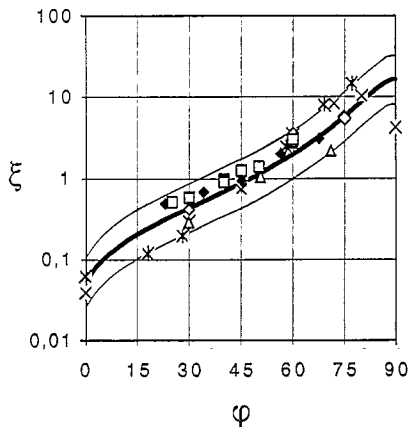


Fig. 3. Effect of the corrugation inclination angle φ on the pressure drop in plate heat exchangers. The friction factor $\xi(\text{Re} = 2000, \varphi)$ as a function of corrugation inclination angle φ is as follows: (open diamond): Okada et al. [4] (1972): model plates: $\varphi = 30^\circ, 45^\circ, 60^\circ, 75^\circ$ (here $\xi = \xi_{\text{model}}(45^\circ) \cdot \Delta p(\varphi)/\Delta p(45^\circ)$, ξ could not be calculated from Δp as in Ref. [4]); (cross): Focke et al. [6] (1985): model plates: $\varphi = 0^\circ, 30^\circ, 45^\circ, 60^\circ, 72^\circ, 80^\circ, 90^\circ$; (star) Gaiser [7] (1990): model plates: $\varphi = 18^\circ, 28^\circ, 45^\circ, 58^\circ, 69^\circ, 77^\circ$; (triangle): Bassiouny [9] (1985): industrial plates (Schmidt, Bretten): $\varphi = 29.75^\circ, (29.75^\circ \text{ and } 71^\circ), 71^\circ$; (square): Bond [8] (1981): diagram for industrial plates: $\varphi = 25^\circ, 30^\circ, 40^\circ, 45^\circ, 60^\circ$, and HEDH [3] (Taborek, 1988): diagram for industrial plates: $\varphi = 30^\circ, 40^\circ, 50^\circ, 60^\circ$; (filled diamond): Heavner et al. [10] (1993): industrial plates (APV): $\varphi = 23^\circ, (23^\circ \text{ and } 45^\circ), 45^\circ, (23^\circ \text{ and } 90^\circ), (45^\circ \text{ and } 90^\circ)$. Curves: model equation (18) for $\xi(\varphi, \text{Re})$ with the friction parameters ('standard set') $(a, b, c) = (3.8, 0.18, 0.36)$: upper curve, $2 \cdot \xi$; lower curve, $0.5 \cdot \xi$.

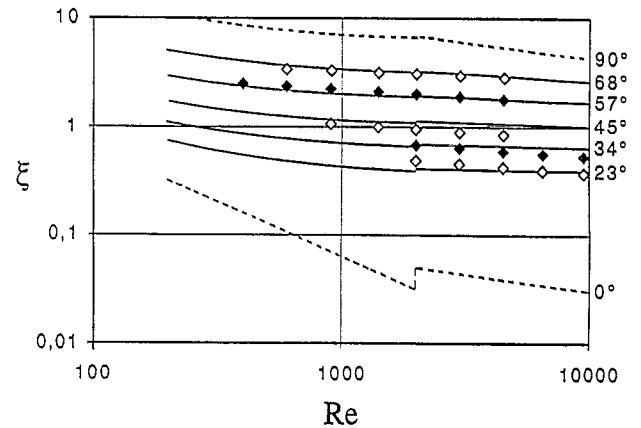


Fig. 4. Plots of the friction factor ξ versus Reynolds number Re with the inclination angle φ as a parameter. Curves calculated from the model equation (18) with (a, b, c) with (1.6, 0.40, 0.36). Symbols: empirical correlations by Heavner et al. (1993) representing their data for technical plates (the values for $\text{Re} = 2000$ are also shown in Fig. 3 as full diamonds).

$$\frac{1}{\sqrt{\xi}} = \frac{\cos \varphi}{\sqrt{b \tan \varphi + c \sin \varphi + \xi_0(\text{Re})/\cos \varphi}} + \frac{1 - \cos \varphi}{\sqrt{\xi_1(\text{Re})}} \quad (18)$$

Fig. 3 shows a comparison of experimentally obtained friction factors for a constant Reynolds number of 2000 (turbulent flow in nearly all cases) as taken from seven different sources with curves calculated from Eq. (18) using Eqs. (6,7) for $\xi_0(\text{Re})$ and Eqs. (9,10) for $\xi_{1,0}(\text{Re})$, while the friction parameters a , b and c have been used to fit the data. From these data it can be seen that a variation of the inclination angle φ from 0° to about 80° results in a change in the pressure drop over about 2.5 decades, i.e. a factor of 300 ($\approx 10^{2.5}$). Focke's experimental data for 90° , where the corrugation patterns of the two plates are exactly in phase, are considerably lower than the maximum. The curve from the model equation does not show this behaviour because ξ_1 in the model equation cannot necessarily be identified with this special case, i.e. $\xi_{1,0}$. To fit the data over the range of inclination angles from 0° to 80° , a ratio of ξ_1 to $\xi_{1,0}$ (i.e. the parameter a) of about 3.8 had to be chosen. The choice of a , however, does not significantly change the values of ξ for angles below 70° . With the exception of the special case of 90° , the model gives the correct trend for $\xi(\varphi)$. The relatively large individual deviation — the thinner lines show one-half and twice the values, respectively, of the mean correlation — are certainly due to the fact that the ratio of the corrugation parameters Λ/\hat{a} (which was 3.56, 4 and 5 for the model plates of Gaiser, Focke and Okada, respectively) will probably have been about twice as large as for the industrial plates used in the other investigations [3,8–10]. Only one source, i.e. Ref. [9], gives values of \hat{a} and Λ of 1.8 mm and 13.78 mm, respectively, i.e. $\Lambda/\hat{a} = 7.66$. Since the friction

¹ It should be mentioned that the empirical correlations [10] given in Table 1 result in (Fanning) friction factors f which are 17.2% higher than the corresponding lines through the data (see Ref. [10], in Fig. 3). The possible reason for this systematic discrepancy between the figure and the table might be the effect of not dividing the values in the table by the area enhancement factor Φ , leading to a friction factor defined with d_a according to Eq. (5) in place of d_h . The value of Φ is not given in Ref. [10], but 1.172 might well be a reasonable value for Φ for an industrial plate. The values shown as symbols in Fig. 4 (and in Fig. 3) have been calculated from the empirical correlations for f (with $\xi = 4f$) of Heavner et al. [10] divided by 1.172 in order to represent the data correctly.

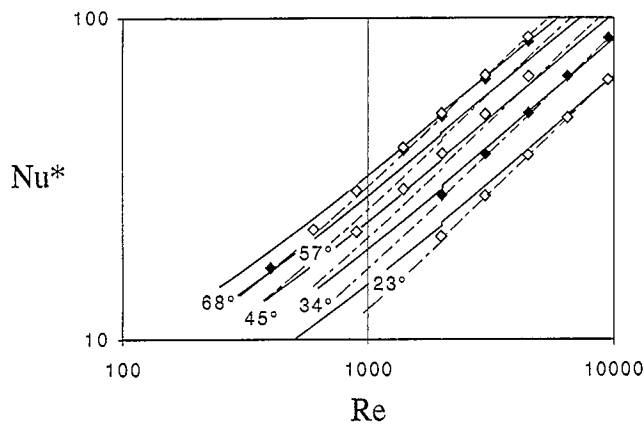


Fig. 7. $Nu^* = Nu \cdot Pr^{-(1/3)}(\eta/\eta_w)^{-(1/6)}$ versus Re with the inclination angle φ as a parameter. Full lines calculated from the theoretical L  v  que equation (19) with Eq. (20), and ξ from Eq. (18) using the standard set of friction parameters $(a, b, c) = (3.8, 0.18, 0.36)$ and $(d_h/\Lambda) = 0.21$. Symbols: empirical correlations by Heavner et al. [10] (1993) representing their data for technical plates (see Table 1); dotted lines: calculated from Eq. (28) (based on the L  v  que analogy, adapted to the experimental results) with ξ as above.

[3,5,6,8,10] or by $Pr^{0.4}$ [4,9] and, partly, by a viscosity ratio correction term $(\eta/\eta_w)^{0.17}$ [3,8,10]. [Some of the authors used the Colburn j -factor ($j = Nu \cdot Re^{-1} \cdot Pr^{-(1/3)}$, $j = c_f \cdot Re^{-p}$) or related groups in place of Nu ; in these cases the exponent m is obtained as $m = 1 - p$]. All the m values which were taken, as far as possible, from the same sources as used in the previous figures, were found to be between 0.55 and 0.72 (shown by the broken horizontal lines in Fig. 6) with the exception of a single much higher value $m = 0.868$ ($= 1 - 0.132$) obtained for straight longitudinal turbulent flow ($\varphi = 0^\circ$) by analogy to a tube flow equation in the paper by Focke et al. [6]. The circles indicate values $m_L = (2 - n)/3$, with n being the exponents in the friction factor formula of the same type as Eq. (10) taken from the empirical correlations of Focke et al. in the turbulent range. These values of m_L would be the theoretical value of m which follow from the generalized L  v  que equation. Except for $\varphi = 0^\circ$, they are indeed not far away from the empirically obtained values of m by the same authors (shown as the crosses X in Fig. 6). The full horizontal line is $m = 2/3$, i.e. the L  v  que exponent for a friction factor which does not depend on Re (m_L for $n = 0$). The average value of all the exponents is not far from $2/3$. Values above $2/3$, which would only be consistent with the L  v  que theory for negative values of n (or friction factors increasing with Re , which may indeed be sometimes found in small ranges of Reynolds numbers between laminar and turbulent flow) are mainly found for technical heat exchanger plates. There seems to be a slight trend to larger

values of m with lower inclination angles φ . This might be interpreted as the start of a transition from a thermally developing to a more and more developed heat transfer, finally leading to higher exponents between 0.8 and 0.9 for turbulent flow in straight ducts ($\varphi = 0^\circ$).

A more direct comparison between theory and experiment is shown in Fig. 7. Curves obtained from Eqs. (19) and (20) for $Nu^* = Nu/Pr^{1/3}$ (the full lines), with the friction factors ξ as calculated from Eq. (18) using the friction parameters (a, b, c) as in Fig. 3, are compared with correlation equations (the various symbols) representing the experimental data of Heavner et al. [10] for $Nu^* = Nu/[Pr^{1/3}(\eta/\eta_w)^{0.17}]$ as functions of the Reynolds number with the inclination angle as a parameter (see Table 1, columns 4 and 5 for the values of c_n and m from Ref. [10]). As the geometrical details of the plates used by these authors were not given in Ref. [10], the parameter d_h/Λ in Eq. (20) has been used to fit the theoretical curves to the experimental data. The resulting value $(d_h/\Lambda)_{fit} = 0.21$ is probably lower than the actual geometric one. A typical value of d_h/Λ for technical plates may be obtained from Bassiouny [9] for example: thus with $\Lambda/\hat{a} = 7.66$, $\Phi = 1.16$, $d_h/\Lambda = 0.45$ is obtained. Assuming that Heavner's plates had similar geometrical parameters, the ratio $(0.45/0.21)^{1/3} \approx 1.29$ would mean that the theoretical prediction (without fitting) might be about 30% higher in this case than the experimental result. Simple application of the L  v  que theory does indeed give the correct order of magnitude for the heat- or mass-transfer coefficients in chevron-type plate heat exchangers. (The dotted lines in this figure have been calculated from Eq. (28) as explained below in Section 5.)

4. On the analogy between heat, mass and momentum transfer

The generalized L  v  que equation (19) may be seen as a special form of the analogy between heat, mass and momentum transfer. The classical analogy has been established for fully developed temperature, concentration and velocity profiles in turbulent flow by Osborne Reynolds, Ludwig Prandtl and — following their routes — by many others. In general, the analogy in its various forms provides practically useful interrelations between the transport phenomena.

For ideal gases, the Reynolds analogy predicts a linear relationship between the heat- and the mass-transfer coefficient and the friction factor:

$$\frac{Nu}{Re \cdot Pr} = \frac{Sh}{Re \cdot Sc} = \frac{\xi}{8} \quad (Pr \approx Sc \approx 1) \quad (23)$$

The Prandtl analogy in its original or more refined modern versions shows that this simple proportionality [for $Pr (Sc) \neq 1$] has to be replaced by expressions such as:

$$\frac{Nu}{Re \cdot Pr} = \frac{(\xi/8)}{1 + c\sqrt{(\xi/8)(Pr^n - 1)}}; \quad \frac{Sh}{Re \cdot Sc} = \frac{(\xi/8)}{1 + c\sqrt{(\xi/8)(Sc^n - 1)}} \quad (24)$$

The linear proportionality between Nu (Sh) and ξ is therefore replaced by a proportionality to ξ^b , with the exponent varying between $b = 1$ [for $Pr (Sc) = 1$] and $b = 0.5$ [for $Pr (Sc) \rightarrow \infty$].

From the preceding sections [see Eq. (19)], it may follow that for the case of developing thermal (or diffusional) boundary layers in a developed velocity profile, the interrelation between Nu (Sh) and ξ may be even weaker, i.e. $b = 0.333$. It is suggested that this special case of the interrelation between heat, mass and momentum transfer be called the L  v  que analogy:

$$\frac{Nu}{Pr^{1/3}} = \frac{Sh}{Sc^{1/3}} = 0.404 \cdot (d/L)^{1/3} \cdot (\xi \cdot Re^2)^{1/3} \quad (25)$$

This may be easily adapted to correlate experimental data for plate heat exchangers by replacing the theoretical constant [including the geometrical parameter d_h/Λ from Eq. (20)] and the theoretical exponent $1/3$ if necessary by appropriate values obtained from experimental results.

5. Practical application of the L  v  que analogy

The most striking consequence of the fact that the ‘L  v  que analogy’ applies so nicely to plate heat exchangers may be seen from the dependency of the heat- and mass-transfer coefficients on the product $\xi \cdot Re^2$, which is directly proportional to the pressure drop Δp [see Eq. (1)]:

$$\xi \cdot Re^2 = \frac{2\Delta p d_h^3 \rho}{L_p \eta^2} \quad (26)$$

This indicates that the heat- and mass-transfer coefficients are independent (or virtually independent) of the individual values of the flow rate (Re) and inclination angle φ . The term $(d/L)^{1/3}$ in Eq. (20) is proportional to $[\sin(2\varphi)]^{1/3}$, which is unity for $\varphi = 45^\circ$ and deviates from this maximum value by less than 10% over the range of inclination angles from $25^\circ \leq \varphi \leq 65^\circ$, resulting in a rather weak individual dependency on φ . A plate heat exchanger with (‘soft’) ‘low phi’ plates (say with $\varphi = 30^\circ$) will have the same heat-transfer coefficient as a one with (‘hard’) ‘high phi’ plates ($\varphi = 60^\circ$, for example) if both are operated with the same pressure difference Δp ! The flow rates will differ, however, by

more than a factor of two. So the ‘high phi’ plates, with a given pressure drop, will produce higher numbers of transfer units (NTU) or higher temperature changes. This is a well known fact and the (nearly) ‘universal’ relationship between the heat-transfer coefficient and the pressure drop for plate heat exchangers has been empirically observed and documented, for example, in a figure showing the heat-transfer coefficient α versus pressure drop Δp for a water-to-water (313 K) application by Cooper and Usher [3] (see Fig. 2 in Section 3.7.10 of their report). This figure, which is said to show an “ $\alpha/\Delta p$ curve representative of plates in general, where α in this case is equal to $2U$, thereby incorporating an approximate allowance for metal resistance”, is an essentially straight line in a log–log plot from $\alpha_{\text{overall}}(\Delta p = 0.1 \text{ bar}) \approx 6300 \text{ W m}^{-2} \text{ K}^{-1}$ via $\alpha_{\text{overall}}(\Delta p = 1.0 \text{ bar}) \approx 12600 \text{ W m}^{-2} \text{ K}^{-1}$ (water, 313 K) to $\alpha_{\text{overall}}(\Delta p = 1.6 \text{ bar}) \approx 14500 \text{ W m}^{-2} \text{ K}^{-1}$. In the form of an equation, this curve may be expressed approximately as $(10^{0.30} = 2)$:

$$\alpha_{\text{overall}} \approx 12600 \text{ W m}^{-2} \text{ K}^{-1} \times (\Delta p / 1 \text{ bar})^{0.30} \quad (\text{water, 313 K}) \quad (27)$$

(The subscript ‘overall’ here means: $\alpha_{\text{overall}} = 2/(2/\alpha + (s/\lambda)_{\text{metal}})$, i.e. this ‘ α ’ is twice the overall heat-transfer coefficient U as stated by Cooper and Usher [3], see above).

The L  v  que analogy would predict an exponent of 0.333 for α as a function of Δp . Empirical evidence from various sources with technical plates would lead to a slightly higher exponent (see Figs. 6 and 7). Using the empirical equations of Heavner et al. [10] for the friction factors ($f = K \cdot Re^{-n}$) and the Nusselt numbers ($Nu^* = c_n \cdot Re^m$) respectively (see Table 1), covering inclination angles from 23° to 68° , one obtains an average exponent of $q = [m/(2-n)]_{\text{average}} = 0.374$ (with maximum deviations of -4.7% to $+5.6\%$) in place of $q = 0.333$ from the direct application of the L  v  que theory. A good semi-empirical equation, representing the heat-transfer data by Heavner et al. [10] together with their data on pressure drop [or with Eq. (18)] is obtained from replacing Re in $Nu^* = c_n \cdot Re^m$ by the value obtained from solving the equation $\xi \cdot Re^2 \sin(2\varphi) = x = c_x \cdot Re^{2-n}$ for the Reynolds number on the right-hand side, i.e. $Re = (x/c_x)^{1/(2-n)}$. So, finally, the empirical equations of Heavner et al. are rewritten in the form $Nu^* = c_q [\xi \cdot Re^2 \sin(2\varphi)]^q$, with the adapted exponents q and constants c_q obtained from the corresponding friction factor correlations as listed in Table 1. Taking the arithmetic mean of the five values of q and the geometric mean of the five constants c_q (which is the appropriate averaging for a set of power laws), one finally obtains a practically useful semi-empirical equation for heat transfer in technical plates, based on the L  v  que analogy and on experimental evidence:

$$\text{Nu} = 0.122\text{Pr}^{1/3}(\eta/\eta_w)^{1/6} \cdot [\xi \cdot \text{Re}^2 \sin(2\varphi)]^{0.374} \quad (28)$$

This equation when used with the model equation (18) for $\xi(\text{Re}, \varphi)$ and the standard set of friction parameters (a, b, c) = (3.8, 0.18, 0.36) in fact represents the heat transfer data by Heavner et al. [10] somewhat better than the original L  v  que equation. This is shown by the dotted curves in Fig. 7.

Eq. (28) may easily be rewritten in terms of Δp , with Eq. (26). Using the physical properties of water at 313 K, as given by Cooper and Usher [3], i.e. $\rho = 1000 \text{ kg m}^{-3}$, $C_p = 4.2 \times 10^3 \text{ J kg}^{-1} \text{ K}^{-1}$, $\eta = 0.65 \times 10^{-3} \text{ Pa s}$ and $\lambda = 0.63 \text{ W K}^{-1} \text{ m}^{-1}$, together with $L_p = 1 \text{ m}$ (or rather Δp replaced by a pressure gradient $\Delta p/L_p$ in bar m^{-1}), $d_h = 4 \text{ mm}$ (the value of d_h has a weak influence here, with a power of only 0.122) and the property ratio correction $(\eta/\eta_w)^{1/6}$ put equal to one, finally gives:

$\alpha_{\text{water}, 313 \text{ K}}$

$$= 19\,677[(\Delta p/1 \text{ bar}) \sin(2\varphi)]^{0.374} \text{ W m}^{-2} \text{ K}^{-1} \quad (29)$$

Using angles of $\varphi = 30^\circ$ and 60° for example [$\sin(2 \times 30^\circ) = \sin(2 \times 60^\circ) = 0.866$], one obtains $\alpha_{\text{water}, 313 \text{ K}} = 18\,646(\Delta p/1 \text{ bar})^{0.374} \text{ W m}^{-2} \text{ K}^{-1}$ and allowing for a typical stainless-steel wall resistance with a conductivity of $\lambda_w = 15 \text{ W K}^{-1} \text{ m}^{-1}$ and a wall thickness of $s_w = 0.75 \text{ mm}$, i.e. $(1/20\,000) \text{ m}^2 \text{K W}^{-1}$, gives $\alpha_{\text{overall}} = 2/[2/\alpha + (s/\lambda)_w]$:

$$\left. \begin{aligned} \alpha_{\text{overall}}(\Delta p = 0.1 \text{ bar}) &= 6590 \text{ W m}^{-2} \text{ K}^{-1} \\ \alpha_{\text{overall}}(\Delta p = 1.0 \text{ bar}) &= 12\,700 \text{ W m}^{-2} \text{ K}^{-1} \\ \alpha_{\text{overall}}(\Delta p = 1.6 \text{ bar}) &= 14\,300 \text{ W m}^{-2} \text{ K}^{-1} \end{aligned} \right\}$$

[water, 313 K, from Eq. (28)]

which are indeed pretty close to the values of 6300, 12 600 and 14 500 $\text{W m}^{-2} \text{ K}^{-1}$ respectively from the above-mentioned “ $\alpha/\Delta p$ curve representative of plates in general” from the *Heat Exchanger Design Handbook* [3].

The practically important result that heat- and mass-transfer coefficients in plate heat exchangers depend essentially on the pressure gradient (or on the product $\xi \cdot \text{Re}^2$), but not separately on both the friction factor $\xi(\varphi, \text{Re})$ and the flow rate (Re), seems to date to have been based only on experience. It may now be understood from the application of theory.

6. Conclusions

On the basis of present knowledge, Eq. (18) and (28) may be recommended for obtaining the friction factors ξ and the heat-transfer coefficients α typically found in technical plate heat exchangers directly as a function of the corrugation inclination angle φ , the Reynolds num-

ber Re , or alternatively as a function of the pressure drop ($\xi \cdot \text{Re}^2$) [see Eq. (26)].

More detailed comparison with the original data should be carried out within the near future in order to test and further improve this practically useful design method based on a physically reasonable flow model for $\xi(\varphi, \text{Re})$ and on a special form of analogy between momentum, heat and mass transfer found from generalizing the L  v  que theory for turbulent flow. The latter idea has been discussed earlier [12] as a more or less academic example of a rigorous theoretical equation that may be applied for turbulent heat transfer. It has been shown, at least for chevron-type plate heat exchangers, that this theory is not only of academic value but is in fact directly applicable for solving practical engineering problems.

7. Nomenclature

a, b, c	friction parameters in Eq. (18), ‘standard set’: (3.8, 0.18, 0.36), –
\hat{a}	amplitude of corrugation (see Fig. 1), m
B	width of corrugation pattern (see Fig. 1), m
B_0, B_1	constants in Eq. (6) and (9), –
c_n, c_q, c_x	constants defined in Table 1, –
C_p	specific heat capacity at constant pressure, $\text{J kg}^{-1} \text{ K}^{-1}$
d_e	equivalent diameter, Eq. (5), $d_e = 4\hat{a}$, m
d_h	hydraulic diameter, Eq. (2), $d_h = 4\hat{a}/\Phi$, m
f	Fanning friction factor, $f = \xi/4$, –
j	Colburn j -factor, $j = \text{Nu} \cdot \text{Re}^{-1} \cdot \text{Pr}^{-(1/3)}$, –
K	constant in Eq. (10), –
L	length between two crossing points, Eq. (20), m
L_p	plate length (see Fig. 1), m
m	exponent in Eq. (22), –
n	exponents in Eqs. (10) and (24), numbers, –
Nu	Nusselt number = $\alpha d_h/\lambda$, –
Nu^*	Nusselt group = $\text{Nu} \cdot \text{Pr}^{-(1/3)} \cdot (\eta/\eta_w)^{-(1/6)}$, –
Pr	Prandtl number = $\eta C_p/\lambda$, –
q	exponent in generalized L��v��que analogy (Table 1), –
Re	Reynolds number = $\rho u d_h/\eta$, –
s	thickness, m
u	flow velocity, m s^{-1}
U	overall heat-transfer coefficient, $\text{W m}^{-2} \text{ K}^{-1}$
X	corrugation parameter = $2\pi\hat{a}/\Lambda$, –
α	heat-transfer coefficient, $\text{W m}^{-2} \text{ K}^{-1}$
α^*	normalized heat- (or mass-) transfer coefficient = $\alpha(\varphi)/\alpha(45^\circ)$, –

